HIGH ORDER SUBSYNCHRONOUS RESONANCE MODELSAND MULTI-MODE S'TABILIZATION
by
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## ABSTRACT

Subsynchronous resonance (SSR) occurs in a series-capacitorcompensated power system when a mechanical mass-spring mode coincides with that of the electrical system. In this thesis, a complete high order model including mass-spring system, series-capacitor-compensated transmission line, synchronous generator, turbines and governors, exciter and voltage regulator is derived. Eigenvalue analysis is used to find the effect of capacitor compensation, conventional lead-lag stabilizer, loading and dampers on SSR. Finally, controllers are designed to stabilize multimode subsynchronous resonance simultaneously over a wide range of capacitor compensation.
ABSTRACT ..... ii
TABLE OF CONTENTS ..... iii
LIST OF TABLES ..... v
LIST OF ILLUSTRATIONS ..... vi
ACKNOWLEDGEMENT ..... vii
NOMENCLATURE ..... viii

1. INTRODUCTION ..... 1
1.1 Subsynchronous Resonance ..... 1
1.2 The Scope of the Thesis ..... 2
2. A COMPLETE POWER SYSTEM MODEL FOR SUBSYNCHRONOUS RESONANCE STUDIES ..... 4
2.1 Introduction ..... 4
2.2 The Steam Turbines and Generator Multi-Mass Torsional System ..... 4
2.3 The Turbine Torques and Speed Governor ..... 8
2.4 The Synchronous Generator ..... 10
2.5 The Exciter and Voltage Regulator ..... 14
2.6 State Equations for the Complete System ..... 16
3. EIGENVALUE ANALYSIS OF THE SSR MODEL ..... 18
3.1 Introduction ..... 18
3.2 The Effect of Capacitor Compensation ..... 18
3.3 The Effect of Conventional Stabilizer ..... 18
3.4 The Effect of Loading ..... 19
3.5 The Effect of Dampers ..... 19
4. MULTI-MODE TORSIONAL OSCILLATIONS STABILIZATION WITH LINEAR OPTIMAL CONTROL ..... 29
4.1 State Equations with Measurable Variables ..... 29
4.2 State Equations in Canonical Form ..... 31
4.3 Linear Optimal Control Design ..... 34

## Page

### 4.4 Stabilization of SSR <br> 35

5. CONCLUSIONS ..... 42
REFERENCES ..... 43
Table Page
3-1 Data for SSR Mode1 ..... 20
3-2 Eigenvalues of SSR model at different degrees of capaci- tor compensation without conventional stabilizer ..... 21
3-3 Effect of damper winding for zero total reactance ..... 28
4-1 Eigenvalues of original system and reduced order models without controller at $30 \%$ compensation ..... 37
4-2 Eigenvalues of reduced 22nd order model with/without controller and original system with the controller at $30 \%$ compensation ..... 38
4-3 Eigenvalues of reduced 22nd order model with/without controller and original system with the controller at 50\% compensation ..... 39
4-4 Eigenvalues of reduced 19 th order model with/without controller and original system with the controller at $30 \%$ compensation ..... 40
Figure Page
2-1 A functional block diagram of the complete system for subsynchronous resonance studies ..... 6
2-2 Mechanical mass and shaft system ..... 7
2-3 Torques of a mass-shaft system ..... 7
2-4 A speed governor model for the steam turbine system ..... 8
2-5 A linear model of the steam turbine system ..... 9
2-6 A synchronous machine model ..... 10
2-7 A single line representation of the transmission line ..... 12
2-8 Exciter and voltage regulator mode1 ..... 14
2-9 A supplementary excitation control ..... 15
3-1 The effect of capacitor compensation without stabilizer. ..... 22
3-2 Enlarged portion of Fig. 3-1 ..... 23
3-3 The effect of capacitor compensation with stabilizer ..... 24
3-4 Enlarged portion of Fig. 3-3 ..... 25
3-5 The effect of loading without stabilizer ..... 26
3-6 The effect of loading with stabilizer ..... 27
4-1 The effect of capacitor compensation with controller ..... 41

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=NOMENCLATURE
```


## General

| X | state vector of unmeasurable model |
| :---: | :---: |
| Z | state vector of measurable model |
| Y | state vector of canonical |
| A | system matrix of X -mode1 |
| F | system matrix of Z -model |
| $\mathrm{F}_{0}$ | system matrix of Y-model |
| B | control matrix of $X$-mode 1 |
| G 。 | control matrix of Z -model |
| $G_{0}$ | control matrix of Y -mode1 |
| U | control vector |
| M | transformation matrix for Z -mode1 |
| T | transformation matrix for Y-model |
| $\lambda$ | eigenvalue |
| j | complex operator, $\sqrt{-1}$ |

## Mass-Spring System

M inertia coefficient $=2 H$
H inertia constant
K shaft stiffness

D damping
$\theta \quad$ rotor angle
$\omega \quad$ rotor speed
$\omega_{0} \quad$ synchronous speed

## Synchronous Machine

i instantaneous value of current
$V$ instantaneous value of voltage
$\psi \quad$ flux-1inkage
R resistance
$X$ reactance
$\delta \quad$ torque angle, rad.
$\omega \quad$ angular velocity, rad./s
Te electrical torque
$i_{t} \quad$ terminal voltage
$P+j Q \quad$ generator output power
Transmission Network
$X_{t}, R_{t} \quad$ reactance and resistance of transformer
$X_{e}, R_{e} \quad$ reactance and resistance of the line
$X_{c} \quad$ reactance of capacitor
Vo infinite bus voltage
Exciter and Voltage Regulator
$\mathrm{K}_{\mathrm{A}} \quad$ regulator gain
$\mathrm{T}_{\mathrm{A}} \quad$ regulator time constant, s
$T_{E} \quad$ exciter time constant, $s$
$V_{r e f . ~ r e f e r e n c e ~ v o l t a g e ~}^{\text {en }}$
Governor and Turbine System
$\mathrm{K}_{\mathrm{g}} \quad$ actuator gain
$\mathrm{T}_{1}, \mathrm{~T}_{2}$ actuator time constant
$\mathrm{T}_{3}$ servomotor time constant
a change in actuator signal
$P_{G V} \quad$ power at gate outlet
$\mathrm{T}_{\mathrm{CH}} \quad$ steam chest time constant
$\mathrm{T}_{\mathrm{RH}} \quad$ reheater time constant
$\mathrm{T}_{\mathrm{CO}} \quad$ cross-over time constant

| $\mathrm{F}_{\mathrm{HP}}$ | high pressure turbine power fraction |
| :---: | :---: |
| $\mathrm{F}_{\text {IP }}$ | intermediate pressure turbine power fraction |
| $\mathrm{F}_{\mathrm{LP} 1}$ | low pressure turbine 1 power fraction |
| ${ }^{\mathrm{F}} \mathrm{LP} 2$ | low pressure turbine 2 power fraction |
| $\mathrm{T}_{\mathrm{HP}}$ | high pressure turbine torque |
| $\mathrm{T}_{\text {IP }}$ | intermediate pressure turbine torque |
| $\mathrm{T}_{\text {LP 1 }}, \mathrm{T}_{\text {LP2 }}$ | low pressure turbine torque |
| Subscripts |  |
| d, q | direct- and quadrature-axis stator quantities |
| f | field circuit quantities |
| $D, Q, G$ | direct- and quadrature-axis damper quantities |
| c | quantities associate with capacitor |
| a | armature phase quantities |
| Superscripts |  |
| -1 | inverse of a matrix |
| $t$ | transpose of a matrix |
| - | differential operator |
| Prefix |  |
| $\Delta$ | linearized quantities |
| s | differential operator |
| p | differential operator |

## 1. INTRODUCTION

### 1.1 Subsynchronous Resonance [1]

To increase the power transfer capability of a power system, the use of series-capacitor-compensated transmission lines is the best alternative to the addition of transmission lines because of environmental considerations and the limited availability of right-of-way. They are also more economical than other methods such as HVDC. However, subsynchronous resonance (SSR) may occur and shaft damage may result. Two turbine shafts were severely damaged [2] at the Mohave generating station of the Southern California Edison Company because of the excessive torsional oscillations caused by interaction between the electrical resonance of the series-capacitor-compensated system and the natural modes of the multi-mass generator turbine mechanical system.

Subsynchronous resonance may occur in a system in the steadystate or transient state due to a system fault or major switching. The former may be called the steady-state subsynchronous resonance and the latter the transient subsynchronous resonance. The main problems are the self excitation, the torsional interaction, and the transient torques [3]. When SSR occurs, the synchronous machine is selfexcited and behaves like an induction generator. If the negative resis- ${ }^{-}$ tance of the machine, as an induction generator, exceeds the total resistance of the external electrical system, self excitation of SSR occurs.

Torsional oscillation is due to the mechanical modes of the multi-mass turbine-generator system. The torsional frequencies are in the subsynchronous range. If the electrical resonant frequency is equal or close to a torsional mode, the rotor oscillations and the induced voltages will build up and the interaction between the electrical and
mechanical systems ensues [1,4].
Transient torques are caused by system disturbances on a series-capacitor-compensated line and the energy stored in the series capacitor produces large subsynchronous currents in the lines. When the frequency of the current coincides with the natural torsional frequency, transient torque results.

After the reported turbine shaft failures [3], corrective measures have been proposed. Some of them are under serious consideration and others already put into practice. Without too much modification to the existing system, the simplestway to avoid the subsynchronous resonance is to reduce the degree of capacitor compensation. Another suggestion is the installation of passive filter units in series with the generator transformer neutral at the high voltage side. Each filter unit is a high-Q parallel resonant circuit tuned to block the subsynchronous current at a particular frequency corresponding to one of the mechanical modes. Additional amortisseur windings on the pole faces can reduce the effective negative resistance [5]. Supplementary excitation control is being considered and the stabilizing signals are derived from rotor speed. Finally a subsynchronous overcurrent relay has been developed for the automatic protection of generating units in case of sustained subsynchronous oscillations.

### 1.2 Scope of the thesis

The widely accepted method for subsynchronous resonance studies in engineering practice consists of a two-step analysis [5]. The electrical and mechanical modes are determined separately. The transient elec... trioal torque from the electrical system is calculated first and then applied to the mechanical system as a forcing function. In this thesis, a complete model including the electrical, mechanical and control systems
will be developed and presented in Chapter 2. By using eigenvalue analysis, the effect of various degrees of compensation, loading conditions and conventional supplementary excitation control on subsynchronous resonance will be examined in Chapter 3. For broad-band frequency multimode subsynchronous resonance control, linear optimal controllers will be designed in Chapter 4. A summary of all important results and conclusions will be presented in Chapter 5.
2. A COMPLETE POWER SYSTEM MODEL FOR SUBSYNCHRONOUS RESONANCE STUDIES

### 2.1 Introduction

For any dynamic or transient stability study of a power system, an accurate model of the system is required. In addition to the individual efforts $[2,7,12]$, a benchmark model has been proposed by the IEEE Subsynchronous Resonance Working Group for SSR studies [20]. In this chapter, a complete subsynchronous resonance model is presented, including steam turbines and generator múlti-mass torsional system, the turbine torques and speed governor, the synchronous generator, the capacitor-compensated transmission lines, and the exciter and voltage regulator. A functional block diagram of the complete system is shown in Fig. 2-1.

### 2.2 The Steam Turbines and Generator Multi-Mass Torsional System

Assume that the steam turbine-generator set consists of one high-pressure steam turbine, one intermediate-pressure turbine, two lowpressure turbines, one generator rotor and one exciter, all mechanically coupled on the same shaft as shown in Fig. 2-2. They comprise a six-mass torsional system. For the purpose of analysis [13], they are considered to have concentrated masses and to be coupled by shafts of negligible mass and known torsional stiffness. Each mass is denoted by a circular disc, as in Fig. 2-3, with an inertia constant $M_{i}$, a positive torsional torque $K_{i}\left(\theta_{i+1}-\theta_{i}\right)$ on the left and a negative torque $-K_{i-1}\left(\theta_{i}-\theta_{i-1}\right)$ on the right. There is an external torque $\mathrm{T}_{\mathrm{i}}$ applied to the mass inna positive direction, an accelerating torque $M_{i} \dot{\omega}_{i}$ in the same direction and a damping torque $D_{i} \omega_{i}$ in the opposite direction. The net accelerating torque becomes

$$
\begin{equation*}
M_{i} \dot{\omega}_{i}=T_{i}-D_{i} \omega_{i}+K_{i}\left(\theta_{i+1}-\theta_{i}\right)-K_{i-1}\left(\theta_{i}-\theta_{i-1}\right) \tag{2-1}
\end{equation*}
$$

where $\quad M_{i}=$ the inertia constant of $i^{\text {th }}$ rotor

$$
\theta_{i}=\text { the rotational displacement for } i^{\text {th }} \text { rotor }
$$

$D_{i}=$ damping coefficient for $i^{\text {th }}$ rotor
$K_{i, i+1}=$ the torsional stiffness of the shaft between the $i^{\text {th }}$ rotor and the $i+1^{\text {th }}$ rotor

By applying equation (2-1) to the six mass turbine-generator system, twelve differential equations are obtained:

High Pressure $\mathrm{p}_{6}=\frac{\mathrm{K}_{56}}{\mathrm{M}_{6}} \theta_{5}-\frac{\mathrm{K}_{56}}{\mathrm{M}_{6}} \theta_{6}-\frac{\mathrm{D}_{66}}{\mathrm{M}_{6}} \hat{\omega}_{6}+\frac{\mathrm{T}_{H P}}{\mathrm{M}_{6}}$

$$
\begin{equation*}
p \theta_{6}=\omega_{6} \omega 0^{-} \tag{2-3}
\end{equation*}
$$



$$
\begin{equation*}
p \theta_{5}=\omega_{5} \omega 0 \tag{2-5}
\end{equation*}
$$

Low Pressure $1 \mathrm{p} \ddot{\omega}_{4}=\frac{\mathrm{K}_{45}}{\mathrm{M}_{4}} \theta_{5}-\frac{\left(\mathrm{K}_{34}+\mathrm{K}_{45}\right)}{\mathrm{M}_{4}} \theta_{4}+\frac{\mathrm{K}_{34}}{\mathrm{M}_{4}} \theta_{3}-\frac{\mathrm{D}_{44}}{\mathrm{M}_{4}} \omega_{4}+\frac{\mathrm{T}_{\mathrm{LP}}}{\mathrm{M}_{4}}$ (2-6)

$$
\begin{equation*}
p \theta_{4}=\omega_{4} \omega 0 \tag{2-7}
\end{equation*}
$$

$$
\begin{equation*}
p \omega=\frac{K_{23}}{M_{2}} \theta_{3}-\frac{\left(K_{12}+K_{23}\right)}{M_{2}} \delta+\frac{K_{12}}{M_{2}} \theta_{1}-\frac{\mathrm{D}_{22}}{M_{2}} \omega-\frac{\mathrm{Te}}{\mathrm{M}_{2}} \tag{2-10}
\end{equation*}
$$

Generator

$$
\begin{equation*}
p \theta_{3}=\omega_{3} \omega o \tag{2-9}
\end{equation*}
$$

$$
\begin{equation*}
p \delta=\omega \omega 0 \tag{2-11}
\end{equation*}
$$



Fig. 2-1 A functional block diagram of the complete system for subsynchronous resonance studies.

The generator has an electric torque output Te , and the exciter electric torque is neglected. Note that while angles are in radians, the speed is in p.u.;

$$
\omega o=1 \text { p.u. }=377 \text { electrical radian } / \text { second }
$$

| High | Intermediate | Low | Low | Generator | Exciter |
| :---: | :---: | :---: | :---: | :---: | :---: |
| pressure | pressure | pressure 1 | pressure 2 |  |  |
| $\theta_{6}, \omega_{6}$ | $\theta$ | ,$\omega_{5}$ | $\theta_{4}, \omega_{4}$ | $\theta_{3}, \omega_{3}$ | $\delta, \omega$ |
| $\theta_{1}, \omega_{1}$ |  |  |  |  |  |



Fig. 2-2 Mechanical mass and shaft system


Fig. 2-3 Torques of a mass-shaft system

### 2.3 The Turbine Torques and Speed Governor

The steam turbine and speed governor representation is based on an IEEE committee report [14]. Usually the speed is sensed between the lowpressure turbine and the generator rotor. Combined with the speed reference, the speed deviation or error signal is derived and relayed through the actuator to activate the servomotor, which in turn opens or closes the steam valves. A block diagram [14] is shown in Fig. 244. Forca linear study, the system equations may be written;

$$
\begin{align*}
& \mathrm{p} \Delta \mathrm{a}=\frac{\mathrm{K}_{\mathrm{g}}}{\mathrm{~T}_{1}} \Delta \omega-\frac{1}{\mathrm{~T}_{1}} \Delta \mathrm{a}  \tag{2-14}\\
& \mathrm{p} \Delta \mathrm{P}_{\mathrm{GV}}=\frac{1}{\mathrm{~T}_{3}} \Delta \mathrm{a}-\frac{1}{\mathrm{~T}_{3}} \Delta \mathrm{P}_{\mathrm{GV}} \tag{2-15}
\end{align*}
$$



Fig. 2-4 A speed governor model for the steam turbine system


Fig. 2-5 A Linear model of the steam turbine system
Fig. 2-5 shows a standard turbine representation for stability studies [14]. The system consists of one high-pressure, one intermediatepressure and two low-pressure turbines. Their output torques are denoted by $T_{H P}, T_{I P}, T_{L P 1} 1^{1}, T_{L P 2}$, respectively. There is a reheater between highpressure and intermediate-pressure stages, and crossover pipings between intermediate-pressure and low-pressure stages. The steam into the turbines flow through the governor-controlled valves at the inlet of the steam chest. The time constants of the steam chest, the reheater and the crossover piping are denoted by $\mathrm{T}_{\mathrm{CH}}, \mathrm{T}_{\mathrm{RH}}$, and $\mathrm{T}_{\mathrm{CO}}$, respectively. $\mathrm{F}_{\mathrm{HP}}, \mathrm{F}_{\mathrm{IP}}$, $\mathrm{F}_{\mathrm{LP} 1}$, and $\mathrm{F}_{\mathrm{LP} 2}$ represent fractions of the total power developed in the various stages. Therefore

$$
\begin{align*}
& \mathrm{p} \Delta \mathrm{~T}_{\mathrm{HP}}=\frac{\mathrm{F}_{\mathrm{HP}}}{\mathrm{~T}_{\mathrm{CH}}} \Delta \mathrm{P}_{\mathrm{GV}}-\frac{1}{\mathrm{~T}_{\mathrm{CH}}} \Delta \mathrm{~T}_{\mathrm{HP}}  \tag{2-16}\\
& \mathrm{p} \Delta \mathrm{~T}_{\mathrm{IP}}=\frac{\mathrm{F}_{\mathrm{IP}}}{\mathrm{~F}_{\mathrm{HP}} \times \mathrm{T}_{\mathrm{RH}}} \Delta \mathrm{~T}_{\mathrm{HP}}-\frac{1}{-\mathrm{T}_{\mathrm{RH}}} \Delta \mathrm{TT} \mathrm{IP}  \tag{2-17}\\
& \mathrm{p} \Delta \mathrm{~T}_{\mathrm{LP} 1}=\frac{\mathrm{F}_{\mathrm{LP} 1}}{\mathrm{~F}_{\mathrm{TPP}} \times \mathrm{T}_{\mathrm{CO}}} \Delta \mathrm{~T}_{\mathrm{IP}}-\frac{1}{\mathrm{~T}_{\mathrm{CO}}} \Delta \mathrm{~T}_{\mathrm{LP} 1} \tag{2-18}
\end{align*}
$$

$$
\begin{equation*}
\Delta \mathrm{T}_{\mathrm{PL} 2}=\frac{\mathrm{F}_{\mathrm{LP} 2}}{\mathrm{~F}_{\mathrm{LP} 1}} \Delta \mathrm{~T}_{\mathrm{LP} 1} \tag{2-19}
\end{equation*}
$$

### 2.4 The Synchronous Generator

The synchronous generator is assumed to have six windings. In addition to the d and q armature windings on the respective axes, there is a field winding $f$, a damper winding $D$ on the d-axis and two damper windings $Q$ and $G$ on the $q$-axis. They are schematically shown in Fig. 2-6.


Fig. 2-6 A synchronous machine model

The voltage equations in the linear form are

$$
\begin{align*}
\Delta \mathrm{V}_{\mathrm{d}} & =\mathrm{p} \Delta \psi_{\mathrm{d}}-\omega o \Delta \psi_{\mathrm{q}}-\psi_{\mathrm{qo}} \Delta \omega-\mathrm{R}_{\mathrm{a}} \Delta i_{\mathrm{d}} \\
\Delta \mathrm{~V}_{\mathrm{q}} & =\mathrm{p} \Delta \psi_{\mathrm{q}}+\omega \Delta \Delta \psi_{\mathrm{d}}+\psi_{\mathrm{do}} \Delta \omega-\mathrm{R}_{\mathrm{a}} \Delta \mathrm{i}_{\mathrm{q}} \\
\Delta \mathrm{~V}_{\mathrm{f}} & =\mathrm{p} \Delta \psi_{\mathrm{f}}+\mathrm{R}_{\mathrm{f}} \Delta \mathbf{i}_{\mathrm{f}} \\
0 & =\mathrm{p} \Delta \psi_{\mathrm{D}}+\mathrm{R}_{\mathrm{D}} \Delta i_{\mathrm{D}} \\
0 & =\mathrm{p} \Delta \psi_{\mathrm{Q}}+\mathrm{R}_{\mathrm{Q}} \Delta i_{\mathrm{Q}} \\
0 & =\mathrm{p} \Delta \psi_{\mathrm{G}}+\mathrm{R}_{\mathrm{G}} \Delta i_{\mathrm{G}} \tag{2-20}
\end{align*}
$$

where the flux linkages are
anand the $\psi^{\prime} s, X^{\prime} s, R^{\prime} s$ and $i^{\prime} s$ are the per unit flux linkages, reactances, resistances and currents respectively.

The saturation in the iron circuit is neglected. The stator transient voltages $\mathrm{p} \psi_{\mathrm{d}}$ an $\mathrm{p} \psi_{\mathrm{q}}$, although normally neglected in stability studies $[15,16]$ are retained in this study because the capacitor compensated transmission lines, to which the armature windings are connected in series, must be described by differential equations.


Fig. 2-7 A single line representation of the transmission line

In Fig. $2-7, V_{d}$ and $V_{q}$ are $d-q$ components of the terminal voltage $\mathrm{V}_{\mathrm{c}}$ is the voltage across the capacitor and $\mathrm{V}_{\mathrm{ct}}$ is the terminal voltage at the capacitor. The transformer is represented by a reactance $X_{t}$ and a resistance $R_{t}$ and the transmission line by a reactance $X_{e}$ and a line resistance $R_{e}$.

Let the terminal voltage equations in $\mathrm{a}-\mathrm{b}-\mathrm{c}$ phase coordinates be

$$
\begin{align*}
{\left[V_{t}\right]_{a, b, c}=} & {\left.[R]_{\left[I_{t}\right.}\right]_{a, b, c}+[L] \frac{d}{d t}\left[I_{t}\right]_{a, b, b}+\left[V_{c}\right]_{a, b, c} } \\
& +\left[V_{o}\right]_{a, b, c} \tag{2-22}
\end{align*}
$$

$$
\text { where } \begin{aligned}
{[R] } & =\text { a resistance matrix: } R_{t}+R_{e} \\
{[L] } & =\text { an inductance matrix: } \frac{X_{t}+X_{e}}{\omega O}
\end{aligned}
$$

Let Park's transformation matrix be

$$
[\mathrm{T}]=\begin{gather*}
\mathrm{a}  \tag{2-23}\\
\mathrm{a} \\
\mathrm{c}
\end{gather*}\left[\begin{array}{ccc}
\cos \theta & -\sin \theta & 1 \\
\cos (\theta-120) & -\sin (\theta-120) & 1 \\
\cos (\theta+120) & -\sin (\theta+120) & 1
\end{array}\right]
$$

and the transformations are

$$
\begin{equation*}
[\mathrm{V}]_{\mathrm{a}, \mathrm{~b}, \mathrm{c}}=[\mathrm{T}][\mathrm{V}]_{\mathrm{d}, \mathrm{q}, \mathrm{o}} \text { and }[\mathrm{I}]_{\mathrm{a}, \mathrm{~b}, \mathrm{c}}=[\mathrm{T}][\mathrm{I}]_{\mathrm{d}, \mathrm{q}, \mathrm{o}} \tag{2-24}
\end{equation*}
$$

Then we have

$$
\begin{gather*}
{[\mathrm{V}]_{\mathrm{d}, \mathrm{q}, \mathrm{o}}=[\mathrm{R}]_{[\mathrm{I}]_{\mathrm{d}, \mathrm{q}, \mathrm{o}}}+[\mathrm{L}]_{\mathrm{dt}}^{\mathrm{d}}[\mathrm{I}]_{\mathrm{d}, \mathrm{q}, \mathrm{o}}+[\mathrm{L}]_{[\mathrm{T}]^{-1} \frac{\mathrm{~d}}{\mathrm{dt}}[\mathrm{~T}] \cdot[\mathrm{I}]_{\mathrm{d}, \mathrm{q}, \mathrm{o}}}} \\
+\left[\mathrm{V}_{\mathrm{c}}\right]_{\mathrm{d}, \mathrm{q}, \mathrm{o}}+\left[\mathrm{V}_{\mathrm{o}}\right]_{\mathrm{d}, \mathrm{q}, \mathrm{o}} \tag{2-25}
\end{gather*}
$$

Note that

$$
[T]^{-1} \frac{\mathrm{~d}}{\mathrm{dt}}[\mathrm{~T}]=\left[\begin{array}{ccc}
0 & -1 & 0  \tag{2-26}\\
1 & 0 & 0 \\
0 & 0 & 1
\end{array}\right] \quad \frac{\mathrm{d} \theta}{\mathrm{dt}}
$$

The terminal voltage equation in $d-q$ coordinates when linearized, becomes
where $V_{c d}$ and $V_{c q}$ are the $d-q$ components of the voltage across the capacitor, and $V_{0}$ the infinite bus voltage. The zero component equation is orthogonal to the other two equations and is usually neglected except for asymmetric loading. The capacitor equations may be written

$$
\begin{equation*}
[I]_{a, b, c}=[C] \frac{d}{d t}\left[V_{c}\right]_{a, b, c} \tag{2-28}
\end{equation*}
$$

After transformation, it becomes

$$
\begin{equation*}
[I]_{d, q, o}=[C] \frac{d}{d t}\left[V_{c}\right]_{d, q, o}+[C][T]^{-1} \frac{d}{d t}[T]\left[V_{c}\right]_{d, q, o} \tag{2-29}
\end{equation*}
$$

which when linearized, gives

$$
\left[\begin{array}{c}
\Delta I_{d}  \tag{2-30}\\
\Delta I_{q}
\end{array}\right]=\frac{1}{\omega_{o} X_{c}}\left[\begin{array}{c}
\Delta V_{c d} \\
\Delta V_{c q}
\end{array}\right]+\frac{1}{X_{c}}\left[\begin{array}{c}
-\Delta V_{c q} \\
\Delta V_{c d}
\end{array}\right]
$$

### 2.5 The Exciter and Voltage Regulator

The exciter and voltage regulator mode1 in this thesis is based on an IEEE committee report [18] with some simplification. The regulator input filter time constant, the saturation function and the stabilizing feedback loop are neglected.


Fig. 2-8 Exciter and Voltage Regulator Model

In Fig. 2-8, $V_{t}$ is the generator terminal voltage,
$\mathrm{U}_{\mathrm{E}}$ the supplementary control, $\mathrm{K}_{\mathrm{A}}$ the voltage regulator gain, $\mathrm{T}_{\mathrm{A}}$ its time constant, $\mathrm{T}_{\mathrm{E}}$ the exciter time constant and $\mathrm{E}_{\mathrm{FD}}$ a per unit output voltage of the exciter. Although the voltage limits are shown in the figure, they will be neglected in linear analysis. Mathematically we have

$$
\begin{align*}
& p \Delta V_{R}=\frac{K_{A}}{T_{A}} \Delta V_{t}+\frac{K_{A}}{T_{A}} u-\frac{1}{T_{A}} \Delta V_{R}  \tag{2-31}\\
& p \Delta E_{F D}=\frac{1}{T_{E}} \Delta V_{R}-\frac{1}{T_{E}} \Delta E_{F D} \tag{2-32}
\end{align*}
$$

where the linearized terminal voltage

$$
\begin{equation*}
\Delta \mathrm{V}_{\mathrm{t}}=\frac{\mathrm{V}_{\mathrm{do}}}{\mathrm{~V}_{\text {to }}} \Delta \mathrm{V}_{\mathrm{d}}+\frac{\mathrm{v}_{\mathrm{qo}}}{\mathrm{v}_{\text {to }}} \Delta \mathrm{v}_{\mathrm{q}} \tag{2-33}
\end{equation*}
$$

Substituting $\Delta V_{\mathrm{d}} \Delta V_{\mathrm{q}}$ from equation (2-27) into equation (2-33) and the results into equation (2-31), we have

$$
\begin{align*}
p \Delta V_{R}= & \frac{V_{d o} K_{A}\left(X_{t}+X_{e}\right)}{V_{t o} T_{A} \omega 0} p \Delta i_{d}+\frac{V_{q o} K_{A}\left(X_{t}+X_{e}\right)}{V_{t o} T_{A}{ }^{\omega \omega 0}} p \Delta i_{q}+\frac{K_{A} V_{d o}}{T_{A} V_{t o}} \Delta V_{c d} \\
& +\frac{K_{A} V_{q o}}{T_{A} V_{t o}} \Delta V_{c q}+\frac{K_{A}}{T_{A} V_{t o}}\left[V_{d o}\left(R_{t}+R_{e}\right)+V_{q o}\left(X_{t}+X_{e}\right)\right] \Delta i_{d} \\
& +\frac{K_{A}}{T_{A} V_{t o}}\left[-V_{d o}\left(X_{t}+X_{e}\right)+V_{q o}\left(R_{t}+R_{e}\right)\right] \Delta i_{q}+\frac{K_{A}}{T_{A}} \Delta u-\frac{1}{T_{A}} \Delta V_{R} \\
& +\frac{K_{A} V_{o}}{T_{A} V_{t o}}\left[V_{d o} \cos \delta_{o}-V_{q o} \sin \delta_{o}\right] \Delta \delta \tag{2-34}
\end{align*}
$$

Fig. 2-9 shows a supplementary excitation control of the leadlag compensation type [19].


Fig. 2-9 A supplementary excitation control

Mathematically,

$$
\begin{align*}
& \mathrm{K}_{\mathrm{S}} \mathrm{~T} \mathrm{p} \Delta \dot{\omega}-\mathrm{T} p \mathrm{~b}=\mathrm{b}  \tag{2-35}\\
& \mathrm{~T}_{\mathrm{x}} \mathrm{p} \cdot \mathrm{~b}-\mathrm{T}_{\mathrm{y}} \mathrm{p} \mathrm{c}=\mathrm{c}-\mathrm{b}  \tag{2-36}\\
& -\mathrm{T}_{\mathrm{y}} \mathrm{p} \mathrm{u}+\mathrm{T}_{\mathrm{x}} \mathrm{p} \mathrm{c}=\mathrm{u}-\mathrm{c} \tag{2-37}
\end{align*}
$$

### 2.6 State Equations for the Complete System

The component system equations previously derived can be combined into a single set of state equations in the form of

$$
\begin{equation*}
\underline{\dot{x}}=[\mathrm{A}] \underline{\mathrm{x}} \tag{2-38}
\end{equation*}
$$

where $\underline{X}$ is the state variable vector and [A] the system matrix. Equation (2-38) can be conveniently partitioned

$$
\left[\begin{array}{c}
\dot{x}_{I}  \tag{2-39}\\
\hdashline \dot{X}_{I I}
\end{array}\right]=\left[\begin{array}{c:c}
A_{I,}, & A_{I,} I I \\
\hdashline A_{I I, I} & A_{I I, I I}
\end{array}\right]\left[\begin{array}{l}
X_{I} \\
\hdashline X_{I I}
\end{array}\right]
$$

where $X_{I}$ contains the state variables of the mechanical system and $X_{I I}$ those of the electrical system; namely
$X_{I}=\left[\omega_{1}, \theta_{1.1}, \omega, \delta, \omega_{3}, \theta_{3}, \omega_{4}, \theta_{4}, \omega_{5}, \theta_{5}, \omega_{6}, \theta_{6}, a, P_{G V}, T_{H P}, T_{I P}, T_{L P 1}\right]$
$X_{I I}=\left[i_{d}, i_{q}, i_{f}, i_{D}, i_{Q}, i_{G}, v_{c d}, V_{c q}, V_{R}, E_{F D}\right]$
$A_{\text {I, II }}$ represents the coupling between the two systems where the interaction occurs through the electrical torque $T_{e}$. Since

$$
\begin{align*}
& T_{e}=\left(\psi_{d} i_{q}-\psi_{q} i_{d}\right) \quad \text { per unit }  \tag{2-40}\\
& \Delta \mathrm{T}_{\mathrm{e}}=\left\{\left(\psi_{\mathrm{do}} \Delta \mathrm{i}_{\mathrm{q}}+\mathrm{i}_{\mathrm{qo}} \Delta \psi_{\mathrm{d}}-\psi_{\mathrm{qo}} \Delta \mathrm{i}_{\mathrm{d}}-\mathrm{i}_{\mathrm{do}} \Delta \psi_{\mathrm{q}}\right) \hat{k}\right\} / \omega_{\mathrm{o}} \\
& =\left\{\left(X_{q}-X_{d}\right) i_{q o} \Delta i_{d}+\left[\left(X_{q}-X_{d}\right) i_{d o}+X_{a d} i_{f o}^{f}\right] \Delta i_{q}+i_{q o} X_{a d} \Delta i_{f}\right. \\
& \left.+i_{q o} X_{a d} \Delta i_{D}-i_{d o} X_{a q} \Delta i_{Q}-i_{d o} X_{a q} \Delta i_{G}\right\} / \Delta \omega_{o} \tag{2-41}
\end{align*}
$$

Next, the partitioned matrices $A_{I I, I}$ and $A_{I I, I I}$ of the electrical system shall be first assembled in the form of

$$
\begin{equation*}
B \dot{X}_{\overline{\bar{I} I}}=C_{I} X_{I}+C_{I I} X_{I I} \tag{2-42}
\end{equation*}
$$

Then we have

$$
\begin{equation*}
\dot{X}_{I I}=B^{-1} C_{I} X_{I}+B^{-1} C_{I I} X_{I I} \tag{2-43}
\end{equation*}
$$

or

$$
\dot{X}_{I I}=A_{I I, I} X_{I}+A_{I I, I I} X_{I I}
$$

where

$$
\begin{equation*}
A_{I I, I}=B^{-1} C_{I} \quad ; \quad A_{I I, I I}=B^{-1} C_{I I} \tag{2-44}
\end{equation*}
$$

Thus we have completed the derivation of the state equations for the overall system.

## 3. EIGENVALUE ANALYSIS OF THE SSR MODEL

### 3.1 Introduction

Eigenvalue analysis technique is useful in investigating the stability of systems. The complex eigenvalues are associated with oscillatory modes of the system and the reall part of the eigenvalues provide the information on system damping. When an eigenvalue has a positive real part, instability of the system is indicated.

In this thesis the effect of capacitor compensation, conventional stabilizer and dampers will be investigated using the data taken from the benchmark model [20] Table 3-1.

### 3.2 The Effect of Capacitor Compensation

Fig. 3-1 and 3-2 show the eigenvalues of the system with various degrees of capacitor compensation, at a particular loading. The pair of eigenvalues corresponding to $\Delta \delta$ and $\Delta \omega$ of the synchronous machine have positive real parts when the compensation is $20 \%$ or less. The natural frequencies of the multi-mass torsional system are approximately 298, 203, 160,127 and 99 radians/second which correspond to $47.4,32.3,25.5,20.2$, and 16 Hz respectively. By changing the degree of compensation, the natural oscillation frequency of the transmission system changes. When the frequency of the electrical mode is closedto a mechanical mode, SSR may occur. At $50 \%$ and $60 \%$ compensation, two mechanical modes are excited simultaneously.

### 3.3 The Effect of Conventional Stabilizer

Fig. 3-3 and 3-4 repeat the study of the effectoof capacitor compensation, but with the addition of a conventional stabilizer of the lead-lag type. Whereas the pair of eigenvalues corresponding to $\Delta \delta$ and
$\Delta \omega$ of the synchronous machine were unstable for compensation below $30 \%$, Fig. 3-1 shows they are substantially moved to the left half of the complex plane with the supplementary excitation control, Fig. 3-3. However, the lowest mechanical mode of $99 \mathrm{rad} / \mathrm{sec}$ is always excited and shifted to the right-half plane. This is in agreement with other findings [21, 22]. The damping of other mechanical modes is decreased slightly.

### 3.4 The Effect of Loading

The effect of different loading with and without stabilizer on SSR is shown in Fig. 3-5 and 3-6 respectively. Most of the eigenvalues do not change except those corresponding to the generator mechanical mode. Generally the system becomes more unstable with more leading power factor. Most utilities operate their systems between 0.9 power factor lagging and unity power factor. For this reason, 0.9 power factor lagging is chosen for the studies here.

### 3.5 The Effect of Dampers

As reported [5] the addition of an amortisseur winding can reduce the possibility of SSR. For this investigation, the additional damper effect is represented by decreased damper impedance. When the total reactance of line and transformer is zero, SSR occurs. The result is shown in Table 3-2. The excited mode is damped out by decreasing the damper impedance which agrees with previous results [5].

Table 3-1 Numerical Values of Mode1 in p.u. system

## Mass-spring System Parameters

$$
\begin{array}{lll}
M_{1}=0.068433 & K_{12}=2.822 & D_{11}=0.1 \\
M_{2}=1.736990 & K_{23}=70.858 & D_{22}=0.1 \\
M_{3}=1.768430 & K_{34}=52.038 & D_{33}=0.1 \\
M_{4}=1.717340 & K_{45}=34.929 & D_{44}=0.1 \\
M_{5}=0.311 .178 & K_{56}=19.303 & D_{55}=0.1 \\
M_{6}=0.185794 & & D_{66}=0.1
\end{array}
$$

## Synchronous Machine Parameters

$$
\begin{array}{lll}
\mathrm{X}_{\mathrm{d}}=1.79 & \mathrm{X}_{\mathrm{f}}=1.6999 & \mathrm{R}_{\mathrm{f}}=0.00105 \\
\mathrm{X}_{\mathrm{ad}}=1.66 & \mathrm{X}_{\mathrm{D}}=1.6657 & \mathrm{R}_{\mathrm{D}}=0.00371 \\
\mathrm{X}_{\mathrm{q}}=1.71 & \mathrm{X}_{\mathrm{Q}}=1.6845 & \mathrm{R}_{\mathrm{Q}}=0.00526 \\
\mathrm{X}_{\mathrm{aq}}=1.58 & \mathrm{X}_{\mathrm{G}}=1.8250 & \mathrm{R}_{\mathrm{G}}=0.01820 \\
\mathrm{R}_{\mathrm{a}}=0.0015 & &
\end{array}
$$

Exciter and Voltage Regulator

$$
\mathrm{K}_{\mathrm{A}}=50 \quad \mathrm{~T}_{\mathrm{E}}=0.002 \quad \mathrm{~T}_{\mathrm{A}}=0.01
$$

Transmission Line Parameters

$$
\begin{array}{lll}
x_{t}=0.14 & R_{t}=0.01 & x_{e}=0.56 \\
R_{e}=0.02 & x_{c} \text { varies from } 0.056-0.56 \\
& & (10 \%-100 \%)
\end{array}
$$

Governing and Turbine System

$$
\begin{array}{lll}
\mathrm{K}_{\mathrm{g}}=25 & \mathrm{~T}_{1}=0.2 & \mathrm{~T}_{2}=0 \\
\mathrm{~T}_{3}=0.3 & \mathrm{~T}_{\mathrm{CH}}=0.3 & \mathrm{~T}_{\mathrm{RH}}=7.0 \\
\mathrm{~T}_{\mathrm{CO}}=0.2 & \mathrm{~F}_{\mathrm{HP}}=0.3 & \mathrm{~F}_{\mathrm{IP}}=0.26 \\
\mathrm{~F}_{\mathrm{LP} 1}=0.22 & \mathrm{~F}_{\mathrm{LP} 2}=0.22 & \mathrm{~F}
\end{array}
$$

## Stabilizer parameters

$$
\begin{array}{lll}
\mathrm{K}_{\mathrm{s}}=20 & \mathrm{~T}=3.0 & \mathrm{~T}_{\mathrm{x}}=0.125 \\
\mathrm{~T}_{\mathrm{y}}=00.05 &
\end{array}
$$

|  | -0.1817 $\pm$ j298.18 | -0.1818 $\pm$ j298.18 | -0.1818 $\pm \mathrm{j} 298.18$ |
| :---: | :---: | :---: | :---: |
|  | -0.2104 $\pm$ j203.20 | +0.1541 $\pm$ j204.35 | +0.1560 $\pm$ j202.68 |
| Shaft modes | -0.2266 $\pm$ j160.66 | -0.2496 $\pm$ j160.72 | +0.9100 $\pm \mathrm{j} 161.42$ |
|  | -0.6679 $\pm$ j127.03 | -0.6706 $\pm \mathrm{j} 127.03$ | -0.6799 $\pm \mathrm{j} 127.08$ |
|  | -0.2660 $\pm 99.13$ | -0.2877 $\pm$ j 99.21 | -0.3545 $\pm 99.79$ |
| Stator/Network | -6.9800 $\pm \mathrm{j} 512.30$ | -7.0224 $\pm$ j542.80 | $-7.0800 \pm j 591.15$ |
|  | -6.0717 $\pm$ j241.01 | -6.1984 $\pm$ j209.20 | -6.8387 $\pm$ j161.47 |
|  | -8.5681 | -8.4404 | -8.1277 |
| Synchronous | -31.578 | -31.920 | -32.808 |
| Machine Rotor | -25.397 | -25.404 | -25.423 |
|  | -2.0196 | -1.9830 | -1.9070 |
| Exciter and | -499.98 | -499.97 | -499.97 |
| Voltage Regulator | -101.97 | -101.91 | -101.76 |
| $\lambda \delta \omega$ | +0.0415 $\pm \mathrm{j} 8.0234$ | $-0.0479 \pm j 8.4801$ | -0.2674 $\pm \mathrm{j} 9.5459$ |
|  | -0.1416 | -0.1417 | -0.1418 |
| Turbine and | -4.6679 | -4.6160 | -4.0496 |
| Governor | -2.9271 | -3.0336 | -3.3335 |
|  | -4.7039 $\pm \mathrm{j} 0.7567$ | -4.6732 $\pm$ j0.6269 | $-4.7939 \pm j 0.3198$ |

Table 3.2 Eigenvalues of SSR model at different degrees of capacitor compensation without conventional stabilizer for $\mathrm{P}=0.9$ p.u. at 0.9 power factor lagging.


Fig. 3-1. The effect of capacitor compensation without stabilizer for $P=0.9$ p.u. at 0.9 power factor lagging. (The symbols $1,2,3, \ldots 9$ respectively correspond to $10,20,30, \ldots 90 \%$ compensation)


Fig. 3-2 Enlarged portion of Fig. 3-1.
(The symbols $1,2,3, \ldots 9$ respectively correspond to $10,20,30, \ldots 90 \%$ compensation)


Fig. 3-3. The effect of capacitor compensation with stabilizer for $P=0.9$ p.u. at 0.9 power factor 1agging. (The symbols $1,2,3 \ldots 9$ respectively correspond to $10,20,30 \ldots 90 \%$ compensation)


Fig. 3-4 Enlarged portion of Fig. 3-3
(The symbols $1,2,3, \ldots 9$ respectively correspond to $10,20,30 \ldots 90 \%$ compensation)


Fig. 3-5 The effect of loading without stabilizer
(The symbols $\underset{X}{\mathbb{D}}=0.8$ p.f. leading, $\mathbb{A}=0.9 \mathrm{p} . \mathrm{f}$. leading, $+\dot{=}$ unity power factor
$X=0.9$ p.f. lagging, $\Delta=0.8$ p.f. lagging)


Fig. 3-6 The effect of loading with stabilizer
(The symbols $\mathbb{O}=0.8$ p.f. leading, $\Delta=0.9$ p.f. leading; $+=$ unity power factor $X=0.9$ p.f. lagging, $\diamond=0.8$ p.f. lagging)
original system
damper
impedance x 0.6
damper impedance x 1.5

| Shaft modes | $-0.1818 \pm$ j298.18 | -0.1818 $\pm$ j298.18 | -0.1818 $\pm$ j298.18 |
| :---: | :---: | :---: | :---: |
|  | -0.0288 $\pm$ j 202.87 | -0.0296 $\pm$ j202.87 | -0.0278 $\pm \mathrm{j} 202.87$ |
|  | $-0.1536 \pm j 160.52$ | $-0.1543 \pm j 160.52$ | -0.1528 $\pm$ j 160.52 |
|  | $-0.6521 \pm j 126.98$ | -0.6522 $\pm$ j126.98 | -0.6518 $\pm$ j126.98 |
|  | $-0.0238 \pm$ j 98.47 | -0.0285 $\pm$ j 98.50 | -0.0163 ${ }_{\text {圭 j } 98.42}$ |
| Stator/Network | $-7.1913 \pm j 715.78$ | $-7.1841 \pm j 718.50$ | $-7.1873 \pm j 712.71$ |
|  | +0.3604 $\pm$ j 37.21 | -0.2149 $\pm$ j 30.47 | +0.9064 $\pm$ j 42.09 |
|  | -5.8109 | -5.1370 | -7.2552 |
| Synchronous | -43.39 | -31.63 | -55.88 |
| Machine Rotor | -25.60 | -25.51 | -25.72 |
|  | -0.5040 | -0.4187 | -0.5597 |
| Exciter and | -499.96 | -499.98 | -499.94 |
| Voltage Regulator | -100.68 | -100.43 | -100.95 |
| $\lambda \delta \omega$ | $-3.8086 \pm$ j 20.07 | $-4.1469 \pm$ j 23.81 | $-3.7030 \pm$ j 18.24 |
|  | -0.1406 | -0.1404 | -0.1407 |
| Turbine | -4.1296 | -4.8816 | -3.8373 |
| Governor | -3.1202 | -3.0225 | -3.1684 |
|  | $-4.5440 \pm j 0.1525$ | -3.7438 $\pm \mathrm{j}-.5651$ | $-4.7916 \pm j-.2688$ |

Table 3.3 Effect of Damper Winding in zero total reactance and $\underset{P}{P}=0.9$ p.u. at 0.9 power factor lagging.
4. MULTI-MODE TORSIONAL OSCILLATIONS STABILIZATION WITH LINEAR OPTIMAL CONTROL

Linear optimal control theory has been applied to the stabilizer design of power systems $[23,8,24]$. For practical applications, the state variables used in the design must be measurable. Another problem of optimal control design is the choice of the weighting matirices $Q$ and $R$ in the cost index. A simple procedure was proposed [8] which requires thesstate equations in the canonical form. The $Q / R$ ratio in the procedure can be judiciously chosen.

### 4.1 State Equations With Measurable Variables

The state equations of the system was written in Chapter 2 in the form

$$
\begin{equation*}
\dot{\mathrm{X}}=\mathrm{AX}+\mathrm{Bu} \tag{4-1}
\end{equation*}
$$

where A was given as (2-39). For an excitation control,

$$
B=\left[\begin{array}{llllll}
0 & 0 & 0 & \ldots & 0 & \frac{K_{A}}{T_{A}} \tag{4-2}
\end{array}\right]^{\frac{\mathrm{t}}{\mathrm{t}}}
$$

as in (2-31). Let

$$
\begin{equation*}
Z=M X \tag{4-3}
\end{equation*}
$$

where $Z$ is the measurable variable vector. Then

$$
\begin{align*}
& \dot{Z}=M X=M A M^{-1} Z+M B u \\
& \dot{Z}=F Z+G u \tag{4-4}
\end{align*}
$$

where
$F=M A M^{-1}$ and $G=M B$
Assume that all mechanical system variables, such as angles and speeds of every turbine rotor and that of generator and exciter; torques output from each stage of turbine and governor system and the electrical system variables such as generator power and current, voltage across the capacitor and to ground (generator side), damper currents and voltage output from voltage regulator and exciter, are measurable, then we have

For electrical power

$$
\begin{equation*}
\Delta \mathrm{P}=\mathrm{V}_{\mathrm{do}} \Delta i_{\mathrm{d}}+\mathrm{V}_{\mathrm{qo}} \Delta i_{\mathrm{q}}+i_{\mathrm{do}} \Delta \mathrm{~V}_{\mathrm{d}}+\mathrm{i}_{\mathrm{qo}} \Delta \mathrm{~V}_{\mathrm{q}} \tag{4-5}
\end{equation*}
$$

By substituting $V_{d}, V_{q}$ from equation (2-25)

$$
\begin{aligned}
\Delta P & =m_{11} \Delta i_{d}+m_{12} \Delta i_{q}+m_{14} \Delta V_{c d}+m_{15} \Delta V_{c q}+m_{16} \Delta \delta \\
\text { where } m_{11} & =V_{d o}+\left(R_{e}+R_{t}\right) i_{d o}+\left(X_{e}+X_{t}\right) i_{q o} \\
m_{12} & =V_{q o}-\left(X_{e}+X_{t}\right) i_{d o}+\left(R_{e}+R_{t}\right) i_{q o} \\
m_{14} & =i_{d o} \\
m_{15} & =i_{q o} \\
m_{16} & =v_{o}\left[i_{d o} \cos \delta-i_{q o} \sin \delta\right]
\end{aligned}
$$

For terminal current

$$
\begin{equation*}
\Delta i_{t}=\frac{i_{\text {do }}}{i_{\text {to }}} \Delta i_{d}+\frac{i_{q o}}{i_{\text {to }}} \Delta i_{q} \tag{4-7}
\end{equation*}
$$

or $\quad \Delta i_{t}=m_{21} \Delta i_{d}+m_{22} \Delta i_{q}$
where $\quad m_{21}=\frac{i_{\text {do }}}{i_{\text {to }}}, \quad m_{22}=\frac{i_{\text {go }}}{i_{\text {to }}}$

For voltage across the capacitor

$$
\begin{equation*}
\Delta V_{c t}=\frac{V_{c d o}}{V_{c o}} \Delta V_{c d}+\frac{V_{c q o}}{V_{c o}} \Delta V_{c q} \tag{4-8}
\end{equation*}
$$

or

$$
\Delta V_{c}=m_{44} \Delta \Delta V_{c d}+m_{45} \Delta V_{c q}
$$

where $\quad m_{44}=\frac{V_{c d o}}{V_{c o}}, \quad m_{45}=\frac{V_{c q o}}{V_{c o}}$

For voltage at the terminal of the capacitor.
As shown in Fig. $2-7, \mathrm{~V}_{\mathrm{ct}}$ is the voltage at the generator side of the capacitor with respect to ground

$$
\begin{equation*}
\Delta V_{c t}=\frac{V_{\text {ctdo }}}{V_{\text {cto }}} \Delta V_{c t d}+\frac{V_{\text {ctqo }}}{V_{\text {cto }}} \Delta V_{\text {ctq }} \tag{4-9}
\end{equation*}
$$

where $V_{c t d}$ and $V_{c t q}$ are the $d, q$ components of $V_{c t}$ and can be expressed in terms of the voltage across the capacitor and infinite bus voltages.
or $\quad \Delta V_{c t}=m_{54} \Delta V_{c d}+m_{55} \Delta V_{c q}+m_{56} \Delta \delta$
where $\quad m_{54}=\frac{V_{\text {ctdo }}}{V_{\text {cto }}}, \quad m_{55}=\frac{V_{\text {ctqo }}}{V_{c \text { to }}}$
$\mathrm{m}_{56}=\left[\mathrm{V}_{\text {ctdo }} \cos \delta-\mathrm{V}_{\text {ctqo }} \sin \delta\right] \frac{\mathrm{V}_{\mathrm{o}}}{\mathrm{V}_{\text {cto }}}$
Besides $\mathrm{m}_{11}, \mathrm{~m}_{22}, \mathrm{~m}_{44}$ and $\mathrm{m}_{55}$, the other main diagonal elements are unity. Other off-diagonal elements are zero except those a1ready derived.

### 4.2 State Equations in Canonical Form

A design procedure has been developed utilising state equations in canonical form [8]:

$$
F_{o}=\left[\begin{array}{ccccccc}
0 & 1 & \cdot & \cdots & \cdot & 0  \tag{4-12}\\
\cdot & 0 & 1 & & & & \cdot \\
\cdot & & & & \cdots & & \\
\cdot & & & & \cdot & & \cdot \\
\cdots & & & & & 1 & \cdot \\
0 & \cdots & \cdot & \cdot & \cdot & 0 & 1 \\
-\alpha_{1} & -\alpha_{2} & \cdot & \cdot & \cdot & -\alpha_{n-1} & \\
c_{n}
\end{array}\right]
$$

Let
$Z=T Y$
'
we shall have

$$
\begin{equation*}
\dot{Y}=T^{-1} \dot{Z}=F_{o} Y+G_{o} U \tag{4-14}
\end{equation*}
$$

where $\quad F_{0}=T^{-1} \mathrm{FT}$
and

$$
G_{0}=T G=\left[\begin{array}{lllll}
0 & 0 & 0 & \ldots & 1
\end{array}\right]^{t}
$$

The transformation matrix T can be found as follows:
Since the eigenvalues remain unchanged with similarity transformation, we shall have

$$
\begin{equation*}
|\lambda I-F|=\left(\lambda-\lambda_{1}\right)\left(\lambda-\lambda_{2}\right) \ldots\left(\lambda-\lambda_{n}\right)=0 \tag{4-17}
\end{equation*}
$$

and

$$
\begin{equation*}
\left|\lambda I-F_{0}\right|=\lambda^{n}+\dot{\alpha}_{n} \lambda^{n-1}+\dot{\alpha}_{n-1} \lambda^{n-2}+\ldots+\alpha_{1}=v 0 \tag{4-18}
\end{equation*}
$$

where $\lambda_{1}, \lambda_{2}, \ldots \lambda_{n}$ are the eigenvalues of the system. The $\alpha$ 's can be determined from (4-16) and (4117),

$$
\begin{align*}
\alpha_{n} & =-\sum_{i=1}^{n} \lambda_{i} \\
\alpha_{n-1} & =\lambda_{1} \lambda_{2}+\lambda_{1} \lambda_{3}+\ldots+\lambda_{2} \lambda_{3}+\ldots+\lambda_{n-1} \lambda_{n} \\
\alpha_{n-2} & =-\lambda_{1} \lambda_{2} \lambda_{3}-\lambda_{1} \lambda_{2} \lambda_{4}-\ldots-\lambda_{n-2} \lambda_{n-1} \lambda_{n} \\
\sim & \\
\cdot & \\
\alpha_{1} & =(-1)^{n}{\underset{i=1}{n} \lambda_{i}}_{\alpha_{1}}^{n} \tag{4-19}
\end{align*}
$$

Let the transformation $T$ matrix be written as

$$
\begin{equation*}
T=\left[T_{1}, T_{2}, T_{3}, \ldots T_{n}\right] \tag{4-20}
\end{equation*}
$$

where $T_{1}, T_{2}, T_{3}, \ldots T_{n}$ are the column vectors of $T$ matrix. From (4-15) and (4-20), we have

$$
T F_{0}=F T
$$

or

$$
\begin{equation*}
\left[T_{1}, T_{2}, T_{3}, \ldots T_{n}\right] F_{0}=F\left[T_{1}, T_{2}, T_{3}, \ldots T_{n}\right] \tag{4-21}
\end{equation*}
$$

From (4-16) and ( $4-21$ ), we have

$$
\begin{equation*}
\mathrm{T}_{\mathrm{n}}=\mathrm{G} \tag{4-22}
\end{equation*}
$$

Hence, we can compute $T_{1}, T_{2}, T_{3} \ldots T_{n}$ by using the following recursive formula

$$
\begin{equation*}
T_{n-i}=F_{n-i+1}+\alpha_{n-i+1} G \quad i==1,2,3, \ldots n-1 \tag{4-23}
\end{equation*}
$$

and

$$
\begin{equation*}
\mathrm{FT}_{1}+\alpha_{1} \mathrm{G}=0 \tag{4-24}
\end{equation*}
$$

The condition of (4-24) may not be met due to the accumulated computation errors. Let $T_{1}, T_{2}, T_{3}, \ldots T_{n}$ be the computed results and $\hat{T}_{1}, \hat{T}_{2}, \hat{T}_{3}$,
$\ldots \hat{T}_{n}$ be correct values

$$
\begin{align*}
& F \hat{T}_{1}+\alpha_{1} G=0 \\
& F T_{1}+\alpha_{1} G=\hat{\varepsilon} \varepsilon \tag{4-25}
\end{align*}
$$

and the error

$$
\begin{equation*}
\hat{T}_{1}-T_{1} \triangleq n_{1} \tag{4-26}
\end{equation*}
$$

Then $\quad n_{1}=-F^{-1} \varepsilon$
Similarly,

$$
\begin{align*}
& \mathrm{F} \hat{\mathrm{~T}}_{2}+\alpha_{2} \mathrm{G}=\hat{\mathrm{T}}_{1} \\
& \mathrm{FT} \mathrm{~T}_{2}+\alpha_{2} \mathrm{G}=\mathrm{T}_{1} \tag{4-28}
\end{align*}
$$

and

$$
\begin{equation*}
n_{2}=F^{-1} n_{1} \tag{4-29}
\end{equation*}
$$

Therefore,

$$
\begin{array}{ll}
n_{i} & =F^{-1} n_{i-1}  \tag{4-30}\\
\hat{T}_{i}=n_{i}+T_{i} & i=2,3,4, \ldots n \\
\end{array}
$$

### 4.3 Linear Optimal Control Design

The system equations in canonical form were

$$
\begin{equation*}
\dot{Y}=F_{o} Y+G_{0} U \tag{4-14}
\end{equation*}
$$

The characteristic equation of the open loop system is

$$
\begin{equation*}
\left|\lambda I-F_{0}\right|=\lambda^{n}+\alpha_{n} \lambda^{n-1}+\alpha_{n-1} \lambda^{n-2}+\ldots+\alpha_{1} \tag{4-31}
\end{equation*}
$$

Let the desired eigenvalues of the closed-loop system be $\hat{\lambda}_{1}, \hat{\lambda}_{2}, \hat{\lambda}_{3}, \ldots \hat{\lambda}_{n}$ The new characteristic equation will be

$$
\begin{align*}
& \left(\lambda-\hat{\lambda}_{1}\right)\left(\lambda-\hat{\lambda}_{2}\right)\left(\lambda=\hat{\lambda}_{3}\right) \ldots\left(\lambda-\hat{\lambda}_{n}\right) \\
& =\lambda^{n}+\hat{\alpha}_{n} \lambda^{n-1}+\hat{\alpha}_{n-1} \lambda^{n-2}+\ldots+\hat{\alpha}_{1}=0 \tag{4-32}
\end{align*}
$$

Since characteristic equation of the closed loop system is

$$
\begin{align*}
& \left|\lambda I-\left(F_{o}-G_{o} S_{o}\right)\right| \\
& =\lambda^{n}+\left(\alpha_{n}+\beta_{n}\right) \lambda^{n-1}+\left(\alpha_{n-1}+\beta_{n-1}\right) \lambda^{n-2}+\ldots+\left(\alpha_{1}+\beta_{1}\right) \\
& =0 \tag{4-33}
\end{align*}
$$

where

$$
\begin{equation*}
U \triangleq-S_{0} Y \tag{4-34}
\end{equation*}
$$

Equating (4-32) to (4-33) gives

$$
\begin{equation*}
\hat{\alpha}_{i}-\alpha_{i}=\beta_{i} \quad i=1,2,3, \ldots n \tag{4-35}
\end{equation*}
$$

and

$$
\begin{equation*}
s_{0}=\left[\beta_{1}, \beta_{2}, \beta_{3} ; \ldots \beta_{n}\right] \tag{4-36}
\end{equation*}
$$

Finally, the linear optimal controller in measurable state variables

$$
U \triangleq-S_{\mathrm{O}} \mathrm{Y}
$$

or

$$
\begin{equation*}
\mathrm{U} \triangleq-\mathrm{S}_{\mathrm{o}} \mathrm{~T}^{-1} \mathrm{Z} \tag{4-37}
\end{equation*}
$$

### 4.4 Stabilization of SSR

Because of the number of state variables which can be measured, the 22 nd order and 19 th reduced order models are used for the linear optimal control design. Eigenvalue analysis shows that all the important mechanical and electrical eigenvalues are essentially unchanged; Table 4-1. Single mechanical mode stabilization

At $30 \%$ compensation and 0.9 power factor lagging of the reduced 22nd order system without stabilizer, the $204 \mathrm{rad} . / \mathrm{sec}$. or 32.5 hertz mechanical mode is excited and has negative damping (eigenvalues with positive real part). By utilizing the design procedure described in this chapter, an optimal controller can be designed to shift the eigenvalues from $+0.1541 \pm \mathrm{j} 204.35$ to $-6.500 \pm \mathrm{j} 204.35$ and another mechanical mode which is barely stable from $-0.08805 \pm \mathrm{j} 8.4938$ to $-6.000 \pm \mathrm{j} 8.4938$. A11 eigenvalues are stabilized as shown in Table 4-2. The controller is 7.8 $\left(7.823 \Delta \omega_{1}, 0.0964 \Delta \theta_{1},-183.005 \Delta \omega,-8.801 \Delta \delta, 192.487 \Delta \omega_{3},-3.398 \Delta \theta_{3}\right.$, $65.534 \Delta \omega_{4}, 7.448 \Delta \theta_{4},-1.336 \Delta \theta_{5},-47.478 \Delta \omega_{6},-2.738 \Delta \theta_{6},-29.746 \Delta \mathrm{P}$, $28.645 \Delta i_{t}, 1.454 \Delta i_{f}, 1.475 \Delta i_{D},-5.931 \Delta i_{Q},-5.929 \Delta i_{G},-60.094 \Delta V_{c}$, $\left.35.987 \Delta \mathrm{~V}_{\mathrm{c}_{\mathrm{t}}},-0.000260 \Delta \mathrm{~V}_{\mathrm{R}},-0.00259 \Delta \mathrm{E}_{\mathrm{FD}}\right)$ ).

Stabilization of two mechanical modes simultaneously
For the same system but.with $50 \%$ compensation, two mechanical modes were excited $+0.1560 \pm j 202.68$ and $+0.9101 \pm j 161.42$ were excited simultaneously. Another optimal controller is designed to shift the two mechanical modes to $-6.500 \pm j 202.68$ and $-3.500 \pm j 161.42$ as shown in Table 4-3. The controller is (1.124 $\omega_{1},-4.959 \Delta \theta_{1},-23.848 \Delta \omega$, $189.462 \Delta \delta, 13.843 \Delta \omega_{3},-321.301 \Delta \theta_{3}, 24.286 \Delta \omega_{4}, 147.568 \Delta \theta_{4},-6.018 \Delta \omega_{5}$, $4.467 \Delta \theta_{5},-9.927 \Delta \omega_{6},-25.064 \Delta \theta_{6},-26.719 \Delta \mathrm{P}, 29.533 \Delta i_{t},-1.030 \Delta i_{f},-1.007 \Delta i_{D}$, $\left.-6.711 \Delta i_{Q},-6.712 \Delta i_{G},-22.422 \Delta V_{c}, 31.169 \Delta V_{c_{t}},-0.000278 \Delta V_{R},-0.00264 \Delta \mathrm{E}_{\mathrm{FD}}\right)$.

Although the two controllers designed by the procedure presented in this chapter have been proved to be effective in stabilizing the system, the damper currents are not directly measurable. Still another linear optimal controller is designed for the system, without the need for damper currents. The equations associated with the damper windings are dropped, resulting in a 19 th order system. The controller is $\left(1.84 \Delta \omega_{1}, 1.01 \Delta \theta_{1}\right.$, $-41.51 \Delta \omega_{,}-30.63 \Delta \delta, 54.51 \Delta \omega_{3}, 39.93 \Delta \theta_{3}, 7.37 \Delta \omega_{4},-5.77 \Delta \theta_{4},-6.41 \Delta \omega_{5}$, $-3.32 \Delta \theta_{5},-7.46 \Delta \omega_{6},-2.16 \Delta \theta_{6},-1.66 \Delta \mathrm{P}, 2.63 \Delta \mathbf{i}_{t},-0.872 \Delta \mathbf{i}_{f},-2.26 \Delta V_{c}$, $\left.-2.35 \Delta \mathrm{~V}_{\mathrm{tc}},-0.000295 \Delta \mathrm{~V}_{\mathrm{R}},-0.00274 \Delta \mathrm{E}_{\overline{\mathrm{FD}}}\right)$, and the eigenvalues of the system with and without the controller are shown in Table 4-4. Finally the controller is tested on the original system for various degrees of compensation. The results are plotted in Fig. 4-1. It is found that the controller designed for the 19 th order model with $30 \%$ compensation, not only can stabilize the original 27 th order system for $30 \%$ compensation but also can stabilize the original system from 10 to $70 \%$ compensation. This proves the effectiveness of such controller design in wide-rangecompensation multi-mode SSR stabilization.

| original | reduced | reduced |
| :--- | :---: | :---: |
| system | 22nd mode1 | 19th model |


|  | $-0.1818 \pm j 298.18$ | $-0.1818 \pm j 298.18$ | $-0.1818 \pm j 298.18$ |
| :--- | :--- | :--- | :--- |
| Shaft modes | $+0.1541 \pm j 204.35$ | $+0.1541 \pm j 204.35$ | $-0.2290 \pm j 203.22$ |
|  | $-0.2496 \pm j 160.72$ | $-0.2496 \pm j 160.72$ | $-0.2273 \pm j 160.66$ |
|  | $-0.6706 \pm j 127.03$ | $-0.6706 \pm j 127.03$ | $-0.6677 \pm j 127.03$ |
|  | $-0.2877 \pm j 99.21$ | $-0.2877 \pm j 99.21$ | $-0.2627 \pm j 99.14$ |
| Stator/Network | $-7.0224 \pm j 542.80$ | $-7.0224 \pm j 542.80$ | $-4.8208 \pm j 514.02$ |
|  | $-6.1984 \pm j 209.20$ | $-6.1984 \pm j 209.20$ | $-3.6580 \pm j 238.75$ |

Table 4-1 Eigenvalues of original system and reduced order models without controller at $30 \%$ compensation and $P=0.9$ p.u. at 0.9 power factor lagging.
reduced 22 nd order model without controller
reduced 22 nd order model with controller
original system with controller

|  | $-0.1818 \pm \div j 298.18$ | $-0.1818 \pm j 298.18$ | $-0.1818 \pm j 298.18$ |
| :---: | :---: | :---: | :---: |
|  | +0.1541 $\pm$ j204.35 | $-6.5000 \pm j 204.35$ | $-6.5000 \pm j 204.35$ |
| Shaft modes | $-0.2496 \pm j 160.72$ | $-3.5000 \pm j 160.72$ | $-3.5000 \pm j 160.72$ |
|  | $-0.6706 \pm j 127.03$ | $-0.6706 \pm j 127.03$ | $-0.6706 \pm j 127.03$ |
|  | $-0.2877 \pm$ j 99.21 | -0.2877 $\pm$ j 99.21 | $-0.2877 \pm$ j 99.21 |
| $\lambda \delta \omega$ | $-0.0881 \pm j 8.4938$ | $-6.0000 \pm j 8.4938$ | $-6.2367 \pm j 8.4158$ |
|  | $-7.0224 \pm j 542.80$ | -7.0224 $\pm$ j542.80 | $-7.0224 \pm j 542.80$ |
| Stator/Network | $-6.1984 \pm j 209.20$ | $-6.1984 \pm j 209.20$ | $-6.1984 \pm j 209.20$ |
|  | -8.4858 | -8.4858 | -9.6038 |
| Synchronous | -31.920 | -31.920 | -31.923 |
| Machine Rotor | -25:404 | -25.404 | -25.404 |
|  | -2. 1855 | -2.1855 | -1.5570 |
| Exciter and | -499.97 | -499.97 | -499.97 |
| Voltage Regulator | -101.91 | -200.00 | -200.00 |
|  |  |  | -0\%1404 |
| Turbine and |  |  | -4.8741 |
| Governor |  |  | $-2.8538$ |
|  |  |  | $-3.9883 \pm j 2.9898$ |

Table 4-2 Eigenvalues of reduced 22 nd order model with/without controller and original system with the controller at $30 \%$ compensation and $P=0.9$ p.u. at 0.9 power factor lagging.
reduced 22 nd order model without controller
reduced $22 n d$ order model with controller
original system with controller

|  | $-0.1818 \pm j 298.18$ | $-0.1818 \pm j 298.18$ | $-0.1818 \pm j 298.18$ |
| :--- | :--- | :--- | :--- |
|  | $+0.1560 \pm j 202.68$ | $-6.5000 \pm j 202.68$ | $-6.5000 \pm j 202.68$ |
| Shaft modes | $+0.9101 \pm j 161.42$ | $-3.5000 \pm j 161.42$ | $-3.5000 \pm j 161.42$ |
|  | $-0.6799 \pm j 127.08$ | $-0.6799 \pm j 127.08$ | $-0.6799 \pm j 127.08$ |
|  | $-0.3545 \pm j 99.49$ | $-0.3545 \pm j 99.49$ | $-0.3545 \pm j 99.49$ |
| $\lambda \delta \omega$ | $-0.2958 \pm j 9.5621$ | $-6.0000 \pm j 9.5621$ | $-6.4682 \pm j 9.7544$ |
|  |  |  |  |

Table 4-3 Eigenvalues of reduced 22 nd order model with/without controller and original system with the controller at $50 \%$ compensation and $\mathrm{P}=.0 .9$ p.u. at 0.9 power factor lagging.

## reduced 19th order model without controller

reduced 19 th order model with controller
original system with controller

|  | -0.1818 $\pm$ j298.18 | -0.1817 $\pm \mathrm{j} 298.18$ | -0.1818 $\pm$ j298.18 |
| :---: | :---: | :---: | :---: |
|  | -0.2290 $\pm$ j203.22 | -6.5000 $\pm$ j203.22 | -0.4968 $\pm$ j 203.19 |
| Shaft modes | -0.2273 $\pm$ j160.66 | -3.5000 $\pm \mathrm{j} 160.66$ | -0.2790 $\pm \mathrm{j} 160.52$ |
|  | $-0.6677 \pm j 127.03$ | $-0.6676 \pm j 127.03$ | -0.6697 $\pm$ j127.04 |
|  | -0.2627 $\pm$ j 99.14 | -0.2628 $\pm$ j 99.14 | -0.2770 $\pm 99.22$ |
| $\lambda \delta \omega$ | -0.2266 $\pm$ j7.9054 | $-6.0000 \pm j 7.9054$ | $-0.1092 \pm j 8.7874$ |
| Stator/Network | -4.8208 $\pm$ j514.02 | -4.8208 $\pm \mathrm{j} 514.02$ | $-7.1255 \pm j 542.54$ |
|  | $-3.6580 \pm \mathrm{j} 238.75$ | $-3.6582 \pm j 238.75$ | -5.9600 $\pm$ j209.43 |
|  | -8.0056 | -8.0056 | -25.4025 |
| Synchronous |  |  | $-2.8136+\mathrm{j} 0.2572$ |
| Machine Rotor |  |  | -2.8136-j0.2572 |
|  |  |  | $-1.6473+j 391.91$ |
| Exciter and | -499.52 | -499.52 | -773.59 |
| Voltage Regulator | -93.682 | -200.00 | -1.6473-- j391.91 |
|  |  |  | -0.1401 |
| Turbine and |  |  | -4.6414 |
| Governor |  |  | -0.2592 |
|  |  |  | -4.7914 $\pm$ j0.9552 |

Table 4-4 Eigenvalues of reduced order model with/without controller and original system with the controller at $30 \%$ compensation and $\mathrm{P}=0.9$ p.u. at 0.9 power factor lagging.


Fig. 4-1 The effect of capacitor compensation with controller for $P=0.9$ p.u. at 0.9 power

## 5. CONCLUSIONS

A high-order power system model for subsynchronous resonance studies is developed. The model includes mass-spring system, synchronous machine, series capacitor compensated transmission lines, turbines and governor, voltage regulator and exciter. The transient terms $p \psi_{d}$ and $\mathrm{p} \psi_{\mathrm{q}}$ are included.

From eigenvalue analysis, it is found that by changing the degree of compensation the frequency of the electrical mode will be changed and that, in some cases, even more than one mechanical mode can be excited at the same time. When a conventional lëad-lag supplementary excitation control for the stabilization of small oscillations is included, it has an adverse effect on the other mechanical modes close to the small oscillation mode. Such finding is in agreement with other previous work [22]. When the damper impedance is decreased, it does reduce the possibility of SSR under ideal conditions [5].

Linear optimal controllers based upon an earlier developed method [8] are designed. Two controllers are designed with a reduced 22nd order model and one with a reduced 19 th order model and the latter controller, not only can stabilize the original 27 th order system for $30 \%$ compensation, but also can stabilize the system for wide-range compensation and multi-mode SSR.

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