# PLASTIC HINGE LENGTH IN HIGH-RISE CONCRETE SHEAR WALLS 

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#### Abstract

The flexural displacement capacity of a concrete shear wall depends on the length of the plastic hinge. Typical building codes and several researchers recommend the use of an equivalent plastic hinge length at the base equal to half the wall length. However, the plastic hinge length is also influenced by parameters other than the wall length. There are currently no recommendations on what should be the plastic hinge length for parallel walls of different lengths in a high-rise building.


A parametric study was conducted to investigate the parameters that affect the length of the plastic hinge in concrete walls. The walls were analyzed using program VecTor2. The analytical model was validated with tests results performed on wall specimens.

The results obtained show that the inelastic curvatures vary linearly over the plastic hinge length. The shape of the strain profile in slender walls after cracking depends on the amount of reinforcement. Longer walls have larger plastic hinge lengths than shorter walls. Compressive axial loads reduce the plastic hinge length, tensile axial loads have the opposite effect. A simple shear model was proposed to estimate the increase in plastic hinge length when the shear stresses are high.

Walls of different lengths interconnected by rigid slabs at various levels have different curvature distributions and plastic hinge lengths. The curvatures in the longer wall do not change whether it is alone or combined with a wall of shorter length. The shorter length wall is subjected to larger curvatures at the base when it is combined. A simple model was proposed to predict the maximum curvature in the shorter wall.

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## LIST OF SYMBOLS

- $A_{g}:$ Gross area of cross-section
- $A_{c}$ : Area of concrete
- $A_{s}$ : Area of reinforcement
- $A_{s c}$ : Area of compression reinforcement
- $A_{s t}$ Area of tension reinforcement
- $a$ : Maximum aggregate size
- $a_{s l}$ : Zero-one variable in Equation 2.46 and 2.47
- $b$ : Width of member
- $\quad b_{w}:$ Web thickness of wall
- $\quad C$ : Cohesion
- $\quad C_{d}$ : Compression softening strain softening factor, defined in Equation 3.10
- $C_{s}$ : Compression softening shear slip factor in Equation 3.9
- $\quad c$ : Neutral axis depth at ultimate moment
- $c_{a}$ : Averaging factor, defined in Equation 3.5
- $\quad c_{b}$ : Coefficient used to reflect bar spacing in Equation 3.43
- $c_{t}$ : Coefficient that incorporates the influence of reinforcement bond characteristics, defined in Equation 3.19
- $d$ : Effective depth of member
- $d_{b}$ : Bar diameter of the tension reinforcement
- $\quad d_{v}$ : Lever arm of tensile force in the longitudinal reinforcement
- $E_{c}$ : Concrete initial tangent stiffness
- $E_{s}$ : Initial tangent stiffness or elastic modulus of reinforcement
- $E_{\text {sec }}$ Concrete secant stiffness
- $E_{\text {sh }}$ : Strain hardening modulus
- $F$ : Lateral force
- $f_{c}^{\prime}:$ Concrete cylinder uniaxial compressive strength
- $f_{c l}$ : Average net concrete axial stress in the principal tensile direction
- $f_{c 1}^{a}$ : Average concrete tensile stress due to tension stiffening
- $f_{c 1}^{b}$ : Average concrete tensile stress due to tension softening
- $\quad f_{c 2}$ : Average net concrete axial stress in the principal compressive direction
- $\quad f_{c 2}^{a}$ : Average concrete compressive stress contribution of unconfined concrete
- $\quad f_{c 2}^{b}:$ Average concrete compressive stress contribution of confined concrete
- $f_{c c}$ : Concrete cube strength
- $f_{c r}$ : Concrete cracking stress
- $f_{\text {crui }}$ Defined in Equation 3.24
- $f_{p}$ : Peak concrete compressive stress
- $f_{s}$ : Reinforcement stress
- $f_{i}$ Tensile strength of steel
- $f_{i}$ : Concrete uniaxial tensile strength
- $f_{u}$ : Ultimate strength of reinforcement
- $\quad f_{y}$ : Yield stress of the tension reinforcement
- $f_{y}^{\prime}$ : Yield stress of the compression reinforcement
- $G_{f}$ Energy required to form a complete crack of unit area
- $H_{w}$ : Total height of wall
- $h$ : Depth or height of the cross-section of member
- $\quad I_{z}$ : Moment of inertia of the reinforcement
- $k_{i}$ : Factor that considers the influence of the tension reinforcement in Equation 2.12, 2.17 and 2.19
- $k_{2}$ : Factor that considers the influence of the axial load, defined in Equation 2.13
- $k_{3}$ : Factor that considers the influence of the concrete strength, defined in Equation 2.14
- $\quad k_{c}$ : Stiffness of the notional concrete foundation, defined in Equation 3.43
- $\quad L$ : Length of member in the longitudinal direction
- $L_{p}$ : Plastic hinge length
- $\quad L_{p, i}$. Plastic hinge length in element " $i$ "
- $L_{p, \text { consti }}$ Plastic hinge length for a constant inelastic curvature
- $L_{p, c y}$ : Plastic hinge length for cyclic loading
- $L_{p, \text { in }}$ : Plastic hinge length for a linearly varying inelastic curvature
- $L_{p, \text { meas: }}$ Measured plastic hinge length
- $L_{p, \text { mon }}$ : Plastic hinge length for monotonic loading
- $L_{p, p r e d}$ : Predicted plastic hinge length
- $L_{r}$ : Distance over which the crack is assumed to be uniformly distributed
- $L_{s}$ : Distance from the section of maximum moment to the section of zero moment or shear span
- $l_{s}$ : Standard length in Equation 2.39 and 2.41
- $\quad l_{w}$ : Horizontal length of wall section
- $\quad$ M: Total bending moment
- $M_{f}$ Moment due to flexure
- $M_{\max }$ : Maximum moment in the length of member
- $M_{u}$ : Ultimate moment
- $\quad M_{\nu}:$ Moment due to shear
- $M_{y}$ : Yield moment
- $\quad m$ : Bond parameter, defined in Equation 3.20
- $\quad N_{v}$ : Axial compression due to shear
- $n$ : Curve fitting parameter for stress-strain response of concrete in compression, defined in Equation 3.2
- $\quad P$ : Axial load
- $\quad P_{u}$ : Axial compressive strength of member without any bending moment
- PER: Percentage ratio of volume of stirrups to volume of concrete core measured outside of stirrups
- $\quad R_{m}$ : Moment ratio, defined in Equation 2.32
- $R_{\delta}$ : Strain ratio, defined in Equation 2.31
- RFT: Defined in Equation 2.43
- $\quad r$ : Ratio of the principal tensile strain to the principal compressive strain
- $\quad$ SPR: Defined in Equation 2.44
- $s$ : Crack spacing
- $\quad V$ : Shear force
- $V_{c}$ : Shear force in concrete
- $\quad V_{d}$ : Dowel force
- $\quad V_{d u}$ : Ultimate dowel force
- $V_{s}$ : Shear force in longitudinal reinforcement
- $\quad V_{z}$ : Shear adjacent to a concentrated load or reaction at the section of maximum moment
- $v$ : Shear stress
- $v_{c i}$. Local shear stress on the crack
- $v_{c i, \text { max }}$ : Maximum local shear stress on the crack
- $v_{c o}$ : Defined in Equation 3.32
- $v_{c r}$. Cracking shear stress
- $v_{d}$ : Shear resistance due to dowel action
- $w$ : Average crack width
- $w_{z}$ : Uniformly distributed load at the section of maximum moment
- $\quad Z_{m}$ : Slope of compression post-peak descending curve, defined in Equation 3.8
- $\quad \alpha$. Calibration parameter in Equation 2.10
- $\alpha_{I}$ : Model parameter in Equation 2.39, 2.40 and 2.41
- $\alpha_{2}$ : Model parameter in Equation 2.39, 2.40 and 2.41
- $\quad \beta$ : Calibration parameter in Equation 2.10
- $\quad \beta_{d}$ : Softening parameter, defined in Equation 3.9
- $\beta_{i}$ : Strength enhancement factor, defined in Equation 3.14
- $\Delta$ : Total displacement
- $\Delta_{f}$ Flexural displacement
- $\Delta_{p}$ : Plastic or inelastic deformation
- $\Delta_{s}$ : Slip displacement
- $\Delta_{u}$ : Ultimate displacement
- $\Delta_{v}$ : Shear displacement
- $\Delta_{y}:$ Yield displacement
- $\Delta \theta_{\sigma}:$ Post-cracking rotation of the principal stress field
- $\Delta \theta_{\varepsilon}:$ Post-cracking rotation of the principal strain field
- $\delta_{s}$ : Shear slip
- $\delta_{s}^{*}$ : Defined in Equation 3.29
- $\varepsilon_{c I}$ : Average net concrete axial strain in the principal tensile direction
- $\varepsilon_{c 2}$ : Average net concrete axial strain in the principal compressive direction
- $\varepsilon_{c e}$ : Concrete strain in the extreme compression fiber at yield curvature
- $\varepsilon_{c h}$ : Characteristic strain
- $\varepsilon_{c r}$ : Concrete cracking strain
- $\varepsilon_{c u}$ : Concrete strain in the extreme compression fiber at ultimate curvature
- $\varepsilon_{o}$ : Concrete compressive strain corresponding to $f_{c}{ }_{c}$
- $\quad \varepsilon_{p}$ : Concrete compressive strain corresponding to $f_{p}$
- $\varepsilon_{s}:$ Reinforcement strain
- $\varepsilon_{s h}$ : Strain at the onset of strain hardening
- $\varepsilon_{u}$ : Reinforcement ultimate strain
- $\quad \varepsilon_{x x}$ : Total axial strain in the $x$-direction
- $\varepsilon_{y}$ : Yield strain
- $\varepsilon_{y y}$ : Total axial strain in the $y$-direction
- $\boldsymbol{\Phi}$ : Internal angle of friction
- $\phi$ : Total curvature
- $\phi_{m a x}$ : Maximum curvature
- $\phi_{\text {max } i:}$ Maximum curvature in element " $i$ "
- $\phi_{p}$ : Plastic or inelastic curvature
- $\phi_{u}$ : Ultimate curvature
- $\phi_{y}$ : Yield curvature
- $\phi_{y, i}$ : Yield curvature in element " $i$ "
- $\gamma_{s}$ : Shear slip strain
- $\quad \gamma_{s}^{a}$ : Shear slip strain determined from the stress-based model
- $\gamma_{s}^{b}$ : Shear slip strain determined from the constant rotation lag model
- $\quad \gamma_{x y}$ : Total shear strain
- $\lambda$ : Parameter that compares the stiffness of the concrete to the stiffness of the reinforcing bar, defined in Equation 3.42
- $\theta$. Total rotation or slope
- $\theta_{c}$ : Angle of the crack
- $\theta_{i}$ : Total rotation or slope in element " $i$ "
- $\theta_{i c}$ : Inclination of the principal stress field at cracking
- $\theta^{l}:$ Rotation lag
- $\theta_{n}$ : Angle between the normal to the crack surface and the longitudinal axis of the reinforcement
- $\theta_{p}$ : Plastic or inelastic rotation or slope
- $\theta_{p, d 2}$ : Plastic or inelastic rotation or slope in the length $d / 2$
- $\theta_{u}$ : Ultimate rotation or slope
- $\quad \theta_{\sigma}$ : Inclination of the principal stress field
- $\quad \rho$ : Tension reinforcement ratio
- $\quad \rho^{\prime}$ : Compression reinforcement ratio
- $\quad \rho_{b}:$ Reinforcement ratio at balanced ultimate strength condition in a member without compression reinforcement
- $\quad \rho_{s}$ : Reinforcement ratio
- $\omega$ : Tension reinforcement index, defined in Equation 2.15
- $\quad \omega^{\prime}$ : Compression reinforcement index, defined in Equation 2.16
- $\omega_{b}$ : Tension reinforcement index for balanced ultimate strength condition, defined in Equation 2.24
- $\xi$ Calibration parameter in Equation 2.10
- $\xi_{L}:$ Model error term in Equation 2.39, 2.40 and 2.41
- $\psi$. Defined in Equation 3.30


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To my parents, Alfredo and Berta

## CHAPTER 1: INTRODUCTION

### 1.1 Research significance

The use of concrete shear walls to provide lateral strength and stiffness in high-rise buildings has become common practice. These walls require enough shear strength capacity and flexural displacement capacity to have an adequate seismic behaviour. The flexural displacement capacity depends on the compression strain capacity of concrete, the neutral axis depth and the length of the plastic hinge.

The inelastic curvatures at the base of the wall are commonly assumed to be constant over a length known as the equivalent plastic hinge length. Different studies done in the past on concrete members have shown that the equivalent plastic hinge length is proportional to the member dimension (e.g.: beam depth, wall length). Building code requirements in ACI 3182005 and CSA A23.3 2004 for determining confinement requirements at the ends of a concrete wall assume the plastic hinge length is equal to half of the wall length.

Several empirical models have been developed to estimate the length of the plastic hinge in reinforced concrete members based on test results. A large number of these models were calibrated so that they will give the real total displacement or real total rotation at failure. These models consider different parameters and provide significantly different predictions. Most of the research done in the past has focused on the influence of the member dimension, the span of the member and the longitudinal reinforcement properties. The effect of the axial load ratio and the strain hardening has also been studied. It is still not clear which are the most relevant parameters that influence the length of the plastic hinge.

Many of these models consider that the plastic hinge length is proportional to the member dimension only. This is due to the fact that the member dimension is commonly associated with the effect of diagonal cracking, which causes the length of the plastic hinge to increase. If the angle of the crack is considered to be $45^{\circ}$, it will extend to a distance from the support equal to the member dimension. Therefore, longer members have larger plastic hinge lengths. One of the objectives of this thesis will be to investigate in more depth how the member dimension and the plastic hinge length are related.

The increase in the plastic hinge length due to the influence of diagonal cracking is difficult to quantify. A simple shear model will be developed to try to estimate the plastic hinge length when the shear stresses are high.

Previous investigations on plastic hinge length in concrete walls have been done for individual members. To the knowledge of the author, there are no previous studies done on concrete wall systems. It is not known how the interaction of the walls that are part of the same system will affect the length of the plastic hinge of these members.

A system of coupled walls is connected together by very stiff beams (coupling beams). When a system of coupled walls with a high degree of coupling is subjected to lateral loading, the shear forces in the coupling beams induce high axial forces in the walls. Some walls are subjected to high axial tension, and others are subjected to high axial compression. The axial loads in the walls are expected to have a significant effect on the plastic hinge length. Walls subjected to compressive and tensile axial forces will be analyzed to study this effect.

Figure 1.1 System of coupled walls subjected to lateral loading


In high-rise buildings, it is common to have systems of walls of different lengths providing lateral resistance in the same direction of lateral loading. These walls are interconnected by rigid slabs at numerous floor levels. As a result, when this system is subjected to lateral loads, the displacement of all these walls is the same at the floor levels.

Figure 1.2 System of walls of different lengths interconnected by rigid slabs


Along with the walls, the gravity columns that do not form part of the lateral force resisting system will also have the same displacement at the floor levels. This tends to increase curvature demand in these elements. The columns must be able to sustain that demand.

An important objective of this thesis will be to analyze walls of different lengths connected together by rigid slabs to study the impact of the connection in the plastic hinge length. The effect on gravity columns will also be studied.

### 1.2 Objective of the thesis

The main objective of this thesis is to investigate the parameters that affect the length of the plastic hinge in concrete walls, using static nonlinear finite element analysis. The parameters considered in this study were the wall length, the distance from the section of maximum moment to the section of zero moment, diagonal cracking due to shear, and the axial load. This parametric study was first performed for individual walls. Then, walls of different lengths interconnected by rigid slabs at numerous levels were analyzed to investigate how this affects the length of the plastic hinge.

The analysis was performed using program VecTor2. VecTor2 is a computer program developed at the University of Toronto to perform nonlinear finite element analysis of reinforced concrete membrane structures, using the constitutive relationships of the Disturbed Stress Field Model. The program uses low-powered elements to model concrete structures. It considers a variety of effects to accurately predict the response, such as compression softening, tension stiffening, slip distortions and strain hardening.

Prior to the parametric study, analytical predictions made by program VecTor2 were verified with experimental results obtained from previous tests performed on cantilever walls to see how well the program predicts the response of these members. In particular, the curvature distributions along the height and the strain profiles along the wall section were examined and compared.

### 1.3 Organization of the thesis

The thesis is divided in seven chapters. Chapter 1 is the introduction. It presents the objective of the thesis.

Chapter 2 is the literature review. It first introduces the concept of plastic hinge length and the formulation of classical plastic hinge analysis. The results from 23 previous investigations on plastic hinge length and the models that were developed are presented. A summary of all these investigations, including the type of members studied and the parameters considered, is presented at the end of the chapter.

Chapter 3 presents the analytical methods. It describes the theoretical bases of program VecTor2, as well as its finite element formulation and the description of the material models for concrete and steel. Only the material models used in this thesis are presented. Finally, a brief description of the pre-processor and post-processor for program VecTor2 is presented.

Chapter 4 shows a comparison of analytical predictions from program VecTor2 with isolated wall test results. Two wall specimens were used for the comparison. The first is a high-rise shear wall tested at the University of British Columbia in 2000; the curvature distributions along the height were compared. The second is a rectangular shear wall tested at Clarkson University in 1995; the strain profiles along the wall section were compared.

Chapter 5 presents a parametric study of concrete walls. The factors that affect the length of the plastic hinge were investigated. This study was performed for two individual walls.

Chapter 6 presents the analysis of three systems of walls of different lengths interconnected by rigid slabs at numerous levels. The influence of the connection in the curvature distribution and the plastic hinge length of the walls were studied.

Chapter 7 presents the conclusions of the thesis.

## CHAPTER 2: LITERATURE REVIEW

### 2.1 Concept of plastic hinge length

Consider a slender reinforced concrete cantilever element subjected to a lateral load at the top, like the one in Figure 2.1. The total lateral displacement at the top, $\Delta$, is comprised of three components: the flexural displacement, $\Delta_{f}$, the shear displacement, $\Delta_{v}$, and the slip displacement, $\Delta_{s}$ :

Figure 2.1 Displacement components of a reinforced concrete cantilever element


Therefore:

$$
\begin{equation*}
\Delta=\Delta_{f}+\Delta_{v}+\Delta_{s} \tag{2.1}
\end{equation*}
$$

The shear and slip displacements are commonly not very important for high-rise buildings. In the classical formulation of plastic hinge analysis, it is considered that the total displacement of a reinforced concrete element after yielding has two components: the yield displacement, $\Delta_{y}$, and the plastic deformation, $\Delta_{p}$ :

Figure 2.2 Displacement components beyond the yield displacement


Therefore:

$$
\begin{equation*}
\Delta=\Delta_{y}+\Delta_{p} \tag{2.2}
\end{equation*}
$$

In practice, it is commonly assumed that the inelastic curvature in the plasticized region, although it has a certain variation, is constant over a length known as the plastic hinge length, $L_{p}$. This is shown in the Figure 2.3:

Figure 2.3 Formulation of classical plastic hinge analysis


From Figure 2.3, the inelastic curvature, $\phi_{p}$, is calculated from the total curvature, $\phi$, by:

$$
\begin{equation*}
\phi_{p}=\phi-\phi_{y} \tag{2.3}
\end{equation*}
$$

Where $\phi_{y}$ is the yield curvature. The inelastic rotation, $\theta_{p}$, can be determined by integrating the inelastic curvatures:

$$
\begin{equation*}
\theta_{p}=\phi_{p} L_{p} \tag{2.4}
\end{equation*}
$$

Considering that the inelastic rotation is concentrated at the centroid of the inelastic curvatures, the inelastic displacement can be expressed as:

$$
\begin{equation*}
\Delta_{p}=\theta_{p}\left(L-\frac{L_{p}}{2}\right)=\left(\phi-\phi_{y}\right) L_{p}\left(L-\frac{L_{p}}{2}\right) \tag{2.5}
\end{equation*}
$$

Combining Equation 2.2 and 2.5, the total displacement is:

$$
\begin{equation*}
\Delta=\Delta_{y}+\left(\phi-\phi_{y}\right) L_{p}\left(L-\frac{L_{p}}{2}\right) \tag{2.6}
\end{equation*}
$$

The yield displacement can be determined by integrating the elastic curvatures, considering a linear variation as shown in Figure 2.3:

$$
\begin{equation*}
\Delta_{y}=\frac{\phi_{y} L^{2}}{3} \tag{2.7}
\end{equation*}
$$

Then, the total displacement is:

$$
\begin{equation*}
\Delta=\frac{\phi_{y} L^{2}}{3}+\left(\phi-\phi_{y}\right) L_{p}\left(L-\frac{L_{p}}{2}\right) \tag{2.8}
\end{equation*}
$$

This is the classical formulation of plastic hinge analysis, and it has been used over the years by many researchers to develop models for plastic hinge length. According to this formulation, the plastic hinge length is defined as the equivalent length over which the inelastic curvature is considered to be
constant. In reality, the inelastic curvature has a certain variation. Paulay and Priestly presented the actual curvature distribution of a prismatic reinforced concrete cantilever element, showing that the real spread of plasticity is longer that the equivalent plastic hinge length (Paulay and Priestly 1992: 139).

Classical plastic hinge analysis is based on the assumption of an elasto-plastic behaviour and an equivalent plastic hinge length. The yield curvature used in Equation 2.8 can be determined considering that the moment-curvature relationship in the plastic hinge zone is elasto-plastic (Priestly and Park 1987: 71). This approximation depends on the actual shape of the moment-curvature relationship, which depends on the type of element. Figure 2.4 shows the difference between the equivalent yield curvature for a column and for a shear wall (Adebar and Ibrahim 2002: 408):

Figure 2.4 Bending moment-curvature relationship of different reinforced concrete elements


The total displacement at the top can also be determined by calculating the curvature distribution along the length of the member, as shown in Figure 2.5 (Priestly and Park 1987: 71):

Figure 2.5 Curvature distribution along the length of a reinforced concrete cantilever element


Then:

$$
\begin{equation*}
\Delta=\int_{0}^{L} \phi(x) x d x \tag{2.9}
\end{equation*}
$$

Equation 2.9 can be used to calculate the total displacement, considering a theoretical curvature distribution like the one in Figure 2.5. However, the actual displacement will be larger than the one calculated with this expression. This is due to the effect of tensile strain penetration and the spread of plasticity due to shear. Tensile strain penetration is the additional rotation in the plastic hinge zone due to the slippage of the longitudinal reinforcement. The extent of the tensile strain penetration depends on the development length of the bar (Paulay and Priestly 1992: 141). The spread of plasticity is caused by the presence of diagonal cracks due to high shear stresses, which produces higher steel strains than the ones due to pure flexure. For a crack angle of $45^{\circ}$, the influence of diagonal cracking is proportional to the depth of the member (Priestly and Park 1987: 71 - 72).

The influence of tensile strain penetration and spread of plasticity can be taken into account in Equation 2.9 by considering a modified curvature distribution, shown in Figure 2.5. However, the most common approach in practice is to use plastic hinge analysis, in which the influence of tensile strain penetration and spread of plasticity is considered implicitly in Equation 2.8 in the parameter $L_{p}$, in order to obtain a good estimate of the real displacement (Priestly and Park 1987: 72). Therefore, many of the models developed to determine the equivalent plastic hinge length in reinforced concrete elements were established on the basis that the plastic hinge length is proportional to the length of the member in the
longitudinal direction, the depth of the member, and the longitudinal reinforcement properties; as it is presented in the following expression (Berry and Eberhard 2003: 13):

$$
\begin{equation*}
L_{p}=\alpha L+\beta h+\xi_{y} d_{b} \tag{2.10}
\end{equation*}
$$

- $L:$ Length of the member in the longitudinal direction.
- $h$ : Depth or height of the cross-section of the member.
- $f_{y}$ : Yield stress of the tension reinforcement.
- $d_{b}:$ Bar diameter of the tension reinforcement.

In a more general sense, $L$ in Equation 2.10 refers to the distance from the critical section to the point of contraflexure (point of zero moment). For the cantilever element shown in Figure 2.5, this distance is equal to the length of the member. When this is not the case, a different terminology will be used. Also, when referring to walls, the length in the longitudinal dimension $(L)$ will be referred to as the height of the wall $\left(H_{w}\right)$, and the depth of the cross-section ( $h$ ) will be referred to as the wall length $\left(l_{w}\right)$.

In Equation 2.10, the member length takes into account the curvature distribution along its length, the member depth considers the spread of plasticity, and the longitudinal reinforcement properties consider the effect of tensile strain penetration (Berry and Eberhard 2003: 13). Several researchers have used the form of Equation 2.10 or other similar to it and have calibrated them with experimental results to obtain the values for $\alpha, \beta$ and $\xi$.

Most of the studies done in plastic hinge length in reinforced concrete elements have been focused in the parameters presented in Equation 2.10. However, some researchers have also investigated the stress-strain properties of steel and the effect of axial compression, which have a significant influence in the length of the plastic hinge. In the case of walls, when subjected to some axial compression, the length over which the longitudinal reinforcement yields is reduced, and so does the plastic hinge length (Paulay and Uzumeri 1975: 596-597).

In this chapter, the models developed by various researchers to determine the plastic hinge length for various kinds of concrete elements will be presented.

### 2.2 Chan (1955)

Chan performed tests on concrete columns to compare the assumption of plastic hinges concentrated at points with the real spread of plasticity. He also investigated the effect of lateral
confinement in the strain capacity of concrete. He reported the results obtained for 23 columns (Chan 1955: 121 -132).

Consider the linear bending moment diagram for a reinforced concrete cantilever element subjected to a lateral load at a certain distance from the base, like the one in Figure 2.6:

Figure 2.6 Bending moment diagram for a reinforced concrete cantilever element


Chan considered the effect of strain hardening to define the length of the plastic hinge. For a linear bending moment diagram like the one shown in Figure 2.6, the length where the yield moment is exceeded is determined by (Chan 1955: $121-122$ ):

$$
\begin{equation*}
\frac{L_{p}}{L_{s}}=1-\frac{M_{y}}{M_{u}} \tag{2.11}
\end{equation*}
$$

- $L_{s}$ : Distance from the section of maximum moment to the section of zero moment.
- $M_{y}:$ Yield moment.
- $M_{u}$ : Ultimate moment.

Three types of specimens where tested: nine members with bent-in transverse reinforcement, seven members with spiral transverse reinforcement, and seven members with welded transverse reinforcement. The most important variables covered the following range (Chan 1955: 124):

- Depth: 6".
- Width: 3 5/8 and 6".
- Span length: 11 1/2, 12 and 52".
- Diameter of tension reinforcement: $1 / 2$ and $5 / 8^{\prime \prime}$ bars.
- Diameter of compression reinforcement: $1 / 2$ and $5 / 8^{\prime \prime}$ bars.
- Diameter of transverse reinforcement: $1 / 8,3 / 16$ and $1 / 4$ " bars.
- Eccentricity of axial load: $1 / 4$ and $1 / 2^{\prime \prime}$.
- Concrete cube strength: Between 2.65 and 5.46 ksi .

Each specimen was pin-ended and was loaded until failure. The axial load in the members with bent-in and spiral transverse reinforcement was held constant. The members with welded transverse reinforcement were subjected additionally to a transverse load at the midspan (Chan 1955: 125).

The test results were used to develop stress-strain relationships for unconfined and confined concrete. Then, using these stress-strain curves, an elasto-plastic stress-strain curve for steel, a yield strain of 0.001 , and an ultimate strain of 0.0035 for unconfined concrete; the yield moment and ultimate moment were calculated for each of the specimens tested. These values were then used in Equation 2.11 to determine the length of the plastic hinge. For members with low axial loads, the plastic hinge length did not vary significantly with the tension reinforcement ratio, and had a mean value of $0.4 L_{s}$. The plastic hinge length was greater for members with high axial loads, due to concrete spalling. The maximum value was approximately $0.7 L_{s}$ (Chan 1955: 129).

### 2.3 Baker (1956)

Baker investigated the plastic deformations of hinges and members in concrete frames. He indicated that a safe estimate of the length of the plastic hinge in columns is between $0.5 h$ and $h$ (Baker 1956: 27).

### 2.4 Cohn and Petcu (1963)

Cohn and Petcu performed tests on continuous concrete beams with two spans to investigate the factors that affect the rotational capacity of plastic hinges. The most important factor studied was the percentage of steel. They reported the results obtained for 10 beams (Cohn and Petcu 1963: 282-290).

Two series of five concrete beams each were tested. The most important variables covered the following range (Cohn and Petcu 1963: 282-284):

- Depth: 8cm.
- Width: 15 cm .
- Span length: 1.5 m .
- Diameter of longitudinal reinforcement: 6, 8 and 10 mm .
- Diameter of stirrups: 6 mm .
- Average yield stress of reinforcement: Between 2550 and $3500 \mathrm{~kg} / \mathrm{cm}^{2}$.
- Average tensile strength of reinforcement: Between 3920 and $4900 \mathrm{~kg} / \mathrm{cm}^{2}$.
- Average concrete cube strength: $240 \mathrm{~kg} / \mathrm{cm}^{2}$.

Each beam was loaded with a concentrated load at a certain distance from the central support until failure. This distance was 40 cm for one of the series, and 60 cm for the other series (Cohn and Petcu 1963: 284).

The test results were used to determine the bending moments at yielding and failure. These were calculated from the measured reactions in the beams using equilibrium equations, considering a linear variation of the bending moment diagram. Then, Equation 2.11 was used to determine the length of the plastic hinge at one side from the central support of the beams (Cohn and Petcu 1963: 284-285). The results obtained varied from 30 to $90 \%$ of the effective depth of the beam (Cohn and Petcu 1963: 290).

### 2.5 Baker and Amarakone (1964)

Baker and Amarakone reported tests results on beams and columns that were performed to investigate how the following parameters influenced the moment-curvature relationship of these members: the strength of concrete and steel, percentage of steel, single loads and double loads, axial force, shear force, transverse reinforcement, and percentage of compression reinforcement. They reported the results obtained for 92 specimens (Baker and Amarakone 1964: 85-142).

Baker and Amarakone proposed the following expression to determine the plastic hinge length for members with unconfined concrete (Baker and Amarakone 1964: 94):

$$
\begin{gather*}
L_{p}=k_{1} k_{2} k_{3}\left(\frac{L_{s}}{d}\right)^{1 / 4} d  \tag{2.12}\\
k_{2}=1+0.5 \frac{P}{P_{u}}  \tag{2.13}\\
k_{3}=0.9-\frac{f_{c c}-13.8}{92} \tag{2.14}
\end{gather*}
$$

- $k_{1}$ : Factor that considers the influence of the tension reinforcement. It is equal to 0.7 for mild steel or 0.9 for cold-worked steel.
- $k_{2}$ : Factor that considers the influence of the axial load.
- $k_{3}$ : Factor that considers the influence of the concrete strength. It is equal to 0.6 for a concrete cube strength of $6 \mathrm{ksi}(41.4 \mathrm{MPa})$ or 0.9 for a concrete cube strength of $2 \mathrm{ksi}(13.8 \mathrm{MPa})$. Linear interpolation may be used between these values, arriving to Equation 2.14.
- $d$ : Effective depth of the member.
- $\quad P$ : Axial load.
- $\quad P_{u}$ : Axial compressive strength of the member without any bending moment.
- $f_{c c}$ : Concrete cube strength, in MPa units.

They indicated that for values of $L / d$ and $L_{s} / d$ commonly used in practice, the plastic hinge length is between $0.4 d$ and $2.4 d$.

Three types of specimens where tested: 32 members with cold-worked steel, 30 members with mild steel, and 32 members subjected to bending and axial load (with both cold-worked and mild steel). The most important variables covered the following range (Baker and Amarakone 1964: 88-93):

- Width: 6, 10 and $12^{\prime \prime}$.
- Depth: 8 and 11".
- Concrete cylinder strength: Between 2.5 and 6.5 ksi .
- Yield stress of reinforcement: Between 37 and 85 ksi .
- Tension reinforcement index: Between 2.5 and 70.4\%.
- Compression reinforcement index: Between 0.96 and 26.3\%.
- Transverse reinforcement ratio: Between 0.05 and $1.26 \%$.
- Axial load ratio (in this case, $P / f^{\prime}, b d$ ): Between 0.164 and 1.2.
- Span length: 55, 80, 110 and 117".

The tension and compression reinforcement index, $\omega$ and $\omega^{\prime}$, respectively, are given by:

$$
\begin{align*}
& \omega=\rho \frac{f_{y}}{f_{c}^{\prime}}  \tag{2.15}\\
& \omega^{\prime}=\rho^{\prime} \frac{f_{y}^{\prime}}{f_{c}^{\prime}} \tag{2.16}
\end{align*}
$$

- $\quad \rho$. Tension reinforcement ratio.
- $f_{c}^{\prime}$ : Concrete cylinder uniaxial compressive strength.
- $\rho^{\prime}$ : Compression reinforcement ratio.
- $f_{y}^{\prime}$ : Yield stress of the compression reinforcement.

The test results were used to compare measured and predicted inelastic rotations. Since concrete behaves differently when it is confined by transverse reinforcement, the authors developed an expression to estimate the inelastic rotation for this case (Baker and Amarakone 1964: 97):

$$
\begin{equation*}
\theta_{p}=0.8\left(\varepsilon_{c u}-\varepsilon_{c e}\right) k_{1} k_{3}\left(\frac{L_{s}}{d}\right) \tag{2.17}
\end{equation*}
$$

- $\varepsilon_{c u}:$ Concrete strain in the extreme compression fiber at ultimate curvature.
- $\varepsilon_{c e}$ : Concrete strain in the extreme compression fiber at yield curvature.

Considering that the neutral axis depth at yielding and failure are equal, Equation 2.4 can be expressed as (Baker and Amarakone 1964: 91):

$$
\begin{equation*}
\theta_{p}=\frac{\varepsilon_{c u}-\varepsilon_{c e}}{c} L_{p} \tag{2.18}
\end{equation*}
$$

Where $c$ is the neutral axis depth at ultimate moment. Therefore, combining Equation 2.17 and 2.18, the plastic hinge length for confined members is:

$$
\begin{equation*}
L_{p}=0.8 k_{1} k_{3}\left(\frac{L_{s}}{d}\right) c \tag{2.19}
\end{equation*}
$$

Although the parameter $k_{2}$ is not included in Equation 2.19, the influence of the axial load is considered implicitly in the ratio $c / d$.

The test results showed that members with cold-worked steel had longer plastic hinge lengths. The ratio $L_{s} / d$ did not have a significant effect on the plastic hinge lengths (Baker and Amarakone 1964: 116). A considerable scatter was observed for the inelastic rotation, because of the variation of the concrete strain at ultimate curvature. Baker and Amarakone indicated that the neutral axis depth has an important influence in the ultimate strain of confined concrete. Therefore, they plotted the experimental
plastic hinge lengths as a function of the ratio $c / d$, and found that there is a linear relation between these two parameters (Baker and Amarakone 1964: 95-97).

### 2.6 Sawyer (1964)

Sawyer developed a design methodology for reinforced concrete frames based on a bilinear moment-curvature relationship (Sawyer 1964: 405-431). Some of the assumptions made to develop his method were the following:

- The maximum moment at any section is equal to the ultimate moment (Sawyer 1964: 409).
- The ratio $M_{y} / M_{u}$ is equal to 0.85 . This value was adopted based on previous test results obtained for beams (Sawyer 1964: 415).
- The plasticity spreads $d / 4$ past the section in which the bending moment is equal to the yield moment (Sawyer 1964: 422).

An expression to calculate the plastic hinge length can be derived from these assumptions (Mendis 2001: 190-191), considering a linear bending moment diagram like the one shown in Figure 2.6. Based on the second assumption, the length where the yield moment is exceeded is $0.15 L_{s}$. The bilinear moment-curvature relationship developed by Sawyer is defined by the points $(0,0),\left(M_{y}, \phi_{y}\right)$ and $\left(M_{u}, \phi_{u}\right)$. Based on this, the inelastic rotation at the section of maximum moment can be obtained by integrating the inelastic curvatures over the plastic region:

$$
\begin{equation*}
\theta_{p}=\frac{1}{2}\left(0.15 L_{s}\right)\left(\phi_{u}-\phi_{y}\right)=0.075 L_{s} \phi_{p} \tag{2.20}
\end{equation*}
$$

Comparing this equation with Equation 2.4, it can be seen that they are both the same. Therefore:

$$
\begin{equation*}
L_{p}=0.075 L_{s} \tag{2.21}
\end{equation*}
$$

Including the spread of plasticity (third assumption), the plastic hinge length is given by:

$$
\begin{equation*}
L_{p}=0.25 d+0.075 L_{s} \tag{2.22}
\end{equation*}
$$

### 2.7 Mattock (1964)

Mattock performed tests on simply supported beams subjected to a concentrated load at the midspan to investigate how their rotational capacity is influenced by the concrete strength, the effective depth, the distance from the section of maximum moment to the section of zero moment, and the amount and yield stress of the reinforcement. He reported the results obtained for 37 beams (Mattock 1964: 143 181).

The most important variables covered the following range (Mattock 1964: 145 - 149):

- Concrete cylinder strength: 4 and 6 ksi.
- Yield stress of tension reinforcement: 47 and 60 ksi.
- Yield stress of compression reinforcement: 50 and 70 ksi .
- Yield stress of stirrups: 50,60 and 70 ksi .
- Width: 6".
- Effective depth: 10 and 20".
- Span length: 55, 110 and 220".
- Tension reinforcement ratio: Between 1 and 3\%.

The test results were used to determine the spread of plasticity of the beam at each side of the midspan. Mattock indicated that a considerable spread of inelastic deformation occurred beyond a distance of half the effective depth, and that it depended on the distance from the section of maximum moment to the section of zero moment (equal to half of the beam length for these beams), the effective depth, and the amount of flexural reinforcement. He used the ratio of the total inelastic rotation in the length $L_{s}, \theta_{p}$, to the inelastic rotation in the length $d / 2, \theta_{p, d / 2}$, as a measure of the spread of plasticity (Mattock 1964: 158-161).

The total inelastic rotation in each beam was obtained from the plastic deformation at the midspan. The inelastic rotation in the length $d / 2$ was obtained from the other measurements made at the midspan.

Mattock plotted the values of $\theta_{\rho} / \theta_{p, d / 2}$ as a function of the parameters previously indicated. Based on his results, he developed the following expression to determine the spread of plasticity, using a least squares fit:

$$
\begin{gather*}
\frac{\theta_{p}}{\theta_{p, d / 2}}=1+\left(1.14 \sqrt{\frac{L_{s}}{d}}-1\right)\left[1-\left(\frac{\omega-\omega^{\prime}}{\omega_{b}}\right) \sqrt{\frac{d}{0.411}}\right]  \tag{2.23}\\
\omega_{b}=\rho_{b} \frac{f_{y}}{f_{c}^{\prime}} \tag{2.24}
\end{gather*}
$$

- $L_{s}$ : Distance from the section of maximum moment to the section of zero moment, in meters.
- $d$ : Effective depth of the member, in meters.
- $\omega_{b}$ : Tension reinforcement index for balanced ultimate strength condition.
- $\rho_{b}$ : Reinforcement ratio at balanced ultimate strength condition in a member without compression reinforcement.

Mattock indicated that in Equation 2.23, the ratio of the difference between the tension and compression reinforcement index to the balanced tension reinforcement index is a measure of the strain hardening. As this ratio increases, the amount of strain hardening reduces, and so does the spread of plasticity. As he indicated, this is because the length of the plastic region is proportional to the difference between the yield moment and the ultimate moment, and this difference strongly depends on the amount of strain hardening in the tension reinforcement. This behaviour was also observed in the test results.

The measured inelastic rotations for the tests beams were compared to the inelastic rotations obtained with Equation 2.23. The inelastic rotation in the length $d / 2$ was calculated using compatibility and equilibrium equations (Mattock 1964: 163-164). The results showed that Equation 2.23 provides a lower bound for most of the beams tested.

The plastic hinge length can be determined by (Mattock 1967: 521):

$$
\begin{equation*}
\frac{\theta_{p, d / 2}}{d / 2}=\frac{\theta_{p}}{L_{p}} \tag{2.25}
\end{equation*}
$$

Therefore, combining Equation 2.23 and 2.25, the plastic hinge length is:

$$
\begin{equation*}
L_{p}=\frac{d}{2}\left\{1+\left(1.14 \sqrt{\frac{L_{s}}{d}}-1\right)\left[1-\left(\frac{\omega-\omega^{\prime}}{\omega_{b}}\right) \sqrt{\frac{d}{0.411}}\right]\right\} \tag{2.26}
\end{equation*}
$$

### 2.8 Corley (1966)

Corley performed tests on simply supported beams subjected to a concentrated load at the midspan to investigate the effect of confinement of the concrete in compression and the effect of the size of the member in their rotational capacity. He reported the results obtained for 40 beams, which were an extension of those reported by Mattock in 1964. The size of the specimens in Corley's research extended the range covered by Mattock (Corley 1966: 121-146).

The most important variables covered the following range (Corley 1966: 123-127):

- Concrete cylinder strength: 4 ksi .
- Yield stress of tension reinforcement: 60 ksi .
- Yield stress of compression reinforcement: 60 ksi .
- Yield stress of stirrups: 50 ksi .
- Width: 3, 9 and 12".
- Effective depth: 5, 10, 24 and 30".
- Span length: 36, 72, 144, 165, 240 and 330 ".
- Tension reinforcement ratio: Between 1 and $3 \%$.
- Transverse reinforcement ratio: Between 0.3 and $9 \%$.

The test results were used to determine the spread of plasticity of the beam at each side of the midspan. Corley used the ratio $\theta_{p} / \theta_{p, d / 2}$ as a measure of the spread of plasticity, as Mattock did in 1964, and used the same procedure to determine it from the test results. He also investigated the same parameters (Corley 1966: 139 -141). These are all presented in Section 2.7.

Corley plotted the values of $\theta_{p} / \theta_{p, d / 2}$ as a function of the parameters indicated by Mattock, and found that the amount of flexural reinforcement did not have a significant influence; contrary to what Mattock determined. A pronounced scatter was found for the other variables. Therefore, Corley considered that using a least squares fit was not appropriate. Based on this, he suggested a simpler expression for the spread of plasticity:

$$
\begin{equation*}
\frac{\theta_{p}}{\theta_{p, d / 2}}=1+\frac{0.064}{\sqrt{d}} \frac{L_{s}}{d} \tag{2.27}
\end{equation*}
$$

Where $L_{s}$ and $d$ are in meters. The measured inelastic rotations for the tests beams reported by Corley and Mattock were compared to the inelastic rotations obtained with Equation 2.27. The inelastic
rotation in the length $d / 2$ was calculated using compatibility and equilibrium equations (Corley 1966: 143 - 144). The results showed that Equation 2.27 provides a lower bound for most of the beams tested.

Combining Equation 2.25 and 2.27, the plastic hinge length is:

$$
\begin{equation*}
L_{p}=0.5 d+0.032 \frac{L_{s}}{\sqrt{d}} \tag{2.28}
\end{equation*}
$$

### 2.9 Mattock (1967)

Mattock, in his discussion of Corley's paper, used the measured values of the total inelastic rotation and the inelastic rotation in the length $d / 2$ that he and Corley determined, to calculate the plastic hinge length, using Equation 2.25 (Mattock 1967: 519-522). He plotted these values as a function of the distance from the section of maximum moment to the section of zero moment. Although the results showed a considerable scatter, Mattock proposed the following equation, which represents reasonably the measured data:

$$
\begin{equation*}
L_{p}=0.5 d+0.05 L_{s} \tag{2.29}
\end{equation*}
$$

### 2.10 ACI-ASCE Committee 428 (1968)

The ACI-ASCE Committee 428, on their progress report on code clauses for "Limit Design", proposed lower and upper bounds for the plastic hinge length in beams and frames (ACI-ASCE Committee 428 1968: 713-715). The plastic hinge length is between the following limits:

$$
\begin{equation*}
\operatorname{Min}\left[R_{\varepsilon}\left(\frac{d}{4}+0.03 L_{s} R_{m}\right), R_{\varepsilon} d\right]<L_{p}<R_{\varepsilon}\left(\frac{d}{2}+0.10 L_{s} R_{m}\right) \tag{2.30}
\end{equation*}
$$

In which:

$$
\begin{align*}
R_{\varepsilon} & =\frac{0.004-\varepsilon_{c e}}{\varepsilon_{c u}-\varepsilon_{c e}}  \tag{2.31}\\
R_{m} & =\frac{M_{\max }-M_{y}}{M_{u}-M_{y}} \tag{2.32}
\end{align*}
$$

- $\quad R_{\varepsilon}:$ Strain ratio.
- $\quad R_{m}$ : Moment ratio.
- $\varepsilon_{c e}$ : Concrete strain in the extreme compression fiber at yield curvature; either calculated, or assumed between 0.001 and 0.002 .
- $\varepsilon_{c u}$ : Concrete strain in the extreme compression fiber at ultimate curvature, neglecting effects of confinement, loading rate and strain gradients. It is assigned a value between 0.003 and 0.004 .
- $M_{\text {max }}$ : Maximum moment in the length of the member.

The following expression was proposed to calculate the distance from the section of maximum moment to the section of zero moment for members subjected to uniformly distributed load:

$$
\begin{equation*}
L_{s}=\frac{4 M_{\max }}{4 V_{z}+\sqrt{w_{z} M_{\max }} R_{m}} \tag{2.33}
\end{equation*}
$$

- $V_{z}$ : Shear adjacent to a concentrated load or reaction at the section of maximum moment.
- $\quad w_{z}$ : Uniformly distributed load at the section of maximum moment.


### 2.11 Priestly, Park and Potangaroa (1981)

Priestly, Park and Potangaroa performed tests on spirally-confined concrete columns to study their behaviour under seismic loading. The effect of confinement reinforcement on the ductility of column plastic hinges was investigated. They reported the results obtained for five columns (Priestly, Park and Potangaroa 1981: 181-202).

The five column specimens were of octagonal shape with the same dimensions: 600 mm diameter, a height of 3.3 m , and longitudinal reinforcement of $16-24 \mathrm{~mm}$ bars equally spaced around the circle (Priestly, Park and Potangaroa 1981: 184 - 187).

The most important variables covered the following range:

- Concrete cylinder strength: Between 26.6 and 32.9 MPa .
- Axial load: Between 1920 and 6770 kN .
- Axial load ratio $\left(P / f^{\prime}{ }_{c} A_{g}\right)$ : Between 0.237 and 0.737 .
- Yield stress of longitudinal reinforcement: 303 and 307 MPa .
- Diameter of spiral reinforcement: 10,12 and 16 mm .
- Spiral reinforcement volumetric ratio: Between 0.75 and 2.61 .
- Yield stress of spiral reinforcement: Between 280 and 423 MPa .

The columns were tested as pin-ended vertical members. A cyclic lateral load was applied to a heavily reinforced central stub located in the middle of the column span to produce a linear bending moment diagram above and below the point of application. A constant axial load was applied during the test. The tests continued until failure.

The test results were used to compare experimental plastic hinge lengths with the values obtained using the expressions proposed by Baker and Amarakone (Equation 2.19) and Corley (Equation 2.28). The experimental plastic hinge lengths were obtained from the curvature distributions over the plastic region (Priestly, Park and Potangaroa 1981: 190-192).

The experimental plastic hinge lengths were calculated by solving Equation 2.5 for $L_{p}$; setting $\phi=$ $\phi_{u}$, where $\phi_{u}$ is ultimate curvature, which for the tests is the average curvature measured on either side of the central stub. An equivalent yield curvature and yield displacement was estimated from the test results. The plastic deformation was determined using Equation 2.2, subtracting the yield displacement from the measured total displacement.

The results showed that the axial load level had very little influence on the experimental plastic hinge lengths. The models proposed by Baker and Amarakone and by Corley predict larger values than the ones measured. In order for the curvatures to be conservatively estimated, the plastic hinge length has to be underestimated (Priestly, Park and Gill 1982: 942). Therefore, these models gave results that were not conservative. For the experimental results, the plastic hinge length had an average value of approximately 0.3 h .

### 2.12 Park, Priestly and Gill (1982)

Park, Priestly and Gill performed tests on square-confined concrete columns to study their behaviour under seismic loading. This investigation was done in parallel with the experimental study carried by Priestly, Park and Potangaroa in 1981. The effect of confinement reinforcement on the ductility of column plastic hinges was also investigated here. They reported the results obtained for four columns (Priestly, Park and Gill 1982: 929-950).

The four column specimens had the same dimensions: cross-sectional area of $550 \mathrm{~mm}^{2}$, a height of 3.3 m , longitudinal reinforcement of $12-24 \mathrm{~mm}$ bars arranged symmetrically around the perimeter, and a longitudinal reinforcement ratio of 1.79\% (Priestly, Park and Gill 1982: 932-934).

The most important variables covered the following range:

- Concrete cylinder strength: Between 23.1 and 41.4 MPa .
- Axial load: Between 1815 and 4265 kN.
- Axial load ratio: Between 0.214 and 0.6 .
- Yield stress of longitudinal reinforcement: 375 MPa .
- Diameter of stirrups: 10 and 12 mm .
- Transverse reinforcement ratio: Between 1.5 and $3.5 \%$.
- Yield stress of stirrups: Between 294 and 316 MPa.

The test procedure used for the columns was very similar to the one used by Priestly, Park and Potangaroa in 1981 (Priestly, Park and Gill 1982: 937). It is presented in Section 2.11.

The test results were used to compare experimental plastic hinge lengths with the values obtained using the expressions proposed by Baker and Amarakone (Equation 2.19) and Corley (Equation 2.28), as it was done by the authors in their research from 1981 (Priestly, Park and Gill 1982: 938 - 942). The same procedure was used to determine the experimental plastic hinge lengths.

The results showed that the axial load level had very little influence on the experimental plastic hinge lengths; the mean value for these was 0.42 h . The models proposed by Baker and Amarakone and by Corley predict larger values than the ones measured, so they were not conservative (as in the research in 1981). Corley's equation was more accurate than Baker and Amarakone's equation, because it does not consider the axial load, which is in agreement with the results of this investigation. On the other hand, Baker and Amarakone research suggested that the plastic hinge length depended on the axial load. For the experimental results, the plastic hinge length can be conservatively estimated as $0.4 h$.

### 2.13 Oesterle, Aristizabal-Ochoa, Shiu and Corley (1984)

Oesterle, Aristizabal-Ochoa, Shiu and Corley conducted tests of isolated reinforced concrete structural walls subjected to inelastic load reversals to study their web crushing strength. In this study, they considered a plastic hinge length equal to the horizontal length of the wall section, $l_{w}$, (Oesterle, Aristizabal-Ochoa, Shiu and Corley 1984: 233).

### 2.14 Paulay (1986)

Paulay presented design procedures for ductile reinforced concrete walls for earthquake resistance. He indicated that the plastic hinge length was primarily a function of the wall length. Based on this, he suggested that the length of the plastic hinge is between $0.5 l_{w}$ and $l_{w}$ (Paulay 1986: 801 -802).

### 2.15 Zahn, Park and Priestly (1986)

Zahn, Park and Priestly performed tests on reinforced concrete bridge columns with different cross-sections subjected to combined axial load and bending to study their strength and ductility. As part of their work, they tested the validity of an equation derived by Priestly and Park to calculate the plastic hinge length (Equation 2.36, presented in Section 2.16). This equation was also derived from tests on concrete bridge columns, and did not take into account the axial load, since it did not have a significant effect according to the results obtained. They reported the results obtained for 14 columns (Zahn, Park and Priestly 1986).

Four types of section shapes were studied: two square sections with face loading, four square sections with diagonal loading, two octagonal sections with circular reinforcement, and six circular hollow sections. The length of all the test units was 3.9 m , and the depth (or diameter) of all the crosssections was 0.4 m (Zahn, Park and Priestly 1986: 62-63).

The most important variables covered the following range (Zahn, Park and Priestly 1986: 211 212):

- Internal diameter of hollow sections: 212, 252 and 292 mm .
- Concrete cylinder strength: Between 27 and 40.1 MPa .
- Axial load ratio: Between 0.08 and 0.43 .
- Yield stress of longitudinal reinforcement: Between 306, 337, 423 and 440 MPa .
- Longitudinal reinforcement ratio: Between 1.51 and 5.48\%.
- Yield stress of stirrups: Between 318, 340 and 466 MPa .
- Transverse reinforcement ratio: Between 0.61 and $3.52 \%$.

The test procedure used for the columns was very similar to the one used by Priestly, Park and Potangaroa in 1981 (Zahn, Park and Priestly 1986: 66-67). It is presented in Section 2.11.

The test results were used to develop an expression to calculate the equivalent plastic hinge length. The experimental plastic hinge lengths were determined separately above and below the central stub, in both directions of loading. These were determined the same way as it was done by the authors in their research from 1981 (Zahn, Park and Priestly 1986: 71-72).

The test observations showed that the inelastic curvatures spread over a longer portion of the column when the axial load was high, due to concrete spalling. Therefore, the authors concluded that the plastic hinge length is a function of this parameter. There was a tendency for the plastic hinge length to be smaller for axial loads ratios lower than 0.3. Based on this, the following expression was proposed to predict the plastic hinge length (Zahn, Park and Priestly 1986: 238-240):

$$
L_{p}=\left\{\begin{array}{l}
\left(0.08 L_{s}+6 d_{b}\right)\left(0.5+1.67 \frac{P}{f_{c}^{\prime} A_{g}}\right) ;  \tag{2.34}\\
0.08 L_{s}+6 d_{b} ;
\end{array} \frac{P}{f_{c}^{\prime} A_{g}}<0.38\right.
$$

For circular hollow columns with one ring of reinforcement and no confinement, the following equation was recommended:

$$
\begin{equation*}
L_{p}=0.06 L_{s}+4.5 d_{b} \tag{2.35}
\end{equation*}
$$

### 2.16 Priestly and Park (1987)

Priestly and Park performed tests on concrete bridge columns with different cross-sections subjected to combined axial load and bending to study their strength and ductility. The influence of the following variables in the seismic behaviour of concrete bridge columns was investigated: the axial load, the amount and yield strength of the transverse reinforcement, and the aspect ratio (Priestly and Park 1987: 61-76).

The section shapes tested were square (face and diagonally loaded), octagon with circular reinforcement (solid or hollow) and hollow square (Priestly and Park 1987: 61). In addition to these, smaller octagonal and rectangular sections were also tested (Priestly and Park 1987: 64).

The most important variables covered the following range (Priestly and Park 1987: 61):

- Depth: Between 400 and 750 mm .
- Thickness of hollow sections: 120 mm .

Two different tests were performed, one for squat columns (with an aspect ratio of two) and one for slender columns (with an aspect ratio of four). Members with square and octagonal sections were tested as squat columns. The test procedure used for these was very similar to the one used by Priestly, Park and Potangaroa in 1981. It is presented in Section 2.11. The hollow-section columns were tested as cantilever elements, a constant axial load and a cyclic lateral load at the top were applied to these (Priestly and Park 1987: 64-66).

The test results were used to develop an expression to calculate the equivalent plastic hinge length. The experimental plastic hinge lengths were determined the same way as it was done by the authors in their research from 1981. These values were correlated with the column parameters, arriving to the following expression (Priestly and Park 1987: 71-73):

$$
\begin{equation*}
L_{p}=0.08 L_{s}+6 d_{b} \tag{2.36}
\end{equation*}
$$

This equation was used to compute plastic hinge lengths for columns with different aspect ratios that were tested by other researchers (including the ones tested by the authors in 1981 and 1982), and were compared with the experimental plastic hinge lengths obtained in these tests. There was a good agreement between the experimental and predicted values for most cases. The average value for all tests was approximately $0.5 h$. The test results in this research also indicated that the plastic hinge length did not have any significant dependence on the axial load ratio, the longitudinal reinforcement ratio, and the yield stress of longitudinal reinforcement.

### 2.17 Paulay and Priestly (1992)

Paulay and Priestly proposed the following expression to estimate the plastic hinge length (Priestly, Seible and Calvi 1996: 308-309):

$$
\begin{equation*}
L_{p}=0.08 L_{s}+0.022 f_{y} d_{b} \geq 0.044 f_{y} d_{b} \tag{2.37}
\end{equation*}
$$

Where $f_{y}$ is in MPa units. The authors indicated that for commonly used beam and column dimensions, Equation 2.37 gives plastic hinge lengths of approximately $0.5 h$. They also indicated that the plastic region where special detailing requirements must be provided to have sufficient rotational capacity
is larger than the calculated equivalent plastic hinge length (Paulay and Priestly 1992: 141-142). For $f_{y}=$ 275 MPa , Equation 2.37 becomes the same as Equation 2.36.

### 2.18 Wallace and Moehle (1992)

Wallace and Moehle presented an analytical procedure to determine the need of confined boundaries in concrete walls subjected to earthquake loading. They stated that the plastic hinge length is usually between $0.5 l_{w}$ and $l_{w}$ (Wallace and Moehle 1992: 1633-1634).

### 2.19 Moehle (1992)

Moehle indicated that the equivalent plastic hinge length in reinforced concrete columns depends on the section depth, aspect ratio, bar diameter, and the axial and shear force. He stated that good correlation with experimental results may be obtained if a plastic hinge length equal to $0.5 h$ is used (Moehle 1992: 411-412).

### 2.20 Paulay and Priestly (1993)

Paulay and Priestly reported tests on ductile concrete walls of rectangular shape subjected to seismic loading to study out-of-plane buckling. In this study, they used the following expression to calculate the plastic hinge length (Paulay and Priestly 1993: 386-387):

$$
\begin{equation*}
L_{p}=0.2 l_{w}+0.044 H_{w} \tag{2.38}
\end{equation*}
$$

Where $H_{w}$ is the total height of the wall. The authors indicated that Equation 2.38 predicts conservatively the plastic hinge length, so that the curvature ductility demands are not underestimated; and that it gives a good approximation of the portion of the height of the wall over which out-of-plane buckling can occur.

### 2.21 Sasani and Der Kiureghian (2001)

Sasani and Der Kiureghian developed probabilistic displacement capacity and demand models for reinforced concrete walls. They derived a model for the plastic hinge length in concrete walls, using the test results reported by Corley in 1966 (Section 2.8) and Mattock in 1967 (Section 2.9). From the 40 beams tested back then, they selected 29 of them with effective depths greater than 0.5 m . This data was
used to estimate the parameters of the following two equations, which are normalized versions of Equation 2.28 and 2.29, respectively (Sasani and Der Kiureghian 2001: 220-221):

$$
\begin{gather*}
\frac{L_{p}}{d}=\alpha_{1}+\alpha_{2} \frac{L_{s} l_{s}^{1 / 2}}{d^{3 / 2}}+\xi_{L}  \tag{2.39}\\
\frac{L_{p}}{d}=\alpha_{1}+\alpha_{2} \frac{L_{s}}{d}+\xi_{L} \tag{2.40}
\end{gather*}
$$

- $l_{s}$ : Standard length equal to $1 \mathrm{~m}(39.4 ")$, inserted to make the model parameters dimensionless.
- $\alpha_{1}, \alpha_{2}$ : Model parameters.
- $\xi_{L}$ : Model error term.

For the 29 beams selected, the mean values of $\alpha_{1}$ and $\alpha_{2}$ for Corley's equation were 0.52 and 0.047 , respectively; and for Mattock's equation they were 0.56 and 0.051 . These results were used in Equation 2.39 and 2.40 to determine plastic hinge lengths as a function of the effective depth and of the distance from the section of maximum moment to the section of zero moment. The results obtained for both models overestimated the plastic hinge length for large values of the effective depth, which are representative of concrete walls. Therefore, the authors explored several different models, and finally arrived to the following expression:

$$
\begin{equation*}
\frac{L_{p}}{d}=\alpha_{1}+\alpha_{2} \frac{\sqrt{L_{s}} l_{s}^{3 / 2}}{d^{2}}+\xi_{L} \tag{2.41}
\end{equation*}
$$

The mean values of $\alpha_{1}$ and $\alpha_{2}$ for this equation were now 0.427 and 0.077 , respectively. These results were used in Equation 2.41 to determine plastic hinge lengths as a function of the effective depth and of the distance from the section of maximum moment to the section of zero moment. The results obtained showed that Equation 2.41 provides a better fit to the data, specially for large values of the effective depth.

### 2.22 Mendis (2001)

Mendis performed tests on simply supported beams subjected to a concentrated load at the midspan to investigate how the plastic hinge length is influenced by the amount of tension, compression and transverse reinforcement, and shear and axial forces. He reported the results obtained for 13 beams (Mendis 2001: 189-195).

The most important variables covered the following range (Mendis 2001: 192-193):

- Span length: 630, 750 and 938 mm .
- Width: 60 mm .
- Depth: 164 mm .
- Axial load: $0,50,100$ and 175 kN .
- Stirrup spacing: $30,50,75$ and 150 mm .
- Amount of tension reinforcement: 3Y12 and 4Y12 bars.
- Amount of compression reinforcement: 2Y12 and 3Y12 bars.
- Concrete cylinder strength: Between 37.4 and 57.9 MPa .

The test results were used to compare experimental plastic hinge lengths with the values obtained using the expressions proposed by Baker and Amarakone (Equation 2.12 and 2.19), Sawyer (Equation 2.22), Mattock (Equation 2.26 and 2.29), Corley (Equation 2.28), ACI-ASCE Committee 428 (Equation 2.30 ), and Park, Priestly and Gill (who suggested a value of $0.4 h$ ). Most of the experimental results were between the ACI-ASCE bounds.

The experimental results showed that the plastic hinge length increased with the length, the shearspan ratio ( $M / V h$ ) and the longitudinal reinforcement ratio; and decreased with the transverse reinforcement ratio. For the specimens tested with axial loads, the plastic hinge length was approximately $0.4 d$, meaning that it was independent of the axial load (Mendis 2001: 193). Park, Priestly and Gill arrived to this same conclusion from their research in 1982 (see Section 2.12).

Comparisons between the experimental and predicted results showed that the expression of Baker and Amarakone for unconfined concrete (Equation 2.12) overestimated the plastic hinge length by a small margin, while their equation for confined concrete (Equation 2.19) underestimated it. The equations proposed by Mattock and Corley significantly overestimated the plastic hinge length. Sawyer's equation also overestimated the plastic hinge length for many specimens. The value recommended by Park, Priestly and Gill gave good predictions for beams with axial loads.

The test results were used to derive an equation to calculate the plastic hinge length for beams without axial loads (Mendis 2001: 193-194):

$$
\begin{equation*}
L_{p}=0.25 d \frac{R F T^{0.2} S P R^{0.5}}{P E R^{0.2}} \tag{2.42}
\end{equation*}
$$

$$
\begin{gather*}
R F T=\left(\frac{A_{s t}-A_{s c}}{b d}\right) \times 100  \tag{2.43}\\
S P R=\frac{L_{s}}{d} \tag{2.44}
\end{gather*}
$$

- PER: Percentage ratio of volume of stirrups to volume of concrete core measured outside of stirrups.
- $A_{s i}$ Area of tension reinforcement.
- $A_{s c}$ : Area of compression reinforcement.
- $b$ : Width of the member.

For members with the same amount of tension and compression reinforcement (like symmetrical columns or walls), Equation 2.42 gives a plastic hinge length of zero, so it is not applicable for these cases.

Mendis also reported measured plastic hinge lengths obtained from tests on high-strength concrete (up to 80 MPa ) beams and columns with low axial load that were performed by other researchers, and compared these results with the ACI-ASCE bounds. Most of the experimental results were between these bounds, the author recommended their use to estimate the plastic hinge length for normal and high-strength concrete beams and columns with low axial loads.

### 2.23 Panagiotakos and Fardis (2001)

Panagiotakos and Fardis reported tests on reinforced concrete members subjected to uniaxial bending, with and without axial loads, to derive expressions for deformations at yielding and failure, in terms of the member geometric and mechanical properties. These members are representative of beams, columns and walls. They reported the results obtained for 1012 specimens (Panagiotakos and Fardis 2001: 135-148).

The experimental database used in this research was the following (Panagiotakos and Fardis 2001: 136-139):

- 266 specimens were representative of beams, they had unsymmetrical reinforcement and were not subjected to axial loads. All the specimens had rectangular cross-sections, except for two of them, which had T-sections. Beams with and without closely spaced stirrups were tested.
- 682 specimens were representative of columns, they had symmetrically reinforced square or rectangular cross-sections, subjected or not to axial load. Columns with and without closely spaced stirrups were tested.
- 61 specimens were representative of walls, with rectangular, barbelled or T-sections. Walls with and without confined boundaries were tested.
- 23 column specimens had diagonal reinforcement, combined or not with longitudinal bars.
- 824 specimens used hot-rolled steel, 129 specimens used heat-treated steel, and 59 specimens used brittle cold-worked steel.

The most important variables covered the following range:

- Concrete cylinder strength: Between 15 and 120 MPa .
- Axial load ratio: Between 0 and 0.95 .
- Shear-span ratio: Between 1 and 6.5 .
- Diagonal reinforcement ratio: Between 0 and $1.125 \%$.
- Hardening ratio $\left(f_{l} / f_{y}\right)$ : Between 1.1 and 1.5 (where $f_{t}$ is the tensile strength of steel).
- Steel strain at peak stress: Between 4 and $15 \%$.

Most of the specimens were tested as simple or double cantilevers, and others were tested as simply supported beams with a concentrated load applied at the midspan. They were subjected to monotonic and cyclic loading. Most tests continued until failure (Panagiotakos and Fardis 2001: 136).

The test results were used to compare measured and predicted total rotations. The results for 875 specimens subjected to monotonic or cyclic loading, for which the values of the rotations at failure were measured and where failure was due to bending, were used for the comparison. From the data used, 242 were monotonic tests and 633 were cyclic tests. All 61 walls specimens were used. There was some slippage of the longitudinal reinforcement for 703 of these tests, most of them for cyclic loading (Panagiotakos and Fardis 2001: 140).

An approach used by the authors to determine the total rotation at failure was through plastic hinge analysis (Panagiotakos and Fardis 2001: 144-147). This was done by rearranging Equation 2.8 . Setting $L=L_{s}$ and $\phi=\phi_{u}$, and dividing Equation 2.8 by $L_{s}$, an expression for the total rotation at failure is obtained:

$$
\begin{equation*}
\theta_{u}=\frac{\phi_{y} L_{s}}{3}+\left(\phi_{u}-\phi_{y}\right) L_{p}\left(1-\frac{L_{p}}{2 L_{s}}\right) \tag{2.45}
\end{equation*}
$$

Equation 2.45 requires knowing the yield and ultimate curvature, and the plastic hinge length. The yield and ultimate curvature were determined from expressions based on basic principles of mechanics, since these expressions predicted the measured curvatures well on average, although with a large scatter. With expressions for the yield and ultimate curvature combined with Equation 2.45, the total rotations at failure for the 875 tests considered were used to arrive for expressions for the plastic hinge length that provided the best fit to this data. For this, the authors decided to use the same form of the expression proposed by Paulay and Priestly in 1992 (Equation 2.37), since the parameters included in this expression were the most significant. They developed the following expressions to determine the plastic hinge length:

For cyclic loading:

$$
\begin{equation*}
L_{p, c y}=0.12 L_{s}+0.014 a_{s l} d_{b} f_{y} \tag{2.46}
\end{equation*}
$$

For monotonic loading:

$$
\begin{equation*}
L_{p, \text { mon }}=1.5 L_{p, c y}=0.18 L_{s}+0.021 a_{s l} d_{b} f_{y} \tag{2.47}
\end{equation*}
$$

- $\quad L_{p, c y}:$ Plastic hinge length for cyclic loading.
- $L_{p, \text { mon }}$ : Plastic hinge length for monotonic loading.
- $a_{s l}$ Zero-one variable. It is equal to one if slippage of the longitudinal reinforcement is possible, and zero if it is not possible.
- $f_{y}$ : Yield stress of the tension reinforcement, in MPa units.

Equation 2.46 and 2.47 were used in conjunction with Equation 2.45 to calculate the total rotation at failure for the 875 tests considered. Equation 2.37 of Paulay and Priestly was also used with Equation 2.45 to calculate the total rotations at failure. These predictions were compared with the experimental results. The quality of the predictions was very similar for these three equations, meaning that the predictions of Equation 2.45 were not very sensitive to the expression used to calculate the plastic hinge length. A considerable scatter of the results was observed.

### 2.24 Thomsen and Wallace (2004)

Thomsen and Wallace conducted tests of slender reinforced concrete walls with rectangularshaped and T-shaped cross-sections with moderate amounts of transverse reinforcement in the boundaries
to evaluate the simplified displacement-based design approach in ACI 318 1999. The premises on which displacement-based design is founded on were verified with the experimental results. They reported the results obtained for four walls (Thomsen and Wallace 2004: 618-630).

The four wall specimens included two with rectangular sections (named RW1 and RW2) and two with T-sections (named TW1 and TW2). All had an aspect ratio of three and were considered as fourstorey walls. The walls were 4 " thick, $48^{\prime \prime}$ long and $144^{\prime \prime}$ high. Additionally, the flanges in the T-shaped walls were 4 " thick and $48^{\prime \prime}$ long, and floor slabs were provided every $36^{\prime \prime}$ over the height for these walls (Thomsen and Wallace 2004: 618-619).

The most important variables covered the following range (Thomsen and Wallace 2004: 621622):

- Average concrete cylinder strength at first storey: 4.6, 4.9, 6.3 and 6.1 ksi .
- Average concrete strain at peak stress: 0.002.
- Concrete rupture strength: Between 13 and $14 \%$ of concrete cylinder strength.
- Diameter of boundary longitudinal reinforcement: \#3 bars.
- Diameter of boundary transverse reinforcement: $3 / 16^{\prime \prime}$ smooth wire.
- Diameter of distributed horizontal and vertical web reinforcement: \#2 bars.
- Average axial load ratio: 0.1.

The wall specimens were tested as cantilever elements. They were subjected to cyclic lateral displacements applied at the top, and a constant compressive axial load (Thomsen and Wallace 2004: 622 -623).

The test results were used to verify the premises on which displacement-based design for slender concrete walls is based on (Thomsen and Wallace 2004: 626-629). Some of these premises are:

- The normal strain profile along the wall length at the critical section is linear.
- The yield curvature is equal to $0.003 / l_{w}$.
- The plastic hinge length is equal to $0.5 l_{w}$.

The experimental strain profiles were obtained from measurements made along the length of the wall. The analytical strain profiles were determined using a simplified version of Equation 2.8. Setting $L$ $=H_{w}$, and considering that the plastic hinge length is very small compared to the height of the wall, Equation 2.8 becomes:

$$
\begin{equation*}
\Delta=\frac{\phi_{y} H_{w}^{2}}{3}+\left(\phi-\phi_{y}\right) L_{p} H_{w} \tag{2.48}
\end{equation*}
$$

Considering a yield curvature of $0.003 / l_{w}$ and a plastic hinge length of $0.5 l_{w}$, and solving Equation 2.48 for the total curvature, the following expression can be derived:

$$
\begin{equation*}
\phi=\left(\frac{\Delta}{H_{w}}-0.001 \frac{H_{w}}{l_{w}}+0.0015\right)\left(\frac{2}{l_{w}}\right) \tag{2.49}
\end{equation*}
$$

Then, the strain profile along the length of the wall is given by multiplying this curvature by the distance to the neutral axis. The strain profiles were computed for different drift levels $\left(\Delta / H_{w}\right)$. The curvature was computed for each drift level, and the neutral axis depth associated to that curvature was obtained from a moment-curvature analysis.

The sensitivity of the strain profile to the assumed yield curvature and plastic hinge length was then analyzed. This was done using the same procedure as before: consider different values for these parameters, plug them into Equation 2.48, consider a drift level, solve for the curvature, determine the neutral axis depth, and calculate the strain profile. Variation of the yield curvature from $0.0025 / l_{w}$ to $0.0035 / l_{w}$ had a negligible impact on the strain profiles. However, the plastic hinge length had a very significant influence. The authors compared the experimental and analytical strain profiles, considering plastic hinge lengths of $0.33 l_{w}, 0.5 l_{w}$ and $0.67 l_{w}$; and determined that plastic hinge lengths between $0.33 l_{w}$ and $0.5 l_{w}$ produced the best agreement between results. For the specimen RW2, the best agreement was found for $0.33 l_{w}$.

The experimental results also showed that the strain profiles are not linear, although the walls are slender. The greater difference between the experimental and analytical strain profiles was in the tension zone; the authors indicated that it was due to the influence of concrete cracking and slippage of the reinforcement.

### 2.25 Summary

An extensive review of the models used to calculate the equivalent plastic hinge length has been presented. These models were developed for beams, columns and walls. Details on the tests performed to arrive at these expressions have been described.

In classical plastic hinge analysis, the plastic hinge length is defined as the equivalent length over which the inelastic curvature is considered to be constant. This definition has been used by most of researchers cited to derive their models. The influence of phenomena like tensile strain penetration and spread of plasticity is considered implicitly through the plastic hinge length.

The plastic hinge length is a function of several parameters. One of the most important is the depth of the member, as it has been included in most of the models presented. Other important parameters are the span of the member, the longitudinal reinforcement properties, the axial load ratio and the strain hardening. However, most researchers have not included the same parameters in their models, due to the characteristics of the specimens tested.

Chan (1955), Cohn and Petcu (1963), Sawyer (1964) and the ACI-ASCE Committee 428 (1968) considered the effect of strain hardening. Mattock (1964) also considered this parameter along with the reinforcement properties. However, Corley (1966) found that these two parameters did not have a significant influence, based on additional test results. Mattock (1967) proposed a new expression that did not include these parameters.

Sawyer (1964), Corley (1966), Mattock (1967) and Sasani and Der Kiureghian (2001) did not include the longitudinal reinforcement properties. On the other hand, Baker and Amarakone (1964), Zahn, Park and Priestly (1986), Priestly and Park (1987), Paulay and Priestly (1992), Mendis (2001) and Panagiotakos and Fardis (2001) considered that these were important parameters.

Baker and Amarakone (1964) included the influence of the type of steel and the concrete strength. They also considered the effect of the axial load, which had a direct relation with the plastic hinge length according to the results of their research. Zahn, Park and Priestly (1986) also considered that the effect of the axial load and concrete strength was important. However, Priestly, Park and Potangaroa (1981), Park, Priestly and Gill (1982), Priestly and Park (1987), Mendis (2001) and Panagiotakos and Fardis (2001) conducted tests for members subjected to axial load and found no significant dependence between this parameter and the plastic hinge length.

The ACI-ASCE Committee 428 (1968) proposed lower and upper bounds for the plastic hinge length, instead of a one equation.

Many researchers arrived to the conclusion that a safe (lower bound) approximation for the plastic hinge length is $0.5 h$ or $0.5 l_{w}$. This is the value given in many concrete codes, including CSA A23.3.

Most of the research done in plastic hinge length has been more focused in beams and columns. Some models have been developed for walls, but have certain limitations. Oesterle, Aristizabal-Ochoa, Shiu and Corley (1984), Paulay (1986) and Wallace and Moehle (1992) suggested values for plastic hinge length between $0.5 l_{w}$ and $l_{w}$. Paulay and Priestly (1993) proposed an equation for the plastic hinge length applicable to walls, which provides conservative predictions of the curvature ductility demand. Sasani and Der Kiureghian (2001) developed plastic hinge length models for reinforced concrete walls using the test results reported by Corley (1966) and Mattock (1967). However, these tests were performed on beams, which have a different behaviour compared to walls. Panagiotakos and Fardis (2001) derived expressions for the plastic hinge length for monotonic and cyclic loading using the results obtained for 875 specimens. Only 61 of these were walls, so these models may not be representative of this type of members. Thomsen and Wallace (2004) obtained an approximation of the plastic hinge length by comparing experimental and analytical strain profiles. The best agreement was found for values between $0.33 l_{w}$ and $0.5 l_{w}$.

Table 2.1 summarizes all the studies previously done on plastic hinge length presented in this chapter, including the type of members studied and the parameters considered by the authors to develop their models:

Table 2.1 Summary of previous research done on plastic hinge length

| Researchers reference | Members studied | Parameters considered |
| :---: | :---: | :---: |
| Chan (1955) | Columns | Span, strain hardening |
| Baker (1956) | Beams and columns | Depth |
| Cohn and Petcu (1963) | Beams | Depth, strain hardening |
| Baker and Amarakone (1964) | Beams and columns | Depth, span, reinforcement <br> properties, axial load ratio |
| Sawyer (1964) | Beams and columns | Depth, span, strain hardening |
| Mattock (1964) | Beams | Depth, span, reinforcement <br> properties, strain hardening |
| Corley (1966) | Beams | Depth, span |
| Mattock (1967) | Beams | Depth, span |
| ACI-ASCE Committee 428 (1968) | Beams and columns | Depth, span, strain hardening |
| Priestly, Park and Potangaroa (1981) | Columns | Depth |
| Park, Priestly and Gill (1982) | Columns | Depth |
| Oesterle, Aristizabal-Ochoa, Shiu <br> and Corley (1984) | Walls | Depth |
| Paulay (1986) | Walls | Depth |
| Zahn, Park and Priestly (1986) | Columns | Span, reinforcement properties, <br> axial load ratio |
| Priestly and Park (1987) | Columns | Depth, span, reinforcement <br> properties |
| Paulay and Priestly (1992) | Beams and columns | Depth, span, reinforcement <br> properties |
| Wallace and Moehle (1992) | Walls | Depth |
| Moehle (1992) | Columns | Depth |
| Paulay and Priestly (1993) | Walls | Depth, span |
| Sasani and Der Kiureghian (2001) | Beams $d>0.5 m$ | Depth, span |
| Mendis (2001) | Beams and columns | Depth, span, reinforcement <br> properties |
| Panagiotakos and Fardis (2001) | Beams, columns and walls | Span, reinforcement properties |
| Thomsen and Wallace (2004) | Walls | Depth |

The models presented in this chapter will be used to determine the equivalent plastic hinge length for a test specimen representative of a shear wall in a high-rise building, in order to compare the results obtained. These results are presented in Chapter 4 (see Section 4.2.8).

## CHAPTER 3: ANALYTICAL METHODS

### 3.1 Introduction to program VecTor2

Program VecTor2 will be the analysis tool used in this research. VecTor2 is a computer program developed to perform nonlinear finite element analysis of two-dimensional reinforced concrete membrane structures subjected to quasi-static loading. This program has been developed by researchers from the University of Toronto. In this chapter, a general overview of program VecTor2 will be presented, focusing basically on the models used in this research. The information presented in this chapter comes directly from the VecTor2 and FormWorks User's Manual (Wong and Vecchio 2002) and the VecTor Analysis Group website.

Along with program VecTor2, program FormWorks is used as the pre-processor for the analysis, and program Augustus is used as the post-processor.

### 3.2 Theoretical bases of program VecTor2

The analysis was performed using the constitutive models of the Disturbed Stress Field Model (Vecchio 2000: 1070-1077), which is a refinement of the Modified Compression Field Theory (Vecchio and Collins 1986: 219-231). Both analytical models can predict the behaviour of reinforced concrete elements subjected to in-plane normal and shear stresses, modeling cracked concrete as an orthotropic material with smeared, rotating cracks. Constitutive models for a variety of effects, such as compression softening and tension stiffening, are included to accurately predict the response (Wong and Vecchio 2002: 2). Additionally, the Disturbed Stress Field Model can consider different directions of the principal stresses and strains, and takes into account crack shear slip deformations (Wong and Vecchio 2002: 13).

### 3.3 Finite element formulation

Program VecTor2 uses low-powered finite elements to model the structure. These include the 3node constant strain triangle (with six degrees of freedom) and the 4-node plane stress rectangle (with eight degrees of freedom) to model concrete, and the 2 -node truss bar element (with four degrees of freedom) to model discrete reinforcement. The reinforcement may be modeled as either smeared within concrete elements or as discrete bars. The stresses, strains and material properties are constant within each element (VecTor Analysis Group).

The finite element solution is based on a secant stiffness formulation that uses a total-load iterative procedure (VecTor Analysis Group). At each load step, the element stiffness matrices are calculated from the current stress-strain state and then assembled, and the nodal loads are calculated. The nodal displacements are determined and then used to calculate one strain tensor for each element, and then the principal strains are determined. These are used in the constitutive relationships for concrete and steel to calculate the stress tensor in each element. The secant moduli are then determined from the new stress-strain state, and compared to the secant moduli from the previous stress-strain state. If convergence is satisfactory, the analysis for that load step is completed and then continues to the next load step. If not, the analysis is repeated using the new stress-strain state. Usually, convergence is achieved after 10 to 30 iterations (Selby, Vecchio and Collins 1996: 306-307). This procedure continues until the specified force or the target displacement is reached, or until the structure becomes unstable.

### 3.4 Models for concrete in compression

The compression response of concrete is described through nonlinear functions relating stresses and strains. Different models are used for the ascending and descending branches of the concrete response in uniaxial compression. These curves are then modified to account for second-order effects (Wong and Vecchio 2002: 45).

Figure 3.1 Concrete compression response


### 3.4.1 Compression pre-peak response

Compression pre-peak response models compute the principal compressive stresses while the principal compressive strain is less than the strain corresponding to the peak compressive stress. The
model used for the ascending curve for concrete compressive stress is the one proposed by Popovics for normal-strength concrete. The stress-strain curve is described by (Wong and Vecchio 2002: 46 - 47):

$$
\begin{gather*}
f_{c 2}=-\left(\frac{\varepsilon_{c 2}}{\varepsilon_{p}}\right) f_{p} \frac{n}{n-1+\left(\varepsilon_{c 2} / \varepsilon_{p}\right)^{n}} ; \varepsilon_{p}<\varepsilon_{c 2}<0  \tag{3.1}\\
n=\frac{E_{c}}{E_{c}-E_{\mathrm{sec}}}  \tag{3.2}\\
E_{\mathrm{sec}}=\frac{f_{p}}{\left|\varepsilon_{p}\right|} \tag{3.3}
\end{gather*}
$$

- $f_{c z}$ : Average net concrete axial stress in the principal compressive direction.
- $\varepsilon_{c}$ : Average net concrete axial strain in the principal compressive direction.
- $\varepsilon_{p}$ : Concrete compressive strain corresponding to $f_{p}$.
- $f_{p}:$ Peak concrete compressive stress.
- $n$ : Curve fitting parameter for stress-strain response of concrete in compression.
- $E_{c}$ : Concrete initial tangent stiffness.
- $E_{\text {sec }}$ : Concrete secant stiffness.


### 3.4.2 Compression post-peak response

Compression post-peak response models compute the principal compressive stresses while the principal compressive strain is greater than the strain corresponding to the peak compressive stress. The compressive stresses are computed as follows (Wong and Vecchio 2002: 52):

$$
\begin{gather*}
f_{c 2}=\left(1-c_{a}\right) f_{c 2}^{a}+c_{a} f_{c 2}^{b} ; \varepsilon_{c 2}<\varepsilon_{p}<0  \tag{3.4}\\
c_{a}=4\left(\frac{f_{p}-f_{c}^{\prime}}{f_{c}^{\prime}}\right) ; 0 \leq c_{a} \leq 1 \tag{3.5}
\end{gather*}
$$

- $\quad c_{a}$ : Averaging factor.
- $f_{c 2}^{a}$ : Average concrete compressive stress contribution of unconfined concrete.
- $\quad f_{c 2}^{b}$ : Average concrete compressive stress contribution of confined concrete.

The stress contribution of unconfined concrete is determined using the Smith-Young model:

$$
\begin{equation*}
f_{c 2}^{a}=-f_{p}\left(\frac{\varepsilon_{c 2}}{\varepsilon_{p}}\right) \operatorname{Exp}\left(1-\frac{\varepsilon_{c 2}}{\varepsilon_{p}}\right) \tag{3.6}
\end{equation*}
$$

The stress contribution of confined concrete is determined using the Modified Park-Kent model, in which the stresses decay linearly (Wong and Vecchio 2002: 53-54):

$$
\begin{gather*}
f_{c 2}^{b}=-\left[f_{p}+Z_{m} f_{p}\left(\varepsilon_{c 2}-\varepsilon_{p}\right)\right]< \begin{cases}0 & ; 0<f_{p}<f_{c}^{\prime} \\
-0.2 f_{p} & ; 0<f_{c}^{\prime}<f_{p}\end{cases}  \tag{3.7}\\
Z_{m}=\frac{0.5}{\left(\frac{3+0.29\left|f_{c}^{\prime}\right|}{145\left|f_{c}^{\prime}\right|-1000}\right)\left(\frac{\varepsilon_{o}}{-0.002}\right)+\left(\frac{\left|f_{c 1}\right|}{170}\right)^{0.9}+\varepsilon_{p}} \tag{3.8}
\end{gather*}
$$

- $\quad Z_{m}$ : Slope of compression post-peak descending curve.
- $f^{\prime}$ : Concrete cylinder uniaxial compressive strength, in MPa units.
- $\varepsilon_{0}$ : Concrete compressive strain corresponding to $f_{c}^{\prime}$.
- $f_{c l}$ : Average net concrete axial stress in the principal tensile direction.


### 3.4.3 Compression softening

Compression softening is the reduction of compressive strength and stiffness of concrete due to transverse cracking and tensile straining. In VecTor2, compression softening is included by calculating a softening parameter, $\beta_{d}$, which varies between zero to one. The model used to determine this parameter is the Vecchio 1992-A ( $\varepsilon_{c 1} / \varepsilon_{c 2}$-Form). This factor is then applied to the uniaxial compressive strength and its corresponding strain to obtain the peak compressive strength and its corresponding strain, respectively; used in the compression response models previously described. The following expressions apply (Wong and Vecchio 2002: 58-61):

$$
\begin{gather*}
\beta_{d}=\frac{1}{1+C_{s} C_{d}} \leq 1  \tag{3.9}\\
C_{d}= \begin{cases}0 & ; r<0.28 \\
0.35(r-0.28)^{0.8} & ; r>0.28\end{cases}  \tag{3.10}\\
r=\frac{-\varepsilon_{c 1}}{\varepsilon_{c 2}} \leq 400  \tag{3.11}\\
f_{p}=\beta_{d} \beta_{l} f_{c}^{\prime} \tag{3.12}
\end{gather*}
$$

$$
\varepsilon_{p}= \begin{cases}\beta_{d} \beta_{l} \varepsilon_{o} & ; \beta_{d} \beta_{l} \varepsilon_{o}<\varepsilon_{o}<0  \tag{3.13}\\ \varepsilon_{c 2} & ; \varepsilon_{o}<\varepsilon_{c 2}<\beta_{d} \beta_{l} \varepsilon_{o}<0 \\ \varepsilon_{o} & ; \varepsilon_{c 2}<\varepsilon_{o}<\beta_{d} \beta_{l} \varepsilon_{o}<0\end{cases}
$$

- $C_{s}$ : Compression softening shear slip factor. It is assigned a value of one if shear slip is not considered, and 0.55 if it is considered.
- $\quad C_{d}:$ Compression softening strain softening factor.
- $r$ : Ratio of the principal tensile strain to the principal compressive strain.
- $\varepsilon_{c l}$ : Average net concrete axial strain in the principal tensile direction.
- $\beta_{i}$ : Strength enhancement factor.

The value of $\varepsilon_{p}$ may be modified in some cases, as shown in Equation 3.13, so that compression response ascends for strains up to $\varepsilon_{o}$ and then descend.

### 3.4.4 Confinement strength

Confinement increases the compressive strength and ductility of concrete. In VecTor2, the effect of confinement is included by calculating a strength enhancement factor, $\beta_{l}$, which is equal or greater than one. The model used to determine this parameter is the Kupfer-Richart model. This factor is then applied to the uniaxial compressive strength and its corresponding strain, to obtain the peak compressive strength and its corresponding strain, respectively; used in the compression response models previously described (Equations 3.12 and 3.13). For the case of biaxial compression, the strength enhancement factor for the direction of the largest compressive stress, $f_{c 2}$, is determined using the following expression (Wong and Vecchio 2002: 77-78):

$$
\begin{equation*}
\beta_{l}=1+0.92\left(\frac{-f_{c 1}}{f_{c}^{\prime}}\right)-0.76\left(\frac{-f_{c 1}}{f_{c}^{\prime}}\right)^{2} \geq 1 ; f_{c 2}<f_{c 1}<0 \tag{3.14}
\end{equation*}
$$

The strength enhancement factor for the direction of $f_{c 1}$ is determined by interchanging $f_{c 1}$ for $f_{c 2}$ in Equation 3.14.

### 3.5 Models for concrete in tension

The tension response of concrete is divided into uncracked and cracked response:

Figure 3.2 Concrete tension response


Before cracking, the response is considered linear-elastic (Wong and Vecchio 2002: 64):

$$
\begin{gather*}
f_{c 1}=E_{c} \varepsilon_{c 1} ; 0<\varepsilon_{c 1}<\varepsilon_{c r}  \tag{3.15}\\
\varepsilon_{c r}=\frac{f_{c r}}{E_{c}} \tag{3.16}
\end{gather*}
$$

- $\varepsilon_{c r}$ : Concrete cracking strain.
- $f_{c r}$ : Concrete cracking stress.

After cracking, VecTor2 does two calculations to determine the average concrete tensile stresses, one due to tension stiffening and the other due to tension softening. The larger of the two values is taken as the average post-cracking tensile stress (Wong and Vecchio 2002: 65):

$$
\begin{equation*}
f_{c 1}=\operatorname{Max}\left(f_{c 1}^{a}, f_{c 1}^{b}\right) ; 0<\varepsilon_{c r}<\varepsilon_{c 1} \tag{3.17}
\end{equation*}
$$

- $f_{c 1}^{a}$ : Average concrete tensile stress due to tension stiffening.
- $f_{c 1}^{b}$ : Average concrete tensile stress due to tension softening.


### 3.5.1 Tension stiffening

Tension stiffening is the presence of post-cracking tensile stresses between cracks in the vicinity of the reinforcement. In VecTor2, for discrete reinforcement elements, their tributary area in which the concrete within exhibits tension stiffening is delineated by a distance equal to 7.5 bar diameters from the
reinforcement element (Wong and Vecchio 2002: 64-65). Tension stiffening is included by decreasing gradually the average stress-strain response of concrete in tension. The Modified Bentz model is used for this purpose. This model is the same as the Bentz 1999 model when the steel is aligned with the $x$ or $y$ axis (no skew steel). The concrete tensile stresses decay following these expressions (Wong and Vecchio 2002: 18):

$$
\begin{gather*}
f_{c 1}^{a}=\frac{f_{c r}}{1+\sqrt{c_{t} \varepsilon_{c 1}}}  \tag{3.18}\\
c_{t}=2.2 m  \tag{3.19}\\
\frac{1}{m}=\sum_{i} \frac{4 \rho_{i}}{d_{b i}}\left|\operatorname{Cos} \theta_{n i}\right| \tag{3.20}
\end{gather*}
$$

- $c_{t}$ : Coefficient that incorporates the influence of reinforcement bond characteristics.
- $m$ : Bond parameter, in millimetres.
- $\theta_{n}$ : Angle between the normal to the crack surface and the longitudinal axis of the reinforcement.


### 3.5.2 Tension softening

Tension softening is the presence of post-cracking tensile stresses in plain concrete. It is included by decreasing gradually the average stress-strain response of concrete in tension, as it is done for tension stiffening. The linear model, which does not consider residual tensile stresses, is used for this purpose. The concrete tensile stresses decrease linearly following these expressions (Wong and Vecchio 2002: 70 -72):

$$
\begin{gather*}
\varepsilon_{c h}=\frac{2 G_{f}}{L_{r} f_{c r}} ; 1.1 \varepsilon_{c r}<\varepsilon_{c h}<10 \varepsilon_{c r}  \tag{3.21}\\
f_{c 1}^{b}=f_{c r}\left(1-\frac{\varepsilon_{c 1}-\varepsilon_{c r}}{\varepsilon_{c h}-\varepsilon_{c r}}\right) \geq 0 \tag{3.22}
\end{gather*}
$$

- $\varepsilon_{c h}$ : Characteristic strain.
- $G_{f}$ : Energy required to form a complete crack of unit area, it is assigned a value of $75 \mathrm{~N} / \mathrm{m}$.
- $L_{r}$ : Distance over which the crack is assumed to be uniformly distributed, it is assigned a value of half the crack spacing.


### 3.5.3 Cracking criterion

The concrete cracking stress usually decreases as the compressive stresses acting transversely increase, so it does not remain constant and may have a different value from the specified concrete uniaxial tensile strength, $f^{\prime}$. The Mohr-Coulomb stress model is used to calculate the cracking stress. The following expressions apply (Wong and Vecchio 2002: 80-82):

$$
\begin{gather*}
C=f_{c}^{\prime} \frac{1-\operatorname{Sin} \Phi}{2 \operatorname{Cos} \Phi}  \tag{3.23}\\
f_{c r u}=2 C \frac{\operatorname{Cos} \Phi}{1+\operatorname{Sin} \Phi}  \tag{3.24}\\
f_{c r}=f_{c r u}\left(1+\frac{f_{c 2}}{f_{c}^{\prime}}\right) ; 0.2 f_{t}^{\prime} \leq f_{c r} \leq f_{t}^{\prime} \tag{3.25}
\end{gather*}
$$

- $C$ : Cohesion.
- $\Phi$ : Internal angle of friction, it is assigned a value of $37^{\circ}$.


### 3.6 Models for slip distortions in concrete

VecTor2 can take into account crack shear slip deformations, as it is formulated in the Disturbed Stress Field Model. When crack shear slip deformations are considered, the crack shear check requirement is eliminated (Wong and Vecchio 2002: 13). The model used to calculate the shear slip is the Hybrid-I model, which combines the Lai-Vecchio stress model (a stress-based model) and the constant rotation lag model (Wong and Vecchio 2002: 88). A hybrid model computes the shear slip strains using both the stress-based model and the constant rotation lag model, and takes the greater value:

$$
\begin{equation*}
\gamma_{s}=\operatorname{Max}\left(\gamma_{s}^{a}, \gamma_{s}^{b}\right) \tag{3.26}
\end{equation*}
$$

- $\gamma_{s}$ : Shear slip strain.
- $\quad \gamma_{s}^{a}$ : Shear slip strain determined from the stress-based model.
- $\quad \gamma_{s}^{b}$ : Shear slip strain determined from the constant rotation lag model.


### 3.6.1 Stress-based model

The stress-based models relate the shear slip along the crack to the local shear stress along the crack. The shear slip strain is then computed from the shear slip with the following expression (Wong and Vecchio 2002: 85-86):

$$
\begin{equation*}
\gamma_{s}^{a}=\frac{\delta_{s}}{s} \tag{3.27}
\end{equation*}
$$

- $\delta_{s}$ : Shear slip.
- $s$ : Crack spacing.

For the Lai-Vecchio stress model, the following expressions used are to calculate the shear slip (Wong and Vecchio 2002: 87):

$$
\begin{gather*}
\delta_{s}=\delta_{s}^{*} \sqrt{\frac{\psi}{1-\psi}} \leq 2 w  \tag{3.28}\\
\delta_{s}^{*}=\frac{0.5 v_{c i, \max }+v_{c o}}{1.8 w^{-0.8}+\left(0.234 w^{-0.707}-0.2\right) f_{c c}}  \tag{3.29}\\
\psi=\frac{v_{c i}}{v_{c i, \max }}  \tag{3.30}\\
v_{c i, \max }=\frac{\sqrt{f_{c}^{\prime}}}{0.31+\frac{24 w}{a+16}}  \tag{3.31}\\
v_{c o}=\frac{f_{c c}}{30}  \tag{3.32}\\
f_{c c}=1.2 f_{c}^{\prime} \tag{3.33}
\end{gather*}
$$

- $\quad w$ : Average crack width, in millimetres.
- $v_{c i}$ Local shear stress on the crack, in MPa units.
- $v_{c i, m a x}$ : Maximum local shear stress on the crack, in MPa units.
- a: Maximum aggregate size, in millimetres.


### 3.6.2 Constant rotation lag model

The constant rotation lag models relate the post-cracking rotation of the principal stress field to the post-cracking rotation of the principal strain field using a rotation lag (Wong and Vecchio 2002: 86):

$$
\Delta \theta_{\sigma}= \begin{cases}\Delta \theta_{\varepsilon} & ;\left|\Delta \theta_{\varepsilon}\right| \leq \theta^{l}  \tag{3.34}\\ \Delta \theta_{\varepsilon}-\theta^{\prime} & ;\left|\Delta \theta_{\varepsilon}\right|>\theta^{l}\end{cases}
$$

- $\Delta \theta_{\sigma}:$ Post-cracking rotation of the principal stress field.
- $\Delta \theta_{\varepsilon}$ : Post-cracking rotation of the principal strain field.
- $\theta^{l}$ : Specified rotation lag. It is assigned a value of $10^{\circ}$ for unreinforced elements, $7.5^{\circ}$ for uniaxially reinforced elements and $5^{\circ}$ for biaxially reinforced elements.

The shear slip strain is then computed with the following expressions:

$$
\begin{gather*}
\theta_{\sigma}=\theta_{i c}+\Delta \theta_{\sigma}  \tag{3.35}\\
\gamma_{s}^{b}=\gamma_{x y} \operatorname{Cos} 2 \theta_{\sigma}+\left(\varepsilon_{y y}-\varepsilon_{x x}\right) \sin 2 \theta_{\sigma} \tag{3.36}
\end{gather*}
$$

- $\quad \theta_{\sigma}$ : Inclination of the principal stress field.
- $\theta_{i c}$ : Inclination of the principal stress field at cracking.
- $\quad \gamma_{x y}$ : Total shear strain.
- $\varepsilon_{x x}$ : Total axial strain in the $x$-direction.
- $\quad \varepsilon_{y y}$ : Total axial strain in the $y$-direction.


### 3.7 Models for reinforcement

The constitutive models for steel use nonlinear functions relating stresses and strains. Effect of strain hardening is considered (Wong and Vecchio 2002: 98).

### 3.7.1 Stress-strain response

The model used for the monotonic stress-strain response of the reinforcement is the ductile steel reinforcement model. The response is trilinear, consisting of a linear-elastic response, a yield plateau, and a linear strain-hardening phase until rupture:

## Figure 3.3 Reinforcement compression and tension response



The reinforcement stress, both in tension and compression, is given by the following expression (Wong and Vecchio 2002: 98-99):

$$
\begin{gather*}
f_{s}= \begin{cases}E_{s} \varepsilon_{s} & ;\left|\varepsilon_{s}\right| \leq \varepsilon_{y} \\
f_{y} & ; \varepsilon_{y}<\left|\varepsilon_{s}\right| \leq \varepsilon_{s h} \\
f_{y}+E_{s h}\left(\varepsilon_{s}-\varepsilon_{s h}\right) & ; \varepsilon_{s h}<\left|\varepsilon_{s}\right| \leq \varepsilon_{u} \\
0 & ; \varepsilon_{u}<\left|\varepsilon_{s}\right|\end{cases}  \tag{3.37}\\
\varepsilon_{y}=\frac{f_{y}}{E_{s}}  \tag{3.38}\\
\varepsilon_{u}=\varepsilon_{s h}+\frac{f_{u}-f_{y}}{E_{s h}} \tag{3.39}
\end{gather*}
$$

- $f_{s}$ : Reinforcement stress.
- $E_{s}$ : Initial tangent stiffness or elastic modulus of reinforcement.
- $\varepsilon_{s}:$ Reinforcement strain.
- $\varepsilon_{y}$ : Yield strain.
- $\varepsilon_{s h}:$ Strain at the onset of strain hardening.
- $E_{s h}$ : Strain hardening modulus.
- $\varepsilon_{u}:$ Reinforcement ultimate strain.
- $f_{u}$ : Ultimate strength of reinforcement.


### 3.7.2 Dowel action

Dowel action is the shear resistance offered by reinforcing bars crossing a crack. In VecTor2, dowel action models are used in conjunction with the stress-based slip distortion models for concrete. The shear resistance due to dowel action is calculated as a function of the shear slip at the crack. This shear resistance is then subtracted from the local shear stress on the crack, which reduces the shear slip. The model used to determine the shear resistance due to dowel action is the Tassios model. The following expressions are used to determine the shear resistance (Wong and Vecchio 2002: 103-104):

$$
\begin{gather*}
V_{d}=E_{s} I_{z} \lambda^{3} \delta_{s} \leq V_{d u}  \tag{3.40}\\
I_{z}=\frac{\pi d_{b}^{2}}{64}  \tag{3.41}\\
\lambda=\sqrt[4]{\frac{k_{c} d_{b}}{4 E_{s} I_{z}}}  \tag{3.42}\\
k_{c}=\frac{127 c_{b} \sqrt{f_{c}^{\prime}}}{d_{b}^{2 / 3}}  \tag{3.43}\\
V_{d u}=1.27 d_{b}^{2} \sqrt{f_{c}^{\prime} f_{y}}  \tag{3.44}\\
v_{d}=\frac{\rho_{s} V_{d}}{A_{s}} \tag{3.45}
\end{gather*}
$$

- $\quad V_{d}$ : Dowel force.
- $\quad I_{z}$ : Moment of inertia of the reinforcement.
- $\quad \lambda$ : Parameter that compares the stiffness of the concrete to the stiffness of the reinforcing bar.
- $\quad V_{d u}$ : Ultimate dowel force.
- $k_{c}$ : Stiffness of the notional concrete foundation.
- $\quad c_{b}$ : Coefficient used to reflect bar spacing, it is assigned a value of 0.8.
- $\quad v_{d}$ : Shear resistance due to dowel action.
- $\rho_{s}$ : Reinforcement ratio.
- $A_{s}$ : Area of reinforcement.


## CHAPTER 4: COMPARISON OF ANALYTICAL PREDICTIONS WITH ISOLATED WALL TEST RESULTS

### 4.1 Scope of analysis

Program VecTor2 was used to analyze isolated walls specimens that have been previously tested. The analytical predictions were compared with the test results available in order to see how well VecTor2 predicts the response of these members. In particular, the curvature distribution and the strain profile at critical sections were examined, since these results were then used to estimate plastic hinge lengths. The objective of this analysis was to test the validity of the results provided by VecTor2, so that the program can then be used to perform a parametric study of concrete walls. Also, the actual curvature distributions and strain profiles were investigated.

Two wall specimens were analyzed. The first is a high-rise shear wall tested at the University of British Columbia in 2000 (Adebar, Ibrahim and Bryson 2004). The second is one of the rectangular shear walls tested at Clarkson University in 1995 (Thomsen and Wallace 1995); the wall analyzed is the rectangular specimen RW2 (see Section 2.24).

### 4.2 High-rise shear wall tested at the University of British Columbia

A test was conducted on a large-scale model of a slender reinforced concrete cantilever shear wall from the core of a high-rise building at the University of British Columbia in 2000 by Adebar, Ibrahim and Bryson. The wall had an aspect ratio of 7.2, had a flanged cross-section, had a low amount of vertical reinforcement, and was subjected to a constant compressive axial load. The purpose of this test was to investigate the influence of cracking on the effective stiffness of the wall used for seismic analysis. Extensive concrete strains measurements were made at the faces of the wall over the cracked region (Adebar, Ibrahim and Bryson 2004: 1-2).

### 4.2.1 Description of the wall specimen

The specimen was a $1 / 4$ scale model of a typical wall from a high-rise building. It was 12200 mm high and 1625 mm long, with a flanged cross-section. The web was 1219 mm long and 127 mm thick, and the two flanges were 203 mm long and 380 mm thick (Adebar, Ibrahim and Bryson 2004: 4 - 5).

The reinforcement was arranged symmetrically. The vertical reinforcement in the two flanges consisted of $5-10 \mathrm{M}$ reinforcing bars enclosed by 10 M ties, spaced at 64 mm in the lower 3 m of the wall
and then spaced at 152 mm over the remaining height of the wall. The vertical reinforcement in each flange was arranged in two layers, the exterior layer had three bars and the other layer had two bars. The clear cover of the ties was 6 mm . The web had 10 M reinforcing bars spaced at 305 mm , vertically and horizontally. A total of four bars were used in the web.

High-rise buildings typically have large perimeter walls below grade that are much larger than the tower walls. These are connected together by diaphragm action of the concrete slabs. Because of this, the section where the maximum bending moment occurs in the tower walls is usually at grade level, and there is no pullout of the vertical reinforcement from the foundation. To simulate this effect, the critical section was located at 426 mm from the base, providing additional vertical reinforcement below a construction joint at this location (Adebar, Ibrahim and Bryson 2004: 5).

The wall specimen was tested in a horizontal position because of the limited height of the laboratory. The base of the wall was post-tensioned to the floor to prevent movement during the test (Adebar, Ibrahim and Bryson 2004: 5-6).

The average cylinder compressive strength was 49 MPa at the time of testing. The average yield strength of the reinforcing steel was 455 MPa , and the average ultimate strength was 650 MPa . The stress-strain response from four bar samples was measured, the yield plateau was very short (Adebar, Ibrahim and Bryson 2004: 6).

### 4.2.2 Instrumentation

A cyclic lateral load was applied at the top of the wall by a hydraulic actuator. Dywidag bars were used to apply a uniformly distributed axial compression load. The wall displacement at the top was measured using linear variable differential transducers. Displacement transducers were also provided to measure movement of the base of the wall (Adebar, Ibrahim and Bryson 2004: 6-7).

Twelve metal targets were glued to the concrete surface over the lower 5.08 m of the wall height along each face to measure average concrete strains. Targets named TE1 to TE12 were located at the east face, and targets named TW1 to TW12 were located at the west face. The relative vertical displacements of the targets were measured with a large digital calliper. On each side of the wall, 10 targets located above the construction joint (TE3 to TE12 and TW3 to TW12) were used to measure strains over the cracked region. Targets TE3 and TW3 were located 59 mm above the construction joint, and the rest were spaced at 500 mm approximately. The two bottom targets located below the construction joint were more
closely spaced. Targets TE2 and TW2 were located 134 mm below the construction joint, and targets TE1 and TW1 were located just above the base (Adebar, Ibrahim and Bryson 2004: 32).

### 4.2.3 Test procedure

A constant compressive axial load of $0.1 A_{8} f^{\prime}$, equal to 1500 kN , was applied to the wall during the test. This load was applied through two hydraulic actuators located below the base of the wall that pulled the Dywidag bars.

The cyclic lateral load at the top was applied at 11.76 m from the base of the wall; that is, 11.33 m from the construction joint. The wall was subjected to 13 displacement levels, which varied from 15 to 300 mm . East was considered the positive direction of loading, and west the negative direction of loading. At each displacement level, four complete displacement cycles (zero, then maximum positive, then maximum negative, and finally zero) were performed (Adebar, Ibrahim and Bryson 2004: 6-7).

### 4.2.4 Test results

The test results were used to determine curvature distributions along the height of the wall for different lateral displacement levels. The curvatures were determined from the axial strains at the faces of the wall, which were determined from the measured relative axial displacements of the targets. The lateral displacements were corrected so that they did not include the effect of the base rotation. Although the authors determined experimental curvatures for the wall, the curvatures presented in this section were recalculated from the original measured data, and differ from the results reported by the authors.

During the test, some rotation at the base of the wall was measured. The total displacements included the lateral displacement component (wall displacement) and the rigid body motion of the wall due to base rotation (Adebar, Ibrahim and Bryson 2004: 8). Measurements from displacement transducers at the base were used to determine displacements due to base rotation to separate these two components. For each total displacement level, the corresponding wall displacement in each direction of loading was different (Adebar, Ibrahim and Bryson 2004: 27).

The first cracking in the wall occurred at a wall displacement of 21 mm (total displacement of 30 mm ) in the positive direction, and yielding of the vertical reinforcement at the construction joint occurred at a wall displacement of 46 mm (total displacement of 60 mm ) in the positive direction (Adebar, Ibrahim and Bryson 2004: 9). The maximum wall displacement was 281 mm (total displacement of 300 mm ) in the positive direction (Adebar, Ibrahim and Bryson 2004: 11).

The relative axial displacements of the targets in the compression and tension face of the wall were measured at the peak displacement of the first cycle for each wall displacement level. The difference between the vertical distance between targets at a particular displacement level and the initial vertical distance between targets (zero reading), divided by this initial distance, gives the average axial strain over that length at the faces of the wall for that displacement level. Then, the difference between the compression and tension strain, divided by the wall length between the two faces, gives the average curvature over the length between targets (Adebar, Ibrahim and Bryson 2004: 12-13). This procedure was used to calculate the curvature distribution at total displacement levels of 120,150 and 200 mm ; for which the corresponding wall displacement levels were the following:

- Wall displacements of 105,132 and 182 mm , respectively; in the east (positive) direction of loading.
- Wall displacements of 104,138 and 187 mm , respectively; in the west (negative) direction of loading.

A significant portion of the concrete cover fell off at a total displacement level of 300 mm and some targets were lost, so the curvature distribution could not be determined for this displacement level (Adebar, Ibrahim and Bryson 2004: 14).

Regarding the zero readings, several target measurements were made before and after the axial compression was applied. The average of all these readings was taken as the zero reading used in the calculations. Measurements that were significantly out of range were neglected to determine the zero readings.

Figure 4.1 shows the experimental curvature distributions for the wall displacement levels mentioned in the east direction:

Figure 4.1 Experimental curvatures (pushing east) for different wall displacement levels


Figure 4.2 shows the experimental curvature distributions for the wall displacement levels mentioned in the west direction:

Figure 4.2 Experimental curvatures (pushing west) for different wall displacement levels


As expected, the maximum curvature is located at the construction joint; that is, at approximately 426 mm from the base. There were differences between the curvatures for both directions of loading; these become very significant near the construction joint for the higher displacement levels. This is because the wall was first pushed to the east and then to the west, so the crack pattern was not symmetrical (Adebar, Ibrahim and Bryson 2004: 10).

An experimental moment-curvature relationship for the wall was developed by the authors, in order to estimate the yield curvature. The yield curvature was taken as the curvature prior to the yield plateau, which was approximately equal to $0.002 \mathrm{rad} / \mathrm{m}$ (Adebar, Ibrahim and Bryson 2004: 15).

### 4.2.5 Analytical model of the wall specimen

The prototype wall was modeled and analyzed using program VecTor2. The purpose of this analysis is to predict the curvature distribution along the height of the wall and compare these predictions with the experimental results. The strain profile along the length of the wall was also studied. Lowpowered rectangular and triangular elements were used to model the concrete, with smeared steel to account for the presence of reinforcement. The constitutive models for concrete and steel described in Chapter 3 were used in the analysis.

Three different material types were used to represent various regions of the wall in the finite element model:

- The first material type was used to represent the flanges in the lower 3 m of the wall, in which the ties were spaced at 64 mm .
- The second material type was used to represent the web of the wall.
- The third material type was used to represent the flanges over the remaining height of the wall, in which the ties were spaced at 152 mm .

As described in Section 4.2.1, the wall had a construction joint at 426 mm from the base; and the maximum curvature occurs at this location, as shown in Figure 4.1 and 4.2. For this reason, the wall was modeled from the construction joint upwards. The horizontal and vertical displacements at the bottom in the analytical model were restrained.

The finite element mesh was more refined near the base (critical section), in order to have a good representation of the strain profile along the length of the wall. The mesh was refined in such a way that the position of certain nodes coincides with the position of the targets. Meshes of $68 \times 59,68 \times 87,87 \times 59$
and $87 \times 87$ rectangular elements were used; 20 elements ( 21 nodes) were used in the transverse direction of the wall. This level of refinement was maintained up to approximately half the height of the wall. Up from this point, 16 elements ( 17 nodes) were used in the transverse direction, and then it was further reduced to eight elements (nine nodes) up to the top of the wall. The transitions were made using triangular elements. All nodes and elements were numbered in the horizontal (short) direction. The complete mesh consisted of 1487 nodes, 1360 rectangular elements and 60 triangular elements.

The material properties used in the analysis were those reported in the description of the wall specimen, presented in Section 4.2.1. For the material properties that were not measured during the test, the values given by default in program VecTor2 were used. For the concrete properties, VecTor2 uses the following expressions to determine the tensile strength, the initial tangent elastic modulus and the cylinder strain at $f_{c}^{\prime}$ (Wong and Vecchio 2002: 146):

$$
\begin{gather*}
f_{t}^{\prime}=0.33 \sqrt{f_{c}^{\prime}}  \tag{4.1}\\
E_{c}=5500 \sqrt{f_{c}^{\prime}}  \tag{4.2}\\
\varepsilon_{o}=1.8+0.0075 f_{c}^{\prime} \tag{4.3}
\end{gather*}
$$

Where $f_{c}^{\prime}, f_{t}^{\prime}$ and $E_{c}$ are in MPa units, and $\varepsilon_{o}$ is in $\mathrm{mm} / \mathrm{m}$. The maximum aggregate size given by default is 10 mm . For the reinforcement properties, a modulus of elasticity of 200000 MPa was considered; while the strain hardening modulus and the strain at the onset of strain hardening were determined from the measured stress-strain curve.

A monotonic lateral load was applied at the top of the wall. This load was applied in a displacement-control mode, in increments of 0.2 mm . Although the actual wall was subjected to cyclic loading, applying a monotonic load seems reasonable, since the envelopes for the monotonic and cyclic response were almost the same. Additionally, the constant axial load of 1500 kN was applied; this load was equally distributed among all the nine nodes at the top. The self-weight of the wall was not considered.

Figure 4.3 shows the finite element model of the wall specimen, created in the pre-processor FormWorks:

Figure 4.3 Finite element model of UBC wall in FormWorks


Table 4.1 shows the material properties in the different regions of the wall:

Table 4.1 Material properties of UBC wall model

| Concrete properties | Material 1 |  | Material 2 |  | Material 3 |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Color |  |  |  |  |  |  |
| Thickness (mm) | 380 |  | 127 |  | 380 |  |
| Cylinder compressive strength ( MPa ) | 49 |  | 49 |  | 49 |  |
| Reinforcement component properties | $\boldsymbol{x}$-direction | $\boldsymbol{y}$-direction | $\boldsymbol{x}$-direction | $\boldsymbol{y}$-direction | $x$-direction | $y$-direction |
| Reinforcement ratio (\%) | 0.831 | 0.65 | 0.259 | 0.259 | 0.346 | 0.65 |
| Reinforcement diameter ( mm ) | 10 | 10 | 10 | 10 | 10 | 10 |
| Yield strength (MPa) | 455 | 455 | 455 | 455 | 455 | 455 |
| Ultimate strength (MPa) | 650 | 650 | 650 | 650 | 650 | 650 |
| Elastic modulus (MPa) | 200000 | 200000 | 200000 | 200000 | 200000 | 200000 |
| Strain hardening modulus (MPa) | 4875 | 4875 | 4875 | 4875 | 4875 | 4875 |
| Strain hardening strain ( $\mathrm{mm} / \mathrm{m}$ ) | 10 | 10 | 10 | 10 | 10 | 10 |

### 4.2.6 Analytical results

The analytical predictions obtained from program VecTor2 were used to determine curvature distributions along the height of the wall for different lateral displacement levels. The curvatures were determined the same way as it was done for the experimental results in order to make comparisons, using the vertical nodal displacements at the faces of the wall obtained from the analysis. Since the model of the wall was fixed at the base, the total displacement and wall displacement are the same.

As previously described, the position of certain nodes at the faces of the wall coincides with the position of the targets. The vertical nodal displacements at these particular nodes were used to calculate the curvatures. The difference in vertical displacements between two consecutive nodes at the face, divided by the vertical distance between these nodes, gives the average axial strain over that length. Then, the difference between the compression and tension strain, divided by the wall length between the two faces, gives the average curvature over the length between nodes. The curvature distribution was determined for the following wall displacement levels:

- Wall displacements of 105,132 and 182 mm ; to be compared with experimental curvatures when the wall is pushed in the east direction.
- Wall displacements of 104,138 and 187 mm ; to be compared with experimental curvatures when the wall is pushed in the west direction.

This way, the experimental and analytical average curvatures were calculated over the same lengths. The only exception is near the construction joint, due to the fact that for the analysis, the wall was modeled from the construction joint upwards. Therefore, the analytical curvature at this location was calculated for a shorter length than for the experimental curvature; this length goes from the location of the construction joint to the location of targets TE3 and TW3, located 59 mm above the construction joint.

### 4.2.7 Comparison of experimental and analytical results

Figure 4.4 to 4.9 show comparisons between the experimental and analytical curvature distributions for the wall displacement levels mentioned. Note that the vertical axis shows the distance from the construction joint, not the height of the wall:

Figure 4.4 Experimental (pushing east) and analytical curvatures for a wall displacement of 105 mm


Figure 4.5 Experimental (pushing west) and analytical curvatures for a wall displacement of $\mathbf{1 0 4 m m}$


Figure 4.6 Experimental (pushing east) and analytical curvatures for a wall displacement of 132 mm


Figure 4.7 Experimental (pushing west) and analytical curvatures for a wall displacement of 138 mm


Figure 4.8 Experimental (pushing east) and analytical curvatures for a wall displacement of $\mathbf{1 8 2 m m}$


Figure 4.9 Experimental (pushing west) and analytical curvatures for a wall displacement of 187 mm


There is a good agreement between the experimental and analytical curvatures prior to yielding for most wall displacement levels. For the higher wall displacement levels, there is a better agreement between the analytical curvatures and the experimental curvatures in the east direction, than with the experimental curvatures in the west direction. A reason for these discrepancies is that the model was constructed from the construction joint upwards and it was fixed at the base, so it is stiffer than the actual wall. Despite these differences, the analytical model seems to predict reasonably well the curvature distribution of the wall.

### 4.2.8 Equivalent plastic hinge length

The models presented in Chapter 2 will be used to compute the equivalent plastic hinge length for the test wall, in order to compare the results obtained and see how accurate these models are. The relevant wall parameters required for these calculations are shown in Table 4.2:

Table 4.2 UBC wall parameters

| Parameter | Symbol | Value |
| :--- | :---: | :---: |
| Height of wall (m) | $L_{s}=H_{w}$ | 11.33 |
| Horizontal length of wall (m) | $h=l_{w}$ | 1.625 |
| Bar diameter of tension reinforcement (mm) | $d_{b}$ | 10 |
| Yield stress of tension reinforcement $(\mathrm{MPa})$ | $f_{y}$ | 455 |
| Concrete cylinder compressive strength $(\mathrm{MPa})$ | $f_{c}$ | 49 |
| Axial load (kN) | $P$ | 1500 |
| Gross area of cross-section $\left(\mathrm{mm}^{2}\right)$ | $A_{g}$ | 309093 |

Baker and Amarakone's equation for unconfined concrete was used (Equation 2.12). Regarding Mattock's equation (Equation 2.26), the second term in parenthesis was neglected, since the wall has the same amount of reinforcement in both flanges ( $\rho=\rho^{\prime}$ ). Regarding the bounds proposed by the ACIASCE Committee 428 (Equation 2.30), since the plastic hinge length is being calculated at failure, the maximum moment is equal to the ultimate moment $\left(R_{m}=1\right)$. In the equation proposed by Sasani and Der Kiureghian (Equation 2.41), the mean values of the model parameters were used, and the model error term was not included ( $\xi_{L}=0$ ). Panagiotakos and Fardis' equation for cyclic loading was used (Equation 2.46), and the second term in the equation was neglected, as it was considered that no significant slippage occurred $\left(a_{s l}=0\right)$. Other required parameters in the calculations are presented in Table 4.3:

Table 4.3 Parameters required in plastic hinge length models

| Parameter | Symbol | Value | Comments |
| :--- | :---: | :---: | :--- |
| Effective depth (m) | $d$ | 1.3 | Calculated as 0.8l <br> to CSA A23.3 |
| Area of reinforcement $\left(\mathrm{mm}^{2}\right)$ | $A_{s}$ | 1400 | $14-10 \mathrm{M}$ bars in the wall |
| Area of concrete $\left(\mathrm{mm}^{2}\right)$ | $A_{c}$ | 307693 |  |
| Axial compressive strength without <br> bending moment (kN) | $P_{u}$ | 15714 | Calculated without using <br> strength reduction factors |
| Concrete cube strength (MPa) | $f_{c c}$ | 58.8 | Calculated with equation 3.33 |
| Tension reinforcement factor (Baker <br> and Amarakone's equation) | $k_{l}$ | 0.7 | Mild steel |
| Axial load factor (Baker and <br> Amarakone's equation) | $k_{2}$ | 1.048 | Calculated with equation 2.13 |
| Concrete strength factor (Baker and <br> Amarakone's equation) | $k_{3}$ | 0.411 | Calculated with equation 2.14 |
| Concrete strain in extreme compression <br> fiber at yield curvature | $\varepsilon_{c e}$ | 0.0015 | Average of recommended <br> values |
| Concrete strain in extreme compression <br> fiber at ultimate curvature | $\varepsilon_{c u}$ | 0.0035 | According to CSA A23.3 |
| Strain ratio (ACI-ASCE equation) | $R_{\varepsilon}$ | 1.25 | Calculated with equation 2.31 |

The equivalent plastic hinge length can also be determined experimentally from the test results available. Although the authors determined experimental plastic hinge lengths, the one presented in this
section was recalculated from the measured data, and differs from the results reported by the authors. In Equation 2.6, setting $L=H_{w}, \Delta=\Delta_{u}$ and $\phi=\phi_{u}$, this equation becomes:

$$
\begin{equation*}
\Delta_{u}=\Delta_{y}+\left(\phi_{u}-\phi_{y}\right) L_{p}\left(H_{w}-\frac{L_{p}}{2}\right) \tag{4.4}
\end{equation*}
$$

Solving Equation 4.4 for $L_{p}$ :

$$
\begin{equation*}
L_{p}=H_{w}-\sqrt{H_{w}^{2}-2\left(\frac{\Delta_{u}-\Delta_{y}}{\phi_{u}-\phi_{y}}\right)} \tag{4.5}
\end{equation*}
$$

As it was mentioned in Section 4.2.4, the maximum wall displacement was 281 mm , the wall yield displacement was 46 mm , and the yield curvature was $0.002 \mathrm{rad} / \mathrm{m}$. Because some targets were lost at the ultimate displacement level, the ultimate curvature could not be determined experimentally. Therefore, a plane sections analysis of the wall was performed to predict the ultimate curvature. The analysis was done considering an elasto-plastic stress-strain curve for steel, the concrete cylinder strength, a concrete ultimate strain of 0.0035 , a compressive axial load of 1500 kN , the CSA A23.3 stress block for concrete in compression, and strength reduction factors of unity. The estimated curvature capacity was $0.0236 \mathrm{rad} / \mathrm{m}$. Using all these values in Equation 4.5, the equivalent plastic hinge length was 1 m or $0.62 l_{w}$.

The predictions obtained are summarized in Table 4.4:

Table 4.4 Comparison of equivalent plastic hinge length models

| Researchers reference | Equation | $L_{p}(\mathrm{~m})$ | $L_{p} / l_{w}$ |
| :---: | :---: | :---: | :---: |
| Baker and Amarakone (1964) | Eq. 2.12 | 0.67 | 0.41 |
| Sawyer (1964) | Eq. 2.22 | 1.17 | 0.72 |
| Mattock (1964) | Eq. 2.26 | 2.19 | 1.35 |
| Corley (1966) | Eq. 2.28 | 0.97 | 0.60 |
| Mattock (1967) | Eq. 2.29 | 1.22 | 0.75 |
| ACI-ASCE lower bound (1968) | Eq. 2.30 | 0.83 | 0.51 |
|  |  | 1.37 |  |
| ACI-ASCE upper bound (1968) | Eq. 2.34 | 0.64 | 0.40 |
| Zahn, Park and Priestly (1986) | E. |  |  |
| Priestly and Park (1987) | Eq. 2.36 | 0.97 | 0.59 |
| Paulay and Priestly (1992) | Eq. 2.37 | 1.01 | 0.62 |
| Paulay and Priestly (1993) | Eq. 2.38 | 2.34 | 1.44 |
| Sasani and Der Kiureghian (2001) | Eq. 2.41 | 0.75 | 0.46 |
| Panagiotakos and Fardis (2001) | Eq. 2.46 | 1.36 | 0.84 |
| CSA A23.3 |  | 0.81 | 0.50 |
| Experimental |  | 1.00 | 0.62 |

Figure 4.10 shows these results graphically:

Figure 4.10 Comparison of equivalent plastic hinge length models


The results obtained using the models for plastic hinge length presented in Table 4.4 vary from $0.4 l_{w}$ to $1.44 l_{w}$. The plastic hinge length given by the code is also presented. Most of these equations estimate a plastic hinge greater than the one given by the code; therefore, the code is conservative compared to most of these. The plastic hinge length obtained experimentally is also larger than the one predicted by the code.

Comparing the experimental and predicted results from Figure 4.10, the equations of Mattock (1964), ACI-ASCE upper bound (1968) and Paulay and Priestly (1993) significantly overestimate the plastic hinge length. The equations of Sawyer (1964), Mattock (1967) and Panagiotakos and Fardis (2001) also overestimate it, but by a smaller margin. The equations of Baker and Amarakone (1964), ACI-ASCE lower bound (1968), Zahn, Park and Priestly (1986) and Sasani and Der Kiureghian (2001) underestimate the plastic hinge length. The best predictions are given by the equations of Corley (1966), Priestly and Park (1987) and Paulay and Priestly (1992).

The experimental plastic hinge length is between the bounds proposed by the ACI-ASCE Committee 428. The lower bound gives a better prediction.

The results provided by all these expressions are significantly different. This is because they have been derived from tests for different types of concrete members and consider different parameters. The last three equations in Table 4.4 (Equation 2.38, 2.41 and 2.46) where derived specifically for walls. These, however, are not providing good predictions compared to other equations.

### 4.2.9 Distribution of inelastic curvatures

Figure 4.4 to 4.9 show the actual distribution of the inelastic curvatures. Since the elastic portion of the curvature was $0.002 \mathrm{rad} / \mathrm{m}$, the inelastic curvatures can be visualized by shifting the vertical axis by this amount. The resulting curvatures suggest that the inelastic curvature over the plastic hinge length are not uniform, as it is commonly assumed; but have a linear variation over a distance equal to approximately the length of the wall $(1625 \mathrm{~mm})$ measured from the construction joint, which is the critical section (Adebar, Ibrahim and Bryson 2004: 15-16). Both the experimental and analytical results are showing this trend.

As seen in Chapter 2, in the classical formulation of plastic hinge analysis, the inelastic curvature is considered to be constant over the equivalent plastic hinge length. This equivalent length is commonly assumed to be $0.5 h$ or $0.5 l_{w}$, according to typical concrete codes and several researchers. However, the actual inelastic curvature has a certain variation. The test results and the analytical predictions from this study are suggesting that this variation may be considered linear over a length of 1.0 h or $1.0 l_{w}$; that is, twice the length normally considered.

If we consider $L_{p, \text { lin }}$ to be the plastic hinge length for a linearly varying inelastic curvature, the resulting inelastic displacement at the top of a cantilever wall can be determined from the following formulation:

## Figure 4.11 Linearly varying inelastic curvatures



The inelastic rotation can be determined by integrating the inelastic curvatures:

$$
\begin{equation*}
\theta_{p}=\frac{\phi_{p} L_{p, l i n}}{2} \tag{4.6}
\end{equation*}
$$

Considering that the inelastic rotation is concentrated at the centroid of the inelastic curvatures, the inelastic displacement can be expressed as:

$$
\begin{equation*}
\Delta_{p}=\theta_{p}\left(H_{w}-\frac{L_{p, l i n}}{3}\right)=\left(\phi-\phi_{y}\right) \frac{L_{p, l i n}}{2}\left(H_{w}-\frac{L_{p, l i n}}{3}\right) \tag{4.7}
\end{equation*}
$$

If we consider $L_{p, \text { const }}$ to be the plastic hinge length for a constant inelastic curvature, and setting $L_{p, \text { lin }}=2 L_{p, \text { const, }}$ Equation 4.7 can be expressed as:

$$
\begin{equation*}
\Delta_{p}=\left(\phi-\phi_{y}\right) L_{p, \text { const }}\left(H_{w}-\frac{2 L_{p, \text { const }}}{3}\right) \tag{4.8}
\end{equation*}
$$

Compare this inelastic displacement with the one obtained for a uniform inelastic curvature:

$$
\begin{equation*}
\Delta_{p}=\left(\phi-\phi_{y}\right) L_{p, \text { const }}\left(H_{w}-\frac{L_{p, c o n s t}}{2}\right) \tag{4.9}
\end{equation*}
$$

If the plastic hinge length is small compared to the height of the wall, both formulations give the same inelastic displacement. However, if this is not the case, the inelastic displacement is greater if a uniform inelastic curvature is considered.

Although many researchers have considered a uniform inelastic curvature to develop their models for plastic hinge length, some have studied the actual curvature distribution in concrete elements. Paulay and Priestly plotted the actual curvature distribution of a prismatic reinforced concrete cantilever element. They showed that the actual extent of plasticity is approximately twice the equivalent plastic hinge length obtained considering a uniform inelastic curvature, and that the actual variation of the inelastic curvature can be reasonably considered as linear; as it was observed in this study (Paulay and Priestly 1992: 139).

In the next chapter, a linearly varying inelastic curvature will be considered to perform a parametric study of concrete walls.

### 4.2.10 Analysis of strain profiles

So far, the length of the plastic hinge has been obtained from the curvature distribution. The concept of curvature (strain gradient) is based on the hypothesis that plane sections remain plane after bending, which is a widely used engineering assumption. During the test, strain measurements were only made at the faces of the wall, and the curvatures were calculated assuming that the strains along the length of the wall had a linear variation. The analytical curvatures were also determined this way in order to compare results. Since this wall is very slender, it is expected than the actual variation of the strain profile will be close to linear. The average vertical strains in the elements along the length of the wall obtained from program VecTor2 were studied to verify if this assumption is true.

First, we will consider the case where the wall is subjected to a constant compressive axial load only (before applying lateral displacements). It is expected that the strain profile in every cross-section will be constant. Figure 4.12 shows the strain profile at the construction joint:

Figure 4.12 Strain profile at construction joint for the case of no bending


Contrary to what it is expected, the strain profile is not constant. In order to investigate why, the strain profile at a certain distance from the construction joint is shown in Figure 4.13:

Figure 4.13 Strain profile at $\mathbf{3 1 9 m m}$ from construction joint for the case of no bending


As we get further from the construction joint, the strain profile becomes almost constant. Therefore, the results are distorted near the construction joint due to the boundary conditions of the model.

Figure 4.14 shows the strain profile for a wall displacement of 12.8 mm at the construction joint, before cracking occurs:

Figure 4.14 Strain profile at construction joint for a wall displacement of $\mathbf{1 2 . 8 m m}$


Prior to cracking, the strain profile is almost linear. There are some distortions near the base due to the boundary conditions, as seen previously. A least-squares fit was used to obtain a linear strain profile; the slope of this line is the curvature, in $\mathrm{rad} / \mathrm{m}$.

Figure 4.15 shows the strain profile for this wall displacement level at 319 mm from the construction joint:

Figure 4.15 Strain profile at 319 mm from the construction joint for a wall displacement of 12.8 mm


At this distance from the construction joint, the strain profile has practically a perfect linear variation. Therefore, it can be concluded that the strain profiles remain linear before cracking.

Figure 4.16 shows the strain profile for a wall displacement of 25.4 mm at the construction joint, after cracking and prior to yielding:

Figure 4.16 Strain profile at construction joint for a wall displacement of $\mathbf{2 5 . 4 \mathrm { mm }}$


After cracking, the strain profile does not have a linear variation. Note that using a least-squares fit to determine the curvature gives a different result that if we use only the strains at the faces of the wall.

Figure 4.17 shows the strain profile for this wall displacement level at 840 mm from the construction joint:

Figure 4.17 Strain profile at $\mathbf{8 4 0 \mathrm { mm }}$ from the construction joint for a wall displacement of $\mathbf{2 5 . 4 \mathrm { mm }}$


At this distance from the construction joint, the strain profile does not have a linear variation either. The distortions in this case are not due to the boundary conditions. Therefore, it can be concluded that after cracking, the strain profile does not remain linear, even though the wall is slender.

Figure 4.18 shows the strain profile for a wall displacement of 48 mm at 840 mm from the construction joint, after yielding:

Figure 4.18 Strain profile at 840 mm from the construction joint for a wall displacement of $\mathbf{4 8 m m}$


As we get further into the nonlinear range, the strain profile is less close to having a linear variation.

Because these observations contradict the results that were expected, a new analysis was carried to study why the strain profiles do not have a linear variation after cracking. As it was previously described, the wall had a low amount of reinforcement. In order to investigate how the amount of reinforcement influences the shape of the strain profiles, a second finite element analysis of the wall specimen, with an increased amount of reinforcement, was performed using program VecTor2. A new model was constructed for this purpose. This model had the same characteristics as the previous one, described in Section 4.2.5, except for the amount of horizontal and vertical reinforcement. The same loads were also applied. The average vertical strains in the elements along the length of the wall obtained from this analysis were examined.

Table 4.5 shows the material properties in the different regions of this new model of the wall:

Table 4.5 Material properties of UBC wall model with added reinforcement

| Concrete properties | Material 1 |  | Material 2 |  | Material 3 |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Color | 380 |  | 127 |  | 380 |  |
| Thickness (mm) | 49 |  | 49 | 49 |  |  |
| Cylinder compressive <br> strength (MPa) |  |  |  |  |  |  |
| Reinforcement <br> component properties | $x$-direction | $y$-direction | $x$-direction | $y$-direction | $x$-direction | $\boldsymbol{y}$-direction |
| Reinforcement ratio (\%) | 1.5 | 2.0 | 0.5 | 0.5 | 1.5 | 2.0 |
| Reinforcement diameter <br> (mm) | 10 | 10 | 10 | 10 | 10 | 10 |
| Yield strength (MPa) | 455 | 455 | 455 | 455 | 455 | 455 |
| Ultimate strength (MPa) | 650 | 650 | 650 | 650 | 650 | 650 |
| Elastic modulus (MPa) | 200000 | 20000 | 200000 | 200000 | 200000 | 200000 |
| Strain hardening <br> modulus (MPa) | 4875 | 4875 | 4875 | 4875 | 4875 | 4875 |
| Strain hardening strain <br> (mm/m) | 10 | 10 | 10 | 10 | 10 | 10 |

Figure 4.19 shows the strain profile for a wall displacement of 41.4 mm at the construction joint for this new analysis, after cracking and prior to yielding:

Figure 4.19 Strain profile at construction joint for a wall displacement of $\mathbf{4 1 . 4 m m}$


Comparing this figure with Figure 4.16, the variation of the strain profile is more close to linear.

Figure 4.20 shows the strain profile for this wall displacement level at 840 mm from the construction joint:

Figure 4.20 Strain profile at $\mathbf{8 4 0 \mathrm { mm }}$ from the construction joint for a wall displacement of $\mathbf{4 1 . 4 \mathrm { mm }}$


At this distance from the construction joint, the strain profile is very close to linear. The distortions in Figure 4.19 are due to the boundary conditions. Therefore, it can be concluded that the amount of reinforcement in the wall has a very significant impact in the shape of the strain profile after cracking.

With this amount of reinforcement, yielding occurs at a wall displacement of 70 mm . Figure 4.21 shows the strain profile for a wall displacement of 105 mm at 840 mm from the construction joint, after yielding:

Figure 4.21 Strain profile at $\mathbf{8 4 0 \mathrm { mm }}$ from the construction joint for a wall displacement of $\mathbf{1 0 5 m m}$


Comparing this figure with Figure 4.18, the variation of the strain profile is more close to linear after yielding at this distance from the construction joint.

Figure 4.22 shows the strain profile for a wall displacement of 281 mm at 840 mm from the construction joint:

Figure 4.22 Strain profile at 840 mm from the construction joint for a wall displacement of $\mathbf{2 8 1 \mathrm { mm }}$


As seen previously in Figure 4.18, as we get further into the nonlinear range, the strain profile is less close to having a linear variation. However, for this case, considering a linear variation is still a fair approximation.

The four main conclusions made from this study can be summarized as the following:

- The boundary conditions of the model distort the shape of the strain profile.
- Before cracking, the strain profile remains linear.
- After cracking, the shape of the strain profile depends on the amount of reinforcement. As the amount of reinforcement is increased, the strain profile is closer to having a linear variation.
- The linearity of the strain profile degrades as the wall goes further into the nonlinear range.

Therefore, considering that the strain profile is linear is not always true. It is important to take this into consideration when we design slender walls with low amounts of reinforcement. For the wall studied, there was no experimental information available regarding the axial strains along the crosssection, so a linear variation had to be assumed to determine the curvatures. Tests results showing the actual variation of axial strains along the cross-section of slender walls will be presented in the next section.

### 4.3 Rectangular shear wall tested at Clarkson University

A series of tests were conducted at Clarkson University in 1995 on slender reinforced concrete walls with rectangular-shaped and T -shaped cross-sections with moderate amounts of transverse reinforcement by Thomsen and Wallace. The purpose of these tests was to evaluate the simplified displacement-based design approach in ACI 3181999 (Thomsen and Wallace 2004: 618-630). A brief description of the tests and the results obtained has already been given in Section 2.24. These tests are of particular interest because extensive measurements of the concrete strains were made along the length of the wall at the base, so the actual variation of the axial strains was determined. Therefore, these test results can be compared with analytical predictions made with program VecTor2. The wall analyzed for these comparisons was the rectangular specimen RW2. The following sections will be focused on this specimen only; additional information needed for the analysis will be presented.

### 4.3.1 Description of the wall specimen

The specimen RW2 was a $1 / 4$ scaled model of a wall in an area of high seismicity. It had an aspect ratio of three and was considered as a four-storey wall. The wall was 102 mm thick, 1219 mm long and 3658 mm high (Thomsen and Wallace 2004: 618-620).

The reinforcement was arranged symmetrically. The boundary zones were 191 mm long each. The vertical reinforcement in the two zones consisted of eight deformed \#3 bars enclosed by $3 / 16^{\text {" }}$ diameter smooth wire hoop and cross-ties spaced at 51 mm . The vertical reinforcement in each zone was arranged in four layers (spaced at 51 mm ) of two bars each. The clear cover of the hoops was 9.5 mm . The web had two deformed \#2 bars in each layer spaced at 191mm, vertically and horizontally. A total of eight bars were used in the web.

The wall had different concrete strengths at the base and at each of the four storeys at the time of testing. At the first storey, the compressive cylinder strength was 43.7 MPa and the rupture strength was 5.63 MPa . All three types of reinforcing steel used had different properties. The yield strength was 434, 448 and 434 MPa for the \#3 bars, \#2 bars and 3/16" smooth wires, respectively; and the ultimate strength was 641,586 and 483 MPa , respectively. The modulus of elasticity was 200000 MPa for all three. The strain at the onset of strain hardening for the \#3 bars was $16 \mathrm{~mm} / \mathrm{m}$ (Thomsen and Wallace 1995: 148 150). The stress-strain response of three types of reinforcing steel was measured (Thomsen and Wallace 1995: 182).

### 4.3.2 Instrumentation

The wall was tested as a cantilever element. It was subjected to cyclic lateral displacements applied at the top by a hydraulic actuator. Additionally, a constant compressive axial load was applied at the top by hydraulic jacks (Thomsen and Wallace 2004: 622-623).

Four wire potentiometers were used to measure lateral displacements at 914 mm intervals along the height of the wall. Seven linear variable differential transducers were provided along the wall length to measure vertical displacements; these were placed vertically over a gage length of 229 mm (Thomsen and Wallace 2004: 624).

### 4.3.3 Test procedure

During the test, the constant axial load applied to the wall was on average $0.07 A_{8} f^{\prime}{ }^{\prime}$, equal to 378 kN , calculated using the concrete cylinder strength of the first storey. However, the axial load had to be constantly adjusted, resulting in a considerable variation of it. The maximum axial load applied was 436 kN (Thomsen and Wallace 1995: 205).

The lateral displacements applied to the wall consisted of 20 cycles. The drift for the 12 first cycles increased from 0.125 to $1.5 \%$, and for the last eight cycles it increased from 1 to $2.5 \%$. The drifts were increased every two cycles (Thomsen and Wallace 1995: 203).

### 4.3.4 Test results

The test results were used to determine strain profiles along the length of the wall for different drifts levels. The axial strains were determined from the displacement readings measured with the seven transducers provided along the wall. These readings were divided by the gage length to obtain axial strains. Some of the readings at the surface were not reliable because of concrete spalling (Thomsen and Wallace 2004: 627). The experimental strain profiles were determined for the positive and negative direction of loading (Thomsen and Wallace 1995: 223).

### 4.3.5 Analytical model of the wall specimen

The wall specimen was modeled and analyzed using program VecTor2. The purpose of this analysis is to predict the strain profile along the length of the wall and compare these predictions with the experimental results obtained by Thomsen and Wallace. Low-powered rectangular elements were used to
model the concrete, with smeared steel to account for the presence of reinforcement. The constitutive models for concrete and steel described in Chapter 3 were used in the analysis.

Two different material types were used to represent various regions of the wall in the finite element model:

- The first material type was used to represent the confined boundaries of the wall.
- The second material type was used to represent the web of the wall.

The analytical model was fixed at the bottom; both the horizontal and vertical displacements were restrained.

The finite element mesh was refined in such a way that the position of certain nodes coincides with the position of the transducers. Meshes of $89 \times 114,102 \times 114$ and $114 \times 114$ rectangular elements were used; 12 elements ( 13 nodes) were used in the transverse direction of the wall. This level of refinement was used to model the whole wall. All nodes and elements were numbered in the horizontal (short) direction. The complete mesh consisted of 429 nodes and 384 rectangular elements.

The material properties used in the analysis were those reported in the description of the wall specimen, presented in Section 4.3.1. The concrete properties from the first storey were considered for the whole wall. The concrete tensile strength was taken as half of the rupture strength, that is, 2.8 MPa . For the material properties that were not measured during the test, the values given by default in program VecTor2 were used. The concrete initial tangent elastic modulus and the cylinder strain at $f_{c}{ }^{\prime}$ where determined with Equation 4.2 and 4.3, respectively. The maximum aggregate size was 10 mm , given by default. For the reinforcement properties, the strain hardening modulus and the strain at the onset of strain hardening were determined from the measured stress-strain curves; except for the strain at the onset of strain hardening for the deformed \#3 bar, which is given in Section 4.3.1.

A monotonic lateral load was applied at the top of the wall. This load was applied in a displacement-control mode, in increments of 0.2 mm . Additionally, a constant axial load of 436 kN was applied; this was the maximum axial load applied during the test. This load was equally distributed among all the 13 nodes at the top. The self-weight of the wall was not considered.

Figure 4.23 shows the finite element model of the wall specimen, created in the pre-processor FormWorks:

Figure 4.23 Finite element model of specimen RW2 in FormWorks


Table 4.6 shows the material properties in the different regions of the wall:

Table 4.6 Material properties of specimen RW2 model

| Concrete properties | Material 1 |  | Material 2 |  |
| :---: | :---: | :---: | :---: | :---: |
| Color | 101.6 |  | 101.6 |  |
| Thickness (mm) |  |  | 43.7 |  |
| Cylinder compressive <br> strength (MPa) | 43.7 |  | 2.8 |  |
| Tensile strength (MPa) | 2.8 |  |  |  |
| Reinforcement <br> component properties | $x$-direction | $y$-direction | $x$-direction | $y$-direction |
| Reinforcement ratio (\%) | 0.69 | 2.95 | 0.33 | 0.33 |
| Reinforcement diameter <br> $(\mathrm{mm})$ | 4.75 | 9.5 | 6.4 | 6.4 |
| Yield strength (MPa) | 434 | 434 | 448 | 448 |
| Ultimate strength (MPa) | 483 | 641 | 586 | 586 |
| Elastic modulus (MPa) | 200000 | 200000 | 200000 | 200000 |
| Strain hardening <br> modulus (MPa) | 1119 | 5750 | 3391 | 3391 |
| Strain hardening strain <br> $(\mathrm{mm} / \mathrm{m})$ | 2.2 | 16 | 2.3 | 2.3 |

### 4.3.6 Analytical results

The analytical predictions obtained from program VecTor2 were used to determine strain profiles along the length of the wall for different drift levels. The axial strains were determined the same way as it was done for the experimental results in order to make comparisons, using the vertical nodal displacements along the wall length obtained from the analysis.

As previously described, the position of certain nodes along the length of the wall coincides with the position of the transducers. The vertical nodal displacements at these particular nodes were used to calculate the axial strains. The difference in vertical displacements between two nodes in the same position along the wall length, divided by the vertical distance between these nodes, gives the average axial strain over that length at that position. This way, the experimental and analytical axial strains were calculated over the same length. The strain profile was determined for drift levels of 1.5 and $2 \%$.

### 4.3.7 Comparison of experimental and analytical results

Figure 4.24 and 4.25 show comparisons between the experimental strain profile, in both directions of loading, and the analytical strain profile; for the two drift levels mentioned:

Figure 4.24 Strain profile at base for a drift of $1.5 \%$


Figure 4.25 Strain profile at base for a drift of $\mathbf{2 \%}$


Both the experimental and the analytical results show that the strain profiles are not linear, although the walls are slender; which confirms the results obtained for the UBC wall test. As it was
discussed previously, the nonlinearity of the strain profile is because this wall had a moderate amount of reinforcement.

Although program VecTor2 predicts reasonably well the shape of the strain profiles along the length of the wall, there are differences between the experimental and analytical results, specially in the tension zone. There is a better agreement between the analytical and the experimental strain profiles in the positive direction of loading. The greater differences in the tension zone, as the authors indicated, are due to the influence of concrete cracking and slippage of the reinforcement (Thomsen and Wallace 2004: 628). The differences in the compression zone are due to the very significant damage and spalling of the concrete cover at the wall boundaries during the test (Thomsen and Wallace 2004: 626).

The analytical predictions in the compression zone can be improved if the cover spalling at the wall boundary is considered in the finite element model. For this, only the concrete inside the core was modeled. The area of concrete in the wall boundary considered in the analysis was delimited by the longitudinal axis of the hoops. This is shown in Figure 4.26:

Figure 4.26 Area of concrete considered in the analysis to account for cover spalling


A second finite element analysis of the wall, considering cover spalling, was performed using program VecTor2. A new model was constructed for this purpose. This model had the same characteristics as the previous one, described in Section 4.3.5, except that a new material type was created for the confined boundary of the wall in the compression zone. This material type had a reduced thickness of 78 mm to account for cover spalling at the sides, as shown in Figure 4.26. The nodes at the far right in the model were moved 12 mm to the left to account for cover spalling at the face of the wall. The same loads were applied for this analysis. The vertical nodal displacements obtained were used to calculate the
axial strains along the length of the wall using the same procedure described in Section 4.3.6, for drift levels of 1.5 and $2 \%$.

Figure 4.27 shows the new finite element model of the wall specimen, created in the preprocessor FormWorks:

Figure 4.27 Finite element model accounting for cover spalling of specimen RW2 in FormWorks


Table 4.7 shows the material properties in the different regions of this new model of the wall:

Table 4.7 Material properties of specimen RW2 model accounting for cover spalling

| Concrete properties | Material 1 |  | Material 2 |  | Material 3 |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  |  |  |
| Thickness (mm) | 101.6 |  | 101.6 |  | 77.75 |  |
| Cylinder compressive strength (MPa) | 43.7 |  | 43.7 |  | 43.7 |  |
| Tensile strength (MPa) | 2.8 |  | 2.8 |  | 2.8 |  |
| Reinforcement component properties | $\boldsymbol{x}$-direction | $\boldsymbol{y}$-direction | $\boldsymbol{x}$-direction | $y$-direction | $\boldsymbol{x}$-direction | $\boldsymbol{y}$-direction |
| Reinforcement ratio (\%) | 0.69 | 2.95 | 0.33 | 0.33 | 0.69 | 2.95 |
| Reinforcement diameter ( mm ) | 4.75 | 9.5 | 6.4 | 6.4 | 4.75 | 9.5 |
| Yield strength (MPa) | 434 | 434 | 448 | 448 | 434 | 434 |
| Ultimate strength (MPa) | 483 | 641 | 586 | 586 | 483 | 641 |
| Elastic modulus (MPa) | 200000 | 200000 | 200000 | 200000 | 200000 | 200000 |
| Strain hardening modulus (MPa) | 1119 | 5750 | 3391 | 3391 | 1119 | 5750 |
| Strain hardening strain ( $\mathrm{mm} / \mathrm{m}$ ) | 2.2 | 16 | 2.3 | 2.3 | 2.2 | 16 |

Figure 4.28 and 4.29 show comparisons between the experimental strain profile, in both directions of loading, and the analytical strain profile; for the two drift levels mentioned:

Figure 4.28 Strain profile accounting for cover spalling at base for a drift of $\mathbf{1 . 5 \%}$


Figure 4.29 Strain profile accounting for cover spalling at base for a drift of $\mathbf{2 \%}$


Comparing these figures with Figure 4.24 and 4.25 , there is an excellent agreement between the experimental and analytical strain profiles in the compression zone. Despite the discrepancies in the tension zone, the analytical model seems to predict reasonably well the strain profile of the wall.

## CHAPTER 5: PARAMETRIC STUDY OF CONCRETE WALLS

### 5.1 Scope of analysis

Program VecTor2 was used to perform a parametric study of concrete walls. The factors that affect the plastic hinge length were investigated. The parameters considered in this study were the wall length, the distance from the section of maximum moment to the section of zero moment (shear span), the diagonal cracking and the axial load.

### 5.2 Description of wall models

Two cantilever wall models were considered for this parametric study, which will be referred to as Wall 1 and Wall 2 throughout this chapter. Wall 1 was twice as long as Wall 2. The dimensions and material properties of these walls were typical of high-rise buildings. The two walls had a rectangular cross-section, and they were 54860 mm high and 508 mm thick.

Wall 1 was 7620 mm long. The boundary zones were 1219 mm long, the vertical reinforcement in these consisted of $24-25 \mathrm{M}$ reinforcing bars enclosed by 15 M ties spaced at 100 mm . The clear cover of the ties was 40 mm . The web had 15 M reinforcing bars spaced at 150 mm vertically and horizontally. The cross-section of Wall 1 is shown in Figure 5.1:

Figure 5.1 Cross-section details of Wall 1


Wall 2 was 3810 mm long. The boundary zones were 610 mm long, the vertical reinforcement in these consisted of $12-25 \mathrm{M}$ reinforcing bars enclosed by 15 M ties spaced at 100 mm . The clear cover of the ties was 40 mm . The web had 15 M reinforcing bars spaced at 150 mm vertically and horizontally. The cross-section of Wall 2 is shown in Figure 5.2:

Figure 5.2 Cross-section details of Wall 2


The concrete cylinder compressive strength of both walls was 40 MPa . The stress-strain curve of all the reinforcing bars was assumed to be the same as the one used for the UBC wall test, whose properties are shown in the bottom five rows of Table 4.1, except that the yield strength was taken as 400 MPa.

Both walls were fixed at the base. They were subjected to a monotonically increasing lateral load at the top; as well as a constant compressive axial load of $0.1 A_{g} f^{\prime}$, equal to 15484 kN for Wall I and 7742 kN for Wall 2.

Using these two wall models as the base cases, the parametric study was performed by changing the wall parameters and observing how this affects the length of the plastic hinge.

### 5.3 Analytical model of walls

Wall 1 and 2 were modeled and analyzed using program VecTor2. Each wall was analyzed separately. Low-powered rectangular and triangular elements were used to model the concrete, with smeared steel to account for the presence of reinforcement. The constitutive models for concrete and steel described in Chapter 3 were used in the analysis.

Two different material types were used to represent various regions of the two walls in the finite element model:

- The first material type was used to represent the confined boundaries of the wall.
- The second material type was used to represent the web of the wall.

The analytical model was fixed at the bottom; both the horizontal and vertical displacements were restrained. The finite element mesh was more refined near the base (critical section) in both walls. Also,
the two walls had the same refinement in the vertical direction up to a half of the height, so that the curvatures could be calculated over the same average length.

For Wall 1, meshes of $305 \times 423$ and $305 \times 203$ rectangular elements were used; 25 elements ( 26 nodes) were used in the transverse direction of the wall. This level of refinement was maintained up to half the height of the wall. Up from this point, 18 elements ( 19 nodes) were used in the transverse direction, then it was further reduced to 10 elements ( 11 nodes), and then to five elements (six nodes) up to the top of the wall. The transitions were made using triangular elements. All nodes and elements were numbered in the horizontal (short) direction. The complete mesh consisted of 2081 nodes, 1927 rectangular elements and 86 triangular elements.

For Wall 2, meshes of $305 \times 423,305 \times 203,288 \times 423$ and $288 \times 203$ rectangular elements were used; 13 elements ( 14 nodes) were used in the transverse direction of the wall. This level of refinement was maintained up to half the height of the wall. Up from this point, eight elements (nine nodes) were used in the transverse direction, and then it was further reduced to five elements (six nodes) up to the top of the wall. The transitions were made using triangular elements. All nodes and elements were numbered in the horizontal (short) direction. The complete mesh consisted of 1276 nodes, 1139 rectangular elements and 30 triangular elements.

The material properties used in the analysis were those reported in the description of the wall models, presented in Section 5.2. For the material properties not mentioned in that section, the values given by default in program VecTor2 were used. The concrete tensile strength, the initial tangent elastic modulus and the cylinder strain at $f_{c}^{\prime}$ where determined with Equation 4.1, 4.2 and 4.3, respectively. The maximum aggregate size was 10 mm , given by default. For the reinforcement properties, these were the same as those for the UBC wall test, except for the yield strength, as it was previously mentioned.

A monotonic lateral load was applied at the top of the walls. This load was applied in a displacement-control mode, in increments of 1 mm . Additionally, a constant axial load of 15484 kN in Wall 1 and 7742 kN in Wall 2 was applied; these loads were equally distributed among the six nodes at the top of each wall. The self-weight of the walls was not considered.

Figure 5.3 shows the finite element model of Wall 1, created in the pre-processor FormWorks:

Figure 5.3 Finite element model of Wall 1 in FormWorks


Figure 5.4 shows the finite element model of Wall 2, created in the pre-processor FormWorks:

Figure 5.4 Finite element model of Wall 2 in FormWorks


Table 5.1 shows the material properties in the different regions of the two walls:

Table 5.1 Material properties of Wall 1 and 2

| Concrete properties | Material 1 |  | Material 2 |  |
| :---: | :---: | :---: | :---: | :---: |
| Color | 508 |  | 508 |  |
| Thickness (mm) | 40 |  | 40 |  |
| Cylinder compressive <br> strength (MPa) |  |  | Reinforcement <br> component properties |  |
| $x$-direction | $y$-direction | $x$-direction | $y$-direction |  |
| Reinforcement ratio (\%) | 0.8 | 2.0 | 0.5 | 0.5 |
| Reinforcement diameter <br> (mm) | 15 | 25 | 15 | 15 |
| Yield strength (MPa) | 400 | 400 | 400 | 400 |
| Ultimate strength (MPa) | 650 | 650 | 650 | 650 |
| Elastic modulus (MPa) | 200000 | 200000 | 200000 | 200000 |
| Strain hardening <br> modulus (MPa) | 4875 | 4875 | 4875 | 4875 |
| Strain hardening strain <br> (mm/m) | 10 | 10 | 10 | 10 |

### 5.4 Analytical results

The analytical predictions obtained from program VecTor2 were used to determine curvature distributions along the height of the walls for different lateral displacement levels. The vertical nodal displacements at the faces of the walls were used to calculate the curvatures, as it was done for the UBC wall test, which is described in Section 4.2.6. The difference in vertical displacements between two nodes at the face, divided by the vertical distance between these nodes, gives the average axial strain over that length. Then, the difference between the compression and tension strain, divided by the wall length between the two faces, gives the average curvature over the length between nodes. The curvatures were calculated over average lengths of 847 or 1050 mm ; that is, between every two or three elements.

The strain profile along the length of the two walls was investigated. For slender walls like the ones being studied, with the amount of reinforcement that they have, we expect the strain profile along the wall length to be close to linear, as seen in Section 4.2.10. The average vertical strains in the elements along the length of the walls obtained from program VecTor2 were studied to verify if this assumption is true. The ultimate drift for Wall 1 was $2 \%$. Figure 5.5 shows the strain profile of Wall 1 for this drift at the base:

Figure 5.5 Strain profile of Wall 1 at base for a drift of 2\%


The strain profile is not linear due to the boundary conditions, as seen in Section 4.2.10. Figure 5.6 shows the strain profile of Wall 1 for a $2 \%$ drift at 1270 mm from the base:

Figure 5.6 Strain profile of Wall $\mathbf{1}$ at $\mathbf{1 2 7 0 m m}$ from the base for a drift of $\mathbf{2 \%}$


At this distance from the base, the strain profile very close to having a linear variation. Figure 5.7 shows the strain profile of Wall 2 for a $2 \%$ drift at the base:

Figure 5.7 Strain profile of Wall 2 at the base for a drift of $\mathbf{2 \%}$


Because Wall 2 is more slender, the strain profile is linear even at the base. From these figures, it can be seen that considering that plane sections remain plane after bending is a reasonable assumption for these two walls. The curvatures were calculated from the strains at the wall faces considering a linear variation. Figure 5.8 shows the curvature distribution up to the mid-height of both walls for a drift of $2 \%$ :

Figure 5.8 Curvatures up to the mid-height of Wall 1 and 2 for a drift of $\mathbf{2 \%}$


As seen in Section 4.2.9, the inelastic curvatures can be well approximated as linearly varying. Wall 2 has larger elastic curvatures because it is more flexible. In order to determine the plastic hinge length from the curvature distribution, the yield curvature of the walls has to be estimated. Once the yield curvature is determined, the inelastic curvatures can be visualized by shifting the vertical axis by this amount, which defines the length of the plastic hinge for a linearly varying inelastic curvature.

There are two ways to determine the yield curvature. One way is to calculate the curvature just above the base, using the procedure described in this section, for the yield displacement or drift (first yield of the longitudinal reinforcement). The yield drifts were determined by looking at the average vertical steel strains in the elements obtained from program VecTor2. For Wall 1, the steel started to yield at a drift of $0.61 \%$, for which the curvature just above the base was $0.00045 \mathrm{rad} / \mathrm{m}$. For Wall 2 , the steel started to yield at a drift of $1.18 \%$, for which the curvature just above the base was $0.0009 \mathrm{rad} / \mathrm{m}$.

A second way to estimate the yield curvature is from the moment-curvature relationships of the walls. To develop this relationship, the bending moment diagram along the height of the wall must be calculated at the same locations where the curvatures in Figure 5.8 were determined. For Wall 1 and 2, since they were subjected to a lateral load at the top, the bending moment diagram was linearly varying and could be easily determined. Figure 5.9 shows the bending moments along the height of both walls for a drift of $2 \%$ :

Figure 5.9 Bending moments along the height of Wall 1 and 2 for a drift of $\mathbf{2 \%}$


Figure 5.10 shows the moment-curvature relationship of Wall 1:

Figure 5.10 Moment-curvature relationship of Wall 1


Figure 5.11 shows the moment-curvature relationship of Wall 2:

Figure 5.11 Moment-curvature relationship of Wall 2


From Figure 5.10 and 5.11 , it can be seen that the yield curvature of $0.00045 \mathrm{rad} / \mathrm{m}$ for Wall 1 and $0.0009 \mathrm{rad} / \mathrm{m}$ for Wall 2 determined previously are approximately the same values estimated from the moment-curvature relationships. Both procedures are giving the same result, so these values were used to estimate the plastic hinge length.

Similarly, the yield moment can be determined using these two procedures. The bending moment just above the base was calculated using the procedure described in this section for the yield drift. For Wall 1, this moment was $90680 \mathrm{kN} . \mathrm{m}$; and for Wall 2, this moment was $22740 \mathrm{kN} . \mathrm{m}$. These were approximately the same values estimated from the moment-curvature relationships shown in Figure 5.10 and 5.11. Therefore, these values were selected as the yield moments.

### 5.5 Revision of analytical results

The moment-curvature relationships of Wall 1 and 2 were used to check the results obtained from program VecTor2, by comparing these with the moment-curvature relationships obtained from a sectional analysis. Program Response-2000 was used for this purpose. Response-2000 is a computer program developed to perform two-dimensional sectional analysis of concrete members. This program was also
developed at the University of Toronto, by Evan Bentz. More information about the program Response2000 can be found in the Response-2000, Shell-2000, Triax-2000, Membrane-2000 User Manual (Bentz 2001). Response-2000 was used to perform a sectional analysis of Wall 1 and 2 and develop their moment-curvature relationships for pure bending.

To compare the results obtained from VecTor2 and Response-2000, the material models used in both programs have to be the same. Response-2000 uses the Modified Compression Field Theory, and has specific constitutive models incorporated in it to account for effects like compression softening and tension stiffening. These models are different from the ones used in the VecTor2 analysis. Using different constitutive models can significantly change the response. Therefore, a new analysis for both walls was performed in VecTor2, using the same constitutive models as in Response-2000, in order to compare results. The new models used in this analysis are the following, the details of these models can be found in the VecTor2 and FormWorks User's Manual (Wong and Vecchio 2002: 45-88):

- Compression pre-peak response: Popovics for high-strength concrete.
- Compression post-peak response: Popovics for high-strength concrete.
- Compression softening: Vecchio-Collins 1986.
- Tension stiffening: Bentz 1999.
- Cracking criterion: Unixial cracking stress.
- Crack slip check: Vecchio-Collins 1986 (this check was not done before because it is not required when using the Disturbed Stress Field Model).
- Slip distortions: Not considered (since the Modified Compression Field Theory is being used).

The moment-curvature relationship from the results of VecTor2 was determined using the same procedure described in Section 5.4. Figure 5.12 shows the moment-curvature relationship of Wall 1 :

Figure 5.12 Moment-curvature relationship of Wall 1 using Response-2000 constitutive models


Figure 5.13 shows the moment-curvature relationship of Wall 2:

Figure 5.13 Moment-curvature relationship of Wall 2 using Response-2000 constitutive models


There is a good agreement between the results provided by VecTor2 and Response-2000. This analysis was only performed to check the VecTor2 predictions. The parametric study in this chapter was done using the constitutive models described in Chapter 3, since these models provided the best agreement with the experimental results from the tests described in Chapter 4.

### 5.6 Influence of wall length

The first parameter studied was the wall length. As it was discussed in Chapter 2, many models developed for plastic hinge length consider that it is proportional to the wall length only, so it is an important parameter to be considered. The results obtained for Wall 1 and 2 will be used to analyze how the plastic hinge length is influenced by the wall length.

As seen in Section 5.4, the yield curvature and yield moment for Wall 1 were $0.00045 \mathrm{rad} / \mathrm{m}$ and $90680 \mathrm{kN} . \mathrm{m}$, respectively; and for Wall 2 they were $0.0009 \mathrm{rad} / \mathrm{m}$ and $22740 \mathrm{kN} . \mathrm{m}$. The yield curvature was used to measure the plastic hinge length for a drift of $2 \%$ from the curvature distribution in Figure 5.8. For Wall 1 , the plastic hinge length was 8.39 m or $1.10 l_{w}$; and for Wall 2 , it was 6.21 m or $1.63 l_{w}$. Therefore, predicting the plastic hinge length as a function of the wall length only does not work for these walls.

A different approach to predict the plastic hinge length is by using moments; that is, the yield moment determined from the moment-curvature relationship defines the length of the plastic hinge. For a linear bending moment diagram like the one shown in Figure 2.6, the length where the yield moment is exceeded can be determined from Equation 2.11, repeated here for convenience:

$$
\begin{equation*}
\frac{L_{p}}{L_{s}}=1-\frac{M_{y}}{M_{\max }} \tag{5.1}
\end{equation*}
$$

From Figure 5.9, the maximum moment in Wall 1 was $105633 \mathrm{kN} . \mathrm{m}$, and for Wall 2 it was $25406 \mathrm{kN} . \mathrm{m}$. Using Equation 5.1, the plastic hinge length for Wall 1 was 7.77 m or $1.02 l_{w}$; and for Wall 2, it was 5.76 m or $1.51 l_{w}$. This flexural prediction matches well with the measured plastic hinge lengths.

The spread of yielding was larger in Wall 1. To explain why a longer wall has a larger plastic hinge length, the tensile steel strains in the wall were observed. The average vertical steel strains in the elements along the height of the walls obtained from program VecTor2 were studied for this. The steel strains were observed in the exterior and interior steel layer in the boundary zones, shown in Figure 5.14:

Figure 5.14 Exterior and interior steel layers in zones in Wall 1 and 2



Wall 2

Figure 5.15 shows the steel strains up to the mid-height of both walls for a drift of $2 \%$ :

Figure 5.15 Steel strains up to the mid-height of Wall 1 and 2 for a drift of $\mathbf{2 \%}$


The steel strains exceed the yield strain ( 0.002 for these walls) at a larger distance from the base in Wall 1 , so the spread of yielding was larger in this wall. The reason why the steel strains are larger in Wall 1 can be explained from the curvature distribution in Figure 5.8. At this drift, the curvatures at the base are approximately the same. For the same curvature, the steel strains will be larger in the longer wall; and therefore, it will have a larger spread of yielding.

The reason why a longer wall has a larger plastic hinge length can also be explained by comparing the moment-curvature relationships of both walls. The plastic hinge length increases as the
difference between the yield and ultimate moment increases, which is also reflected in Equation 5.1. By plotting the moment-curvature relationships together, it can be observed that the slope of the postyielding phase is bigger in Wall 1; therefore, it has a larger plastic hinge length. This is shown in Figure 5.16:

Figure 5.16 Moment-curvature relationships of Wall 1 and 2


It can be concluded that longer walls have larger plastic hinge lengths because there is a bigger difference between the yield and ultimate moment for these. However, predicting the plastic hinge length as a function of the wall length only does not provide good results. Better predictions can be made by using a flexural prediction, like the one shown in Equation 5.1.

The measured (from curvature distribution) and predicted (using Equation 5.1) plastic hinge lengths were determined for different drift levels, using the same procedure described in Section 5.4 and this section. The results are shown in Table 5.2 and 5.3:

Table 5.2 Predicted and measured plastic hinge lengths for Wall 1

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W1 Predicted | W1 Measured |
| 2.0 | 105632.9 | 90680.0 | 54.860 | 7.77 | 8.39 |
| 1.8 | 104908.8 | 90680.0 | 54.860 | 7.44 | 8.08 |
| 1.6 | 104316.3 | 90680.0 | 54.860 | 7.17 | 7.40 |
| 1.4 | 103345.3 | 90680.0 | 54.860 | 6.72 | 7.18 |
| 1.2 | 102281.0 | 90680.0 | 54.860 | 6.22 | 6.53 |

Table 5.3 Predicted and measured plastic hinge lengths for Wall 2

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W2 Predicted | W2 Measured |
| 2.0 | 25405.7 | 22740.0 | 54.860 | 5.76 | 6.21 |
| 1.8 | 25136.9 | 22740.0 | 54.860 | 5.23 | 5.54 |
| 1.6 | 24769.3 | 22740.0 | 54.860 | 4.49 | 4.56 |
| 1.4 | 24050.6 | 22740.0 | 54.860 | 2.99 | 3.12 |
| 1.2 | 23079.6 | 22740.0 | 54.860 | 0.81 | 1.02 |

The flexural prediction provides good results for all drift levels.

### 5.7 Influence of shear span

The second parameter studied was the shear span. This parameter has also been included in many models for plastic hinge length presented in Chapter 2, so it has a significant importance.

So far, the analysis for Wall 1 and 2 had been performed for a shear span equal to the wall height. To study the influence of the shear span, three new analyses for each wall were performed in VecTor2 for the following shear spans:

- $35659 \mathrm{~mm}(2 / 3$ of wall height).
- 27430 mm ( $1 / 2$ of wall height).
- 19201 mm ( $1 / 3$ of wall height).

The models used for these analyses had the same characteristics as the previous ones, described in Section 5.3, except that the lateral load was now applied at these heights measured from the base. The results from the analysis were used to determine the curvatures and bending moments for different drift levels, using the same procedure described in Section 5.4 and 5.6.

Table 5.4 to 5.9 show the measured (from curvature distribution) and predicted (using Equation 5.1) plastic hinge lengths for different drift levels. These drifts were measured at the location of the lateral load ( $\Delta / L_{s}$ ):

Table 5.4 Predicted and measured plastic hinge lengths for Wall 1 for a shear span of $\mathbf{3 5 6 5 9 m m}$

| Drift (\%) | $\mathcal{M}_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W1 Measured |  |
| 1.6 | 106852.2 | 90680.0 | 35.659 | 5.40 | 6.06 |
| 1.4 | 106338.7 | 90680.0 | 35.659 | 5.25 | 5.72 |
| 1.2 | 105650.5 | 90680.0 | 35.659 | 5.05 | 5.74 |
| 1.0 | 103803.3 | 90680.0 | 35.659 | 4.51 | 5.01 |

Table 5.5 Predicted and measured plastic hinge lengths for Wall 2 for a shear span of $\mathbf{3 5 6 5 9} \mathbf{m m}$

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W2 Measured |  |
| 1.6 | 25941.9 | 22740.0 | 35.659 | 4.40 | 4.74 |
| 1.4 | 25774.3 | 22740.0 | 35.659 | 4.20 | 4.29 |
| 1.2 | 25275.1 | 22740.0 | 35.659 | 3.58 | 3.61 |
| 1.0 | 24494.2 | 22740.0 | 35.659 | 2.55 | 2.61 |

Table 5.6 Predicted and measured plastic hinge lengths for Wall 1 for a shear span of $\mathbf{2 7 4 3 0} \mathbf{m m}$

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W1 Measured |  |
| 1.6 | 108123.6 | 90680.0 | 27.430 | 4.43 | 5.55 |
| 1.4 | 107983.7 | 90680.0 | 27.430 | 4.40 | 5.37 |
| 1.2 | 106905.7 | 90680.0 | 27.430 | 4.16 | 5.31 |
| 1.0 | 105764.6 | 90680.0 | 27.430 | 3.91 | 4.61 |
| 0.8 | 104069.4 | 90680.0 | 27.430 | 3.53 | 4.00 |

Table 5.7 Predicted and measured plastic hinge lengths for Wall 2 for a shear span of $\mathbf{2 7 4 3 0} \mathbf{m m}$

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W2 Predicted | W2 Measured |
| 1.6 | 26585.2 | 22740.0 | 27.430 | 3.97 | 4.28 |
| 1.4 | 26288.9 | 22740.0 | 27.430 | 3.70 | 3.98 |
| 1.2 | 26025.6 | 22740.0 | 27.430 | 3.46 | 3.50 |
| 1.0 | 25603.2 | 22740.0 | 27.430 | 3.07 | 3.06 |
| 0.8 | 24610.2 | 22740.0 | 27.430 | 2.08 | 2.16 |

Table 5.8 Predicted and measured plastic hinge lengths for Wall 1 for a shear span of 19201 mm

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W1 Measured |  |
| 1.2 | 108372.4 | 90680.0 | 19.201 | 3.13 | 4.45 |
| 1.0 | 107362.4 | 90680.0 | 19.201 | 2.98 | 4.31 |
| 0.8 | 106567.5 | 90680.0 | 19.201 | 2.86 | 3.40 |
| 0.6 | 104616.6 | 90680.0 | 19.201 | 2.56 | 3.01 |

Table 5.9 Predicted and measured plastic hinge lengths for Wall 2 for a shear span of 19201 mm

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W2 Predicted | W2 Measured |
| 1.2 | 26766.2 | 22740.0 | 19.201 | 2.89 | 3.16 |
| 1.0 | 26282.3 | 22740.0 | 19.201 | 2.59 | 2.80 |
| 0.8 | 25911.7 | 22740.0 | 19.201 | 2.35 | 2.47 |
| 0.6 | 24840.3 | 22740.0 | 19.201 | 1.62 | 1.63 |

It can be observed that as the shear span gets smaller, the measured plastic hinge length becomes larger than the predicted one; specially for Wall 1 at high drifts. The reason for this is that the shear stress becomes larger as the shear span decreases, because the lateral force required to produce the same maximum moment at the base becomes larger. This produces more diagonal cracking, which increases the tensile forces in the longitudinal reinforcement, and so, the length of the plastic hinge. Therefore, a pure flexural prediction does not work when shear stresses are high. However, for low shear stresses, the prediction is good, independent of the shear span. The influence of the shear span is strongly related to the influence of diagonal cracking, which will be studied in the next section.

### 5.8 Influence of diagonal cracking

The third parameter studied was the diagonal cracking. As it was discussed in Chapter 2, the reason why many of the models developed for plastic hinge length consider that it is proportional to the wall length is because the longer it is, more significant is the influence of diagonal cracking. Therefore, diagonal cracking was related to the wall length. In this study, diagonal cracking was related to the magnitude of the shear stress.

The results of all the previous analyses made for Wall 1 and 2 were used to study the influence of diagonal cracking. Additionally, two new analyses for each wall were performed in VecTor2, this time reducing the thickness of the web, so that the shear stresses become higher (that is, they were now flanged walls). These new analyses were performed for the following shear spans:

- $\quad 27430 \mathrm{~mm}$ ( $1 / 2$ of wall height).
- $19201 \mathrm{~mm}(1 / 3$ of wall height).

The models used for these analyses had the same characteristics as the previous ones, described in Section 5.3, with the lateral load being applied at these heights measured from the base. The only difference was that the web thickness was reduced from 508 mm to 254 mm . The results from the analysis were used to determine the curvatures and bending moments for different drift levels, using the same procedure described in Section 5.4 and 5.6.

Table 5.10 shows the material properties in the different regions of these new models of the two walls:

Table 5.10 Material properties of Wall 1 and 2 with a web thickness of $\mathbf{2 5 4 m m}$

| Concrete properties | Material 1 |  | Material 2 |  |
| :---: | :---: | :---: | :---: | :---: |
| Color | 508 |  | 254 |  |
| Thickness (mm) | 50 |  |  |  |
| Cylinder compressive <br> strength (MPa) | 40 |  | 40 |  |
| Reinforcement <br> component properties | $x$-direction | $\boldsymbol{y}$-direction | $x$-direction | $\boldsymbol{y}$-direction |
| Reinforcement ratio (\%) | 0.8 | 2.0 | 0.5 | 0.5 |
| Reinforcement diameter <br> $(\mathrm{mm})$ | 15 | 25 | 15 | 15 |
| Yield strength (MPa) | 400 | 400 | 400 | 400 |
| Ultimate strength (MPa) | 650 | 650 | 650 | 650 |
| Elastic modulus (MPa) | 200000 | 200000 | 200000 | 200000 |
| Strain hardening <br> modulus (MPa) | 4875 | 4875 | 4875 | 4875 |
| Strain hardening strain <br> $(\mathrm{mm} / \mathrm{m})$ | 10 | 10 | 10 | 10 |

Since the walls in these new analyses had a different web thickness, the moment-curvature relationship was also different. The yield curvature and yield moment were determined using the same procedures described in Section 5.4. Figure 5.17 shows the moment-curvature relationship for Wall 1 with a web thickness of 254 mm :

Figure 5.17 Moment-curvature relationship of Wall 1 with a web thickness of $\mathbf{2 5 4 m m}$


Figure 5.18 shows the moment-curvature relationship for Wall 2 with a web thickness of 254 mm :

Figure 5.18 Moment-curvature relationship of Wall 2 with a web thickness of 254 mm


The yield curvature of both walls is approximately the same, and yield moment has decreased. The yield curvature and yield moment for Wall 1 were now $0.00045 \mathrm{rad} / \mathrm{m}$ and $84700 \mathrm{kN} . \mathrm{m}$, respectively; and for Wall 2 they were now $0.0009 \mathrm{rad} / \mathrm{m}$ and $21420 \mathrm{kN} . \mathrm{m}$. Table 5.11 to 5.14 show the measured (from curvature distribution) and predicted (using Equation 5.1) plastic hinge lengths for different drift levels. Again, these drifts were measured at the location of the lateral load $\left(\Delta / L_{s}\right)$ :

Table 5.11 Predicted and measured plastic hinge lengths for Wall 1 for a web thickness of $\mathbf{2 5 4} \mathbf{m m}$ and a shear span of 27430 mm

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | W1 Predicted | W1 Measured |  |
| 1.8 | 100292.3 | 84700.0 | 27.430 | 4.26 | 6.26 |
| 1.6 | 99837.0 | 84700.0 | 27.430 | 4.16 | 5.83 |
| 1.4 | 97971.7 | 84700.0 | 27.430 | 3.72 | 5.06 |
| 1.2 | 98202.1 | 84700.0 | 27.430 | 3.77 | 4.72 |
| 1.0 | 97801.7 | 84700.0 | 27.430 | 3.67 | 4.35 |
| 0.8 | 96663.3 | 84700.0 | 27.430 | 3.39 | 3.89 |

Table 5.12 Predicted and measured plastic hinge lengths for Wall 2 for a web thickness of 254 mm and a shear span of $\mathbf{2 7 4 3 0 \mathrm { mm }}$

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W2 Predicted | W2 Measured |
| 1.8 | 25112.2 | 21420.0 | 27.430 | 4.03 | 4.42 |
| 1.6 | 24815.9 | 21420.0 | 27.430 | 3.75 | 4.13 |
| 1.4 | 24431.9 | 21420.0 | 27.430 | 3.38 | 3.72 |
| 1.2 | 24149.4 | 21420.0 | 27.430 | 3.10 | 3.30 |
| 1.0 | 23716.0 | 21420.0 | 27.430 | 2.66 | 2.79 |
| 0.8 | 23161.9 | 21420.0 | 27.430 | 2.06 | 2.06 |

Table 5.13 Predicted and measured plastic hinge lengths for Wall 1 for a web thickness of $\mathbf{2 5 4} \mathbf{m m}$ and a shear span of 19201 mm

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W1 Predicted | W1 Measured |
| 1.8 | 101277.6 | 84700.0 | 19.201 | 3.14 | 5.86 |
| 1.6 | 101004.9 | 84700.0 | 19.201 | 3.10 | 5.57 |
| 1.4 | 100235.0 | 84700.0 | 19.201 | 2.98 | 4.98 |
| 1.2 | 99442.0 | 84700.0 | 19.201 | 2.85 | 4.63 |
| 1.0 | 99117.5 | 84700.0 | 19.201 | 2.79 | 4.00 |
| 0.8 | 97243.5 | 84700.0 | 19.201 | 2.48 | 3.55 |
| 0.6 | 96679.0 | 84700.0 | 19.201 | 2.38 | 3.34 |

Table 5.14 Predicted and measured plastic hinge lengths for Wall 2 for a web thickness of 254 mm and a shear span of 19201 mm

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W2 Predicted | W2 Measured |
| 1.8 | 25796.5 | 21420.0 | 19.201 | 3.26 | 3.68 |
| 1.6 | 25500.8 | 21420.0 | 19.201 | 3.07 | 3.47 |
| 1.4 | 25376.0 | 21420.0 | 19.201 | 2.99 | 3.49 |
| 1.2 | 24945.9 | 21420.0 | 19.201 | 2.71 | 3.16 |
| 1.0 | 24441.0 | 21420.0 | 19.201 | 2.37 | 2.81 |
| 0.8 | 24108.8 | 21420.0 | 19.201 | 2.14 | 2.37 |
| 0.6 | 23296.6 | 21420.0 | 19.201 | 1.55 | 1.63 |

It is clear from these results that as the shear span decreases and the shear stresses become higher, the plastic hinge length is much larger than the one determined from the pure flexural prediction. To make a better prediction, we have to account for the diagonal cracking. A measure of the diagonal cracking in the walls is the average shear stress, determined by:

$$
\begin{equation*}
v=\frac{V}{0.8 l_{w} b_{w}} \tag{5.2}
\end{equation*}
$$

- $v$ : Shear stress.
- $\quad V$ : Shear force.
- $b_{w}$ : Web thickness of wall.

Although the lateral force acting on the walls keeps increasing until they reach their capacity, its increase after yielding is not very large. Therefore, the average shear stress in the walls after yielding remained approximately constant. Table 5.15 shows the average shear stress after yielding in Wall 1 and 2 calculated with Equation 5.2 for the six analyses presented so far in this chapter. The yield drifts (first yield of the longitudinal reinforcement) are also shown; these were determined by looking at the average vertical steel strains in the elements obtained from program VecTor2:

Table 5.15 Yield drifts and shear stresses of Wall 1 and 2

| $L_{s}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $\nu(\mathrm{MPa})$ |  | $\Delta_{y} / L_{s}(\%)$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | Wall 1 | Wall 2 | Wall 1 | Wall 2 |
| 54860.0 | 508.0 | 0.60 | 0.30 | 0.61 | 1.18 |
| 35659.0 | 508.0 | 0.95 | 0.45 | 0.43 | 0.79 |
| 27430.0 | 508.0 | 1.25 | 0.60 | 0.35 | 0.63 |
| 19201.0 | 508.0 | 1.80 | 0.85 | 0.28 | 0.47 |
| 27430.0 | 254.0 | 2.30 | 1.15 | 0.37 | 0.65 |
| 19201.0 | 254.0 | 3.35 | 1.65 | 0.30 | 0.49 |

The results obtained so far show that the plastic hinge length increases as the lateral displacement or drift increases. This trend is plotted in Figure 5.19, which shows the measured plastic hinge lengths for Wall 1 and 2 as a function of the total drift for the six analyses presented so far in this chapter. The continuous lines show the results for Wall 1, and the discontinuous lines show the results for Wall 2:

Figure 5.19 Measured plastic hinge lengths vs. total drift for Wall 1 and 2


Figure 5.20 shows the measured plastic hinge lengths for Wall 1 and 2 as a function of the plastic drift. The plastic drift was calculated as the difference between the total drift and the yield drift:

Figure 5.20 Measured plastic hinge lengths vs. plastic drift for Wall 1 and 2


The quality of the predictions was evaluated by plotting the ratio of the measured to predicted plastic hinge length for different drift levels and shear stresses. This is shown in Figure 5.21 and 5.22:

Figure 5.21 Ratio of measured to predicted plastic hinge lengths vs. drift for Wall 1 for different shear stresses without accounting for shear


Figure 5.22 Ratio of measured to predicted plastic hinge lengths vs. drift for Wall 2 for different shear stresses without accounting for shear


The predictions for Wall 2 are good for all cases because the shear stresses are not very high, since this wall is more slender than Wall 1. For Wall 1, it can be seen that the measured plastic hinge lengths are very large for high shear stresses.

As seen in Section 2.1, diagonal cracking changes the curvature distribution along the height (see Figure 2.5). This has an influence in the moment-curvature relationship. Figure 5.23 shows the momentcurvature relationship for Wall 1 for low and high shear stresses. These relations were developed for the same web thickness but different shear spans (see Table 5.15):

Figure 5.23 Moment-curvature relationships of Wall 1 for different shear stresses


This difference in the moment-curvature relationship is the reason why the pure flexural prediction, which is based on the moment-curvature relationship developed for low shear stresses (pure bending), is not providing good results for Wall 1 for high shear stresses.

In order to improve the predictions for Wall 1, a very simple shear model was developed. As it was mentioned previously, high shear stresses produce more diagonal cracking, which increases the tensile forces in the longitudinal reinforcement. Therefore, we must take into account these additional tensile forces. A simple way to determine the additional tension due to shear (shift) is using the following expression:

$$
\begin{equation*}
\frac{N_{v}}{2}=\frac{V_{s}}{2} \operatorname{Cot} \theta_{c}+V_{c} \operatorname{Cot} \theta_{c} \tag{5.3}
\end{equation*}
$$

- $\quad N_{v}$ : Axial compression due to shear.
- $\quad V_{s}$ : Shear force in the longitudinal reinforcement.
- $\theta_{c}$ : Angle of the crack.
- $V_{c}$ : Shear force in the concrete.

It was consider that the total shear force was carried by the reinforcement ( $V=V_{s}$ and $V_{c}=0$ ), because it gives a smaller shift (which is more conservative, since this gives a smaller plastic hinge length). Also, it was assumed that $\theta_{c}=45^{\circ}$. Therefore, the additional tension due to shear is:

$$
\begin{equation*}
\frac{N_{v}}{2}=\frac{V}{2} \tag{5.4}
\end{equation*}
$$

The additional moment due to shear is then given by:

$$
\begin{equation*}
M_{v}=\frac{V}{2} d_{v} \tag{5.5}
\end{equation*}
$$

- $\quad M_{\nu}$ : Moment due to shear.
- $\quad d_{\nu}$ : Lever arm of the tensile force in the longitudinal reinforcement.

For these walls, it was considered that:

$$
\begin{equation*}
d_{v}=0.8 l_{w} \tag{5.6}
\end{equation*}
$$

So, for high shear stresses, the bending moment diagram is shifted by an amount equal to $M_{v}$. The total bending moment at any section is given by:

$$
\begin{equation*}
M=M_{f}+M_{v} \tag{5.7}
\end{equation*}
$$

Where $M_{f}$ is the moment due to flexure. This simple model was used to predict the plastic hinge length in Wall 1 for high shear stresses. In typical concrete codes, the following expression is proposed for the cracking shear stress:

$$
\begin{equation*}
v_{c r}=0.17 \sqrt{f_{c}^{\prime}} \tag{5.8}
\end{equation*}
$$

Where $v_{c r}$ and $f^{\prime}{ }_{c}$ are in MPa units. For a concrete cylinder strength of 40 MPa , used for Wall 1 and 2, the cracking shear stress is 1.08 MPa . However, Figure 5.21 for Wall 1 shows that for shear stresses up to 1.25 MPa , the pure flexural prediction provides good results. Also, Figure 5.22 for Wall 2 shows that the pure flexural prediction provides good results in all cases, for which the highest shear stress is 1.65 MPa . Therefore, the shear model should be used for shear stresses that are higher than these.

Figure 5.21 suggests that the influence of diagonal cracking in the plastic hinge length is significant for shear stresses higher or equal than 1.8 MPa ; this value is equal to 0.29 times the square root of $f^{\prime}$. Based on this, the cracking shear stress should be:

$$
\begin{equation*}
v_{c r}=0.3 \sqrt{f_{c}^{\prime}} \tag{5.9}
\end{equation*}
$$

The following expression was proposed to account for shear stresses:

$$
M_{v}= \begin{cases}0 & ; v<v_{c r}  \tag{5.10}\\ \frac{V}{2} d_{v} & ; v \geq v_{c r}\end{cases}
$$

Where $v_{c r}$ is calculated with Equation 5.9. The plastic hinge length in Wall 1 was predicted for different drift levels and for shear stresses of $1.8,2.3$ and 3.35 MPa , using this model. For the linear bending moment diagram of Wall 1 , since it is now shifted by $M_{v}$, the yield moment is located at a larger distance from the base, so the plastic hinge length is longer. The moment due to shear was first calculated with Equation 5.5, and then added to the maximum bending moment at the base. Then, this moment was applied in Equation 5.1 to determine the plastic hinge length. The results are shown in Table 5.16 to 5.18:

Table 5.16 Predicted and measured plastic hinge lengths for Wall 1 for a shear span of 19201 mm using shear

| Drift (\%) | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ | $d_{v}(\mathrm{~mm})$ | $M_{v}(\mathrm{kN} . \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1.2 | 5644.1 | 7620.0 | 508.0 | 1.82 | 6096.0 | 17203.2 |
| 1.0 | 5591.5 | 7620.0 | 508.0 | 1.81 | 6096.0 | 17042.9 |
| 0.8 | 5550.1 | 7620.0 | 508.0 | 1.79 | 6096.0 | 16916.7 |
| 0.6 | 5448.5 | 7620.0 | 508.0 | 1.76 | 6096.0 | 16607.0 |
| Drift (\%) | $M_{f}(\mathrm{kN} . \mathrm{m})$ | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN.m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
|  |  | 108372.4 | 125575.6 | 90680.0 | 19.201 | 5.34 |
| 1.2 | 1081 Predicted | W1 Measured |  |  |  |  |
| 1.0 | 107362.4 | 124405.3 | 90680.0 | 19.201 | 5.21 | 4.35 |
| 0.8 | 106567.5 | 123484.2 | 90680.0 | 19.201 | 5.10 | 3.40 |
| 0.6 | 104616.6 | 121223.7 | 90680.0 | 19.201 | 4.84 | 3.01 |

Table 5.17 Predicted and measured plastic hinge lengths for Wall 1 for a web thickness of $\mathbf{2 5 4 m m}$ and a shear span of $\mathbf{2 7 4 3 0 \mathrm { mm }}$ using shear model

| Drift (\%) | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ | $d_{v}(\mathrm{~mm})$ | $M_{v}(\mathrm{kN} . \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 3656.3 | 7620.0 | 254.0 | 2.36 | 6096.0 | 11144.4 |
| 1.6 | 3639.7 | 7620.0 | 254.0 | 2.35 | 6096.0 | 11093.8 |
| 1.4 | 3571.7 | 7620.0 | 254.0 | 2.31 | 6096.0 | 10886.5 |
| 1.2 | 3580.1 | 7620.0 | 254.0 | 2.31 | 6096.0 | 10912.1 |
| 1.0 | 3565.5 | 7620.0 | 254.0 | 2.30 | 6096.0 | 10867.6 |
| 0.8 | 3524.0 | 7620.0 | 254.0 | 2.28 | 6096.0 | 10741.2 |
| Drift (\%) | $M_{f}(\mathrm{kN} . \mathrm{m})$ | $M_{m a x}(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ |  | $L_{p}(\mathrm{~m})$ |
|  |  |  |  | W1 Predicted | W 1 Measured |  |
| 1.8 | 100292.3 | 111436.7 | 84700.0 | 27.430 | 6.58 | 6.26 |
| 1.6 | 99837.0 | 110930.8 | 84700.0 | 27.430 | 6.49 | 5.83 |
| 1.4 | 97971.7 | 108858.3 | 84700.0 | 27.430 | 6.09 | 5.06 |
| 1.2 | 98202.1 | 109114.3 | 84700.0 | 27.430 | 6.14 | 4.72 |
| 1.0 | 97801.7 | 108669.3 | 84700.0 | 27.430 | 6.05 | 4.35 |
| 0.8 | 96663.3 | 107404.5 | 84700.0 | 27.430 | 5.80 | 3.89 |

Table 5.18 Predicted and measured plastic hinge lengths for Wall 1 for a web thickness of 254 mm and a shear span of 19201 mm using shear model

| Drift (\%) | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ | $d_{v}(\mathrm{~mm})$ | $M_{v}(\mathrm{kN} . \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1.8 | 5274.6 | 7620.0 | 254.0 | 3.41 | 6096.0 | 16077.0 |
| 1.6 | 5260.4 | 7620.0 | 254.0 | 3.40 | 6096.0 | 16033.7 |
| 1.4 | 5220.3 | 7620.0 | 254.0 | 3.37 | 6096.0 | 15911.5 |
| 1.2 | 5179.0 | 7620.0 | 254.0 | 3.34 | 6096.0 | 15785.6 |
| 1.0 | 5162.1 | 7620.0 | 254.0 | 3.33 | 6096.0 | 15734.1 |
| 0.8 | 5064.5 | 7620.0 | 254.0 | 3.27 | 6096.0 | 15436.6 |
| 0.6 | 5035.1 | 7620.0 | 254.0 | 3.25 | 6096.0 | 15347.0 |
| Drift (\%) | $M_{f}(\mathrm{kN} . \mathrm{m})$ | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| $)$ |  |  |  |  |  |  |
| 1.8 | 101277.6 | 117354.6 | 84700.0 | 19.201 | 51 Predicted | W1 Measured |
| 1.6 | 101004.9 | 117038.6 | 84700.0 | 19.201 | 5.31 | 5.86 |
| 1.4 | 100235.0 | 116146.5 | 84700.0 | 19.201 | 5.20 | 4.57 |
| 1.2 | 99442.0 | 115227.6 | 84700.0 | 19.201 | 5.09 | 4.63 |
| 1.0 | 99117.5 | 114851.6 | 84700.0 | 19.201 | 5.04 | 4.00 |
| 0.8 | 97243.5 | 112680.1 | 84700.0 | 19.201 | 4.77 | 3.55 |
| 0.6 | 96679.0 | 112025.9 | 84700.0 | 19.201 | 4.68 | 3.34 |

The quality of these predictions was evaluated by plotting the ratio of the measured to predicted plastic hinge length. This is shown in Figure 5.24:

Figure 5.24 Ratio of measured to predicted plastic hinge lengths vs. drift for Wall 1 for different shear stresses using the shear model


Although the predictions are now very large for low drifts (which is conservative), they are good for high drifts, which are of more interest. Considering the simplicity of the model used, the results are satisfactory. The reason why this shear model does not work for low drifts is because at this point, there are not many diagonal cracks formed in the wall, so their influence is still not significant.

It can be concluded that the diagonal cracking in walls has a significant influence in the plastic hinge length. For high shear stresses, the actual plastic hinge length is larger than the one predicted using a pure flexural prediction. The effect of shear needs to be included to estimate the length of the plastic hinge.

### 5.9 Influence of axial load

The fourth parameter studied was the axial load. As seen in Chapter 2, the axial load was also considered in some of the models developed for plastic hinge length. Different conclusions were made regarding the influence of axial load. Some researchers concluded that it has a significant influence in the plastic hinge length, while others did not find any significant dependence.

Studying the influence of axial load is of particular importance for coupled walls, which are subjected to high tensile and compressive axial loads. Previous studies presented in Chapter 2 were done
for compressive axial loads only, not tensile axial loads. In this study, the influence of both tensile and compressive axial loads was investigated.

The first analysis made for Wall 1 and 2 , for a shear span equal to the wall height and a web thickness of 508 mm , described in Section 5.3, were performed for a compressive axial load ratio ( $P / f^{\prime}{ }_{c} A_{g}$ ) of 0.1 . To study the influence of compressive axial loads, three new analyses for each wall were performed in VecTor2 for the following axial load ratios:

- $\quad 0.3$ ( 46452 kN for Wall 1 and 23226 kN for Wall 2).
- $\quad 0.2$ ( 30968 kN for Wall 1 and 15484 kN for Wall 2).
- Zero.

To study the influence of tensile axial loads, two additional analyses for each wall were performed. The tensile axial load applied cannot exceed the pure tension capacity of the walls $\left(f_{y} A_{s}\right)$, equal to 15200 kN for Wall 1 and 7680 kN for Wall 2 . The following axial load ratios were selected:

- 0.02 ( 3800 kN for Wall 1 and 1920 kN for Wall 2), which is equal to $1 / 4$ of the pure tension capacity.
- $0.05(7600 \mathrm{kN}$ for Wall 1 and 3840 kN for Wall 2), which is equal to $1 / 2$ of the pure tension capacity.

The models used for these analyses had the same characteristics as the previous ones, described in Section 5.3. The only difference was the magnitude of the constant axial loads; these loads were also equally distributed among all the six nodes at the top of each wall. The results from the analysis were used to determine the curvatures and bending moments for different drift levels, using the same procedure described in Section 5.4 and 5.6.

Since the axial load was different for all these analyses, the moment-curvature relationship was also different in all cases. The yield curvature and yield moment were also different; these were determined using the same procedures described in Section 5.4. Figure 5.25 shows the moment-curvature relationship for Wall 1 for the six axial load ratios considered (the negative sign indicates compression):

Figure 5.25 Moment-curvature relationships of Wall 1 for different axial load ratios


Figure 5.26 shows the moment-curvature relationship for Wall 2 for the six axial load ratios considered:

Figure 5.26 Moment-curvature relationships of Wall 2 for different axial load ratios


Table 5.19 shows the yield curvature and yield moment for each of the cases analyzed:

Table 5.19 Yield curvature and yield moment of Wall 1 and 2 for different axial load ratios

| $P$ Pff $A_{g}$ | $P(\mathrm{kN})$ |  | $\phi_{y}(\mathrm{rad} / \mathrm{m})$ |  | $M_{v}(\mathrm{kN} . \mathrm{m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Wall 1 | Wall 2 | Wall 1 | Wall 2 | Wall 1 | Wall 2 |
| -0.30 | -46452.0 | -23226.0 | 0.00062 | 0.00120 | 157840.0 | 39480.0 |
| -0.20 | -30968.0 | -15484.0 | 0.00053 | 0.00104 | 127380.0 | 31800.0 |
| -0.10 | -15484.0 | -7742.0 | 0.00045 | 0.00090 | 90680.0 | 22740.0 |
| 0.00 | 0.0 | 0.0 | 0.00037 | 0.00075 | 47910.0 | 12150.0 |
| 0.02 | 3800.0 | 1920.0 | 0.00035 | 0.00070 | 36270.0 | 9300.0 |
| 0.05 | 7600.0 | 3840.0 | 0.00032 | 0.00064 | 23900.0 | 6040.0 |

Table 5.20 to 5.29 show the measured (from curvature distribution) and predicted (using Equation 5.1) plastic hinge lengths for different drift levels. The results for Wall 1 and 2 for a compressive axial load ratio of 0.1 were presented in Table 5.2 and 5.3 , respectively:

Table 5.20 Predicted and measured plastic hinge lengths for Wall 1 for a compressive axial load ratio of 0.3

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  |  |
| 1.2 | 170170.2 | 157840.0 | 54.860 | 3.98 | 4.28 |
| 1.0 | 168431.2 | 157840.0 | 54.860 | 3.45 | 3.43 |
| 0.8 | 160306.4 | 157840.0 | 54.860 | 0.84 | 0.78 |

Table 5.21 Predicted and measured plastic hinge lengths for Wall 2 for a compressive axial load ratio of $\mathbf{0 . 3}$

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W2 Measured |  |
| 2.0 | 42522.0 | 39480.0 | 54.860 | 3.92 | 3.94 |
| 1.8 | 41781.4 | 39480.0 | 54.860 | 3.02 | 3.03 |
| 1.6 | 40722.6 | 39480.0 | 54.860 | 1.67 | 1.30 |

Table 5.22 Predicted and measured plastic hinge lengths for Wall 1 for a compressive axial load ratio of 0.2

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W1 Predicted | W1 Measured |
| 1.4 | 141950.3 | 127380.0 | 54.860 | 5.63 | 5.93 |
| 1.2 | 141741.8 | 127380.0 | 54.860 | 5.56 | 5.64 |
| 1.0 | 139481.6 | 127380.0 | 54.860 | 4.76 | 4.71 |
| 0.8 | 134066.9 | 127380.0 | 54.860 | 2.74 | 2.61 |

Table 5.23 Predicted and measured plastic hinge lengths for Wall 2 for a compressive axial load ratio of 0.2

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W2 Measured |  |
| 2.0 | 35472.5 | 31800.0 | 54.860 | 5.68 | 5.38 |
| 1.8 | 34781.2 | 31800.0 | 54.860 | 4.70 | 4.50 |
| 1.6 | 33826.7 | 31800.0 | 54.860 | 3.29 | 3.17 |
| 1.4 | 32614.3 | 31800.0 | 54.860 | 1.37 | 1.23 |

Table 5.24 Predicted and measured plastic hinge lengths for Wall 1 for no axial load

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W1 Measured |  |
| 2.0 | 57690.8 | 47910.0 | 54.860 | 9.30 | 10.10 |
| 1.8 | 56730.7 | 47910.0 | 54.860 | 8.53 | 9.32 |
| 1.6 | 55929.8 | 47910.0 | 54.860 | 7.87 | 8.59 |
| 1.4 | 55145.3 | 47910.0 | 54.860 | 7.20 | 7.87 |
| 1.2 | 54278.5 | 47910.0 | 54.860 | 6.44 | 6.97 |

Table 5.25 Predicted and measured plastic hinge lengths for Wall 2 for no axial load

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W2 Measured |  |
| 2.0 | 13720.5 | 12150.0 | 54.860 | 6.28 | 6.24 |
| 1.8 | 13391.3 | 12150.0 | 54.860 | 5.09 | 5.78 |
| 1.6 | 13314.5 | 12150.0 | 54.860 | 4.80 | 4.99 |
| 1.4 | 12957.9 | 12150.0 | 54.860 | 3.42 | 4.00 |
| 1.2 | 12683.6 | 12150.0 | 54.860 | 2.31 | 2.21 |

Table 5.26 Predicted and measured plastic hinge lengths for Wall 1 for a tensile axial load ratio of $\mathbf{0 . 0 2}$

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W1 Predicted | W1 Measured |
| 2.0 | 44381.7 | 36270.0 | 54.860 | 10.03 | 10.01 |
| 1.8 | 43778.3 | 36270.0 | 54.860 | 9.41 | 9.73 |
| 1.6 | 42906.0 | 36270.0 | 54.860 | 8.48 | 8.86 |
| 1.4 | 42116.0 | 36270.0 | 54.860 | 7.81 | 8.06 |
| 1.2 | 41260.2 | 36270.0 | 54.860 | 6.64 | 6.84 |

Table 5.27 Predicted and measured plastic hinge lengths for Wall 2 for a tensile axial load ratio of $\mathbf{0 . 0 2}$

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W2 Measured |  |
| 2.0 | 10461.8 | 9300.0 | 54.860 | 6.09 | 6.18 |
| 1.8 | 10275.3 | 9300.0 | 54.860 | 5.21 | 5.65 |
| 1.6 | 10209.4 | 9300.0 | 54.860 | 4.89 | 5.05 |
| 1.4 | 9918.7 | 9300.0 | 54.860 | 3.42 | 3.82 |
| 1.2 | 9627.9 | 9300.0 | 54.860 | 1.87 | 2.17 |

Table 5.28 Predicted and measured plastic hinge lengths for Wall 1 for a tensile axial load ratio of 0.05

| Drift (\%) | $M_{\max }(\mathrm{kN} . \mathrm{m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W1 Measured |  |
| 2.0 | 30381.5 | 23900.0 | 54.860 | 11.70 | 11.19 |
| 1.8 | 29778.0 | 23900.0 | 54.860 | 10.83 | 10.29 |
| 1.6 | 29031.9 | 23900.0 | 54.860 | 9.70 | 9.17 |
| 1.4 | 28423.0 | 23900.0 | 54.860 | 8.73 | 8.35 |
| 1.2 | 27792.1 | 23900.0 | 54.860 | 7.68 | 7.03 |

Table 5.29 Predicted and measured plastic hinge lengths for Wall $\mathbf{2}$ for a tensile axial load ratio of $\mathbf{0 . 0 5}$

| Drift (\%) | $M_{\max }(\mathrm{kN.m})$ | $M_{y}(\mathrm{kN} . \mathrm{m})$ | $L_{s}(\mathrm{~m})$ | $L_{p}(\mathrm{~m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | W2 Measured |  |
| 2.0 | 6906.9 | 6040.0 | 54.860 | 6.89 | 6.04 |
| 1.8 | 6868.5 | 6040.0 | 54.860 | 6.62 | 5.87 |
| 1.6 | 6665.5 | 6040.0 | 54.860 | 5.15 | 5.14 |
| 1.4 | 6616.1 | 6040.0 | 54.860 | 4.78 | 3.79 |
| 1.2 | 6292.4 | 6040.0 | 54.860 | 2.20 | 1.18 |

It can be observed that the flexural prediction provides good results for the different axial load ratios and drift levels. Figure 5.27 shows the measured plastic hinge lengths for Wall 1 as a function of the total drift:

Figure 5.27 Measured plastic hinge lengths vs. total drift for Wall 1 for different axial load ratios


Figure 5.28 shows the measured plastic hinge lengths for Wall 2 as a function of the total drift:

Figure 5.28 Measured plastic hinge lengths vs. total drift for Wall 2 for different axial load ratios


These plots show that the plastic hinge length of the wall reduces when it is subjected to compressive axial forces, and increases when it is subjected to tensile axial forces. This is because the compression load reduces the length in which the inelastic curvatures occur, and so, the length of the plastic hinge. The tension load has the opposite effect.

The quality of the predictions was evaluated by plotting the ratio of the measured to predicted plastic hinge length for different drift levels and axial loads. This is shown in Figure 5.29 and 5.30:

Figure 5.29 Ratio of measured to predicted plastic hinge lengths vs. drift for Wall 1 for different axial load ratios


Figure 5.30 Ratio of measured to predicted plastic hinge lengths vs. drift for Wall $\mathbf{2}$ for different axial load ratios


Most of the predictions shown in these figures are good. Some of the predictions for Wall 2 are not very good for the lowest drifts. This is because the wall at these drifts has just started yielding, the plastic hinge length is very small, and the ratio of measured to predicted becomes either too large or too small. For large drifts, all the predictions are good.

It can be concluded that the length of the plastic hinge reduces with the addition of compression and increases with the addition of tension. The flexural prediction can be used to estimate the plastic hinge length in walls subjected to compressive and tensile axial loads.

### 5.10 Summary

Table 5.30 shows the measured plastic hinge lengths at maximum drift as a function of the parameters studied for the 22 analyses presented in this chapter. These are measured plastic hinge lengths considering a linearly varying inelastic curvature:

Table 5.30 Measured plastic hinge lengths at maximum drift

| Wall | $L_{s}(\mathrm{~m})$ | $v(\mathrm{MPa})$ | $P / f_{c}^{f} A_{g}$ | $L_{p}(\mathrm{~m})$ | $L_{p} / l_{w}$ | $L_{p} / L_{s}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Wall 1 | 54.860 | 0.60 | -0.10 | 8.39 | 1.10 | 0.15 |
| Wall 2 | 54.860 | 0.30 | -0.10 | 6.21 | 1.63 | 0.11 |
| Wall 1 | 35.659 | 0.95 | -0.10 | 6.06 | 0.80 | 0.17 |
| Wall 2 | 35.659 | 0.45 | -0.10 | 4.74 | 1.25 | 0.13 |
| Wall 1 | 27.430 | 1.25 | -0.10 | 5.55 | 0.73 | 0.20 |
| Wall 2 | 27.430 | 0.60 | -0.10 | 4.28 | 1.12 | 0.16 |
| Wall 1 | 19.201 | 1.80 | -0.10 | 4.45 | 0.58 | 0.23 |
| Wall 2 | 19.201 | 0.85 | -0.10 | 3.16 | 0.83 | 0.16 |
| Wall 1 | 27.430 | 2.30 | -0.10 | 6.26 | 0.82 | 0.23 |
| Wall 2 | 27.430 | 1.15 | -0.10 | 4.42 | 1.16 | 0.16 |
| Wall 1 | 19.201 | 3.35 | -0.10 | 5.86 | 0.77 | 0.30 |
| Wall 2 | 19.201 | 1.65 | -0.10 | 3.68 | 0.97 | 0.19 |
| Wall 1 | 54.860 | 1.00 | -0.30 | 4.28 | 0.56 | 0.08 |
| Wall 2 | 54.860 | 0.50 | -0.30 | 3.94 | 1.04 | 0.07 |
| Wall 1 | 54.860 | 0.80 | -0.20 | 5.93 | 0.78 | 0.11 |
| Wall 2 | 54.860 | 0.40 | -0.20 | 5.38 | 1.41 | 0.10 |
| Wall 1 | 54.860 | 0.32 | 0.00 | 10.10 | 1.33 | 0.18 |
| Wall 2 | 54.860 | 0.16 | 0.00 | 6.24 | 1.64 | 0.11 |
| Wall 1 | 54.860 | 0.25 | 0.02 | 10.01 | 1.31 | 0.18 |
| Wall 2 | 54.860 | 0.12 | 0.02 | 6.18 | 1.62 | 0.11 |
| Wall 1 | 54.860 | 0.17 | 0.05 | 11.19 | 1.47 | 0.20 |
| Wall 2 | 54.860 | 0.08 | 0.05 | 6.04 | 1.59 | 0.11 |

## CHAPTER 6: SYSTEMS OF WALLS OF DIFFERENT LENGTHS CONNECTED TOGETHER BY RIGID SLABS

### 6.1 Scope of analysis

Program VecTor2 was used to investigate plastic hinge lengths in a system of walls of different lengths interconnected by rigid slabs at various levels. The parametric study in Chapter 5 was performed for Wall 1 and 2 separately. A new analysis for these two walls was performed, being under the same conditions as the ones described in Section 5.2, except that they were now combined together, to see how this influences the length of the plastic hinge. Additional analyses with walls of other lengths were also performed.

In high-rise buildings, it is common to have parallel walls of different lengths providing lateral resistance. These walls, as well as the gravity columns, are interconnected by rigid slabs at numerous floor levels. As a result, the lateral displacement of all these elements is the same at these levels.

To study how the plastic hinge length in walls of different lengths is influenced when they are connected together, three new analyses were performed in VecTor2 for the following wall systems:

- The first system consisted of Wall 1 and 2 combined together. Wall 1 was two times longer than Wall 2.
- The second system consisted of Wall 1 combined together with a wall that was four times shorter. This wall will be referred to as Wall 3.
- The third system consisted of Wall 1 combined together with a column that was eight times shorter. This column will be referred to as Column 1.


### 6.2 Wall 1 combined with Wall 2

The description of Wall 1 and 2 and their individual analytical models have already been given in Section 5.2 and 5.3 , respectively. The description of the analytical model for these two walls combined together and the results obtained will be presented in the following sections.

### 6.2.1 Description of wall system model

The model used to analyze Wall 1 and 2 combined is presented in Figure 6.1. The slabs were provided every 2743 mm , resulting in a 20 storey building. The thickness of the slabs was 203 mm .

Figure 6.1 Model of Wall 1 and 2 combined


### 6.2.2 Analytical model of wall system

Wall 1 and 2 were modeled and analyzed together using program VecTor2. The finite element mesh used for each wall, as well as the boundary conditions, material properties and loads, were the same as the ones described in Section 5.3.

The slabs interconnecting the walls at each floor level were modeled using truss bar elements. This type of element in VecTor2 is typically used to model the reinforcing steel as discrete bars (see Section 3.3), so reinforcement properties have to be assigned to it. These elements were provided with a very high strength and stiffness, so that they do not yield and do not deform axially, in order to force the lateral displacements in both walls to be the same. A new material type was created to represent this region of the wall system.

Since the slabs are very stiff, the strains in the transverse direction of the wall at the floor levels are very small. To simulate that effect in VecTor2, the rectangular elements located at each storey level were also provided with very stiff properties. These elements had a height equal to the slab thickness ( 203 mm ). These were only provided up to half the height of the wall, where the mesh was more refined. A new material type was created to represent this region of the wall system.

All nodes and elements were numbered in the horizontal (short) direction of the wall system. The complete mesh consisted of 3357 nodes, 3066 rectangular elements, 116 triangular elements and 20 truss elements.

A monotonic lateral load was applied at the top of the Wall 1. This load was applied in a displacement-control mode, in increments of 1 mm . Additionally, a constant axial load of 15484 kN in Wall 1 and 7742 kN in Wall 2 was applied; these loads were equally distributed among the six nodes at the top of each wall. The self-weight of the walls was not considered.

Figure 6.2 shows the finite element model of Wall 1 and 2 , created in the pre-processor FormWorks:

Figure 6.2 Finite element model of Wall 1 and 2 in FormWorks


Figure 6.3 shows a more detailed view of the elements representing the slabs at each storey in the finite element model:

Figure 6.3 Detail of elements representing the slabs at each storey in finite element model of Wall 1 and 2


Table 6.1 and 6.2 show the material properties in the different regions of the wall system:

Table 6.1 Material properties in rectangular and triangular elements of Wall 1 and 2

| Concrete properties | Material 1 |  | Material 2 |  | Material 3 |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Color | 508 |  | 508 |  | 2000 |  |
| Thickness (mm) | 40 |  | 40 |  | 40 |  |
| Cylinder compressive <br> strength (MPa) | 40 |  |  |  |  |  |
| Reinforcement <br> component properties | $x$-direction | $y$-direction | $x$-direction | $y$-direction | $x$-direction | $y$-direction |
| Reinforcement ratio (\%) | 0.8 | 2.0 | 0.5 | 0.5 | 0.5 | 0.5 |
| Reinforcement diameter <br> $(\mathrm{mm})$ | 15 | 25 | 15 | 15 | 15 | 15 |
| Yield strength (MPa) | 400 | 400 | 400 | 400 | 400 | 400 |
| Ultimate strength (MPa) | 650 | 650 | 650 | 650 | 650 | 650 |
| Elastic modulus (MPa) | 200000 | 200000 | 200000 | 200000 | 200000 | 200000 |
| Strain hardening <br> modulus $(M P a)$ | 4875 | 4875 | 4875 | 4875 | 4875 | 4875 |
| Strain hardening strain <br> $(\mathrm{mm} / \mathrm{m})$ | 10 | 10 | 10 | 10 | 10 | 10 |

Table 6.2 Material properties in truss elements of Wall 1 and 2

| Reinforcement properties | Material 4 |
| :---: | :---: |
| Color |  |
| Cross-sectional area $\left(\mathrm{mm}^{2}\right)$ | 2500000 |
| Reinforcement diameter $(\mathrm{mm})$ | 55 |
| Yield strength $(\mathrm{MPa})$ | 4000 |
| Ultimate strength $(\mathrm{MPa})$ | 4000 |
| Elastic modulus $(\mathrm{MPa})$ | 1000000 |
| Strain hardening modulus $(\mathrm{MPa})$ | 1 |
| Strain hardening strain $(\mathrm{mm} / \mathrm{m})$ | 80 |

### 6.2.3 Analytical results

The results from the analysis were used to determine the curvatures, using the same procedure described in Section 5.4 and 5.6.

The ultimate drift for the wall system was $2 \%$. Figure 6.4 shows the curvature distribution up to the mid-height of both walls for this drift, when they are alone and combined:

Figure 6.4 Curvatures up to the mid-height of Wall 1 and $\mathbf{2}$ alone and combined for a drift of $\mathbf{2 \%}$


Several observations can be made for this figure. Wall 1 is pulling Wall 2 along, resulting in equal curvatures in the elastic range. After yielding, a more complex phenomenon is occurring. The curvature distribution in Wall 1 remains approximately the same, while the curvatures in Wall 2 at the base have increased considerably. The curvature distribution will be examined in more detail in the next section.

The influence of the interconnection in the moment-curvature relationship of the walls was analyzed. To develop this relationship, the bending moment diagram along the height of the wall must be calculated. The truss elements representing the slabs forced Wall 1 and 2 to displace the same at the storey levels, resulting in high axial forces in these elements. Therefore, both walls were now subjected to a system of lateral forces at each storey level, and Wall 1 is additionally subjected to the lateral load
acting at the top. The axial forces in the truss elements obtained from program VecTor2 were used to determine the system of lateral forces. Then, the bending moment diagram was calculated. Figure 6.5 shows the bending moments along the height of both walls for a drift of $2 \%$ :

Figure 6.5 Bending moments along the height of Wall 1 and 2 alone and combined for a drift of $\mathbf{2 \%}$


Figure 6.6 shows the moment-curvature relationship of Wall 1:

Figure 6.6 Moment-curvature relationship of Wall 1 alone and combined


Figure 6.7 shows the moment-curvature relationship of Wall 2:

Figure 6.7 Moment-curvature relationship of Wall 2 alone and combined


It can be seen that the moment-curvature relationship does not change when the walls are connected, so this is not causing the differences in the curvature distributions shown in Figure 6.4. Therefore, as seen in Section 5.4, the yield curvature and yield moment for Wall 1 were $0.00045 \mathrm{rad} / \mathrm{m}$ and $90680 \mathrm{kN} . \mathrm{m}$, respectively; and for Wall 2 they were $0.0009 \mathrm{rad} / \mathrm{m}$ and $22740 \mathrm{kN} . \mathrm{m}$.

The deflected shapes and slopes of both walls were also investigated. The total displacement at each storey, as well as its flexural and shear displacement components, were determined. The total horizontal displacements at any level or location were obtained directly from program VecTor2. The flexural displacement at a particular storey was calculated by integrating the curvatures along the height up to that level, using the second moment-area theorem:

$$
\begin{equation*}
\Delta_{f}=\int \phi(x) x d x \tag{6.1}
\end{equation*}
$$

This integral was solved with the following numerical scheme:

Figure 6.8 Numerical scheme for curvature integration


Then:

$$
\begin{equation*}
\Delta_{f}=\sum_{i}\left(\frac{\phi_{i-1}+\phi_{i}}{2}\right)\left(x_{i}+\frac{\Delta x_{i}}{2}\right) \Delta x_{i} \tag{6.2}
\end{equation*}
$$

The shear displacement at a particular storey was then calculated as the difference between the total and flexural displacement:

$$
\begin{equation*}
\Delta_{v}=\Delta-\Delta_{f} \tag{6.3}
\end{equation*}
$$

The slopes were calculated at the same location as the curvatures, by integrating the curvatures along the height using the first moment-area theorem:

$$
\begin{equation*}
\theta=\int \phi(x) d x \tag{6.4}
\end{equation*}
$$

Using the numerical scheme from Figure 6.8:

$$
\begin{equation*}
\theta=\sum_{i}\left(\frac{\phi_{i-1}+\phi_{i}}{2}\right) \Delta x_{i} \tag{6.5}
\end{equation*}
$$

### 6.2.4 Discussion of analytical results

The curvature distribution in Figure 6.4 shows that the curvatures in Wall 1 remain almost the same when it is connected with Wall 2 , and its plastic hinge length has increased slightly. In Wall 2, the curvatures have increased considerably at the base, and its plastic hinge has been reduced. Using the yield curvature to measure the plastic hinge length for a drift of $2 \%$ from the curvature distribution when the walls are connected together, it was equal to 9.60 m for Wall 1 and 4.64 m for Wall 2 . The change in the plastic hinge lengths can also be observed from the distribution of tensile steel strains in the wall. The average vertical steel strains in the elements along the height of the walls obtained from program VecTor2 were studied for this. The steel strains were observed in the exterior and interior steel layer in the boundary zones, which were shown in Figure 5.14. Figure 6.9 shows the steel strains up to the midheight of Wall 1, alone and combined, for a drift of $2 \%$ :

Figure 6.9 Steel strains up to the mid-height of Wall 1 alone and combined for a drift of $\mathbf{2 \%}$


Figure 6.10 shows the steel strains up to the mid-height of Wall 2, alone and combined, for a drift of $2 \%$ :

Figure 6.10 Steel strains up to the mid-height of Wall 2 alone and combined for a drift of $\mathbf{2 \%}$


For Wall 1, the steel strains in both layers are approximately the same, but they exceed the yield strain ( 0.002 for these walls) at a slightly larger distance from the base when it is combined. For Wall 2 , the steel strains are larger for the combined case, and they exceed the yield strain at a slightly smaller distance from the base.

All these observations are revealing that the curvature distributions and plastic hinge lengths in both walls are different, even when they are combined together by rigid slabs. In order to understand this phenomenon, the deflected shapes and slopes of both walls were studied.

Figure 6.11 shows the total lateral displacements at the faces of both walls (right face for Wall 1 and left face for Wall 2) up to the fourth storey when they are combined, for a drift of $2 \%$; these were obtained directly from program VecTor2:

Figure 6.11 Displacements at wall faces up to the fourth storey of Wall $\mathbf{1}$ and $\mathbf{2}$ combined for a drift of $\mathbf{2 \%}$


Although the lateral displacements in both walls are equal at the storey levels, these are different between these levels. So, the deflected shapes are not strictly the same. The bigger differences between the displacements in both walls are located at the bottom. Figure 6.12 shows a more detailed view of the lateral displacements at the faces of both walls up to the first storey when they are combined, for a drift of $2 \%$ :

Figure 6.12 Displacements at wall faces up to the first storey of Wall 1 and $\mathbf{2}$ combined for a drift of $\mathbf{2 \%}$


The displacements are not the same between storey levels. Wall 1 has larger displacements than Wall 2. It can also be seen that Wall 2 ends up having bigger slopes. The variation of the slope in Wall 2 is larger than in Wall 1, which is the reason why the curvatures are larger in this wall for the combined case.

Table 6.3 shows the displacement components up to the fourth storey of both walls when they are combined for a drift of $2 \%$ :

Table 6.3 Displacement components up to the fourth storey of Wall 1 and 2 combined for a drift of $\mathbf{2 \%}$

| Storey | Height <br> $(\mathrm{m})$ | Wall 1 |  |  | Wall 2 |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | $\Delta_{f}(\mathrm{~mm})$ | $\Delta_{v}(\mathrm{~mm})$ | $\Delta(\mathrm{mm})$ | $\Delta_{f}(\mathrm{~mm})$ | $\Delta_{v}(\mathrm{~mm})$ | $\Delta(\mathrm{mm})$ |
| 0 |  | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 1 | 2.74 | 8.02 | 10.59 | 18.61 | 10.38 | 8.23 | 18.61 |
| 2 | 5.49 | 34.02 | 18.84 | 52.87 | 38.88 | 13.99 | 52.87 |
| 3 | 8.23 | 70.91 | 23.46 | 94.37 | 74.26 | 20.11 | 94.37 |
| 4 | 10.97 | 112.96 | 27.48 | 140.44 | 114.45 | 25.99 | 140.44 |

Figure 6.13 shows these results graphically:

Figure 6.13 Displacement components up to the fourth storey of Wall 1 and 2 combined for a drift of $\mathbf{2 \%}$


The total displacements are the same at the storey heights, but the flexural displacement is larger in Wall 2 at the lower levels. Wall 1 has higher shear displacements at the lower levels. Therefore, the displacements in Figure 6.12 in this wall up to the first storey are mostly due to shear. Although Wall 1 has larger total displacements at the bottom, Wall 2 has larger flexural displacements. The difference in shear displacements is causing the slopes to be different, as shown in Figure 6.14:

Figure 6.14 Slopes up to the mid-height of Wall 1 and 2 combined for a drift of 2\%


The slopes in Wall 2 are larger at the lower levels; this was also deduced by looking at Figure 6.12. The inelastic rotation in Wall 2 is concentrated in a shorter height than in Wall 1.

It can be concluded that the curvature distribution of Wall 1 and 2 when they are connected are different because of two reasons. First, the displacements between storey levels are different in both walls, so the deflected shapes are not the same. Second, the shear displacements at the lower levels are larger in Wall 1 than in Wall 2 when they are connected; resulting in lower flexural displacements, lower slopes, and lower curvatures for Wall 1 . Since the curvatures are not the same, the plastic hinge lengths are also different.

Another important conclusion is that when walls of different lengths are combined together, the curvature demand for the shorter wall is much larger than when it is alone. This wall has to be able to sustain that demand.

In order to predict the maximum curvature in Wall 2 for the combined case, a very simple model was developed, based on the observations made from this study. It is shown in figure 6.15:

Figure 6.15 Proposed model to determine maximum curvature in shorter wall


The following assumptions were made regarding this model:

- The inelastic curvatures are linearly varying.
- The plastic hinge length and the maximum curvature in Wall 1 (longer wall) do not change when it is combined with Wall 2 (shorter wall).
- The rotations or slopes in both walls at a distance equal to the plastic hinge length of Wall 1 from the section of maximum curvature are the same.

Based on these assumptions, the maximum curvature in Wall 2 can be estimated. From Figure 6.15, the total rotation in Wall 1 is determined by integrating the curvatures:

$$
\begin{equation*}
\theta_{1}=\left(\frac{\phi_{y, 1}+\phi_{\max , 1}}{2}\right) L_{p, 1} \tag{6.6}
\end{equation*}
$$

The total rotation in Wall 2 is determined similarly:

$$
\begin{equation*}
\theta_{2}=\left(\frac{\phi_{y, 1}+\phi_{y, 2}}{2}\right)\left(L_{p, 1}-L_{p, 2}\right)+\left(\frac{\phi_{y, 2}+\phi_{\max , 2}}{2}\right) L_{p, 2} \tag{6.7}
\end{equation*}
$$

Since we are considering that the rotations are the same:

$$
\begin{equation*}
\theta_{1}=\theta_{2} \tag{6.8}
\end{equation*}
$$

Replacing Equation 6.6 and 6.7 in Equation 6.8, and solving for $\phi_{\text {max } 2,}$, we finally get:

$$
\begin{equation*}
\phi_{\max , 2}=\left(\phi_{\max , 1}+\phi_{y, 2}\right)\left(\frac{L_{p, 1}}{L_{p, 2}}\right)+\phi_{y, 1} \tag{6.9}
\end{equation*}
$$

The measured values of the parameters in Equation 6.9 were used to predict the maximum curvature in Wall 2, and then it was compared with the measured value, obtained from Figure 6.4. The maximum curvature, yield curvature and plastic hinge length for Wall 1 when it is alone are already known. The yield curvature of Wall 2 is also known. The plastic hinge length used for Wall 2 was the one measured in the combined case, that is, 4.64 m . The result is shown in Table 6.4:

Table 6.4 Predicted and measured maximum curvature for Wall 2 combined

| Drift (\%) | $L_{p, I}(\mathrm{~m})$ | $L_{p, 2}(\mathrm{~m})$ | $\phi_{y, I}(\mathrm{rad} / \mathrm{m})$ | $\phi_{p, 2}(\mathrm{rad} / \mathrm{m})$ | $\phi_{\text {max, }, l}(\mathrm{rad} / \mathrm{m})$ | $\phi_{\text {max }, 2}(\mathrm{rad} / \mathrm{m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | Predicted |  |  |  |  |
| 2.0 | 8.39 | 4.64 | 0.00045 | 0.00090 | 0.00326 | 0.00472 | 0.00456 |

The proposed model provides a good prediction of the maximum curvature in Wall 2.

### 6.3 Wall 1 combined with Wall 3

Wall 1 was then combined with a wall that was four times shorter. This wall will be referred to as Wall 3 throughout this chapter. Wall 3 was first analyzed individually, and then it was connected together with Wall 1 to compare the results obtained. The description of the analytical model for Wall 3 alone and combined together with Wall 1, and the results obtained, will be presented in the following sections.

### 6.3.1 Description of wall system model

Wall 3 was first analyzed alone as a cantilever element. It had a rectangular cross-section, and it was 54860 mm high, 508 mm thick and 1905 mm long. The boundary zones were 305 mm long, the vertical reinforcement in these consisted of $6-25 \mathrm{M}$ reinforcing bars enclosed by 15 M ties spaced at 100 mm . The clear cover of the ties was 40 mm . The web had 15 M reinforcing bars spaced at 150 mm vertically and horizontally. The cross-section of Wall 3 is shown in Figure 6.16:

Figure 6.16 Cross-section details of Wall 3


The material properties of Wall 3 were the same ones used for Wall 1 and 2, described in Section 5.2. The wall was fixed at the base. It was subjected to a monotonically increasing lateral load at the top; as well as a constant compressive axial load of $0.2 A_{g} f^{\prime} c$, equal to 7742 kN . This is the same axial load used for Wall 2. Smaller walls usually have a larger tributary area, so the axial load ratio was increased for Wall 3.

The model used to analyze walls 1 and 3 combined was the same as the one presented in Figure 6.1 , except that Wall 2 was replaced by Wall 3 . The slabs were provided every 2743 mm , resulting in a 20 storey building. The thickness of the slabs was 203 mm .

### 6.3.2 Analytical model of wall system

Wall 3 was first modeled and analyzed individually using program VecTor2. The model was very similar to the one created for Wall 1 and 2, described in Section 5.3. Low-powered rectangular and triangular elements were used to model the concrete, with smeared steel to account for the presence of reinforcement. The constitutive models for concrete and steel described in Chapter 3 were used in the analysis. The same two material types from Wall 1 and 2 were used to represent the confined boundaries and the web of Wall 3 in the finite element model.

The analytical model was fixed at the bottom; both the horizontal and vertical displacements were restrained. The finite element mesh was more refined near the base (critical section). Wall 3 had the same refinement as Wall 1 in the vertical direction up to a half of the height, so that the curvatures could be calculated over the same average length.

Meshes of $305 \times 423,305 \times 203,324 \times 423$ and $324 \times 203$ rectangular elements were used; six elements (seven nodes) were used in the transverse direction of the wall. This level of refinement was
maintained up to half the height of the wall. Up from this point, five elements (six nodes) were used in the transverse direction. The transitions were made using triangular elements. All nodes and elements were numbered in the horizontal (short) direction. The complete mesh consisted of 857 nodes, 717 rectangular elements and seven triangular elements.

The material properties used in the analysis were the same ones used for Wall 1 and 2, presented in Section 5.3.

A monotonic lateral load was applied at the top of the wall. This load was applied in a displacement-control mode, in increments of 1 mm . Additionally, a constant axial load of 7742 kN was applied; this load was equally distributed among the six nodes at the top of the wall. The self-weight of the wall was not considered.

Figure 6.17 shows the finite element model of Wall 3, created in the pre-processor FormWorks:

Figure 6.17 Finite element model of Wall 3 in FormWorks


The material properties in the different regions of the wall were the same ones used for Wall 1 and 2 alone, presented in Table 5.1.

Wall 1 and 3 were then modeled and analyzed together using program VecTor2. The finite element mesh used for each wall, as well as the boundary conditions, material properties and loads, were the same as the ones described in Section 5.3 and this section.

As it was done for the model of Wall 1 and 2 combined together, described in Section 6.2.2, the slabs interconnecting the walls at each floor level were modeled using truss bar elements with very stiff properties. The rectangular elements located at each storey level, which had a height equal to the slab thickness, were also provided with very stiff properties. The same material types were used for these elements.

All nodes and elements were numbered in the horizontal (short) direction of the wall system. The complete mesh consisted of 2938 nodes, 2644 rectangular elements, 93 triangular elements and 20 truss elements.

A monotonic lateral load was applied at the top of the Wall 1. This load was applied in a displacement-control mode, in increments of 1mm. Additionally, a constant axial load of 15484 kN in Wall 1 and 7742 kN in Wall 3 was applied; these loads were equally distributed among the six nodes at the top of each wall. The self-weight of the walls was not considered.

Figure 6.18 shows the finite element model of Wall 1 and 3, created in the pre-processor FormWorks:

Figure 6.18 Finite element model of Wall 1 and 3 in FormWorks


The detail of the elements representing the slabs at each storey level in the finite element model was the same as the one shown in Figure 6.3. The material properties in the different regions of the wall system were the same ones used for Wall 1 and 2 combined, presented in Table 6.1 and 6.2.

### 6.3.3 Analytical results

The results from the analysis were used to determine the curvatures, using the same procedure described in Section 5.4 and 5.6.

The ultimate drift for the wall system was $2 \%$. Figure 6.19 shows the curvature distribution up to the mid-height of both walls for this drift, when they are alone and combined:

Figure 6.19 Curvatures up to the mid-height of Wall 1 and 3 alone and combined for a drift of $\mathbf{2 \%}$


The same phenomenon observed for Wall 1 and 2, presented in Section 6.2.3, is seen here. The curvature distribution in Wall 1 remains approximately the same, while the curvatures in Wall 3 at the base have increased considerably.

The moment-curvature relationship of Wall 3 was also determined, using the procedure described in Section 5.4 when it is alone, and the procedure described in Section 6.2 .3 when it is combined. It is presented in Figure 6.20:

Figure 6.20 Moment-curvature relationship of Wall 3 alone and combined


It can be seen that the moment-curvature relationship does not change when the walls are connected. The yield curvature for Wall 3 is $0.00207 \mathrm{rad} / \mathrm{m}$.

### 6.3.4 Discussion of analytical results

The curvature distribution in Figure 6.19 shows that the curvatures in Wall 1 remain almost the same when it is connected with Wall 3 , and its plastic hinge length has increased slightly. In Wall 3, the curvatures have increased considerably at the base. This is the same phenomenon observed for Wall 1 and 2 combined together. However, when Wall 3 is alone, the curvatures at $2 \%$ drift do not reach the yield curvature. Therefore, it is not yielding, so no plastic hinge is formed. When combined with Wall 1 , the curvatures in Wall 3 increase, causing it to yield and develop a plastic hinge. This can also be observed from the distribution of tensile steel strains in the wall. The average vertical steel strains in the elements along the height of the walls obtained from program VecTor2 were studied for this. The steel strains were observed in the exterior steel layer in the boundary zones. Figure 6.21 shows the steel strains up to the mid-height of Wall 3, alone and combined, for a drift of $2 \%$ :

Figure 6.21 Steel strains up to the mid-height of Wall 3 alone and combined for a drift of $\mathbf{2 \%}$


The steel strains do not reach the yield strain when Wall 3 is alone, but they do when it is combined.

Using the yield curvature to measure the plastic hinge length for a drift of $2 \%$ from the curvature distribution when the walls are connected together, it was equal to 9.58 m for Wall 1 and 2.06 m for Wall 3.

In conclusion, the curvature distribution and plastic hinge length of Wall 1 and 3 when they are connected are different, as it was observed for Wall 1 and 2 connected, for the same reasons explained in Section 6.2.4. Also, the curvature demand for the shorter wall is much larger when it is combined than when it is alone, and it has to be able to sustain that demand.

The maximum curvature in Wall 3 was predicted using the model presented in Figure 6.15. The measured values of the parameters in Equation 6.9 were used in the prediction, and then it was compared with the measured value, obtained from Figure 6.19. The maximum curvature, yield curvature and plastic hinge length for Wall 1 when it is alone are already known. The yield curvature of Wall 3 is also known. The plastic hinge length used for Wall 3 was the one measured in the combined case, that is, 2.06 m . The result is shown in Table 6.5:

Table 6.5 Predicted and measured maximum curvature for Wall 3 combined

| Drift (\%) | $L_{p, I}(\mathrm{~m})$ | $L_{p, 2}(\mathrm{~m})$ | $\phi_{y, 1}(\mathrm{rad} / \mathrm{m})$ | $\phi_{y, 2}(\mathrm{rad} / \mathrm{m})$ | $\phi_{\max , I}(\mathrm{rad} / \mathrm{m})$ | $\phi_{\text {max } 2}(\mathrm{rad} / \mathrm{m})$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | Predicted | Measured |  |  |  |
| 2.0 | 8.39 | 2.06 | 0.00045 | 0.00207 | 0.00326 | 0.00530 | 0.00581 |

The proposed model provides a good prediction of the maximum curvature in Wall 3.

### 6.4 Wall 1 combined with Column 1

Wall 1 was finally combined with a column that was eight times shorter. This column will be referred to as Column 1 throughout this chapter. Column 1 was first analyzed individually, and then it was connected together with Wall 1 to compare the results obtained. The description of the analytical model for Column 1 alone and combined together with Wall 1, and the results obtained, will be presented in the following sections.

### 6.4.1 Description of wall system model

Column 1 was first analyzed alone as a cantilever element. It had a rectangular cross-section, and it was 54860 mm high, 508 mm thick and 953 mm long. The vertical reinforcement consisted of $20-25 \mathrm{M}$ reinforcing bars enclosed by 15 M ties spaced at 100 mm . The clear cover of the ties was 40 mm . The cross-section of Column 1 is shown in Figure 6.22:

Figure 6.22 Cross-section details of Column 1


The material properties of Column 1 were the same ones used for Wall 1 and 2, described in Section 5.2. The wall was fixed at the base. It was subjected to a monotonically increasing lateral load at the top; as well as a constant compressive axial load of $0.4 A_{g} f^{\prime} c$, equal to 7742 kN . This is the same axial load used for Wall 2 and 3 . Gravity columns typically have a larger tributary area compared to walls, so the axial load ratio was increased for Column 1.

The model used to analyze Wall 1 and Column 1 combined was the same as the one presented in Figure 6.1, except that Wall 2 was replaced by Column 1. The slabs were provided every 2743 mm , resulting in a 20 storey building. The thickness of the slabs was 203 mm .

### 6.4.2 Analytical model of wall system

Column 1 was first modeled and analyzed individually using program VecTor2. The model was very similar to the one created for Wall 1 and 2, described in Section 5.3. Low-powered rectangular and triangular elements were used to model the concrete, with smeared steel to account for the presence of reinforcement. The constitutive models for concrete and steel described in Chapter 3 were used in the analysis. Only one material type was used to represent the whole column in the finite element model, it was the same one used to represent the confined boundaries of Wall 1 and 2.

The analytical model was fixed at the bottom; both the horizontal and vertical displacements were restrained. The finite element mesh was more refined near the base (critical section). Column 1 had the same refinement as Wall 1 in the vertical direction up to a half of the height, so that the curvatures could be calculated over the same average length.

Meshes of $318 \times 423$ and $318 \times 203$ rectangular elements were used; three elements (four nodes) were used in the transverse direction of the wall. This level of refinement was maintained up to half the height of the wall. Up from this point, two elements (three nodes) were used in the transverse direction, and then it was further reduced to one element (two nodes) up to the top of the wall. The transitions were made using triangular elements. All nodes and elements were numbered in the horizontal (short) direction. The complete mesh consisted of 345 nodes, 242 rectangular elements and eight triangular elements.

The material properties used in the analysis were the same ones used for Wall 1 and 2, presented in Section 5.3.

A monotonic lateral load was applied at the top of the column. This load was applied in a displacement-control mode, in increments of 1 mm . Additionally, a constant axial load of 7742 kN was applied; this load was equally distributed in the two nodes at the top of the column. The self-weight of the column was not considered.

Figure 6.23 shows the finite element model of Column 1, created in the pre-processor FormWorks:

Figure 6.23 Finite element model of Column 1 in FormWorks


The material properties of the column were the same ones used for the material 1 in Wall 1 and 2 alone, presented in Table 5.1.

Wall 1 and Column 1 were then modeled and analyzed together using program VecTor2. The finite element mesh used for each of them, as well as the boundary conditions, material properties and loads, were the same as the ones described in Section 5.3 and this section.

As it was done for the model of Wall 1 and 2 combined together, described in Section 6.2.2, the slabs interconnecting the wall and the column at each floor level were modeled using truss bar elements with very stiff properties. The rectangular elements located at each storey level, which had a height equal to the slab thickness, were also provided with very stiff properties. The same material types were used for these elements.

All nodes and elements were numbered in the horizontal (short) direction of the wall system. The complete mesh consisted of 2426 nodes, 2169 rectangular elements, 94 triangular elements and 20 truss elements.

A monotonic lateral load was applied at the top of the Wall 1 . This load was applied in a displacement-control mode, in increments of 1 mm . Additionally, a constant axial load of 15484 kN in Wall 1 and 7742 kN in Column 1 was applied; these loads were equally distributed among the six nodes at the top of the wall and the two nodes at the top of the column. The self-weight of the wall and the column was not considered.

Figure 6.24 shows the finite element model of Wall 1 and Column 1, created in the pre-processor FormWorks:

Figure 6.24 Finite element model of Wall 1 and Column 1 in FormWorks


The detail of the elements representing the slabs at each storey level in the finite element model was the same as the one shown in Figure 6.3. The material properties in the different regions of the wall system were the same ones used for Wall 1 and 2 combined, presented in Table 6.1 and 6.2.

### 6.4.3 Analytical results

The results from the analysis were used to determine the curvatures, using the same procedure described in Section 5.4 and 5.6.

The ultimate drift for the wall system was $2 \%$. Figure 6.25 shows the curvature distribution up to the mid-height of the wall and the column for this drift, when they are alone and combined:

Figure 6.25 Curvatures up to the mid-height of Wall 1 and Column 1 alone and combined for a drift of 2\%


The same phenomenon observed for the previous wall systems, presented in Section 6.2.3 and 6.3.3, is seen here. The curvature distribution in Wall 1 remains approximately the same, while the curvatures in Column 1 at the base have increased considerably.

### 6.4.4 Discussion of analytical results

The curvature distribution in Figure 6.25 shows that the curvatures in Wall 1 remain almost the same when it is connected with Column 1, and its plastic hinge length has increased slightly. In Column 1, the curvatures have increased considerably at the base. This is the same phenomenon observed for Wall 1 and 2 and Wall 1 and 3 combined together. However, when Column 1 is alone, the curvatures at $2 \%$ drift do not reach the curvature at cracking. The column remains uncracked, so no plastic hinge is formed.

This is due to the high axial load applied to it. When combined with Wall 1, the curvatures in Column 1 increase, causing it to crack, but they are not large enough to make the column yield and develop a plastic hinge. This can also be observed from the distribution of tensile steel strains in the wall. The average vertical steel strains in the elements along the height of the walls obtained from program VecTor2 were studied for this. The steel strains were observed in the exterior steel layer. Figure 6.26 shows the steel strains up to the mid-height of Column 1, alone and combined, for a drift of $2 \%$ :

Figure 6.26 Steel strains up to the mid-height of Column 1 alone and combined for a drift of $\mathbf{2 \%}$


The steel strains are of compression when Column 1 is alone, but they are of tension when it is combined. The tensile strains in the combined case do not reach the yield strain.

In conclusion, as seen for the previous wall systems in Sections 6.2.4 and 6.3.4, the curvature distributions of Wall 1 and Column 1 when they are connected are different. Also, the curvature demand for the column is much larger when it is combined than when it is alone, and it has to be able to sustain that demand. It is important to consider this effect in the design of gravity columns in high-rise buildings.

## CHAPTER 7: CONCLUSIONS

The plastic hinge length is a function of several parameters. Past research has focused primarily on the influence of member depth, member span and longitudinal reinforcement properties. Some studies have also considered the influence of axial load ratio and strain hardening. Many researchers arrived to the conclusion that a good lower bound approximation for the plastic hinge length is $0.5 h$ or $0.5 l_{w}$, which is the value given in many concrete codes, including CSA A23.3.

Several empirical models have been developed to predict the length of the plastic hinge. These expressions provide very different results because they have been derived from tests for different types of concrete members and consider different parameters. Past research has focused mainly on beams and columns, not walls.

Program VecTor2 is a powerful analysis tool that provides good predictions of the response of reinforced concrete members. Analytical predictions of curvatures distributions and strain profiles obtained from VecTor2 have been verified with experimental results from tests performed on wall specimens.

The curvature distribution along the height of walls, determined experimentally and analytically, has been investigated. The results indicate that the distribution of inelastic curvatures can be well approximated as linearly varying.

The concept of curvature is based on the hypothesis that plane sections remain plane after bending, which is commonly assumed for slender concrete members. The strain profile along the length of walls has also been investigated to verify this assumption. The results indicate that before cracking, the strain profile remains linear. However, after cracking, the shape of the strain profile depends on the amount of reinforcement. As the amount of reinforcement increases, the strain profile is closer to have a linear variation. Also, the linearity of the strain profile degrades as the wall goes further into the nonlinear range.

The effect of the length of the wall in the plastic hinge length is more related to the magnitude of the steel strains than to the effect of diagonal cracking. Longer walls have larger plastic hinge lengths because they have larger steel strains than shorter walls. There is a bigger difference between the yield and ultimate moment for longer walls, which increases the slope of the post-yielding phase of the moment-curvature relationship.

For low shear stresses, a pure flexural prediction can be used to estimate the plastic hinge length. When the shear stresses are high, the actual plastic hinge length is longer due to the effect of diagonal cracking. The effect of shear has to be included to estimate the length of the plastic hinge for these cases. A simple shear model has been proposed, which provides reasonable results for high drifts.

The length of the plastic hinge reduces with the addition of axial compression, and increases with the addition of axial tension.

In a system of two walls of different lengths interconnected by rigid slabs at numerous floor levels, the curvature distributions along the height of the walls are different because of two reasons. First, the displacements between storey levels are different in both walls, specially at the lower levels, so the deflected shapes are not the strictly the same. Second, the shear displacements at the lower levels are larger in the longer wall; resulting in lower flexural displacements, lower slopes, and lower curvatures for this wall. Since the curvatures are not the same, the plastic hinge lengths are also different.

When two walls of different lengths are combined together, the curvature demand of the shorter wall is much larger than when it is alone for the same drift level. The curvature distribution in the longer wall remains the same. The shorter wall has to be designed to sustain that larger curvature demand. A simple model to predict the maximum curvature in the shorter wall has been proposed. This effect is of particular importance in the design of gravity columns in high-rise buildings, which may be subjected to tensile strains and also have to sustain larger curvature demands, even if they do not provide lateral resistance.

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## APPENDIX A: UBC WALL TEST

Table A. 1 List of zero readings

| Target |  | Reading 1 <br> $(\mathrm{mm})$ | Reading 2 <br> $(\mathrm{mm})$ | Reading 3 <br> $(\mathrm{mm})$ | Reading 4 <br> $(\mathrm{mm})$ | Reading 5 <br> $(\mathrm{mm})$ | Reading 6 <br> $(\mathrm{mm})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second | TW1 | TW2 | 288.37 | 289.53 | 289.71 | 289.55 |
| TW2 | TW3 | 194.74 | 194.08 | 193.95 | 194.01 | 193.57 | 289.63 |
| TW3 | TW4 | 521.09 | 521.15 | 521.12 | 521.04 | 521.00 | 521.03 |
| TW4 | TW5 | 521.06 | 521.08 | 521.08 | 521.04 | 521.04 | 521.04 |
| TW5 | TW6 | 520.93 | 520.91 | 520.77 | 520.70 | 520.65 | 520.61 |
| TW6 | TW7 | 521.51 | 521.33 | 521.42 | 521.28 | 521.00 | 521.23 |
| TW7 | TW8 | 520.11 | 521.20 | 520.10 | 520.03 | 521.30 | 519.83 |
| TW8 | TW9 | 520.40 | 520.82 | 520.42 | 520.34 | 520.05 | 520.28 |
| TW9 | TW10 | 520.62 | 520.52 | 520.56 | 520.49 | 520.48 | 520.44 |
| TW10 | TW11 | 520.52 | 520.50 | 520.44 | 521.74 | 521.00 | 521.20 |
| TW11 | TW12 | 520.59 | 520.40 | 520.42 | 520.78 | 521.00 | 521.82 |
| TE1 | TE2 | 291.37 | 294.36 | 294.78 | 294.73 | 294.80 | 294.82 |
| TE2 | TE3 | 191.04 | 191.61 | 191.24 | 191.98 | 190.92 | 191.12 |
| TE3 | TE4 | 521.10 | 521.10 | 521.28 | 521.30 | 521.09 | 521.00 |
| TE4 | TE5 | 521.06 | 520.89 | 521.07 | 520.90 | 520.95 | 520.89 |
| TE5 | TE6 | 521.25 | 521.07 | 521.26 | 521.02 | 521.04 | 520.97 |
| TE6 | TE7 | 521.06 | 521.12 | 521.08 | 521.06 | 521.09 | 521.03 |
| TE7 | TE8 | 521.03 | 521.46 | 521.16 | 520.99 | 520.95 | 521.08 |
| TE8 | TE9 | 521.06 | 521.07 | 521.34 | 520.98 | 521.05 | 521.26 |
| TE9 | TE10 | 521.33 | 521.16 | 521.14 | 521.02 | 521.11 | 521.16 |
| TE10 | TE11 | 521.16 | 521.24 | 521.13 | 521.05 | 521.10 | 521.22 |
| TE11 | TE12 | 521.33 | 521.28 | 521.15 | 521.25 | 521.41 | 521.07 |

Marked cells show measurements out of range, these were neglected.

Table A. 2 Mean values of zero readings

| Target |  | Reading 1 <br> $(\mathrm{mm})$ | Reading 2 <br> $(\mathrm{mm})$ | Reading 3 <br> $(\mathrm{mm})$ | Reading 4 <br> $(\mathrm{mm})$ | Reading 5 <br> $(\mathrm{mm})$ | Reading 6 <br> $(\mathrm{mm})$ | Mean value <br> $(\mathrm{mm})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |

Table A. 3 Experimental (pushing east) curvatures for a wall displacement of $\mathbf{1 0 5 m m}$

| Target |  | Reading <br> $(\mathrm{mm})$ | Zero reading <br> $(\mathrm{mm})$ | Cumulative <br> $(\mathrm{mm})$ | Height <br> $(\mathrm{mm})$ | Strain | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |

Wall length: $\qquad$ mm

Table A. 4 Experimental (pushing west) curvatures for a wall displacement of 104 mm

| Target |  | Reading <br> $(\mathrm{mm})$ | Zero reading <br> $(\mathrm{mm})$ | Cumulative <br> $(\mathrm{mm})$ | Height <br> $(\mathrm{mm})$ | Strain | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second | (W2 | TW2 | 288.76 | 289.60 | 289.60 | 146.07 |
| TW1 | TW | -0.0029 | 0.00433 |  |  |  |  |
| TW2 | TW3 | 193.67 | 193.94 | 483.54 | 388.43 | -0.0014 | 0.01050 |
| TW3 | TW4 | 520.59 | 521.07 | 1004.61 | 745.27 | -0.0009 | 0.00364 |
| TW4 | TW5 | 520.65 | 521.06 | 1525.67 | 1266.32 | -0.0008 | 0.00259 |
| TW5 | TW6 | 520.38 | 520.76 | 2046.43 | 1787.29 | -0.0007 | 0.00149 |
| TW6 | TW7 | 521.02 | 521.35 | 2567.78 | 2308.37 | -0.0006 | 0.00052 |
| TW7 | TW8 | 519.50 | 520.02 | 3087.80 | 2829.24 | -0.0010 | 0.00155 |
| TW8 | TW9 | 520.11 | 520.36 | 3608.16 | 3349.88 | -0.0005 | 0.00039 |
| TW9 | TW10 | 520.28 | 520.52 | 4128.68 | 3870.67 | -0.0005 | 0.00031 |
| TW10 | TW11 | 520.87 | 520.73 | 4649.41 | 4391.55 | 0.0003 | 0.00010 |
| TW11 | TW12 | 520.58 | 520.64 | 5170.05 | 4912.50 | -0.0001 | 0.00006 |
| Top of wall | - | - | - | 11760.00 | 0.0000 | 0.00000 |  |
| TE1 | TE2 | 295.92 | 294.70 | 294.70 | 146.07 | 0.0041 | 0.00433 |
| TE2 | TE3 | 194.18 | 191.19 | 485.88 | 388.43 | 0.0157 | 0.01050 |
| TE3 | TE4 | 523.75 | 521.15 | 1007.03 | 745.27 | 0.0050 | 0.00364 |
| TE4 | TE5 | 522.75 | 520.96 | 1527.99 | 1266.32 | 0.0034 | 0.00259 |
| TE5 | TE6 | 521.98 | 521.10 | 2049.09 | 1787.29 | 0.0017 | 0.00149 |
| TE6 | TE7 | 521.18 | 521.07 | 2570.16 | 2308.37 | 0.0002 | 0.00052 |
| TE7 | TE8 | 521.84 | 521.04 | 3091.21 | 2829.24 | 0.0015 | 0.00155 |
| TE8 | TE9 | 521.21 | 521.13 | 3612.33 | 3349.88 | 0.0002 | 0.00039 |
| TE9 | TE10 | 521.18 | 521.15 | 4133.49 | 3870.67 | 0.0001 | 0.00031 |
| TE10 | TE11 | 521.20 | 521.15 | 4654.64 | 4391.55 | 0.0001 | 0.00010 |
| TE11 | TE12 | 521.24 | 521.25 | 5175.88 | 4912.50 | 0.0000 | 0.00006 |
| Top of wall | - | - | - | 11760.00 | 0.0000 | 0.00000 |  |

Wall length: $\qquad$ mm

Table A. 5 Experimental (pushing east) curvatures for a wall displacement of $\mathbf{1 3 2 m m}$

| Target |  | Reading <br> $(\mathrm{mm})$ | Zero reading <br> $(\mathrm{mm})$ | Cumulative <br> $(\mathrm{mm})$ | Height <br> $(\mathrm{mm})$ | Strain | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |

Wall length: $\qquad$ mm

Table A. 6 Experimental (pushing west) curvatures for a wall displacement of 138 mm

| Target |  | Reading <br> $(\mathrm{mm})$ | Zero reading <br> $(\mathrm{mm})$ | Cumulative <br> $(\mathrm{mm})$ | Height <br> $(\mathrm{mm})$ | Strain | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second | (m) | TW2 | 288.95 | 289.60 | 289.60 | 146.07 |
| TW1 | -0.0022 | 0.00307 |  |  |  |  |  |
| TW2 | TW3 | 193.70 | 193.94 | 483.54 | 388.43 | -0.0012 | 0.01472 |
| TW3 | TW4 | 520.47 | 521.07 | 1004.61 | 745.27 | -0.0012 | 0.00557 |
| TW4 | TW5 | 520.74 | 521.06 | 1525.67 | 1266.32 | -0.0006 | 0.00375 |
| TW5 | TW6 | 520.35 | 520.76 | 2046.43 | 1787.29 | -0.0008 | 0.00191 |
| TW6 | TW7 | 520.98 | 521.35 | 2567.78 | 2308.37 | -0.0007 | 0.00048 |
| TW7 | TW8 | 519.61 | 520.02 | 3087.80 | 2829.24 | -0.0008 | 0.00172 |
| TW8 | TW9 | 520.12 | 520.36 | 3608.16 | 3349.88 | -0.0005 | 0.00055 |
| TW9 | TW10 | 520.44 | 520.52 | 4128.68 | 3870.67 | -0.0002 | 0.00012 |
| TW10 | TW11 | 521.11 | 520.73 | 4649.41 | 4391.55 | 0.0007 | 0.00041 |
| TW11 | TW12 | 520.65 | 520.64 | 5170.05 | 4912.50 | 0.0000 | 0.00013 |
| Top of wall | - | - | - | 11760.00 | 0.0000 | 0.00000 |  |
| TE1 | TE2 | 295.51 | 294.70 | 294.70 | 146.07 | 0.0028 | 0.00307 |
| TE2 | TE3 | 195.52 | 191.19 | 485.88 | 388.43 | 0.0227 | 0.01472 |
| TE3 | TE4 | 525.26 | 521.15 | 1007.03 | 745.27 | 0.0079 | 0.00557 |
| TE4 | TE5 | 523.82 | 520.96 | 1527.99 | 1266.32 | 0.0055 | 0.00375 |
| TE5 | TE6 | 522.31 | 521.10 | 2049.09 | 1787.29 | 0.0023 | 0.00191 |
| TE6 | TE7 | 521.11 | 521.07 | 2570.16 | 2308.37 | 0.0001 | 0.00048 |
| TE7 | TE8 | 522.09 | 521.04 | 3091.21 | 2829.24 | 0.0020 | 0.00172 |
| TE8 | TE9 | 521.35 | 521.13 | 3612.33 | 3349.88 | 0.0004 | 0.00055 |
| TE9 | TE10 | 521.18 | 521.15 | 4133.49 | 3870.67 | 0.0001 | 0.00012 |
| TE10 | TE11 | 521.18 | 521.15 | 4654.64 | 4391.55 | 0.0001 | 0.00041 |
| TE11 | TE12 | 521.15 | 521.25 | 5175.88 | 4912.50 | -0.0002 | 0.00013 |
| T0p of wall | - | - | - | 11760.00 | 0.0000 | 0.00000 |  |

Wall length: $\qquad$ mm

Table A. 7 Experimental (pushing east) curvatures for a wall displacement of $\mathbf{1 8 2 m m}$

| Target |  | Reading <br> $(\mathrm{mm})$ | Zero reading <br> $(\mathrm{mm})$ | Cumulative <br> $(\mathrm{mm})$ | Height <br> $(\mathrm{mm})$ | Strain | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second | ( 289.65 | 289.60 | 289.60 | 146.07 | 0.0002 | 0.00151 |
| TW1 | TW2 | 289.65 | 483.54 | 388.43 | 0.0209 | 0.01402 |  |
| TW2 | TW3 | 198.00 | 193.94 | 521.07 | 1004.61 | 745.27 | 0.0148 |
| TW3 | TW4 | 528.79 | 0.01021 |  |  |  |  |
| TW4 | TW5 | 525.59 | 521.06 | 1525.67 | 1266.32 | 0.0087 | 0.00600 |
| TW5 | TW6 | 523.16 | 520.76 | 2046.43 | 1787.29 | 0.0046 | 0.00340 |
| TW6 | TW7 | 522.12 | 521.35 | 2567.78 | 2308.37 | 0.0015 | 0.00132 |
| TW7 | TW8 | 520.98 | 520.02 | 3087.80 | 2829.24 | 0.0019 | 0.00157 |
| TW8 | TW9 | 520.45 | 520.36 | 3608.16 | 3349.88 | 0.0002 | 0.00053 |
| TW9 | TW10 | 521.09 | 520.52 | 4128.68 | 3870.67 | 0.0011 | 0.00108 |
| TW10 | TW11 | 521.15 | 520.73 | 4649.41 | 4391.55 | 0.0008 | 0.00080 |
| TW11 | TW12 | 520.87 | 520.64 | 5170.05 | 4912.50 | 0.0004 | 0.00071 |
| Top of wall | - | - | - | 11760.00 | 0.0000 | 0.00000 |  |
| TE1 | TE2 | 294.03 | 294.70 | 294.70 | 146.07 | -0.0023 | 0.00151 |
| TE2 | TE3 | 190.83 | 191.19 | 485.88 | 388.43 | -0.0019 | 0.01402 |
| TE3 | TE4 | 520.22 | 521.15 | 1007.03 | 745.27 | -0.0018 | 0.01021 |
| TE4 | TE5 | 520.41 | 520.96 | 1527.99 | 1266.32 | -0.0011 | 0.00600 |
| TE5 | TE6 | 520.62 | 521.10 | 2049.09 | 1787.29 | -0.0009 | 0.00340 |
| TE6 | TE7 | 520.72 | 521.07 | 2570.16 | 2308.37 | -0.0007 | 0.00132 |
| TE7 | TE8 | 520.68 | 521.04 | 3091.21 | 2829.24 | -0.0007 | 0.00157 |
| TE8 | TE9 | 520.77 | 521.13 | 3612.33 | 3349.88 | -0.0007 | 0.00053 |
| TE9 | TE10 | 520.81 | 521.15 | 4133.49 | 3870.67 | -0.0007 | 0.00108 |
| TE10 | TE11 | 520.89 | 521.15 | 4654.64 | 4391.55 | -0.0005 | 0.00080 |
| TE11 | TE12 | 520.88 | 521.25 | 5175.88 | 4912.50 | -0.0007 | 0.00071 |
| T0p of wall | - | - | - | 11760.00 | 0.0000 | 0.00000 |  |

Wall length: $\qquad$ mm

Table A. 8 Experimental (pushing west) curvatures for a wall displacement of 187 mm

| Target |  | Reading <br> $(\mathrm{mm})$ | Zero reading <br> $(\mathrm{mm})$ | Cumulative <br> $(\mathrm{mm})$ | Height <br> $(\mathrm{mm})$ | Strain | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second | (WW1 | TW2 | 288.86 | 289.60 | 289.60 | 146.07 |
| TW2 | TW3 | 193.53 | 193.94 | 483.54 | 388.43 | -0.0025 | 0.00341 |
| TW21 | 0.02253 |  |  |  |  |  |  |
| TW3 | TW4 | 520.33 | 521.07 | 1004.61 | 745.27 | -0.0014 | 0.00808 |
| TW4 | TW5 | 520.52 | 521.06 | 1525.67 | 1266.32 | -0.0010 | 0.00577 |
| TW5 | TW6 | 520.29 | 520.76 | 2046.43 | 1787.29 | -0.0009 | 0.00278 |
| TW6 | TW7 | 520.95 | 521.35 | 2567.78 | 2308.37 | -0.0008 | 0.00125 |
| TW7 | TW8 | 519.67 | 520.02 | 3087.80 | 2829.24 | -0.0007 | 0.00159 |
| TW8 | TW9 | 520.07 | 520.36 | 3608.16 | 3349.88 | -0.0006 | 0.00061 |
| TW9 | TW10 | 520.28 | 520.52 | 4128.68 | 3870.67 | -0.0005 | 0.00034 |
| TW10 | TW11 | 520.86 | 520.73 | 4649.41 | 4391.55 | 0.0002 | 0.00008 |
| TW11 | TW12 | 520.96 | 520.64 | 5170.05 | 4912.50 | 0.0006 | 0.00034 |
| Top of wall | - | - | - | 11760.00 | 0.0000 | 0.00000 |  |
| TE1 | TE2 | 295.58 | 294.70 | 294.70 | 146.07 | 0.0030 | 0.00341 |
| TE2 | TE3 | 197.78 | 191.19 | 485.88 | 388.43 | 0.0345 | 0.02253 |
| TE3 | TE4 | 527.25 | 521.15 | 1007.03 | 745.27 | 0.0117 | 0.00808 |
| TE4 | TE5 | 525.31 | 520.96 | 1527.99 | 1266.32 | 0.0083 | 0.00577 |
| TE5 | TE6 | 522.98 | 521.10 | 2049.09 | 1787.29 | 0.0036 | 0.00278 |
| TE6 | TE7 | 521.73 | 521.07 | 2570.16 | 2308.37 | 0.0013 | 0.00125 |
| TE7 | TE8 | 522.04 | 521.04 | 3091.21 | 2829.24 | 0.0019 | 0.00159 |
| TE8 | TE9 | 521.35 | 521.13 | 3612.33 | 3349.88 | 0.0004 | 0.00061 |
| TE9 | TE10 | 521.20 | 521.15 | 4133.49 | 3870.67 | 0.0001 | 0.00034 |
| TE10 | TE11 | 521.21 | 521.15 | 4654.64 | 4391.55 | 0.0001 | 0.00008 |
| TE11 | TE12 | 521.28 | 521.25 | 5175.88 | 4912.50 | 0.0001 | 0.00034 |
| Top of wall | - | - | - | 11760.00 | 0.0000 | 0.00000 |  |

Wall length: $\qquad$ mm

Table A. 9 Analytical curvatures for a wall displacement of 105 mm

| Node |  | Node 1 ydispl (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | Difference (mm) | Height (mm) | Strain | Curvature $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1 | 22 | 0.00 | 0.78 | 58.71 | 58.71 | 29.36 | 0.0134 | 0.00985 |
| 22 | 148 | 0.78 | 6.87 | 579.63 | 520.92 | 319.17 | 0.0117 | 0.00822 |
| 148 | 274 | 6.87 | 9.51 | 1100.55 | 520.92 | 840.09 | 0.0051 | 0.00375 |
| 274 | 400 | 9.51 | 10.27 | 1621.47 | 520.92 | 1361.01 | 0.0015 | 0.00139 |
| 400 | 526 | 10.27 | 10.90 | 2142.39 | 520.92 | 1881.93 | 0.0012 | 0.00118 |
| 526 | 652 | 10.90 | 11.35 | 2663.31 | 520.92 | 2402.85 | 0.0009 | 0.00091 |
| 652 | 778 | 11.35 | 11.62 | 3184.23 | 520.92 | 2923.77 | 0.0005 | 0.00066 |
| 778 | 904 | 11.62 | 11.75 | 3705.15 | 520.92 | 3444.69 | 0.0002 | 0.00046 |
| 904 | 1030 | 11.75 | 11.86 | 4226.07 | 520.92 | 3965.61 | 0.0002 | 0.00041 |
| 1030 | 1156 | 11.86 | 11.91 | 4746.99 | 520.92 | 4486.53 | 0.0001 | 0.00033 |
| Top | of wall | - | - | - | - | 11330.00 | 0.0000 | 0.00000 |
| 21 | 42 | 0.00 | -0.16 | 58.71 | 58.71 | 29.36 | -0.0027 | 0.00985 |
| 42 | 168 | -0.16 | -1.03 | 579.63 | 520.92 | 319.17 | -0.0017 | 0.00822 |
| 168 | 294 | -1.03 | -1.57 | 1100.55 | 520.92 | 840.09 | -0.0010 | 0.00375 |
| 294 | 420 | -1.57 | -1.98 | 1621.47 | 520.92 | 1361.01 | -0.0008 | 0.00139 |
| 420 | 546 | -1.98 | -2.34 | 2142.39 | 520.92 | 1881.93 | -0.0007 | 0.00118 |
| 546 | 672 | -2.34 | -2.67 | 2663.31 | 520.92 | 2402.85 | -0.0006 | 0.00091 |
| 672 | 798 | -2.67 | -2.96 | 3184.23 | 520.92 | 2923.77 | -0.0006 | 0.00066 |
| 798 | 924 | -2.96 | -3.22 | 3705.15 | 520.92 | 3444.69 | -0.0005 | 0.00046 |
| 924 | 1050 | -3.22 | -3.46 | 4226.07 | 520.92 | 3965.61 | -0.0005 | 0.00041 |
| 1050 | 1176 | -3.46 | -3.68 | 4746.99 | 520.92 | 4486.53 | -0.0004 | 0.00033 |
| Top of wall |  | - | - | - | - | 11330.00 | 0.0000 | 0.00000 |

Wall length: $\qquad$ mm

Table A.10 Analytical curvatures for a wall displacement of 104 mm

| Node |  | Node 1 ydispl (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | Difference (mm) | Height (mm) | Strain | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1 | 22 | 0.00 | 0.78 | 58.71 | 58.71 | 29.36 | 0.0133 | 0.00981 |
| 22 | 148 | 0.78 | 6.82 | 579.63 | 520.92 | 319.17 | 0.0116 | 0.00816 |
| 148 | 274 | 6.82 | 9.37 | 1100.55 | 520.92 | 840.09 | 0.0049 | 0.00365 |
| 274 | 400 | 9.37 | 10.13 | 1621.47 | 520.92 | 1361.01 | 0.0015 | 0.00139 |
| 400 | 526 | 10.13 | 10.77 | 2142.39 | 520.92 | 1881.93 | 0.0012 | 0.00118 |
| 526 | 652 | 10.77 | 11.21 | 2663.31 | 520.92 | 2402.85 | 0.0009 | 0.00091 |
| 652 | 778 | 11.21 | 11.49 | 3184.23 | 520.92 | 2923.77 | 0.0005 | 0.00067 |
| 778 | 904 | 11.49 | 11.61 | 3705.15 | 520.92 | 3444.69 | 0.0002 | 0.00046 |
| 904 | 1030 | 11.61 | 11.72 | 4226.07 | 520.92 | 3965.61 | 0.0002 | 0.00041 |
| 1030 | 1156 | 11.72 | 11.77 | 4746.99 | 520.92 | 4486.53 | 0.0001 | 0.00033 |
| Top | of wall | - | - | - | - | 11330.00 | 0.0000 | 0.00000 |
| 21 | 42 | 0.00 | -0.16 | 58.71 | 58.71 | 29.36 | -0.0026 | 0.00981 |
| 42 | 168 | -0.16 | -1.02 | 579.63 | 520.92 | 319.17 | -0.0017 | 0.00816 |
| 168 | 294 | -1.02 | -1.56 | 1100.55 | 520.92 | 840.09 | -0.0010 | 0.00365 |
| 294 | 420 | -1.56 | -1.97 | 1621.47 | 520.92 | 1361.01 | -0.0008 | 0.00139 |
| 420 | 546 | -1.97 | -2.33 | 2142.39 | 520.92 | 1881.93 | -0.0007 | 0.00118 |
| 546 | 672 | -2.33 | -2.66 | 2663.31 | 520.92 | 2402.85 | -0.0006 | 0.00091 |
| 672 | 798 | -2.66 | -2.95 | 3184.23 | 520.92 | 2923.77 | -0.0006 | 0.00067 |
| 798 | 924 | -2.95 | -3.21 | 3705.15 | 520.92 | 3444.69 | -0.0005 | 0.00046 |
| 924 | 1050 | -3.21 | -3.45 | 4226.07 | 520.92 | 3965.61 | -0.0005 | 0.00041 |
| 1050 | 1176 | -3.45 | -3.67 | 4746.99 | 520.92 | 4486.53 | -0.0004 | 0.00033 |
| Top of wall |  | - | - | - | - | 11330.00 | 0.0000 | 0.00000 |

Wall length: $\qquad$ mm

Table A. 11 Analytical curvatures for a wall displacement of $\mathbf{1 3 2 m m}$

| Node |  | Node 1 ydispl (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | Difference (mm) | Height (mm) | Strain | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1 | 22 | 0.00 | 0.93 | 58.71 | 58.71 | 29.36 | 0.0158 | 0.01171 |
| 22 | 148 | 0.93 | 8.10 | 579.63 | 520.92 | 319.17 | 0.0138 | 0.00967 |
| 148 | 274 | 8.10 | 12.41 | 1100.55 | 520.92 | 840.09 | 0.0083 | 0.00580 |
| 274 | 400 | 12.41 | 13.80 | 1621.47 | 520.92 | 1361.01 | 0.0027 | 0.00217 |
| 400 | 526 | 13.80 | 14.47 | 2142.39 | 520.92 | 1881.93 | 0.0013 | 0.00123 |
| 526 | 652 | 14.47 | 15.01 | 2663.31 | 520.92 | 2402.85 | 0.0010 | 0.00103 |
| 652 | 778 | 15.01 | 15.28 | 3184.23 | 520.92 | 2923.77 | 0.0005 | 0.00067 |
| 778 | 904 | 15.28 | 15.47 | 3705.15 | 520.92 | 3444.69 | 0.0004 | 0.00054 |
| 904 | 1030 | 15.47 | 15.58 | 4226.07 | 520.92 | 3965.61 | 0.0002 | 0.00042 |
| 1030 | 1156 | 15.58 | 15.63 | 4746.99 | 520.92 | 4486.53 | 0.0001 | 0.00033 |
| Top | of wall | - | - | - | - | 11330.00 | 0.0000 | 0.00000 |
| 21 | 42 | 0.00 | -0.19 | 58.71 | 58.71 | 29.36 | -0.0032 | 0.01171 |
| 42 | 168 | -0.19 | -1.20 | 579.63 | 520.92 | 319.17 | -0.0020 | 0.00967 |
| 168 | 294 | -1.20 | -1.81 | 1100.55 | 520.92 | 840.09 | -0.0012 | 0.00580 |
| 294 | 420 | -1.81 | -2.25 | 1621.47 | 520.92 | 1361.01 | -0.0009 | 0.00217 |
| 420 | 546 | -2.25 | -2.63 | 2142.39 | 520.92 | 1881.93 | -0.0007 | 0.00123 |
| 546 | 672 | -2.63 | -2.96 | 2663.31 | 520.92 | 2402.85 | -0.0006 | 0.00103 |
| 672 | 798 | -2.96 | -3.26 | 3184.23 | 520.92 | 2923.77 | -0.0006 | 0.00067 |
| 798 | 924 | -3.26 | -3.53 | 3705.15 | 520.92 | 3444.69 | -0.0005 | 0.00054 |
| 924 | 1050 | -3.53 | -3.77 | 4226.07 | 520.92 | 3965.61 | -0.0005 | 0.00042 |
| 1050 | 1176 | -3.77 | -4.00 | 4746.99 | 520.92 | 4486.53 | -0.0004 | 0.00033 |
| Top of wall |  | - | - | - | - | 11330.00 | 0.0000 | 0.00000 |

Wall length: $\qquad$ mm

Table A. 12 Analytical curvatures for a wall displacement of 138 mm

| Node |  | Node 1 ydispl (mm) | Node 2 y displ (mm) | Node 2 ycoord (mm) | Difference (mm) | Height (mm) | Strain | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1 | 22 | 0.00 | 0.97 | 58.71 | 58.71 | 29.36 | 0.0165 | 0.01221 |
| 22 | 148 | 0.97 | 8.56 | 579.63 | 520.92 | 319.17 | 0.0146 | 0.01021 |
| 148 | 274 | 8.56 | 13.17 | 1100.55 | 520.92 | 840.09 | 0.0089 | 0.00617 |
| 274 | 400 | 13.17 | 14.62 | 1621.47 | 520.92 | 1361.01 | 0.0028 | 0.00225 |
| 400 | 526 | 14.62 | 15.29 | 2142.39 | 520.92 | 1881.93 | 0.0013 | 0.00124 |
| 526 | 652 | 15.29 | 15.84 | 2663.31 | 520.92 | 2402.85 | 0.0010 | 0.00104 |
| 652 | 778 | 15.84 | 16.11 | 3184.23 | 520.92 | 2923.77 | 0.0005 | 0.00067 |
| 778 | 904 | 16.11 | 16.29 | 3705.15 | 520.92 | 3444.69 | 0.0004 | 0.00054 |
| 904 | 1030 | 16.29 | 16.41 | 4226.07 | 520.92 | 3965.61 | 0.0002 | 0.00042 |
| 1030 | 1156 | 16.41 | 16.46 | 4746.99 | 520.92 | 4486.53 | 0.0001 | 0.00033 |
| Top | of wall | - | - | - | - | 11330.00 | 0.0000 | 0.00000 |
| 21 | 42 | 0.00 | -0.20 | 58.71 | 58.71 | 29.36 | -0.0034 | 0.01221 |
| 42 | 168 | -0.20 | -1.25 | 579.63 | 520.92 | 319.17 | -0.0020 | 0.01021 |
| 168 | 294 | -1.25 | -1.86 | 1100.55 | 520.92 | 840.09 | -0.0012 | 0.00617 |
| 294 | 420 | -1.86 | -2.30 | 1621.47 | 520.92 | 1361.01 | -0.0009 | 0.00225 |
| 420 | 546 | -2.30 | -2.68 | 2142.39 | 520.92 | 1881.93 | -0.0007 | 0.00124 |
| 546 | 672 | -2.68 | -3.02 | 2663.31 | 520.92 | 2402.85 | -0.0006 | 0.00104 |
| 672 | 798 | -3.02 | -3.32 | 3184.23 | 520.92 | 2923.77 | -0.0006 | 0.00067 |
| 798 | 924 | -3.32 | -3.58 | 3705.15 | 520.92 | 3444.69 | -0.0005 | 0.00054 |
| 924 | 1050 | -3.58 | -3.83 | 4226.07 | 520.92 | 3965.61 | -0.0005 | 0.00042 |
| 1050 | 1176 | -3.83 | -4.06 | 4746.99 | 520.92 | 4486.53 | -0.0004 | 0.00033 |
| Top of wall |  | - | - | - | - | 11330.00 | 0.0000 | 0.00000 |

Wall length: $\qquad$ mm

Table A. 13 Analytical curvatures for a wall displacement of 182 mm

| Node |  | Node 1 ydispl (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | $\begin{aligned} & \text { Difference } \\ & (\mathrm{mm}) \end{aligned}$ | Height (mm) | Strain | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1 | 22 | 0.00 | 1.14 | 58.71 | 58.71 | 29.36 | 0.0193 | 0.01458 |
| 22 | 148 | 1.14 | 10.31 | 579.63 | 520.92 | 319.17 | 0.0176 | 0.01235 |
| 148 | 274 | 10.31 | 17.78 | 1100.55 | 520.92 | 840.09 | 0.0143 | 0.00965 |
| 274 | 400 | 17.78 | 20.51 | 1621.47 | 520.92 | 1361.01 | 0.0052 | 0.00380 |
| 400 | 526 | 20.51 | 21.23 | 2142.39 | 520.92 | 1881.93 | 0.0014 | 0.00132 |
| 526 | 652 | 21.23 | 21.82 | 2663.31 | 520.92 | 2402.85 | 0.0011 | 0.00111 |
| 652 | 778 | 21.82 | 22.18 | 3184.23 | 520.92 | 2923.77 | 0.0007 | 0.00078 |
| 778 | 904 | 22.18 | 22.37 | 3705.15 | 520.92 | 3444.69 | 0.0004 | 0.00056 |
| 904 | 1030 | 22.37 | 22.49 | 4226.07 | 520.92 | 3965.61 | 0.0002 | 0.00044 |
| 1030 | 1156 | 22.49 | 22.55 | 4746.99 | 520.92 | 4486.53 | 0.0001 | 0.00034 |
| Top | wall | - | - | - | - | 11330.00 | 0.0000 | 0.00000 |
| 21 | 42 | 0.00 | -0.26 | 58.71 | 58.71 | 29.36 | -0.0044 | 0.01458 |
| 42 | 168 | -0.26 | -1.54 | 579.63 | 520.92 | 319.17 | -0.0025 | 0.01235 |
| 168 | 294 | -1.54 | -2.24 | 1100.55 | 520.92 | 840.09 | -0.0013 | 0.00965 |
| 294 | 420 | -2.24 | -2.72 | 1621.47 | 520.92 | 1361.01 | -0.0009 | 0.00380 |
| 420 | 546 | -2.72 | -3.12 | 2142.39 | 520.92 | 1881.93 | -0.0008 | 0.00132 |
| 546 | 672 | -3.12 | -3.46 | 2663.31 | 520.92 | 2402.85 | -0.0007 | 0.00111 |
| 672 | 798 | -3.46 | -3.77 | 3184.23 | 520.92 | 2923.77 | -0.0006 | 0.00078 |
| 798 | 924 | -3.77 | -4.05 | 3705.15 | 520.92 | 3444.69 | -0.0005 | 0.00056 |
| 924 | 1050 | -4.05 | -4.30 | 4226.07 | 520.92 | 3965.61 | -0.0005 | 0.00044 |
| 1050 | 1176 | -4.30 | -4.53 | 4746.99 | 520.92 | 4486.53 | -0.0004 | 0.00034 |
| Top of wall |  | - | - | - | - | 11330.00 | 0.0000 | 0.00000 |

Wall length: $\qquad$ mm

Table A. 14 Analytical curvatures for a wall displacement of 187 mm

| Node |  | Node 1 y displ (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | Difference (mm) | Height (mm) | Strain | Curvature (rad/m) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1 | 22 | 0.00 | 1.16 | 58.71 | 58.71 | 29.36 | 0.0197 | 0.01487 |
| 22 | 148 | 1.16 | 10.52 | 579.63 | 520.92 | 319.17 | 0.0180 | 0.01260 |
| 148 | 274 | 10.52 | 18.15 | 1100.55 | 520.92 | 840.09 | 0.0147 | 0.00986 |
| 274 | 400 | 18.15 | 21.17 | 1621.47 | 520.92 | 1361.01 | 0.0058 | 0.00414 |
| 400 | 526 | 21.17 | 21.91 | 2142.39 | 520.92 | 1881.93 | 0.0014 | 0.00134 |
| 526 | 652 | 21.91 | 22.50 | 2663.31 | 520.92 | 2402.85 | 0.0011 | 0.00112 |
| 652 | 778 | 22.50 | 22.86 | 3184.23 | 520.92 | 2923.77 | 0.0007 | 0.00079 |
| 778 | 904 | 22.86 | 23.06 | 3705.15 | 520.92 | 3444.69 | 0.0004 | 0.00056 |
| 904 | 1030 | 23.06 | 23.18 | 4226.07 | 520.92 | 3965.61 | 0.0002 | 0.00044 |
| 1030 | 1156 | 23.18 | 23.24 | 4746.99 | 520.92 | 4486.53 | 0.0001 | 0.00035 |
| Top | wall | - | - | - | - | 11330.00 | 0.0000 | 0.00000 |
| 21 | 42 | 0.00 | -0.26 | 58.71 | 58.71 | 29.36 | -0.0045 | 0.01487 |
| 42 | 168 | -0.26 | -1.57 | 579.63 | 520.92 | 319.17 | -0.0025 | 0.01260 |
| 168 | 294 | -1.57 | -2.28 | 1100.55 | 520.92 | 840.09 | -0.0014 | 0.00986 |
| 294 | 420 | -2.28 | -2.77 | 1621.47 | 520.92 | 1361.01 | -0.0009 | 0.00414 |
| 420 | 546 | -2.77 | -3.17 | 2142.39 | 520.92 | 1881.93 | -0.0008 | 0.00134 |
| 546 | 672 | -3.17 | -3.51 | 2663.31 | 520.92 | 2402.85 | -0.0007 | 0.00112 |
| 672 | 798 | -3.51 | -3.82 | 3184.23 | 520.92 | 2923.77 | -0.0006 | 0.00079 |
| 798 | 924 | -3.82 | -4.10 | 3705.15 | 520.92 | 3444.69 | -0.0005 | 0.00056 |
| 924 | 1050 | -4.10 | -4.35 | 4226.07 | 520.92 | 3965.61 | -0.0005 | 0.00044 |
| 1050 | 1176 | -4.35 | -4.58 | 4746.99 | 520.92 | 4486.53 | -0.0004 | 0.00035 |
| Top of wall |  | - | - | - | - | 11330.00 | 0.0000 | 0.00000 |

Wall length: $\qquad$ mm

Table A. 15 Plane sections analysis to determine ultimate curvature


| Concrete properties |  | Wall dimensions |  |
| :---: | :---: | :---: | :---: |
| $f_{c}^{\prime}(\mathrm{MPa})$ | 49.0 | $l_{f 1}(\mathrm{~mm})$ | 203.0 |
| $\varepsilon_{c u}$ | 0.0035 | $t_{f 1}(\mathrm{~mm})$ | 380.0 |
| $\phi_{c}$ | 1.00 | $l_{f 2}(\mathrm{~mm})$ | 203.0 |
| Reinforcement <br> properties |  | $t_{f 2}(\mathrm{~mm})$ | 380.0 |
| $\left(l_{w}(\mathrm{~mm})\right.$ | 1219.0 |  |  |
| $f_{y}(\mathrm{MPa})$ | 455.0 | $t_{w}(\mathrm{~mm})$ |  |
| $E_{s}(\mathrm{MPa})$ | 200000.0 | Axial load |  |
| $\phi_{s}$ | 1.00 | $P(\mathrm{kN})$ |  |


| Stress block factors |  | Yield strain |  | Section centroid |  | Forces in concrete |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\alpha_{l}$ | 0.777 | $\varepsilon_{y}$ | 0.0023 | $x(\mathrm{~mm})$ | 812.50 | $c(\mathrm{~mm})$ | 148.23 |
| $\beta_{1}$ | 0.848 |  |  |  |  | $a(\mathrm{~mm})$ | 125.62 |
|  |  |  |  |  |  | $P_{c}(\mathrm{kN})$ | -1816.31 |
|  |  |  |  |  |  | $M_{c}$ (kN.m) | 1361.67 |


| Forces in reinforcement |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Layer | $x(\mathrm{~mm})$ | $A_{s}\left(\mathrm{~mm}^{2}\right)$ | $\varepsilon_{s}$ | $f_{s}(\mathrm{MPa})$ | $P_{s}(\mathrm{kN})$ | $y(\mathrm{~mm})$ | $M_{s}(\mathrm{kN} . \mathrm{m})$ |
| 1 | 21.00 | 300.0 | -0.0030 | -455.00 | -125.09 | -791.50 | 99.01 |
| 2 | 182.00 | 200.0 | 0.0008 | 159.49 | 31.90 | -630.50 | -20.11 |
| 3 | 355.00 | 100.0 | 0.0049 | 455.00 | 45.50 | -457.50 | -20.82 |
| 4 | 660.00 | 100.0 | 0.0121 | 455.00 | 45.50 | -152.50 | -6.94 |
| 5 | 965.00 | 100.0 | 0.0193 | 455.00 | 45.50 | 152.50 | 6.94 |
| 6 | 1270.00 | 100.0 | 0.0265 | 455.00 | 45.50 | 457.50 | 20.82 |
| 7 | 1443.00 | 200.0 | 0.0306 | 455.00 | 91.00 | 630.50 | 57.38 |
| 8 | 1604.00 | 300.0 | 0.0344 | 455.00 | 136.50 | 791.50 | 108.04 |


| Results |  |
| :---: | :---: |
| $P_{r}(\mathrm{kN})$ | -1500.00 |
| $M_{r}(\mathrm{kN} . \mathrm{m})$ | 1605.98 |
| $\phi_{u}(\mathrm{rad} / \mathrm{m})$ | 0.0236 |

APPENDIX B: CLARKSON UNIVERSITY WALL TEST

Table B.1 Analytical strain profile at base for a drift of $\mathbf{1 . 5 \%}$

| Node |  | Node 1 y- <br> displ (mm) | Node 2 y- <br> displ (mm) | Distance along <br> wall length (mm) | Distance between <br> nodes (mm) | Strain <br> $(\mathrm{mm} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second | 0.00 | 5.71 | 0.00 | 228.60 | 24.96 |
| 1 | 27 | 0.00 | 5.07 | 177.80 | 228.60 | 22.20 |
| 3 | 29 | 0.00 | 4.92 | 381.00 | 228.60 | 21.52 |
| 5 | 31 | 0.00 | 3.26 | 609.60 | 228.60 | 14.27 |
| 7 | 33 | 0.00 | 1.59 | 838.20 | 228.60 | 6.93 |
| 9 | 35 | 0.00 | 0.27 | 1041.40 | 228.60 | 1.18 |
| 11 | 37 | 0.00 | -1.48 | 1219.20 | 228.60 | -6.47 |
| 13 | 39 |  |  |  |  |  |

Table B. 2 Analytical strain profile at base for a drift of $\mathbf{2 \%}$

| Node |  | Node 1 y- <br> displ (mm) | Node 2 y- <br> displ (mm) | Distance along <br> wall length (mm) | Distance between <br> nodes (mm) | Strain <br> $(\mathrm{mm} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second | 0.00 | 6.56 | 0.00 | 228.60 | 28.67 |
| 1 | 27 | 0.00 | 177.80 | 228.60 | 24.49 |  |
| 3 | 29 | 0.00 | 5.60 | 381.00 | 228.60 | 22.48 |
| 5 | 31 | 0.00 | 5.14 | 609.60 | 228.60 | 15.93 |
| 7 | 33 | 0.00 | 3.64 | 838.20 | 228.60 | 10.45 |
| 9 | 35 | 0.00 | 2.39 | 1041.40 | 228.60 | 1.69 |
| 11 | 37 | 0.00 | 0.39 | 1219.20 | 228.60 | -9.41 |
| 13 | 39 | 0.00 | -2.15 |  |  |  |

Table B. 3 Analytical strain profile accounting for cover spalling at base for a drift of $\mathbf{1 . 5 \%}$

| Node |  | Node l y- <br> displ (mm) | Node 2 y- <br> displ (mm) | Distance along <br> wall length (mm) | Distance between <br> nodes (mm) | Strain <br> $(\mathrm{mm} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second | 0.00 | 5.49 | 0.00 | 228.60 | 24.02 |
| 1 | 27 | 0.00 | 4.52 | 177.80 | 228.60 | 19.77 |
| 3 | 29 | 0.00 | 3.18 | 381.00 | 228.60 | 13.89 |
| 5 | 31 | 0.00 | 3.03 | 609.60 | 228.60 | 13.23 |
| 7 | 33 | 0.00 | 1.48 | 838.20 | 228.60 | 6.48 |
| 9 | 35 | 0.00 | -0.23 | 1041.40 | 228.60 | -0.99 |
| 11 | 37 | 0.00 | -1.76 | 1219.20 | 228.60 | -7.69 |
| 13 | 39 |  |  |  |  |  |

Table B. 4 Analytical strain profile accounting for cover spalling at base for a drift of $\mathbf{2 \%}$

| Node |  | Node 1 y- <br> displ (mm) | Node 2 y- <br> displ (mm) | Distance along <br> wall length (mm) | Distance between <br> nodes (mm) | Strain <br> $(\mathrm{mm} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second | 0.00 | 6.40 | 0.00 | 228.60 | 27.98 |
| 1 | 27 | 0.00 | 5.18 | 177.80 | 228.60 | 22.66 |
| 3 | 29 | 0.00 | 3.41 | 381.00 | 228.60 | 14.90 |
| 5 | 31 | 0.00 | 3.47 | 609.60 | 228.60 | 15.18 |
| 7 | 33 | 0.00 | 2.26 | 838.20 | 228.60 | 9.88 |
| 9 | 35 | 0.00 | -0.35 | 1041.40 | 228.60 | -1.54 |
| 11 | 37 | 0.00 | -2.33 | 1219.20 | 228.60 | -10.20 |
| 13 | 39 |  |  |  |  |  |

## APPENDIX C: CALCULATIONS FOR PARAMETRIC STUDY

Table C. 1 Curvatures up to the mid-height of Wall 1 for a drift of $\mathbf{2 \%}$

| Node |  | Node 1 ydispl (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | $\begin{aligned} & \text { Difference } \\ & (\mathrm{mm}) \end{aligned}$ | Height (mm) | Strain | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1 | 53 | 0.00 | 15.39 | 846.60 | 846.60 | 423.30 | 0.0182 | 0.00326 |
| 53 | 105 | 15.39 | 30.70 | 1693.20 | 846.60 | 1269.90 | 0.0181 | 0.00320 |
| 105 | 183 | 30.70 | 49.08 | 2743.00 | 1049.80 | 2218.10 | 0.0175 | 0.00271 |
| 183 | 235 | 49.08 | 62.44 | 3589.60 | 846.60 | 3166.30 | 0.0158 | 0.00237 |
| 235 | 287 | 62.44 | 74.59 | 4436.20 | 846.60 | 4012.90 | 0.0143 | 0.00214 |
| 287 | 365 | 74.59 | 87.95 | 5486.00 | 1049.80 | 4961.10 | 0.0127 | 0.00190 |
| 365 | 417 | 87.95 | 94.19 | 6332.60 | 846.60 | 5909.30 | 0.0074 | 0.00117 |
| 417 | 469 | 94.19 | 97.11 | 7179.20 | 846.60 | 6755.90 | 0.0035 | 0.00064 |
| 469 | 547 | 97.11 | 99.87 | 8229.00 | 1049.80 | 7704.10 | 0.0026 | 0.00052 |
| 547 | 599 | 99.87 | 101.58 | 9075.60 | 846.60 | 8652.30 | 0.0020 | 0.00042 |
| 599 | 651 | 101.58 | 103.17 | 9922.20 | 846.60 | 9498.90 | 0.0019 | 0.00040 |
| 651 | 729 | 103.17 | 105.05 | 10972.00 | 1049.80 | 10447.10 | 0.0018 | 0.00038 |
| 729 | 781 | 105.05 | 106.53 | 11818.60 | 846.60 | 11395.30 | 0.0018 | 0.00037 |
| 781 | 833 | 106.53 | 107.97 | 12665.20 | 846.60 | 12241.90 | 0.0017 | 0.00036 |
| 833 | 911 | 107.97 | 109.70 | 13715.00 | 1049.80 | 13190.10 | 0.0016 | 0.00034 |
| 911 | 963 | 109.70 | 111.03 | 14561.60 | 846.60 | 14138.30 | 0.0016 | 0.00033 |
| 963 | 1015 | 111.03 | 112.30 | 15408.20 | 846.60 | 14984.90 | 0.0015 | 0.00032 |
| 1015 | 1093 | 112.30 | 113.79 | 16458.00 | 1049.80 | 15933.10 | 0.0014 | 0.00030 |
| 1093 | 1145 | 113.79 | 114.92 | 17304.60 | 846.60 | 16881.30 | 0.0013 | 0.00029 |
| 1145 | 1197 | 114.92 | 115.99 | 18151.20 | 846.60 | 17727.90 | 0.0013 | 0.00027 |
| 1197 | 1275 | 115.99 | 117.22 | 19201.00 | 1049.80 | 18676.10 | 0.0012 | 0.00026 |
| 1275 | 1327 | 117.22 | 118.14 | 20047.60 | 846.60 | 19624.30 | 0.0011 | 0.00024 |
| 1327 | 1379 | 118.14 | 118.99 | 20894.20 | 846.60 | 20470.90 | 0.0010 | 0.00023 |
| 1379 | 1457 | 118.99 | 119.96 | 21944.00 | 1049.80 | 21419.10 | 0.0009 | 0.00022 |
| 1457 | 1509 | 119.96 | 120.67 | 22790.60 | 846.60 | 22367.30 | 0.0008 | 0.00020 |
| 1509 | 1561 | 120.67 | 121.31 | 23637.20 | 846.60 | 23213.90 | 0.0008 | 0.00019 |
| 1561 | 1639 | 121.31 | 122.03 | 24687.00 | 1049.80 | 24162.10 | 0.0007 | 0.00017 |
| 1639 | 1691 | 122.03 | 122.55 | 25533.60 | 846.60 | 25110.30 | 0.0006 | 0.00016 |
| 1691 | 1743 | 122.55 | 123.01 | 26380.20 | 846.60 | 25956.90 | 0.0005 | 0.00015 |
| 1743 | 1821 | 123.01 | 123.52 | 27430.00 | 1049.80 | 26905.10 | 0.0005 | 0.00014 |
| 26 | 78 | 0.00 | -5.66 | 846.60 | 846.60 | 423.30 | -0.0067 | 0.00326 |
| 78 | 130 | -5.66 | -10.98 | 1693.20 | 846.60 | 1269.90 | -0.0063 | 0.00320 |
| 130 | 208 | -10.98 | -14.31 | 2743.00 | 1049.80 | 2218.10 | -0.0032 | 0.00271 |
| 208 | 260 | -14.31 | -16.24 | 3589.60 | 846.60 | 3166.30 | -0.0023 | 0.00237 |
| 260 | 312 | -16.24 | -17.93 | 4436.20 | 846.60 | 4012.90 | -0.0020 | 0.00214 |
| 312 | 390 | -17.93 | -19.76 | 5486.00 | 1049.80 | 4961.10 | -0.0017 | 0.00190 |
| 390 | 442 | -19.76 | -21.07 | 6332.60 | 846.60 | 5909.30 | -0.0015 | 0.00117 |
| 442 | 494 | -21.07 | -22.26 | 7179.20 | 846.60 | 6755.90 | -0.0014 | 0.00064 |
| 494 | 572 | -22.26 | -23.63 | 8229.00 | 1049.80 | 7704.10 | -0.0013 | 0.00052 |
| 572 | 624 | -23.63 | -24.65 | 9075.60 | 846.60 | 8652.30 | -0.0012 | 0.00042 |
| 624 | 676 | -24.65 | -25.63 | 9922.20 | 846.60 | 9498.90 | -0.0012 | 0.00040 |
| 676 | 754 | -25.63 | -26.78 | 10972.00 | 1049.80 | 10447.10 | -0.0011 | 0.00038 |
| 754 | 806 | -26.78 | -27.67 | 11818.60 | 846.60 | 11395.30 | -0.0011 | 0.00037 |
| 806 | 858 | -27.67 | -28.53 | 12665.20 | 846.60 | 12241.90 | -0.0010 | 0.00036 |
| 858 | 936 | -28.53 | -29.56 | 13715.00 | 1049.80 | 13190.10 | -0.0010 | 0.00034 |
| 936 | 988 | -29.56 | -30.36 | 14561.60 | 846.60 | 14138.30 | -0.0009 | 0.00033 |
| 988 | 1040 | -30.36 | -31.13 | 15408.20 | 846.60 | 14984.90 | -0.0009 | 0.00032 |
| 1040 | 1118 | -31.13 | -32.06 | 16458.00 | 1049.80 | 15933.10 | -0.0009 | 0.00030 |


| Node |  | Node 1 y- <br> displ (mm) | Node 2 y- <br> displ (mm) | Node 2 y- <br> coord (mm) | Difference <br> $(\mathrm{mm})$ | Height <br> $(\mathrm{mm})$ | Strain | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second | ( m (118 | 1170 | -32.06 | -32.78 | 17304.60 | 846.60 | 16881.30 |
| 118 | -0.0009 | 0.00029 |  |  |  |  |  |  |
| 1170 | 1222 | -32.78 | -33.48 | 18151.20 | 846.60 | 17727.90 | -0.0008 | 0.00027 |
| 1222 | 1300 | -33.48 | -34.31 | 19201.00 | 1049.80 | 18676.10 | -0.0008 | 0.00026 |
| 1300 | 1352 | -34.31 | -34.96 | 20047.60 | 846.60 | 19624.30 | -0.0008 | 0.00024 |
| 1352 | 1404 | -34.96 | -35.59 | 20894.20 | 846.60 | 20470.90 | -0.0007 | 0.00023 |
| 1404 | 1482 | -35.59 | -36.34 | 21944.00 | 1049.80 | 21419.10 | -0.0007 | 0.00022 |
| 1482 | 1534 | -36.34 | -36.93 | 22790.60 | 846.60 | 22367.30 | -0.0007 | 0.00020 |
| 1534 | 1586 | -36.93 | -37.49 | 23637.20 | 846.60 | 23213.90 | -0.0007 | 0.00019 |
| 1586 | 1664 | -37.49 | -38.16 | 24687.00 | 1049.80 | 24162.10 | -0.0006 | 0.00017 |
| 1664 | 1716 | -38.16 | -38.68 | 25533.60 | 846.60 | 25110.30 | -0.0006 | 0.00016 |
| 1716 | 1768 | -38.68 | -39.18 | 26380.20 | 846.60 | 25956.90 | -0.0006 | 0.00015 |
| 1768 | 1846 | -39.18 | -39.78 | 27430.00 | 1049.80 | 26905.10 | -0.0006 | 0.00014 |

Wall length: 7620 mm

Table C. 2 Interpolation of curvatures at storey heights of Wall 1 for a drift of $\mathbf{2 \%}$


Table C. 3 Bending moments along the height of Wall 1 for a drift of $\mathbf{2 \%}$

| Storey | Height (m) | Force (kN) | Shear (kN) | Moment incr. (kN.m) | Moment (kN.m) |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 0 | 0.00 | 0.0 | 1925.5 | 815.1 | 105632.9 |
|  | 0.42 | 0.0 | 1925.5 | 1630.1 | 104817.9 |
|  | 1.27 | 0.0 | 1925.5 | 1825.8 | 103187.7 |
|  | 2.22 | 0.0 | 1925.5 | 1010.7 | 101362.0 |
| 1 | 2.74 | 0.0 | 1925.5 | 815.1 | 100351.3 |
|  | 3.17 | 0.0 | 1925.5 | 1630.1 | 99536.2 |
|  | 4.01 | 0.0 | 1925.5 | 1825.8 | 97906.1 |
|  | 4.96 | 0.0 | 1925.5 | 1010.7 | 96080.3 |
| 2 | 5.49 | 0.0 | 1925.5 | 815.1 | 95069.6 |
|  | 5.91 | 0.0 | 1925.5 | 1630.1 | 94254.6 |
|  | 6.76 | 0.0 | 1925.5 | 1825.8 | 92624.4 |
|  | 7.70 | 0.0 | 1925.5 | 1010.7 | 90798.7 |
| 3 | 8.23 | 0.0 | 1925.5 | 815.1 | 89788.0 |
|  | 8.65 | 0.0 | 1925.5 | 1630.1 | 88972.9 |
|  | 9.50 | 0.0 | 1925.5 | 1825.8 | 87342.8 |
|  | 10.45 | 0.0 | 1925.5 | 1010.7 | 85517.0 |
| 4 | 10.97 | 0.0 | 1925.5 | 815.1 | 84506.3 |
|  | 11.40 | 0.0 | 1925.5 | 1630.1 | 83691.3 |
|  | 12.24 | 0.0 | 1925.5 | 1825.8 | 82061.2 |
|  | 13.19 | 0.0 | 1925.5 | 1010.7 | 80235.4 |
| 5 | 13.72 | 0.0 | 1925.5 | 815.1 | 79224.7 |
|  | 14.14 | 0.0 | 1925.5 | 1630.1 | 78409.6 |
|  | 14.98 | 0.0 | 1925.5 | 1825.8 | 76779.5 |
|  | 15.93 | 0.0 | 1925.5 | 1010.7 | 74953.7 |
| 6 | 16.46 | 0.0 | 1925.5 | 815.1 | 73943.1 |
|  | 16.88 | 0.0 | 1925.5 | 1630.1 | 73128.0 |
|  | 17.73 | 0.0 | 1925.5 | 1825.8 | 71497.9 |
|  | 18.68 | 0.0 | 1925.5 | 1010.7 | 69672.1 |
| 7 | 19.20 | 0.0 | 1925.5 | 815.1 | 68661.4 |
|  | 19.62 | 0.0 | 1925.5 | 1630.1 | 67846.3 |
|  | 20.47 | 0.0 | 1925.5 | 1825.8 | 66216.2 |
|  | 21.42 | 0.0 | 1925.5 | 1010.7 | 64390.5 |
| 8 | 21.94 | 0.0 | 1925.5 | 815.1 | 63379.8 |
|  | 22.37 | 0.0 | 1925.5 | 1630.1 | 62564.7 |
|  | 23.21 | 0.0 | 1925.5 | 1825.8 | 60934.6 |
|  | 24.16 | 0.0 | 1925.5 | 1010.7 | 59108.8 |
| 9 | 24.69 | 0.0 | 1925.5 | 815.1 | 58098.1 |
|  | 25.11 | 0.0 | 1925.5 | 1630.1 | 57283.0 |
|  | 25.96 | 0.0 | 1925.5 | 1825.8 | 55652.9 |
|  | 26.91 | 0.0 | 1925.5 | 1010.7 | 53827.2 |
| 10 | 27.43 | 0.0 | 1925.5 | 5281.6 | 52816.5 |
| 11 | 30.17 | 0.0 | 1925.5 | 5281.6 | 47534.8 |
| 12 | 32.92 | 0.0 | 1925.5 | 5281.6 | 42253.2 |
| 13 | 35.66 | 0.0 | 1925.5 | 5281.6 | 36971.5 |
| 14 | 38.40 | 0.0 | 1925.5 | 5281.6 | 31689.9 |
| 15 | 41.15 | 0.0 | 1925.5 | 5281.6 | 26408.2 |
| 16 | 43.89 | 0.0 | 1925.5 | 5281.6 | 21126.6 |
| 17 | 46.63 | 0.0 | 1925.5 | 5281.6 | 15844.9 |
| 18 | 49.37 | 0.0 | 1925.5 | 5281.6 | 10563.3 |
| 19 | 52.12 | 0.0 | 1925.5 | 5281.6 | 5281.6 |
| 20 | 54.86 | 1925.5 | 1925.5 | 0.0 | 0.0 |

Table C. 4 Curvatures up to the mid-height of Wall 2 for a drift of $\mathbf{2 \%}$

| Node |  | Node 1 y displ (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | $\begin{gathered} \text { Difference } \\ (\mathrm{mm}) \end{gathered}$ | Height (mm) | Strain | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1 | 29 | 0.00 | 7.52 | 846.60 | 846.60 | 423.30 | 0.0089 | 0.00300 |
| 29 | 57 | 7.52 | 14.01 | 1693.20 | 846.60 | 1269.90 | 0.0077 | 0.00257 |
| 57 | 99 | 14.01 | 20.58 | 2743.00 | 1049.80 | 2218.10 | 0.0063 | 0.00214 |
| 99 | 127 | 20.58 | 24.25 | 3589.60 | 846.60 | 3166.30 | 0.0043 | 0.00157 |
| 127 | 155 | 24.25 | 27.94 | 4436.20 | 846.60 | 4012.90 | 0.0044 | 0.00153 |
| 155 | 197 | 27.94 | 30.86 | 5486.00 | 1049.80 | 4961.10 | 0.0028 | 0.00109 |
| 197 | 225 | 30.86 | 32.74 | 6332.60 | 846.60 | 5909.30 | 0.0022 | 0.00091 |
| 225 | 253 | 32.74 | 34.56 | 7179.20 | 846.60 | 6755.90 | 0.0021 | 0.00088 |
| 253 | 295 | 34.56 | 36.55 | 8229.00 | 1049.80 | 7704.10 | 0.0019 | 0.00080 |
| 295 | 323 | 36.55 | 38.09 | 9075.60 | 846.60 | 8652.30 | 0.0018 | 0.00076 |
| 323 | 351 | 38.09 | 39.58 | 9922.20 | 846.60 | 9498.90 | 0.0018 | 0.00074 |
| 351 | 393 | 39.58 | 41.34 | 10972.00 | 1049.80 | 10447.10 | 0.0017 | 0.00071 |
| 393 | 421 | 41.34 | 42.69 | 11818.60 | 846.60 | 11395.30 | 0.0016 | 0.00068 |
| 421 | 449 | 42.69 | 43.98 | 12665.20 | 846.60 | 12241.90 | 0.0015 | 0.00065 |
| 449 | 491 | 43.98 | 45.49 | 13715.00 | 1049.80 | 13190.10 | 0.0014 | 0.00062 |
| 491 | 519 | 45.49 | 46.63 | 14561.60 | 846.60 | 14138.30 | 0.0014 | 0.00059 |
| 519 | 547 | 46.63 | 47.71 | 15408.20 | 846.60 | 14984.90 | 0.0013 | 0.00057 |
| 547 | 589 | 47.71 | 48.96 | 16458.00 | 1049.80 | 15933.10 | 0.0012 | 0.00054 |
| 589 | 617 | 48.96 | 49.90 | 17304.60 | 846.60 | 16881.30 | 0.0011 | 0.00051 |
| 617 | 645 | 49.90 | 50.77 | 18151.20 | 846.60 | 17727.90 | 0.0010 | 0.00048 |
| 645 | 687 | 50.77 | 51.76 | 19201.00 | 1049.80 | 18676.10 | 0.0009 | 0.00045 |
| 687 | 715 | 51.76 | 52.49 | 20047.60 | 846.60 | 19624.30 | 0.0009 | 0.00042 |
| 715 | 743 | 52.49 | 53.16 | 20894.20 | 846.60 | 20470.90 | 0.0008 | 0.00040 |
| 743 | 785 | 53.16 | 53.92 | 21944.00 | 1049.80 | 21419.10 | 0.0007 | 0.00037 |
| 785 | 813 | 53.92 | 54.47 | 22790.60 | 846.60 | 22367.30 | 0.0007 | 0.00034 |
| 813 | 841 | 54.47 | 54.98 | 23637.20 | 846.60 | 23213.90 | 0.0006 | 0.00032 |
| 841 | 883 | 54.98 | 55.54 | 24687.00 | 1049.80 | 24162.10 | 0.0005 | 0.00030 |
| 883 | 911 | 55.54 | 55.94 | 25533.60 | 846.60 | 25110.30 | 0.0005 | 0.00028 |
| 911 | 939 | 55.94 | 56.30 | 26380.20 | 846.60 | 25956.90 | 0.0004 | 0.00026 |
| 939 | 981 | 56.30 | 56.71 | 27430.00 | 1049.80 | 26905.10 | 0.0004 | 0.00024 |
| 14 | 42 | 0.00 | -2.16 | 846.60 | 846.60 | 423.30 | -0.0025 | 0.00300 |
| 42 | 70 | -2.16 | -3.96 | 1693.20 | 846.60 | 1269.90 | -0.0021 | 0.00257 |
| 70 | 112 | -3.96 | -5.93 | 2743.00 | 1049.80 | 2218.10 | -0.0019 | 0.00214 |
| 112 | 140 | -5.93 | -7.31 | 3589.60 | 846.60 | 3166.30 | -0.0016 | 0.00157 |
| 140 | 168 | -7.31 | -8.57 | 4436.20 | 846.60 | 4012.90 | -0.0015 | 0.00153 |
| 168 | 210 | -8.57 | -10.00 | 5486.00 | 1049.80 | 4961.10 | -0.0014 | 0.00109 |
| 210 | 238 | -10.00 | -11.07 | 6332.60 | 846.60 | 5909.30 | -0.0013 | 0.00091 |
| 238 | 266 | -11.07 | -12.08 | 7179.20 | 846.60 | 6755.90 | -0.0012 | 0.00088 |
| 266 | 308 | -12.08 | -13.28 | 8229.00 | 1049.80 | 7704.10 | -0.0011 | 0.00080 |
| 308 | 336 | -13.28 | -14.21 | 9075.60 | 846.60 | 8652.30 | -0.0011 | 0.00076 |
| 336 | 364 | -14.21 | -15.10 | 9922.20 | 846.60 | 9498.90 | -0.0011 | 0.00074 |
| 364 | 406 | -15.10 | -16.18 | 10972.00 | 1049.80 | 10447.10 | -0.0010 | 0.00071 |
| 406 | 434 | -16.18 | -17.01 | 11818.60 | 846.60 | 11395.30 | -0.0010 | 0.00068 |
| 434 | 462 | -17.01 | -17.82 | 12665.20 | 846.60 | 12241.90 | -0.0010 | 0.00065 |
| 462 | 504 | -17.82 | -18.81 | 13715.00 | 1049.80 | 13190.10 | -0.0009 | 0.00062 |
| 504 | 532 | -18.81 | -19.58 | 14561.60 | 846.60 | 14138.30 | -0.0009 | 0.00059 |
| 532 | 560 | -19.58 | -20.32 | 15408.20 | 846.60 | 14984.90 | -0.0009 | 0.00057 |
| 560 | 602 | -20.32 | -21.22 | 16458.00 | 1049.80 | 15933.10 | -0.0009 | 0.00054 |


| Node |  | Node 1 y- <br> displ (mm) | Node 2 y- <br> displ (mm) | Node 2 y- <br> coord (mm) | Difference <br> $(\mathrm{mm})$ | Height <br> $(\mathrm{mm})$ | Strain | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second | -21.22 | -21.91 | 17304.60 | 846.60 | 16881.30 | -0.0008 | 0.00051 |
| 602 | 630 | -22 | -22.59 | 18151.20 | 846.60 | 17727.90 | -0.0008 | 0.00048 |
| 630 | 658 | -21.91 | -23.40 | 19201.00 | 1049.80 | 18676.10 | -0.0008 | 0.00045 |
| 658 | 700 | -22.59 | -24.02 | 20047.60 | 846.60 | 19624.30 | -0.0007 | 0.00042 |
| 700 | 728 | -23.40 | -24.00 | 20.62 | 20894.20 | 846.60 | 20470.90 | -0.0007 |
| 728 | 756 | -24.02 | -24.62 | 0.00040 |  |  |  |  |
| 756 | 798 | -24.62 | -25.35 | 21944.00 | 1049.80 | 21419.10 | -0.0007 | 0.00037 |
| 798 | 826 | -25.35 | -25.90 | 22790.60 | 846.60 | 22367.30 | -0.0007 | 0.00034 |
| 826 | 854 | -25.90 | -26.44 | 23637.20 | 846.60 | 23213.90 | -0.0006 | 0.00032 |
| 854 | 896 | -26.44 | -27.09 | 24687.00 | 1049.80 | 24162.10 | -0.0006 | 0.00030 |
| 896 | 924 | -27.09 | -27.58 | 25533.60 | 846.60 | 25110.30 | -0.0006 | 0.00028 |
| 924 | 952 | -27.58 | -28.06 | 26380.20 | 846.60 | 25956.90 | -0.0006 | 0.00026 |
| 952 | 994 | -28.06 | -28.63 | 27430.00 | 1049.80 | 26905.10 | -0.0005 | 0.00024 |

Wall length: $\quad 3810 \mathrm{~mm}$

Table C. 5 Interpolation of curvatures at storey heights of Wall 2 for a drift of $\mathbf{2 \%}$


Table C. 6 Bending moments along the height of Wall 2 for a drift of $\mathbf{2 \%}$

| Storey | Height (m) | Force (kN) | Shear (kN) | Moment incr. (kN.m) | Moment (kN.m) |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 0 | 0.00 | 0.0 | 463.1 | 196.0 | 25405.7 |
|  | 0.42 | 0.0 | 463.1 | 392.1 | 25209.6 |
|  | 1.27 | 0.0 | 463.1 | 439.1 | 24817.6 |
|  | 2.22 | 0.0 | 463.1 | 243.1 | 24378.5 |
| 1 | 2.74 | 0.0 | 463.1 | 196.0 | 24135.4 |
|  | 3.17 | 0.0 | 463.1 | 392.1 | 23939.4 |
|  | 4.01 | 0.0 | 463.1 | 439.1 | 23547.3 |
|  | 4.96 | 0.0 | 463.1 | 243.1 | 23108.2 |
| 2 | 5.49 | 0.0 | 463.1 | 196.0 | 22865.1 |
|  | 5.91 | 0.0 | 463.1 | 392.1 | 22669.1 |
|  | 6.76 | 0.0 | 463.1 | 439.1 | 22277.0 |
|  | 7.70 | 0.0 | 463.1 | 243.1 | 21837.9 |
| 3 | 8.23 | 0.0 | 463.1 | 196.0 | 21594.8 |
|  | 8.65 | 0.0 | 463.1 | 392.1 | 21398.8 |
|  | 9.50 | 0.0 | 463.1 | 439.1 | 21006.7 |
|  | 10.45 | 0.0 | 463.1 | 243.1 | 20567.6 |
| 4 | 10.97 | 0.0 | 463.1 | 196.0 | 20324.5 |
|  | 11.40 | 0.0 | 463.1 | 392.1 | 20128.5 |
|  | 12.24 | 0.0 | 463.1 | 439.1 | 19736.4 |
|  | 13.19 | 0.0 | 463.1 | 243.1 | 19297.3 |
| 5 | 13.72 | 0.0 | 463.1 | 196.0 | 19054.2 |
|  | 14.14 | 0.0 | 463.1 | 392.1 | 18858.2 |
|  | 14.98 | 0.0 | 463.1 | 439.1 | 18466.2 |
|  | 15.93 | 0.0 | 463.1 | 243.1 | 18027.0 |
| 6 | 16.46 | 0.0 | 463.1 | 196.0 | 17784.0 |
|  | 16.88 | 0.0 | 463.1 | 392.1 | 17587.9 |
|  | 17.73 | 0.0 | 463.1 | 439.1 | 17195.9 |
|  | 18.68 | 0.0 | 463.1 | 243.1 | 16756.8 |
| 7 | 19.20 | 0.0 | 463.1 | 196.0 | 16513.7 |
|  | 19.62 | 0.0 | 463.1 | 392.1 | 16317.7 |
|  | 20.47 | 0.0 | 463.1 | 439.1 | 15925.6 |
|  | 21.42 | 0.0 | 463.1 | 243.1 | 15486.5 |
| 8 | 21.94 | 0.0 | 463.1 | 196.0 | 15243.4 |
|  | 22.37 | 0.0 | 463.1 | 392.1 | 15047.4 |
|  | 23.21 | 0.0 | 463.1 | 439.1 | 14655.3 |
|  | 24.16 | 0.0 | 463.1 | 243.1 | 14216.2 |
| 9 | 24.69 | 0.0 | 463.1 | 196.0 | 13973.1 |
|  | 25.11 | 0.0 | 463.1 | 392.1 | 13777.1 |
|  | 25.96 | 0.0 | 463.1 | 439.1 | 13385.0 |
|  | 26.91 | 0.0 | 463.1 | 243.1 | 12945.9 |
| 10 | 27.43 | 0.0 | 463.1 | 1270.3 | 12702.8 |
| 11 | 30.17 | 0.0 | 463.1 | 1270.3 | 11432.5 |
| 12 | 32.92 | 0.0 | 463.1 | 1270.3 | 10162.3 |
| 13 | 35.66 | 0.0 | 463.1 | 1270.3 | 8892.0 |
| 14 | 38.40 | 0.0 | 463.1 | 1270.3 | 7621.7 |
| 15 | 41.15 | 0.0 | 463.1 | 1270.3 | 6351.4 |
| 16 | 43.89 | 0.0 | 463.1 | 1270.3 | 5081.1 |
| 17 | 46.63 | 0.0 | 463.1 | 1270.3 | 3810.8 |
| 18 | 49.37 | 0.0 | 463.1 | 1270.3 | 2540.6 |
| 19 | 52.12 | 0.0 | 463.1 | 1270.3 | 1270.3 |
| 20 | 54.86 | 463.1 | 463.1 | 0.0 | 0.0 |

Table C. 7 Average shear stress in Wall 1

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 2.0 | 0.61 | 1.39 | 1925.5 | 7620.0 | 508.0 | 0.62 |
| 1.8 | 0.61 | 1.19 | 1912.3 | 7620.0 | 508.0 | 0.62 |
| 1.6 | 0.61 | 0.99 | 1901.5 | 7620.0 | 508.0 | 0.61 |
| 1.4 | 0.61 | 0.79 | 1883.8 | 7620.0 | 508.0 | 0.61 |
| 1.2 | 0.61 | 0.59 | 1864.4 | 7620.0 | 508.0 | 0.60 |

Table C. 8 Average shear stress in Wall 2

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 2.0 | 1.18 | 0.82 | 463.1 | 3810.0 | 508.0 | 0.30 |
| 1.8 | 1.18 | 0.62 | 458.2 | 3810.0 | 508.0 | 0.30 |
| 1.6 | 1.18 | 0.42 | 451.5 | 3810.0 | 508.0 | 0.29 |
| 1.4 | 1.18 | 0.22 | 438.4 | 3810.0 | 508.0 | 0.28 |
| 1.2 | 1.18 | 0.02 | 420.7 | 3810.0 | 508.0 | 0.27 |

Table C. 9 Average shear stress in Wall 1 for a shear span of 35659 mm

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $\boldsymbol{l}_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1.6 | 0.43 | 1.17 | 2996.5 | 7620.0 | 508.0 | 0.97 |
| 1.4 | 0.43 | 0.97 | 2982.1 | 7620.0 | 508.0 | 0.96 |
| 1.2 | 0.43 | 0.77 | 2962.8 | 7620.0 | 508.0 | 0.96 |
| 1.0 | 0.43 | 0.57 | 2911.0 | 7620.0 | 508.0 | 0.94 |

Table C. 10 Average shear stress in Wall 2 for a shear span of 35659 mm

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1.6 | 0.79 | 0.81 | 727.5 | 3810.0 | 508.0 | 0.47 |
| 1.4 | 0.79 | 0.61 | 722.8 | 3810.0 | 508.0 | 0.47 |
| 1.2 | 0.79 | 0.41 | 708.8 | 3810.0 | 508.0 | 0.46 |
| 1.0 | 0.79 | 0.21 | 686.9 | 3810.0 | 508.0 | 0.44 |

Table C. 11 Average shear stress in Wall 1 for a shear span of $\mathbf{2 7 4 3 0 \mathrm { mm }}$

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1.6 | 0.35 | 1.25 | 3941.8 | 7620.0 | 508.0 | 1.27 |
| 1.4 | 0.35 | 1.05 | 3936.7 | 7620.0 | 508.0 | 1.27 |
| 1.2 | 0.35 | 0.85 | 3897.4 | 7620.0 | 508.0 | 1.26 |
| 1.0 | 0.35 | 0.65 | 3855.8 | 7620.0 | 508.0 | 1.25 |
| 0.8 | 0.35 | 0.45 | 3794.0 | 7620.0 | 508.0 | 1.23 |

Table C. 12 Average shear stress in Wall 2 for a shear span of 27430 mm

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1.6 | 0.63 | 0.97 | 969.2 | 3810.0 | 508.0 | 0.63 |
| 1.4 | 0.63 | 0.77 | 958.4 | 3810.0 | 508.0 | 0.62 |
| 1.2 | 0.63 | 0.57 | 948.8 | 3810.0 | 508.0 | 0.61 |
| 1.0 | 0.63 | 0.37 | 933.4 | 3810.0 | 508.0 | 0.60 |
| 0.8 | 0.63 | 0.17 | 897.2 | 3810.0 | 508.0 | 0.58 |

Table C. 13 Average shear stress in Wall 1 for a shear span of 19201 mm

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1.2 | 0.28 | 0.92 | 5644.1 | 7620.0 | 508.0 | 1.82 |
| 1.0 | 0.28 | 0.72 | 5591.5 | 7620.0 | 508.0 | 1.81 |
| 0.8 | 0.28 | 0.52 | 5550.1 | 7620.0 | 508.0 | 1.79 |
| 0.6 | 0.28 | 0.32 | 5448.5 | 7620.0 | 508.0 | 1.76 |

Table C. 14 Average shear stress in Wall 2 for a shear span of 19201 mm

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1.2 | 0.47 | 0.73 | 1394.0 | 3810.0 | 508.0 | 0.90 |
| 1.0 | 0.47 | 0.53 | 1368.8 | 3810.0 | 508.0 | 0.88 |
| 0.8 | 0.47 | 0.33 | 1349.5 | 3810.0 | 508.0 | 0.87 |
| 0.6 | 0.47 | 0.13 | 1293.7 | 3810.0 | 508.0 | 0.84 |

Table C. 15 Average shear stress in Wall 1 for a web thickness of 254 mm and a shear span of 27430 mm

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1.8 | 0.37 | 1.43 | 3656.3 | 7620.0 | 254.0 | 2.36 |
| 1.6 | 0.37 | 1.23 | 3639.7 | 7620.0 | 254.0 | 2.35 |
| 1.4 | 0.37 | 1.03 | 3571.7 | 7620.0 | 254.0 | 2.31 |
| 1.2 | 0.37 | 0.83 | 3580.1 | 7620.0 | 254.0 | 2.31 |
| 1.0 | 0.37 | 0.63 | 3565.5 | 7620.0 | 254.0 | 2.30 |
| 0.8 | 0.37 | 0.43 | 3524.0 | 7620.0 | 254.0 | 2.28 |

Table C. 16 Average shear stress in Wall 2 for a web thickness of $\mathbf{2 5 4 m m}$ and a shear span of 27430 mm

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1.8 | 0.65 | 1.15 | 915.5 | 3810.0 | 254.0 | 1.18 |
| 1.6 | 0.65 | 0.95 | 904.7 | 3810.0 | 254.0 | 1.17 |
| 1.4 | 0.65 | 0.75 | 890.7 | 3810.0 | 254.0 | 1.15 |
| 1.2 | 0.65 | 0.55 | 880.4 | 3810.0 | 254.0 | 1.14 |
| 1.0 | 0.65 | 0.35 | 864.6 | 3810.0 | 254.0 | 1.12 |
| 0.8 | 0.65 | 0.15 | 844.4 | 3810.0 | 254.0 | 1.09 |

Table C. 17 Average shear stress in Wall 1 for a web thickness of $\mathbf{2 5 4 m m}$ and a shear span of 19201 mm

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1.8 | 0.30 | 1.50 | 5274.6 | 7620.0 | 254.0 | 3.41 |
| 1.6 | 0.30 | 1.30 | 5260.4 | 7620.0 | 254.0 | 3.40 |
| 1.4 | 0.30 | 1.10 | 5220.3 | 7620.0 | 254.0 | 3.37 |
| 1.2 | 0.30 | 0.90 | 5179.0 | 7620.0 | 254.0 | 3.34 |
| 1.0 | 0.30 | 0.70 | 5162.1 | 7620.0 | 254.0 | 3.33 |
| 0.8 | 0.30 | 0.50 | 5064.5 | 7620.0 | 254.0 | 3.27 |
| 0.6 | 0.30 | 0.30 | 5035.1 | 7620.0 | 254.0 | 3.25 |

Table C. 18 Average shear stress in Wall 2 for a web thickness of $\mathbf{2 5 4 m m}$ and a shear span of $\mathbf{1 9 2 0 1 m m}$

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1.8 | 0.49 | 1.31 | 1343.5 | 3810.0 | 254.0 | 1.74 |
| 1.6 | 0.49 | 1.11 | 1328.1 | 3810.0 | 254.0 | 1.72 |
| 1.4 | 0.49 | 0.91 | 1321.6 | 3810.0 | 254.0 | 1.71 |
| 1.2 | 0.49 | 0.71 | 1299.2 | 3810.0 | 254.0 | 1.68 |
| 1.0 | 0.49 | 0.51 | 1272.9 | 3810.0 | 254.0 | 1.64 |
| 0.8 | 0.49 | 0.31 | 1255.6 | 3810.0 | 254.0 | 1.62 |
| 0.6 | 0.49 | 0.11 | 1213.3 | 3810.0 | 254.0 | 1.57 |

Table C. 19 Average shear stress in Wall 1 for a compressive axial load ratio of $\mathbf{0 . 3}$

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1.2 | 0.78 | 0.42 | 3101.9 | 7620.0 | 508.0 | 1.00 |
| 1.0 | 0.78 | 0.22 | 3070.2 | 7620.0 | 508.0 | 0.99 |
| 0.8 | 0.78 | 0.02 | 2922.1 | 7620.0 | 508.0 | 0.94 |

Table C. 20 Average shear stress in Wall 2 for a compressive axial load ratio of $\mathbf{0 . 3}$

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 2.0 | 1.51 | 0.49 | 775.1 | 3810.0 | 508.0 | 0.50 |
| 1.8 | 1.51 | 0.29 | 761.6 | 3810.0 | 508.0 | 0.49 |
| 1.6 | 1.51 | 0.09 | 742.3 | 3810.0 | 508.0 | 0.48 |

Table C. 21 Average shear stress in Wall 1 for a compressive axial load ratio of $\mathbf{0 . 2}$

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1.4 | 0.69 | 0.71 | 2587.5 | 7620.0 | 508.0 | 0.84 |
| 1.2 | 0.69 | 0.51 | 2583.7 | 7620.0 | 508.0 | 0.83 |
| 1.0 | 0.69 | 0.31 | 2542.5 | 7620.0 | 508.0 | 0.82 |
| 0.8 | 0.69 | 0.11 | 2443.8 | 7620.0 | 508.0 | 0.79 |

Table C. 22 Average shear stress in Wall $\mathbf{2}$ for a compressive axial load ratio of $\mathbf{0 . 2}$

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 2.0 | 1.34 | 0.66 | 646.6 | 3810.0 | 508.0 | 0.42 |
| 1.8 | 1.34 | 0.46 | 634.0 | 3810.0 | 508.0 | 0.41 |
| 1.6 | 1.34 | 0.26 | 616.6 | 3810.0 | 508.0 | 0.40 |
| 1.4 | 1.34 | 0.06 | 594.5 | 3810.0 | 508.0 | 0.38 |

Table C. 23 Average shear stress in Wall 1 for no axial load

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 2.0 | 0.55 | 1.45 | 1051.6 | 7620.0 | 508.0 | 0.34 |
| 1.8 | 0.55 | 1.25 | 1034.1 | 7620.0 | 508.0 | 0.33 |
| 1.6 | 0.55 | 1.05 | 1019.5 | 7620.0 | 508.0 | 0.33 |
| 1.4 | 0.55 | 0.85 | 1005.2 | 7620.0 | 508.0 | 0.32 |
| 1.2 | 0.55 | 0.65 | 989.4 | 7620.0 | 508.0 | 0.32 |

Table C. 24 Average shear stress in Wall 2 for no axial load

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 2.0 | 1.08 | 0.92 | 250.1 | 3810.0 | 508.0 | 0.16 |
| 1.8 | 1.08 | 0.72 | 244.1 | 3810.0 | 508.0 | 0.16 |
| 1.6 | 1.08 | 0.52 | 242.7 | 3810.0 | 508.0 | 0.16 |
| 1.4 | 1.08 | 0.32 | 236.2 | 3810.0 | 508.0 | 0.15 |
| 1.2 | 1.08 | 0.12 | 231.2 | 3810.0 | 508.0 | 0.15 |

Table C. 25 Average shear stress in Wall 1 for a tensile axial load ratio of 0.02

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 2.0 | 0.55 | 1.45 | 809.0 | 7620.0 | 508.0 | 0.26 |
| 1.8 | 0.55 | 1.25 | 798.0 | 7620.0 | 508.0 | 0.26 |
| 1.6 | 0.55 | 1.05 | 782.1 | 7620.0 | 508.0 | 0.25 |
| 1.4 | 0.55 | 0.85 | 767.7 | 7620.0 | 508.0 | 0.25 |
| 1.2 | 0.55 | 0.65 | 752.1 | 7620.0 | 508.0 | 0.24 |

Table C. 26 Average shear stress in Wall 2 for a tensile axial load ratio of 0.02

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 2.0 | 1.11 | 0.89 | 190.7 | 3810.0 | 508.0 | 0.12 |
| 1.8 | 1.11 | 0.69 | 187.3 | 3810.0 | 508.0 | 0.12 |
| 1.6 | 1.11 | 0.49 | 186.1 | 3810.0 | 508.0 | 0.12 |
| 1.4 | 1.11 | 0.29 | 180.8 | 3810.0 | 508.0 | 0.12 |
| 1.2 | 1.11 | 0.09 | 175.5 | 3810.0 | 508.0 | 0.11 |

Table C. 27 Average shear stress in Wall 1 for a tensile axial load ratio of 0.05

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 2.0 | 0.58 | 1.42 | 553.8 | 7620.0 | 508.0 | 0.18 |
| 1.8 | 0.58 | 1.22 | 542.8 | 7620.0 | 508.0 | 0.18 |
| 1.6 | 0.58 | 1.02 | 529.2 | 7620.0 | 508.0 | 0.17 |
| 1.4 | 0.58 | 0.82 | 518.1 | 7620.0 | 508.0 | 0.17 |
| 1.2 | 0.58 | 0.62 | 506.6 | 7620.0 | 508.0 | 0.16 |

Table C. 28 Average shear stress in Wall 2 for a tensile axial load ratio of 0.05

| Drift (\%) | Yield drift <br> $(\%)$ | Plastic drift <br> $(\%)$ | $V(\mathrm{kN})$ | $l_{w}(\mathrm{~mm})$ | $b_{w}(\mathrm{~mm})$ | $v(\mathrm{MPa})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 2.0 | 1.17 | 0.83 | 125.9 | 3810.0 | 508.0 | 0.08 |
| 1.8 | 1.17 | 0.63 | 125.2 | 3810.0 | 508.0 | 0.08 |
| 1.6 | 1.17 | 0.43 | 121.5 | 3810.0 | 508.0 | 0.08 |
| 1.4 | 1.17 | 0.23 | 120.6 | 3810.0 | 508.0 | 0.08 |
| 1.2 | 1.17 | 0.03 | 114.7 | 3810.0 | 508.0 | 0.07 |

Table C. 29 Curvatures up to the mid-height of Wall 1 combined with Wall 2 for a drift of 2\%

| Node |  | Node 1 ydispl (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | Difference (mm) | $\begin{gathered} \text { Height } \\ (\mathrm{mm}) \end{gathered}$ | Strain | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1 | 81 | 0.00 | 15.30 | 846.60 | 846.60 | 423.30 | 0.0181 | 0.00317 |
| 81 | 161 | 15.30 | 30.51 | 1693.20 | 846.60 | 1269.90 | 0.0180 | 0.00311 |
| 161 | 281 | 30.51 | 47.37 | 2743.00 | 1049.80 | 2218.10 | 0.0161 | 0.00248 |
| 281 | 361 | 47.37 | 59.96 | 3589.60 | 846.60 | 3166.30 | 0.0149 | 0.00229 |
| 361 | 441 | 59.96 | 71.56 | 4436.20 | 846.60 | 4012.90 | 0.0137 | 0.00209 |
| 441 | 561 | 71.56 | 82.29 | 5486.00 | 1049.80 | 4961.10 | 0.0102 | 0.00156 |
| 561 | 641 | 82.29 | 88.74 | 6332.60 | 846.60 | 5909.30 | 0.0076 | 0.00123 |
| 641 | 721 | 88.74 | 94.21 | 7179.20 | 846.60 | 6755.90 | 0.0065 | 0.00106 |
| 721 | 841 | 94.21 | 98.66 | 8229.00 | 1049.80 | 7704.10 | 0.0042 | 0.00073 |
| 841 | 921 | 98.66 | 101.03 | 9075.60 | 846.60 | 8652.30 | 0.0028 | 0.00055 |
| 921 | 1001 | 101.03 | 102.93 | 9922.20 | 846.60 | 9498.90 | 0.0022 | 0.00046 |
| 1001 | 1121 | 102.93 | 104.52 | 10972.00 | 1049.80 | 10447.10 | 0.0015 | 0.00034 |
| 1121 | 1201 | 104.52 | 106.27 | 11818.60 | 846.60 | 11395.30 | 0.0021 | 0.00043 |
| 1201 | 1281 | 106.27 | 107.81 | 12665.20 | 846.60 | 12241.90 | 0.0018 | 0.00038 |
| 1281 | 1401 | 107.81 | 109.28 | 13715.00 | 1049.80 | 13190.10 | 0.0014 | 0.00031 |
| 1401 | 1481 | 109.28 | 110.77 | 14561.60 | 846.60 | 14138.30 | 0.0018 | 0.00037 |
| 1481 | 1561 | 110.77 | 112.19 | 15408.20 | 846.60 | 14984.90 | 0.0017 | 0.00035 |
| 1561 | 1681 | 112.19 | 113.52 | 16458.00 | 1049.80 | 15933.10 | 0.0013 | 0.00028 |
| 1681 | 1761 | 113.52 | 114.82 | 17304.60 | 846.60 | 16881.30 | 0.0015 | 0.00033 |
| 1761 | 1841 | 114.82 | 116.04 | 18151.20 | 846.60 | 17727.90 | 0.0014 | 0.00031 |
| 1841 | 1961 | 116.04 | 117.17 | 19201.00 | 1049.80 | 18676.10 | 0.0011 | 0.00024 |
| 1961 | 2041 | 117.17 | 118.26 | 20047.60 | 846.60 | 19624.30 | 0.0013 | 0.00028 |
| 2041 | 2121 | 118.26 | 119.27 | 20894.20 | 846.60 | 20470.90 | 0.0012 | 0.00026 |
| 2121 | 2241 | 119.27 | 120.18 | 21944.00 | 1049.80 | 21419.10 | 0.0009 | 0.00020 |
| 2241 | 2321 | 120.18 | 121.06 | 22790.60 | 846.60 | 22367.30 | 0.0010 | 0.00023 |
| 2321 | 2401 | 121.06 | 121.85 | 23637.20 | 846.60 | 23213.90 | 0.0009 | 0.00021 |
| 2401 | 2521 | 121.85 | 122.56 | 24687.00 | 1049.80 | 24162.10 | 0.0007 | 0.00017 |
| 2521 | 2601 | 122.56 | 123.21 | 25533.60 | 846.60 | 25110.30 | 0.0008 | 0.00019 |
| 2601 | 2681 | 123.21 | 123.79 | 26380.20 | 846.60 | 25956.90 | 0.0007 | 0.00017 |
| 2681 | 2801 | 123.79 | 124.32 | 27430.00 | 1049.80 | 26905.10 | 0.0005 | 0.00014 |
| 26 | 106 | 0.00 | -5.17 | 846.60 | 846.60 | 423.30 | -0.0061 | 0.00317 |
| 106 | 186 | -5.17 | -10.01 | 1693.20 | 846.60 | 1269.90 | -0.0057 | 0.00311 |
| 186 | 306 | -10.01 | -12.98 | 2743.00 | 1049.80 | 2218.10 | -0.0028 | 0.00248 |
| 306 | 386 | -12.98 | -15.18 | 3589.60 | 846.60 | 3166.30 | -0.0026 | 0.00229 |
| 386 | 466 | -15.18 | -17.04 | 4436.20 | 846.60 | 4012.90 | -0.0022 | 0.00209 |
| 466 | 586 | -17.04 | -18.81 | 5486.00 | 1049.80 | 4961.10 | -0.0017 | 0.00156 |
| 586 | 666 | -18.81 | -20.30 | 6332.60 | 846.60 | 5909.30 | -0.0018 | 0.00123 |
| 666 | 746 | -20.30 | -21.65 | 7179.20 | 846.60 | 6755.90 | -0.0016 | 0.00106 |
| 746 | 866 | -21.65 | -23.00 | 8229.00 | 1049.80 | 7704.10 | -0.0013 | 0.00073 |
| 866 | 946 | -23.00 | -24.18 | 9075.60 | 846.60 | 8652.30 | -0.0014 | 0.00055 |
| 946 | 1026 | -24.18 | -25.26 | 9922.20 | 846.60 | 9498.90 | -0.0013 | 0.00046 |
| 1026 | 1146 | -25.26 | -26.38 | 10972.00 | 1049.80 | 10447.10 | -0.0011 | 0.00034 |
| 1146 | 1226 | -26.38 | -27.38 | 11818.60 | 846.60 | 11395.30 | -0.0012 | 0.00043 |
| 1226 | 1306 | -27.38 | -28.32 | 12665.20 | 846.60 | 12241.90 | -0.0011 | 0.00038 |
| 1306 | 1426 | -28.32 | -29.30 | 13715.00 | 1049.80 | 13190.10 | -0.0009 | 0.00031 |
| 1426 | 1506 | -29.30 | -30.19 | 14561.60 | 846.60 | 14138.30 | -0.0011 | 0.00037 |
| 1506 | 1586 | -30.19 | -31.03 | 15408.20 | 846.60 | 14984.90 | -0.0010 | 0.00035 |
| 1586 | 1706 | -31.03 | -31.90 | 16458.00 | 1049.80 | 15933.10 | -0.0008 | 0.00028 |


| Node |  | Node 1 y- <br> displ (mm) | Node 2 y- <br> displ (mm) | Node 2 y- <br> coord (mm) | Difference <br> $(\mathrm{mm})$ | Height <br> $(\mathrm{mm})$ | Strain | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1706 | 1786 | -31.90 | -32.70 | 17304.60 | 846.60 | 16881.30 | -0.0009 | 0.00033 |
| 1786 | 1866 | -32.70 | -33.45 | 18151.20 | 846.60 | 17727.90 | -0.0009 | 0.00031 |
| 1866 | 1986 | -33.45 | -34.24 | 19201.00 | 1049.80 | 18676.10 | -0.0007 | 0.00024 |
| 1986 | 2066 | -34.24 | -34.95 | 20047.60 | 846.60 | 19624.30 | -0.0008 | 0.00028 |
| 2066 | 2146 | -34.95 | -35.62 | 20894.20 | 846.60 | 20470.90 | -0.0008 | 0.00026 |
| 2146 | 2266 | -35.62 | -36.32 | 21944.00 | 1049.80 | 21419.10 | -0.0007 | 0.00020 |
| 2266 | 2346 | -36.32 | -36.96 | 22790.60 | 846.60 | 22367.30 | -0.0007 | 0.00023 |
| 2346 | 2426 | -36.96 | -37.56 | 23637.20 | 846.60 | 23213.90 | -0.0007 | 0.00021 |
| 2426 | 2546 | -37.56 | -38.18 | 24687.00 | 1049.80 | 24162.10 | -0.0006 | 0.00017 |
| 2546 | 2626 | -38.18 | -38.74 | 25533.60 | 846.60 | 25110.30 | -0.0007 | 0.00019 |
| 2626 | 2706 | -38.74 | -39.28 | 26380.20 | 846.60 | 25956.90 | -0.0006 | 0.00017 |
| 2706 | 2826 | -39.28 | -39.84 | 27430.00 | 1049.80 | 26905.10 | -0.0005 | 0.00014 |

Wall length: 7620 mm

Table C. 30 Interpolation of curvatures at storey heights of Wall 1 combined with Wall 2 for a drift of 2\%


Table C. 31 Displacement components at the first storey of Wall 1 combined with Wall 2 for a drift of $\mathbf{2 \%}$

| Storey | 1 |  |  |  |
| :---: | :---: | :---: | :---: | :---: |
| Height (m) | 2.74 |  |  |  |
| Height (m) | Curvature (rad/m) | $\Delta \mathrm{x}(\mathrm{m})$ | x (m) | Flexural displ (mm) |
| 0.42 | 0.00317 |  | 2.32 |  |
| 1.27 | 0.00311 | 0.85 | 1.47 | 5.04 |
| 2.22 | 0.00248 | 0.95 | 0.52 | 2.65 |
| 2.74 | 0.00238 | 0.52 | 0.00 | 0.33 |
|  |  |  | Total | 8.02 |
|  |  | Total dis | ent (mm) | 18.61 |
|  |  | Shear di | ent (mm) | 10.59 |

Table C. 32 Displacement components at the second storey of Wall 1 combined with Wall 2 for a drift of $\mathbf{2 \%}$

| Storey |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Height $(\mathrm{m})$ | 5.49 |  |  |  |  |
| Height $(\mathrm{m})$ | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ | $\Delta \mathrm{x}(\mathrm{m})$ | $\mathrm{x}(\mathrm{m})$ | Flexural <br> displ $(\mathrm{mm})$ |  |
| 0.42 | 0.00317 |  | 5.06 |  |  |
| 1.27 | 0.00311 | 0.85 | 4.22 | 12.33 |  |
| 2.22 | 0.00248 | 0.95 | 3.27 | 9.91 |  |
| 2.74 | 0.00238 | 0.52 | 2.74 | 3.83 |  |
| 3.17 | 0.00229 | 0.42 | 2.32 | 2.50 |  |
| 4.01 | 0.00209 | 0.85 | 1.47 | 3.52 |  |
| 4.96 | 0.00156 | 0.95 | 0.52 | 1.73 |  |
| 5.49 | 0.00138 | 0.52 | 0.00 | 0.20 |  |
|  |  |  | Total | 34.02 |  |
|  |  | Total displacement (mm) | 52.87 |  |  |
|  |  | Shear displacement (mm) |  |  |  |

Table C. 33 Displacement components at the third storey of Wall 1 combined with Wall $\mathbf{2}$ for a drift of $\mathbf{2 \%}$

| Storey | 3 |  |  |  |
| :---: | :---: | :---: | :---: | :---: |
| Height (m) | 8.23 |  |  |  |
| Height (m) | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) | $\Delta \mathrm{x}(\mathrm{m})$ | x (m) | Flexural displ (mm) |
| 0.42 | 0.00317 |  | 7.81 |  |
| 1.27 | 0.00311 | 0.85 | 6.96 | 19.63 |
| 2.22 | 0.00248 | 0.95 | 6.01 | 17.18 |
| 2.74 | 0.00238 | 0.52 | 5.49 | 7.32 |
| 3.17 | 0.00229 | 0.42 | 5.06 | 5.21 |
| 4.01 | 0.00209 | 0.85 | 4.22 | 8.60 |
| 4.96 | 0.00156 | 0.95 | 3.27 | 6.47 |
| 5.49 | 0.00138 | 0.52 | 2.74 | 2.32 |
| 5.91 | 0.00123 | 0.42 | 2.32 | 1.40 |
| 6.76 | 0.00106 | 0.85 | 1.47 | 1.84 |
| 7.70 | 0.00073 | 0.95 | 0.52 | 0.84 |
| 8.23 | 0.00063 | 0.52 | 0.00 | 0.09 |
|  |  |  | Total | 70.91 |
|  |  | Total di | nent (mm) | 94.37 |
|  |  | Shear di | nent (mm) | 23.46 |

Table C. 34 Displacement components at the fourth storey of Wall 1 combined with Wall $\mathbf{2}$ for a drift of 2\%

| Storey | 4 |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Height (m) | 10.97 |  |  |  |  |
| Height (m) | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ | $\Delta \mathrm{x}(\mathrm{m})$ | $\mathrm{x}(\mathrm{m})$ | Flexural <br> displ (mm) |  |
| 0.42 | 0.00317 |  | 10.55 |  |  |
| 1.27 | 0.00311 | 0.85 | 9.70 | 26.92 |  |
| 2.22 | 0.00248 | 0.95 | 8.75 | 24.44 |  |
| 2.74 | 0.00238 | 0.52 | 8.23 | 10.82 |  |
| 3.17 | 0.00229 | 0.42 | 7.81 | 7.92 |  |
| 4.01 | 0.00209 | 0.85 | 6.96 | 13.69 |  |
| 4.96 | 0.00156 | 0.95 | 6.01 | 11.22 |  |
| 5.49 | 0.00138 | 0.52 | 5.49 | 4.44 |  |
| 5.91 | 0.00123 | 0.42 | 5.06 | 2.91 |  |
| 6.76 | 0.00106 | 0.85 | 4.22 | 4.49 |  |
| 7.70 | 0.00073 | 0.95 | 3.27 | 3.16 |  |
| 8.23 | 0.00063 | 0.52 | 2.74 | 1.07 |  |
| 8.65 | 0.00055 | 0.42 | 2.32 | 0.63 |  |
| 9.50 | 0.00046 | 0.85 | 1.47 | 0.81 |  |
| 10.45 | 0.00034 | 0.95 | 0.52 | 0.38 |  |
| 10.97 | 0.00039 | 0.52 | 0.00 | 0.05 |  |

Table C. 35 Slopes up to the mid-height of Wall 1 combined with Wall 2 for a drift of 2\%

| Height (m) | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ | $\Delta \mathrm{x}(\mathrm{m})$ | Slope incr. <br> $(\mathrm{rad})$ | Slope (rad) |
| :---: | :---: | :---: | :---: | :---: |
| 0.42 | 0.00317 |  |  |  |
| 1.27 | 0.00311 | 0.85 | 0.0027 | 0.0027 |
| 2.22 | 0.00248 | 0.95 | 0.0026 | 0.0053 |
| 3.17 | 0.00229 | 0.95 | 0.0023 | 0.0076 |
| 4.01 | 0.00209 | 0.85 | 0.0019 | 0.0094 |
| 4.96 | 0.00156 | 0.95 | 0.0017 | 0.0112 |
| 5.91 | 0.00123 | 0.95 | 0.0013 | 0.0125 |
| 6.76 | 0.00106 | 0.85 | 0.0010 | 0.0134 |
| 7.70 | 0.00073 | 0.95 | 0.0008 | 0.0143 |
| 8.65 | 0.00055 | 0.95 | 0.0006 | 0.0149 |
| 9.50 | 0.00046 | 0.85 | 0.0004 | 0.0153 |
| 10.45 | 0.00034 | 0.95 | 0.0004 | 0.0157 |
| 11.40 | 0.00043 | 0.95 | 0.0004 | 0.0161 |
| 12.24 | 0.00038 | 0.85 | 0.0003 | 0.0164 |
| 13.19 | 0.00031 | 0.95 | 0.0003 | 0.0167 |
| 14.14 | 0.00037 | 0.95 | 0.0003 | 0.0171 |
| 14.98 | 0.00035 | 0.85 | 0.0003 | 0.0174 |
| 15.93 | 0.00028 | 0.95 | 0.0003 | 0.0177 |
| 16.88 | 0.00033 | 0.95 | 0.0003 | 0.0179 |
| 17.73 | 0.00031 | 0.85 | 0.0003 | 0.0182 |
| 18.68 | 0.00024 | 0.95 | 0.0003 | 0.0185 |
| 19.62 | 0.00028 | 0.95 | 0.0002 | 0.0187 |
| 20.47 | 0.00026 | 0.85 | 0.0002 | 0.0189 |
| 21.42 | 0.00020 | 0.95 | 0.0002 | 0.0192 |
| 22.37 | 0.00023 | 0.95 | 0.0002 | 0.0194 |
| 23.21 | 0.00021 | 0.85 | 0.0002 | 0.0196 |
| 24.16 | 0.00017 | 0.95 | 0.0002 | 0.0197 |
| 25.11 | 0.00019 | 0.95 | 0.0002 | 0.0199 |
| 25.96 | 0.00017 | 0.85 | 0.0002 | 0.0201 |
| 26.91 | 0.00014 | 0.95 | 0.0001 | 0.0202 |

Table C. 36 Bending moments along the height of Wall 1 combined with Wall 2 for a drift of 2\%

| Storey | Height (m) | Force (kN) | Shear (kN) | Moment incr. (kN.m) | Moment (kN.m) |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 0 | 0.00 | 0.0 | 1726.0 | 730.6 | 106307.1 |
|  | 0.42 | 0.0 | 1726.0 | 1461.2 | 105576.5 |
|  | 1.27 | 0.0 | 1726.0 | 1636.6 | 104115.3 |
|  | 2.22 | 0.0 | 1726.0 | 906.0 | 102478.7 |
| 1 | 2.74 | -44.9 | 1726.0 | 740.1 | 101572.7 |
|  | 3.17 | 0.0 | 1770.9 | 1499.2 | 100832.6 |
|  | 4.01 | 0.0 | 1770.9 | 1679.2 | 99333.4 |
|  | 4.96 | 0.0 | 1770.9 | 929.5 | 97654.2 |
| 2 | 5.49 | 321.0 | 1770.9 | 681.7 | 96724.6 |
|  | 5.91 | 0.0 | 1449.9 | 1227.5 | 96043.0 |
|  | 6.76 | 0.0 | 1449.9 | 1374.8 | 94815.5 |
|  | 7.70 | 0.0 | 1449.9 | 761.1 | 93440.7 |
| 3 | 8.23 | -90.4 | 1449.9 | 632.9 | 92679.6 |
|  | 8.65 | 0.0 | 1540.3 | 1304.0 | 92046.8 |
|  | 9.50 | 0.0 | 1540.3 | 1460.5 | 90742.7 |
|  | 10.45 | 0.0 | 1540.3 | 808.5 | 89282.2 |
| 4 | 10.97 | -252.7 | 1540.3 | 705.5 | 88473.7 |
|  | 11.40 | 0.0 | 1793.0 | 1518.0 | 87768.2 |
|  | 12.24 | 0.0 | 1793.0 | 1700.1 | 86250.3 |
|  | 13.19 | 0.0 | 1793.0 | 941.1 | 84550.2 |
| 5 | 13.72 | -166.0 | 1793.0 | 794.1 | 83609.0 |
|  | 14.14 | 0.0 | 1959.0 | 1658.5 | 82814.9 |
|  | 14.98 | 0.0 | 1959.0 | 1857.5 | 81156.4 |
|  | 15.93 | 0.0 | 1959.0 | 1028.3 | 79298.9 |
| 6 | 16.46 | -69.8 | 1959.0 | 844.0 | 78270.6 |
|  | 16.88 | 0.0 | 2028.8 | 1717.6 | 77426.6 |
|  | 17.73 | 0.0 | 2028.8 | 1923.7 | 75709.0 |
|  | 18.68 | 0.0 | 2028.8 | 1064.9 | 73785.3 |
| 7 | 19.20 | -14.2 | 2028.8 | 861.8 | 72720.4 |
|  | 19.62 | 0.0 | 2043.0 | 1729.6 | 71858.6 |
|  | 20.47 | 0.0 | 2043.0 | 1937.2 | 70129.0 |
|  | 21.42 | 0.0 | 2043.0 | 1072.4 | 68191.8 |
| 8 | 21.94 | 30.5 | 2043.0 | 858.3 | 67119.4 |
|  | 22.37 | 0.0 | 2012.5 | 1703.8 | 66261.1 |
|  | 23.21 | 0.0 | 2012.5 | 1908.3 | 64557.3 |
|  | 24.16 | 0.0 | 2012.5 | 1056.4 | 62649.1 |
| 9 | 24.69 | 67.4 | 2012.5 | 837.6 | 61592.7 |
|  | 25.11 | 0.0 | 1945.1 | 1646.7 | 60755.1 |
|  | 25.96 | 0.0 | 1945.1 | 1844.3 | 59108.3 |
|  | 26.91 | 0.0 | 1945.1 | 1021.0 | 57264.0 |
| 10 | 27.43 | 112.5 | 1945.1 | 5181.1 | 56243.0 |
| 11 | 30.17 | -42.1 | 1832.6 | 5084.6 | 51061.9 |
| 12 | 32.92 | -89.7 | 1874.7 | 5265.3 | 45977.3 |
| 13 | 35.66 | -89.4 | 1964.4 | 5511.0 | 40712.0 |
| 14 | 38.40 | -52.8 | 2053.8 | 5706.0 | 35201.1 |
| 15 | 41.15 | -22.6 | 2106.6 | 5809.4 | 29495.1 |
| 16 | 43.89 | -11.9 | 2129.2 | 5856.7 | 23685.7 |
| 17 | 46.63 | -26.6 | 2141.1 | 5909.5 | 17829.0 |
| 18 | 49.37 | -48.3 | 2167.7 | 6012.2 | 11919.4 |
| 19 | 52.12 | 124.9 | 2216.0 | 5907.2 | 5907.2 |
| 20 | 54.86 | 2091.1 | 2091.1 | 0.0 | 0.0 |

Table C. 37 Curvatures up to the mid-height of Wall 2 combined with Wall 1 for a drift of $\mathbf{2 \%}$

| Node |  | Node 1 ydispl (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | Difference (mm) | Height (mm) | Strain | Curvature $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 27 | 107 | 0.00 | 11.38 | 846.60 | 846.60 | 423.30 | 0.0134 | 0.00456 |
| 107 | 187 | 11.38 | 21.65 | 1693.20 | 846.60 | 1269.90 | 0.0121 | 0.00388 |
| 187 | 307 | 21.65 | 31.63 | 2743.00 | 1049.80 | 2218.10 | 0.0095 | 0.00296 |
| 307 | 387 | 31.63 | 36.59 | 3589.60 | 846.60 | 3166.30 | 0.0059 | 0.00198 |
| 387 | 467 | 36.59 | 39.24 | 4436.20 | 846.60 | 4012.90 | 0.0031 | 0.00121 |
| 467 | 587 | 39.24 | 41.02 | 5486.00 | 1049.80 | 4961.10 | 0.0017 | 0.00074 |
| 587 | 667 | 41.02 | 42.80 | 6332.60 | 846.60 | 5909.30 | 0.0021 | 0.00086 |
| 667 | 747 | 42.80 | 44.34 | 7179.20 | 846.60 | 6755.90 | 0.0018 | 0.00076 |
| 747 | 867 | 44.34 | 45.78 | 8229.00 | 1049.80 | 7704.10 | 0.0014 | 0.00059 |
| 867 | 947 | 45.78 | 47.08 | 9075.60 | 846.60 | 8652.30 | 0.0015 | 0.00065 |
| 947 | 1027 | 47.08 | 48.26 | 9922.20 | 846.60 | 9498.90 | 0.0014 | 0.00060 |
| 1027 | 1147 | 48.26 | 49.35 | 10972.00 | 1049.80 | 10447.10 | 0.0010 | 0.00046 |
| 1147 | 1227 | 49.35 | 50.25 | 11818.60 | 846.60 | 11395.30 | 0.0011 | 0.00049 |
| 1227 | 1307 | 50.25 | 51.07 | 12665.20 | 846.60 | 12241.90 | 0.0010 | 0.00045 |
| 1307 | 1427 | 51.07 | 51.83 | 13715.00 | 1049.80 | 13190.10 | 0.0007 | 0.00035 |
| 1427 | 1507 | 51.83 | 52.45 | 14561.60 | 846.60 | 14138.30 | 0.0007 | 0.00037 |
| 1507 | 1587 | 52.45 | 53.03 | 15408.20 | 846.60 | 14984.90 | 0.0007 | 0.00035 |
| 1587 | 1707 | 53.03 | 53.56 | 16458.00 | 1049.80 | 15933.10 | 0.0005 | 0.00028 |
| 1707 | 1787 | 53.56 | 54.03 | 17304.60 | 846.60 | 16881.30 | 0.0005 | 0.00031 |
| 1787 | 1867 | 54.03 | 54.45 | 18151.20 | 846.60 | 17727.90 | 0.0005 | 0.00029 |
| 1867 | 1987 | 54.45 | 54.85 | 19201.00 | 1049.80 | 18676.10 | 0.0004 | 0.00023 |
| 1987 | 2067 | 54.85 | 55.20 | 20047.60 | 846.60 | 19624.30 | 0.0004 | 0.00025 |
| 2067 | 2147 | 55.20 | 55.51 | 20894.20 | 846.60 | 20470.90 | 0.0004 | 0.00024 |
| 2147 | 2267 | 55.51 | 55.81 | 21944.00 | 1049.80 | 21419.10 | 0.0003 | 0.00019 |
| 2267 | 2347 | 55.81 | 56.06 | 22790.60 | 846.60 | 22367.30 | 0.0003 | 0.00021 |
| 2347 | 2427 | 56.06 | 56.28 | 23637.20 | 846.60 | 23213.90 | 0.0003 | 0.00019 |
| 2427 | 2547 | 56.28 | 56.47 | 24687.00 | 1049.80 | 24162.10 | 0.0002 | 0.00015 |
| 2547 | 2627 | 56.47 | 56.64 | 25533.60 | 846.60 | 25110.30 | 0.0002 | 0.00017 |
| 2627 | 2707 | 56.64 | 56.79 | 26380.20 | 846.60 | 25956.90 | 0.0002 | 0.00016 |
| 2707 | 2827 | 56.79 | 56.91 | 27430.00 | 1049.80 | 26905.10 | 0.0001 | 0.00013 |
| 40 | 120 | 0.00 | -3.32 | 846.60 | 846.60 | 423.30 | -0.0039 | 0.00456 |
| 120 | 200 | -3.32 | -5.55 | 1693.20 | 846.60 | 1269.90 | -0.0026 | 0.00388 |
| 200 | 320 | -5.55 | -7.42 | 2743.00 | 1049.80 | 2218.10 | -0.0018 | 0.00296 |
| 320 | 400 | -7.42 | -8.84 | 3589.60 | 846.60 | 3166.30 | -0.0017 | 0.00198 |
| 400 | 480 | -8.84 | -10.08 | 4436.20 | 846.60 | 4012.90 | -0.0015 | 0.00121 |
| 480 | 600 | -10.08 | -11.28 | 5486.00 | 1049.80 | 4961.10 | -0.0011 | 0.00074 |
| 600 | 680 | -11.28 | -12.28 | 6332.60 | 846.60 | 5909.30 | -0.0012 | 0.00086 |
| 680 | 760 | -12.28 | -13.19 | 7179.20 | 846.60 | 6755.90 | -0.0011 | 0.00076 |
| 760 | 880 | -13.19 | -14.10 | 8229.00 | 1049.80 | 7704.10 | -0.0009 | 0.00059 |
| 880 | 960 | -14.10 | -14.90 | 9075.60 | 846.60 | 8652.30 | -0.0009 | 0.00065 |
| 960 | 1040 | -14.90 | -15.64 | 9922.20 | 846.60 | 9498.90 | -0.0009 | 0.00060 |
| 1040 | 1160 | -15.64 | -16.39 | 10972.00 | 1049.80 | 10447.10 | -0.0007 | 0.00046 |
| 1160 | 1240 | -16.39 | -17.05 | 11818.60 | 846.60 | 11395.30 | -0.0008 | 0.00049 |
| 1240 | 1320 | -17.05 | -17.69 | 12665.20 | 846.60 | 12241.90 | -0.0007 | 0.00045 |
| 1320 | 1440 | -17.69 | -18.34 | 13715.00 | 1049.80 | 13190.10 | -0.0006 | 0.00035 |
| 1440 | 1520 | -18.34 | -18.92 | 14561.60 | 846.60 | 14138.30 | -0.0007 | 0.00037 |
| 1520 | 1600 | -18.92 | -19.48 | 15408.20 | 846.60 | 14984.90 | -0.0007 | 0.00035 |
| 1600 | 1720 | -19.48 | -20.06 | 16458.00 | 1049.80 | 15933.10 | -0.0006 | 0.00028 |


| Node |  | Node 1 y- <br> displ (mm) | Node 2 y- <br> displ (mm) | Node 2 y- <br> coord (mm) | Difference <br> $(\mathrm{mm})$ | Height <br> $(\mathrm{mm})$ | Strain | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second | din | 1800 | -20.06 | -20.58 | 17304.60 | 846.60 | 16881.30 |
| 1720 | -0.0006 | 0.00031 |  |  |  |  |  |  |
| 1800 | 1880 | -20.58 | -21.08 | 18151.20 | 846.60 | 17727.90 | -0.0006 | 0.00029 |
| 1880 | 2000 | -21.08 | -21.61 | 19201.00 | 1049.80 | 18676.10 | -0.0005 | 0.00023 |
| 2000 | 2080 | -21.61 | -22.08 | 20047.60 | 846.60 | 19624.30 | -0.0006 | 0.00025 |
| 2080 | 2160 | -22.08 | -22.53 | 20894.20 | 846.60 | 20470.90 | -0.0005 | 0.00024 |
| 2160 | 2280 | -22.53 | -23.00 | 21944.00 | 1049.80 | 21419.10 | -0.0004 | 0.00019 |
| 2280 | 2360 | -23.00 | -23.42 | 22790.60 | 846.60 | 22367.30 | -0.0005 | 0.00021 |
| 2360 | 2440 | -23.42 | -23.82 | 23637.20 | 846.60 | 23213.90 | -0.0005 | 0.00019 |
| 2440 | 2560 | -23.82 | -24.24 | 24687.00 | 1049.80 | 24162.10 | -0.0004 | 0.00015 |
| 2560 | 2640 | -24.24 | -24.62 | 25533.60 | 846.60 | 25110.30 | -0.0004 | 0.00017 |
| 2640 | 2720 | -24.62 | -24.98 | 26380.20 | 846.60 | 25956.90 | -0.0004 | 0.00016 |
| 2720 | 2840 | -24.98 | -25.36 | 27430.00 | 1049.80 | 26905.10 | -0.0004 | 0.00013 |

Wall length: 3810 mm

Table C. 38 Interpolation of curvatures at storey heights of Wall 2 combined with Wall 1 for a drift of $\mathbf{2 \%}$

| Storey | Height <br> (mm) | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: |
|  | 423.30 | 0.00456 |
|  | 1269.90 | 0.00388 |
|  | 2218.10 | 0.00296 |
| 1 | 2743.00 | 0.00242 |
|  | 3166.30 | 0.00198 |
|  | 4012.90 | 0.00121 |
|  | 4961.10 | 0.00074 |
| 2 | 5486.00 | 0.00081 |
|  | 5909.30 | 0.00086 |
|  | 6755.90 | 0.00076 |
|  | 7704.10 | 0.00059 |
| 3 | 8229.00 | 0.00062 |
|  | 8652.30 | 0.00065 |
|  | 9498.90 | 0.00060 |
|  | 10447.10 | 0.00046 |
| 4 | 10972.00 | 0.00047 |
|  | 11395.30 | 0.00049 |
|  | 12241.90 | 0.00045 |
|  | 13190.10 | 0.00035 |
| 5 | 13715.00 | 0.00036 |
|  | 14138.30 | 0.00037 |
|  | 14984.90 | 0.00035 |
|  | 15933.10 | 0.00028 |
| 6 | 16458.00 | 0.00029 |
|  | 16881.30 | 0.00031 |
|  | 17727.90 | 0.00029 |
|  | 18676.10 | 0.00023 |
| 7 | 19201.00 | 0.00024 |
|  | 19624.30 | 0.00025 |
|  | 20470.90 | 0.00024 |
|  | 21419.10 | 0.00019 |
| 8 | 21944.00 | 0.00020 |
|  | 22367.30 | 0.00021 |
|  | 23213.90 | 0.00019 |
|  | 24162.10 | 0.00015 |
| 9 | 24687.00 | 0.00016 |
|  | 25110.30 | 0.00017 |
|  | 25956.90 | 0.00016 |
|  | 26905.10 | 0.00013 |
| 10 | 27430.00 | 0.00011 |

Table C. 39 Displacement components at the first storey of Wall $\mathbf{2}$ combined with Wall 1 for a drift of $\mathbf{2 \%}$

| Storey | 1 |  |  |  |
| :---: | :---: | :---: | :---: | :---: |
| Height (m) | 2.74 |  |  |  |
| Height (m) | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) | $\Delta x(\mathrm{~m})$ | $x$ (m) | Flexural displ (mm) |
| 0.42 | 0.00456 |  | 2.32 |  |
| 1.27 | 0.00388 | 0.85 | 1.47 | 6.77 |
| 2.22 | 0.00296 | 0.95 | 0.52 | 3.24 |
| 2.74 | 0.00242 | 0.52 | 0.00 | 0.37 |
|  |  |  | Total | 10.38 |
|  |  | Total dis | ment (mm) | 18.61 |
|  |  | Shear dis | ent (mm) | 8.23 |

Table C. 40 Displacement components at the second storey of Wall 2 combined with Wall 1 for a drift of $\mathbf{2 \%}$

| Storey |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Height $(\mathrm{m})$ | 5.49 |  |  |  |  |
| Height $(\mathrm{m})$ | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ | $\Delta \mathrm{x}(\mathrm{m})$ | $\mathrm{x}(\mathrm{m})$ | Flexural <br> displ $(\mathrm{mm})$ |  |
| 0.42 | 0.00456 |  | 5.06 |  |  |
| 1.27 | 0.00388 | 0.85 | 4.22 | 16.56 |  |
| 2.22 | 0.00296 | 0.95 | 3.27 | 12.13 |  |
| 2.74 | 0.00242 | 0.52 | 2.74 | 4.24 |  |
| 3.17 | 0.00198 | 0.42 | 2.32 | 2.36 |  |
| 4.01 | 0.00121 | 0.85 | 1.47 | 2.56 |  |
| 4.96 | 0.00074 | 0.95 | 0.52 | 0.92 |  |
| 5.49 | 0.00081 | 0.52 | 0.00 | 0.11 |  |
|  |  | Total | 38.88 |  |  |

Table C. 41 Displacement components at the third storey of Wall 2 combined with Wall 1 for a drift of $\mathbf{2 \%}$


Table C. 42 Displacement components at the fourth storey of Wall 2 combined with Wall 1 for a drift of 2\%

: Table C. 43 Slopes up to the mid-height of Wall 2 combined with Wall 1 for a drift of 2\%

| Height (m) | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ | $\Delta \mathrm{x}(\mathrm{m})$ | Slope incr. <br> $(\mathrm{rad})$ | Slope (rad) |
| :---: | :---: | :---: | :---: | :---: |
| 0.42 | 0.00456 |  |  |  |
| 1.27 | 0.00388 | 0.85 | 0.0036 | 0.0036 |
| 2.22 | 0.00296 | 0.95 | 0.0032 | 0.0068 |
| 3.17 | 0.00198 | 0.95 | 0.0023 | 0.0092 |
| 4.01 | 0.00121 | 0.85 | 0.0013 | 0.0105 |
| 4.96 | 0.00074 | 0.95 | 0.0009 | 0.0114 |
| 5.91 | 0.00086 | 0.95 | 0.0008 | 0.0122 |
| 6.76 | 0.00076 | 0.85 | 0.0007 | 0.0129 |
| 7.70 | 0.00059 | 0.95 | 0.0006 | 0.0135 |
| 8.65 | 0.00065 | 0.95 | 0.0006 | 0.0141 |
| 9.50 | 0.00060 | 0.85 | 0.0005 | 0.0146 |
| 10.45 | 0.00046 | 0.95 | 0.0005 | 0.0151 |
| 11.40 | 0.00049 | 0.95 | 0.0004 | 0.0156 |
| 12.24 | 0.00045 | 0.85 | 0.0004 | 0.0160 |
| 13.19 | 0.00035 | 0.95 | 0.0004 | 0.0164 |
| 14.14 | 0.00037 | 0.95 | 0.0003 | 0.0167 |
| 14.98 | 0.00035 | 0.85 | 0.0003 | 0.0170 |
| 15.93 | 0.00028 | 0.95 | 0.0003 | 0.0173 |
| 16.88 | 0.00031 | 0.95 | 0.0003 | 0.0176 |
| 17.73 | 0.00029 | 0.85 | 0.0003 | 0.0178 |
| 18.68 | 0.00023 | 0.95 | 0.0002 | 0.0181 |
| 19.62 | 0.00025 | 0.95 | 0.0002 | 0.0183 |
| 20.47 | 0.00024 | 0.85 | 0.0002 | 0.0185 |
| 21.42 | 0.00019 | 0.95 | 0.0002 | 0.0187 |
| 22.37 | 0.00021 | 0.95 | 0.0002 | 0.0189 |
| 23.21 | 0.00019 | 0.85 | 0.0002 | 0.0191 |
| 24.16 | 0.00015 | 0.95 | 0.0002 | 0.0192 |
| 25.11 | 0.00017 | 0.95 | 0.0002 | 0.0194 |
| 25.96 | 0.00016 | 0.85 | 0.0001 | 0.0195 |
| 26.91 | 0.00013 | 0.95 | 0.0001 | 0.0197 |

Table C. 44 Bending moments along the height of Wall 2 combined with Wall 1 for a drift of $\mathbf{2 \%}$

| Storey | Height (m) | Force (kN) | Shear (kN) | Moment incr. (kN.m) | Moment (kN.m) |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 0 | 0.00 | 0.0 | 693.4 | 293.5 | 26421.1 |
|  | 0.42 | 0.0 | 693.4 | 587.0 | 26127.6 |
|  | 1.27 | 0.0 | 693.4 | 657.5 | 25540.6 |
|  | 2.22 | 0.0 | 693.4 | 364.0 | 24883.1 |
| 1 | 2.74 | 44.9 | 693.4 | 284.0 | 24519.1 |
|  | 3.17 | 0.0 | 648.5 | 549.0 | 24235.1 |
|  | 4.01 | 0.0 | 648.5 | 614.9 | 23686.1 |
|  | 4.96 | 0.0 | 648.5 | 340.4 | 23071.2 |
| 2 | 5.49 | -321.0 | 648.5 | 342.4 | 22730.8 |
|  | 5.91 | 0.0 | 969.5 | 820.8 | 22388.4 |
|  | 6.76 | 0.0 | 969.5 | 919.3 | 21567.6 |
|  | 7.70 | 0.0 | 969.5 | 508.9 | 20648.3 |
| 3 | 8.23 | 90.4 | 969.5 | 391.3 | 20139.4 |
|  | 8.65 | 0.0 | 879.1 | 744.2 | 19748.2 |
|  | 9.50 | 0.0 | 879.1 | 833.6 | 19003.9 |
|  | 10.45 | 0.0 | 879.1 | 461.4 | 18170.3 |
| 4 | 10.97 | 252.7 | 879.1 | 318.6 | 17708.9 |
|  | 11.40 | 0.0 | 626.4 | 530.3 | 17390.3 |
|  | 12.24 | 0.0 | 626.4 | 594.0 | 16860.0 |
|  | 13.19 | 0.0 | 626.4 | 328.8 | 16266.0 |
| 5 | 13.72 | 166.0 | 626.4 | 230.0 | 15937.2 |
|  | 14.14 | 0.0 | 460.4 | 389.8 | 15707.2 |
|  | 14.98 | 0.0 | 460.4 | 436.6 | 15317.4 |
|  | 15.93 | 0.0 | 460.4 | 241.7 | 14880.9 |
| 6 | 16.46 | 69.8 | 460.4 | 180.1 | 14639.2 |
|  | 16.88 | 0.0 | 390.6 | 330.7 | 14459.1 |
|  | 17.73 | 0.0 | 390.6 | 370.4 | 14128.4 |
|  | 18.68 | 0.0 | 390.6 | 205.0 | 13758.0 |
| 7 | 19.20 | 14.2 | 390.6 | 162.3 | 13553.0 |
|  | 19.62 | 0.0 | 376.4 | 318.7 | 13390.7 |
|  | 20.47 | 0.0 | 376.4 | 356.9 | 13072.0 |
|  | 21.42 | 0.0 | 376.4 | 197.6 | 12715.1 |
| 8 | 21.94 | -30.5 | 376.4 | 165.8 | 12517.5 |
|  | 22.37 | 0.0 | 406.9 | 344.5 | 12351.7 |
|  | 23.21 | 0.0 | 406.9 | 385.8 | 12007.3 |
|  | 24.16 | 0.0 | 406.9 | 213.6 | 11621.4 |
| 9 | 24.69 | -67.4 | 406.9 | 186.5 | 11407.9 |
|  | 25.11 | 0.0 | 474.3 | 401.5 | 11221.4 |
|  | 25.96 | 0.0 | 474.3 | 449.7 | 10819.8 |
|  | 26.91 | 0.0 | 474.3 | 249.0 | 10370.1 |
| 10 | 27.43 | -112.5 | 474.3 | 1455.3 | 10121.1 |
| 11 | 30.17 | 42.1 | 586.8 | 1551.9 | 8665.8 |
| 12 | 32.92 | 89.7 | 544.7 | 1371.1 | 7114.0 |
| 13 | 35.66 | 89.4 | 455.0 | 1125.5 | 5742.9 |
| 14 | 38.40 | 52.8 | 365.6 | 930.4 | 4617.4 |
| 15 | 41.15 | 22.6 | 312.8 | 827.0 | 3687.0 |
| 16 | 43.89 | 11.9 | 290.2 | 779.7 | 2860.0 |
| 17 | 46.63 | 26.6 | 278.3 | 726.9 | 2080.3 |
| 18 | 49.37 | 48.3 | 251.7 | 624.2 | 1353.4 |
| 19 | 52.12 | -124.9 | 203.4 | 729.2 | 729.2 |
| 20 | 54.86 | 328.3 | 328.3 | 0.0 | 0.0 |

Table C. 45 Curvatures up to the mid-height of Wall $\mathbf{3}$ for a drift of $\mathbf{2 \%}$

| Node |  | Node 1 ydispl (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | $\begin{aligned} & \text { Difference } \\ & (\mathrm{mm}) \end{aligned}$ | Height (mm) | Strain | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1 | 15 | 0.00 | 1.21 | 846.60 | 846.60 | 423.30 | 0.0014 | 0.00148 |
| 15 | 29 | 1.21 | 2.39 | 1693.20 | 846.60 | 1269.90 | 0.0014 | 0.00143 |
| 29 | 50 | 2.39 | 3.76 | 2743.00 | 1049.80 | 2218.10 | 0.0013 | 0.00137 |
| 50 | 64 | 3.76 | 4.80 | 3589.60 | 846.60 | 3166.30 | 0.0012 | 0.00130 |
| 64 | 78 | 4.80 | 5.76 | 4436.20 | 846.60 | 4012.90 | 0.0011 | 0.00124 |
| 78 | 99 | 5.76 | 6.86 | 5486.00 | 1049.80 | 4961.10 | 0.0011 | 0.00118 |
| 99 | 113 | 6.86 | 7.69 | 6332.60 | 846.60 | 5909.30 | 0.0010 | 0.00112 |
| 113 | 127 | 7.69 | 8.45 | 7179.20 | 846.60 | 6755.90 | 0.0009 | 0.00107 |
| 127 | 148 | 8.45 | 9.35 | 8229.00 | 1049.80 | 7704.10 | 0.0009 | 0.00103 |
| 148 | 162 | 9.35 | 10.03 | 9075.60 | 846.60 | 8652.30 | 0.0008 | 0.00098 |
| 162 | 176 | 10.03 | 10.66 | 9922.20 | 846.60 | 9498.90 | 0.0008 | 0.00095 |
| 176 | 197 | 10.66 | 11.41 | 10972.00 | 1049.80 | 10447.10 | 0.0007 | 0.00092 |
| 197 | 211 | 11.41 | 11.96 | 11818.60 | 846.60 | 11395.30 | 0.0006 | 0.00086 |
| 211 | 225 | 11.96 | 12.47 | 12665.20 | 846.60 | 12241.90 | 0.0006 | 0.00083 |
| 225 | 246 | 12.47 | 13.06 | 13715.00 | 1049.80 | 13190.10 | 0.0006 | 0.00079 |
| 246 | 260 | 13.06 | 13.47 | 14561.60 | 846.60 | 14138.30 | 0.0005 | 0.00074 |
| 260 | 274 | 13.47 | 13.84 | 15408.20 | 846.60 | 14984.90 | 0.0004 | 0.00070 |
| 274 | 295 | 13.84 | 14.25 | 16458.00 | 1049.80 | 15933.10 | 0.0004 | 0.00066 |
| 295 | 309 | 14.25 | 14.55 | 17304.60 | 846.60 | 16881.30 | 0.0003 | 0.00062 |
| 309 | 323 | 14.55 | 14.82 | 18151.20 | 846.60 | 17727.90 | 0.0003 | 0.00060 |
| 323 | 344 | 14.82 | 15.14 | 19201.00 | 1049.80 | 18676.10 | 0.0003 | 0.00059 |
| 344 | 358 | 15.14 | 15.37 | 20047.60 | 846.60 | 19624.30 | 0.0003 | 0.00055 |
| 358 | 372 | 15.37 | 15.58 | 20894.20 | 846.60 | 20470.90 | 0.0002 | 0.00053 |
| 372 | 393 | 15.58 | 15.83 | 21944.00 | 1049.80 | 21419.10 | 0.0002 | 0.00052 |
| 393 | 407 | 15.83 | 15.98 | 22790.60 | 846.60 | 22367.30 | 0.0002 | 0.00047 |
| 407 | 421 | 15.98 | 16.11 | 23637.20 | 846.60 | 23213.90 | 0.0002 | 0.00046 |
| 421 | 442 | 16.11 | 16.26 | 24687.00 | 1049.80 | 24162.10 | 0.0001 | 0.00044 |
| 442 | 456 | 16.26 | 16.37 | 25533.60 | 846.60 | 25110.30 | 0.0001 | 0.00042 |
| 456 | 470 | 16.37 | 16.46 | 26380.20 | 846.60 | 25956.90 | 0.0001 | 0.00041 |
| 470 | 491 | 16.46 | 16.56 | 27430.00 | 1049.80 | 26905.10 | 0.0001 | 0.00039 |
| 7 | 21 | 0.00 | -1.18 | 846.60 | 846.60 | 423.30 | -0.0014 | 0.00148 |
| 21 | 35 | -1.18 | -2.30 | 1693.20 | 846.60 | 1269.90 | -0.0013 | 0.00143 |
| 35 | 56 | -2.30 | -3.67 | 2743.00 | 1049.80 | 2218.10 | -0.0013 | 0.00137 |
| 56 | 70 | -3.67 | -4.73 | 3589.60 | 846.60 | 3166.30 | -0.0013 | 0.00130 |
| 70 | 84 | -4.73 | -5.77 | 4436.20 | 846.60 | 4012.90 | -0.0012 | 0.00124 |
| 84 | 105 | -5.77 | -7.03 | 5486.00 | 1049.80 | 4961.10 | -0.0012 | 0.00118 |
| 105 | 119 | -7.03 | -8.01 | 6332.60 | 846.60 | 5909.30 | -0.0012 | 0.00112 |
| 119 | 133 | -8.01 | -8.96 | 7179.20 | 846.60 | 6755.90 | -0.0011 | 0.00107 |
| 133 | 154 | -8.96 | -10.13 | 8229.00 | 1049.80 | 7704.10 | -0.0011 | 0.00103 |
| 154 | 168 | -10.13 | -11.03 | 9075.60 | 846.60 | 8652.30 | -0.0011 | 0.00098 |
| 168 | 182 | -11.03 | -11.93 | 9922.20 | 846.60 | 9498.90 | -0.0011 | 0.00095 |
| 182 | 203 | -11.93 | -13.01 | 10972.00 | 1049.80 | 10447.10 | -0.0010 | 0.00092 |
| 203 | 217 | -13.01 | -13.86 | 11818.60 | 846.60 | 11395.30 | -0.0010 | 0.00086 |
| 217 | 231 | -13.86 | -14.68 | 12665.20 | 846.60 | 12241.90 | -0.0010 | 0.00083 |
| 231 | 252 | -14.68 | -15.68 | 13715.00 | 1049.80 | 13190.10 | -0.0010 | 0.00079 |
| 252 | 266 | -15.68 | -16.46 | 14561.60 | 846.60 | 14138.30 | -0.0009 | 0.00074 |
| 266 | 280 | -16.46 | -17.21 | 15408.20 | 846.60 | 14984.90 | -0.0009 | 0.00070 |
| 280 | 301 | -17.21 | -18.12 | 16458.00 | 1049.80 | 15933.10 | -0.0009 | 0.00066 |


| Node |  | Node 1 y- <br> displ (mm) | Node 2 y- <br> displ (mm) | Node 2 y- <br> coord (mm) | Difference <br> $(\mathrm{mm})$ | Height <br> $(\mathrm{mm})$ | Strain | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second | -18.12 | -18.83 | 17304.60 | 846.60 | 16881.30 | -0.0008 | 0.00062 |
| 301 | 315 | -12 | -2.53 | 18151.20 | 846.60 | 17727.90 | -0.0008 | 0.00060 |
| 315 | 329 | -18.83 | -190 | 19201.00 | 1049.80 | 18676.10 | -0.0008 | 0.00059 |
| 329 | 350 | -19.53 | -20.37 | -21.04 | 20047.60 | 846.60 | 19624.30 | -0.0008 |
| 350 | 364 | -20.37 | -21.69 | 20894.20 | 846.60 | 20470.90 | -0.0008 | 0.0055 |
| 364 | 378 | -21.04 | -22.47 | 21944.00 | 1049.80 | 21419.10 | -0.0008 | 0.00052 |
| 378 | 399 | -21.69 | -220 |  |  |  |  |  |
| 399 | 413 | -22.47 | -23.09 | 22790.60 | 846.60 | 22367.30 | -0.0007 | 0.00047 |
| 413 | 427 | -23.09 | -23.69 | 23637.20 | 846.60 | 23213.90 | -0.0007 | 0.00046 |
| 427 | 448 | -23.69 | -24.42 | 24687.00 | 1049.80 | 24162.10 | -0.0007 | 0.00044 |
| 448 | 462 | -24.42 | -24.99 | 25533.60 | 846.60 | 25110.30 | -0.0007 | 0.00042 |
| 462 | 476 | -24.99 | -25.55 | 26380.20 | 846.60 | 25956.90 | -0.0007 | 0.00041 |
| 476 | 497 | -25.55 | -26.24 | 27430.00 | 1049.80 | 26905.10 | -0.0007 | 0.00039 |

Wall length: 1905 mm

Table C. 46 Interpolation of curvatures at storey heights of Wall $\mathbf{3}$ for a drift of $\mathbf{2 \%}$


Table C. 47 Bending moments along the height of Wall $\mathbf{3}$ for a drift of $\mathbf{2 \%}$

| Storey | Height (m) | Force (kN) | Shear (kN) | Moment incr. (kN.m) | Moment (kN.m) |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 0 | 0.00 | 0.0 | 135.4 | 57.3 | 7428.0 |
|  | 0.42 | 0.0 | 135.4 | 114.6 | 7370.7 |
|  | 1.27 | 0.0 | 135.4 | 128.4 | 7256.1 |
|  | 2.22 | 0.0 | 135.4 | 71.1 | 7127.7 |
| 1 | 2.74 | 0.0 | 135.4 | 57.3 | 7056.6 |
|  | 3.17 | 0.0 | 135.4 | 114.6 | 6999.3 |
|  | 4.01 | 0.0 | 135.4 | 128.4 | 6884.7 |
|  | 4.96 | 0.0 | 135.4 | 71.1 | 6756.3 |
| 2 | 5.49 | 0.0 | 135.4 | 57.3 | 6685.2 |
|  | 5.91 | 0.0 | 135.4 | 114.6 | 6627.9 |
|  | 6.76 | 0.0 | 135.4 | 128.4 | 6513.3 |
|  | 7.70 | 0.0 | 135.4 | 71.1 | 6384.9 |
| 3 | 8.23 | 0.0 | 135.4 | 57.3 | 6313.8 |
|  | 8.65 | 0.0 | 135.4 | 114.6 | 6256.5 |
|  | 9.50 | 0.0 | 135.4 | 128.4 | 6141.9 |
|  | 10.45 | 0.0 | 135.4 | 71.1 | 6013.5 |
| 4 | 10.97 | 0.0 | 135.4 | 57.3 | 5942.4 |
|  | 11.40 | 0.0 | 135.4 | 114.6 | 5885.1 |
|  | 12.24 | 0.0 | 135.4 | 128.4 | 5770.5 |
|  | 13.19 | 0.0 | 135.4 | 71.1 | 5642.1 |
| 5 | 13.72 | 0.0 | 135.4 | 57.3 | 5571.0 |
|  | 14.14 | 0.0 | 135.4 | 114.6 | 5513.7 |
|  | 14.98 | 0.0 | 135.4 | 128.4 | 5399.1 |
|  | 15.93 | 0.0 | 135.4 | 71.1 | 5270.7 |
| 6 | 16.46 | 0.0 | 135.4 | 57.3 | 5199.6 |
|  | 16.88 | 0.0 | 135.4 | 114.6 | 5142.3 |
|  | 17.73 | 0.0 | 135.4 | 128.4 | 5027.7 |
|  | 18.68 | 0.0 | 135.4 | 71.1 | 4899.3 |
| 7 | 19.20 | 0.0 | 135.4 | 57.3 | 4828.2 |
|  | 19.62 | 0.0 | 135.4 | 114.6 | 4770.9 |
|  | 20.47 | 0.0 | 135.4 | 128.4 | 4656.3 |
|  | 21.42 | 0.0 | 135.4 | 71.1 | 4527.9 |
| 8 | 21.94 | 0.0 | 135.4 | 57.3 | 4456.8 |
|  | 22.37 | 0.0 | 135.4 | 114.6 | 4399.5 |
|  | 23.21 | 0.0 | 135.4 | 128.4 | 4284.9 |
|  | 24.16 | 0.0 | 135.4 | 71.1 | 4156.5 |
| 9 | 24.69 | 0.0 | 135.4 | 57.3 | 4085.4 |
|  | 25.11 | 0.0 | 135.4 | 114.6 | 4028.1 |
|  | 25.96 | 0.0 | 135.4 | 128.4 | 3913.5 |
|  | 26.91 | 0.0 | 135.4 | 71.1 | 3785.1 |
| 10 | 27.43 | 0.0 | 135.4 | 371.4 | 3714.0 |
| 11 | 30.17 | 0.0 | 135.4 | 371.4 | 3342.6 |
| 12 | 32.92 | 0.0 | 135.4 | 371.4 | 2971.2 |
| 13 | 35.66 | 0.0 | 135.4 | 371.4 | 2599.8 |
| 14 | 38.40 | 0.0 | 135.4 | 371.4 | 2228.4 |
| 15 | 41.15 | 0.0 | 135.4 | 371.4 | 1857.0 |
| 16 | 43.89 | 0.0 | 135.4 | 371.4 | 1485.6 |
| 17 | 46.63 | 0.0 | 135.4 | 371.4 | 1114.2 |
| 18 | 49.37 | 0.0 | 135.4 | 371.4 | 742.8 |
| 19 | 52.12 | 0.0 | 135.4 | 371.4 | 371.4 |
| 20 | 54.86 | 135.4 | 135.4 | 0.0 | 0.0 |

Table C. 48 Curvatures up to the mid-height of Wall 1 combined with Wall 3 for a drift of $\mathbf{2 \%}$

| Node |  | Node 1 ydispl (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | $\begin{gathered} \text { Difference } \\ (\mathrm{mm}) \end{gathered}$ | Height (mm) | Strain | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1 | 67 | 0.00 | 15.02 | 846.60 | 846.60 | 423.30 | 0.0177 | 0.00310 |
| 67 | 133 | 15.02 | 29.99 | 1693.20 | 846.60 | 1269.90 | 0.0177 | 0.00307 |
| 133 | 232 | 29.99 | 47.07 | 2743.00 | 1049.80 | 2218.10 | 0.0163 | 0.00251 |
| 232 | 298 | 47.07 | 60.20 | 3589.60 | 846.60 | 3166.30 | 0.0155 | 0.00238 |
| 298 | 364 | 60.20 | 71.74 | 4436.20 | 846.60 | 4012.90 | 0.0136 | 0.00209 |
| 364 | 463 | 71.74 | 82.30 | 5486.00 | 1049.80 | 4961.10 | 0.0101 | 0.00155 |
| 463 | 529 | 82.30 | 89.14 | 6332.60 | 846.60 | 5909.30 | 0.0081 | 0.00130 |
| 529 | 595 | 89.14 | 95.15 | 7179.20 | 846.60 | 6755.90 | 0.0071 | 0.00114 |
| 595 | 694 | 95.15 | 99.06 | 8229.00 | 1049.80 | 7704.10 | 0.0037 | 0.00066 |
| 694 | 760 | 99.06 | 101.27 | 9075.60 | 846.60 | 8652.30 | 0.0026 | 0.00053 |
| 760 | 826 | 101.27 | 103.17 | 9922.20 | 846.60 | 9498.90 | 0.0022 | 0.00046 |
| 826 | 925 | 103.17 | 104.79 | 10972.00 | 1049.80 | 10447.10 | 0.0015 | 0.00034 |
| 925 | 991 | 104.79 | 106.54 | 11818.60 | 846.60 | 11395.30 | 0.0021 | 0.00042 |
| 991 | 1057 | 106.54 | 108.08 | 12665.20 | 846.60 | 12241.90 | 0.0018 | 0.00038 |
| 1057 | 1156 | 108.08 | 109.54 | 13715.00 | 1049.80 | 13190.10 | 0.0014 | 0.00030 |
| 1156 | 1222 | 109.54 | 111.02 | 14561.60 | 846.60 | 14138.30 | 0.0017 | 0.00036 |
| 1222 | 1288 | 111.02 | 112.43 | 15408.20 | 846.60 | 14984.90 | 0.0017 | 0.00035 |
| 1288 | 1387 | 112.43 | 113.74 | 16458.00 | 1049.80 | 15933.10 | 0.0013 | 0.00027 |
| 1387 | 1453 | 113.74 | 115.03 | 17304.60 | 846.60 | 16881.30 | 0.0015 | 0.00032 |
| 1453 | 1519 | 115.03 | 116.24 | 18151.20 | 846.60 | 17727.90 | 0.0014 | 0.00030 |
| 1519 | 1618 | 116.24 | 117.35 | 19201.00 | 1049.80 | 18676.10 | 0.0011 | 0.00024 |
| 1618 | 1684 | 117.35 | 118.43 | 20047.60 | 846.60 | 19624.30 | 0.0013 | 0.00028 |
| 1684 | 1750 | 118.43 | 119.42 | 20894.20 | 846.60 | 20470.90 | 0.0012 | 0.00026 |
| 1750 | 1849 | 119.42 | 120.33 | 21944.00 | 1049.80 | 21419.10 | 0.0009 | 0.00020 |
| 1849 | 1915 | 120.33 | 121.19 | 22790.60 | 846.60 | 22367.30 | 0.0010 | 0.00023 |
| 1915 | 1981 | 121.19 | 121.97 | 23637.20 | 846.60 | 23213.90 | 0.0009 | 0.00021 |
| 1981 | 2080 | 121.97 | 122.68 | 24687.00 | 1049.80 | 24162.10 | 0.0007 | 0.00017 |
| 2080 | 2146 | 122.68 | 123.34 | 25533.60 | 846.60 | 25110.30 | 0.0008 | 0.00019 |
| 2146 | 2212 | 123.34 | 123.92 | 26380.20 | 846.60 | 25956.90 | 0.0007 | 0.00017 |
| 2212 | 2311 | 123.92 | 124.44 | 27430.00 | 1049.80 | 26905.10 | 0.0005 | 0.00013 |
| 26 | 92 | 0.00 | -4.98 | 846.60 | 846.60 | 423.30 | -0.0059 | 0.00310 |
| 92 | 158 | -4.98 | -9.82 | 1693.20 | 846.60 | 1269.90 | -0.0057 | 0.00307 |
| 158 | 257 | -9.82 | -12.86 | 2743.00 | 1049.80 | 2218.10 | -0.0029 | 0.00251 |
| 257 | 323 | -12.86 | -15.11 | 3589.60 | 846.60 | 3166.30 | -0.0027 | 0.00238 |
| 323 | 389 | -15.11 | -17.04 | 4436.20 | 846.60 | 4012.90 | -0.0023 | 0.00209 |
| 389 | 488 | -17.04 | -18.86 | 5486.00 | 1049.80 | 4961.10 | -0.0017 | 0.00155 |
| 488 | 554 | -18.86 | -20.40 | 6332.60 | 846.60 | 5909.30 | -0.0018 | 0.00130 |
| 554 | 620 | -20.40 | -21.77 | 7179.20 | 846.60 | 6755.90 | -0.0016 | 0.00114 |
| 620 | 719 | -21.77 | -23.13 | 8229.00 | 1049.80 | 7704.10 | -0.0013 | 0.00066 |
| 719 | 785 | -23.13 | -24.30 | 9075.60 | 846.60 | 8652.30 | -0.0014 | 0.00053 |
| 785 | 851 | -24.30 | -25.37 | 9922.20 | 846.60 | 9498.90 | -0.0013 | 0.00046 |
| 851 | 950 | -25.37 | -26.47 | 10972.00 | 1049.80 | 10447.10 | -0.0010 | 0.00034 |
| 950 | 1016 | -26.47 | -27.46 | 11818.60 | 846.60 | 11395.30 | -0.0012 | 0.00042 |
| 1016 | 1082 | -27.46 | -28.37 | 12665.20 | 846.60 | 12241.90 | -0.0011 | 0.00038 |
| 1082 | 1181 | -28.37 | -29.34 | 13715.00 | 1049.80 | 13190.10 | -0.0009 | 0.00030 |
| 1181 | 1247 | -29.34 | -30.21 | 14561.60 | 846.60 | 14138.30 | -0.0010 | 0.00036 |
| 1247 | 1313 | -30.21 | -31.03 | 15408.20 | 846.60 | 14984.90 | -0.0010 | 0.00035 |
| 1313 | 1412 | -31.03 | -31.90 | 16458.00 | 1049.80 | 15933.10 | -0.0008 | 0.00027 |


| Node |  | Node 1 y- <br> displ (mm) | Node 2 y- <br> displ (mm) | Node 2 y- <br> coord (mm) | Difference <br> $(\mathrm{mm})$ | Height <br> $(\mathrm{mm})$ | Strain | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second | -31.90 | -32.68 | 17304.60 | 846.60 | 16881.30 | -0.0009 | 0.00032 |
| 1412 | 1478 | -32.68 | -33.43 | 18151.20 | 846.60 | 17727.90 | -0.0009 | 0.00030 |
| 1478 | 1544 | -33.43 | -34.21 | 19201.00 | 1049.80 | 18676.10 | -0.0007 | 0.00024 |
| 1544 | 1643 | -33.26 | -34.91 | 20047.60 | 846.60 | 19624.30 | -0.0008 | 0.00028 |
| 1643 | 1709 | -34.21 | -34.91 | -35.58 | 20894.20 | 846.60 | 20470.90 | -0.0008 |
| 1709 | 1775 | -35.58 | -36.29 | 21944.00 | 1049.80 | 21419.10 | -0.0007 | 0.00026 |
| 1775 | 1874 | -36.29 | -36.92 | 22790.60 | 846.60 | 22367.30 | -0.0007 | 0.00023 |
| 1874 | 1940 | -36.92 | -37.52 | 23637.20 | 846.60 | 23213.90 | -0.0007 | 0.00021 |
| 1940 | 2006 | -36.92 | -38.14 | 24687.00 | 1049.80 | 24162.10 | -0.0006 | 0.00017 |
| 2006 | 2105 | -37.52 | -38.14 | -38.71 | 25533.60 | 846.60 | 25110.30 | -0.0007 |
| 2105 | 2171 | -38.14 | -38.00019 |  |  |  |  |  |
| 2171 | 2237 | -38.71 | -39.24 | 26380.20 | 846.60 | 25956.90 | -0.0006 | 0.00017 |
| 2237 | 2336 | -39.24 | -39.80 | 27430.00 | 1049.80 | 26905.10 | -0.0005 | 0.00013 |

Wall length: $\quad 7620 \mathrm{~mm}$

Table C. 49 Curvatures up to the mid-height of Wall $\mathbf{3}$ combined with Wall 1 for a drift of $\mathbf{2 \%}$

| Node |  | Node 1 y displ (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | Difference (mm) | Height (mm) | Strain | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 27 | 93 | 0.00 | 6.60 | 846.60 | 846.60 | 423.30 | 0.0078 | 0.00581 |
| 93 | 159 | 6.60 | 11.04 | 1693.20 | 846.60 | 1269.90 | 0.0052 | 0.00399 |
| 159 | 258 | 11.04 | 12.83 | 2743.00 | 1049.80 | 2218.10 | 0.0017 | 0.00169 |
| 258 | 324 | 12.83 | 14.30 | 3589.60 | 846.60 | 3166.30 | 0.0017 | 0.00170 |
| 324 | 390 | 14.30 | 15.56 | 4436.20 | 846.60 | 4012.90 | 0.0015 | 0.00149 |
| 390 | 489 | 15.56 | 16.65 | 5486.00 | 1049.80 | 4961.10 | 0.0010 | 0.00108 |
| 489 | 555 | 16.65 | 17.37 | 6332.60 | 846.60 | 5909.30 | 0.0009 | 0.00101 |
| 555 | 621 | 17.37 | 17.95 | 7179.20 | 846.60 | 6755.90 | 0.0007 | 0.00087 |
| 621 | 720 | 17.95 | 18.36 | 8229.00 | 1049.80 | 7704.10 | 0.0004 | 0.00060 |
| 720 | 786 | 18.36 | 18.62 | 9075.60 | 846.60 | 8652.30 | 0.0003 | 0.00059 |
| 786 | 852 | 18.62 | 18.82 | 9922.20 | 846.60 | 9498.90 | 0.0002 | 0.00052 |
| 852 | 951 | 18.82 | 18.96 | 10972.00 | 1049.80 | 10447.10 | 0.0001 | 0.00040 |
| 951 | 1017 | 18.96 | 19.07 | 11818.60 | 846.60 | 11395.30 | 0.0001 | 0.00042 |
| 1017 | 1083 | 19.07 | 19.15 | 12665.20 | 846.60 | 12241.90 | 0.0001 | 0.00039 |
| 1083 | 1182 | 19.15 | 19.22 | 13715.00 | 1049.80 | 13190.10 | 0.0001 | 0.00032 |
| 1182 | 1248 | 19.22 | 19.27 | 14561.60 | 846.60 | 14138.30 | 0.0001 | 0.00035 |
| 1248 | 1314 | 19.27 | 19.31 | 15408.20 | 846.60 | 14984.90 | 0.0000 | 0.00033 |
| 1314 | 1413 | 19.31 | 19.33 | 16458.00 | 1049.80 | 15933.10 | 0.0000 | 0.00027 |
| 1413 | 1479 | 19.33 | 19.33 | 17304.60 | 846.60 | 16881.30 | 0.0000 | 0.00029 |
| 1479 | 1545 | 19.33 | 19.32 | 18151.20 | 846.60 | 17727.90 | 0.0000 | 0.00028 |
| 1545 | 1644 | 19.32 | 19.30 | 19201.00 | 1049.80 | 18676.10 | 0.0000 | 0.00023 |
| 1644 | 1710 | 19.30 | 19.27 | 20047.60 | 846.60 | 19624.30 | 0.0000 | 0.00025 |
| 1710 | 1776 | 19.27 | 19.23 | 20894.20 | 846.60 | 20470.90 | 0.0000 | 0.00023 |
| 1776 | 1875 | 19.23 | 19.17 | 21944.00 | 1049.80 | 21419.10 | -0.0001 | 0.00019 |
| 1875 | 1941 | 19.17 | 19.10 | 22790.60 | 846.60 | 22367.30 | -0.0001 | 0.00021 |
| 1941 | 2007 | 19.10 | 19.03 | 23637.20 | 846.60 | 23213.90 | -0.0001 | 0.00019 |
| 2007 | 2106 | 19.03 | 18.94 | 24687.00 | 1049.80 | 24162.10 | -0.0001 | 0.00016 |
| 2106 | 2172 | 18.94 | 18.85 | 25533.60 | 846.60 | 25110.30 | -0.0001 | 0.00017 |
| 2172 | 2238 | 18.85 | 18.75 | 26380.20 | 846.60 | 25956.90 | -0.0001 | 0.00016 |
| 2238 | 2337 | 18.75 | 18.63 | 27430.00 | 1049.80 | 26905.10 | -0.0001 | 0.00013 |
| 33 | 99 | 0.00 | -2.77 | 846.60 | 846.60 | 423.30 | -0.0033 | 0.00581 |
| 99 | 165 | -2.77 | -4.76 | 1693.20 | 846.60 | 1269.90 | -0.0024 | 0.00399 |
| 165 | 264 | -4.76 | -6.36 | 2743.00 | 1049.80 | 2218.10 | -0.0015 | 0.00169 |
| 264 | 330 | -6.36 | -7.64 | 3589.60 | 846.60 | 3166.30 | -0.0015 | 0.00170 |
| 330 | 396 | -7.64 | -8.77 | 4436.20 | 846.60 | 4012.90 | -0.0013 | 0.00149 |
| 396 | 495 | -8.77 | -9.85 | 5486.00 | 1049.80 | 4961.10 | -0.0010 | 0.00108 |
| 495 | 561 | -9.85 | -10.76 | 6332.60 | 846.60 | 5909.30 | -0.0011 | 0.00101 |
| 561 | 627 | -10.76 | -11.59 | 7179.20 | 846.60 | 6755.90 | -0.0010 | 0.00087 |
| 627 | 726 | -11.59 | -12.39 | 8229.00 | 1049.80 | 7704.10 | -0.0008 | 0.00060 |
| 726 | 792 | -12.39 | -13.08 | 9075.60 | 846.60 | 8652.30 | -0.0008 | 0.00059 |
| 792 | 858 | -13.08 | -13.72 | 9922.20 | 846.60 | 9498.90 | -0.0008 | 0.00052 |
| 858 | 957 | -13.72 | -14.36 | 10972.00 | 1049.80 | 10447.10 | -0.0006 | 0.00040 |
| 957 | 1023 | -14.36 | -14.93 | 11818.60 | 846.60 | 11395.30 | -0.0007 | 0.00042 |
| 1023 | 1089 | -14.93 | -15.48 | 12665.20 | 846.60 | 12241.90 | -0.0006 | 0.00039 |
| 1089 | 1188 | -15.48 | -16.05 | 13715.00 | 1049.80 | 13190.10 | -0.0005 | 0.00032 |
| 1188 | 1254 | -16.05 | -16.56 | 14561.60 | 846.60 | 14138.30 | -0.0006 | 0.00035 |
| 1254 | 1320 | -16.56 | -17.06 | 15408.20 | 846.60 | 14984.90 | -0.0006 | 0.00033 |
| 1320 | 1419 | -17.06 | -17.57 | 16458.00 | 1049.80 | 15933.10 | -0.0005 | 0.00027 |


| Node |  | Node 1 ydispl (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | $\begin{aligned} & \text { Difference } \\ & (\mathrm{mm}) \end{aligned}$ | Height (mm) | Strain | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1419 | 1485 | -17.57 | -18.04 | 17304.60 | 846.60 | 16881.30 | -0.0006 | 0.00029 |
| 1485 | 1551 | -18.04 | -18.50 | 18151.20 | 846.60 | 17727.90 | -0.0005 | 0.00028 |
| 1551 | 1650 | -18.50 | -18.98 | 19201.00 | 1049.80 | 18676.10 | -0.0005 | 0.00023 |
| 1650 | 1716 | -18.98 | -19.41 | 20047.60 | 846.60 | 19624.30 | -0.0005 | 0.00025 |
| 1716 | 1782 | -19.41 | -19.83 | 20894.20 | 846.60 | 20470.90 | -0.0005 | 0.00023 |
| 1782 | 1881 | -19.83 | -20.27 | 21944.00 | 1049.80 | 21419.10 | -0.0004 | 0.00019 |
| 1881 | 1947 | -20.27 | -20.67 | 22790.60 | 846.60 | 22367.30 | -0.0005 | 0.00021 |
| 1947 | 2013 | -20.67 | -21.05 | 23637.20 | 846.60 | 23213.90 | -0.0005 | 0.00019 |
| 2013 | 2112 | -21.05 | -21.46 | 24687.00 | 1049.80 | 24162.10 | -0.0004 | 0.00016 |
| 2112 | 2178 | -21.46 | -21.83 | 25533.60 | 846.60 | 25110.30 | -0.0004 | 0.00017 |
| 2178 | 2244 | -21.83 | -22.19 | 26380.20 | 846.60 | 25956.90 | -0.0004 | 0.00016 |
| 2244 | 2343 | -22.19 | -22.57 | 27430.00 | 1049.80 | 26905.10 | -0.0004 | 0.00013 |

Wall length: 1905 mm

Table C. 50 Interpolation of curvatures at storey heights of Wall $\mathbf{3}$ combined with Wall 1 for a drift of $\mathbf{2 \%}$

| Storey | Height <br> (mm) | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: |
|  | 423.30 | 0.00581 |
|  | 1269.90 | 0.00399 |
|  | 2218.10 | 0.00169 |
| 1 | 2743.00 | 0.00170 |
|  | 3166.30 | 0.00170 |
|  | 4012.90 | 0.00149 |
|  | 4961.10 | 0.00108 |
| 2 | 5486.00 | 0.00104 |
|  | 5909.30 | 0.00101 |
|  | 6755.90 | 0.00087 |
|  | 7704.10 | 0.00060 |
| 3 | 8229.00 | 0.00059 |
|  | 8652.30 | 0.00059 |
|  | 9498.90 | 0.00052 |
|  | 10447.10 | 0.00040 |
| 4 | 10972.00 | 0.00041 |
|  | 11395.30 | 0.00042 |
|  | 12241.90 | 0.00039 |
|  | 13190.10 | 0.00032 |
| 5 | 13715.00 | 0.00034 |
|  | 14138.30 | 0.00035 |
|  | 14984.90 | 0.00033 |
|  | 15933.10 | 0.00027 |
| 6 | 16458.00 | 0.00028 |
|  | 16881.30 | 0.00029 |
|  | 17727.90 | 0.00028 |
|  | 18676.10 | 0.00023 |
| 7 | 19201.00 | 0.00024 |
|  | 19624.30 | 0.00025 |
|  | 20470.90 | 0.00023 |
|  | 21419.10 | 0.00019 |
| 8 | 21944.00 | 0.00020 |
|  | 22367.30 | 0.00021 |
|  | 23213.90 | 0.00019 |
|  | 24162.10 | 0.00016 |
| 9 | 24687.00 | 0.00017 |
|  | 25110.30 | 0.00017 |
|  | 25956.90 | 0.00016 |
|  | 26905.10 | 0.00013 |
| 10 | 27430.00 | 0.00011 |

Table C. 51 Bending moments along the height of Wall $\mathbf{3}$ combined with Wall 1 for a drift of $\mathbf{2} \%$

| Storey | Height (m) | Force (kN) | Shear (kN) | Moment incr. (kN.m) | Moment (kN.m) |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 0 | 0.00 | 0.0 | 632.3 | 267.7 | 9621.2 |
|  | 0.42 | 0.0 | 632.3 | 535.3 | 9353.6 |
|  | 1.27 | 0.0 | 632.3 | 599.5 | 8818.2 |
|  | 2.22 | 0.0 | 632.3 | 331.9 | 8218.7 |
| 1 | 2.74 | 70.3 | 632.3 | 252.8 | 7886.8 |
|  | 3.17 | 0.0 | 562.0 | 475.8 | 7634.0 |
|  | 4.01 | 0.0 | 562.0 | 532.9 | 7158.2 |
|  | 4.96 | 0.0 | 562.0 | 295.0 | 6625.4 |
| 2 | 5.49 | 58.8 | 562.0 | 225.4 | 6330.4 |
|  | 5.91 | 0.0 | 503.2 | 426.0 | 6104.9 |
|  | 6.76 | 0.0 | 503.2 | 477.1 | 5678.9 |
|  | 7.70 | 0.0 | 503.2 | 264.1 | 5201.8 |
| 3 | 8.23 | 155.3 | 503.2 | 180.1 | 4937.6 |
|  | 8.65 | 0.0 | 347.9 | 294.5 | 4757.5 |
|  | 9.50 | 0.0 | 347.9 | 329.9 | 4463.0 |
|  | 10.45 | 0.0 | 347.9 | 182.6 | 4133.1 |
| 4 | 10.97 | 114.7 | 347.9 | 123.0 | 3950.5 |
|  | 11.40 | 0.0 | 233.2 | 197.4 | 3827.5 |
|  | 12.24 | 0.0 | 233.2 | 221.1 | 3630.1 |
|  | 13.19 | 0.0 | 233.2 | 122.4 | 3408.9 |
| 5 | 13.72 | 62.5 | 233.2 | 85.5 | 3286.5 |
|  | 14.14 | 0.0 | 170.7 | 144.5 | 3201.1 |
|  | 14.98 | 0.0 | 170.7 | 161.9 | 3056.5 |
|  | 15.93 | 0.0 | 170.7 | 89.6 | 2894.7 |
| 6 | 16.46 | 19.8 | 170.7 | 68.1 | 2805.1 |
|  | 16.88 | 0.0 | 150.9 | 127.8 | 2737.0 |
|  | 17.73 | 0.0 | 150.9 | 143.1 | 2609.3 |
|  | 18.68 | 0.0 | 150.9 | 79.2 | 2466.2 |
| 7 | 19.20 | 9.7 | 150.9 | 61.8 | 2387.0 |
|  | 19.62 | 0.0 | 141.2 | 119.5 | 2325.1 |
|  | 20.47 | 0.0 | 141.2 | 133.9 | 2205.6 |
|  | 21.42 | 0.0 | 141.2 | 74.1 | 2071.7 |
| 8 | 21.94 | 15.1 | 141.2 | 56.6 | 1997.6 |
|  | 22.37 | 0.0 | 126.1 | 106.8 | 1941.0 |
|  | 23.21 | 0.0 | 126.1 | 119.6 | 1834.3 |
|  | 24.16 | 0.0 | 126.1 | 66.2 | 1714.7 |
| 9 | 24.69 | 24.5 | 126.1 | 48.2 | 1648.5 |
|  | 25.11 | 0.0 | 101.6 | 86.0 | 1600.3 |
|  | 25.96 | 0.0 | 101.6 | 96.3 | 1514.3 |
|  | 26.91 | 0.0 | 101.6 | 53.3 | 1418.0 |
| 10 | 27.43 | -30.9 | 101.6 | 321.1 | 1364.6 |
| 11 | 30.17 | 51.3 | 132.5 | 293.1 | 1043.6 |
| 12 | 32.92 | 27.9 | 81.2 | 184.5 | 750.5 |
| 13 | 35.66 | 13.9 | 53.3 | 127.1 | 566.0 |
| 14 | 38.40 | 4.9 | 39.4 | 101.4 | 438.9 |
| 15 | 41.15 | 2.5 | 34.5 | 91.2 | 337.5 |
| 16 | 43.89 | -3.0 | 32.0 | 91.9 | 246.3 |
| 17 | 46.63 | 12.8 | 35.0 | 78.4 | 154.4 |
| 18 | 49.37 | 55.0 | 22.2 | -14.5 | 76.0 |
| 19 | 52.12 | -131.6 | -32.8 | 90.5 | 90.5 |
| 20 | 54.86 | 98.8 | 98.8 | 0.0 | 0.0 |

Table C. 52 Curvatures up to the mid-height of Column 1 for a drift of $\mathbf{2 \%}$

| Node |  | Node 1 ydispl (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | Difference (mm) | Height (mm) | Strain | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1 | 9 | 0.00 | 0.01 | 846.60 | 846.60 | 423.30 | 0.0000 | 0.00113 |
| 9 | 17 | 0.01 | 0.00 | 1693.20 | 846.60 | 1269.90 | 0.0000 | 0.00109 |
| 17 | 29 | 0.00 | 0.00 | 2743.00 | 1049.80 | 2218.10 | 0.0000 | 0.00109 |
| 29 | 37 | 0.00 | -0.02 | 3589.60 | 846.60 | 3166.30 | 0.0000 | 0.00105 |
| 37 | 45 | -0.02 | -0.05 | 4436.20 | 846.60 | 4012.90 | 0.0000 | 0.00104 |
| 45 | 57 | -0.05 | -0.08 | 5486.00 | 1049.80 | 4961.10 | 0.0000 | 0.00103 |
| 57 | 65 | -0.08 | -0.13 | 6332.60 | 846.60 | 5909.30 | 0.0000 | 0.00099 |
| 65 | 73 | -0.13 | -0.17 | 7179.20 | 846.60 | 6755.90 | -0.0001 | 0.00098 |
| 73 | 85 | -0.17 | -0.24 | 8229.00 | 1049.80 | 7704.10 | -0.0001 | 0.00097 |
| 85 | 93 | -0.24 | -0.30 | 9075.60 | 846.60 | 8652.30 | -0.0001 | 0.00094 |
| 93 | 101 | -0.30 | -0.37 | 9922.20 | 846.60 | 9498.90 | -0.0001 | 0.00092 |
| 101 | 113 | -0.37 | -0.46 | 10972.00 | 1049.80 | 10447.10 | -0.0001 | 0.00092 |
| 113 | 121 | -0.46 | -0.54 | 11818.60 | 846.60 | 11395.30 | -0.0001 | 0.00088 |
| 121 | 129 | -0.54 | -0.63 | 12665.20 | 846.60 | 12241.90 | -0.0001 | 0.00087 |
| 129 | 141 | -0.63 | -0.75 | 13715.00 | 1049.80 | 13190.10 | -0.0001 | 0.00086 |
| 141 | 149 | -0.75 | -0.86 | 14561.60 | 846.60 | 14138.30 | -0.0001 | 0.00082 |
| 149 | 157 | -0.86 | -0.97 | 15408.20 | 846.60 | 14984.90 | -0.0001 | 0.00081 |
| 157 | 169 | -0.97 | -1.11 | 16458.00 | 1049.80 | 15933.10 | -0.0001 | 0.00080 |
| 169 | 177 | -1.11 | -1.24 | 17304.60 | 846.60 | 16881.30 | -0.0002 | 0.00077 |
| 177 | 185 | -1.24 | -1.38 | 18151.20 | 846.60 | 17727.90 | -0.0002 | 0.00075 |
| 185 | 197 | -1.38 | -1.55 | 19201.00 | 1049.80 | 18676.10 | -0.0002 | 0.00074 |
| 197 | 205 | -1.55 | -1.70 | 20047.60 | 846.60 | 19624.30 | -0.0002 | 0.00071 |
| 205 | 213 | -1.70 | -1.86 | 20894.20 | 846.60 | 20470.90 | -0.0002 | 0.00069 |
| 213 | 225 | -1.86 | -2.06 | 21944.00 | 1049.80 | 21419.10 | -0.0002 | 0.00069 |
| 225 | 233 | -2.06 | -2.23 | 22790.60 | 846.60 | 22367.30 | -0.0002 | 0.00066 |
| 233 | 241 | -2.23 | -2.41 | 23637.20 | 846.60 | 23213.90 | -0.0002 | 0.00064 |
| 241 | 253 | -2.41 | -2.64 | 24687.00 | 1049.80 | 24162.10 | -0.0002 | 0.00063 |
| 253 | 261 | -2.64 | -2.83 | 25533.60 | 846.60 | 25110.30 | -0.0002 | 0.00060 |
| 261 | 269 | -2.83 | -3.03 | 26380.20 | 846.60 | 25956.90 | -0.0002 | 0.00059 |
| 269 | 281 | -3.03 | -3.28 | 27430.00 | 1049.80 | 26905.10 | -0.0002 | 0.00058 |
| 4 | 12 | 0.00 | -0.90 | 846.60 | 846.60 | 423.30 | -0.0011 | 0.00113 |
| 12 | 20 | -0.90 | -1.79 | 1693.20 | 846.60 | 1269.90 | -0.0010 | 0.00109 |
| 20 | 32 | -1.79 | -2.89 | 2743.00 | 1049.80 | 2218.10 | -0.0010 | 0.00109 |
| 32 | 40 | -2.89 | -3.76 | 3589.60 | 846.60 | 3166.30 | -0.0010 | 0.00105 |
| 40 | 48 | -3.76 | -4.62 | 4436.20 | 846.60 | 4012.90 | -0.0010 | 0.00104 |
| 48 | 60 | -4.62 | -5.69 | 5486.00 | 1049.80 | 4961.10 | -0.0010 | 0.00103 |
| 60 | 68 | -5.69 | -6.53 | 6332.60 | 846.60 | 5909.30 | -0.0010 | 0.00099 |
| 68 | 76 | -6.53 | -7.37 | 7179.20 | 846.60 | 6755.90 | -0.0010 | 0.00098 |
| 76 | 88 | -7.37 | -8.40 | 8229.00 | 1049.80 | 7704.10 | -0.0010 | 0.00097 |
| 88 | 96 | -8.40 | -9.22 | 9075.60 | 846.60 | 8652.30 | -0.0010 | 0.00094 |
| 96 | 104 | -9.22 | -10.04 | 9922.20 | 846.60 | 9498.90 | -0.0010 | 0.00092 |
| 104 | 116 | -10.04 | -11.04 | 10972.00 | 1049.80 | 10447.10 | -0.0010 | 0.00092 |
| 116 | 124 | -11.04 | -11.84 | 11818.60 | 846.60 | 11395.30 | -0.0009 | 0.00088 |
| 124 | 132 | -11.84 | -12.63 | 12665.20 | 846.60 | 12241.90 | -0.0009 | 0.00087 |
| 132 | 144 | -12.63 | -13.60 | 13715.00 | 1049.80 | 13190.10 | -0.0009 | 0.00086 |
| 144 | 152 | -13.60 | -14.37 | 14561.60 | 846.60 | 14138.30 | -0.0009 | 0.00082 |
| 152 | 160 | -14.37 | -15.14 | 15408.20 | 846.60 | 14984.90 | -0.0009 | 0.00081 |
| 160 | 172 | -15.14 | -16.08 | 16458.00 | 1049.80 | 15933.10 | -0.0009 | 0.00080 |


| Node |  | Node 1 y- <br> displ (mm) | Node 2 y- <br> displ (mm) | Node 2 y- <br> coord (mm) | Difference <br> $(\mathrm{mm})$ | Height <br> $(\mathrm{mm})$ | Strain | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 172 | 180 | -16.08 | -16.83 | 17304.60 | 846.60 | 16881.30 | -0.0009 | 0.00077 |
| 180 | 188 | -16.83 | -17.57 | 18151.20 | 846.60 | 17727.90 | -0.0009 | 0.00075 |
| 188 | 200 | -17.57 | -18.48 | 19201.00 | 1049.80 | 18676.10 | -0.0009 | 0.00074 |
| 200 | 208 | -18.48 | -19.21 | 20047.60 | 846.60 | 19624.30 | -0.0009 | 0.00071 |
| 208 | 216 | -19.21 | -19.92 | 20894.20 | 846.60 | 20470.90 | -0.0008 | 0.00069 |
| 216 | 228 | -19.92 | -20.81 | 21944.00 | 1049.80 | 21419.10 | -0.0008 | 0.00069 |
| 228 | 236 | -20.81 | -21.51 | 22790.60 | 846.60 | 22367.30 | -0.0008 | 0.00066 |
| 236 | 244 | -21.51 | -22.20 | 23637.20 | 846.60 | 23213.90 | -0.0008 | 0.00064 |
| 244 | 256 | -22.20 | -23.06 | 24687.00 | 1049.80 | 24162.10 | -0.0008 | 0.00063 |
| 256 | 264 | -23.06 | -23.74 | 25533.60 | 846.60 | 25110.30 | -0.0008 | 0.00060 |
| 264 | 272 | -23.74 | -24.41 | 26380.20 | 846.60 | 25956.90 | -0.0008 | 0.00059 |
| 272 | 284 | -24.41 | -25.24 | 27430.00 | 1049.80 | 26905.10 | -0.0008 | 0.00058 |

Col. length: $\quad 952.5 \mathrm{~mm}$

Table C. 53 Curvatures up to the mid-height of Wall 1 combined with Column 1 for a drift of $\mathbf{2 \%}$

| Node |  | Node 1 ydispl (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | $\begin{aligned} & \text { Difference } \\ & (\mathrm{mm}) \end{aligned}$ | Height (mm) | Strain | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1 | 61 | 0.00 | 15.45 | 846.60 | 846.60 | 423.30 | 0.0182 | 0.00316 |
| 61 | 121 | 15.45 | 30.84 | 1693.20 | 846.60 | 1269.90 | 0.0182 | 0.00315 |
| 121 | 211 | 30.84 | 47.94 | 2743.00 | 1049.80 | 2218.10 | 0.0163 | 0.00253 |
| 211 | 271 | 47.94 | 60.94 | 3589.60 | 846.60 | 3166.30 | 0.0154 | 0.00237 |
| 271 | 331 | 60.94 | 72.75 | 4436.20 | 846.60 | 4012.90 | 0.0140 | 0.00213 |
| 331 | 421 | 72.75 | 83.68 | 5486.00 | 1049.80 | 4961.10 | 0.0104 | 0.00160 |
| 421 | 481 | 83.68 | 91.00 | 6332.60 | 846.60 | 5909.30 | 0.0086 | 0.00137 |
| 481 | 541 | 91.00 | 97.35 | 7179.20 | 846.60 | 6755.90 | 0.0075 | 0.00120 |
| 541 | 631 | 97.35 | 100.38 | 8229.00 | 1049.80 | 7704.10 | 0.0029 | 0.00055 |
| 631 | 691 | 100.38 | 102.43 | 9075.60 | 846.60 | 8652.30 | 0.0024 | 0.00050 |
| 691 | 751 | 102.43 | 104.19 | 9922.20 | 846.60 | 9498.90 | 0.0021 | 0.00044 |
| 751 | 841 | 104.19 | 105.72 | 10972.00 | 1049.80 | 10447.10 | 0.0015 | 0.00033 |
| 841 | 901 | 105.72 | 107.38 | 11818.60 | 846.60 | 11395.30 | 0.0020 | 0.00041 |
| 901 | 961 | 107.38 | 108.89 | 12665.20 | 846.60 | 12241.90 | 0.0018 | 0.00037 |
| 961 | 1051 | 108.89 | 110.33 | 13715.00 | 1049.80 | 13190.10 | 0.0014 | 0.00030 |
| 1051 | 1111 | 110.33 | 111.76 | 14561.60 | 846.60 | 14138.30 | 0.0017 | 0.00035 |
| 1111 | 1171 | 111.76 | 113.12 | 15408.20 | 846.60 | 14984.90 | 0.0016 | 0.00033 |
| 1171 | 1261 | 113.12 | 114.38 | 16458.00 | 1049.80 | 15933.10 | 0.0012 | 0.00026 |
| 1261 | 1321 | 114.38 | 115.62 | 17304.60 | 846.60 | 16881.30 | 0.0015 | 0.00031 |
| 1321 | 1381 | 115.62 | 116.77 | 18151.20 | 846.60 | 17727.90 | 0.0014 | 0.00029 |
| 1381 | 1471 | 116.77 | 117.82 | 19201.00 | 1049.80 | 18676.10 | 0.0010 | 0.00023 |
| 1471 | 1531 | 117.82 | 118.84 | 20047.60 | 846.60 | 19624.30 | 0.0012 | 0.00026 |
| 1531 | 1591 | 118.84 | 119.78 | 20894.20 | 846.60 | 20470.90 | 0.0011 | 0.00025 |
| 1591 | 1681 | 119.78 | 120.63 | 21944.00 | 1049.80 | 21419.10 | 0.0008 | 0.00019 |
| 1681 | 1741 | 120.63 | 121.44 | 22790.60 | 846.60 | 22367.30 | 0.0010 | 0.00022 |
| 1741 | 1801 | 121.44 | 122.16 | 23637.20 | 846.60 | 23213.90 | 0.0009 | 0.00020 |
| 1801 | 1891 | 122.16 | 122.82 | 24687.00 | 1049.80 | 24162.10 | 0.0006 | 0.00016 |
| 1891 | 1951 | 122.82 | 123.41 | 25533.60 | 846.60 | 25110.30 | 0.0007 | 0.00018 |
| 1951 | 2011 | 123.41 | 123.95 | 26380.20 | 846.60 | 25956.90 | 0.0006 | 0.00016 |
| 2011 | 2101 | 123.95 | 124.42 | 27430.00 | 1049.80 | 26905.10 | 0.0005 | 0.00013 |
| 26 | 86 | 0.00 | -4.96 | 846.60 | 846.60 | 423.30 | -0.0059 | 0.00316 |
| 86 | 146 | -4.96 | -9.92 | 1693.20 | 846.60 | 1269.90 | -0.0059 | 0.00315 |
| 146 | 236 | -9.92 | -13.06 | 2743.00 | 1049.80 | 2218.10 | -0.0030 | 0.00253 |
| 236 | 296 | -13.06 | -15.36 | 3589.60 | 846.60 | 3166.30 | -0.0027 | 0.00237 |
| 296 | 356 | -15.36 | -17.29 | 4436.20 | 846.60 | 4012.90 | -0.0023 | 0.00213 |
| 356 | 446 | -17.29 | -19.12 | 5486.00 | 1049.80 | 4961.10 | -0.0017 | 0.00160 |
| 446 | 506 | -19.12 | -20.67 | 6332.60 | 846.60 | 5909.30 | -0.0018 | 0.00137 |
| 506 | 566 | -20.67 | -22.02 | 7179.20 | 846.60 | 6755.90 | -0.0016 | 0.00120 |
| 566 | 656 | -22.02 | -23.36 | 8229.00 | 1049.80 | 7704.10 | -0.0013 | 0.00055 |
| 656 | 716 | -23.36 | -24.52 | 9075.60 | 846.60 | 8652.30 | -0.0014 | 0.00050 |
| 716 | 776 | -24.52 | -25.57 | 9922.20 | 846.60 | 9498.90 | -0.0012 | 0.00044 |
| 776 | 866 | -25.57 | -26.64 | 10972.00 | 1049.80 | 10447.10 | -0.0010 | 0.00033 |
| 866 | 926 | -26.64 | -27.60 | 11818.60 | 846.60 | 11395.30 | -0.0011 | 0.00041 |
| 926 | 986 | -27.60 | -28.50 | 12665.20 | 846.60 | 12241.90 | -0.0011 | 0.00037 |
| 986 | 1076 | -28.50 | -29.45 | 13715.00 | 1049.80 | 13190.10 | -0.0009 | 0.00030 |
| 1076 | 1136 | -29.45 | -30.30 | 14561.60 | 846.60 | 14138.30 | -0.0010 | 0.00035 |
| 1136 | 1196 | -30.30 | -31.10 | 15408.20 | 846.60 | 14984.90 | -0.0009 | 0.00033 |
| 1196 | 1286 | -31.10 | -31.95 | 16458.00 | 1049.80 | 15933.10 | -0.0008 | 0.00026 |


| Node |  | Node 1 y displ (mm) | Node 2 y displ (mm) | Node 2 y coord (mm) | Difference(mm) | Height (mm) | Strain | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 1286 | 1346 | -31.95 | -32.71 | 17304.60 | 846.60 | 16881.30 | -0.0009 | 0.00031 |
| 1346 | 1406 | -32.71 | -33.43 | 18151.20 | 846.60 | 17727.90 | -0.0009 | 0.00029 |
| 1406 | 1496 | -33.43 | -34.19 | 19201.00 | 1049.80 | 18676.10 | -0.0007 | 0.00023 |
| 1496 | 1556 | -34.19 | -34.88 | 20047.60 | 846.60 | 19624.30 | -0.0008 | 0.00026 |
| 1556 | 1616 | -34.88 | -35.53 | 20894.20 | 846.60 | 20470.90 | -0.0008 | 0.00025 |
| 1616 | 1706 | -35.53 | -36.21 | 21944.00 | 1049.80 | 21419.10 | -0.0006 | 0.00019 |
| 1706 | 1766 | -36.21 | -36.82 | 22790.60 | 846.60 | 22367.30 | -0.0007 | 0.00022 |
| 1766 | 1826 | -36.82 | -37.40 | 23637.20 | 846.60 | 23213.90 | -0.0007 | 0.00020 |
| 1826 | 1916 | -37.40 | -38.00 | 24687.00 | 1049.80 | 24162.10 | -0.0006 | 0.00016 |
| 1916 | 1976 | -38.00 | -38.55 | 25533.60 | 846.60 | 25110.30 | -0.0006 | 0.00018 |
| 1976 | 2036 | -38.55 | -39.06 | 26380.20 | 846.60 | 25956.90 | -0.0006 | 0.00016 |
| 2036 | 2126 | -39.06 | -39.60 | 27430.00 | 1049.80 | 26905.10 | -0.0005 | 0.00013 |

Wall length: $\quad 7620 \mathrm{~mm}$

Table C. 54 Curvatures up to the mid-height of Column 1 combined with Wall 1 for a drift of 2\%

| Node |  | Node 1 ydispl (mm) | Node 2 y displ (mm) | Node 2 ycoord (mm) | $\begin{aligned} & \text { Difference } \\ & (\mathrm{mm}) \end{aligned}$ | Height (mm) | Strain | Curvature ( $\mathrm{rad} / \mathrm{m}$ ) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second |  |  |  |  |  |  |  |
| 27 | 87 | 0.00 | 1.91 | 846.60 | 846.60 | 423.30 | 0.0023 | 0.00550 |
| 87 | 147 | 1.91 | 3.31 | 1693.20 | 846.60 | 1269.90 | 0.0017 | 0.00404 |
| 147 | 237 | 3.31 | 4.16 | 2743.00 | 1049.80 | 2218.10 | 0.0008 | 0.00238 |
| 237 | 297 | 4.16 | 4.37 | 3589.60 | 846.60 | 3166.30 | 0.0002 | 0.00164 |
| 297 | 357 | 4.37 | 4.49 | 4436.20 | 846.60 | 4012.90 | 0.0001 | 0.00141 |
| 357 | 447 | 4.49 | 4.52 | 5486.00 | 1049.80 | 4961.10 | 0.0000 | 0.00102 |
| 447 | 507 | 4.52 | 4.47 | 6332.60 | 846.60 | 5909.30 | -0.0001 | 0.00097 |
| 507 | 567 | 4.47 | 4.37 | 7179.20 | 846.60 | 6755.90 | -0.0001 | 0.00083 |
| 567 | 657 | 4.37 | 4.19 | 8229.00 | 1049.80 | 7704.10 | -0.0002 | 0.00059 |
| 657 | 717 | 4.19 | 3.98 | 9075.60 | 846.60 | 8652.30 | -0.0002 | 0.00055 |
| 717 | 777 | 3.98 | 3.75 | 9922.20 | 846.60 | 9498.90 | -0.0003 | 0.00050 |
| 777 | 867 | 3.75 | 3.48 | 10972.00 | 1049.80 | 10447.10 | -0.0003 | 0.00039 |
| 867 | 927 | 3.48 | 3.21 | 11818.60 | 846.60 | 11395.30 | -0.0003 | 0.00040 |
| 927 | 987 | 3.21 | 2.93 | 12665.20 | 846.60 | 12241.90 | -0.0003 | 0.00038 |
| 987 | 1077 | 2.93 | 2.63 | 13715.00 | 1049.80 | 13190.10 | -0.0003 | 0.00031 |
| 1077 | 1137 | 2.63 | 2.33 | 14561.60 | 846.60 | 14138.30 | -0.0004 | 0.00033 |
| 1137 | 1197 | 2.33 | 2.03 | 15408.20 | 846.60 | 14984.90 | -0.0004 | 0.00032 |
| 1197 | 1287 | 2.03 | 1.69 | 16458.00 | 1049.80 | 15933.10 | -0.0003 | 0.00026 |
| 1287 | 1347 | 1.69 | 1.38 | 17304.60 | 846.60 | 16881.30 | -0.0004 | 0.00028 |
| 1347 | 1407 | 1.38 | 1.05 | 18151.20 | 846.60 | 17727.90 | -0.0004 | 0.00027 |
| 1407 | 1497 | 1.05 | 0.70 | 19201.00 | 1049.80 | 18676.10 | -0.0003 | 0.00022 |
| 1497 | 1557 | 0.70 | 0.37 | 20047.60 | 846.60 | 19624.30 | -0.0004 | 0.00023 |
| 1557 | 1617 | 0.37 | 0.03 | 20894.20 | 846.60 | 20470.90 | -0.0004 | 0.00022 |
| 1617 | 1707 | 0.03 | -0.34 | 21944.00 | 1049.80 | 21419.10 | -0.0004 | 0.00018 |
| 1707 | 1767 | -0.34 | -0.69 | 22790.60 | 846.60 | 22367.30 | -0.0004 | 0.00019 |
| 1767 | 1827 | -0.69 | -1.05 | 23637.20 | 846.60 | 23213.90 | -0.0004 | 0.00018 |
| 1827 | 1917 | -1.05 | -1.44 | 24687.00 | 1049.80 | 24162.10 | -0.0004 | 0.00015 |
| 1917 | 1977 | -1.44 | -1.80 | 25533.60 | 846.60 | 25110.30 | -0.0004 | 0.00016 |
| 1977 | 2037 | -1.80 | -2.17 | 26380.20 | 846.60 | 25956.90 | -0.0004 | 0.00016 |
| 2037 | 2127 | -2.17 | -2.56 | 27430.00 | 1049.80 | 26905.10 | -0.0004 | 0.00014 |
| 30 | 90 | 0.00 | -2.53 | 846.60 | 846.60 | 423.30 | -0.0030 | 0.00550 |
| 90 | 150 | -2.53 | -4.38 | 1693.20 | 846.60 | 1269.90 | -0.0022 | 0.00404 |
| 150 | 240 | -4.38 | -5.91 | 2743.00 | 1049.80 | 2218.10 | -0.0015 | 0.00238 |
| 240 | 300 | -5.91 | -7.03 | 3589.60 | 846.60 | 3166.30 | -0.0013 | 0.00164 |
| 300 | 360 | -7.03 | -8.04 | 4436.20 | 846.60 | 4012.90 | -0.0012 | 0.00141 |
| 360 | 450 | -8.04 | -9.03 | 5486.00 | 1049.80 | 4961.10 | -0.0009 | 0.00102 |
| 450 | 510 | -9.03 | -9.86 | 6332.60 | 846.60 | 5909.30 | -0.0010 | 0.00097 |
| 510 | 570 | -9.86 | -10.63 | 7179.20 | 846.60 | 6755.90 | -0.0009 | 0.00083 |
| 570 | 660 | -10.63 | -11.39 | 8229.00 | 1049.80 | 7704.10 | -0.0007 | 0.00059 |
| 660 | 720 | -11.39 | -12.05 | 9075.60 | 846.60 | 8652.30 | -0.0008 | 0.00055 |
| 720 | 780 | -12.05 | -12.68 | 9922.20 | 846.60 | 9498.90 | -0.0007 | 0.00050 |
| 780 | 870 | -12.68 | -13.34 | 10972.00 | 1049.80 | 10447.10 | -0.0006 | 0.00039 |
| 870 | 930 | -13.34 | -13.93 | 11818.60 | 846.60 | 11395.30 | -0.0007 | 0.00040 |
| 930 | 990 | -13.93 | -14.51 | 12665.20 | 846.60 | 12241.90 | -0.0007 | 0.00038 |
| 990 | 1080 | -14.51 | -15.13 | 13715.00 | 1049.80 | 13190.10 | -0.0006 | 0.00031 |
| 1080 | 1140 | -15.13 | -15.69 | 14561.60 | 846.60 | 14138.30 | -0.0007 | 0.00033 |
| 1140 | 1200 | -15.69 | -16.25 | 15408.20 | 846.60 | 14984.90 | -0.0007 | 0.00032 |
| 1200 | 1290 | -16.25 | -16.84 | 16458.00 | 1049.80 | 15933.10 | -0.0006 | 0.00026 |


| Node |  | Node 1 y- <br> displ (mm) | Node 2 y- <br> displ (mm) | Node 2 y- <br> coord (mm) | Difference <br> $(\mathrm{mm})$ | Height <br> $(\mathrm{mm})$ | Strain | Curvature <br> $(\mathrm{rad} / \mathrm{m})$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| First | Second | -16.84 | -17.38 | 17304.60 | 846.60 | 16881.30 | -0.0006 | 0.00028 |
| 1290 | 1350 | -17.38 | -17.92 | 18151.20 | 846.60 | 17727.90 | -0.0006 | 0.00027 |
| 1350 | 1410 | -17.92 | -18.49 | 19201.00 | 1049.80 | 18676.10 | -0.0005 | 0.00022 |
| 1410 | 1500 | -17.92 | -19.01 | 20047.60 | 846.60 | 19624.30 | -0.0006 | 0.00023 |
| 1500 | 1560 | -18.49 | -19.53 | 20894.20 | 846.60 | 20470.90 | -0.0006 | 0.00022 |
| 1560 | 1620 | -19.01 | -19.53 | -20.08 | 21944.00 | 1049.80 | 21419.10 | -0.0005 |
| 1620 | 1710 | -20.00018 |  |  |  |  |  |  |
| 1710 | 1770 | -20.08 | -20.59 | 22790.60 | 846.60 | 22367.30 | -0.0006 | 0.00019 |
| 1770 | 1830 | -20.59 | -21.09 | 23637.20 | 846.60 | 23213.90 | -0.0006 | 0.00018 |
| 1830 | 1920 | -21.09 | -21.63 | 24687.00 | 1049.80 | 24162.10 | -0.0005 | 0.00015 |
| 1920 | 1980 | -21.63 | -22.12 | 25533.60 | 846.60 | 25110.30 | -0.0006 | 0.00016 |
| 1980 | 2040 | -22.12 | -22.61 | 26380.20 | 846.60 | 25956.90 | -0.0006 | 0.00016 |
| 2040 | 2130 | -22.61 | -23.14 | 27430.00 | 1049.80 | 26905.10 | -0.0005 | 0.00014 |

Col. length: 952.5 mm

## UBC Civil Engineering Grad Students, 2004/2005

(Contains some students who graduated in 2003/04. Many students did not have their photo taken)

## LEGEND: SPECIALIZATION GROUP ID

CM Construction Management
EF Environmental Fluid Mechanics
EQ Earthquake Engineering
G Geotechnical Engineering
GE Environmental Geotechnics

H Hydrotechnical Engineering
M Materials Engineering
PC Pollution Control
S Structural Engineering
T Transportation Engineering


Reem Hared
Abdul-Hafidh
MIng CM


Yapo
Alle-Ando
MASc EF


Alireza
Biparva
MASc M


Christian
Brampton
MIng PC


Ali
Amini Asalemi
PhD G


Cynthia Evelyn
Bluteau
MASc EF


Kelly Lynn
Bush
MASc PC


Parmeshwaree Bahadoorsingh PhD PC


Alfredo Guillemo
Bohr
MASc S


Jessica Erin
Campbell
MASc G

100 to match his preference 245 :ce to match
title pase


OK par domed

