by

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# A THESIS SUBMITTED IN PARTIAL FULFILLMENT OF THE REQUIREMENTS FOR THE DEGREE OF DOCTOR OF PHILOSOPHY <br> in <br> THE FACULTY OF GRADUATE STUDIES <br> Department of Chemical Engineering 

We accept this thesis as conforming to the required standard

THE UNIVERSITY OF BRITISH COLUMBIA

November 1995
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Date April 25,1996


#### Abstract

Kinetics of thermal pyrolysis of both CANMET and Syncrude pitches from heavy oil upgrading have been studied with Thermogravimetric Analysis (TGA), and with Pyroprobe-Gas Chromatography (Pyroprobe-GC). In the latter technique samples are pyrolyzed at high heating rates and products analyzed with in-line gas chromatography.

Experiments with TGA were carried out at atmospheric pressure and at temperatures between 700 and $950{ }^{\circ} \mathrm{C}$. The heating rates were 25,50100 and $150{ }^{\circ} \mathrm{C} / \mathrm{min}$. The sample weight was varied between 3 and 17.2 mg . The effects of sample weight, heating rate and final temperature on the weight loss as a function of time were examined.

Experiments with Pyroprobe-GC were carried out at atmospheric pressure and at temperatures between 500 and $1000{ }^{\circ} \mathrm{C}$. The heating rates were $600,3000,30000,300000$ ${ }^{\circ} \mathrm{C} / \mathrm{min}$, using a sample weight of about 5 mg . The accumulated pyrolysis products were analyzed and lumped into major groups for yield estimation based on number of carbon atoms. The final weight of residue was also determined. The effects of the final temperatures on the yield of each major group were examined.

At temperatures below $150{ }^{\circ} \mathrm{C}$, there is little pyrolysis of either pitch. At higher temperatures, the pyrolysis takes place in two following stages, with a first stage of low activation energy barrier and low pre-exponential factor, and the second stage of higher activation energy and pre-exponential factor. Higher conversion to volatiles was achieved with Syncrude pitch than with CANMET pitch. Heating rates had a minor effect on the weight loss. The total weight loss decreased slightly with the increase of sample weight, and final temperatures.

The most abundant components of the pyrolysis products were species lighter than $\mathbf{C}_{7}$, which are primarily gases. The $\mathrm{C}_{10}$ group yield was strongly influenced by heating rates. Higher molecular weight components $\mathrm{C}_{11}, \mathrm{C}_{12}, \mathrm{C}_{13}$, and $\mathrm{C}_{14}$ were also detected. The pyrolysis products


from Syncrude pitch consisted of higher yields of lighter components $\left(\mathrm{C}_{7}\right)$ than those from CANMET pitch.

A general first order equation for the kinetics of volatile release under temperature programmed conditions is widely used in the pyrolysis literature. Interpretation of results via the single stage integral method and methods due to Coats-Redfern, Chen-Nuttall and Friedman were tested with the TGA data and found inadequate. The single stage first order model of Anthony and Howard, which incorporates a Gaussian distribution of activation energies also failed. An adequate description of the pitch pyrolysis kinetics was achieved using a 2-stage first order model with the integral analysis method. The 2 -stage model reflects changes in the chemical constitution or structure as conversion proceeds using two sets of kinetic parameters. This feature is essential to describe the dependence of devolatilization rates on remaining volatile content. The transition between these two stages is a sharp one, occurring at about $450{ }^{\circ} \mathrm{C}$ for both CANMET and Syncrude pitches. The magnitude of the activation energies suggested that both stages are kinetically controlled. The analysis methods of Coats-Redfern, Chen-Nuttall, and Friedman were also tested as two stage methods and found to be inadequate to describe the pitch pyrolysis kinetics in the temperature range studied.

The pre-exponential factors and activation energies from the different kinetic methods exhibited the compensation effect, in which the values of the derived pre-exponential factors and activation energies are related. This mutual dependence prompted an the examination of the accuracy of these kinetic parameters, and a search for a single set of parameters for each stage of the pitch pyrolysis. It was found that the accuracy of these kinetic parameters derived by different analysis procedures are not identical, and a single set of kinetic parameters for each stage can be obtained with adequate fitting of the experimental data.

Indexing terms: Pitch, Residuum, TGA, Pyroprobe-GC, Pyrolysis, Kinetics, Modeling,
Kinetic Compensation Effect.

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## NOMENCLATURE

a

B
b

C
E, $\mathrm{E}_{\mathrm{o}}$
$\mathrm{E}_{\mathrm{i}}$
$\mathrm{f}(\mathrm{E}) \quad$ distribution function of activation energy
$H_{P T}$
$h_{L V} \quad$ total heat of volatile evaporation per unit mass of coal, $\mathrm{kJ} / \mathrm{kg}$ coal
$\mathrm{k}, \mathrm{k}_{1}, \mathrm{k}_{2}, \mathrm{k}_{3} \quad$ rate constant represented by an Arrhenius expression, $\min ^{-1}$
$\mathrm{k}_{0}, \mathrm{k}_{01}, \mathrm{k}_{02} \quad$ pre-exponential constant of Arrhenius equation, $\min ^{-1}$
$\mathrm{k}_{\mathrm{oi}}$
$\bar{k}$
$\mathrm{m}_{\mathrm{C}}(0) \quad$ initial mass of coal, kg
$\mathrm{m}_{\mathrm{C}} \quad$ mass of coal at any time $\mathrm{t}, \mathrm{kg}$
$\mathrm{m}_{\mathrm{C}}$ (final) mass of coal at the end of pyrolysis reaction, kg
$m_{I} \quad$ mass of the reactive intermediate, kg
$\mathrm{m}_{\mathrm{T}} \quad$ mass of tar, kg
n
no. of reaction stages which are first order reaction
R
gas constant, $8.314 \mathrm{~J} / \mathrm{mol} . \mathrm{K}$
$\mathrm{R}^{2}$
linear regression constant
kinetic compensation constant in Chapter 2
fitting parameter in Chapter 5
maximum possible devolatilization rate, $\mathrm{S}^{-1}$
kinetic compensation constant in Chapter 2
fitting parameter in Chapter 5, $\mathrm{b}=-\mathrm{E} / \mathrm{R}$
C heating rate, ${ }^{\circ} \mathrm{C} / \mathrm{min}$
activation energy, J/mol
activation energy of ith stage of reaction, $\mathrm{J} / \mathrm{mol}$
rate of heat supply for volatile evaporation per unit mass, $\mathrm{kJ} / \mathrm{s} . \mathrm{kg}$ coal
pre-exponential factor of ith stage reaction, $\min ^{-1}$
nominal rates $\bar{k}=(d V / d t) /\left(V^{*}-V\right), \min ^{-1}$
standard deviation of activation energy, J/mol

| s.e.e. | standard deviation error |
| :---: | :---: |
| T | temperature, ${ }^{\circ} \mathrm{C}$ ( or K) |
| Ti | critical temperature, at which reaction behavior is undergoing visible change in terms of the ratio $\mathrm{dV} / \mathrm{dT}$ or rate $\mathrm{dV} / \mathrm{dt}$ due to the change of reacting residue, K |
| To | initial temperature for TGA, $50{ }^{\circ} \mathrm{C}$ |
| $\mathrm{T}_{\mathrm{P}}$ | coal particle temperature, ${ }^{\circ} \mathrm{C}(\mathrm{K})$ |
| TV | volatile evaporation temperature, ${ }^{\circ} \mathrm{C}$ (or K) |
| V | volatile released at time $\mathrm{t}, \%$ (or kg ) |
| $\mathrm{V}_{\mathrm{t}=0}$ | volatile yield at $\mathrm{t}=0$ minutes, \% |
| $\mathrm{V}_{\mathrm{t}=10}$ | volatile yield at $\mathbf{t}=10$ minutes, \% |
| V* | total volatile yield, \% (or kg ) |
| X | reciprocal of temperature $1 / \mathrm{T}, \mathrm{K}^{-1}$ |
| X ${ }_{\text {v }}$ | fraction of volatile material to be released |
| $Y$ | LHS of each of the single overall first order reaction methods |
| $\alpha, \alpha_{1}, \alpha_{2}$ | kinetic compensation constant |
| $\alpha_{1}, \alpha_{2}$ | mass stoichiometric factors representing the extents of devolatilization via reaction 1 and 2 respectively in Chapter 2 |
| $\alpha_{i}$ | constant used to characterize the gradual change of the chemical structure of reacting residue. $\alpha_{i}=1$ when $T_{i-1} \leq T<T_{i}$, otherwise $\alpha_{i}=0$ |
| $\beta$ | shape parameter of Weibull distribution |
| $\beta, \beta_{1}, \beta_{2}$ | kinetic compensation constant |
| $\eta$ | scale parameter of Weibull distribution |
| $\gamma$ | threshold or location parameter of Weibull distribution |

## ACKNOWLEDGMENTS

I am grateful to my supervisor, Professor A. P. Watkinson, for his guidance and interest in this research. I am also indebted to many others who helped during the course of this work:

To the University of British Columbia which contributed financial support in the form of University Graduate Fellowships.

To the Natural Science and Engineering Research Council of Canada which provided financial support in the form of research grants.

To the Combustion Group of the Department of Chemical Engineering, UBC, which provided the CANMET pitch sample.

To Syncrude Canada Ltd. which provided the Syncrude pitch sample.
Special thanks are due to my mother, sisters and younger brother for their encouragement, understanding and patience.

## CHAPTER 1 INTRODUCTION

Pyrolysis of high molecular mass carbon and hydrogen containing materials is viewed as depolymerization in parallel with thermal decomposition of functional groups. The primary products compete for the donatable hydrogen for stabilization [1]. Pyrolysis is the first step in some conversion processes for hydrocarbon containing materials such as coal, heavy petroleum, and oil shale. It is the step which is most dependent on the properties of the hydrocarbons [2]. In combustion and gasification, pyrolysis precedes reaction by oxygen, steam, hydrogen or carbon dioxide [3]. In coking processes, pyrolysis of petroleum, semi-solids (mainly residua) and solids (mainly coals) results in the formation of a complete range of products from solids to gas. In addition to its importance in the hydrocarbon conversion process, analysis of pyrolysis products can supply important clues to the structure of the parent hydrocarbon.

The last several decades have seen an improvement in the understanding of coal and biomass pyrolysis in processes such as gasification, combustion, and liquefaction [1-4]. More rigorous information has also been developed for the light hydrocarbons. For complex feedstocks such as above, the approach taken to pyrolysis has been mainly semi-empirical. The literature contains relatively fewer attempts to deal with moderately heavy hydrocarbon feedstocks and the related mechanism involved, especially the secondary reactions which are often ignored in coal pyrolysis. Secondary reaction refers to the cyclization and condensation of the pyrolysis volatiles before leaving the reacting hydrocarbon matrix. For coal, secondary reactions are complex, being influenced by coal type, heating rate, residence time, temperature, intra- and extra-particle heat and mass transfer, and physical structure of the reacting coal. Further, these reactions can be heterogeneous (vapor-solid, vapor-liquid, or liquid-solid) or homogeneous (vapor phase or liquid phase) [5].

The pyrolysis of coal and biomass has been widely investigated since the late 1970's to maximize the liquid product yield in order to find a substitute for petroleum or for generation of chemicals. This has resulted in the development of several coal conversion processes $[6,7]$ in which the knowledge of pyrolysis is used to predict the product yields and distribution quite reasonably and successfully. For biomass, the complexity of the liquid products generally defies prediction.

The knowledge of pitch pyrolysis is also quite limited and has been borrowed from that for coal. Most known technology for processing of bitumen, coal, petroleum, and oil sands produces pitch. Pitch is commonly used to describe the liquid or semi-liquid fraction of a reaction product that boils above $524^{\circ} \mathrm{C}$ and which arises as a by-product from processing of crude oil or bitumen. Its relatively high $\mathrm{H} / \mathrm{C}$ atomic ratio (about 1.0 compared to about 0.5 for coke and 0.3 to 0.9 for coal) $[8,9]$ suggests that it should be possible to produce liquids by additional processing. Furthermore an appreciation of pitch pyrolysis might lead to new methods of pitch utilization. Thus, there is a clear need for further study in this field, to clarify the behavior of pitch in pyrolysis, which will improve the understanding of the processes and mechanisms involved, and hopefully lead to a proper way to process pitch, and generate economic and environmental profit.

As Canada, and other countries rely increasingly on heavy oils, residues from upgrading will become more of a disposal problem. Hydrogen or fuel gas production via gasification is a possible route to utilization. When pitches are heated prior to gasification their large volatile content is released, leaving a char. To understand kinetics, information on volatile yields and composition as function of temperature, atmosphere and pitch type is essential [10, 11]. Syncrude and CANMET processes represent two major bitumen processes which subsequently produce pitch as by-product. CANMET pitch is the residue of Cold Lake bitumen from CANMET hydrocracking process, where an additive is used to inhibit coke formation. This process was
demonstrated at 5000 bpd in Petro-Canada's Montreal refinery and about $10 \%$ of the feed ends up as pitch during the upgrading process [12-14]. This demonstration was successful and the technology is ready for commercialization. Syncrude pitch is the residue of Athabasca bitumen from Syncrude LC-Fining process. LC-Fining is a hydroprocessing process where $\mathrm{H}_{2}$ and catalysts are added in to upgrade bitumen at temperature $375-530^{\circ} \mathrm{C}$ and pressure $1100-1600 \mathrm{psi}$, and currently operating at $715 \mathrm{~m}^{3} / \mathrm{D}$ of bitumen. About $4 \%$ of the feed ends up as pitch [15].

The objectives of this study were to investigate the pyrolysis reaction mechanism and product distribution for different pitch types, and to formulate a model for the mechanism under conditions of different heating rates, final temperatures and reaction times. The study is concerned with pitch pyrolysis over a range of heating rates and for final temperatures from 700 to $1000^{\circ} \mathrm{C}$, and under normal pressure in an inert atmosphere, so that the first step in atmospheric pressure gasification, pyrolysis and combustion processes can be simulated.

Pitch pyrolysis at low heating rates is studied using a TGA, and at rapid heating rates with a Pyroprobe-equipped gas chromatograph. With TGA, the weight loss rate is investigated quantitatively at different final temperatures and heating rates less than $150^{\circ} \mathrm{C} / \mathrm{min}$. Diffusional effects inside the pitch samples are studied by changing the initial pitch sample weight (or pitch sample thickness inside the TGA sample holder). Thus pyrolysis kinetics and relative parameters can be derived from the data. With the Pyroprobe-equipped chromatograph, weight loss is also obtained at different final temperatures and heating rates up to $300,000^{\circ} \mathrm{C} / \mathrm{min}$. The Pyroprobe equipped chromatograph permits in situ GC analysis, in which the volatile composition is investigated as a function of reaction conditions. With the two procedures, pyrolysis kinetics and reaction parameters can be investigated under a wide range of conditions.

## Chapter 2 Literature Review

### 2.1 Chemical Structure of Pitch

The pyrolysis of relatively simple hydrocarbon compounds is complex and only partly understood. Therefore, it comes as little surprise that the knowledge of chemical mechanisms for pyrolysis of relatively undefined materials such as pitch or coal is lacking.

The characterization of pyrolysis products of coal and/or pitch is a sizable task, as these are usually present as gases, liquids and solids. The number of distinct chemical species is very large, and to facilitate data analysis one must usually resort to judiciously grouping the products into a few key classes of compounds.

With pitch, the characterization of the reactant is as difficult, if not more difficult than, the characterization of the products of the process. Because pitch is a somewhat heterogeneous and only partially soluble in most solvents, many of the traditional chemical and spectroscopic techniques for organic structure determination can not be applied easily or unambiguously. Therefore there is still a fair amount of debate over what constitutes a representative structure for a pitch "molecule". The chemical structures of the pitches studied have not been determined directly in this work. Rather, the structural characteristics must be inferred from a knowledge of the more traditional classification parameters for pitch.

The literature on the structure determination of petroleum derived pitch also contains information on "coal extracts" and other solvated coal and pitch fractions. However, the fraction that is soluble in a given solvent does not represent the total pitch or coal, since solubilization is unlikely to preserve its basic structure. The information is therefore difficult to apply.

### 2.1.1 The Carbon-Hydrogen Structure

It is generally accepted that an important characteristic of pitch or coal structure is its aromaticity, defined as the fraction of carbon in the pitch or coal which is aromatic in nature. A large number of approaches have been employed to determine the aromaticity and the average number of rings in the condensed polycyclic aromatic "clusters", as a function of carbon content. Various physical techniques [16] have been employed in studying the structure of coal/pitch structure.

From empirical studies on many hydrocarbons Van Krevelen [17] and several coworkers developed several ingenious correlations between measurable physical properties and some much more difficult to measure structural parameters.

Great advances have been achieved during the last decade in the application of NMR in pitch characterization. For measurements on pitch solutions, the main problem is the fact that pitches are not completely soluble in solvents suitable for NMR. Solid state NMR has the disadvantage of insufficient spectral results. As has been shown by Komatsu [18], these disadvantages can be overcome by measuring the spectrum at a temperature above the softening point of the pitch. The method has been applied to various types of pitches using ${ }^{13} \mathrm{C}$ NMR. Wellresolved spectra characterized by a high signal to noise ratio were obtained. Moreover, the measuring time could be markedly shorted compared with the measuring time necessary in organic solutions. ${ }^{13} \mathrm{C}$ NMR not only provides the important aromaticity figure but also detailed information on the aliphatic functional groups present in pitches.

Of course aromaticity alone does not completely characterize the carbon skeletal structure. Information on the distribution of aromatic and nonaromatic carbon is also necessary. It should also certainly be noted that using the total carbon content for the coals or pitch masks
potentially significant differences among the maceral fractions in coals, or the difference among fractions of different solubility in pitches.

Unfortunately data on total hydrogen distribution is not plentiful and its reliability is frequently questioned. Chemical techniques have provided some of the necessary data (such as that for hydroaromatic hydrogen and phenolic hydrogen), while spectroscopic techniques, such as
${ }^{1} \mathrm{H}$ NMR and IR , have provided others.

### 2.1.2 Solvent Fractionation of Pitches

Solvent fractionation is the most widely used method in pitch characterization. Solvent fractionation uses organic solvents of increasing polarity such as $n$-pentane, benzene and tetrahydrofuran (THF) to give fractions of increasing molar mass and heteroatom content. Three typical fractions are: $n$-pentane solubles, benzene insolubles and asphaltenes (benzene soluble, $n$ pentane insoluble). The fractions can be separated further by chromatographic methods and characterized by a variety of spectroscopic and chemical methods to provide details of individual components and average structures.

Chromatography is widely used in the separation, fractionation and characterization of complex mixtures of organic molecules. Size exclusion chromatography provides a separation mainly on the basis of molecular size which corresponds to separation on the basis of molar mass. It has been extensively used for coal and petroleum derivatives. However, separation occurs partially on the basis of functionality when THF is used as solvent, as well as on molecular size. This makes determination of molar mass distributions unreliable with high concentrations of pitch or tar present.

SARA (saturates, aromatics, resins and asphaltenes) separation is a traditional characterization method for hydrocarbon residuum, based on solubility/polarity of compounds. A
discussion of the chemical structures found in the SARA fractions can be found [19]. In general, the components are alkyl-substituted polycyclic structures related to steranes and hopanes derived from squalene precursors, or to terpenoid skeletons. Following the progression from saturates to aromatics to resins to asphaltenes, these fractions show increased aromaticity, average molecular weight, and heteroatomic content. There is also some overlap of structures and properties between neighboring fractions. Furthermore, variations in the relative amounts of the SARA fractions are accompanied by variations in the physical properties of bitumen. The chemical structures in the different fractions are believed to be related through various diagenetic processes such biodegradation and thermal maturation. However, each SARA fraction is fundamentally different to the extent that it can exhibit some specific chemical attributes.

The studies of SARA fractions have led to insights into the processing of petroleum residuum (pitch etc.). However, attempts to correlate SARA fractions with the processibility of residua have generally also been unsuccessful [20]. The determination of average molecular structures held some promise of providing insight into residuum conversion chemistry.

A promising approach to gain some understanding of the complex chemistry of residuum upgrading [21] by coking, hydrocracking and hydrotreating appears to be to use a combination of yield data obtained over a wide range of conversions, together with average molecular structural data and the extensive knowledge of molecular structures in residua.

### 2.2 Chemistry of Pyrolysis and Secondary Reaction

### 2.2.1 Chemical Thermodynamics of the Pyrolytic Reactions

To understand the chemistry of the pyrolysis reaction and the criteria for its chemical control, it is necessary to compare the thermodynamic stability of the various carbon compounds. The comparative stability of the various hydrocarbon groups may serve as a basis for discussing
the probable sequence of a decomposition reaction. Among the three major groups of hydrocarbons, i.e., paraffins, olefins and aromatics, the low molecular weight paraffins are the most stable hydrocarbons up to about $500{ }^{\circ} \mathrm{C}$ and among these, methane exhibits the greatest stability. Above $800{ }^{\circ} \mathrm{C}$, the aromatics become the most stable hydrocarbons. In this temperature range $\left(500-800^{\circ} \mathrm{C}\right)$, the thermodynamic stability of the olefins lies between that of the paraffins and aromatics. The stability of paraffins decreases with increasing chain length. In the higher temperature region ( $>800^{\circ} \mathrm{C}$ ), the same holds true for the olefins. The alkylated aromatics compounds are less stable than the pure aromatics. With increasing length of the side chain, the stability decreases. Contrary to this, the stability of aromatics increases with increasing number of rings, i.e. with increasing molecular size.

The mean bond energies of organic compounds are obtained by referring the energies of formation to the gaseous elements involved, i.e. carbon and hydrogen, and by then dividing by the number of bonds. Table 2.1 gives bonds energies obtained from thermodynamic data, as compared with those derived from quantum mechanical calculations based on bond length and force constants. Both methods give approximately the same values for the individual bond types.

Table 2.1 Bond Energies Obtained from Thermodynamic Data and from Quantum Mechanical Calculations [22]

| From thermodynamic data |  | From quantum mech. calculations |  |  |  |
| :--- | :---: | :---: | :---: | :---: | :---: |
|  | Mean <br> bond <br> energy <br> $\mathrm{kJ} / \mathrm{mol}$ | Mean <br> bond <br> energy <br> $\mathrm{kJ} / \mathrm{mol}$ | Distance <br> between <br> nuclei, $\AA$ | Force <br> constant <br> dynes <br> $\mathrm{cm}^{-1} \times 10^{-5}$ |  |
| $\mathrm{C}-\mathrm{C}$ | $\mathrm{C}_{2} \mathrm{H}_{6}$ | 325.10 | - | - | - |
| $\mathrm{C}=\mathrm{C}$ | $\mathrm{C}_{2} \mathrm{H}_{4}$ | 585.76 | 597.89 | 1.337 | 9.8 |
| $\mathrm{C} \equiv \mathrm{C}$ | $\mathrm{C}_{2} \mathrm{H}_{2}$ | 808.77 | 811.70 | 1.205 | 15.6 |
| $\mathrm{C}-\mathrm{C}$ | $\mathrm{C}_{6} \mathrm{H}_{6}$ | 517.98 | - | - | -- |
| $\mathrm{C}-\mathrm{H}$ | $\mathrm{CH}_{4}$ | 410.87 | 412.96 | 1.094 | 4.88 |
| $\mathrm{O}-\mathrm{H}$ | $\mathrm{HO}_{-1}$ | 457.73 | 462.75 | 0.98 | 7.6 |
| $\mathrm{C}=\mathrm{O}$ | $\mathrm{CH}_{2} \mathrm{O}$ | 683.25 | 694.54 | 1.21 | 12.1 |

A comparative consideration shows that the $\mathrm{C}-\mathrm{H}$ bond is more stable than the $\mathrm{C}-\mathrm{C}$ bond. Also shown is the higher bond energy of the C-C double bond and the C-C triple bond, which explains the dehydrogenation tendency towards olefins and the stability of acetylene at high temperature.

The high stability of ring compounds and especially of aromatics is due to the resonance energy. The resonance energy increases with increasing molecular size of the ring system, thus explaining the driving force for the chemical condensation of low molecular weight aromatics to polycyclic aromatic systems with the accompanying release of hydrogen. Examples are given in the following Table 2.2.

Table 2.2 Resonance Energies of Cyclic Compounds [22]

| Compound | Energy <br> $\mathrm{kJ} / \mathrm{mol}$ | Compound | Energy <br> $\mathrm{kJ} / \mathrm{mol}$ |
| :--- | :--- | :--- | :---: |
| Benzene | 150.62 | Quinoline | 288.70 |
| Naphthalene | 255.22 | Biphenyl | 299.16 |
| Anthracene | 351.46 | Aniline | 167.36 |
| Phenanthrene | 384.93 | Furan | 92.05 |
| Toluene | 146.44 | Pyrrole | 102.51 |
| Styrene | 158.99 | Indole | 205.02 |
| Phenol | 150.62 | Thiophene | 117.15 |
| Pyridine | 179.91 | Cyclooctatraene | 25.10 |

This qualitative thermodynamic consideration suggests the following trends for the course of pyrolysis of hydrocarbons with the increase of temperature [22]:

1. Cracking of all nonaromatic hydrocarbons to smaller molecules (cracking and dehydrogenation reactions).
2. Cyclization of all hydrocarbon chains to form aromatics. The first and the second reaction trends apply in the same way to aromatics with side chains which can undergo cracking or cyclization.
3. Condensation reactions of aromatics to form polycyclic aromatic systems.

These three principal types of reactions occur in all known technical processes dealing with the formation of carbon via pyrolysis reactions.

### 2.2.2 Pyrolysis of Unsubstituted Aromatics

Unsubstituted aromatics primarily exhibit direct ring condensation, i.e., the formation of diarenes and triarenes. Whenever sterically possible, chemical condensation can proceed to polycyclic products. Unsubstituted aromatics having the anthracene configuration are more reactive with respect to chemical condensation.

In summary [22], the pyrolysis of hydrocarbons takes place via aromatic intermediates. Results on the pyrolysis of well-defined, pure aromatics have shown the following effects:

1. Unsubstituted aromatics react by chemical condensation to form polynuclear aromatics, the aromatics having an anthracene configuration being the most reactive.
2. Alkyl-substituted aromatics are more reactive than unsubstituted ones, the effect being more pronounced the greater the number and the length of the alkyl groups.
3. The alkyl groups are the positions where the formation of the new aromatic systems takes place.
4. The highest reactivity is exhibited by aromatics containing five-numbered ring systems.

The existing investigations, pertaining to gas phase pyrolysis in a flow system, show that in the early stages the order of reaction is approximately unity for benzene, naphthalene and biphenyl. The apparent first order rate constants for these three aromatics are found to be of the same order of magnitude. The apparent activation energies amount to approximately 292.88 $\mathrm{kJ} / \mathrm{mol}$ to $334.72 \mathrm{~kJ} / \mathrm{mol}$.

### 2.2.3 Pyrolysis of Mixture of Hydrocarbons

The great complexity of chemical reactions occurring during the pyrolysis of hydrocarbons can be recognized not only from thermodynamic considerations but also from technological experiences gained in different processes. In view of the various mixtures of hydrocarbons used as raw materials and because of the insufficient analytical control of a technical pyrolysis, these processes do not reveal the chemistry in detail. Nevertheless, they provide a fair picture as to process parameters such as temperature, residence time and yield upon pyrolysis. Free radical reactions control the pyrolysis of most organic substances.

### 2.2.3.1 Pyrolysis of Crude Oil Fractions to Volatile Products

It has been found that the tendency to pyrolysis increases from the paraffins to the olefins and further to the naphthalene and the alkylated aromatics, up to the nonsubstituted aromatics. In case of purely thermal pyrolysis, mild conditions around $400^{\circ} \mathrm{C}$ lead primarily to a fracture of the C-C bond, preferentially in the middle of the molecule chain. With increasing temperatures, the position of the fracture shifts towards the chain end, thus producing long chain olefins and increased portions of highly volatile fragments.

Paraffins undergo pyrolysis leading to the formation of saturated and nonsaturated fragments between $400^{\circ} \mathrm{C}$ and $600^{\circ} \mathrm{C}$. Depending on the length of the main chain, isoparaffins primarily lose their branches and then behave like straight chain paraffins. Ring paraffins lose parts of their side chains, thus leading to unsaturated fragments. At temperatures above $600{ }^{\circ} \mathrm{C}$, naphthalene rings can be broken to form straight chain olefins. Cycloparaffins containing three carbon atoms are broken most easily, whereas cyclopentane is most stable. Cycloparaffins containing six carbon atoms in the ring become stabilized by aromatization. In the case of
alkylated aromatics, the rupture of the side chains is promoted with increasing length of these chains.

With rising temperature, the rupture of the $\mathrm{C}-\mathrm{H}$ bond is more strongly enhanced than rupture of the $\mathrm{C}-\mathrm{C}$ bond. Thus the formation of very small fragments down to hydrogen is favored and diolefins and triolefins with good thermal stability are formed. Above $550^{\circ} \mathrm{C}$, long chain olefins disintegrate, leading to shorter molecules and partial aromatization.

The first step of the hydrocarbon pyrolysis, namely, the decomposition to nonaromatic hydrocarbons takes place in the low temperature region between 400 and $700^{\circ} \mathrm{C}$, whereas the aromatization occurs between 700 and $900{ }^{\circ} \mathrm{C}$. These results are however valid only for short contact times at the described temperature.

Similar experiences pertain to the coal coking processes. The volatile hydrocarbons released at pyrolysis temperatures between 400 and $500{ }^{\circ} \mathrm{C}$ consist mainly of noncyclic compounds. In the high temperature range, however, the volatile products found in the coal tar are extremely aromatic.

### 2.2.3.2 Pyrolysis of SARA Fractions

Evidence from GC, HPLC, and FTIR analysis [23-29] suggests that SARA fractions (saturates, aromatics, resins, and asphaltenes) from heavy hydrocarbon undergo dealkalyation and aromatization when pyrolyzed at temperatures $362{ }^{\circ} \mathrm{C}$ to $418{ }^{\circ} \mathrm{C}$. Aromatization and dealkylation of the polycyclic, saturated structures in the saturate fraction lead directly to the production of aromatic compounds. Further aromatization and dealkylation of the aromatic fraction result in resin production. Resins and asphaltenes have chemical and structural similarities. The thermal pyrolysis of the resins and asphaltenes results in further condensation of the polycyclic structure and fragmentation and finally leads to the formation of coke. This process was proven to involve
bond scission and radical reactions. The pyrolysis of each of the SARA fractions appeared first order. The observed apparent activation energy of pyrolysis for aromatics, resins and asphaltenes is $108,135,150 \mathrm{~kJ} / \mathrm{mol}$ respectively [23]. These values fall in the wide range of values 29.1 $\mathrm{kJ} / \mathrm{mol}$ and $286 \mathrm{~kJ} / \mathrm{mol}$ [29] for bitumen pyrolysis. Observed values of apparent activation energies depend upon many factors, including the structure and complexity of the kinetic model. The method used to prepare a particular fraction will influence both its chemical composition and behavior, and consequently the values for any kinetic parameters that characterize it. The more chemically-varied the species contained in a particular fraction, the greater will be the number of reactions within the fraction and consequently, the lower will be the observed globe kinetic parameters [13, 30].

### 2.3 Pyrolysis Models and Comparison

One aim of modeling is to predict the pyrolysis behavior a priori in a conversion system as a function of parameters (temperature, heating rate, pressure, particle size etc.) thus facilitating the design of conversion reactors. Systematic research to this end during recent decades has advanced our knowledge to a stage where reasonable predictions are possible through modeling [31, 32]. These studies have provided valuable insight into the kinetics and the mechanism of the pyrolysis process. The modeling of pyrolysis is relatively straightforward when the chemical reaction is the only process occurring within the reactor and the feed species is simple. There are, of course, different levels of complexity of kinetic models. For simple hydrocarbons, pyrolysis models are based on the free radical mechanism. For propane pyrolysis, for example, the scheme of Trimm and Turner [33] includes one initiation reaction, thirty one propagation reactions, and nine termination reactions. These involved no species of greater molecular weight than $\mathrm{C}_{4} \mathrm{H}_{10}$, and no coke formation. At a less complex level, Sundaram and Froment [34] describe propane
cracking by ten reactions using molecular species rather than free radicals. This reaction scheme yields information on product distributions, but does not represent the mechanism as such. For higher molecular weight hydrocarbon feeds, or complex mixtures such as pitch, it is not feasible to write a kinetic model which reflects all steps in the actual mechanism. For example, the Kumar and Kunzru [35] scheme for naphtha pyrolysis incorporates twenty two reactions which are written in terms of molecular species. Each reaction requires a specified pre-exponential factor and activation energy. However, pyrolysis can involve extra transport steps which introduce complexity.

The review by Jamaludin et al. [36] considers the present understanding of kinetic models, and the review by Suuberg [37] considers the present understanding of general pyrolysis models including the mass transfer limitations of coal pyrolysis. Analogous models are applied to biomass pyrolysis.

At the time of writing, no accurate model has been developed to completely describe pitch pyrolysis. The following work which primarily involves coal pyrolysis is reviewed as that which bears most relevance to the system under investigation. Rather than dealing with individual species, this approach deals with the volatile matter as one or two components. The application of the models to pitch involves some changes, and certain steps which are valid for coal, would not apply to pitch. The two competing reaction model of pyrolysis is shown to be a simple, but effective method for predicting the weight loss due to devolatilization at high temperature and high heating rates for coal pyrolysis.

### 2.3.1 Constant Evaporation Rate Model

The constant evaporation model is probably the simplest existing pyrolysis model. Proposed by Baum and Street [38], it assumes that pyrolysis does not begin until the particle
temperature exceeds a vaporization temperature $\mathrm{T}_{\mathrm{V}}$, taken as $327{ }^{\circ} \mathrm{C}$. Above $\mathrm{T}_{\mathrm{V}}$, the rate of pyrolysis is controlled by the total heat of evaporation of the volatile, up to an empirically determined maximum value.

The rate of pyrolysis in terms of the fraction of volatile material $X_{V}$ to be released, can be expressed as

$$
\begin{array}{ll}
\frac{d X_{V}}{d t}=0 ; & \mathrm{T}_{\mathrm{P}}<\mathrm{T}_{\mathrm{V}} \text { or } \mathrm{X}_{\mathrm{V}}=1.0 \\
\frac{d X_{V}}{d t}=-H_{P T} / h_{l V} ; & \mathrm{T}_{\mathrm{P}} \geq \mathrm{T}_{\mathrm{V}} \text { and } \mathrm{X}_{\mathrm{V}}<1.0, \mathrm{H}_{\mathrm{PT}}<\mathrm{Bh}_{\mathrm{IV}} \\
\frac{d X_{V}}{d t}=-B ; & \mathrm{T}_{\mathrm{P}} \geq \mathrm{T}_{\mathrm{V}} \text { and } \mathrm{X}_{\mathrm{V}}<1.0, \mathrm{H}_{\mathrm{PT}} \geq \mathrm{Bh}_{\mathrm{IV}} \tag{2.2c}
\end{array}
$$

Where $h_{L V}$ is total heat of volatile evaporation per unit mass of coal, $\mathrm{kJ} / \mathrm{kg}$ coal, $H_{P T}$ is rate of heat supply for volatile evaporation per unit mass, $\mathrm{kJ} / \mathrm{s} . \mathrm{kg}$ coal. B is the maximum possible devolatilization rate. $\mathrm{T}_{\mathrm{P}}$ is the coal particle temperature and $\mathrm{T}_{\mathrm{V}}$ the volatile evaporation temperature. Lochwood et al. [39] observed that good predictions are obtained only for coal when $\mathrm{B}<\mathrm{H}_{\mathrm{PT}} / \mathrm{h}_{\mathrm{l}}$, while Jamaludin found that using considerably higher values of B compared to the recommended value of $10 \mathrm{~s}^{-1}$ did not appreciably change the predicted temperature [36].

By defining $m_{C}(0)$ as the initial mass of coal $(\mathrm{kg}), m_{C}$ as the mass of coal at any time $t$ $(\mathrm{kg}), m_{\mathrm{C}}$ (final) as the mass of coal at the end of pyrolysis reaction $(\mathrm{kg})$, then the volatile released at time $t$ is

$$
\begin{equation*}
V=\frac{m_{C}(0)-m_{C}}{m_{C}(0)} \tag{2.2d}
\end{equation*}
$$

and the total volatile yield is

$$
\begin{equation*}
V^{*}=\frac{m_{C}(0)-m_{C}(\text { final })}{m_{C}(0)} \tag{2.2e}
\end{equation*}
$$

and the fraction of volatile material $X_{V}$ to be released is

$$
\begin{equation*}
X_{V}=\frac{m_{c}-m_{c}(\text { final })}{m_{c}(0)-m_{c}(\text { final })} \tag{2.2f}
\end{equation*}
$$

### 2.3.2 Single Overall Reaction Model

This model, proposed first by Badzioch and Hawksley [40] for coal pyrolysis, has been widely used due to its simplicity and effectiveness. It is based on the following simplified reaction scheme

$$
\mathrm{C} \xrightarrow{k} \mathrm{~V}+\mathrm{R}
$$

coal $C$ pyrolyzes to produce volatiles $V$ and solid residue $R$. The reaction is assumed first order, the pyrolysis rate being proportional to the volatile matter yet to be released ( $\mathrm{V}^{*}-\mathrm{V}$ )

$$
\begin{equation*}
\frac{d V}{d t}=k\left(V^{*}-V\right) \tag{2.3a}
\end{equation*}
$$

Where $\mathrm{V}^{*}$ is the total volatile fraction, and the rate constant k is represented by an Arrhenius expression:

$$
\begin{equation*}
k=k_{o} \exp \left[-E / R T_{P}\right] \tag{2.3b}
\end{equation*}
$$

The fractional devolatilization at any time is obtained by integrating the above equation, then

$$
\begin{equation*}
\frac{V}{V^{*}}=1-\exp \left[-\int_{0}^{t} k d t\right] \tag{2.3c}
\end{equation*}
$$

### 2.3.3 Two Competing Reaction Model

This model, proposed by Kobayashi et al. [41], and Ubhahayakar et al. [42], represents the overall coal pyrolysis process by two mutually competing first order reactions as:


The rate of weight loss of the coal (maf basis) is given by

$$
\begin{equation*}
\frac{d m_{C}}{d t}=-\left(k_{1}+k_{2}\right) m_{C} \tag{2.4a}
\end{equation*}
$$

so that at any time $t$ the mass of material yet to be pyrolyzed is

$$
\begin{equation*}
m_{C}=m_{C}(0) \exp \left[-\int_{0}^{\iota}\left(k_{1}+k_{2}\right) d t^{\prime}\right] \tag{2.4b}
\end{equation*}
$$

Where the rate of devolatilization at any time is

$$
\begin{equation*}
\frac{d V}{d t}=\left(\alpha_{1} k_{1}+\alpha_{2} k_{2}\right) m_{C} \tag{2.4c}
\end{equation*}
$$

Where $\alpha_{1}$ and $\alpha_{2}$ are mass stoichiometric factors representing the extents of devolatilization via reaction 1 and 2 respectively. The extent of devolatilization at time $t$ is obtained as

$$
\begin{equation*}
V(t)=m_{C}(0) \int_{0}^{t}\left(\alpha_{1} k_{1}+\alpha_{2} k_{2}\right) \exp \left[-\int_{0}^{t}\left(k_{1}+k_{2}\right) d t^{\prime \prime}\right] d t^{\prime} \tag{2.4d}
\end{equation*}
$$

The rate constants $k_{1}$ and $k_{2}$ have Arrhenius form, and are such that reaction 1 has a lower activation energy than reaction 2, with the effect that secondary reaction becomes operational only at higher temperature to effect volatile yields in excess of $\alpha_{1}$.

### 2.3.4 Three Reaction Models

The three reaction models, first proposed by Wen and Dutta [43], considers of three parts representing devolatilization, cracking and deposition. The pyrolysis products are gases, tar and solid residual. Tars are defined as species heavier than $\mathrm{C}_{6}$, and gases those lighter than $\mathrm{C}_{6}$. The proposed reaction scheme is:


The rate of weight loss of the coal particle is given by

$$
\begin{equation*}
\frac{d m_{C}}{d t}=-k_{1} m_{C} \tag{2.5a}
\end{equation*}
$$

By integration, then

$$
\begin{equation*}
m_{C}(t)=m_{C}(0) \exp \left[-\int_{0}^{t} k_{1} d t^{\prime}\right] \tag{2.5b}
\end{equation*}
$$

The net rate of production of tar is

$$
\begin{equation*}
\frac{d m_{T}}{d t}=\alpha_{1} k_{1} m_{C}-\left(k_{2}+k_{3}\right) m_{T} \tag{2.5c}
\end{equation*}
$$

Where $m_{T}$ is the mass of tar, then the yield of tar at any time is given by integration of the above equation.

$$
\begin{equation*}
m_{T}(t)=\alpha_{1} m_{C}(0) \exp \left[-\int_{0}^{t}\left(k_{2}+k_{3}\right) d t^{\prime}\right] \int_{0}^{t} k_{1} \exp \left[\int_{0}^{t}\left(k_{2}+k_{3}-k_{1}\right) d t^{\prime \prime}\right] d t^{\prime} \tag{2.5d}
\end{equation*}
$$

The corresponding rate of production of volatile is

$$
\begin{equation*}
\frac{d V}{d t}=k_{2} m_{T} \tag{2.5e}
\end{equation*}
$$

By integration the above equation, with $m_{T}$ given by the previous equation, gives the volatile yield at any time $t$. For isothermal conditions, the expression for the volatile yield simplifies to

$$
\begin{equation*}
V(t)=\frac{\alpha_{1} k_{2} m_{C}(0)}{k_{2}+k_{3}-k_{1}}\left\{\left[1-\exp \left(-k_{1} t\right)\right]-\frac{k_{1}}{k_{2}+k_{3}}\left[1-\exp \left(-\left(k_{2}+k_{3}\right) t\right)\right]\right\} \tag{2.5f}
\end{equation*}
$$

which further simplifies to (by assuming $k_{1}$ much smaller than $k_{2}, k_{3}$ )

$$
\begin{equation*}
V(t)=\frac{\alpha_{1} m_{C}(0)}{1+\left(k_{3} / k_{2}\right)}\left[1-\exp \left(-k_{1} t\right)\right] \tag{2.5~g}
\end{equation*}
$$

A reaction scheme similar to the above was proposed by Niksa et al. [44], using a nonisothermal kinetic analysis, similar to that of Jüntgen and Van Heek [45, 46], to show that faster devolatilization rates were obtained at higher heating rates and they adopted the following competitive scheme to account for the enhanced yield at high heating rates as:


The rate of decomposition of the coal particle is given by the equation

$$
\begin{equation*}
\frac{d m_{C}}{d t}=-k_{1} m_{C} \tag{2.6a}
\end{equation*}
$$

the rate of production of the volatile and the intermediate is then given as

$$
\begin{align*}
& \frac{d V_{1}}{d t}=\alpha_{1} k_{1} m_{C}  \tag{2.6b}\\
& \frac{d m_{I}}{d t}=\left(1-\alpha_{1}\right) k_{1} m_{C}-\left(k_{2}+k_{3}\right) m_{I}  \tag{2.6c}\\
& \frac{d V_{2}}{d t}=k_{3} m_{I} \tag{2.6d}
\end{align*}
$$

Where $m_{I}$ is the mass of the reactive intermediate.
Nsakala et al. [47] proposed the following parallel consecutive reactions scheme based on pyrolysis of lignite at $800^{\circ} \mathrm{C}$ as:


Coal particle $C$ is assumed to consist of two distinct components, $C_{1}$ and $C_{2}$, of different ease of pyrolysis. In their analysis, Nsakala et al. [48] ignored the secondary cracking of $\mathrm{V}_{1}$. If, therefore, components $\mathrm{C}_{1}$ and $\mathrm{C}_{2}$ decompose isothermally by independent first order reactions

$$
\begin{align*}
& C_{1}=C_{01} \exp \left(-k_{1} t\right)  \tag{2.7a}\\
& C_{2}=C_{02} \exp \left(-k_{2} t\right) \tag{2.7b}
\end{align*}
$$

Where $\mathrm{C}_{01}$ and $\mathrm{C}_{02}$ are the initial mass of $\mathrm{C}_{1}$ and $\mathrm{C}_{2}$. The total weight loss is obtained from

$$
\begin{equation*}
V=V_{1}+V_{2} \tag{2.7c}
\end{equation*}
$$

where $V_{1}$ and $V_{2}$ are volatile product from $C_{1}$ and $C_{2}$ components respectively and

$$
\begin{align*}
& V_{1}=C_{01}-C_{1}  \tag{2.7d}\\
& V_{2}=C_{02}-C_{2}-R  \tag{2.7e}\\
& V=C_{01}\left[1-\exp \left(-k_{1} t\right)\right]+\frac{C_{02}}{1+R / V_{2}}\left[1-\exp \left(-k_{2} t\right)\right] \tag{2.7f}
\end{align*}
$$

At infinite time

$$
\begin{equation*}
V^{*}=C_{01}+\frac{C_{02}}{1+R / V_{2}} \tag{2.7~g}
\end{equation*}
$$

or

$$
\begin{equation*}
1-\frac{V}{V^{*}}=\left(\frac{C_{01}}{V^{*}}\right) \exp \left(-k_{1} t\right)+\left(1-C_{01} / V^{*}\right) \exp \left(-k_{2} t\right) \tag{2.7h}
\end{equation*}
$$

### 2.3.5 Multiple Parallel Reaction Model

The powerful multiple parallel reaction model, originally proposed by Pitt [48], was later adopted by Anthony and Howard $[49,50]$ to fit their data. The merit of this model is that it only needs one more adjustable parameter than the single reaction model. The reactions envisaged were

$$
\text { Coal } \rightarrow \sum_{i=1}^{n} C_{i} \xrightarrow{k_{i}} \sum_{i=1}^{n}\left(V_{i}+R_{i}\right)
$$

The reactions are assumed to have the same pre-exponential factor but different activation energies. The weight loss due to devolatilization at any time is

$$
\begin{equation*}
\frac{V}{V^{*}}=1-\int_{0}^{\infty} \exp \left[-\int_{0}^{t} k(E) d t\right] f(E) d E \tag{2.8a}
\end{equation*}
$$

$f(E)$, denoting the distribution function of activation energy, is assumed to be Gaussian and given by

$$
\begin{equation*}
f(E)=\left(\frac{1}{s \sqrt{2 \pi}}\right) \exp \left[-\left(E-E_{0}\right)^{2} /\left(2 s^{2}\right)\right] \tag{2.8b}
\end{equation*}
$$

Where $E_{0}$ is the mean activation energy and $s$ the standard deviation.

Instead of using a Gaussian distribution, Laskshmannan [51] used the Weibull distribution to model the kinetics of petroleum generation over a geological time scale. The probability density function $f(E)$ (as applied to describe the distribution of activation energies) for this distribution is given by:

$$
\begin{align*}
f(E) & =\left(\frac{\beta}{\eta}\right)\left(\frac{E-\gamma}{\eta}\right)^{\beta-1} \exp \left[-\left(\frac{E-\gamma}{\eta}\right)^{\beta}\right] \text { for } E \geq \gamma, \eta>0 \text { and } \beta>0 \\
& =0 \text { for all other values of } E, \eta \text { and } \beta \tag{2.8c}
\end{align*}
$$

where $E$ is the activation energy expressed in $\mathrm{kcal} / \mathrm{mol}$. There are three parameters, namely, $\eta$, the scale parameter; $\beta$, the shape parameter; and $\gamma$, the threshold or location parameter characterizing the distribution. A number of different distributions can be generated by a suitable choice of these parameters. For $\beta=1$, the Weibull distribution coincides with the exponential distribution. For $\beta>1$, the distribution becomes 'bell shaped", but becomes positively skewed. As $\beta$ increases, the Weibull distribution approaches the Gaussian distribution more and more closely. In fact, for $\beta=4$, the Weibull and Gaussian distributions become almost indistinguishable. This model may be useful for process chemical engineering applications, such as combustion and pyrolysis of coal, oil shale, bitumen and pitch. Unlike the Gaussian distribution, the Weibull distribution is well suited to represent many empirical distributions.

Noting the limitation of these distributions, Miura [52] proposed a mathematical procedure to estimate $f(\mathrm{E})$ from experimental data without assuming any form of distribution. This procedure requires only three sets of experimental data. The procedure to estimate $f(E)$ and $k_{0}$ is summarized as follows:

1. Measure $\mathrm{V} / \mathrm{V}^{*}$ vs T relationships at three different heating rates at least.
2. Calculate nominal rates $\bar{k}=(d V / d t) /\left(V^{*}-V\right)$ at several but same $\mathrm{V} / \mathrm{V}^{*}$ values at different heating rates, then make Arrhenius plots of $\bar{k}$ at the same $\mathrm{V} / \mathrm{V}^{*}$ values.
3. Determine activation energies from the Arrhenius plots at different levels of $\mathrm{V} / \mathrm{V}^{*}$ and then plot $\mathrm{V} / \mathrm{V}^{*}$ against the activation energy E .
4. Differentiate $V / V^{*}$ by $E$ to obtain $f(E)$.
5. Calculate $k_{0}$ corresponding to each $E$ value at all the heating rates using equation $0.5447 \alpha E / k_{o} R T^{2}=e^{-E / R T}$, then employ the averaged $\mathrm{k}_{0}$ value as a true $\mathrm{k}_{0}$ value.

### 2.3.6 Complex Models

In order to model more accurately the gross fundamental mechanism involved, Reidelbelbach and Summerfield [53,54] formulated a model which included six competitive /consecutive reactions. This was later modified by Antal et al. [55] to correct the abnormally high activation energy for the activation step. The reaction scheme is expressed as follows:


Several consideration went into the model, e.g. reaction 1 was proposed to limit decomposition of coal at low temperature. Further decomposition can then proceed by two routines depending on the heating rate and the temperature. The tar production step (reaction 2 ) was assigned a low activation energy as tar evolves at comparatively low temperatures. Similarly,
the experimental observation for increased gas/tar ratios and increased yield at high temperature, etc., were also accommodated. Reidelbach and Summerfield achieved good agreement with the experiment data of Badzioch and Hawsley [40] using a simplified version of the model.

### 2.3.7 Detailed Models

Detailed models of coal pyrolysis attempt to describe the evolution of individual volatile species. One such model is that formulated by Suuberg et al. [56-58] assuming nine volatile products to be formed via fifteen different reactions. The activation energies of the individual reactions when synthesized into a composite distribution function, were found to agree well with the corresponding Gaussian distribution obtained by Anthony et al. [49] solely from the total weight loss data. Tar was assumed to be either converted to coke and light hydrocarbons by secondary reactions, or evolve from the coal particles, as in the two competitive reactions below:


Soloman and coworkers [1-3] have been working towards providing a fundamental basis for pyrolysis reactions though the concept of 'functional groups'. The overall reaction is:


Thus a representative sample of the functional groups evolves without decomposition leaving coal molecule to form tar, while light primary gases are formed by decomposition of some functional groups. These two processes are assumed to be competitive. A single rate is used for tar evolution, and a separate rate for each gaseous species. Distributed (Gaussian) rate kinetics
are used for the gaseous species evolution, and secondary reaction of tar and tar evolution are represented by a separate set of competing reactions. The evolution of ten species (excluding tar) are represented by 15 reactions. Time and temperature dependent devolatilization of coal was predicted by the model using a coal independent set of kinetic parameters and the structural composition.

### 2.3.8 The Application of Models in Pyrolysis Kinetics

The global pyrolysis kinetics applied to pitch or any other hydrocarbon is generally intended to predict the overall rate and yield of volatile release (i.e. mass loss) from the sample. For a first order process this is given as Equation 2.3: $d V / d t=k\left(V^{*}-V\right)$, where for temperature programmed experiments, $\mathrm{T}=\mathrm{f}(\mathrm{t})$. For linear rise in temperature $\mathrm{T}-\mathrm{T}_{0}=\mathrm{Ct}$ where C is heating rate. It has been reported that different volatile products are released depending on the temperature ranges [59] or the temperature histories [60]. This fact has not diminished the interest in the global kinetics for various reasons. One reason is that under certain conditions, tar is a dominant product of pyrolysis for a significant part of the process [59], so that prediction of total mass loss would allow prediction of tar release rate. A second reason is that global kinetics are looked to as offering a clue to the key mechanistic steps in the overall pyrolysis process [61].

Carrasco [62] conducted an extensive review of the different computing methods (used to analyze Equation 2.3) in the literature leading to the determination of the kinetic parameters of thermal decomposition reactions and compared the results obtained by using those methods for coal. Those methods do not reproduce the values of activation energy and reaction order when the same data are taken for computation. Due to the above mentioned shortcomings of these methods listed in Appendix A, these methods are of little use for pitch or bitumen pyrolysis studies, except for the integral method. Table 2.3 summarizes some of the methods used for
analyzing data via Equation 2.3 and the Anthony-Howard model (Equation 2.8). However, there is no comparison of the kinetic parameters derived from the methods listed in Table 2.3 reported yet for pitch. A detailed description of these methods is found in Chapter 5.

Table 2.3 Summary of the Analysis Methods Used in Constant Heating Rate Pyrolysis

| Integral Method | $-\ln \left(\frac{V^{*}-V}{V^{*}}\right)=\frac{k_{o}}{C}\left[T e^{-E / R T}+\frac{E}{R} E_{i}\left(-\frac{E}{R T}\right)\right]$ | (2.9) |
| :---: | :---: | :---: |
| Friedman <br> Method | $\ln \left(\frac{C}{V^{*}} \frac{d V}{d T}\right)-\ln \left(1-\frac{V}{V^{*}}\right)=\ln k_{o}-\frac{E}{R T}$ <br> The values of $\mathrm{dV} / \mathrm{dT}$ is calculated by using two adjacent pairs of the volatile and temperature data: $\frac{d V}{d T}=\frac{V_{i+1}-V_{i}}{T_{i+1}-T_{i}}$ | (2.10) (2.10a) |
| CoatsRedfern Method | $\ln \left(\frac{-C \ln \left(1-\frac{V}{V^{*}}\right)}{R T^{2}}\right)=\ln \frac{k_{o}}{E}\left(1-\frac{2 R T}{E}\right)-\frac{E}{R T}$ | (2.11) |
| Chen-Nuttall Method | $\ln \left(\frac{-C(E+2 R T)}{R T^{2}} \ln \left(1-\frac{V}{V^{*}}\right)\right)=\ln k_{o}-\frac{E}{R T}$ | (2.12) |
| AnthonyHoward Model (1976) | $\begin{aligned} & \frac{V}{V^{*}}=1-\int_{0}^{\infty} \exp \left[-\int_{0}^{t} k(E) d t\right] f(E) d E \\ & f(E)=\left(\frac{1}{s \sqrt{2 \pi}}\right) \exp \left[-\left(E-E_{0}\right)^{2} /\left(2 s^{2}\right)\right] \end{aligned}$ | (2.8a) (2.8b) |

where the heating rate $\mathrm{C}=\mathrm{dT} / \mathrm{dt}$ in the above table.

### 2.4 Compensation Effect of Kinetic Parameters

On determining the kinetic parameters from the thermoanalytical curves with the single overall reaction model (Equation 2.3), variations in the kinetic parameters are encountered due to the variation in physico-chemical properties (such as sample size), measuring conditions and the mathematical methods employed to derive the kinetic parameters. Thus high values of activation energy would be compensated by high values of the pre-exponential factors to give the same rate
constant $k$ value. Further analysis of the variation of the kinetic parameters for a series of reactions leads to a general result of a mutual dependence of the kinetic parameters, termed as the kinetic compensation effect expressed by:

$$
\begin{equation*}
\ln k_{o}=\alpha E+\beta \tag{2.13}
\end{equation*}
$$

The above equation indicates the linear dependence between the values of the logarithmic pre-exponential factor $\operatorname{lnk}_{0}$ and the activation energy $E$ with the constants $\alpha$ and $\beta$. The simple relationship of the above equation is reproduced on the Arrhenius coordinates, lnk vs. $1 / \mathrm{T}$, with an intersection point called the isokinetic points ( $1 / \mathrm{T}_{\mathrm{iso}}, \operatorname{lnk}_{\mathrm{iso}}$ ) [63]. Using the isokinetic relationship, the above equation is rewritten as:

$$
\begin{equation*}
\ln k=a+b \frac{1}{T} \tag{2.14}
\end{equation*}
$$

The kinetic compensation effect was first identified by Constable [64] from studies of dehydrogenation of ethanol on copper. Subsequently, a large number of further examples of comparable patterns of kinetic behavior have been described for many diverse surface heterogeneous catalytic reactions. Occurrence of such a compensation behavior between $\operatorname{lnk}_{0}$ and $E$ has been widely investigated in recent years. In particular, the existence of the compensation effect in thermal dehydration and decomposition reactions of solid inorganic and organic materials has been reported [65]: Numerous papers have dealt with the variation of the apparent kinetic parameters using Equations 2.13 and 2.14. In addition, comparable relationships were found during these analyses of reported kinetic data. Additional trends could be also recognized if the survey was extended further or experimental measurements obtained for additional systems [66]. However, despite these many and various examples of compensation behavior, there remain important difficulties in establishing the range of meaningful application and the usefulness of Equations 2.13 and 2.14 in the understanding of the significance of kinetic observation. Although
the present state of understanding of the kinetic compensation effect can be found in many historical surveys [66-71], no single theoretical explanation of compensation behavior has been recognized as having general application. The factors to which references are made most frequently are surface heterogeneity in catalytic reactions and the occurrence of two or more concurrent and/or consecutive reactions in thermal decomposition processes. The causes of the kinetic compensation effect in thermal decomposition reactions may be classified into the three categories discussed below: sample physico-chemical properties, measuring conditions, and the mathematical methods used to derive the kinetic parameters. At present time, however, doubt remains concerning the general theoretical implications of the compensation relation despite the very many reported instances of obedience of Equation 2.13. Accordingly, this short review emphasizes the interrelation between kinetic characteristics and the chemistry of thermal decomposition processes.

### 2.4.1 Effect of Sample Physico-Chemical Properties on the Kinetic Compensation Effect

A typical example of the physico-chemical interpretation of the kinetic compensation effect is seen for the thermal decomposition of $\mathrm{CaCO}_{3}$, under various partial pressures of $\mathrm{CO}_{2}$. In 1935, Zawadski and Bretzsnajder [72] originally pointed out the variation in E with $\mathrm{CO}_{2}$ partial pressure. Another example is seen for the thermal decomposition of $\mathrm{CaC}_{2} \mathrm{O}_{4} \bullet \mathrm{H}_{2} \mathrm{O}$, with various sample sizes. The activation energy was found to decrease with the increase of sample size [73]. A theoretical interpretation for this effect was attempted by Pavlyuchenko and Prodan [74]. The kinetic behavior was reinvestigated experimentally by Wist [75] and analyzed by Roginski and Chatji [76] from a viewpoint of chemical statistics. Attempts have been made to explain the empirical kinetic compensation effect by using the physico-chemical variables, such as partial pressure of a gas [77], bond energy due to the different metals and ligands [78-80], defect
concentration [81], chemical composition [82], impurities [83] etc., other than reaction rate and temperature. Guarini et al. [84] pointed out that nonlinearity of the Arrhenius plot increases with the sample size, and recommended extrapolation to zero mass to avoid the kinetic compensation effect. Sample size dependent variations in the Arrhenius parameters have been explained by the effect of gradients in temperature and gaseous pressure [73]. In thermal analysis, however, the physico-chemical properties are difficult to identify quantitatively, because of the macroscopic character of the kinetic data derived from TGA curves. Without quantitative identification of the physico-chemical properties, estimation of the linear interdependence of Equation 2.13 does not provide meaningful kinetic interpretation, but only shows an empirical observation of the mutual dependence of the kinetic parameters.

### 2.4.2 Effect of Experimental Conditions on the Kinetic Compensation Effect

One of the examples is also seen for the thermal decomposition of $\mathrm{CaC}_{2} \mathrm{O}_{4} \bullet \mathrm{H}_{2} \mathrm{O}$, under various heating rates [73, 85, 86]. It was found that the activation energy $E$ decreased with the increase of heating rate. It is generally accepted that the experimentally resolved shape of a TGA curve changes with the measuring conditions applied, such as heating rates, atmosphere, etc. [85]. In many cases, the kinetic parameters obtained from such a TGA curve are also dependent on the measuring conditions applied, showing empirically the kinetic compensation effect. The kinetic compensation effect caused by the effect of heating rate is rather common for the thermal decomposition of solids with gaseous products $[87,88]$. On discussing the kinetic compensation effect obtained from different measuring conditions, both effects of heating rate on the sample physico-chemical properties and the changes in the sample caused by reaction itself should be taken into consideration. The latter is closely connected with the reliability of the experimentally resolved shape of the TGA curve as a source of kinetic data [89, 90], because such changes in the
sample is not controlled, in a strict sense, in conventional TGA measurements. A typical example can be seen for hydrocarbon pyrolysis in which the chemical structure and makeup is undergoing constant change.

### 2.4.3 Analysis of One TGA Experiment with Different Models or Methods

Discussion of the mutual dependence of the kinetic parameters has also been attempted from the mathematical and statistical points of view. Because the kinetic parameters have meaning only in relation to the mathematical functions of the kinetic model, these are distorted by an inappropriate kinetic model function. Criado and Gonzalez [91] reported that sets of kinetic parameters calculated using inappropriate kinetic model functions show mutual dependence. The degree of the distortion was further discussed on the basis of an empirical analysis [92, 93] and a mathematical approximation [94]. Reexamination of the kinetic compensation effect of this type was performed by Somasekharan and Kalpagam [95], who suggested the correspondence between the isokinetic temperature and the maximum TGA peak.

However, application of the Arrhenius equation to complicated solid-state processes has been questioned [96]. Hulett [97] made a search for the nonlinearity of the Arrhenius plot, determining that any derivations from a straight line in the plot of $\ln k(T)$ vs. $1 / T$ are to be considered as almost certain evidence that the observed process is complex. Drawing the theoretical TGA curves, correlation of the kinetic parameters and its effect on the TGA curves were noticed by Sesták [98] and further analyzed by Zsakó [99]. Exner [100] first suggested that it is not correct to determine the kinetic compensation effect by a linear regression of E vs. $\operatorname{lnk}_{0}$, because these quantities are mutually dependent. Agrawal [101] proposed dividing the kinetic compensation effect into two groups by the existence of an isokinetic point: one arising from physico-chemical factors and the other from computational and experimental artifacts. Because
$\mathrm{k}(\mathrm{T})$ and T can be determined independently, the plot of $\operatorname{lnk}(\mathrm{T})$ vs. $1 / \mathrm{T}$ is statistically correct. However, Agrawal's procedure of distinguishing a false kinetic compensation effect from a true one was criticized by Sesták [102] and was shown by Zsakó and Somasekharan [103] to be incorrect. Garn's view is that the kinetic compensation effect is simply a consequence of trying to describe a complex process by computing one of the kinetic parameters in Equation 2.3 and dumping the results of computed variations into the remaining 'constant', accepting changes of many orders of magnitude without question or test [104].

### 2.4.4 Interaction of the Causes

According to the procedure of TGA kinetics of thermal decomposition reactions, the sample physico-chemical properties, experimental conditions and the resulting mutual dependence of the kinetic parameters seem to be interpreted separately [105]. However, the causes seem to be interrelated and inseparable. The TGA curve is a response of a certain averaged behavior of the respective reaction steps involved for the case of the thermal decomposition. The mutual relationship of the consecutive and/or concurrent steps may change with the experimental conditions applied (such as heating rates) and the changes in the sample, influencing the overall characteristics of the reaction. The variation in the overall behavior for a reaction is only detected as changes in the position and shapes of the experimentally resolved TGA curves. The kinetic parameters calculated from these macroscopic data are projected on the Arrhenius coordinates through a particular projection system, i.e., the general kinetic equation. The variation in the respective kinetic parameters apparently results from changes in the experimental and physicochemical factors. However, the resulting mutual dependence of the kinetic parameters, usually stated as the kinetic compensation effect, seems to be connected with the properties of the mathematical methods used to analyzed the general kinetic equation (Equation 2.3). In such a
case, not knowing the properties of the general kinetic equation concerning the kinetic compensation effect, interpretation of the mutually dependent variation of the Arrhenius parameters connected with the physico-chemical properties of the kinetic process is likely to lead to a speculative conclusion. However, recognition of the kinetic compensation effect would give some insights to the relationship between the logarithm of pre-exponential factor, $\operatorname{lnk}_{0}$, and activation energy E , and further give guidelines of application and explanation of the kinetic parameters. The magnitude of the rate constant is therefore of more importance than that of each of the kinetic parameters $\mathrm{k}_{\mathrm{o}}$ and E .

## Chapter 3 Experimental Procedures and Apparatus

### 3.1 Introduction

In this chapter, the experimental procedures which outline each operational step employed in the present work are discussed. The first part of this chapter deals with the materials, sample preparation and characterization. This is followed by the description of the experimental apparatus. Finally, the experimental techniques are presented.

### 3.2 Sample Preparation and Characterization

The CANMET pitch was obtained from Combustion Group of Department of Chemical Engineering at UBC, which obtained the pitch sample from CANMET in barrels for combustion study. The Syncrude pitch was obtained from the sample bank of Syncrude Canada Ltd. in 10kg containers. Representative samples were then taken from CANMET pitch barrels and Syncrude pitch containers and stored in a refrigerator for the subsequent characterization analysis, TGA study and Pyroprobe-GC study.

Each of the two pitch samples was used as received. Representative samples of CANMET and Syncrude pitches were sent to MicroAnalytical of Delta, Vancouver for ultimate analysis. Results are given in Table 3.1, along with the proximate analysis determined by TGA and solvent fractionation with pentane and benzene. The latter were determined by dissolving 5 mg of pitch sample into 200 ml pentane and benzene respectively in an ultrasonic bath ( $\sim 25^{\circ} \mathrm{C}$ for 30 min ). The pentane or benzene soluble fractions were clarified over filter paper and the insolubles washed and dried at room temperature for 12 hours. The weight of the insolubles was recorded. The atomic ratios were also calculated and given in the same table. It is evident that the chemical structure and makeup of Syncrude pitch are different from those of CANMET pitch. Syncrude pitch has higher H/C, S/C atomic ratios and lower N/C, O/C atomic ratios. This observation is in
good agreement with the low pentane and benzene insolubles. It is expected that the pyrolysis behavior of these two pitches might be different due to those chemical differences. Both pitches contain limited amounts of ash and oxygen. Syncrude pitch contains more sulfur than the CANMET pitch.

Table 3.1 Pitch Characterization Analysis -- Ultimate Analysis

| As received | CANMET Pitch |  | Syncrude Pitch | Suncor <br> Pitch [8] | Maya[106]Residuum |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | This Work | Lim [10] |  |  |  |
| Carbon \% | 85.32 | 86.2 | 82.72 | 82.8 | 83.6 |
| Hydrogen \% | 9.33 | 7.1 | 10.35 | 7.9 | 9.3 |
| Nitrogen \% | 0.82 | 1.1 | 0.52 | 1.0 | 0.5 |
| Sulfur \% | 2.39 | 2.8 | 4.73 | 5.8 | 5.8 |
| Oxygen \% | 1.12 | 1.0 | 0.97 | -- | 0.5 |
| Others \% | 1.02 | 1.8 | 0.71 | -- | -- |
|  | 100.00 | 100.00 | 100.00 | 97.5 | 99.7 |
| H/C | 1.31 | 0.99 | 1.50 | 1.145 | 1.335 |
| N/C | 0.0082 | 0.011 | 0.0054 | 0.0104 | 0.0051 |
| O/C | 0.0098 | 0.0087 | 0.0088 | -- | 0.0045 |
| S/C | 0.011 | 0.0012 | 0.021 | 0.0263 | 0.026 |
| Proximate Analysis \% |  |  |  |  |  |
| Volatile | 81.17 | -- | 90.11 | -- | -- |
| Fixed Carbon | 18.65 | -- | 8.65 | -- | -- |
| Ash | 0.18 | -- | 1.24 | 1.6 | -- |
| - Solvent Fractionation \% |  |  |  |  |  |
| Pentane insolubles | 45.15 | -- | 33.58 | 39.5 | -- |
| Asphaltene | 35.65 | -- | 25.77 | -- | -- |
| Benzene insolubles | 9.50 | -- | 7.81 | -- | -- |

The ultimate analysis results of CANMET and Syncrude pitch samples in this research are similar to those reported by previous workers [8, 10, 106].

### 3.3 Experimental Apparatus

The CANMET pitch and Syncrude pitch were pyrolyzed with TGA at low heating rates using U.H.P. Nitrogen as purge gas, and with Pyroprobe-GC at high heating rates using U.H.P. Helium as carrier gas. The volatile yield (or the weight loss) was recorded with TGA dynamically as a function of temperature via a computer. The weight of the pitch sample and the residue of

Pyroprobe-GC pyrolysis was recorded before and after each run at the selected operating conditions. The weight loss in the Pyroprobe pyrolysis can then be calculated by subtracting the residue weight from the initial sample weight. The volatiles were swept into the on-line GC for analysis of the chemical composition.

### 3.3.1 Thermogravimetric Analysis (TGA) Setup and Operation

The pyrolysis of CANMET and Syncrude pitches was performed on a Perkin-Elmer TGS2 TGA. The model TGS-2 (referred to as the TGA in the following text) is designed for accurately recording the weight loss (or volatile content) of a sample as it is subject to a precisely controlled temperature environment. It is capable of controlled heating rates of 0.31 to 320 ${ }^{\circ} \mathrm{C} / \mathrm{min}$. It is a completely modular system consisting of the following units: the thermo-balance analyzer; the electronic balance control unit; the heater control unit; the temperature program control unit, data acquisition computer, plotter and purge gas system.

The balance system consists of a Perkin-Elmer AR-2 recording balance (including an analyzer and balance control unit) which can be used together with a recorder as a recording balance independently of the other components. The temperature program control is the unit which provides the control over the starting temperature, heating rate, stopping temperature and holding time. The heater control unit is a power supply source which provides the controls for calibrating the furnace so that the sample temperature is that temperature indicated on the programmer readout. It provides thermocouple circuitry for monitoring the temperature of the sample environment.

In order to record the weight loss versus temperature information, the temperature was calibrated each two weeks and when the furnace was changed. An inert purge gas was also used to avoid oxidation of samples and volatiles during each run.

The TGA temperature is controlled through a closed-loop, heater-sensor resistance thermometer circuit, using the furnace winding as both sensor and heater. Reproducible, linear temperature programs are thus achieved. A calibration must be performed, however, to make sure that the temperature at any given moment is that specified during experimental runs.

The calibration is first performed at the factory, where adjustments are made to assure the temperature of the sample agrees with the program temperature. After operating the instrument for a period of time, calibration is also necessary to assure best temperature control accuracy.

The calibration can be accomplished by changing the heater control unit range and zero settings to force agreement between the program temperatures and the thermocouple temperatures (or a magnetic transition standard, a Curie point calibration standard). However, a more convenient method was used, employing a calibration routine built in the heater control unit. This routine automatically checks and corrects the thermocouple temperature at three program temperature points.

The calibration routine forces correspondence between the program and sensor temperature at $\mathrm{T} \operatorname{MIN}\left(50^{\circ} \mathrm{C}\right), \mathrm{T} \operatorname{MAX}\left(1000^{\circ} \mathrm{C}\right)$ and the temperature midway between T MIN and T MAX. The calibration sequence is begun by pressing the CALIBRATE and RESET keys on the control unit keypad. The control unit then programs to T MIN , waits for thermal equilibrium, and measures the difference between the sample temperatures and program temperatures. It then corrects the furnace set-point, allows equilibrium, and again checks for agreement. This procedure is repeated until the discrepancy is less than $0.5^{\circ} \mathrm{C}$. The above procedure is repeated for the intermediate temperature and T MAX and the TGA is then considered calibrated.

The control unit forces the sample temperature and the program temperature to agree exactly at 3 points, and approximates a correction for the rest of the scale. The control unit
interpolates correction between T MIN and T MAX, so that the TGA is calibrated for the whole temperature range. When the calibration is completed, the program temperature and actual temperature agrees within $2^{\circ} \mathrm{C}$ or better $[8,62]$.

The positions of the furnace and the sample pan are very important for correct temperature control. The position of the furnace itself can be changed horizontally or vertically by using the adjustments under the furnace support assembly. The ideal position of the furnace is in the center of the furnace assembly as shown in Figure 3.1. A more detailed sketch of the TGA furnace is also shown in the same figure. The top of the furnace should be 10 mm below the anticonvection shield and the top of the stirrup should be recessed by about 1 to 2 mm into the furnace. The bottom of the sample pan should be 2 mm above the tip of the thermocouple. If it is not, another hangdown wire should be prepared, having the appropriate length in order to obtain the best performance.


Figure 3.1 The relative position of the furnace and sample pan on the left and the TGA furnace sketch on the right

The thermal balance was continuously purged with inert U.H.P. Nitrogen gas when samples were being pyrolyzed in order to prevent decomposition products from flowing up and contaminating the balance mechanism and oxidation. A 20 minute purge was also applied before each run. The U.H.P. Nitrogen flowrate was set $100 \mathrm{~mL} / \mathrm{min}$ and checked before each run.

The TGA had been calibrated at the factory so that when the instrument is set up using the proper configuration of furnace height, hangdown wire length, the temperature accuracy should be within one percent over the temperature range of the instrument. Temperature calibration was always made using a U.H.P. Nitrogen gas to achieve the same conductivity as an experimental run.

The pitch sample was applied to the sample pan carefully into a thin layer to achieve a better temperature uniformity and therefore temperature readings.

Once the temperature calibration was achieved, the temperature control unit was used to control the pyrolysis temperature. Different heating rates and final temperatures were used to study their effects on the pyrolysis of CANMET and Syncrude pitches. The heating rates employed in this study are $25,50,100$, and $150^{\circ} \mathrm{C} / \mathrm{min}$, the final temperatures $700,750,800$, $850,900,950^{\circ} \mathrm{C}$. The following temperature program was used to achieve this conditions:

- Purge the TGA system for 20 minutes at room temperature before starting the run and then increase the furnace temperature to $50^{\circ} \mathrm{C}$.
- Hold at $50^{\circ} \mathrm{C}$ for 5 minutes and then ramp to the final temperature at each selected heating rate.
- Hold at that final temperature for 10 minutes, then terminate the run and decrease the temperature to room temperature.

The sample temperature and weight (of sample as well as residue after certain pyrolysis) at any time was recorded using a computer data logger. The weight of sample was also recorded
at the beginning of each run. At any time, the remaining sample weight was recorded as the percentages of the original sample weight. The information of weight and temperature was then recorded into the computer, printed out as hardcopies, and converted into data files. The data files were used in the subsequent analysis and modeling.

### 3.3.2 Pyroprobe-GC

The pyroprobe-GC is a relatively new type of equipment constructed for dynamic analysis of pyrolysis products from the probe by using in-line Gas Chromatography. The main advantages of this piece of equipment are the temperature programmable probe, high temperature ramping rates and small quantity of samples required in the GC analysis. The Pyroprobe-GC consists of the following modular units: Pyroprobe 1000 controller, Pyroprobe interface, Varian GC 3600, Computer Workstation, and gas system, as shown in Figure 3.2.

The CDS Instruments Pyroprobe 1000 is a resistively heated platinum filament pyrolyzer which prepares samples for analysis by gas chromatography. The Pyroprobe 1000 controller calculates the resistance of the filament and supplies the proper voltage needed to achieve the setpoint temperature. Heating rates are selectable in increments of $0.01{ }^{\circ} \mathrm{C}$ per millisecond to 20 ${ }^{\circ} \mathrm{C}$ per millisecond. Final temperature ranges in $1^{\circ} \mathrm{C}$ increments to a maximum of $1400{ }^{\circ} \mathrm{C}$. Final holding time may be selected from 0.01 seconds to 99.99 seconds. All parameters are entered by simple key strokes on the front panel of the controller module.

Samples may be pyrolyzed using a variety of filament designs. The standard model Pyroprobe 1000 includes a coil element and a ribbon element. The coil element, which heats samples held in a quartz tube, was used to pyrolyze the pitch samples in order to record the weight of the sample and the residue to calculate the volatile yield.

The gas chromatograph interface for the Pyroprobe is a heated chamber which houses the probe during pyrolysis. This chamber attaches to the injection port of the gas chromatograph by means of a welded needle nut assembly which replaces the septum retainer. Carrier gas is brought into the interface, sweeps through the heated chamber containing the probe and exits through the needle nut assembly into the injection port of the gas chromatograph.


Figure 3.2 The Pyroprobe-GC setup

All flow entering the injection port comes from the interface. It is important to remember that the Pyroprobe interface is plumbed upstream from the column, and opening the chamber for probe placement permits air to enter the chromatographic system. Therefore, probe placement and removal should be performed when the column is cool to prevent oxidation of the column liquid phase.

The Pyroprobe interface was installed (Figure 3.3) by inserting it between the gas chromatograph carrier gas flow controller and the injection port. The standard interface has three gas fittings and one electrical connection. The electrical connector attaches to the rear of the Pyroprobe controller to supply current to heat the interface and permits temperature monitoring. The three gas fittings are: 1) a large opening in the front for the interface to accept the probe; 2) a
$1 / 8^{\prime \prime}$ Swagelok fitting which attaches the interface to the welded needle nut assembly of the injection port of the gas chromatograph; and 3) a length of stainless steel tubing with a $1 / 16^{\prime \prime}$ Swagelok fitting to connect to GC carrier flow. The large opening for the probe may be sealed with an interface retainer to permit syringe injections directly into the interface. A more detailed sketch of the pyroprobe head is shown in Figure 3.3a.


Figure 3.3 The installation of Pyroprobe interface into the GC injection port.


Figure 3.3a The sketch of Pyroprobe with pitch sample applied on the inner surface of quartz tube

The $1 / 16^{\prime \prime}$ stainless steel tubing must be connected to the carrier gas for the GC column. Flow is disconnected from the injection port and the inlet there capped while the flow is connected to the Swagelok fitting on the end of the $1 / 16^{\prime \prime}$ tubing. This will bring GC flow into the
interface, where it proceeds through the probe chamber and then into the injection port through the needle nut assembly. For pyrolysis, the probe seal in the collar of the probe makes a gas tight connection while the probe is in the interface. This seal was checked and replaced regularly to insure sealing.

A sample of around 5 mg was applied uniformly onto the middle section of the innersurface of the quartz tube which then was inserted into the Pyroprobe heating coil. The quartz tube is 1 "long and $1 / 8^{\prime \prime}$ in diameter. The heating coil is interfaced with the GC station as shown in Figure 3.3. The pyrolysis product is purged into the GC injection port by Helium carrier gas.

Proper sample handling plays a very important role in achieving reproducible pyrolysis. Best results are obtained by using as small a sample as possible to prevent thermal gradient effects and to insure that the sample is completely pyrolyzed. It is important to remember that the Pyroprobe is being used as a sample introduction device for the gas chromatograph and the sample size should be consistent with what is generally injected onto the column. The best reproducibility was obtained using samples of about 5 mg .

The Pyroprobe 1000 was used to control the heating rates and final temperature. The temperature was calibrated according to the calibration number of the heating coil supplied by the manufacturer. Heating rates employed in this study are $600,3000,30000$ and $300000{ }^{\circ} \mathrm{C} / \mathrm{min}$, and final temperatures are $500,600,700,800,900,1000{ }^{\circ} \mathrm{C}$. Pyrolysis times used are $0,5,10$ seconds. The following temperature program was used:

- Purge the Interface for 20 minutes at room temperature with U.H.P. Helium.
- Ramp to the final temperature at the selected heating rate.
- Hold at that final temperature for the selected pyrolysis time, then terminate the run and decrease the temperature to room temperature.
- Through the experiments, the interface temperature was kept at $50^{\circ} \mathrm{C}$.

In the GC, a J\&W DB-5HT fused silica capillary column was used. It is comprised of three major parts. Polymide is used to coat the exterior of the fused silica tubing to protect the fused silica tubing from breaking. The stationary phase is a polymer that is evenly coated onto the inner wall of the tubing. The predominant stationary phases are silicon based polymers (polysiloxanes), polyethlene glycols (PEG, Carbowax ${ }^{\mathrm{TM}}$ ) and solid adsorbents. The liquid phase in this column is DB-5HT. The column is 30 meters long with a diameter of 0.255 mm , and a film thickness of $0.10 \mu \mathrm{~m}$. The column can be operated from $-60^{\circ} \mathrm{C}$ to $400{ }^{\circ} \mathrm{C}$. In this setup the column was installed to FID and PID detectors. U.H.P. Helium is selected as the carrier gas for this capillary column. The carrier gas flow rate was then optimized during test runs as $1 \mathrm{~mL} / \mathrm{min}$.

The operation of the GC is controlled using the computer workstation. The GC and the Pyroprobe were started at the same time for each run. The GC analysis results were also gathered through this computer. The results can be printed out as hardcopies (including chromatograph and analysis results).

Due to the fact that this piece of equipment had not been widely used in the pyrolysis kinetic studies, a great deal of effort was required to configure the equipment and optimize the experimental conditions. This step consumed some four months of experimental time. The optimal conditions for pitch pyrolysis were found to be:

- Purging the interface chamber for 20 min . before starting a run.
- GC column temperature program: $40^{\circ} \mathrm{C}$ for 10 min ., ramping to $120^{\circ} \mathrm{C}$ at the rate of $2^{\circ} \mathrm{C} / \mathrm{min}$. and holding the final temperature for 10 min .
- GC column carrier gas flow rate $1 \mathrm{~mL} / \mathrm{min}$ U.H.P. Helium.
- The Hydrogen flow rate is $20 \mathrm{~mL} / \mathrm{min}$, and the air flow rate is $375 \mathrm{~mL} / \mathrm{min}$.

A summary of the Pyroprobe-GC parameters used by the computer program is listed in Appendix B.

The weights of the sample and residue were recorded before and after each run. The volatile yield was then calculated by subtracting the residue weight from original sample weight at each condition. The FID analysis results of the released volatiles were logged with the computer workstation and used for subsequent recalculation and analysis for both pitches.

A typical Chromatogram is shown in Figure 3.4. The insert is the enlargement of the chromatogram for the period 8 to 45 min . The peaks indicate the major products. As can be seen, most of the pyrolysis products elutes within 5 min . Other products were also identified between retention time 8 and 45 min as shown in Figure 3.4. Syncrude pitch pyrolysis volatile analysis showed a similar chromatogram.

It is clear that it is difficult to identify each of the large number of peaks in Figure 3.4. A grouping scheme was therefore employed to simplify the identification and quantification processes. Similar lumping schemes have been successfully used in coal pyrolysis to estimate the yields of tar and gases [ $1,17,50]$. Inseparable peaks were therefore grouped into six single peaks. The retention time of those groups are listed in Table 3.2 for the volatile of both pitches. The identification of species and quantification of yields are discussed in the following section.

Table 3.2 Important Peak Lumps on the Pyrolysis-GC Chromatograms

| Lump No. | Retention Time min. | Mid-point min. |
| :---: | :---: | :---: |
| 1 | $0.01-4.73$ | 2.370 |
| 2 | $8.84-15.86$ | 12.350 |
| 3 | $19.17-23.97$ | 21.570 |
| 4 | $24.24-31.72$ | 27.980 |
| 5 | $32.11-36.01$ | 34.060 |
| 6 | $37.49-42.66$ | 40.070 |



Figure 3.4 Chromatogram of CANMET pitch volatiles


Figure 3.5 Chromatogram of standard sample

### 3.3.3 Peak Identification and Quantification

In order to identify species from the chromatograms, standard samples of paraffin $\mathrm{C}_{6}-\mathrm{C}_{\mathbf{1 6}}$ and aromatics $\mathrm{C}_{6}-\mathrm{C}_{14}$ were obtained and analyzed individually for retention time. The retention times of the peaks of interest for both CANMET and Syncrude pitch volatiles match those of paraffin: n-Heptane, n-Decane, n-Undecane, n-Dodecane, $n$-Tridecane, and n-Tetradecane. A standard sample was then designed according to the individual retention time of each standard sample and the characteristics of the chromatogram obtained for CANMET pitch and Syncrude pitch pyrolysis products. The standard sample consists of equal amount of $n$-Heptane, $n$-Decane, n-Undecane, $n$-Dodecane, $n$-Tridecane, and $n$-Tetradecane ( $\mathrm{C}_{7}, \mathrm{C}_{10}, \mathrm{C}_{11}, \mathrm{C}_{12}, \mathrm{C}_{13}, \mathrm{C}_{14}$ ). The standard sample analysis chromatogram is shown in Figure 3.5. The insert is the enlargement of the chromatogram from 9 to 45 min . The retention times of aromatics were detected separately and listed in Table 3.3 for comparison. As can be seen, the retention times fall into those of the volatile lumps and close to that of each paraffin component with the same carbon number. The retention time of each component in this standard sample is listed in Table 3.3.

Table 3.3 Retention Time of Each Component

| Paraffins | Retention Time min | Aromatics | Retention Time min |
| :--- | :---: | :--- | :---: |
| Hexane $\mathrm{C}_{6}$ | 1.489 | Benzene $\mathrm{C}_{6}$ | 1.780 |
| Heptane $\mathrm{C}_{7}$ | 1.933 | Toluene $\mathrm{C}_{7}$ |  |
| Octane $\mathrm{C}_{8}$ | 2.959 | Xylene $\mathrm{C}_{8}$ | $4.684(\mathrm{p})$ |
| Nonane $\mathrm{C}_{9}$ | 6.208 | Cumene $\mathrm{C}_{9}$ | 6.473 |
| Decane $\mathrm{C}_{10}$ | 12.025 | Butylbenzene $\mathrm{C}_{10}$ | 13.500 |
| Undecane $\mathrm{C}_{11}$ | 20.882 |  |  |
| Dodecane $\mathrm{C}_{12}$ | 27.894 |  |  |
| Tridecane $\mathrm{C}_{13}$ | 34.051 |  |  |
| Tetradecane $\mathrm{C}_{14}$ | 40.030 | Octylbenzene $\mathrm{C}_{14}$ | 43.212 |

The peak identification was based on two criteria:

- the time at which the peak elutes (retention time) and
- the size of the peak (response)

Both these criteria were used to identify not only peaks of interest, but also to eliminate from consideration those peaks that are not analytically significant (because of retention time or relative size). The quantification was then performed according to an external standard.

External standard calculation allows one to determine the absolute amount of the compounds of interest, without regard to the total area or height, or the area or height of any other peaks in the chromatogram. The peaks of interest must be identified in a peak table, and the detector response is calibrated to these peaks by injecting a known amount of each compound in a run to determine the Calibration Factor.

Peak lump to 4.73 minutes may contain lighter gases up to $\mathrm{C}_{9}$. However, it was impossible to separate this lump into detailed peaks in a practical time scale with the column being used since the wide spectrum of the components in the volatile. It was therefore lumped as one peak and estimated using the response factor of $\mathrm{C}_{7}$. The yield therefore obtained is a rough estimation. The heavier components were lumped in the same fashion. The yield of each is also an estimation.

Following identification of the peaks in the chromatogram, the yields were calculated according to the parameters specified through the computer station. The results can be calculated to meet the analytical requirements. The yields of each component were then calculated using an external standard as outlined in the Varian Star Computer System User Handbook.

In the external standard calculations, peaks were reported in amounts. The calculation in this study gave results in weight (mg). External standard calculation was also done in two stages. First, Calibration Factors developed during a Calibration run are stored in the computer program, then, during an Analysis run, these factors are used to produce the final calculated results.

Calibration Factors for External Standard calculation are absolute factors that are not relative to any component and are based upon an absolute amount injected. The following equation is the formula used to develop Calibration Factors for External Standard calculations:

$$
F A C T O R_{i}=\frac{A M O U N T_{i} \times A M T S T D}{A R E A_{i}} \times 10000
$$

AMOUNT $_{\mathrm{i}}: \quad$ Peak $_{\mathrm{i}}$ AMOUNT in Peak Table.
AMT STD: Amount Standard 1.000, constant
AREA $_{i}: \quad$ the Peak $_{i}$ area.
10000: constant used to calculate the scale factor.
The following equation shows the formula used for External Standard calculations during an analysis run.

$$
R E S U L T S_{i}=\frac{A R E A_{i} \times F A C T O R_{i}}{D I V I S O R \times 10000} \times M L T P L R
$$

AREA $_{i}: \quad$ Peak $_{i}$ area.
DIVISOR: Divisor 1.000, constant
FACTOR $_{i}: \quad$ Peak $_{\mathrm{i}}$ FACTOR in Peak Table is used for identified Peaks
MLTPLR: Multiplier 1.000, constant
10000: constant used to compensate for scaled factor.
RESULT $_{\mathrm{i}}$ : Final External Standard calculation results, mg.
The operation parameters used with TGA, Pyroprobe and GC are summarized in
Table 3.4.
Table 3.4 The Summary of Operation Parameters Used by TGA, Pyroprobe and GC

|  | TGA | Pyroprobe | GC |
| :---: | :---: | :---: | :---: |
| Purge Time min | 20 | 20 | 20 |
| Purge Gas/Flow Rate $\mathrm{mL} / \mathrm{min}$ | 100 | 1 | 1 |
| Initial <br> Temperature | 50 ( 5 min ) | 50 | 40 (10 min) |
| Heating Rate ${ }^{\circ} \mathrm{C} /$ min | 25,50,100, 150 | $\begin{aligned} & 600,3000,30000, \\ & 300000 \\ & \hline \end{aligned}$ | 2 |
| Final Temperature ${ }^{\circ} \mathrm{C}$ | $\begin{aligned} & \hline 700,750,800, \\ & 850,900,950 \\ & \hline \end{aligned}$ | $\begin{array}{\|l\|} \hline 500,600,700, \\ 800,900,1000 \\ \hline \end{array}$ | 120 |
| Holding Time | 10 min | 0, 5, 10 s | 10 min |

## Chapter 4 Experimental Results

### 4.1 TGA Experimental Results

### 4.1.1 TGA Pyrolysis of CANMET Pitch

The TGA pyrolysis of CANMET pitch was performed under different experimental conditions to study the effects of sample weight, heating rate, and final pyrolysis temperature. The sample weight was varied between 4.4 and 17.2 mg . The heating rates employed were $25,50,100$ and $150{ }^{\circ} \mathrm{C} / \mathrm{min}$ and final temperatures of 700 to $950{ }^{\circ} \mathrm{C}$ in $50^{\circ} \mathrm{C}$ increments. Each run was performed with a 10 minute holding time at final temperature.

### 4.1.1.1 Effect of Sample Weight

The sample weight effect on CANMET pitch pyrolysis was investigated under heating rates of $50^{\circ} \mathrm{C} / \mathrm{min}$ and $100^{\circ} \mathrm{C} / \mathrm{min}$ and final temperature $900^{\circ} \mathrm{C}$ for different sample weights ranging from 4.4 to 17.2 mg . The operating conditions and experimental results are provided in Tables 4.1.1 and 4.1.2. $\mathrm{V}_{\mathrm{t}=0}$ and $\mathrm{V}_{\mathrm{t}=10}$ refer to the total volatile yield (or weight loss) in percentage of the original sample weight at 0 minute and 10 minutes pyrolysis reaction time at the final temperature. For this pitch, some $80 \%$ is converted into volatiles, and about $20 \%$ is left as solid residue under these conditions. The shapes of the chromatograms will be discussed in Section 4.1.3. Here just the final residue numbers are discussed.

Figures 4.1.1 and 4.1.2 show that the weight loss decreased (the solid residue increased) with increases in sample weight. This may indicate an internal mass transfer effect. With larger sample sizes, the volatile release from the residue matrix may be hindered, resulting in more char. It appears that the decrease is not linear and the weight loss exhibited a shallow minimum at a sample weight of about 14 mg for runs at $100^{\circ} \mathrm{C} / \mathrm{min}$ and 15 mg at $50^{\circ} \mathrm{C} / \mathrm{min}$, at both zero and ten minute holding times. Since the slight increase appears at both heating rates, a polynomial
rather than a straight line fit was done to illustrate the general trend of data. The weight loss reduced from $82.89 \%$ at sample weight 4.406 mg to $79.85 \%$ at sample weight 15.78 mg for $\mathrm{t}=0$ minute, while the weight loss reduced from $83.64 \%$ at sample weight 4.406 mg to $80.03 \%$ at sample weight 15.78 mg for $\mathrm{t}=10$ minutes for runs at $100^{\circ} \mathrm{C} / \mathrm{min}$ heating rate. For runs at 50 ${ }^{\circ} \mathrm{C} / \mathrm{min}$ heating rate, the weight loss reduced from $84.38 \%$ at sample weight 4.979 mg to $80.07 \%$ at sample weight 17.17 mg for $\mathrm{t}=0$ minute, while the weight loss reduced from $82.89 \%$ at sample weight 4.979 mg to $79.9 \%$ at sample weight 17.17 mg for $\mathfrak{t}=10$ minutes. It is also clearly shown that the effect of holding time at any sample weight on the total weight loss is not significant for CANMET pitch, i.e. essentially all the reaction occurs during heating to the final temperature for each heating rate. A longer holding time may result in more residual $\mathrm{H}_{2}$ release from solid char, but the amount is very small. This is in good agreement with the analysis of Nguyen [107], where only $1.56 \%$ of $\mathrm{H}_{2}$ content was observed in the delayed coke. It is generally believed that the weight loss at this stage is caused by the $\mathrm{H}_{2}$ release from the remaining char [ $9,17,30$ ]. At lower heating rate, the results appear more scattered (Figure 4.1.2), and it may be caused by the longer pyrolysis time. It is expected that at sample weights below 14 mg , the pyrolysis process may be dominated by chemical reaction processes while at higher sample weights diffusional effect may occur. For reference, a single spherical particle of pitch of 14 mg would have a diameter of 1.5 mm . The statistical analysis of the sample size is shown no mass transfer effect in the range of $7.774 \sim 12.034 \mathrm{mg}$ at $100^{\circ} \mathrm{C} / \mathrm{min}$ and $8.011 \sim 13.157 \mathrm{mg}$ at $50^{\circ} \mathrm{C} / \mathrm{min}$ (Appendix I). The difference of weight loss as shown in Figures 4.1.1 and 4.1.2 (and figures in the following sections) is believed not the consequences of experimental errors. For the remaining work, the size of about 9 mg is used. It is believed that the results reflect the intrinsic kinetics and are not significantly affected by mass transfer. As will be subsequently shown, the calculated activation energy is greater than the range $8 \sim 24 \mathrm{~kJ} / \mathrm{mol}$ typical of diffusion processes.


Figure 4.1.1 Sample weight effect on CANMET pitch pyrolysis with TGA at $900^{\circ} \mathrm{C}$ and $100^{\circ} \mathrm{C} / \mathrm{min}$


Figure 4.1.2 Sample weight effect on CANMET pitch pyrolysis with TGA at $900^{\circ} \mathrm{C}$ and $50^{\circ} \mathrm{C} / \mathrm{min}$

### 4.1.1.2 Effect of Heating Rate

To study the heating rate effect on the pyrolysis total weight loss with TGA, the heating rates were set at $25,50,100$ and $150^{\circ} \mathrm{C} / \mathrm{min}$ with final temperatures of $700^{\circ} \mathrm{C}$ and $800^{\circ} \mathrm{C}$. The sample weight was held constant around 8.216 to 9.668 mg in order to minimize the sample size effect. The operating conditions for the experiments are provided in Table 4.1.3. The volatile yield is the weight loss which occurred when the final temperature was reached, i.e. the holding time was zero.

Figure 4.1.3 shows the heating rate results with different final set temperatures. From this plot, it is observed that the total weight loss decreases weakly as heating rate is increased at both temperatures. At the same heating rate, the weight loss (volatile yield) is essentially the same for both temperatures, especially at heating rates smaller than $100^{\circ} \mathrm{C} / \mathrm{min}$. The total weight loss reached $80.84 \%$ and $77.92 \%$ at $25^{\circ} \mathrm{C} / \mathrm{min}$ and $150^{\circ} \mathrm{C} / \mathrm{min}$ for $800^{\circ} \mathrm{C}$ respectively. A decrease of $3 \%$ is observed due to the increase of the heating rate by a factor of six. The weight loss at 700 ${ }^{\circ} \mathrm{C}$ decreased to $79.93 \%$ at $150{ }^{\circ} \mathrm{C} / \mathrm{min}$ from $81.59 \%$ at $25^{\circ} \mathrm{C} / \mathrm{min}$. However, the decrease in total weight loss caused by either the temperature and heating rates is marginal. This indicates that the pyrolysis process is nearly complete at the temperature of $700^{\circ} \mathrm{C}$ and further increase of the temperature does not significantly increase the total weight loss. In the range of low heating rates studied with TGA, the volatile "precursors" apparently have enough time to decompose and evolve from the sample, therefore no significant difference of weight loss is observed.

The effect of heating rate was also studied at much higher level with the Pyroprobe using heating rates of $600,3000,30,000$ and $300,000{ }^{\circ} \mathrm{C} / \mathrm{min}$ and final temperatures of $700{ }^{\circ} \mathrm{C}$ and 800 ${ }^{\circ} \mathrm{C}$. The sample weight was held around 5.02 to 5.58 mg . The operating conditions are provided in Table 4.1.4. The volatile yield is the weight loss occurred when the final temperature is reached.


Figure 4.1.3 Heating rate effect on CANMET pitch pyrolysis with TGA
Figure 4.1.4 shows that with the Pyroprobe, the weight loss decreases nonlinearly with the increase of heating rate. The weight loss decreased from $49.06 \%$ for heating rate $600^{\circ} \mathrm{C} / \mathrm{min}$ to $2 \%$ for $300,000^{\circ} \mathrm{C} / \mathrm{min}$ at $800^{\circ} \mathrm{C}$, while the weight loss decreased from $12 \%$ for heating rate 600 ${ }^{\circ} \mathrm{C} / \mathrm{min}$ to $1.89 \%$ for $300,000^{\circ} \mathrm{C} / \mathrm{min}$ at $700^{\circ} \mathrm{C}$. At very high heating rates, the weight loss is essentially the same for the two final temperatures. This suggests that the reaction time is an important factor. At high heating rates ( $>10,000{ }^{\circ} \mathrm{C} / \mathrm{min}$ ), the reaction time is extremely short, and the difference of weight loss is small. At low heating rates, the reaction time is long, the difference of weight loss is therefore greater. At very low heating rates, the components have enough time to undergo chemical changes, then the same weight loss would be observed. This is shown in TGA results at $\leq 100^{\circ} \mathrm{C} / \mathrm{min}$.

Figure 4.1.5 compares the results using the TGA and the Pyroprobe. Results using the two procedures appear consistent. The weight loss decreased with increased heating rates over the full range studied, i.e., $25^{\circ} \mathrm{C} / \mathrm{min}$ to $300,000^{\circ} \mathrm{C} / \mathrm{min}$. The weight loss decreased from $81.79 \%$ for
heating rate $25^{\circ} \mathrm{C} / \mathrm{min}$ to $1.8 \%$ for $300,000{ }^{\circ} \mathrm{C} / \mathrm{min}$ at $700^{\circ} \mathrm{C}$ and decreased from $80.84 \%$ for heating rate $25^{\circ} \mathrm{C} / \mathrm{min}$ to $2 \%$ for $300,000^{\circ} \mathrm{C} / \mathrm{min}$ at $800^{\circ} \mathrm{C}$. It is also observed that the temperature is a significant parameter between heating rate $100^{\circ} \mathrm{C} / \mathrm{min}$ to $30,000^{\circ} \mathrm{C} / \mathrm{min}$, which indicates the pyrolysis is reaction controlled. At heating rates higher than $30,000{ }^{\circ} \mathrm{C} / \mathrm{min}$, the weight loss is much less than that at heating rate lower than $150^{\circ} \mathrm{C} / \mathrm{min}$. The effect of heating rate may be due to pyrolysis reaction times. At heating rates above $30,000^{\circ} \mathrm{C} / \mathrm{min}$, it takes less than 1.6 seconds to reach the final temperature of $800^{\circ} \mathrm{C}$, while it takes 320 seconds to reach the same temperature at $150^{\circ} \mathrm{C} / \mathrm{min}$.

The rapid drop-off in Figures 4.1.5 and 4.1.9 which occurs for CANMET pitch heated to $700^{\circ} \mathrm{C}$ and for Syncrude pitch may be caused by some combination of time and temperature effect. However, the reasons that it did not occur for CANMET pitch heated to $800^{\circ} \mathrm{C}$ are not obvious.


Figure 4.1.4 Heating rate effect on CANMET pitch pyrolysis with Pyroprobe-GC


Figure 4.1.5 Heating rate effect on CANMET pitch pyrolysis with TGA and Pyroprobe-GC

### 4.1.1.3 Effect of Final Temperature

The effect of final temperature was studied at the heating rate $100^{\circ} \mathrm{C} / \mathrm{min}$ and final temperatures of $700,750,800,850,900$ and $950{ }^{\circ} \mathrm{C}$ with 10 minute holding time. The sample weight was held roughly constant ( $8.13 \sim 11.31 \mathrm{mg}$ ) for all the runs. The operating conditions are provided in Table 4.1.5. Volatile yields are reported for both zero and ten minute holding time.

Figure 4.1 .6 is the weight loss at zero and ten minute holding times vs. final temperature plot. It is observed that the weight loss decreased slightly, reached a minimum and then increased with the increase of temperature. At 0 min holding time, the weight loss decreased from $79.74 \%$ at $700{ }^{\circ} \mathrm{C}$ to minimal weight loss $79.01 \%$ at $850^{\circ} \mathrm{C}$ and then increased to $81.58 \%$ at $950{ }^{\circ} \mathrm{C}$. While it decreased from $80.20 \%$ at $750{ }^{\circ} \mathrm{C}$ to minimal weight loss $79.39 \%$ at $850{ }^{\circ} \mathrm{C}$ and then increased to $81.58 \%$ at $950^{\circ} \mathrm{C}$ for 10 minute holding time. An increase of weight loss of less than $0.5 \%$ was observed over the 10 minute holding time. The residue is already solid char at the temperature $700^{\circ} \mathrm{C}$. That may indicate that pyrolysis of the pitch samples is nearly complete and
that further weight loss may be caused by the release of residue hydrogen in the char matrix at high temperature. The minimal weight loss at $850^{\circ} \mathrm{C}$ reflects the complexity of the pitch pyrolysis chemistry. Similar phenomena was also observed by van Krevelen [17]. However, it is yet to be investigated.

### 4.1.2 TGA Pyrolysis of Syncrude Pitch

The TGA pyrolysis of Syncrude pitch was performed under different experimental conditions to study the effects of heating rate, pyrolysis temperature and sample weight. The sample weight was controlled between 3 and 16 mg . The heating rates employed are $25,50,100$ and $150^{\circ} \mathrm{C} / \mathrm{min}$ and predefined final temperatures of $700,750,800,850,900$ and $950{ }^{\circ} \mathrm{C}$. Each run was also performed with a 10 minute holding time.


Figure 4.1.6 Final temperature effect on CANMET pitch pyrolysis with TGA at $100^{\circ} \mathrm{C} / \mathrm{min}$

### 4.1.2.1 Effect of Sample Weight

The sample weight effect on Syncrude pitch pyrolysis was investigated under the final temperatures of $700,800^{\circ} \mathrm{C}$ and heating rate $100^{\circ} \mathrm{C} / \mathrm{min}$ for different sample weight from 3 mg to 16 mg . The operating conditions are provided in Table 4.1.6. Volatile yields are about $90 \%$, leaving $10 \%$ of the pitch as non-volatile residue under these conditions.

Figures 4.1 .7 and 4.1 .8 show that the weight loss decreased as the increase of sample weight for Syncrude pitch, as also observed for CANMET pitch. It is also observed that only slightly higher weight loss is obtained at 10 minute holding time over 0 minute (Table 4.1.6), i.e., almost all reactions occur during the heatup period. These results are in good agreement with those of CANMET pitch. With sample weight increasing from 3 to 14 mg , the weight loss decreased from $92.73 \%$ to $89.89 \%$ for 0 minute holding time while it decreased from $93.51 \%$ to $90.4 \%$ for 10 minute holding time for runs at $700^{\circ} \mathrm{C}$. For runs at $800{ }^{\circ} \mathrm{C}$, the weight loss decreased from $91.63 \%$ to $90.29 \%$ for 0 minute holding time, while it decreased from $91.89 \%$ to $91.36 \%$ for 10 minutes. The decrease of weight loss happened mostly with sample weights from 3 to 8 mg . Only a very slight decrease of volatile yield was observed with further increases of sample weight.

A comparison of the above results in Figures 4.1.7 and 4.1.8 show that higher weight loss is obtained under higher final pyrolysis temperature for the sample weight higher than 6 mg , while lower weight loss is observed under higher final pyrolysis temperature for a sample size less than 6 mg . This indicated a very complex reaction mechanism and the temperature plays a very important role.


Figure 4.1.7 Sample weight effect on Syncrude pitch pyrolysis with TGA at $100^{\circ} \mathrm{C} / \mathrm{min}$ and 0 min


Figure 4.1.8 Sample weight effect on Syncrude pitch pyrolysis with TGA at $100^{\circ} \mathrm{C} / \mathrm{min}$ at 10 min .

### 4.1.2.2 Effect of Heating Rate

The heating rate was varied from 25 to $300,000{ }^{\circ} \mathrm{C} / \mathrm{min}$ while that final temperature was held constant at $800^{\circ} \mathrm{C}$ (Table 4.1.7). The volatile yield is the weight loss which had occurred when the final temperature was reached.

Figure 4.1 .9 is the comparison of the weight loss results of TGA and Pyroprobe. It is observed that the weight loss decreased with the heating rates over the range studied. The weight loss decreased from $90.6 \%$ for heating rates less than $150{ }^{\circ} \mathrm{C} / \mathrm{min}$ to below $9 \%$ above $600^{\circ} \mathrm{C} / \mathrm{min}$ at a final temperature of $800{ }^{\circ} \mathrm{C}$. The trend of results observed is in rough agreement with those of CANMET pitch pyrolysis shown in Figure 4.1.5. At heating rates higher than $3000{ }^{\circ} \mathrm{C} / \mathrm{min}$, the weight loss is much less than that at heating rates lower than $150{ }^{\circ} \mathrm{C} / \mathrm{min}$ due to the different pyrolysis reaction times. The slower the heating rate, the longer the reaction time, and the more weight loss occurs.


Figure 4.1.9 Heating rate effect on Syncrude pitch pyrolysis with TGA and Pyroprobe-GC ( 0 minute after reaching $800^{\circ} \mathrm{C}$ )

### 4.1.2.3 Effect of Final Temperature

The effect of final temperature on weight loss was studied at the heating rates of 50 and $150^{\circ} \mathrm{C} / \mathrm{min}$ and final temperatures of 750,850 and $950{ }^{\circ} \mathrm{C}$ for holding time 0 and 10 minutes. The sample weight was held in a range of 6.9 to 7.5 mg for runs under those conditions (Table 4.1.8).

Figures 4.1.10 and 4.1.11 are the weight loss vs. final temperature plots for runs at different heating rates and holding times. At zero holding time, the weight loss increased slightly with the increase of temperature at the higher heating rate. The weight loss increased from $90.18 \%$ at $750{ }^{\circ} \mathrm{C}$ to $92.66 \%$ at $950{ }^{\circ} \mathrm{C}$ for the heating rate $150{ }^{\circ} \mathrm{C} / \mathrm{min}$, while the weight loss remained essentially constant at $91 \%$ from $750^{\circ} \mathrm{C}$ to $950^{\circ} \mathrm{C}$ for the heating rate $50^{\circ} \mathrm{C} / \mathrm{min}$. More total weight loss is observed at $150^{\circ} \mathrm{C} / \mathrm{min}$ than $50^{\circ} \mathrm{C} / \mathrm{min}$ at temperature higher than $800^{\circ} \mathrm{C}$. At $50{ }^{\circ} \mathrm{C} / \mathrm{min}$ and $850^{\circ} \mathrm{C}$, the weight loss was lowest, but the sample size was larger, and from Figure 4.1.2 with CANMET pitch, one should expect a lower weight loss. For 10 minute holding time (Figure 4.1.11), the results are essentially similar to those of at zero holding time.


Figure 4.1.10 Final temperature effect on Syncrude pitch pyrolysis with TGA at 0 min


Figure 4.1.11 Final temperature effect on Syncrude pitch pyrolysis with TGA at 10 min
From Table 4.1.8, slightly higher weight loss was observed at 10 minute holding time. The effect of final temperature, as well as holding time, is in accordance with those of CANMET pitch.

### 4.1.3 TGA Pyrolysis Characteristics

In experiments presented in this section, the pyrolysis heating rate was varied while other parameters such as the final temperature and the sample weight were held constant. The sample weight for CANMET pitch is 8.129 to 10.12 mg and the sample weight for Syncrude pitch is 9.904 to 11.90 mg to permit a direct comparison (Table 4.1.9). The total weight loss $\left(\mathrm{V}^{*}\right)$ is also listed in the table for each run. The volatile yield $\left(\mathrm{V}^{*}\right)$ is obtained when the final temperature is reached. The dynamic weight change during the time of heating is discussed.

CANMET and Syncrude pitches both showed similar patterns in the TGA pyrolysis plots. This pattern differs from results found with oil shale or coals. Figures 4.1.12 to 4.1 .18 show the nonisothermal devolatilization TGA curves of CANMET pitch and Syncrude pitch at $800^{\circ} \mathrm{C}$ final
temperature and heating rates of $25,50,100$ and $150^{\circ} \mathrm{C} / \mathrm{min}$. The nonisothermal devolatilization weight loss vs. temperature behavior is shown in Figures 4.1.12 and 4.1.16 for each pitch respectively. It is observed that a slightly higher weight loss is obtained at a lower heating rate at a given temperature, or a higher temperature is required to reach the same amount of weigh loss for a higher heating rate. However, the effect of the heating rates is not systematic, nor significant. For CANMET pitch, weight loss at 25 and $50^{\circ} \mathrm{C} / \mathrm{min}$ is noticeably higher than those at 100 , and $150{ }^{\circ} \mathrm{C} / \mathrm{min}$, while the weight loss is roughly the same for heating rates 25 and 50 ${ }^{\circ} \mathrm{C} / \mathrm{min}$ at the same temperature as shown in Figure 4.1.12. For Syncrude pitch (Figure 4.1.16), the weight loss is almost the same at heating rates 25 and $50^{\circ} \mathrm{C} / \mathrm{min}$. Also the weight loss is roughly the same at 100 and $150{ }^{\circ} \mathrm{C} / \mathrm{min}$. However the weight loss at 25 and $50{ }^{\circ} \mathrm{C} / \mathrm{min}$ is generally higher that that at 100 and $150{ }^{\circ} \mathrm{C} / \mathrm{min}$ at the same temperature. Similar behavior was also observed by Milosavljevic [61], but heating rate as such was not considered to be the main reason for the difference. He claimed that the chemical reaction itself caused the change and difference. This seems reasonable in the present case as well. Heating rates do affect the temperature history, however, it is the chemical reaction at the specific temperature which causes formation of volatiles and the weight loss. This is also observed in Figures 4.1.13 and 4.1.17, which show the weight loss results vs. time. As can be seen, the heating rate changed the reaction time, but it did not change the volatile evolution pattern with temperature of either CANMET pitch or Syncrude pitch.

Figures 4.1.15 and 4.1.19 showed the weight loss per degree of temperature rise $\mathrm{dW} / \mathrm{dT}$ vs. temperature for each pitch. This derivative was calculated with the following formula:

$$
\left(\frac{d W}{d T}\right)_{i}=\frac{W_{i+1}-W_{i}}{T_{i+1}-T_{i}}
$$

The above equation indicates that $\mathrm{dW} / \mathrm{dT}$ is the average value of weight loss in a very small temperature interval and represents the weight loss rate divided by the heating rate. $\mathrm{dW} / \mathrm{dT}$ is also negative because the pyrolysis is a weight loss process with temperature. It is clearly shown that the $\mathrm{dW} / \mathrm{dT}$ changes with the temperature in a nonlinear manner, passing through three major stages for each type of pitch. At temperatures lower than $150^{\circ} \mathrm{C}, \mathrm{dW} / \mathrm{dT}$ is roughly equal to 0 as observed in Figures 4.1.15 and 4.1.19 for CANMET and Syncrude pitches respectively. This indicates that there is no chemical or physical reaction taking place below this temperature, and that the content of water and low molecular components is negligible. At temperatures between $1500^{\circ} \mathrm{C}$ to about $400^{\circ} \mathrm{C}$, the weight loss $\mathrm{dW} / \mathrm{dT}$ slowly decreased to a steady value, which is more evident for the Syncrude pitch results, then $\mathrm{dW} / \mathrm{dT}$ decreased rather dramatically to its minimum, which occurs at temperatures between $500^{\circ} \mathrm{C}$ and $600^{\circ} \mathrm{C}$. The ratio $\mathrm{dW} / \mathrm{dT}$ then went through the last stage of changing, increasing from its minimum to a very small absolute value at approximately $600^{\circ} \mathrm{C}$. At this condition pyrolysis is nearly complete and further increases of the temperature did not affect the total weight loss significantly. This indicated that the temperature is an important parameter and the change of temperature affects the behavior of the pitch pyrolysis process. It is clearly shown that the pyrolysis process takes place as a two stage process and therefore there are two weight loss peaks as observed in these two plots. However these two stages of pyrolysis overlap and this feature can be easily missed in Figure 4.1.15 for CANMET pitch as they are not clearly separated. This two-peak weight loss feature, i.e. two-stage reaction characteristics is more clearly shown in Figure 4.1.19 for Syncrude pitch. The peak weight loss temperature is also very close to a fixed value for all the heating rates studied for each pitch as shown in Figures 4.1.15 and 4.1.19. This further suggests the chemical nature of the pyrolysis. The first peak temperature is not clearly identifiable for CANMET pitch, but lies in the range of $400^{\circ} \mathrm{C}$ and $450{ }^{\circ} \mathrm{C}$ for Syncrude pitch. The second maximum weight loss rate temperature is
clearly identifiable for both CANMET pitch and Syncrude pitch. The second peak temperature for CANMET pitch is between $500^{\circ} \mathrm{C}$ to $600^{\circ} \mathrm{C}$. It is even better defined for Syncrude pitch in the temperature range of $500{ }^{\circ} \mathrm{C}$ and $550{ }^{\circ} \mathrm{C}$. The weight loss for CANMET pitch at $400{ }^{\circ} \mathrm{C}$ is between $5 \%$ and $25 \%$ depending on the heating rate, while the weight loss for Syncrude pitch is between $20 \%$ and $40 \%$ at the same temperature. The total weight loss for CANMET pitch and Syncrude pitch is $80 \%$ and $90 \%$ at $800{ }^{\circ} \mathrm{C}$ respectively. The most weight loss therefore occurred at temperatures between $400^{\circ} \mathrm{C}$ and $600^{\circ} \mathrm{C}$. The weight loss in this temperature range is $65 \%$ to $75 \%$ for CANMET pitch and $50 \%$ to $70 \%$ for Syncrude pitch respectively. Figures 4.1.13 and 4.1.17 showed the weight loss vs. time for each pitch at different heating rates. Figures 4.1.14 and 4.1.18 showed the weight loss rate $\mathrm{dW} / \mathrm{dt}$ vs. time for each pitch at different heating rate. It is also observed that the pyrolysis occurs in stages at different time scales with changes in heating rate. The two peak weight loss character is also identified in these two plots, attesting the results in Figures 4.1.15 and 4.1.19.


Figure 4.1.12 CANMET pitch weight loss vs. temperature at different heating rates and final temperature $800^{\circ} \mathrm{C}$ measured via TGA


Figure 4.1.13 CANMET pitch weight loss vs. time at different heating rates and final temperature $800^{\circ} \mathrm{C}$ measured via TGA


Figure 4.1.14 CANMET pitch weight loss rate vs. time at different heating rates and final temperature $800^{\circ} \mathrm{C}$ measured via TGA


Figure 4.1.15 CANMET pitch weight loss $\mathrm{dW} / \mathrm{dT}$ vs. temperature at different heating rates and final temperature $800^{\circ} \mathrm{C}$ measured via TGA


Figure 4.1.16 Syncrude pitch weight loss vs. temperature at different heating rates and final temperature $800^{\circ} \mathrm{C}$ measured via TGA


Figure 4.1.17 Syncrude pitch weight loss vs. time at different heating rates and final temperature $800^{\circ} \mathrm{C}$ measured via TGA


Figure 4.1.18 Syncrude pitch weight loss rate vs. time at different heating rates and final temperature $800^{\circ} \mathrm{C}$ measured via TGA


Figure 4.1.19 Syncrude pitch weight loss $\mathrm{dW} / \mathrm{dT}$ vs. temperature at different heating rates and final temperature $800^{\circ} \mathrm{C}$ measured via TGA

### 4.1.4 Discussion and Conclusion

It is shown that the heating rates slightly affect the weight loss, however, it is believed that the temperature history, not the heating rate as such causes the difference. Temperature is the significant factor causing the reactions to take place and produce the weight loss. The devolatilization step is not instantaneous, as little weight loss occurred at the highest heating rate.

The importance of the temperature history is more significantly noticed among the runs of Pyroprobe experiments where total reaction time in the heatup was short, i.e. a few seconds. Low heating rates produce longer reaction times of the order of minutes, caused more extensive pyrolysis reaction, and therefore resulted in a higher weight loss (or volatile yield).

At temperatures below $150^{\circ} \mathrm{C}$, there is little weight loss, suggesting that no pyrolysis take place. The weight loss takes place in two following stages with two different, distinct patterns of chemical and physical change. In the first stage, a low peak weight loss rate was observed, while
in the second stage a higher peak weight loss rate was observed. These features appear unique to pitch pyrolysis, as they have not been reported for coal or shale pyrolysis.

The total weight loss (volatile yield) decreases slightly with the increase of sample weight over the range studied for both CANMET pitch and Syncrude pitch.

With TGA, more than $80 \%$ of residue conversion can be achieved for CANMET pitch, while more than $90 \%$ of residue conversion can be achieved for Syncrude pitch.

Table 4.1.1 Experimental Conditions for Runs at Different Sample Weight with TGA

| Run\# | Heating Rate ${ }^{\circ} \mathrm{C} / \mathrm{min}$ | $\begin{gathered} \text { Final Temp } \\ { }^{\circ} \mathrm{C} \end{gathered}$ | Sample Weight mg | $\begin{aligned} & \mathbf{V}_{\mathrm{R}=0} \\ & \mathrm{wt} \% \\ & \hline \end{aligned}$ | $\begin{aligned} & V_{k=10} \\ & w+\% \end{aligned}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Canl1 | 100 | 900 | 4.406 | 82.8 | 83.64 |
| Can20 | 100 | 900 | 5.702 | 82.51 | 82.82 |
| Can 18 | 100 | 900 | 6.441 | 82.05 | 83.50 |
| Can45 | 100 | 900 | 7.074 | 80.60 | 81.55 |
| Can15 | 100 | 900 | 7.774 | 79.80 | 80.47 |
| Can 16 | 100 | 900 | 7.981 | 81.65 | 81.79 |
| Can 17 | 100 | 900 | 10.719 | 80.27 | 80.50 |
| Can8 | 100 | 900 | 11.162 | 79.33 | 79.57 |
| Can38 | 100 | 900 | 12.034 | 79.59 | 79.83 |
| Can 19 | 100 | 900 | 13.680 | 78.70 | 79.03 |
| Can14 | 100 | 900 | 15.784 | 79.85 | 80.03 |

Table 4.1.2 Experimental Conditions for Runs at Different Sample Weight with TGA

| Run\# | Heating Rate ${ }^{\circ} \mathrm{C} / \mathrm{min}$ | $\begin{gathered} \text { Final Temp } \\ { }^{\circ} \mathrm{C} \end{gathered}$ | Sample Weight mg | $\begin{aligned} & V_{t=0} \\ & w t \% \end{aligned}$ | $\begin{aligned} & \mathrm{V}_{\mathrm{E}=10} \\ & \mathrm{wt} \% \\ & \hline \end{aligned}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Can28 | 50 | 900 | 4.979 | 82.89 | 84.38 |
| Can21 | 50 | 900 | 6.360 | 82.76 | 83.32 |
| Can7 | 50 | 900 | 6.723 | 82.48 | 86.29 |
| Can27 | 50 | 900 | 8.011 | 80.51 | 80.65 |
| Can27 | 50 | 900 | 8.943 | 81.12 | 81.38 |
| Can10 | 50 | 900 | 10.179 | 80.60 | 81.82 |
| Can25 | 50 | 900 | 11.162 | 80.25 | 80.45 |
| Can23 | 50 | 900 | 11.735 | 80.88 | 81.02 |
| Can35 | 50 | 900 | 12.022 | 80.43 | 80.74 |
| Can22 | 50 | 900 | 12.699 | 79.97 | 80.35 |
| Can13 | 50 | 900 | 13.157 | 80.84 | 81.71 |
| Can31 | 50 | 900 | 14.042 | 78.47 | 78.87 |
| Can24 | 50 | 900 | 14.729 | 77.85 | 78.26 |
| Can30 | 50 | 900 | 15.596 | 78.49 | 78.87 |
| Can12 | 50 | 900 | 17.175 | 79.10 | 80.07 |


| Run\# | Heating Rate ${ }^{\circ} \mathrm{C} / \mathrm{min}$ | Final Temp. ${ }^{\circ} \mathrm{C}$ | Sample Weight mg | Volatile wt \% |
| :---: | :---: | :---: | :---: | :---: |
| Can54 | 25 | 700 | 9.368 | 81.78 |
| Can61 | 50 | 700 | 8.835 | 80.66 |
| Can53 | 100 | 700 | 7.896 | 79.74 |
| Can60 | 150 | 700 | 8.923 | 79.93 |
| Can48 | 25 | 800 | 8.878 | 80.84 |
| Can33 | 50 | 800 | 8.224 | 80.79 |
| Can41 | 100 | 800 | 10.304 | 79.30 |
| Can58 | 150 | 800 | 9.109 | 77.59 |

Table 4.1.4 Experimental Conditions for Runs at Different Heating Rates with Pyroprobe

| Run\# | Heating Rate ${ }^{\circ} \mathrm{C} /$ min | Final Temp. ${ }^{\circ} \mathrm{C}$ | Sample Weight mg | Volatile wt \% |
| :---: | :---: | :---: | :---: | :---: |
| Cam069 | 600 | 700 | 5.0 | 12.00 |
| Cam051 | 3,000 | 700 | 5.6 | 10.71 |
| Cam033 | 30,000 | 700 | 5.2 | 9.62 |
| Cam015 | 300,000 | 700 | 5.3 | 1.88 |
| Cam070 | 600 | 800 | 5.3 | 49.06 |
| Cam052 | 3,000 | 800 | 5.8 | 29.31 |
| Cam034 | 30,000 | 800 | 5.2 | 7.69 |
| Cam016 | 300,000 | 800 | 5.0 | 2.00 |

Table 4.1.5 Experimental Conditions for Runs at Different Final Temperature with TGA

| Run\# | Heating Rate ${ }^{\circ} \mathrm{C} / \mathrm{min}$ | Final Temp. ${ }^{\circ} \mathrm{C}$ | Sample Weight mg | $\begin{gathered} V_{t=0} \\ \mathbf{w t} \% \end{gathered}$ | $\begin{aligned} & V_{k=10} \\ & w t \% \end{aligned}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Can53 | 100 | 700 | 7.896 | 79.74 | 80.20 |
| Can42 | 100 | 750 | 9.171 | 79.54 | 79.81 |
| Can41 | 100 | 800 | 10.304 | 79.30 | 79.72 |
| Can40 | 100 | 850 | 10.723 | 79.01 | 79.39 |
| Can38 | 100 | 900 | 12.034 | 79.59 | 79.83 |
| Can52 | 100 | 950 | 8.199 | 81.23 | 81.58 |

Table 4.1.6 Experimental Conditions for Runs at Different Sample Weight with TGA

| Run\# | Heating Rate ${ }^{\circ} \mathrm{C} / \mathrm{min}$ | Final Temp. ${ }^{\circ} \mathrm{C}$ | Sample Weight mg | $\begin{aligned} & V_{t=0} \\ & \text { wt } \% \\ & \hline \end{aligned}$ | $\begin{aligned} & V_{t=10} \\ & w t \% \end{aligned}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Syn13 | 100 | 700 | 3.010 | 92.73 | 93.51 |
| Syn14 | 100 | 700 | 7.852 | 90.54 | 90.84 |
| Syn16 | 100 | 700 | 11.029 | 90.28 | 90.60 |
| Syn 15 | 100 | 700 | 13.932 | 89.83 | 90.40 |
| Syn 17 | 100 | 800 | 4.321 | 91.63 | 91.89 |
| Syn19 | 100 | 800 | 6.797 | 91.10 | 91.39 |
| Syn18 | 100 | 800 | 11.376 | 90.58 | 90.94 |
| Syn20 | 100 | 800 | 15.534 | 90.29 | 91.36 |

Table 4.1.7 Experimental Conditions for Runs at
Different Heating Rates with TGA and Pyroprobe

| Run\# | Heating Rate ${ }^{\circ} \mathrm{C} / \mathrm{min}$ | $\begin{gathered} \text { Final Temp } \\ { }^{\circ} \mathrm{C} \\ \hline \end{gathered}$ | Sample Weight mg | $\begin{aligned} & \mathbf{V}_{\mathrm{t}=0} \\ & \mathbf{w} \% \end{aligned}$ | Equipment |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Syn43 | 25 | 800 | 10.477 | 91.03 | TGA |
| Syn 29 | 50 | 800 | 11.708 | 90.70 | TGA |
| Syn 18 | 100 | 800 | 11.376 | 90.59 | TGA |
| Syn8 | 150 | 800 | 10.053 | 90.62 | TGA |
| Syn070 | 600 | 800 | 4.600 | 8.70 | Pyroprobe |
| Syn052 | 3,000 | 800 | 2.400 | 8.33 | Pyroprobe |
| Syn034 | 30,000 | 800 | 3.600 | 2.78 | Pyroprobe |
| Syn016 | 300,000 | 800 | 4.900 | 4.08 | Pyroprobe |

Table 4.1.8 Experimental Conditions for Runs at Different Final Temperature with TGA

| Run\# | Heating Rate ${ }^{\circ} \mathrm{C} /$ min | Final Temp. ${ }^{\circ} \mathrm{C}$ | Sample Weight mg | $\begin{aligned} & V_{t=0} \\ & w t \% \end{aligned}$ | $\begin{aligned} & V_{t=10} \\ & w t \% \end{aligned}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Syn27 | 50 | 750 | 7.604 | 90.96 | 91.02 |
| Syn32 | 50 | 850 | 7.134 | 90.61 | 90.87 |
| Syn33 | 50 | 950 | 6.942 | 91.01 | 91.21 |
| Syn10 | 150 | 750 | 7.606 | 90.18 | 90.51 |
| Syns | 150 | 850 | 6.920 | 91.19 | 91.22 |
| Syn4 | 150 | 950 | 7.262 | 92.66 | 93.05 |

Table 4.1.9 The Pyrolysis Conditions for CANMET Pitch and

| Run\# | Heating Rate ${ }^{\circ} \mathrm{C} / \mathrm{min}$ | $\begin{aligned} & \text { Final Temp. } \\ & { }^{\circ} \mathrm{C} \end{aligned}$ | $\begin{gathered} \mathbf{v}^{*} \\ \mathbf{w t} \% \\ \hline \end{gathered}$ |
| :---: | :---: | :---: | :---: |
| CANMET Pitch |  |  |  |
| Can48 | 25 | 800 | 80.84 |
| Can33 | 50 | 800 | 80.79 |
| Can41 | 100 | 800 | 79.30 |
| Can58 | 150 | 800 | 77.59 |
| Syncrude Pitch |  |  |  |
| Syn43 | 25 | 800 | 91.03 |
| Sym29 | 50 | 800 | 90.70 |
| Syn18 | 100 | 800 | 90.58 |
| Syn8 | 150 | 800 | 90.62 |

### 4.2 Pyroprobe-GC Pyrolysis of CANMET and Syncrude Pitch

CANMET and Syncrude pitches were studied with the Pyroprobe-GC. The yield of volatiles was determined by the difference between the sample weight and residue weight after pyrolysis. The yield of each major group of components was determined following the method outlined in the experiment techniques section in Chapter 3. The experimental conditions are summarized in each of the following sections. The mass balance of each run in this section is in the range of 95 to $105 \%$.

### 4.2.1 Pyroprobe-GC Pyrolysis of CANMET Pitch

The Pyroprobe-GC pyrolysis of CANMET pitch was performed under different experimental conditions to study the effects of heating rates, pyrolysis reaction temperatures and holding times. The sample weight was kept relatively constant around 5 mg in order to limit the sample size effects. The heating rates are $300000,30000,3000^{\circ} \mathrm{C} / \mathrm{min}$, the holding times are 10 , 5 and 0 s . The combinations of these operating parameters are listed in Table 4.2.1. Each combination of these parameters was performed at the final temperatures of $500,600,700,800$, $900,1000^{\circ} \mathrm{C}$.

Table 4.2.1 Experimental Conditions for Runs at Different Holding Times

| Holding Time s | Heating Rate ${ }^{\circ} \mathrm{C} / \mathrm{min}$ |
| :---: | :---: |
| 10.0 | 300,000 |
| 5.0 | 300,000 |
| 0.0 | 300,000 |
| 10.0 | 30,000 |
| 5.0 | 30,000 |
| 0.0 | 30,000 |
| 10.0 | 3,000 |
| 5.0 | 3,000 |
| 0.0 | 3,000 |

### 4.2.1.1 Effect of Experimental Conditions on the Total Weight Loss

The total weight loss vs. holding time is an important characteristic in hydrocarbon pyrolysis. The effect of holding times on the total weight loss is shown in Figures 4.2.1 to 4.2.3.

Figures 4.2.1 to 4.2.3 show that the weight loss (volatile yield) generally increases as the increase of temperature, with maximum weight loss observed at heating rate 30,000 and 3,000 ${ }^{\circ} \mathrm{C} / \mathrm{min}$. At the heating rate of $300,000{ }^{\circ} \mathrm{C} / \mathrm{min}$ as shown in Figure 4.2 .1 , higher weight loss is observed for a longer holding time at temperatures below $800^{\circ} \mathrm{C}$, i. e., more weight loss is observed after 10 s than 5 or 0 s . At temperatures higher than $800^{\circ} \mathrm{C}$, however, about the same amount of weight loss is observed at 10 and 5 s . That may indicate that the pyrolysis is nearly complete at these conditions. Little weight loss is observed at 0 s .

At heating rate of $30,000^{\circ} \mathrm{C} / \mathrm{min}$ as shown in Figure 4.2 .2 , it is observed that the weight loss vs. temperature at different holding times is not linear. The maximum weight loss is reached at $900{ }^{\circ} \mathrm{C}$ for holding time 10 and 5 s . Higher weight loss is also observed under longer holding time. About $5 \%$ more weight loss is observed at 10 s holding time than 5 s holding time.

At heating rate of $3000^{\circ} \mathrm{C} / \mathrm{min}$ as shown in Figure 4.2.3, it is observed that the weight loss vs. temperature at different holding times is not linear. The maximum weight loss is observed at $700^{\circ} \mathrm{C}$ for holding time 10 s , and at $800^{\circ} \mathrm{C}$ for 5 s . At temperatures lower than $800^{\circ} \mathrm{C}$, more weigh loss is observed under a longer holding time. At temperatures higher than $900^{\circ} \mathrm{C}$, weight loss becomes less sensitive to the holding time. More weight loss is observed at 0 s for heating rate $3000^{\circ} \mathrm{C} / \mathrm{min}$ than that for 300,000 and $30,000^{\circ} \mathrm{C} / \mathrm{min} .50 .9 \%$ weight loss is observed at 1000 ${ }^{\circ} \mathrm{C}$ and 0 s for heating rate $3000^{\circ} \mathrm{C} / \mathrm{min}$, while less than $5 \%$ weight loss is observed for both 300,000 and $30,000^{\circ} \mathrm{C} / \mathrm{min}$ at the same temperature. This further indicates the importance of holding times. To reach $1000^{\circ} \mathrm{C}$, it takes $19,1.9$ and 0.19 s for $3000,30,000$ and $300,000{ }^{\circ} \mathrm{C} / \mathrm{min}$ respectively.


Figure 4.2.1 CANMET pitch pyrolysis total loss (yield) vs. temperature at different pyrolysis holding times with heating rate $300,000^{\circ} \mathrm{C} / \mathrm{min}$


Figure 4.2.2 CANMET pitch pyrolysis total loss (yield) vs. temperature at different pyrolysis holding times with heating rate $30,000^{\circ} \mathrm{C} / \mathrm{min}$


Figure 4.2.3 CANMET pitch pyrolysis total loss (yield) vs. temperature at different pyrolysis holding times with heating rate $3000^{\circ} \mathrm{C} / \mathrm{min}$

### 4.2.1.2 Effect of Experimental Conditions on the $\mathrm{C}_{7}$ Yield

Figures 4.2 . 4 to 4.2 .6 show that the $C_{7}$ yield generally increases as the increase of temperature, while maximum yield was observed at $3000^{\circ} \mathrm{C} / \mathrm{min}$. At the heating rate of 300,000 ${ }^{\circ} \mathrm{C} / \mathrm{min}$ as shown in Figure 4.2.4, higher $\mathrm{C}_{7}$ yield is observed at a longer holding time in the temperature range studied. The $\mathrm{C}_{7}$ yield is $54.46 \%, 27.92 \%$ and $0 \%$ at $1000^{\circ} \mathrm{C}$ for holding time 10,5 and 0 s respectively.

At heating rate $30,000^{\circ} \mathrm{C} / \mathrm{min}$ as shown in Figure 4.2 .5 , it is observed that the $\mathrm{C}_{7}$ yield vs. temperature at different holding times is not linear. It is also observed that the $\mathrm{C}_{7}$ yield at 10 second holding time is very close to that at 5 second holding time. A maximum $\mathrm{C}_{7}$ yield, $40.94 \%$, is observed at $900^{\circ} \mathrm{C}$ for holding time $5 \mathrm{~s} . \mathrm{C}_{7}$ yield reached $47.77 \%$ and $39.19 \%$ at $1000^{\circ} \mathrm{C}$ for 10 and 5 second holding time respectively. The $\mathrm{C}_{7}$ yield at 0 s is negligible, as also observed in Figure 4.2.4.


Figure 4.2.4 CANMET pitch pyrolysis $\mathrm{C}_{7}$ yield vs. temperature at different pyrolysis holding times with heating rate $300,000^{\circ} \mathrm{C} / \mathrm{min}$


Figure 4.2.5 CANMET pitch pyrolysis $\mathrm{C}_{7}$ yield vs. temperature at different pyrolysis holding times with heating rate $30,000 .{ }^{\circ} \mathrm{C} / \mathrm{min}$


Figure 4.2.6 CANMET pitch pyrolysis $\mathrm{C}_{7}$ yield vs. temperature at different pyrolysis holding times with heating rate $3000^{\circ} \mathrm{C} / \mathrm{min}$

At heating rate of $3000^{\circ} \mathrm{C} / \mathrm{min}$ as shown in Figure 4.2 .6 , it is observed that the $\mathrm{C}_{7}$ yield vs. temperature is not linear. It is also observed that maximum $\mathrm{C}_{7}$ yield is reached at different temperature for different holding times. Maximum $C_{7}$ yield is reached at a lower temperature for a longer holding time. Maximum $\mathrm{C}_{7}$ yield, $43.75 \%$, is reached at $700^{\circ} \mathrm{C}$ for 10 second holding time, while maximum $\mathrm{C}_{7}$ yield, $40.14 \%$ and $41,75 \%$, is reached at $800^{\circ} \mathrm{C}$ and $1000{ }^{\circ} \mathrm{C}$ for holding time 5 and 0 s respectively. Secondary pyrolysis is clearly observed for the $\mathrm{C}_{7}$ lump of compounds. At temperature lower than $750^{\circ} \mathrm{C}$, it is observed that longer holding time resulted in higher $\mathrm{C}_{7}$ yield, while at temperature higher than $900^{\circ} \mathrm{C}$, longer holding time resulted in lower $\mathrm{C}_{7}$ yield. It is also observed that $\mathrm{C}_{7}$ yield increased dramatically at temperatures above $700{ }^{\circ} \mathrm{C}$ for 0 second holding time and $41.75 \%$ is obtained at $1000^{\circ} \mathrm{C}$ for holding time 0 s . The maximum $\mathrm{C}_{7}$ yields at different conditions is essentially the same and may indicate the secondary reactions of some of the components in the sample.

### 4.2.1.3 Effect of Experimental Conditions on the $\mathrm{C}_{10}$ Yield

Figures 4.2.7 to 4.2 .9 show that $\mathrm{C}_{10}$ yield generally increases as temperature, with the maximum yield observed at 800 to $900^{\circ} \mathrm{C}$. At heating rate $300,000{ }^{\circ} \mathrm{C} / \mathrm{min}$ as shown in Figure 4.2.7, higher $\mathrm{C}_{10}$ yield is observed at higher heating rate in the temperature range studied. The $\mathrm{C}_{10}$ yield is not sensitive to temperatures lower than $600^{\circ} \mathrm{C}$. As also observed in Figure 4.2.5, $\mathrm{C}_{10}$ yield is negligible for 0 s in the temperature range studied. Maximum $\mathrm{C}_{10}$ yield is also observed at $800^{\circ} \mathrm{C}$ for holding time 10 s.

At heating rate $30,000^{\circ} \mathrm{C} / \mathrm{min}$ as shown in Figure 4.2.8, it is observed that the $\mathrm{C}_{10}$ yield vs. temperature at different holding times is not linear. It is also observed that the $\mathrm{C}_{10}$ yield at 10 second holding time is very close to that at 5 second holding time. A maximum $\mathrm{C}_{10}$ yield, $2.3 \%$ and $2.2 \%$, is observed at $900^{\circ} \mathrm{C}$ for holding time 10 and 5 s respectively. The $\mathrm{C}_{10}$ yield at 0 s is, as also observed in Figure 4.2.7, negligible.

At heating rate of $3000^{\circ} \mathrm{C} / \mathrm{min}$ as shown in Figure 4.2.9, it is observed that the $\mathrm{C}_{10}$ yield vs. temperature is not linear. It is also observed that maximum $\mathrm{C}_{10}$ yield is reached at different temperature for different holding times. Maximum $C_{10}$ yield is reached at about the same temperature $900{ }^{\circ} \mathrm{C}$ for holding time 10 and 5 s respectively. The maximum $\mathrm{C}_{10}$ yield is $1.8 \%$ and $1.7 \%$ for holding time 10 and 5 s respectively. At temperature lower than $60{ }^{\circ} \mathrm{C}$, it is observed that $\mathrm{C}_{10}$ yield is not sensitive to the temperature. It is also observed that $\mathrm{C}_{10}$ yield for holding time 10 s is close to that for holding time 5 s in the temperature range from 600 to $900^{\circ} \mathrm{C} . \mathrm{C}_{10}$ yield increased significantly at temperature higher than $800^{\circ} \mathrm{C}$ and reached maximum yield $1.6 \%$ at 900 ${ }^{\circ} \mathrm{C}$ for holding time 0 s .

Secondary pyrolysis of $\mathrm{C}_{10}$ lump is also evident as shown in Figures 4.2.7, 4.2.8 and 4.2.9.


Figure 4.2.7 CANMET pitch pyrolysis $\mathrm{C}_{10}$ yield vs. temperature at different pyrolysis holding times with heating rate $300,000^{\circ} \mathrm{C} / \mathrm{min}$


Figure 4.2.8 CANMET pitch pyrolysis $\mathrm{C}_{10}$ yield vs. temperature at different pyrolysis holding times with heating rate $30,000{ }^{\circ} \mathrm{C} / \mathrm{min}$


Figure 4.2.9 CANMET pitch pyrolysis $\mathrm{C}_{10}$ yield vs. temperature at different pyrolysis holding times with heating rate $3000^{\circ} \mathrm{C} / \mathrm{min}$

### 4.2.1.4 Effect of Experimental Conditions on the $\mathrm{C}_{11}$ Yield

Figure 4.1.10 shows that the $\mathrm{C}_{11}$ yield generally increases to $800^{\circ} \mathrm{C}$ and then decreases as temperature at the heating rate of $300,000^{\circ} \mathrm{C} / \mathrm{min}$, with maximum $\mathrm{C}_{11}$ yield observed at $800^{\circ} \mathrm{C}$. Maximum $\mathrm{C}_{11}$ yield of $2.7 \%$ and $1 \%$ is obtained at $800^{\circ} \mathrm{C}$ for holding time 10 and 0 s respectively. There is no $\mathrm{C}_{11}$ observed at temperatures lower than $600^{\circ} \mathrm{C}$ for holding time 10 s , and $700^{\circ} \mathrm{C}$ for holding time 0 s . It is worth noting that the $\mathrm{C}_{11}$ yield is negligible at $1000{ }^{\circ} \mathrm{C}$ for holding time 10 s and temperatures higher than $900^{\circ} \mathrm{C}$ for holding time 0 s , indicating that the $\mathrm{C}_{11}$ lump depleted due to further pyrolysis (i.e. secondary reactions).


Figure 4.2.10 CANMET pitch pyrolysis $\mathrm{C}_{11}$ yield vs. temperature at different pyrolysis holding times with heating rate $300,000{ }^{\circ} \mathrm{C} / \mathrm{min}$

### 4.2.1.5 Effect of Experimental Conditions on the $\mathrm{C}_{12}$ Yield

Figure 4.2.11 shows that the $\mathrm{C}_{12}$ yield generally increases to $800^{\circ} \mathrm{C}$ and then decreases as temperature at the heating rate of $300,000{ }^{\circ} \mathrm{C} / \mathrm{min}$, with maximum $\mathrm{C}_{12}$ yields observed at $800^{\circ} \mathrm{C}$. Maximum $\mathrm{C}_{12}$ yield of $3.7 \%$ and $1 \%$ is obtained at $800{ }^{\circ} \mathrm{C}$ for holding times 10 and 0 s respectively. There is no $\mathrm{C}_{12}$ observed at temperature lower than $600^{\circ} \mathrm{C}$ for holding time 10 s , $700^{\circ} \mathrm{C}$ for holding time 0 s . It is also worth noting that the $\mathrm{C}_{12}$ yield depleted at $900^{\circ} \mathrm{C}$ due to its further pyrolysis at 0 second holding time and also significantly decreased at temperatures above $800^{\circ} \mathrm{C}$ for 10 second holding time.


Figure 4.2.11 CANMET pitch pyrolysis $\mathrm{C}_{12}$ yield vs. temperature at different pyrolysis holding times with heating rate $300,000{ }^{\circ} \mathrm{C} / \mathrm{min}$

### 4.2.1.6 Effect of Experimental Conditions on the $C_{13}$ Yield

Figure 4.2 .12 shows the $\mathrm{C}_{13}$ yield generally increases to $700^{\circ} \mathrm{C}$ and then decreases as temperature at the heating rate of $30,000{ }^{\circ} \mathrm{C} / \mathrm{min}$, with the maximum $\mathrm{C}_{13}$ yield observed $700^{\circ} \mathrm{C}$ for the holding time studied. Higher $C_{13}$ yield is also observed at a longer holding time. The maximum $\mathrm{C}_{13}$ yield, observed at $700^{\circ} \mathrm{C}$, is $2.9 \%, 2.4 \%$ and $2.1 \%$ for holding times 10,5 and 0 s . $\mathrm{C}_{13}$ yield decreased as further increase of temperature. This again indicates secondary pyrolysis of $\mathrm{C}_{13}$ lump at higher temperature.


Figure 4.2.12 CANMET pitch pyrolysis $\mathrm{C}_{13}$ yield vs. temperature at different pyrolysis holding times with heating rate $30,000^{\circ} \mathrm{C} / \mathrm{min}$

### 4.2.1.7 Effect of Experimental Conditions on the $\mathrm{C}_{14}$ Yield

Figures 4.1.13 and 4.1.14 show that the $C_{14}$ yield generally increases to certain temperatures and then decreases as temperature. At the heating rate of $300,000^{\circ} \mathrm{C} / \mathrm{min}$ as shown in Figure 4.2.13, maximum $\mathrm{C}_{14}$ yield is obtained at $800^{\circ} \mathrm{C}$ for holding time 10 s and $900{ }^{\circ} \mathrm{C}$ for holding time 5 s . The maximum yields are $1.7 \%$ and $1.3 \%$ respectively. There is no $\mathrm{C}_{14}$ observed at the temperature range studied for holding time 0 s .

At heating rate $30,000{ }^{\circ} \mathrm{C} / \mathrm{min}$ as shown in Figure 4.2 .14 , it is observed that the $\mathrm{C}_{14}$ yield vs. temperature at different holding times is not linear. Maximum $\mathrm{C}_{14}$ yield is observed at $700{ }^{\circ} \mathrm{C}$ for holding times 10 and 5 s . The maximum yields are $3.89 \%$ and $3.91 \%$ respectively. $\mathrm{C}_{14}$ yield increased as temperature in the range from 500 to $700^{\circ} \mathrm{C}$, decreased in the range from 700 to $1000{ }^{\circ} \mathrm{C} . \mathrm{C}_{14}$ yield for holding time 10 s is close to that for holding time 5 s at the same temperature. Secondary reaction is also evident for $\mathrm{C}_{14}$ as shown in Figures 4.2.13 and 4.2.14.


Figure 4.2.13 CANMET pitch pyrolysis $\mathrm{C}_{14}$ yield vs. temperature at different pyrolysis holding times with heating rate $300,000^{\circ} \mathrm{C} / \mathrm{min}$


Figure 4.2.14 CANMET pitch pyrolysis $\mathrm{C}_{14}$ yield vs. temperature at different pyrolysis holding times with heating rate $30,000^{\circ} \mathrm{C} / \mathrm{min}$

### 4.2.2 Pyroprobe-GC Pyrolysis of Syncrude Pitch

The Pyroprobe-GC pyrolysis of Syncrude pitch was again performed under different experimental conditions to study the effects of heating rates, pyrolysis reaction temperatures and holding times. The sample weight was kept relatively constant around 5 mg in order to limit the sample size effect. The heating rates are $300,000,30,000,3000,600^{\circ} \mathrm{C} / \mathrm{min}$, the holding time is 10,5 and 0 s . The combinations of these operating parameters are listed in Table 4.2.2. Each combination of these parameters was performed at the final temperatures of $500,600,700,800$, $900,1000^{\circ} \mathrm{C}$.

Table 4.2.2 Experimental Conditions for Runs at Different Holding Times

| Holding Time s | Heating Rate ${ }^{\circ} \mathrm{C} / \mathrm{min}$ |
| :---: | :---: |
| 10.0 | 300,000 |
| 5.0 | 300,000 |
| 0.0 | 300,000 |
| 10.0 | 30,000 |
| 5.0 | 30,000 |
| 0.0 | 30,000 |
| 10.0 | 3000 |
| 5.0 | 3000 |
| 0.0 | 3000 |
| 10.0 | 600 |
| 5.0 | 600 |
| 0.0 | 600 |

### 4.2.2.1 Effect of Experimental Conditions on the Total Weight Loss

The total weight loss vs. holding time is an important character for Syncrude pitch pyrolysis as well. The effect of holding times on the total weight loss is shown in Figure 4.2.15.

Figure 4.2.15 shows that the weight loss generally increases as the increase of temperature at $300,000{ }^{\circ} \mathrm{C} / \mathrm{min}$, with maximum yield observed for 5 s holding time. Higher weight loss is observed for 10 s than 5 s or 0 s . It is also noted that the weight loss is not significant at holding time 0 s .


Figure 4.2.15 Syncrude pitch pyrolysis total weight loss vs. temperature at different pyrolysis holding times with heating rate $300,000{ }^{\circ} \mathrm{C} / \mathrm{min}$

### 4.2.2.2 Effect of Experimental Conditions on the $\mathrm{C}_{7}$ Yield

Figure 4.16 shows that the $C_{7}$ yield increases as temperature at the heating rate of 300,000 ${ }^{\circ} \mathrm{C} / \mathrm{min}$, with maximum yield observed for 5 s holding time. Higher $\mathrm{C}_{7}$ yield is observed at a longer holding time. The $\mathrm{C}_{7}$ yield reached $75 \%$ and $60 \%$ at 10 and 5 s holding time respectively, while no $C_{7}$ was detected at all at 0 s . Comparison with Figure 4.1 .5 shows that at high heating rates the $C_{7}$ lump comprises essentially all the weight loss.


Figure 4.2.16 Syncrude pitch pyrolysis $\mathrm{C}_{7}$ yield vs. temperature at different pyrolysis holding times with heating rate $300,000{ }^{\circ} \mathrm{C} / \mathrm{min}$

### 4.2.2.3 Effect of Experimental Conditions on the $\mathrm{C}_{10}, \mathrm{C}_{11}, \mathrm{C}_{12}, \mathrm{C}_{13}$, and $\mathrm{C}_{14}$ Yield

Higher yield of $\mathrm{C}_{10}$ and $\mathrm{C}_{11}$ is generally obtained at a lower holding time and higher temperature at the heating rate of $300,000^{\circ} \mathrm{C} / \mathrm{min}$. However, the yields of these lumps are rather small. The $\mathrm{C}_{10}$ yield reached only $0.7 \%$ and $0.5 \%$ at holding time of 10 and 5 s respectively, while the $\mathrm{C}_{11}$ yield reached only $0.225 \%$ and $0.07 \%$ for holding time 10 and 5 s . At heating rate 600 , $3000,30,000{ }^{\circ} \mathrm{C} / \mathrm{min}$, little $\mathrm{C}_{10}$ and $\mathrm{C}_{11}$ was detected.

The heating rate effect is not an important parameter for $\mathrm{C}_{12}, \mathrm{C}_{13}, \mathrm{C}_{14}$ yield. The increase of heating rates did not show any significant effect on $\mathrm{C}_{12}, \mathrm{C}_{13}, \mathrm{C}_{14}$ yield as observed in the CANMET pitch pyrolysis. The quantity of each of the lumps is not abundant to determine accurately.

### 4.2.3 Discussion and Conclusion

It is shown that under Pyroprobe pyrolysis conditions, the pyrolysis reaction time is a very important operating parameter. At the highest heating rate $\left(300,000^{\circ} \mathrm{C} / \mathrm{min}\right)$ employed in this study, there is little pyrolysis, i.e. weight loss, is observed for both CANMET and Syncrude pitches, while at heating rate of $3000{ }^{\circ} \mathrm{C} / \mathrm{min}$, the weight loss is rather significant when the final temperature is just reached ( 0 s isothermal reaction time). In the latter case, some 10 to $50 \%$ of volatile yield was observed at different final temperature. The latter case is somewhat similar to the TGA experiment results, and the effect of the heating period on the pyrolysis of either CANMET or Syncrude pitch should not be ignored. At heating rate of $30,000^{\circ} \mathrm{C} / \mathrm{min}$, the weight loss results are rather close to those at $300,000{ }^{\circ} \mathrm{C} / \mathrm{min}$, while they are generally higher than those at $3000{ }^{\circ} \mathrm{C} / \mathrm{min}$. The effect of the heating rate combined with the final temperature is therefore expected to be interrelated and remains as a topic of research for high heating rate pyrolysis. However, a different pyrolysis mechanism is also expected for the high heating rate pyrolysis.

The most abundant component of the volatile is shown experimentally the hydrocarbons with less than 10 carbons, which is grouped as single lump as $\mathrm{C}_{7}$ in this study. At each heating rate and final temperature, the amount of $\mathrm{C}_{7}$ is becoming significant at temperatures higher than $700{ }^{\circ} \mathrm{C}$. As high as $50 \%$ volatile yield of this group was detected for CANMET pitch and secondary reaction is observed at heating rate $3000{ }^{\circ} \mathrm{C} / \mathrm{min}$. At the Pyroprobe pyrolysis conditions, the volatile may undergo secondary pyrolysis when being purged through the quartz tube. Similar trend is also observed for Syncrude pitch pyrolysis with Pyroprobe-GC.

The yield of $\mathrm{C}_{10}$ compounds is very strongly influenced by the heating rates. At the highest heating rate $\left(300,000^{\circ} \mathrm{C} / \mathrm{min}\right)$, less than $5 \%$ volatile yield of this group of components was detected, while as high as $23 \%$ volatile yield of the group was detected at $30,000{ }^{\circ} \mathrm{C} / \mathrm{min}$. This again attests the influence of the reaction time and heating rates. The amount of $\mathrm{C}_{10}$ detected from

Syncrude pitch pyrolysis with Pyroprobe-GC is much less than that of CANMET pitch. This is in agreement with the difference of "chemical structure" or "chemical makeup" of these two pitches, where proximate analysis, ultimate analysis and fractionation also show that Syncrude pitch contains more low molecular components than CANMET pitch.

Higher yields of $\mathrm{C}_{11}, \mathrm{C}_{12}, \mathrm{C}_{13}$ and $\mathrm{C}_{14}$ groups was also detected at lower heating rates, a similar trend as that of $\mathrm{C}_{10}$ group. While the yield of $\mathrm{C}_{14}$ is much less that those of $\mathrm{C}_{11}, \mathrm{C}_{12}$ and $\mathrm{C}_{13} \mathrm{C}_{14}$ is the heaviest group of compound detected in the Pyroprobe-GC pyrolysis. This may indicate that the volatile is mostly compounds lighter than $\mathrm{C}_{14}$. The yield of these groups from Syncrude pitch pyrolysis with Pyroprobe-GC is also significantly less than those from CANMET pitch pyrolysis. This is in agreement with the $\mathrm{C}_{10}$ yield.

The different yields of each lumped group between CANMET pitch and Syncrude pitch is in good agreement with the difference of the chemical nature of these two pitch samples. This is also in agreement with the TGA pyrolysis results in which the TGA pyrolysis curves showed different patterns between the above two samples.

Secondary reaction of the product lumps is evident for both CANMET and Syncrude pitch. At high temperatures, heavy lumps such as $\mathrm{C}_{14}$, are prone to pyrolysis into smaller molecules before leaving the quartz tube.

## Chapter 5 Modeling of Experimental Results

### 5.1 Introduction of Pyrolysis Kinetic Models

A number of mechanisms which have been proposed in the literature for pyrolysis were described in the literature review. However, the single overall first order reaction mechanism has been accepted most widely due to its simplicity and adequacy to explain the pyrolysis behavior and to model the process mathematically. The single overall first order reaction model assumed that de-volatilization takes place as a single first order reaction and the mechanism does not change during pyrolysis process. It is widely used to describe and explain the pyrolysis processes of coal, oil shale, bitumen, biomass and other hydrocarbons, due to its mathematical simplicity. A number of first order reaction models were thus proposed to that effect.

The general expression for the first order mechanism is given as:

$$
\begin{equation*}
\frac{d V}{d t}=k_{o} e^{-E / R T}\left(V^{*}-V\right) \tag{5.1}
\end{equation*}
$$

Under nonisothermal conditions, such as those in the TGA experiments, the temperature at any time during the heating period is given by the following expression;

$$
\begin{equation*}
T=C t+T_{o} \tag{5.2}
\end{equation*}
$$

where $T_{0}$ is the initial temperature of the experiment. Substituting time term dt with temperature dT , i.e. $\mathrm{dT}=\mathrm{C}^{*} \mathrm{dt}$, the general expression is then given as:

$$
\begin{equation*}
\frac{d V}{d T}=\frac{k_{o}}{C} e^{-E / R T}\left(V^{*}-V\right) \tag{5.3}
\end{equation*}
$$

where:
$V^{*}$ maximum volatile content released at the final temperature, $w t \%$
T pyrolysis temperature, K
V volatile content released at temperature T, wt\%
$t$ pyrolysis reaction time, min.

E activation energy of the single overall first order reaction, $\mathrm{J} / \mathrm{mol}$
$\mathrm{k}_{0} \quad$ pre-exponential factor of the single overall first order reaction, $\mathrm{min}^{-1}$.
R gas constant, 8.314 J/mol.K
C pyrolysis heating rate, $\mathrm{K} / \mathrm{min}$.

### 5.1.1 Overall First Order Reaction Model

A number of methods have been suggested to extract values of $k_{0}$ and $E$ for Equation 5.3 from experiments in which V is measured as a function of T at constant heating rate. Since they use the experimental data in different forms, they tend to give different results for the reaction parameters.

### 5.1.1.1 Integral Method

This method estimates the values of $E$ and $k_{0}$ of a reaction from the overall volatile yield vs. temperature curves. Shih and Sohn [108] used this method to determine the kinetic parameters for oil shale pyrolysis. The general expression is rearranged as:

$$
\begin{equation*}
\frac{d V}{V^{*}-V}=\frac{k_{o}}{C} e^{-E / R T} d T \tag{5.4}
\end{equation*}
$$

Integrating the above expression in the temperature range of interest, we then get

$$
\begin{equation*}
\int_{0}^{V} \frac{d V}{V^{*}-V}=\int_{T_{o}}^{T} \frac{k_{o}}{C} e^{-E / R T} d T \tag{5.5}
\end{equation*}
$$

where $\mathrm{T}_{0}$ is the initial temperature. In the current study the $\mathrm{T}_{\mathrm{o}}$ is chosen as $50^{\circ} \mathrm{C}$, and the rate as well as the total volatile yield at this temperature is negligible; therefore the temperature limit $T_{0}$ can be by replaced by 0 .

Integration of the above equation gives, with $\mathrm{T}_{\mathrm{o}}$ assumed to be 0 K .

$$
\begin{equation*}
-\ln \left(\frac{V^{*}-V}{V^{*}}\right)=\frac{k_{o}}{C}\left[T e^{-E / R T}+\frac{E}{R} E_{i}\left(-\frac{E}{R T}\right)\right] \tag{5.6}
\end{equation*}
$$

The exponential integral $\mathrm{E}_{\mathrm{i}}(-\mathrm{E} / \mathrm{RT}$ ) can be approximated by (Appendix C):

$$
\begin{equation*}
E_{i}\left(-\frac{E}{R T}\right)=-\frac{e^{-E / R T}}{E / R T}\left(1-\frac{1!}{E / R T}+\frac{2!}{(E / R T)^{2}}-\cdots\right) \tag{5.7}
\end{equation*}
$$

If the first three terms of the approximation are used, the above integration becomes:

$$
\begin{equation*}
-\ln \left(\frac{V^{*}-V}{V^{*}}\right) \approx \frac{k_{o} R T^{2}}{C E}\left(1-\frac{2 R T}{E}\right) e^{-E / R T} \tag{5.8}
\end{equation*}
$$

Dividing both sides of the above equation by $\mathrm{RT}^{2}(1-2 \mathrm{RT} / \mathrm{E}) / \mathrm{C}$ and taking the logarithm, then

$$
\begin{equation*}
\ln \left(\frac{-C \ln \left(1-\frac{V}{V^{*}}\right)}{R T^{2}}\right)-\ln \left(1-\frac{2 R T}{E}\right) \approx \ln \frac{k_{o}}{E}-\frac{E}{R T} \tag{5.9}
\end{equation*}
$$

The values of $E$ and $k_{o}$ can be obtained by repeated least squares fit of the above equation to the experimental data. By first using an approximate E in the left hand side of the above equation, the least squares fit can therefore be performed with the FORTRAN program in Appendix $D$. The value of $E$ thus obtained is then used as the new value on the left hand side and successively a more accurate value of $E$ is obtained until no improvement in the value of $E$ takes place. The values of $E$ and $k_{0}$ are therefore obtained.

From the above equation, the volatile yield V (Appendix C) can be obtained as

$$
\begin{equation*}
V=V^{*}\left\{1-\exp \left[-\frac{k_{o} R T^{2}}{C E} e^{-E / R T}\left(1-\frac{2 R T}{E}\right)\right]\right\} \tag{5.10}
\end{equation*}
$$

### 5.1.1.2 Friedman Method

This method determines the values of E and $\mathrm{k}_{0}$ from the ratio $\mathrm{dV} / \mathrm{dT}$ vs. temperature. Rewriting the general expression (Equation 5.3) as:

$$
\begin{equation*}
\frac{C}{V^{*}} \frac{d V}{d T}=k_{o} e^{-E / R T} \frac{\left(V^{*}-V\right)}{V^{*}} \tag{5.11}
\end{equation*}
$$

Taking the logarithm and rearranging,

$$
\begin{equation*}
\ln \left(\frac{C}{V^{*}} \frac{d V}{d T}\right)-\ln \left(1-\frac{V}{V^{*}}\right)=\ln k_{o}-\frac{E}{R T} \tag{5.12}
\end{equation*}
$$

The values of $E$ and $k_{0}$ can be obtained by fitting the above equation to the experimental data, using least squares fitting program in Appendix D. The values of $\mathrm{dV} / \mathrm{dT}$ are calculated by using two adjacent pairs of the volatile and temperature data:

$$
\begin{equation*}
\left(\frac{d V}{d T}\right)_{i}=\frac{V_{i+1}-V_{i}}{T_{i+1}-T_{i}} \tag{5.13}
\end{equation*}
$$

Where $\mathrm{i}=1, \mathrm{n}-1$. The number of data points in a run is n , and $(\mathrm{dV} / \mathrm{dT})_{\mathrm{n}}=(\mathrm{dV} / \mathrm{dT})_{\mathrm{n}-1}$.
The volatile yield can then be calculated from Equation 5.10 with the values of $E$ and $k_{0}$ obtained.

### 5.1.1.3 Coats-Redfern Method

This method is the same as the integral method except that the term of 2RT/E is ignored in Equation 5.9. This simplifies the mathematical procedure, and is based on the assumption of 2RT/E $\ll 1$.

$$
\begin{equation*}
\ln \left(\frac{-C \ln \left(1-\frac{V}{V^{*}}\right)}{R T^{2}}\right)=\ln \frac{k_{o}}{E}-\frac{E}{R T} \tag{5.14}
\end{equation*}
$$

The values of $E$ and $k_{0}$ can be obtained by fitting the above equation to the experimental data, using the program in Appendix D. The volatile yield can be obtained from Equation 5.10 by using the E and $\mathrm{k}_{0}$ values thus obtained.

### 5.1.1.4 Chen-Nuttall Method

This method assumed the initial temperature to be zero K . The initial temperature of this investigation $\left(50^{\circ} \mathrm{C}\right)$ was taken to be close enough to 0 such that the rate as well as the volatile yield was negligible. The general expression is then given as:

$$
\begin{equation*}
\int_{0}^{V} \frac{d V}{V^{*}-V}=\int_{0}^{T} \frac{k_{o}}{C} e^{-E / R T} d T \tag{5.15}
\end{equation*}
$$

Integration of the above equation gives:

$$
\begin{equation*}
\ln \left(1-\frac{V}{V^{*}}\right)=-\frac{k_{o}}{C} \frac{R T^{2}}{E+2 R T} e^{-E / R T} \tag{5.16}
\end{equation*}
$$

Multiplying both sides of the above equation by $-\mathrm{C}(\mathrm{E}+2 \mathrm{RT}) / \mathrm{RT}^{2}$ and taking logarithms gives:

$$
\begin{equation*}
\ln \left(\frac{-C(E+2 R T)}{R T^{2}} \ln \left(1-\frac{V}{V^{*}}\right)\right)=\ln k_{o}-\frac{E}{R T} \tag{5.17}
\end{equation*}
$$

The values of $E$ and $k_{0}$ can be obtained by repeated least squares fit of the above equation to the experimental data with the same procedure as that of the integral method. By first using an approximate $E$ in the left hand side of the above equation, the least squares fit can therefore be performed with the FORTRAN program in Appendix D and the value of E thus obtained is used in the calculation of the values of the left hand side of the equation and successively a more accurate value of $E$ is obtained until no improvement in the value of $E$ takes place. The values of $E$ and $k_{0}$ are therefore obtained. The volatile yield can also be calculated from Equation 5.10 with the values of $E$ and $k_{0}$ obtained.

### 5.1.2 Multi-First-Order Reaction Model

One of the principal shortcomings of the above four methods is the tacit assumption that a single activation energy and a single pre-exponential factor can adequately describe the evolution of the pyrolysis products. For the case of fossil fuel and especially pitch pyrolysis it is physically
realistic to expected evolution of products (for example $\mathrm{CH}_{4}$ and $\mathrm{H}_{2}$ ) from a wide range of chemically nonequivalent sources. Hence more than one rate constant would be required to describe the pyrolysis process. Anthony and Howard [30] proposed a model to deal with this situation in an attempt to explain the coal devolatilization mechanism. Their model describes the evolution of products by a number of parallel, first order rate processes, each represented by a rate constant k. To simplify the problem, Anthony and Howard [30] assumed that the rate constants have the same pre-exponential factor, and differ only in activation energy, and that the number of parallel reactions is sufficiently large for the activation energies to be described by a Gaussian distribution function. The model and its assumptions have been described in the literature review section in more detail.

Integration of the general expression in the activation range of 0 to $\infty$ gives:

$$
\begin{align*}
V & =V^{*}\left(1-\frac{1}{s(2 \pi)^{0.5}} \int_{0}^{\infty} \exp \left\{\frac{k_{o} R T^{2}}{C E}\left[\exp \left(-\frac{E}{R T}\right)\left(1-\frac{2 R T}{E}\right)\right]\right\}\right.  \tag{5.18}\\
& \exp \left[-0.5\left(\frac{E-E_{0}}{s}\right)^{2}\right] d E
\end{align*}
$$

Due to the complex and nonlinear nature of the model function, nonlinear regression must be used to fit the experimental data for $\mathrm{E}_{0}, \mathrm{k}_{0}$ and s . The Levenberg-Marquart method is thus used in this work. This method adjusts $\mathrm{k}_{\mathrm{o}}, \mathrm{E}_{\mathrm{o}}$ and s within the calculation. Some authors [52] used a fixed $\mathrm{k}_{\mathrm{o}}$ value to simplify the mathematical process and reduce the computing time. However their approach resulted in questionable kinetic parameters. This Levenberg-Marquart method is proven a good nonlinear method. It requires the derivatives of V with respect to each of the three parameters: $k_{0}, E_{0}$ and $s$. In order to use the Levenberg-Marquart method, the derivatives with respect of each parameter must be derived in a specific range of activation energy, using the following general mathematical formula [109]:

$$
\frac{d}{d t} \int_{a(t)}^{b(t)} f(x, t) d x=\int_{a(t)}^{b(t)} f_{t}^{\prime}(x, t) d t+f(b, t) \frac{d b}{d t}-f(a, t) \frac{d a}{d t}
$$

The range of the activation energy is selected as $E_{0}-4 s$ to $E_{0}+4 s$. Further increases in the range of activation energy did not improve the precision of parameters and the accuracy of the volatile prediction. These derivatives have been derived as part of this work as is shown below:

$$
\begin{align*}
\frac{d V}{d E_{0}}= & -\frac{V^{*}}{s(2 \pi)^{0.5}}\left[\int_{E_{0}-4 s}^{E_{0}+4 s} \exp \left\{-\frac{k_{o} R T^{2}}{C E} \exp \left(-\frac{E}{R T}\right)\left(1-\frac{2 R T}{E}\right)\right\} \exp \left\{-0.5\left(\frac{E-E_{0}}{s}\right)^{2}\right\}\right. \\
& \left(\frac{E-E_{0}}{s^{2}}\right) d E+\exp (-8) \exp \left\{-\frac{k_{o} R T^{2}}{C\left(E_{0}+4 s\right)} \exp \left(-\frac{E_{0}+4 s}{R T}\right)\left(1-\frac{2 R T}{E_{0}+4 s}\right)\right\}  \tag{5.19}\\
& \left.-\exp (-8) \exp \left\{-\frac{k_{o} R T^{2}}{C\left(E_{0}-4 s\right)} \exp \left(-\frac{E_{0}-4 s}{R T}\right)\left(1-\frac{2 R T}{E_{0}-4 s}\right)\right\}\right] \\
\frac{d V}{d s}= & \frac{V^{*}}{s^{2}(2 \pi)^{0.5}} \int_{E_{0}-4 s}^{E_{0}+4 s} \exp \left\{-\frac{k_{o} R T^{2}}{C E} \exp \left(-\frac{E}{R T}\right)\left(1-\frac{2 R T}{E}\right)\right\} \exp \left\{-0.5\left(\frac{E-E_{0}}{s}\right)^{2}\right\} d E \\
& -\frac{V^{*}}{s(2 \pi)^{0.5}} \int_{E_{0}-4 s}^{E_{0}+4 s} \exp \left\{-\frac{k_{o} R T^{2}}{C E} \exp \left(-\frac{E}{R T}\right)\left(1-\frac{2 R T}{E}\right)\right\} \exp \left\{-0.5\left(\frac{E-E_{0}}{s}\right)^{2}\right\} \frac{\left(E-E_{0}\right)^{2}}{s^{3}} d E \\
& -\frac{4 V^{*}}{s(2 \pi)^{0.5}} \exp (-8) \exp \left\{-\frac{k_{o} R T^{2}}{C\left(E_{0}+4 s\right)} \exp \left(-\frac{E_{0}+4 s}{R T}\right)\left(1-\frac{2 R T}{E_{0}+4 s}\right)\right\} \\
& -\frac{4 V^{*}}{s(2 \pi)^{0.5}} \exp (-8) \exp \left\{-\frac{k_{o} R T^{2}}{C\left(E_{0}-4 s\right)} \exp \left(-\frac{E_{0}-4 s}{R T}\right)\left(1-\frac{2 R T}{E_{0}-4 s}\right)\right\}
\end{align*}
$$

$$
\begin{equation*}
\frac{d V}{d k_{o}}=-\frac{V^{*}}{s(2 \pi)^{0.5}} \int_{E_{0}-4 s}^{E_{0}+4 s} \exp \left\{-\frac{k_{o} R T^{2}}{C E} \exp \left(-\frac{E}{R T}\right)\left(1-\frac{2 R T}{E}\right)\right\} \tag{5.20}
\end{equation*}
$$

$$
\begin{equation*}
\left(-\frac{R T^{2}}{C E}\right) \exp \left(-\frac{E}{R T}\right)\left(1-\frac{2 R T}{E}\right) \exp \left\{-0.5\left(\frac{E-E_{0}}{s}\right)^{2}\right\} d E \tag{5.21}
\end{equation*}
$$

A FORTRAN program was written to solve the above ODEs and the procedures outlined in Numerical Recipe [110] were followed. The FORTRAN program is listed in Appendix D.

### 5.1.3 Mathematical Methods for Overall Single First Order Reaction Model

In order to use the first order model to fit the parameters, the experimental data, i.e. the measured volatile contents need to be converted according to each method into the form:

$$
\begin{equation*}
Y=a+b X \tag{5.22}
\end{equation*}
$$

$Y$ is the LHS of each of the single overall first order reaction methods, $b$ is equal to $-\mathrm{E} / \mathrm{R}$, and X is the reciprocal of temperature $1 / \mathrm{T}$ in K . The $Y$ and $a$ for each method are listed in Table 5.1.

Table 5.1 $Y$ and $a$ Formulas for the Overall Single First Order Reaction Model

| Methods | $Y$ | $a$ |
| :---: | :---: | :---: |
| Integral | $Y=\ln \left(\frac{-C \ln \left(1-\frac{V}{V^{*}}\right)}{R T^{2}}\right)-\ln \left(1-\frac{2 R T}{E}\right)$ (5.23) | $a=\ln \left(k_{d} / E\right)$ |
| Friedman | $Y=\ln \left(\frac{C}{V^{*}} \frac{d V}{d T}\right)-\ln \left(1-\frac{V}{V^{*}}\right) \quad$ (5.24) | $a=\ln k_{o}$ |
| Coats-Redfern | $Y=\ln \left(\frac{-C \ln \left(1-\frac{V}{V^{*}}\right)}{R T^{2}}\right)$ | $a=\ln \left(k_{d} / E\right)$ |
| Chen-Nuttall | $Y=\ln \left(\frac{-C(E+2 R T)}{R T^{2}} \ln \left(1-\frac{V}{V^{*}}\right)\right)(5.26)$ | $a=\ln k_{o}(5.26 \mathrm{a})$ |

The $Y$ values for the integral method and Chen-Nuttall method were calculated with the first guess of $E$, and then iterated for the best fit for the activation $E$ and pre-exponential factor $k_{0}$. The values of $E$ and $k_{0}$ for the Friedman and Coats-Redfern methods are obtained by using least squares to fit the above equations to experimental data.

### 5.2 Testing of the Basic Models

The volatile yield was checked against the prediction of the four different methods and one model described in the previous section. Each method was used to fit to the experimental data for the pre-exponential factor, the activation energy as well as standard activation energy distribution
for the Anthony and Howard model. The values of the $k_{0}, E$, as well as $s$ were then used to predict the volatile yield. These values are listed for CANMET pitch, along with the results of kinetic parameters for Moroccan oil shale pyrolysis [111] in Table 5.2.


There is a close agreement between the values obtained in this study and those obtained by Thakur and Nuttall [111] except for the $k_{o}$ value of the Anthony Howard model. The kinetic parameters also compare favorably with the literature [112-115] for kerogen pyrolysis to bitumen.

Having obtained the kinetic parameters, the volatile yields can then be predicted using Equation 5.10, which were computed using the program in the Appendix D. The predicted and experimental results are plotted in Figure 5.1. It is clear that these models all failed to predict the volatile contents at temperatures higher than $200{ }^{\circ} \mathrm{C}$ even though the values of the kinetic parameters are well within the expected range for hydrocarbon pyrolysis and agree well with the literature, and the standard deviation (s.e.e.) is small enough. However, the s.e.e. is misleading because it is the average error (Equation 5.31). The difference between experimental data and the model prediction, is up to $30 \%$ as is observed in Figure 5.1. It comes as no surprise that these models failed. The fact that the chemical nature of the "pitch" is changing continuously as the pyrolysis progresses has long been overlooked. Schuckler [116] reported that the activation energy increases markedly with the increase of fractional volatilization $V / V^{*}$. This drastic change
in activation energies coupled with the unusually high preexponential factors at $\mathrm{V} / \mathrm{V}^{*}$ of 0.8 and 0.9 suggested a significant change in the pyrolysis mechanism at high volatile levels. The discrepancy at high temperature in Figure 5.1 is also supported by Thakur and Nuttall [111], who reported that two sets of kinetic parameters are required to fit their experimental data over the whole range.

The Anthony and Howard [30] model takes account of the expected change of activation energy in the fashion of a Gaussian distribution with a constant pre-exponential factor. Although this assumption reflects the fact that the activation energy increases in the pyrolysis process, it does not adequately reflect the rate constant change of either CANMET pitch or Syncrude pitch pyrolysis quantitatively and mechanistically. The additional parameter, $s$, is insufficient to fit the experimental results.

In examining the $Y$ values for each overall single reaction model, it is clear that the assumption of the linear relation between Y values and X is not valid for each of the methods, as shown in Figures 5.2, 5.3, 5.4, and 5.5. Inflection points are observed at $X$ value of 0.0014 (450 ${ }^{\circ} \mathrm{C}$ ). This observation is in accordance to the fact that the ratio of pyrolysis $\mathrm{dV} / \mathrm{dT}$ is dramatically increased at $450{ }^{\circ} \mathrm{C}$ as shown in Figures 4.1.15 and 4.1.19 in Chapter 4. Single step reaction models applied to CANMET pitch over the whole temperature range failed to predicted this basic feature. The fitting results of these models to Syncrude pitch showed similar results, in that the single step model failed to predict the change of pyrolysis rate and volatile yield. The results obtained in the present study indicate that the thermal pyrolysis reactions of pitches are complex to the extent that they can not be described as a single overall first order reaction. Hence the above models (the overall single reaction model analyzed with four different mathematical methods, the Anthony and Howard model analyzed with the Levenberg-Marquart nonlinear
regression method) can not be used to fit the TGA data of CANMET and Syncrude pitch pyrolysis.


Figure 5.1 Comparison of model prediction and experimental volatile for CANMET pitch at $50^{\circ} \mathrm{C} / \mathrm{min}$. and $700^{\circ} \mathrm{C}$ with first order reaction models


Figure 5.2 Comparison of model predicted $Y$ results and experimental $Y$ results for CANMET pitch at $50^{\circ} \mathrm{C} / \mathrm{min}$. and $700^{\circ} \mathrm{C}$ with integral method


Figure 5.3 Comparison of model predicted $Y$ results and experimental $Y$ results for CANMET pitch at $50^{\circ} \mathrm{C} / \mathrm{min}$. and $700^{\circ} \mathrm{C}$ with Coats-Redfern method


Figure 5.4 Comparison of model predicted $Y$ results and experimental $Y$ results for CANMET pitch at $50^{\circ} \mathrm{C} / \mathrm{min}$. and $700^{\circ} \mathrm{C}$ with Chen-Nuttall method


Figure 5.5 Comparison of model predicted $Y$ results and experimental $Y$ results for CANMET pitch at $50^{\circ} \mathrm{C} / \mathrm{min}$. and $700^{\circ} \mathrm{C}$ with Friedman method

### 5.3 2-Stage First Order Reaction Model

Multi-step behavior has been clearly identified in the present data as well as results of Rajeshwar [113], Thakur and Nuttall [111] and Schuckler [116]. Rajeshwar [113], Thakur and Nuttall [111] analyzed their oil shale pyrolysis data with the assumption that the thermal decomposition proceeds in two consecutive steps via a soluble bitumen intermediate, while Schuckler [116] analyzed pyrolysis data of heavy residuum fractions within several volatile conversion intervals to evaluate the kinetic parameters, which indicated a multiple step mechanism instead. Campbell et al. [117] employed nonlinear least squares fit of nonisothermal thermogravimetry data to derive kinetic parameters for a Colorado oil shale sample. Herrell and Arnold [118] report the use of nonisothermal TGA for the study of Chattanooga shale. In both these studies the kinetic data have been interpreted in terms of single step decomposition mechanisms. Such an interpretation, however seems to be contradictory to the conclusions
reached in most of the early studies which indicate that the thermal decomposition of oil shale kerogen proceeds in two consecutive steps. It is noted however, that the concept of reaction order and pre-exponential factor in solid-state kinetics assumes a different significance from that adopted in homogenous reaction kinetics. Topochemical considerations restrict values of the reaction order to $0,1 / 2,2 / 3$, and 1 in solid state kinetics [119, 120]. Normally the order of pyrolysis of a sufficiently small sample is considered to be unity [121]. However, a model for a multi-step process such as that identified in the pyrolysis/thermal decomposition process is not yet a available. In order to describe the pitch pyrolysis and take into account the activation energy change in the model, it is important that the model reflect those features as shown in Figures 5.6 and 5.7.

Figure 5.6 shows the ratio $\mathrm{dV} / \mathrm{dT}$ and the rate $\mathrm{dV} / \mathrm{dt}$ vs. the remaining volatile content $\mathrm{V}^{*}$ V for different heating rates at the final temperature $800^{\circ} \mathrm{C}$ for CANMET pitch. It is shown that the ratio $\mathrm{dV} / \mathrm{dT}$ increase linearly with the increase of the remaining volatile content, up to $25 \%$ remaining volatile content, and then decreases approximately linearly with the increase of the remaining volatile content. It is also noted that the heating rate does not show any influence on the volatile yield rate, i.e. the reaction mechanism. The same value of maximum $\mathrm{dV} / \mathrm{dT}$ is reached at about $25 \%$ remaining volatile content for each heating rate. This suggests that the pyrolysis process of CANMET pitch is chemically controlled. This further indicates that the pyrolysis takes place in two stages with differing mechanisms. In the beginning of the pyrolysis, the rate increases with temperature, and the decrease of the remaining volatile content, up to the maximum value which occurs at the remaining volatile content of $25 \%$. Then the ratio $\mathrm{dV} / \mathrm{dT}$ decreases with increasing temperature, and the decrease of remaining volatile content.

Figure 5.7 shows the ratio $\mathrm{dV} / \mathrm{dT}$ and the rate $\mathrm{dV} / \mathrm{dt}$ vs. the remaining volatile content $\mathrm{V}^{*}$ V for different heating rates at the final temperature $800^{\circ} \mathrm{C}$ for Syncrude pitch. The ratio $\mathrm{dV} / \mathrm{dT}$
increases roughly linearly with the increase of the remaining volatile content, up to about $25 \%$ remaining volatile content, which is the amount of the remaining volatile content also observed for the CANMET pitch pyrolysis. However unlike the CANMET pitch pyrolysis, the ratio dV/dT vs. the remaining $\mathrm{V}^{*}$ - V does not show a single linear relationship to the end of the pyrolysis process. Instead, the ratio $\mathrm{dV} / \mathrm{dT}$ vs. the remaining volatile content $\mathrm{V}^{*}-\mathrm{V}$ decreases approximately linearly to $55 \%$ remaining volatile content, then maintains a steady value $\mathrm{dV} / \mathrm{dT}$ up to $75 \%$ remaining volatile content, and then decreases to nil. This is because there are more lower molecular weight components in the Syncrude pitch than in the CANMET pitch shown by the lower pentane solubles and higher $\mathrm{H} / \mathrm{C}$ atomic ratio in Table 3.1. At the beginning of pyrolysis of the Syncrude pitch, the value increases with the temperature and the decrease of the remaining of the volatile content, then the ratio $\mathrm{dV} / \mathrm{dT}$ maintains a steady value in the range of remaining volatile content of $55 \%$ to $70 \%$. This suggests that lower molecular components undergo mild and rather quick chemical changes in the narrow temperature interval of $300^{\circ} \mathrm{C}$ to $450{ }^{\circ} \mathrm{C}$. The steady value in $\mathrm{dV} / \mathrm{dT}$ is unlikely to be caused by physical changes, such as distillation, because the temperature is too high for distillation of most components existing in pitch samples. The relationship of $\mathrm{dV} / \mathrm{dT}$ vs. $\mathrm{V}^{*}-\mathrm{V}$ of Syncrude pitch shows some similarities to that of CANMET pitch, suggesting a similar pyrolysis pathway, at least up to remaining volatile content of $45 \%$.

Similar patterns as that observed from results of $\mathrm{dV} / \mathrm{dT}$ are also observed in the pyrolysis rate $\mathrm{dV} / \mathrm{dt}$ plots Figure 5.6 b and Figure 5.7 b . The difference in these two graphs as a function of heating rate is as expected, and is caused by the difference of time scale of the pyrolysis process.


Figure 5.6 The devolatilization ratio $\mathrm{dV} / \mathrm{dT}$ vs. the remaining volatile at different heating rates and $800^{\circ} \mathrm{C}$ for CANMET pitch


Figure 5.6 b The devolatilization rate $\mathrm{dV} / \mathrm{dt}$ vs. the remaining volatile at different heating rates and $800^{\circ} \mathrm{C}$ for CANMET pitch


Figure 5.7 The devolatilization ratio $\mathrm{dV} / \mathrm{dT}$ vs. the remaining volatile at different heating rates and $800^{\circ} \mathrm{C}$ for Syncrude pitch


Figure 5.7b The devolatilization rate $\mathrm{dV} / \mathrm{dt}$ vs. the remaining volatile at different heating rates and $800^{\circ} \mathrm{C}$ for Syncrude pitch

### 5.3.1 Multi-Stage First Order Reaction Model and its Assumptions

In order to model the pitch pyrolysis data, it is assumed that the pyrolysis of the pitch samples takes place as a multi-step first order thermal decomposition with regard to the volatile content remaining in the "residue" and is a chemically controlled process. It is also assumed that at some critical temperature, the kinetic parameters undergo change as the "reaction" shifts from one stage to the other stage of the pyrolysis process. In each stage, only one type of reaction dominates, and the kinetics parameters remain relatively constant. Therefore each stage of the reaction can be modeled as a single overall first order reaction. As the reaction proceeds and the temperature increases, the chemical nature of the "active reacting matrix" gradually undergoes change due to the depletion of the "component" which dominated the reaction behavior in that stage. This causes the significant change of the reaction behavior. The critical temperatures at which the subsequent stage begins should be identifiable from the pyrolysis rate or weight loss ratio in the case of TGA experiments, such as are shown in Figures 5.6 and 5.7. With these assumptions in mind, the total volatile content can therefore be given by the following expression:

$$
\begin{align*}
& \frac{d V}{d t}=\sum_{i=1}^{n} \alpha_{i} k_{o i} e^{-E_{i} / R T}\left(V^{*}-V\right)  \tag{5.27}\\
& T=C t+323.16 \tag{5.28}
\end{align*}
$$

therefore:

$$
\begin{equation*}
\frac{d V}{d T}=\sum_{i=1}^{n} C \alpha_{i} k_{o i} e^{-E_{l} / R T}\left(V^{*}-V\right) \tag{5.29}
\end{equation*}
$$

where:
n no. of reaction stages which are first order reaction
$\alpha_{i}$ constant used to characterize the gradual change of the chemical structure of reacting residue.

$$
\alpha_{i}=1 \text { when } T_{i-1} \leq T<T_{i} \text {, otherwise } \alpha_{i}=0
$$ critical temperature, at which reaction behavior is undergoing visible change in terms of the ratio $\mathrm{dV} / \mathrm{dT}$ or rate $\mathrm{dV} / \mathrm{dt}$ due to the change of reacting residue, $K$

$\mathrm{E}_{\mathrm{i}} \quad$ activation energy of ith stage of reaction, $\mathrm{J} / \mathrm{mol}$
$\mathrm{k}_{\mathrm{oi}}$ pre-exponential factor of ith stage reaction, $\min ^{-1}$.

### 5.3.2 Application of the Multi-Stage Model

As observed in the TGA results of CANMET and Syncrude pitch pyrolysis, the pyrolysis behavior is shifted at about $450^{\circ} \mathrm{C}$ into a second stage as shown in the rate plots and weight loss plot. This two step feature is also observed by Rajeshwar [113] and Thakur and Nuttall [111] for oil shale pyrolysis, and Schuckler [116] for vacuum residuum pyrolysis. The need for a two stage reaction analysis was evident by their results, but two stage analysis was not implemented. The multi-stage expression can therefore be simplified to a 2-stage pyrolysis mechanism as follows:

$$
\begin{equation*}
\frac{d V}{d T}=\sum_{i=1}^{2} C \alpha_{i} k_{o i} e^{-E_{l} / R T}\left(V^{*}-V\right) \tag{5.30}
\end{equation*}
$$

where

$$
\begin{array}{ll}
\alpha_{1}=1, \alpha_{2}=0 & \text { when } \mathrm{T}<450^{\circ} \mathrm{C} \\
\alpha_{1}=0, \alpha_{2}=1 & \text { when } \mathrm{T} \geq 450^{\circ} \mathrm{C}
\end{array}
$$

This approach which was developed in this work, differs from the multi-parallel reactions discussed in Section 2.3: previous works have assumed that the reactions take place as mutually competing first order reactions. The 2 -stage first order reaction model was applied to the overall single first order reaction methods described earlier, and fitted to experimental data of both CANMET and Syncrude pitch. The TGA data have been divided into two stages: stage 1 corresponding to the first stage of the pyrolysis reaction in the temperature range $50^{\circ} \mathrm{C}$ (initial

TGA pyrolysis temperature) to $450^{\circ} \mathrm{C}$, and stage 2 corresponding to the second stage of pyrolysis reaction in the temperature range of $450{ }^{\circ} \mathrm{C}$ to the final pyrolysis temperature. Each stage was fitted to the model for the kinetic parameters for CANMET pitch and Syncrude pitch pyrolysis with the FORTRAN program, listed in Appendix E. One run was initially chosen for each pitch. The kinetic parameters are listed in Table 5.3 for runs at $25^{\circ} \mathrm{C} / \mathrm{min}$ and $800^{\circ} \mathrm{C}$ :

Table 5.3 Kinetic Parameters for the Nonisothermal Pyrolysis of CANMET Pitch and Syncrude Pitch at $25^{\circ} \mathrm{C} / \mathrm{min}$. and $800^{\circ} \mathrm{C}$

|  | First stage |  | Second stage |  | s.e.e. |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | $\mathrm{E}_{1} \mathrm{~kJ} / \mathrm{mol}$ | $\mathrm{k}_{01} \min ^{-1}$ | $\mathrm{E}_{2} \mathrm{~kJ} / \mathrm{mol}$ | $\mathrm{k}_{02} \min ^{-1}$ |  |
| Run\# Can48 | CANMET Pitch |  |  |  |  |
| 2-Integral | 21.89 | 5.534 | 71.34 | 4.448* $10^{4}$ | 1.42 |
| 2-Coats-Redfern | 18.31 | 1.207 | 69.93 | $2.865 * 10^{4}$ | 8.66 |
| 2-Chen-Nuttall | 19.81 | 2.663 | 70.91 | $4.004^{*} 10^{4}$ | 4.46 |
| 2-Friedman | 18.35 | 2.169 | 39.42 | 2.221* $10^{2}$ | 5.68 |
| Run\# Syn43 |  | Syn | de Pitch |  |  |
| 2-Integral | 30.82 | 51.80 | 67.66 | $2.555 * 10^{4}$ | 1.94 |
| 2-Coats-Redfern | 28.92 | 22.28 | 66.14 | $1.597 * 10^{4}$ | 5.23 |
| 2-Chen-Nuttall | 29.96 | 38.04 | 67.18 | $2.271 * 10^{4}$ | 2.06 |
| 2-Friedman | 22.43 | 7.251 | 101.5 | $3.875 * 10^{6}$ | 9.74 |

The values of the $E_{1}, k_{01}, E_{2}, k_{02}$, i.e. the kinetic parameters determined by each of the 2stage reaction methods, are in reasonable agreement except for Friedman method. The 2-stage integral method gives the best fit for both CANMET and Syncrude pitch. Table 5.3 also indicates the significant change of kinetic parameters between the first stage and second stage reactions, as expected. The activation energies of the second stage are about 2 to 4 times those of the first stage. Having obtained these parameters, the volatile contents were calculated according to each method and the predicted results (Appendices E and F) were plotted, along with experimental results in Figures 5.10 and 5.13. Only the 2-stage integral method gave good predictions of the volatile content over the complete range. The other three 2 -stage methods failed to predict the volatile content reasonably. The effect of the number of significant digits and a change of $k$ in the range of $\pm 2 \%$ was examined (Appendix l ). A change of the number of significant digits or k did not affect the fitting results and the superiority of integral method to other methods. The kinetic parameters were then reported in four significant digits and the s.e.e. in three significant digits.

Examination of the fitted $Y$ values for each 2-stage reaction method revealed that only the 2-stage integral method fitted the Y value calculated from experimental results as shown in Figures 5.8, 5.9, 5.11 and 5.12.

The 2-stage Coats-Redfern method analysis was performed by fitting Equation 5.14 to the 2 stages of TGA experimental data, with the term 2RT/E ignored. This term, ranging from 0.293 to 0.656 for the first stage reaction and from 0.172 to 0.255 for the second stage reaction of CANMET pitch pyrolysis and from 0.186 to 0.416 for the first stage reaction and from 0.182 to 0.270 for the second reaction of Syncrude pitch pyrolysis, is not small enough to be ignored in the linear regression fitting for the kinetic parameters. To do so, introduces a large error, and results in erroneous kinetic parameters and therefore wrong volatile yield predictions. Given a small value of $2 \mathrm{RT} / \mathrm{E}$ of 0.05 , the activation energy E is $423.4 \mathrm{~kJ} / \mathrm{mol}$ at $1000^{\circ} \mathrm{C}$, and $107.4 \mathrm{~kJ} / \mathrm{mol}$ at $50^{\circ} \mathrm{C}$. The error thus introduced to the predicted volatile content would be negligible for this case. However, the obtained activation energy of $423.4 \mathrm{~kJ} / \mathrm{mol}$ is unrealistically high. The simplification may be quite satisfactory when the thermal energy RT is significantly less than the activation energy. This case is often found for thermal decomposition of solids where either the temperature is low or the activation energy of the process is greater than RT. However, if RT tends to E, as is observed in this study, i.e., with low activation barriers and high temperatures, it is necessary to take a great number of terms in the integral analytical solution. It is clearly indicated that the assumption of $2 \mathrm{RT} / \mathrm{E} \ll 1$ is not valid for the case of pitch pyrolysis. The results obtained in the present study indicate that the thermal pyrolysis reactions of pitches are complex to the extent that they can not be described by the 2-stage Coats-Redfern method.

The 2-stage Friedman method analysis was performed by fitting Equation 5.15 to the 2 stages of the TGA experimental data, with the dV/dT calculated with experimental data in each stage by Equation 5.13. However the value of $d V / d T$ has been noted to be a sensitive index of the
reaction rate. The error introduced into the method is even significant at the second stage of reaction. Nonlinear behavior was observed for both CANMET pitch and Syncrude pitch as shown in Figures 5.9 and 5.12. In the second stage, the rate of the weight loss changes dramatically as the temperature is increased, and the ratio $\mathrm{dV} / \mathrm{dT}$ is less accurate. The standard error of deviation is observed as high as 5.68 for CANMET pitch and 9.74 for Syncrude pitch. The difference between the predicted and experimental volatile content is observed as high as $20 \%$ for CANMET pitch and $25 \%$ for Syncrude pitch as shown in Figures 5.10 and 5.13. The Friedman method is handicapped by the necessity of differentiating the raw TGA data, which is prone to error. Application of this method for the analysis of nonisothermal TGA data for pitch pyrolysis would lead, therefore, to incomplete, even wrong, information on the pyrolysis parameters.

The 2-stage Chen-Nuttall method analysis was performed by fitting Equation 5.17 to the 2 stages of the TGA experimental data, with the iterative linear regression technique. However, the Y values for this method are rather sensitive to the activation energy. The results indicate less satisfactory fitting than the integral method, even though the standard deviation s.e.e. of this method is rather close to that of 2-stage integral method for Syncrude pitch pyrolysis.

In the derivation of the Least Squares Fitting Equation 5.22, it is assumed that all measurements have the same standard deviation, s.e.e., and that the equation does fit well, then fitting for the parameters to minimize this deviation error and finally recomputing the standard deviation s.e.e.

$$
\begin{equation*}
\text { s.e.e. }=\sqrt{\sum_{i=1}^{n}\left(V_{i \exp }-V_{i f t}\right)^{2} /(n-2)} \tag{5.31}
\end{equation*}
$$

Where $\mathrm{V}_{\mathrm{i}}$ exp is the experimental volatile content, $\mathrm{V}_{\mathrm{i} \text { fit }}$ is the model predicted volatile content at data point i and n is the total number of data points.

Obviously, this approach prohibits assessment of goodness-of-fit, a fact frequently missed. When the standard deviation is too large, it indicates that the fitting is not successful, as can be seen in Table 5.2 and Figure 5.1. However, a small s.e.e. does not suggest any goodness-of-fit when the standard deviation is well within the experimental error. Further examination is always necessary to ensure the validity of the modeling results, as well as of the kinetic parameters.

The 2-stage integral method does not have the shortcomings mentioned above. The Y values calculated from the experimental data fitted linearly to $1 / T$ for both CANMET and Syncrude pitches, as shown in Figures 5.8 and 5.11. The predicted volatile contents compare closely to the experimental results for both CANMET pitch and Syncrude pitch at all the temperature investigated in this study as shown in Figures 5.10 and 5.13, with s.e.e. 1.4 and 1.9 respectively: The results obtained therefore suggest that the thermal pyrolysis reactions of these pitches can best be described by a 2-stage integral method. This analysis method is further tested for different pyrolysis conditions for its validity.


Figure 5.8 Comparison of model predicted $Y$ results and experimental $Y$ results for CANMET pitch at $25^{\circ} \mathrm{C} / \mathrm{min}$. and $800^{\circ} \mathrm{C}$ with 2-stage model


Figure 5.9 Comparison of model predicted $Y$ results and experimental $Y$ results for CANMET pitch at $25^{\circ} \mathrm{C} / \mathrm{min}$. and $800^{\circ} \mathrm{C}$ with 2-stage model


Figure 5.10 Comparison of model prediction and experimental volatile for CANMET pitch at $25^{\circ} \mathrm{C} / \mathrm{min}$. and $800^{\circ} \mathrm{C}$ with 2 -stage first order reaction model


Figure 5.11 Comparison of model predicted $Y$ results and experimental Y results for Syncrude pitch at $25^{\circ} \mathrm{C} / \mathrm{min}$. and $800^{\circ} \mathrm{C}$ with 2-stage model


Figure 5.12 Comparison of model predicted Y results and experimental Y results for Syncrude pitch at $25^{\circ} \mathrm{C} / \mathrm{min}$. and $800^{\circ} \mathrm{C}$ with 2-stage model


Figure 5.13 Comparison of model prediction and experimental volatile for Syncrude pitch at $25^{\circ} \mathrm{C} / \mathrm{min}$. and $800^{\circ} \mathrm{C}$ with 2-stage first order reaction model

### 5.4 2-Stage First Order Reaction Model for Pitch Pyrolysis

Using the least squares curve fitting of experimental data to the 2 -stage integral method, the kinetic parameters $E_{1}, k_{01}, E_{2}, k_{02}$, and s.e.e. were computed using iterative techniques for a number of different experiments. The values of these parameters are listed in Table 5.4.

Table 5.4 Kinetic Parameters for the Nonisothermal Pyrolysis of CANMET Pitch and Syncrude Pitch at $800^{\circ} \mathrm{C}$ and Different Heating Rates with 2-Integral Method

| Run\# | ${ }^{\circ} \mathrm{C} / \mathrm{min}$. | First stage |  | Second stage |  | s.e.e. |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | $\mathrm{E}_{1} \mathrm{~kJ} / \mathrm{mol}$ | $\mathrm{k}_{01}$ min. ${ }^{-1}$ | $\mathrm{E}_{2} \mathrm{~kJ} / \mathrm{mol}$ | $\mathrm{k}_{02} \min ^{-1}$ |  |
|  |  | CANMET Pitch |  |  |  |  |
| Can48 | 25 | 21.89 | 5.534 | 71.34 | $4.448 * 10^{4}$ | 1.42 |
| Can33 | 50 | 20.90 | 7.649 | 64.47 | $2.444 * 10^{4}$ | 2.12 |
| Can41 | 100 | 26.91 | 39.63 | 72.11 | $1.111 * 10^{5}$ | 1.44 |
| Can58 | 150 | 46.64 | 552.3 | 96.65 | $3.511 * 10^{6}$ | 0.97 |
|  |  |  | Sync | de Pitch |  |  |
| Syn43 | 25 | 30.82 | 51.80 | 67.66 | $2.554 * 10^{4}$ | 1.94 |
| Syn 29 | 50 | 37.57 | 298.2 | 76.57 | $1.964 * 10^{5}$ | 2.07 |
| Syn18 | 100 | 44.16 | $1.326 * 10^{3}$ | 65.51 | $3.523 * 10^{4}$ | 2.97 |
| Syn8 | 150 | 46.14 | $2.549 * 10^{3}$ | 69.80 | $1.031 * 10^{5}$ | 2.85 |

Having obtained the values of $E_{1}, k_{01}, E_{2}, k_{02}$, from this table, the volatile content and the Y values for both CANMET pitch and Syncrude pitch predicted by the 2-stage integral method were computed using Equation 5.22 and Equation 5.23 respectively.

The Y values obtained experimentally and predicted by the 2 -stage integral method for runs at different heating rates are plotted in Figures 5.14 and 5.15, as a function of $1 / \mathrm{T}$. The 2stage integral method fits adequately and linearly the Y versus $1 / \mathrm{T}$ data for both CANMET pitch and Syncrude pitch. It is also noted that it is not safe to fit all the data from different runs to find a set of unique activation energy $E$ and pre-exponential factor $k_{0}$, regardless of the heating rates. The scatter of the data points prohibits this. It is more evidently noted in Figure 5.14 for CANMET pitch pyrolysis at low temperatures. However the heating rate did not show a systematic influence.

Similarly, the prediction of the 2 -stage integral method for the volatile content is shown in Figures 5.16 and 5.17 as a function of pyrolysis time and in Figures 5.18 and 5.19 as a function of pyrolysis temperature for runs at different heating rates, for CANMET pitch and Syncrude pitch respectively. The experimental data fitted the 2 -stage integral method well at different heating rates over the entire temperature range. The close agreement between the experimental volatile contents and the predicted volatile contents suggests that the 2 -stage integral method describes the pitch pyrolysis adequately. The magnitude of the standard deviation also supports this observation.

The volatile yield rate $\mathrm{dV} / \mathrm{dt}$ is also computed with the kinetic parameters obtained as shown in Table 5.4 and compared with the yield rate $\mathrm{dV} / \mathrm{dt}$ calculated from the experimental data. The results are plotted in Figures 5.20 and 5.21. The close agreement between the predicted volatile yield rate and the rate calculated from experimental data is in accordance with that of the volatile content versus $t$ curve, but is a more rigorous test.


Figure 5.14 Comparison of model predicted $Y$ results and experimental $Y$ results for CANMET pitch at different heating rates and $800^{\circ} \mathrm{C}$ with 2 -stage integral method


Figure 5.15 Comparison of model predicted Y results and experimental Y results for Syncrude pitch at different heating rates and $800^{\circ} \mathrm{C}$ with 2-stage integral method


Figure 5.16 Comparison of model prediction and experimental volatile for CANMET pitch at different heating rates and $800^{\circ} \mathrm{C}$ with 2-stage integral method


Figure 5.17 Comparison of model prediction and experimental volatile for Syncrude pitch at different heating rates and $800^{\circ} \mathrm{C}$ with 2-stage integral method


Figure 5.18 Comparison of model prediction and experimental volatile for CANMET pitch at different heating rates and $800^{\circ} \mathrm{C}$ with 2-stage integral method


Figure 5.19 Comparison of model prediction and experimental volatile for Syncrude pitch at different heating rates and $800^{\circ} \mathrm{C}$ with 2-stage integral method


Figure 5.20 Comparison of model prediction $\mathrm{dV} / \mathrm{dt}$ and experimental $\mathrm{dV} / \mathrm{dt}$ for CANMET pitch at different heating rates and $800^{\circ} \mathrm{C}$ with 2-stage integral method


Figure 5.21 Comparison of model prediction $\mathrm{dV} / \mathrm{dt}$ and experimental $\mathrm{dV} / \mathrm{dt}$ for Syncrude pitch at different heating rates and $800^{\circ} \mathrm{C}$ with 2-stage integral method

### 5.5 Testing of the 2-Stage Integral Method

In order to further examine the validity of the 2 -stage integral method, it was used to predict the volatile yield at different pyrolysis conditions, other than the runs used to fit for the kinetic parameters. The kinetic parameters obtained at conditions of $25,50,100$ and $150{ }^{\circ} \mathrm{C} / \mathrm{min}$ and final temperature $800^{\circ} \mathrm{C}$ were used to fit runs at the same heating rates but different final temperature ranging from $750^{\circ} \mathrm{C}$ to $950^{\circ} \mathrm{C}$ for CANMET pitch and Syncrude pitch respectively. The kinetic parameters at heating rate $100^{\circ} \mathrm{C} / \mathrm{min}$ and final temperature $800^{\circ} \mathrm{C}$ were used to predict the volatile yield for CANMET pitch runs at the same heating rate but different final temperature 750,850 and $950^{\circ} \mathrm{C}$, while the kinetic parameters at heating rate $50^{\circ} \mathrm{C} / \mathrm{min}$ and 800 ${ }^{\circ} \mathrm{C}$ were used to predict the volatile yield for Syncrude pitch runs at the same heating rate but final temperature of 750,850 and $950^{\circ} \mathrm{C}$. The experimental conditions are listed in Table 5.5. The s.e.e. values are also listed in the table as the indication of the goodness of the model prediction. The s.e.e values calculated with other methods are also listed in the table for comparison.

Table 5.5 Experimental Conditions and Model Predicted Results of CANMET Pitch and Syncrude Pitch Pyrolysis

| Run\# | T ${ }^{\circ} \mathrm{C}$ | V* | s.e.e. |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | Integral | C-R | C-N | FM |
| CANMET Pitch at $100{ }^{\circ} \mathrm{C} / \mathrm{min}$ |  |  |  |  |  |  |
| Can42 | 750 | 79.54 | 2.05 | 4.24 | 1.95 | 4.60 |
| Can40 | 850 | 79.01 | 4.87 | 2.70 | 4.09 | 3.12 |
| Can52 | 950 | 81.23 | 1.39 | 5.25 | 2.28 | 4.94 |
| Syncrude Pitch at $50{ }^{\circ} \mathrm{C} / \mathrm{min}$ |  |  |  |  |  |  |
| Syn 27 | 750 | 90.96 | 2.57 | 5.38 | 2.63 | 9.07 |
| Syn 32 | 850 | 90.61 | 2.04 | 5.15 | 2.01 | 8.64 |
| Syn33 | 950 | 91.01 | 4.49 | 8.75 | 5.26 | 12.19 |

Integral $=2$-stage integral method, $\mathrm{C}-\mathrm{R}=2$-stage Coats-Redfern method, $\mathrm{C}-\mathrm{N}=2$-stage Chen-Nuttall method, FM $=2$-stage Friedman method

It is observed from the s.e.e. values that the prediction is in good agreement with the experimental volatile content. Further examination of Figure 5.22 to Figure 5.27 proved that the
model indeed predicted the volatile content well. The prediction of CANMET pitch pyrolysis volatile content was calculated with the kinetic parameters obtained at $100^{\circ} \mathrm{C} / \mathrm{min}$ and $800^{\circ} \mathrm{C}$ and plotted in Figures 5.22, 5.23, and 5.24, along with the experimental volatile content for comparison. The prediction of Syncrude pitch pyrolysis volatile content was calculated with the kinetic parameters obtained at $50^{\circ} \mathrm{C} / \mathrm{min}$ and $800^{\circ} \mathrm{C}$ and plotted in Figures $5.25,5.26,5.27$, along with the experimental volatile contents for comparison. It is shown that the agreement between the prediction and experimental data is very good, which is supported by the s.e.e. values. This indicates that the 2 -stage integral method can describe the pitch pyrolysis, and the kinetic parameters derived from this model are independent of pyrolysis conditions such as final temperature. The results thus support the assumption that pyrolysis is a chemical reaction controlled process.


Figure 5.22 Comparison of model prediction and experimental volatile for CANMET pitch at $100^{\circ} \mathrm{C} / \mathrm{min}$ and $750^{\circ} \mathrm{C}$ with 2-stage integral method


Figure 5.23 Comparison of model prediction and experimental volatile for CANMET pitch at $100^{\circ} \mathrm{C} / \mathrm{min}$ and $850^{\circ} \mathrm{C}$ with 2-stage integral method


Figure 5.24 Comparison of model prediction and experimental volatile for CANMET pitch at $100^{\circ} \mathrm{C} / \mathrm{min}$ and $950^{\circ} \mathrm{C}$ with 2-stage integral method


Figure 5.25 Comparison of model prediction and experimental volatile for Syncrude pitch at $50^{\circ} \mathrm{C} / \mathrm{min}$ and $750^{\circ} \mathrm{C}$ with 2 -stage integral method


Figure 5.26 Comparison of model prediction and experimental volatile for Syncrude pitch at $50^{\circ} \mathrm{C} / \mathrm{min}$ and $850^{\circ} \mathrm{C}$ with 2-stage integral method


Figure 5.27 Comparison of model prediction and experimental volatile for Syncrude pitch at $50^{\circ} \mathrm{C} / \mathrm{min}$ and $950^{\circ} \mathrm{C}$ with 2-stage integral method

### 5.6 Discussion and Conclusions

Although the pattern of volatile release for pitches pyrolyzed under TGA conditions is complex, an adequate description of the kinetics is possible. The pyrolysis takes place in 2 stages, with a first stage of low activation energy barrier and lower pre-exponential factor, and the second stage of higher activation energy and pre-exponential factor. It is recommended that the process be modeled with a 2 -stage reaction model with the integral method analysis. It is demonstrated that the overall single stage reaction model with analysis by the integral, CoatsRedfern, Chen-Nuttall and Friedman methods, as well as Anthony and Howard's distributed activation energy model, are not sufficient to fit the TGA pyrolysis data and predict the course of the pitch pyrolysis process. It is also found that the single stage reaction model analyzed with these methods does not reproduce the values of activation energy and pre-exponential factors when the pitch TGA pyrolysis data at different conditions are taken for computation. This
phenomena has also been observed by Dahr [122], Natu [123] and Carrasco [62]. By contrast, the two-stage first order model with constants fitted by the integral analysis method provides a good description of the volatilization behavior.

The kinetic expressions obtained represent the global process, and are intended for numerical modeling or engineering calculations. These parameters have only limited validity and can not be used to pin-point the rate controlling mechanism. The true reaction chemistry undoubtedly is much more complex than the multiple stage first order reactions assumed above. It is known that the pyrolysis of any hydrocarbon residual is a very intricate and complex phenomena composed of various elementary reactions that are different to analyze separately and whose quantitative contributions to the global pyrolysis process are virtually impossible to evaluate. For these reasons, even if the overall process has no ideal significance with regard to the reaction mechanism, it is useful as a means of quantifying the rate of reaction and for design purposes. Caution must therefore be used to avoid over-interpreting these rate equations in terms of the fundamental microscopic chemistry of the system.

The apparent activation energies calculated from this study for pitch fall approximately midway between values reported by others for oil shale decomposition. Values of $31.6 \mathrm{~kJ} / \mathrm{mol}$, $38.4 \mathrm{~kJ} / \mathrm{mol}$ and $62.3 \mathrm{~kJ} / \mathrm{mol}$ [111], $108.10 \mathrm{~kJ} / \mathrm{mol}$ and $209.50 \mathrm{~kJ} / \mathrm{mol}$ [113] are reported. Since $\leftarrow$ the strength of typical single bonds to carbon are about $335-420 \mathrm{~kJ} / \mathrm{mol}$, the question often arises as to why the activation energies for thermal decomposition of such residues are so much lower. The answer is that the activation energies for decomposition of heterogeneous organic material can not generally be interpreted in terms of a specific bond-breaking process (e.g. C-C vs. C-H vs. C-O etc.). Often, activation energies in the ranges of $42-84 \mathrm{~kJ} / \mathrm{mol}$ are reported with an indication that these are essentially effective activation energies for a sum of different reactions that occur simultaneously. When there are radicals involved in the pyrolysis, the activation energy
can be reduced to as low as $21-42 \mathrm{~kJ} / \mathrm{mol}$ [22]. The activation energies for each reaction may be much higher. As a result, the development of a detailed mechanistic picture on the basis of a few effective activation energies is usually fruitless.

Heck and DiGuiseppi [124] observed that the critical element of 2-stage hydrocracking of residuum is believed to be the balancing of the cracking and hydrogenation activities during the initial $50 \%$ conversion. It is during this initial conversion that the residuum is most active, free radicals are formed at the highest rate and hydrogen demand is highest. Gray et al. [125] found that the initial conversion of asphaltenes occurs largely as a result of cracking relatively long aliphatic fragments away from a largely aromatic core. The aliphatic/aromatic bonds broken during this initial conversion process are relatively facile, especially when the aliphatic chains are longer than one or two carbon atoms. The conversion path is best illustrated [124] by the relatively rapid decrease in average molecular size and increase in aromaticity that occurs during the initial cracking of the large aliphatic moieties away from the largely aromatic core. The remaining conversion, which proceeds more slowly, involves the cracking away of smaller aliphatic moieties. Stubington's results [126] using bagasse suggested that the pyrolysis mechanism changed at certain pyrolysis level, which can be expressed as time, conversion of carbon or the remaining volatile content. At a certain devolatilization level, a set of different kinetic parameters is required to describe the change of the pyrolysis mechanism. These findings support the 2 -stage pitch pyrolysis mechanism with the low activation energy barrier for the first stage and high activation energy barrier for the second stage.

The 2 -stage model reflects changes in the chemical constitution or structures as conversion proceeds by using two values of activation energy and pre-exponential factor. This feature is essential to describe pitch dependence of devolatilization rates on the remaining volatile content. The abundance of radicals in the bridges of non-aromatics accelerates their conversion
rates, which has two ramifications: First, gases are expelled rapidly at low temperature and, second, extensive cross-linking inhibits the production of tar precursors [127]. In contrast, bridges in aromatics have very little radical content, so they decompose at relatively high temperatures at significantly slower rates. The transition between these two limiting cases is a sharp one, occurring at a certain temperature (remaining volatile content) level. Consequently for nonaromatic and aromatic components, small differences in the radical content causes appreciable difference in rates and yields, compounding the acute sensitivity of the labile bridge fraction to carbon content. These findings in the present study clearly demonstrated that the chemical constitution of pitch affects product evolution rates and yield at any stage of devolatilization.

The magnitude of the activation energies in both stages suggests that the pyrolysis of pitch was kinetically controlled under the reaction conditions studied. The dependence of $\mathrm{dV} / \mathrm{dT}$ on $\mathrm{V}^{*}-\mathrm{V}$ is also in accordance with that.

In summary, the overall single first order model and the Gaussian distributed activation energy model are not adequate to describe pyrolysis of CANMET and Syncrude pitches due to the mechanism change of the pitch pyrolysis at an intermediate temperature, and high volatile yield. These models have been developed for relatively low volatile content material and processes such as coal pyrolysis. The 2-stage first order reaction model with the integral analysis method is proven adequate to describe the pitch pyrolysis process and gives lower activation energy and preexponential factor for the first stage, and higher activation energy and preexponential factor for the second stage of pyrolysis. These kinetic parameters can be extrapolated to different temperature range. However, the compensation effect of the kinetic parameters is observed and is discussed in the next chapter.

It was noted in applying the different analysis methods to the pyrolysis kinetics, that when the activation energy was low, the pre-exponential factor was also low. The mutual dependence of the activation energy $E$ and the pre-exponential factor $k_{0}$, termed the compensation effect, has been reported for catalytic kinetics [68,69], thermal aging process of polymers [128-130], and some $\mathrm{CaC}_{2} \mathrm{O}_{4} \bullet \mathrm{H}_{2} \mathrm{O}$ pyrolysis processes [86] as described in Chapter 2. The mutual dependence of the kinetic parameters does not occur in simple reactions. The compensation effect is associated with the following two criteria:
A) The logarithm of the pre-exponential factor, $\operatorname{lnk}_{0}$, is linearly proportional to the activation energy $E$, given by the following equation, where the $\alpha$ and $\beta$ are the compensation constants:

$$
\begin{equation*}
\ln k_{o}=\alpha+\beta E \tag{6.1}
\end{equation*}
$$

B) The logarithm of the reaction rate constant, lnk, is linearly proportional to the reciprocal of the reaction temperature $1 / \mathrm{T}$, and all the $\operatorname{lnk} \mathrm{vs}$. $1 / \mathrm{T}$ lines generated in different temperature programmed experiments intersect at one point $\mathrm{T}_{\mathrm{i}}$, the isokinetic temperature. This results in the following equation, where the a and b are isokinetic constants:

$$
\begin{equation*}
\ln k=a+b \frac{1}{T} \tag{6.2}
\end{equation*}
$$

As pointed out in the literature review, the second criterion is a special case of the first one. The existence of the second criterion guarantees the existence of the first criterion and the compensation effect. However, the existence of the first criterion guarantees the existence of the compensation effect, but not the second criterion.

The most common identification of a compensation effect comes from the observation of a linear correlation between the activation energy and the logarithm of the pre-exponential factor
[66]. The application of statistical methods to the recognition of a linear relationship between values of the activation energy and the logarithm of the pre-exponential factor has been described by Exner [100], who suggests that a single point of intersection in the lnk vs. 1/T plots could be used for a sound statistical test, since lnk and T are statistically independent. This is the basis of the isokinetic relationship. It is evident that for a set of experimental data one may infer from such a point of intersection the linearity between the activation energy and the logarithm of the preexponential factor, but the reverse may not be true.

In this work, the compensation effect was investigated at different pyrolysis conditions and for different single overall first reaction models and 2-stage first order model for both CANMET and Syncrude pitch. The accuracy of the kinetic parameters was also examined, comparing the resulting standard deviation error (s.e.e.). The possibility of the existence of one unique set of these kinetic parameters was therefore investigated.

### 6.1 Compensation Effect of Kinetic Parameters Derived from Overall First Order Model

Since the single overall first order reaction model (analyzed with integral, Coats-Redfern, Chen-Nuttall and Friedman methods) was inadequate to describe the pyrolysis kinetics, results on compensation effect are not discussed in detail. Figures 6.1 and 6.2 show that a good linear correlation of Equation 6.1 was obtained, however, Equation 6.2 was not met since an isokinetic temperature within the operating temperature range was not found. This compensation effect has also been observed in the studies of thermal degradation of polymers with different mathematical methods [130]. The single overall first order model analyzed by the different mathematical methods in this work did not reproduce the kinetic parameters, and these parameters derived from each of these methods follow the compensation effect.


Figure 6.1 CANMET pitch TGA pyrolysis kinetic parameters at $50^{\circ} \mathrm{C} / \mathrm{min}$ and $700^{\circ} \mathrm{C}$ with different methods (overall first order)


Figure 6.2 CANMET pitch pyrolysis reaction rate constant as a function of temperature at heating rate $50^{\circ} \mathrm{C} / \mathrm{min}$ and final temperature $700^{\circ} \mathrm{C}$

### 6.2 Compensation Effect of Kinetic Parameters Derived from 2-Stage Reaction Model

The compensation effect for the 2 -stage reaction model analyzed with different methods, at different heating rates such as $25,50,100$ and $150^{\circ} \mathrm{C} / \mathrm{min}$ and final temperature of $800{ }^{\circ} \mathrm{C}$, was investigated and the values of the compensation effect parameters (Equation 6.1) for the first stage: $\alpha_{1}$ and $\beta_{1}$, for the second stage $\alpha_{2}$ and $\beta_{2}$, are listed in Tables 6.1 and 6.2 for CANMET and Syncrude pitch. The square of the regression coefficients, $\mathbf{R}^{2}$, are also listed in these tables indicting the linearity of the fitting. Four data points derived via each 2-stage method were used in the fitting of each run.

Table 6.1 Compensation Parameters for CANMET Pitch Pyrolysis at Different Heating Rates and $800^{\circ} \mathrm{C}$

|  | First stage |  |  | Second stage |  |  |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: |
| Method | $\alpha_{1}$ | $\beta_{1}{ }^{*} 10^{-4}$ | $\mathrm{R}^{2}$ | $\alpha_{2}$ | $\beta_{2}{ }^{*} 10^{-4}$ | $\mathrm{R}^{2}$ |
| Integral | -1.544 | 1.711 | 0.952 | 0.008 | 1.558 | 0.978 |
| Coats-Redfern | -2.824 | 1.930 | 0.959 | -0.406 | 1.586 | 0.980 |
| Chen-Nuttall | -2.128 | 1.826 | 0.956 | -0.087 | 1.566 | 0.979 |
| Friedman | -2.444 | 1.980 | 0.984 | -1.169 | 1.693 | 0.999 |
| All methods | -2.360 | 1.903 | 0.957 | -0.789 | 1.650 | 0.991 |

Table 6.2 Compensation Parameters for Syncrude Pitch Pyrolysis at Different Heating Rates and $800^{\circ} \mathrm{C}$

|  | First stage |  |  | Second stage |  |  |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: |
| Method | $\alpha_{1}$ | $\beta_{1}{ }^{*} 10^{-4}$ | $\mathrm{R}^{2}$ | $\alpha_{2}$ | $\beta_{2}{ }^{*} 10^{-4}$ | $\mathrm{R}^{2}$ |
| Integral | -3.718 | 2.492 | 0.998 | -1.220 | 1.761 | 0.795 |
| Coats-Redfern | -4.367 | 2.591 | 0.998 | -1.554 | 1.781 | 0.809 |
| Chen-Nuttall | -3.923 | 2.530 | 0.998 | -1.278 | 1.765 | 0.801 |
| Friedman | -4.361 | 2.852 | 0.998 | 0.705 | 1.471 | 0.977 |
| All methods | -3.308 | 2.384 | 0.979 | 0.516 | 1.497 | 0.961 |

Figures 6.3 and 6.4 show that the compensation effect Equation 6.1 fits data for each method at each stage of pyrolysis adequately. For all the cases investigated for CANMET pitch the $\mathrm{R}^{2}$ coefficient is greater than 0.95 , whereas for Syncrude pitch the $\mathrm{R}^{2}$ is greater than 0.998 for the first stage and greater than 0.795 for the second stage.


Figure 6.3 CANMET pitch TGA pyrolysis kinetic parameters at different heating rates and $800^{\circ} \mathrm{C}$ with 2-stage first order model analyzed with different methods


Figure 6.4 Syncrude pitch TGA pyrolysis kinetic parameters at different heating rates and $800^{\circ} \mathrm{C}$ with 2 -stage first order model analyzed with different methods


Figure 6.5 CANMET pitch TGA pyrolysis kinetic parameters at different heating rates and $800^{\circ} \mathrm{C}$ with 2 -stage first order model


Figure 6.6 Syncrude pitch TGA pyrolysis kinetic parameters at different heating rates and at $800^{\circ} \mathrm{C}$ with 2-stage first order model

The regression results for parameters via 2-stage model are plotted in Figure 6.5 and Figure 6.6 for CANMET and Syncrude pitch respectively. Clearly one set of constants fits data from all methods in each stage, with $\mathrm{R}^{2}$ coefficient is greater than 0.957 for all the cases investigated.

The physical meaning of the compensation effect parameters $\alpha$ and $\beta$ has been a topic of research and it is beyond the scope of this research to explore it in detail. However, it is noticed that the parameter $\beta$ is rather constant for each stage of the pyrolysis and the parameter $\beta$ of the first stage is larger than the $\beta$ parameter of the second stage, for both of the two pitches studied. By contrast the parameter $\alpha$ changes with the model over a large range. For decarboxylation of solids, Muraishi [65] has stated that whereas the parameter $\beta$ is related to the bond strength of the metal leaving group in the three dicarboxylates investigated in his work, the parameter $\alpha$ is related to the structure of and defects in the starting material or to the mobility of the crystal lattice in the dicarboxylate thermal decomposition. The parameter $\alpha$ obtained in the present work showed complex tendencies among the 2 -stage of pyrolysis process and different mathematical methods used with the 2 stage model to derive the kinetic parameters. However, the parameter $\alpha$ of the first stage of the pyrolysis process of both pitches studied is smaller than that of the second stage of the pyrolysis process, which may suggest the chemical structure difference between these two stages. This difference of the chemical structure at different level of pyrolysis has been also observed experimentally [125]. Although there are slight differences in the parameter $\beta$ obtained from the different methods, the parameter $\beta$ obtained may indicate similar " bond strength" and therefore suggest there is one type of reaction dominant in each stage. The bond strength is therefore different according to the parameter $\beta$ between the first stage and second stage.

The $R^{2}$ coefficient of the first stage of Syncrude pitch is the highest in all the cases studied. This is in good agreement with the results as shown in Figure 5.7 in Chapter 5. The experimental results clearly show a consecutive pyrolysis process: at the beginning of the pyrolysis, the ratio $\mathrm{dV} / \mathrm{dT}$ increases with the decrease of the $\mathrm{V}^{*}-\mathrm{V}$ and up to a point where $\mathrm{dV} / \mathrm{dT}$ kept roughly the same before going into the next stage of pyrolysis. In the modeling process, this consecutive process was not divided into more detailed stages for the simplicity of modeling and limiting the parameters introduced into the kinetic model due to the fact that the yield of volatile at this stage is much less than that at lower $\mathrm{V}^{*}-\mathrm{V}$. This experimental evidence supports a common belief that a consecutive process is one of causes of the compensation effect. This is further supported by the results with CANMET pitch. When the pyrolysis process was fitted with the overall single first order model, the linear regression of the kinetic parameter $k_{o}$ and $E$ to the compensation equation resulted in a $\mathrm{R}^{2}$ of 0.986 . When the pyrolysis experimental results were fitted with 2 -stage model, the linear regression of the kinetic parameter $\mathrm{k}_{0}$ and E for each stage and each method resulted in $R^{2}$ coefficient from 0.952 to 0.999 . The $R^{2}$ coefficient is smaller than that obtained from the overall single first order model, except for the second stage of the Friedman method for CANMET pitch pyrolysis. It is, therefore, evident that the 2 stage behavior in the overall single first order model resulted in the higher $R^{2}$ coefficient. Similarly the lower $R^{2}$ coefficient derived for the second stage kinetic parameters $\mathrm{k}_{0}$ and E of Syncrude pitch pyrolysis suggests a lesser degree of multi-stage behavior, i.e., lesser heterogeneity of reactions. The activation energy of the second stage of Syncrude pitch pyrolysis changes over a very small range with changes of pyrolysis conditions and methods used to derive this parameter.

The compensation effect was also assessed via Equation 6.2, and the results calculated with the kinetic parameters derived from 2-stage reaction model analyzed with integral method are shown in Figure 6.7 and Figure 6.8 for CANMET and Syncrude pitch respectively, as a
function of the reciprocal of the pyrolysis temperature $1 / T$. The logarithm of reaction rate constants at different heating rates shows linear relationship with the reciprocal temperature $1 / \mathrm{T}$ in each temperature range, however the lines of $\ln k \sim 1 / T$ do not intersect at one single point for either pitch in both stages. The isokinetic temperature was therefore not observed. This suggests that the second criterion for the compensation effect does not hold for CANMET and Syncrude pitch pyrolysis. As described by Kral [68, 69], the second criterion is a special case, and the existence of the compensation effect does not guarantee it to be true.

Similarly, calculations were done with the kinetic parameters derived from Coats-Redfern, Chen-Nuttall and Friedman methods as found in Appendix G. As before, the isokinetic temperature was not clearly observed. The inaccuracy of these methods used to derive the kinetic parameters has been cited as a cause of the compensation effect, however the less accurate models did not result in an isokinetic temperature. It is worth noting that although the $\operatorname{lnk} \sim 1 / \mathrm{T}$ lines appeared to intersect at a single point during the first stage pyrolysis of Syncrude pitch, with all 2-stage methods, a narrow temperature range was observed rather than a single point.


Figure 6.7 CANMET pitch pyrolysis reaction rate constant as a function of temperature at different heating rates and final temperature $800^{\circ} \mathrm{C}$ with 2-stage reaction model analyzed with integral method


Figure 6.8 Syncrude pitch pyrolysis reaction rate constant as a function of temperature at different heating rates and final temperature $800^{\circ} \mathrm{C}$ with 2-stage reaction model analyzed with integral method

### 6.3 The Relationship of Standard Errors and Kinetic Parameters

Even though the kinetic parameters $\ln _{0}$ and E follow a linear relationship, the standard deviation errors (s.e.e.) of the experimental volatile content and the model predicted volatile yield via different pairs of the kinetic parameters were not identical. The standard deviation error was calculated with the related model and the kinetic parameters (Appendix E), and is plotted against the activation energy obtained with the different analysis methods and at different pyrolysis conditions in Figures 6.9 and 6.10. For CANMET pitch (Figure 6.9), the s.e.e. for the first stage decreases with the increase of the activation energy and passes through a weak minimum. This minimum is not really well defined. For the second stage pyrolysis of CANMET pitch pyrolysis, a minimum was also observed. It is evident that there is an optimal value of activation energy for each stage of pyrolysis reaction at which minimum s.e.e. can be achieved. Because $\operatorname{lnk}_{0}$ is linearly
proportional to E , there exists a relative value of $\mathrm{k}_{0}$. Therefore there is an unique set of optimal values of $E$ and $k_{0}$ for the first and second stage reaction which minimize the s.e.e. For CANMET pitch these minimal values for the pyrolysis kinetic parameters obtained from Figure 6.9 are:

$$
\mathrm{E}_{1}=40.2 \mathrm{~kJ} / \mathrm{mol}, \mathrm{k}_{01}=197.4 \mathrm{~min}^{-1}, \mathrm{E}_{2}=86.6 \mathrm{~kJ} / \mathrm{mol}, \mathrm{k}_{02}=7.31 * 10^{5} \mathrm{~min}^{-1}
$$

It should be noted that for the first stage, a range of values of E could be applied. For Syncrude pitch pyrolysis process as shown in Figure 6.10, the trends are different in that no minima are evident, but optimal values for the pyrolysis kinetic parameters can be obtained as:

$$
\mathrm{E}_{1}=45.7 \mathrm{~kJ} / \mathrm{mol}, \mathrm{k}_{01}=1.96^{*} 10^{3} \mathrm{~min}^{-1}, \mathrm{E}_{2}=67.6 \mathrm{~kJ} / \mathrm{mol}, \mathrm{k}_{02}=4.19^{*} 10^{4} \mathrm{~min}^{-1}
$$

The activation energy values are within the wide range of published kinetic parameters [127] in which an activation energy range of $42-84 \mathrm{~kJ} / \mathrm{mol}$ was reported for kerogen-to-bitumen pyrolysis. It is also worth noting that these values are very close to the kinetic parameter values derived with integral method for each stage.

With the above kinetic parameters, the volatile yields were calculated for different pyrolysis conditions with Equation 5.10 in Chapter 5 and the results are listed in Appendix H and plotted in Figures 6.11 to 6.14 . With the predicted volatile yield, the s.e.e. can therefore be calculated. The results are listed in Tables 6.3 and 6.4 are plotted in Figures 6.11 to 6.14 .


Figure 6.9 CANMET pitch TGA pyrolysis s.e.e. as a function of $E$ at different conditions and with different methods


Figure 6.10 Syncrude pitch TGA pyrolysis s.e.e. as a function of $E$ at different conditions and different methods

Figures 6.11 and 6.12 show that the predicted volatile contents at high heating rates and high temperature are in very good agreement with the experimental values, while the prediction at low temperatures for both samples and at the low heating rate of $25^{\circ} \mathrm{C} / \mathrm{min}$ for CANMET pitch is acceptable. Figures 6.13 and 6.14 also show that at different final temperatures the predicted volatile contents at high temperature for each run are in very good agreement with the experimental volatile yields. The prediction is generally better than that shown in Figures 6.11 and 6.12, and much better than the prediction of the overall single reaction model as shown in Figure 5.1. It is therefore possible to predict the volatile content with one set of unique kinetic parameters for the 2 stage reaction model regardless of the pyrolysis conditions and the methods used to fit the experimental results.

Table 6.3 Experimental Conditions and Model Predicted Results of CANMET Pitch and Syncrude Pitch Pyrolysis

| Heating |  |  | One set $\mathrm{k}_{\mathbf{o}}$ |  |  | Integral |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Run\# | Rate ${ }^{\circ} \mathrm{C} / \mathrm{min}$ | $\mathrm{V}^{*} \%$ | E, | s.e.e |  |  |
| S.e.e |  |  |  |  |  |  |
| Can48 | 25 | 80.84 | 4.10 | 1.42 |  |  |
| Can33 | 50 | 80.79 | 5.62 | 2.12 |  |  |
| Can41 | 100 | 79.30 | 8.69 | 1.44 |  |  |
| Can58 | 150 | 77.59 | 1.57 | 0.97 |  |  |
| Syncrude Pitch at $800^{\circ} \mathrm{C}$ |  |  |  |  |  |  |
| Syn43 | 25 | 91.03 | 10.45 | 1.94 |  |  |
| Syn29 | 50 | 90.07 | 5.62 | 2.07 |  |  |
| Syn18 | 100 | 90.58 | 4.01 | 2.97 |  |  |
| Syn8 | 150 | 90.62 | 9.98 | 2.85 |  |  |



Figure 6.11 Comparison of experimental data and model prediction for CANMET pitch at different heating rates and $800^{\circ} \mathrm{C}$ with a single set of kinetic parameters


Figure 6.12 Comparison of experimental data and model prediction for Syncrude pitch at different heating rates and $800^{\circ} \mathrm{C}$ with a single set of kinetic parameters

Table 6.4 Experimental Conditions and Model Predicted Results of CANMET Pitch and Syncrude Pitch Pyrolysis

|  |  |  | One set $\mathrm{k}_{0}$ | Integral |
| :---: | :---: | :---: | :---: | :---: |
| Run\# | T ${ }^{\circ} \mathrm{C}$ | V* \% | E, s.e.e. | s.e.e |
| CANMET Pitch at $100{ }^{\circ} \mathrm{C} / \mathrm{min}$ |  |  |  |  |
| Can42 | 750 | 79.54 | 8.13 | 2.05 |
| Can40 | 850 | 79.01 | 4.34 | 4.87 |
| Can52 | 950 | 81.23 | 8.93 | 1.39 |
| Syncrude Pitch at $50{ }^{\circ} \mathrm{C} / \mathrm{min}$ |  |  |  |  |
| Syn27 | 750 | 90.96 | 5.40 | 2.57 |
| Syn32 | 850 | 90.61 | 5.94 | 2.04 |
| Syn33 | 950 | 91.01 | 5.97 | 4.49 |



Figure 6.13 Comparison of model prediction and experimental volatile content for CANMET pitch at $100^{\circ} \mathrm{C} / \mathrm{min}$ and $750^{\circ} \mathrm{C}, 850^{\circ} \mathrm{C}$, and $950^{\circ} \mathrm{C}$ respectively


Figure 6.14 Comparison of model prediction and experimental volatile content for Syncrude pitch at $50^{\circ} \mathrm{C} / \mathrm{min}$ and $750^{\circ} \mathrm{C}, 850^{\circ} \mathrm{C}$ and $950^{\circ} \mathrm{C}$ respectively

### 6.4 Discussion and Conclusion

Clearly for both CANMET and Syncrude pitches, evidence for the compensation effect Equation 6.1 was obtained for the kinetic parameters derived from the single first order reaction model, and the kinetic parameters derived from each stage of the 2 -stage kinetic model with different mathematical methods. It seems that the compensation effect is caused by the heating rates for each method used to analyzed the kinetic model equation, as shown in Figures 6.3 and 6.4. Since for these experiments were performed at the same atmosphere and roughly the same sample weight for each pitch type, these physico-chemical factors are therefore excluded. However, one set of compensation effect constants was found to be able to fit all the kinetic parameters derived from the 2-stage kinetic model with all the mathematical methods, as shown in Figures 6.5 and 6.6. This further indicates that the effects of heating rates and mathematical
methods are inseparable factors causing the kinetic compensation effect. Equation 6.2 was not met in the temperature range studied. The isokinetic temperature $\mathrm{T}_{\mathrm{i}}$ was therefore not observed as a result. Such an isokinetic temperature was not observed either for TVC-70 polymer thermal degradation [129]. The isokinetic temperature is more commonly observed in the catalysis kinetics and is often explained in terms of the temperature at which the catalyst was prepared [68, 69]. It is not surprising that the isokinetic temperature is not observed in pitch pyrolysis kinetics, since the temperature at which the pitch was prepared, has a totally different meaning. It seems safe to say that the measuring conditions and the methods used to analyze the kinetic models cause the kinetic compensation effect in this work and these factors are inseparable. This work however is not intended to investigate this effect in detail and identify the underlying factors as well as the mechanism of the kinetic compensation effect.

The Arrhenius equation of the kinetic parameters, which is rigorously valid for homogeneous reactions, is widely used for heterogeneous reactions, such as hydrocarbon pyrolysis, although such an extrapolation is not justified. Indeed, heterogeneous systems are characterized by supplementary problems due to complication of heterogeneous reactions. It has been shown [130] that for a series of related heterogeneous reactions, the compensation effect holds between the activation energy and pre-exponential factors. Compensation effects occur either for a series of reactions or for a given reaction when the operational parameters are changed. According to Garn [131-133], the common element of the reported cases of compensation effects is the existence of a main reaction which remains unaltered, in which a parameter regarded as a secondary factor changes the modification of the reaction rate with temperature. Audouin and Verdu [128] reported that a compensation effect appears only when the overall kinetic equation for thermal degradation is composed of many steps. It has been suggested that for such system at each moment a new material undergoes degradation and that
each reaction is characterized by a specific value of the activation energy. In catalytic reactions, the reaction rates have been proven associated with the distribution and concentration of active sites [134, 135].

For pyrolysis, the activation energy $E$ value has been observed to change with its conversion in nonisothermal experiments with oil shale [116], and in the two stages of pyrolysis in the present study. Pitch pyrolysis is such a process in which a series of reactions occur consecutively and/or concurrently, in the meantime the concentration of the active radicals decreases with the extent of reaction. When an inappropriate method is used to derive the two kinetic parameters, the error of one parameter caused by the method would be dumped to the other. However, these two parameters are related through the Arrhenius relationship. Since the pyrolysis rate is dependent on the remaining volatile content (or the reactive residue structure), the pyrolysis rate constants should be independent of the mathematical methods. The change of one parameter would be compensated to give the same rate. The existence of the compensation effect can therefore be attested. Similarly at different heating rates, the same "component" may undergo pyrolysis at different temperatures under TGA conditions. When one method is used to derive the kinetic parameters, the accuracy of the parameters to reflect the "true kinetics of that component" may be affected. Again the average kinetic behavior is retained by the Arrhenius relationship and consequently causing the compensation effect. However, the importance of the compensation effect may lie in the fact that the kinetic parameters $k_{o}$ and $E$ are interrelated for the pitch pyrolysis. This requires that the kinetic parameters of pitch pyrolysis be interpreted and compared with as a pair. One of the parameters may not be able to describe the whole picture of the pitch pyrolysis process. Care must also be exercised when using the reported kinetic parameters in research or design work.

It is also noted that these methods did not reproduce the kinetic parameters $k_{0}$ and $E$ at the operating conditions studied. The standard deviation caused by each pair of these parameters is not identical and it is possible to minimize the standard deviation through choosing the best pair of kinetic parameters $\mathrm{k}_{\mathrm{o}}$ and E .

At the onset of the research, no adequate data were available for the kinetics of the pitch pyrolysis and no mathematical models were available for the pitch pyrolysis mechanism. The primary goal of this research has been fulfilled in that the kinetic data for these processes have been outlined and a relevant kinetic model proposed.

### 7.1 SUMMARY OF FINDINGS

The principal observations and conclusions resulting from this study are listed below:

1. Heating rates were found to slightly affect the weight loss at a given temperature. The temperature history is the significant factor governing the extent to which the reactions take place and produce the weight loss. The devolatilization step is not instantaneous, as little weight loss occurred at the highest heating rates where the heating time took of the order of a few seconds.
2. The pyrolysis takes place in stages. At temperatures below $150^{\circ} \mathrm{C}$, there is little weight loss. The weight loss takes place in two following stages with two different, distinct patterns of chemical and physical change. In the first stage, the rate of the total weight loss increased with the temperature. In the second stage, the rate decreased with the temperature. These features appear unique to pitch pyrolysis, as they have not been reported for coal or shale pyrolysis.
3. The total weight loss (volatile yield) using thermogravimetric analysis decreased slightly with the increase of sample weight over the range of 3 to 17 mg for both CANMET pitch and Syncrude pitch. More than $80 \%$ of residue conversion was achieved for CANMET pitch, while more than $90 \%$ of residue conversion was achieved for Syncrude pitch.
4. Under Pyroprobe pyrolysis conditions, the pyrolysis time is a very important operating parameter. At the highest heating rate $\left(300,000{ }^{\circ} \mathrm{C} / \mathrm{min}\right)$ employed in this study, little pyrolysis was observed for both CANMET and Syncrude pitches up to $700^{\circ} \mathrm{C}$, while at heating rate of $600^{\circ} \mathrm{C} / \mathrm{min}$, the weight loss was rather significant when the final temperature was just reached ( 0 min isothermal reaction time). Higher heating rates exhibit complex effects on the weight loss and the secondary pyrolysis of the volatiles.
5. The most abundant component of the volatiles is shown experimentally to be hydrocarbons with less than the 10 carbons, which is grouped as single lump, $C_{7}$, in this study. At each heating rate and final temperature, the amount of $\mathrm{C}_{7}$ became significant at temperatures higher than $700^{\circ} \mathrm{C}$. As high as $50 \%$ volatile yield of this group in the total volatiles was detected for CANMET pitch and secondary reaction is observed at heating rate $3000{ }^{\circ} \mathrm{C} / \mathrm{min}$. At the Pyroprobe pyrolysis conditions, the volatiles may undergo secondary pyrolysis when being purged through the quartz tube. A similar trend is also observed for Syncrude pitch pyrolysis with the Pyroprobe-GC.
6. The yield of $\mathrm{C}_{10}$ compounds is very strongly influenced by the heating rates. At the highest heating rate $\left(300,000{ }^{\circ} \mathrm{C} / \mathrm{min}\right)$, less than $5 \%$ volatile yield of this group of components was detected, while as high as $25 \%$ volatile yield of $\mathrm{C}_{10}$ was detected at lower heating rates. This again attests to the influence of the reaction time and heating rates. The amount of $\mathrm{C}_{10}$ detected from Syncrude pyrolysis with Pyroprobe-GC is much less than that of CANMET pitch, which is in agreement with the differences of chemical structure or makeup of these two pitches. Higher yields of $\mathrm{C}_{11}, \mathrm{C}_{12}, \mathrm{C}_{13}$ and $\mathrm{C}_{14}$ groups were also detected at lower heating rates, a similar trend as that of $\mathrm{C}_{10}$ group. The yield of $\mathrm{C}_{14}$ is much less that those of $\mathrm{C}_{11}, \mathrm{C}_{12}$ and $\mathrm{C}_{13} . \mathrm{C}_{14}$ is the heaviest group of compounds detected in the Pyroprobe-GC pyrolysis, which suggests that the volatiles are mostly compounds lighter than $\mathrm{C}_{14}$. The yield of these
groups from Syncrude pitch pyrolysis with Pyroprobe-GC is also significantly less than those from CANMET pitch pyrolysis. This is in agreement with the $\mathrm{C}_{10}$ yield.
7. Although the pattern of volatile release for pitches pyrolyzed under TGA conditions is complex, an adequate description of the kinetics is possible by methods developed in this work. The pyrolysis takes place in 2 stages, with a first stage of low activation energy barrier and lower pre-exponential factor, and the second stage of higher activation energy and preexponential factor. It is recommended that the process be modeled with a 2 -stage first order reaction model using integral analysis method. It is demonstrated that the overall single stage reaction model using integral, Coats-Redfern, Chen-Nuttall and Friedman methods as well as Anthony and Howard's distributed activation energy model, are not sufficient to fit the TGA pyrolysis data and predict the course of the pitch pyrolysis process over the full range of conversion. It is also found that these single stage methods do not give similar values of activation energy and pre-exponential factors when data at different TGA conditions are taken for computation.
8. The 2-stage model reflects changes in the chemical constitution or structures as conversion proceeds by using two sets of activation energy. This feature is essential to describe pitch dependence of devolatilization rates on the remaining volatile content. The transition between these two stages is a sharp one, occurring at $450^{\circ} \mathrm{C}$ for both CANMET and Syncrude pitches. The magnitude of the activation energies suggests that the pyrolysis of pitch was kinetically controlled under the reaction conditions studied. The dependence of $\mathrm{dV} / \mathrm{dT}$ on $\mathrm{V}^{*}-\mathrm{V}$ is also in accordance with that. The activation energy of the second stage is higher than that of the first stage.
9. For both CANMET and Syncrude pitches, correlation between $k_{0}$ and $E$ values obtained via the different methods was observed. One set of compensation effect constants was found to
be able to fit all the kinetic parameters derived from all the 2 -stage kinetic analysis methods. An isokinetic temperature $T_{i}$ was not observed. These methods did not give similar kinetic parameters $k_{0}$ and $E$ at the operating conditions studied. The standard deviation caused by each pair of these parameters was not identical and it was possible to minimize the standard deviation through choosing the best pair of kinetic parameters $\mathbf{k}_{\mathbf{o}}$ and $E$.

### 7.2 RECOMMENDATIONS

The following recommendations are offered for further work and future application:

1. To achieve more detailed GC analysis of the Pyroprobe pyrolysis products, a longer column should be used and the gaseous and liquid components should be analyzed separately using cryogenic focus. The $\mathrm{C}_{7}$ should also be analyzed in detail with GC for gas components since it is the major lump for both CANMET and Syncrude Pitch.
2. To correlate the volatile yield with operating conditions such as heating rates, final temperature, and sample weights, a wider heating rate range should be used, such as heating rates as low as a few degrees per minute. Large sample weight ( $>18 \mathrm{mg}$ ) should also be used to study the internal mass transfer effect.
3. The critical temperature, dividing the two pyrolysis stage of pitch pyrolysis should be further studied with a variety of pitch samples of different origin. This temperature may be dependent on the pitch sample used.
4. To achieve a higher conversion and more light volatile yields, reactive pyrolysis environments such as hydrogen or steam should be used.
5. The use of a pilot-scale pyrolyzer is needed to explore the applicability of the 2 -stage mechanism and the related 2-stage first order integral method.
6. The application of the kinetic parameters of pitch pyrolysis should be concerned with the methods used to derived these values and the accuracy of the model predictions. The kinetic parameters $k_{0}$ and $E$ should be compared with and used as a pair. The magnitude of one of these parameters may not be adequate to describe the characteristics of a pitch pyrolysis process.

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APPENDICES

## APPENDIX A Methods Available for Computing Kinetic Parameters

Carrasco [62] has compared the activation energy results obtained by using the general analytical solution with those evaluated by means of established methods which were classified in three categories: integral (Table A.1), differential (Table A.2), and special methods (Table A.3). The comparison of the results and the related methods are summarized in the following tables. The accuracy (inaccuracy) of these methods was considered as the consequences of the simplification. These methods are however of no use due to the inaccuracy and their oversimplification for pitch pyrolysis.

Table A. 1 Summary of the Integral Methods [62]

| Author | Method | E |
| :---: | :---: | :---: |
| Analytical solution | $\begin{align*} & \ln \left[\frac{1-(1-f)^{1-n}}{(1-n) T^{2} \sum_{i=1}^{\infty}(-1)^{i+1} i!\left(\frac{R T}{E}\right)^{i-1}}\right]=\ln \left(\frac{A^{*} R}{\beta E}\right)-\frac{E}{R} \frac{1}{T} ; n \neq 1  \tag{2.9a}\\ & \ln \left[\frac{1-(1-f)}{T^{2} \sum_{i=1}^{\infty}(-1)^{i+1} i!\left(\frac{R T}{E}\right)^{i-1}}\right]=\ln \left(\frac{A^{*} R}{\beta E}\right)-\frac{E}{R} \frac{1}{T} ; n=1 \tag{2.9b} \end{align*}$ | 100\% |
| van Krevelen <br> et al. (1951) | $\begin{align*} & \ln \left[\frac{1-(1-f)^{1-n}}{1-n}\right]=\ln \left[\frac{A^{*}}{\beta}\left(\frac{0.368}{T_{m}}\right)^{E / R T_{m}} \frac{1}{\frac{E}{R T_{m}}+1}\right]+  \tag{2.10a}\\ & \left(\frac{E}{R T_{m}}+1\right) \ln T ; n \neq 1 \\ & \ln [-\ln (1-f)]=\ln \left(\frac{A^{*}}{\beta}\left(\frac{0.368}{T_{m}}\right)^{E / R T_{m}} \frac{1}{\frac{E}{R T_{m}}+1}\right]+  \tag{2.10b}\\ & \left(\frac{E}{R T_{m}}+1\right) \ln T ; n=1 \end{align*}$ | 108\% |
| Kissinger (1957) | $\begin{align*} & \ln \left[\frac{\beta}{T_{m}^{2}} \frac{1}{n\left(1-f_{m}\right)^{n-1}}\right]=\ln \left(\frac{A^{*} R}{E}\right) \frac{E}{R} \frac{1}{T_{m}} ; n \neq 1  \tag{2.11a}\\ & \ln \left[\frac{\beta}{T_{m}^{2}}\right]=\ln \left(\frac{A^{*} R}{E}\right) \frac{E}{R} \frac{1}{T_{m}} ; n=1 \tag{2.11b} \end{align*}$ | 80\% |

Table A. 1 (Continued)

| Horowitz and Metzger (1963) | $\begin{aligned} & \ln \left[\frac{1-(1-f)^{1-n}}{1-n}\right]=\frac{E}{R T_{s}^{2}} \theta ; n \neq 1 \\ & \ln [-\ln (1-f)]=\frac{E}{R T_{s}^{2}} \theta ; n=1 \end{aligned}$ <br> where: $\theta=T-T_{s}\left(T_{s}\right.$ is $T$ at which $f=(1-1 / e)$ for $n=1$ and $\mathrm{T}_{\mathrm{s}}=\mathrm{T}_{\mathrm{m}}$ for $\mathrm{n} \neq 1$ $A=\frac{\beta E}{R T_{s}^{2}} \exp \left(\frac{E}{R T_{s}}\right)$ | $\begin{aligned} & \text { (2.12a) } \\ & (2.12 b) \\ & (2.12 c) \end{aligned}$ | 116\% |
| :---: | :---: | :---: | :---: |
| Coats and Redfern (1965) (zero-order reaction) | $\begin{aligned} & \ln \left[\frac{f}{T^{2}\left(1-\frac{2 R T}{E}\right)}\right]=\ln \left(\frac{A^{*} R}{\beta E}\right)-\frac{E}{R} \frac{1}{T} \\ & \ln \left[\frac{f}{T^{2}}\right]=\ln \left(\frac{A^{*} R}{\beta E}\right)-\frac{E}{R} \frac{1}{T} ;(\text { when } R T \ll E) \end{aligned}$ | $\begin{aligned} & (2.13 a) \\ & (2.13 b) \end{aligned}$ | 94\% |

Table A. 2 Summary of Differential Methods [62]

| Author | Method | E |  |
| :--- | :---: | :---: | :---: |
| Classical | $\ln \left(\frac{\left(\frac{d f}{d T}\right)}{(1-f)^{n}}\right]=\ln \left(\frac{A^{*}}{\beta}\right)-\frac{E}{R} \frac{1}{T}$ | $(2.14)$ | Close |
| Multiple <br> linear <br> regression | $\ln \left(\frac{d f}{d T}\right)=\ln \left(\frac{A^{*}}{\beta}\right)+n \ln (1-f)-\frac{E}{R} \frac{1}{T}$ | $(2.15)$ | Close |
| Freeman and <br> Carroll <br> $(1958)$ | $\frac{\Delta \ln \left(\frac{d f}{d T}\right)}{\Delta \ln (1-f)}=n-\frac{E}{R} \frac{\Delta\left(\frac{1}{T}\right)}{\Delta \ln (1-f)}$ | $(2.16)$ | $90 \%$ <br> to |
| Vachusca <br> and Voboril <br> (1971) | $\frac{T^{2}\left(\frac{d^{2} f}{d T^{2}}\right)}{\left(\frac{d f}{d T}\right)}=\frac{E}{R}-n \frac{T^{2}\left(\frac{d f}{d T}\right)}{(1-f)}$ | $(2.17)$ | $110 \%$ |

Table A. 3 Summary of Special Methods [62]

| Author | Method | E |
| :---: | :---: | :---: |
| Reich (1964) | $\begin{equation*} E=\frac{R \ln \left[\frac{\beta_{1}}{\beta_{2}}\left(\frac{T_{1}}{T_{2}}\right)^{2}\right]}{\frac{1}{T_{1}}-\frac{1}{T_{2}}} \tag{2.18} \end{equation*}$ <br> $\mathrm{T}_{1}$ and $\mathrm{T}_{2}$ are measured at the same conversion value of two different heating rate runs. | 82\% |
| Friedman (1969) | $\begin{align*} & n=\frac{\ln \frac{(d f / d T)_{m}}{d f / d T}}{\frac{T_{m}\left(T_{m}-T\right)(d f / d T)_{m}}{T\left(1-f_{m}\right)}-\ln \left(\frac{1-f}{1-f m}\right)}  \tag{2.19a}\\ & \frac{E}{R}=\frac{n T_{m}^{2}(d f / d T)_{m}}{1-f_{m}}  \tag{2.19b}\\ & A=\frac{\beta\left(\frac{d f}{d T}\right)_{m}}{\left(1-f_{m}\right)^{n}} \exp \left(\frac{E}{R T_{m}}\right) \tag{2.19c} \end{align*}$ | 107\% |
| Reich and Stivala (1978) | $\begin{align*} & \ln \left[\frac{1-\left(1-f_{1}\right)^{1-n}}{1-\left(1-f_{2}\right)^{1-n}}\left(\frac{T_{2}}{T_{1}}\right)^{2}\right]=\frac{E}{R}\left(\frac{1}{T_{2}}-\frac{1}{T_{1}}\right) ; n \neq 1  \tag{2.20a}\\ & \ln \left[\frac{\ln \left(1-f_{1}\right)}{\ln \left(1-f_{2}\right)}\left(\frac{T_{2}}{T_{1}}\right)^{2}\right]=\frac{E}{R}\left(\frac{1}{T_{2}}-\frac{1}{T_{1}}\right) ; n=1  \tag{2.20b}\\ & A=\frac{\beta E K}{(1-n) R} ; n \neq 1 \tag{2.20c} \end{align*}$ <br> where $\operatorname{lnK}$ is the intercept of the line: <br> $\ln \left[\frac{1-(1-f)^{1-n}}{T^{2}}\right]$ vs $\frac{1}{T}$ | 98\% |
| Popescu and Segal (1983) | $\begin{align*} & E=R \frac{T_{0 i} T_{0 j}}{T_{0 j}-T_{0 i}} \ln \left[\left(\frac{\Delta T_{i}}{\Delta T_{j}}\right)^{2} \frac{\beta_{1}}{\beta_{2}}\right]  \tag{2.21a}\\ & \Delta T=T_{1}-T_{0} \end{align*}$ <br> $\mathrm{T}_{0}$ and $\mathrm{T}_{1}$ are characteristic temperatures $\begin{align*} & n=\frac{E}{R}\left(\frac{T_{1}-T_{m}}{T_{m}^{2}}\right)  \tag{2.21c}\\ & A=\frac{\beta E}{R(\Delta T)^{2}} \exp \left(\frac{E}{R T_{0}}\right) \end{align*}$ | 80\% |

## APPENDIX B GC Computer Station Method Parameters

```
**********************************************************************************
Varian GC Star Workstation - Method Listing Thu Jan 05 17:15:06 1995
Method: C:\STAR\PHILIP\PHILIPC.MTH
*********************************************************************************
    ADC Board
**************************************
Module Address: 16
            Autozero at Start : Y5.
                Channel A Name : PID
                    Channel B Name : FID
                    Channel A Full Scale : }10\mathrm{ VOLTS
                        Channel A Full Scale : }10\mathrm{ VOLTS
************************************
    GC 3600
***********************************
Module Address : 17
GC Injector A
Injector Type : Isothermal
Initial GC Injector Temperature
Initial GC Injector Hold Time
GC Injector Oven On?
GC Injector B
Injector Type : Isothermal
Initial GC Injector Temperature : 220 degree C
Initial GC Injector Hold Time : 0.00 minutes
GC Injector Oven On?
Coolant To Injector Value On?
Coolant Timeout
: No
: INFINITE
GC Auxiliary
Injector Type : Not used
GC Column
Column Oven On? : Yes
Initial Column Temperate
40 degree C
Initial Column Hold Time
10.00 minutes
Thermal Stabilization Time
3.00 minutes
Coolant To Injector Value On?
Coolant Timeout
No
: INFINITE
GC Column Program 1
Final Temperature
120 degree C
Rate
: 2.0 degrees C/minute
Hold Time
15.00 minutes
GC Column A Parameters
Installed? : Yes
Length
: Yes 
Diameter
255 microns
Carrier Gas : Helium
GC Column B Parameters
Installed?
```

| Length | : | 30 meters |  |
| :---: | :---: | :---: | :---: |
| Diameter | : | 255 microns |  |
| Carrier Gas | : | Helium |  |
| GC Detector Heater A |  |  |  |
| Detector Heater On? | : | Yes |  |
| Detector Temperature | : | 300 degrees C |  |
| GC Detector Heater B |  |  |  |
| Detector Heater On? | : | Yes |  |
| Detector Temperature | : | 250 degrees C |  |
| GC. Detector A |  |  |  |
| Detector Type | : | PID |  |
| Detector On? | : | Yes |  |
| Attenuation | : | 1 |  |
| Detector Range | : | 12 |  |
| Autozero at GC Ready? | : | Yes |  |
| GC Detector B |  |  |  |
| Detector Type | : | FID |  |
| Detector On? | : | Yes |  |
| Attenuation | : | 1 |  |
| Detector Range | : | 12 |  |
| Autozero at GC Ready? | : | Yes |  |
| Autosampler |  |  |  |
| Autosampler Type | : | Not used |  |
| GC Relays |  |  |  |
| Relay Time Program | : | Do Not Use |  |
| GC Stripchart |  |  |  |
| Stripchart On? | : | No |  |
| ********************************* |  |  |  |
| ADC Board |  |  |  |
| **************************** |  |  |  |
| Module Address : 16 |  |  |  |
| Integration Parameters |  |  |  |
| Run Mode | : | Analysis |  |
| Multiplier | : | 1.000000 |  |
| Divisor | : | 1.000000 |  |
| Amount Standard | : | 1.000000 |  |
| Unidentified Peak Factor | : | 0.000000 |  |
| Measurement | : | Peak Area |  |
| Calculation | : | External Standard |  |
| Report Unidentified Peaks | : | NO |  |
| Subtract Blank Baseline | : | Yes |  |
| Peak Rejection Value | : | 0 Counts |  |
| S/N Ratio | : | 5 |  |
| Tangent Height \% | : | 5\% |  |
| Initial Peak Width | : | 2 sec |  |
| Response factor Tolerance | : | Update All Response | Factors |
| Minimum Reference Window | : | 0.10 minutes |  |
| Percent Reference Window | : | 2.08 |  |
| Minimum NonReference Window | : | 0.10 minutes |  |
| Percent NonReference Window | : | 2.08 |  |
| Unretained Peak Time | : | 0.000 minutes |  |


| Peak Table |  |  |  |  |  |  |
| :--- | ---: | :--- | :--- | :--- | :--- | :--- |
| Name | Time | Factor | Amount | Ref. | Std. | RRT |
| C7 | 2.370 | 0.0028412 | 0.5700000 | N | N | N |
| C10 | 12.350 | 0.0094076 | 0.6083000 | N | N | N |
| C11 | 21.569 | 0.0098415 | 0.6175000 | N | N | N |
| C12 | 27.979 | 0.0134153 | 0.6258000 | N | N | N |
| C13 | 34.059 | 0.0128364 | 0.6308000 | N | N | N |
| C14 | 40.075 | 0.0209216 | 0.6357000 | N | N | N |

TimeEvents Table

| Group Event | : | 0.01 | until | 4.73 |
| :---: | :---: | :---: | :---: | :---: |
| Group Event | : | 8.84 | until | 15.86 |
| Group Event | : | 19.17 | until | 23.97 |
| Group Event | : | 24.24 | until | 31.72 |
| Group Event | : | 32.11 | until | 36.01 |
| Group Event | : | 37.49 | until | 42.66 |
| Inhibit Integrate | : | 42.67 | until | 60.00 |
| Report Format |  |  |  |  |
| Title | : | Standard Sample GC Analysis |  |  |
| Start Retention Time | : | 0.00 minutes |  |  |
| End Retention Time | : | 65.00 minutes |  |  |
| Initial Attenuation | : | 32 |  |  |
| Initial Zero Offset | : | 0 |  |  |
| Length in Pages | : | 1 |  |  |
| Initial Chart Speed | : | Off $\mathrm{cm} / \mathrm{min}$ |  |  |
| Minutes per Tick | : | 1.0 |  |  |
| Autoscale | : | On |  |  |
| Time Events | : | Off |  |  |
| Chromatogram Events | : | Off |  |  |
| Retention Times | : | On |  |  |
| Peak Names | : | On |  |  |
| Baseline | : | Off |  |  |
| Units | : | mg |  |  |
| Number of Decimal Digits | : | 4 |  |  |
| Run Log | : | Off |  |  |
| Error Log | : | On |  |  |
| Notes | : | Off |  |  |
| ASCII File Convert | : | Off |  |  |
| Print Chromatogram | : | On |  |  |
| Print Results | : | On |  |  |
| Copies | : | 1 |  |  |
| Sample Information |  |  |  |  |
| Detector Bunch | : | 8 points |  |  |
| Monitor Length | : | 64 points |  |  |
| Data File Name | - | stan |  |  |
| Channel | : | FID 10 VOLTS |  |  |
| Blank Baseline |  |  |  |  |
| Baseline Compression Factor | : | 128 |  |  |
| Baseline Points | : | 152 |  |  |
| Baseline Bunch Size | : | 8 |  |  |
| Baseline Frequency | : | 40.00 Hz |  |  |

## APPENDIX C Comparison of Equation 5.5 and Equation 5.6 Evaluated with Different Numbers of Terms of Integral $E_{i}(-E / R T)$

The accuracy of term V from Equation 5.6 was compared with that from Equation 5.6, using the kinetic parameters derived from CANMET pitch pyrolysis at $50^{\circ} \mathrm{C} / \mathrm{min}$ and $700^{\circ} \mathrm{C}$. The V term was integrated with Equation 5.5 and calculated with Equation 5.6, in which different numbers of terms of integral (Equation 5.7) were used. The results of $V$ were plotted in Figure C.1. As can be seen that the V term calculated with 3 to 8 terms of integral (Equation 5.7) is very close to that from the integration of Equation 5.5. It is therefore reasonable to use 3 terms of integral to estimate Equation 5.6 and simplify the mathematical process of Equation 5.5, and the accuracy of V thus obtained will not significantly affected.


Figure C.1 Comparison of V evaluated for Equation 5.5 and Equation 5.6


```
                    95.19D0,93.11D0,91.01D0,88.39D0,85.18D0,82.21D0,
                        78.23D0,73.51D0,68.8D0,65.87D0,61.13D0,55.71D0,
                        50.84D0,46.5DO,40.38D0,35.34D0,30.66DO,25.81D0,
                22.86D0,21.25D0,19.99D0,19.66D0,19.36D0,19.36D0/
    DATA T/0.D0,49.85D0,124.5DO,143.15D0,176.35D0,201.2D0,
                226.1D0,251.DO,275.D0,298.65D0,325.65D0,
                348.45D0,375.4D0,402.35D0,425.15D0,437.6D0,452.1D0,
                466.6D0,477.D0,485.3D0,495.65D0,503.95DO,512.25DO,
                    524.7DO,537.1D0,549.55D0,576.5D0,601.4D0,676.05DO,700.DO/
            DATA MM,N,L1,L2,R,VO,C/2,2,2,2,8.314DO,80.66D0,50.DO/
            OPEN(UNIT = 3, FILE = 'FIT.DAT',
    1 ACCESS = 'SEQUENTIAL', STATUS = 'NEW')
    E=250.D3
    DO 10 I=1,M
    V(I)=100.D0-V(I)
    T(I)=T(I)+273.16D0
    X(I)=1.D0/T(I)
10 CONTINUE
    DO }12\textrm{I}=1+\textrm{MM},
    X1(I-MM)=X (I)
12 CONTINUE
C
C
C
20 EOLD=E
    DO . }30\textrm{I}=1,\textrm{M
    Y(I)=DLOG(-C*DLOG(1.DO-V(I)/VO)/(R*T(I)*T(I)))
    1 -DLOG(1.DO-2.DO*R*T(I)/E)
30 CONTINUE
    CALL FLSQP(X,Y,M,N,A,VAR)
    E=-A(2)*R
    IF(DABS (E-EOLD).LT.0.1D-4) THEN
    RA=EXP(A(1))*E
    GOTO 40
    ENDIF
    GOTO 20
40 CONTINUE
    DO 60 I=1,M
    YFIT(I) =A(1)+A(2)*X(I)
    VFIT(I)=VO* (I.DO-EXP (-RA*R*T (I)*T (I)*EXP(-E/(R*T(I)))* (1.D0-
    1 2.DO*R*T(I)/E)/(C*E)))
6 0 ~ C O N T I N U E ~
    M1=M-MM
    E1=E
6 2 ~ E O L D = E 1
    DO 64 I=1+MM,M
    Y1 (I-MM)=DLOG(-C*DLOG(1.DO-V(I)/VO)/(R*T(I)*T(I)))
    1 -DLOG(1.D0-2.DO*R*T(I)/E1)
6 4 ~ C O N T I N U E ~
    CALL FLSQP(X1,Y1,M1,N,A,VAR)
    E1=-A(2)*R
    IF(DABS(E1-EOLD).GE.0.1D-4) GOTO 62
    RA1=EXP(A(1))*E1
    DO }66\textrm{I}=1,M
    YFIT1(I) =A(1) +A(2)*X1(I)
6 6 ~ C O N T I N U E ~
    DO }68 I=1,
    VFIT1 (I) =VO* (1.D0-EXP(-RA1*R*T (I) *T(I)*EXP(-E1/(R*T (I)))
    1 *(1.D0-2.DO*R*T(I)/E1)/(C*E1)))
68 CONTINUE
```

```
C FRIEDMAN METHOD
C
    MD=M-1
    DO }70\textrm{I}=1,\textrm{MD
    VD(I)=(V(I+1)-V(I))/(T(I+1)-T(I))
    7 0 ~ C O N T I N U E ~
    VD (M) = VD (MD)
    DO }80\textrm{I}=1+\textrm{L}1,M-L
    YD (I) =DLOG (C/VO*VD (I)) -DLOG(1.DO-V(I)/VO)
    80 CONTINUE
    ML=M-L1-L2
    DO 90 I=1,ML
    XD(I) =X(I+L1)
    YDD(I)=YD(I+L1)
    90 CONTINUE
    CALL FLSQP(XD,YDD,ML,N,AD,VARD)
    ED=-AD(2)*R
    RAD=EXP(AD(1))
    DO 100 I=1,ML
    YDDF (I) =AD (1) +AD (2)* XD (I)
    100 CONTINUE
    DO }110\textrm{I}=1,
    VDD(I)=VO*(1.D0-EXP(-RAD*R*T(I)*T(I)*EXP(-ED/(R*T(I)))*(1.DO-
    1 2.D0*R*T(I)/ED)/(C*ED)))
    110 CONTINUE
    ML1=ML-L1M1
    DO 112 I=1,ML1
    XD1 (I) =XD(I)
    YDD1 (I) =YDD (I)
    112 CONTINUE
    CALL FLSQP(XD1,YDD1,ML1,N,AD,VARD)
    ED1=-AD (2)*R
    RAD1=EXP(AD (1))
    DO 114 I=1,ML1
    YDDF1 (I) =AD (1) +AD (2) *XD1 (I)
    114 CONTINUE
    DO 116 I=1,M
    VDD1 (I) =VO* (1. D0-EXP(-RAD1*R*T(I)*T(I)*EXP (-ED1/(R*T(I)))*
    1 (1.D0-2.DO*R*T(I)/ED1)/(C*ED1)))
    116 CONTINUE
C
C
    COATS AND REDFERN METHOD
    DO 120 I=1,M
    YCR(I)=DLOG (-C*DLOG(1.DO-V(I)/VO)/(R*T (I)*T(I)))
120 CONTINUE
    CALL FLSQP(X,YCR,M,N,ACR,VAR)
    ECR=-R*ACR (2)
    RCR=ECR*EXP(ACR(1))
    DO 130 I=1,M
    YCRF(I) =ACR(1) +ACR (2) *X(I)
    VCR(I)=VO* (1.D0-EXP(-RCR*R*T(I)*T(I)*EXP(-ECR/(R*T(I)))* (I.D0-
    1 2.D0*R*T(I)/ECR)/(C*ECR)))
130 CONTINUE
    DO 132 I=1+MM,M
    YCR1 (I-MM) =DLOG (-C*DLOG(1.DO-V(I)/VO)/(R*T(I) *T(I)))
132 CONTINUE
    CALL FLSQP (X1,YCR1,M1,N,ACR,VAR)
    ECR1=-R*ACR(2)
    RCR1=ECR1*EXP(ACR(1))
    DO }134\textrm{I}=1,\textrm{M}
    YCRE1 (I) =ACR(1) +ACR(2)*X1 (I)
134 CONTINUE
```

```
            DO 136 I=1,M
            VCR1 (I) =VO* (1.D0-EXP(-RCR1*R*T(I)*T(I)*EXP(-ECR1/(R*T(I)))
            1 *(1.D0-2.D0*R*T(I)/ECR1)/(C*ECR1)))
    136 CONTINUE
C
C CHEN-NUTTAL METHOD
C
    ECN=E
    140 EOLD=ECN
    DO 150 I=1,M
    YCN (I)=DLOG (-C* (ECN+2.DO*R*T(I))*DLOG(1.DO-V(I)/VO)/(T(I)*T(I)*R))
    150 CONTINUE
    CALL FLSQP(X,YCN,M,N,ACN,VAR)
    ECN=-R*ACN (2)
    IF(DABS (ECN-EOLD) .GE.O.1D-4) GOTO 140
    RCN=EXP (ACN (1))
    DO 160 I=1,M
    YCNF (I) =ACN (1) +ACN (2.)*X(I)
    VCN (I) =VO* (1.D0-EXP (-RCN*R*T(I) *T (I)*EXP (-ECN/(R*T (I)))* (1.DO-
    1 2.DO*R*T(I)/ECN)/(C*ECN)))
    160 CONTINUE
    ECN1=E1
    162 EOLD=ECN1
        DO. 164 I=1+MM,M
        YCN1 (I-MM) =DLOG (-C* (ECN1+2.D0*R*T(I))*DLOG(1.DO-V(I)/VO)/(T (I)*
    1 T(I)*R))
    164 CONTINUE
    CALL FLSQP (X1, YCN1,M1,N,ACN,VAR)
    ECN1=-R*ACN (2)
    IF(DABS(ECN1-EOLD).GE.0.1D-4) GOTO 162
    RCN1=EXP (ACN (1))
    DO 166 I=1,M1
    YCNF1(I)=ACN(1) +ACN (2)*X1 (I)
    166 CONTINUE
        DO 168 I=1,M
            VCN1(I) =VO*(1.D0-EXP(-RCN1*R*T(I)*T(I)*EXP(-ECN1/(R*T(I)))*
        1 (1.D0-2.D0*R*T(I)/ECN1)/(C*ECN1)))
    168 CONTINUE
C
    WRITE (3,200)
200 FORMAT (4X,'T', 8X, 'V', 8X,'VFIT', 6X,'VDD',7X,'VCR',7X,'VCN',
    1 8X,'VD')
    DO 220 I=1,M
    WRITE(3,210)T(I),V(I),VEIT(I),VDD(I),VCR(I),VCN(I),VD(I)
210 FORMAT (E7.2,6F10.6)
220 CONTINUE
    WRITE (3,230)
230 FORMAT(6X,'X', 10X, 'Y', 8X, 'YFIT',7X, 'YCR',7X,'YCRF', 8X, 'YCN',
    1 7X,'YCNF')
    DO 250 I=M,1,-1
    WRITE (3,240) X(I),Y(I),YFIT(I),YCR(I),YCRE (I),YCN(I),YCNF(I)
240 FORMAT(F10.5,6F11.6)
250 CONTINUE
    WRITE (3,260)
260 FORMAT (10X,'X',13X,'YDD',11X,'YDDF')
    DO 280 I=ML,1,-1
    WRITE(3,270) XD(I),YDD(I),YDDF(I)
270 FORMAT (3F15.8)
280 CONTINUE
    WRITE(3,290)
290 FORMAT(35X,'A',14X,'E')
```

```
    WRITE(3,300) RA,E
    300 FORMAT('INTEGRAL METHOD',10X,2D15.3)
        WRITE(3,310) RAD,ED
    310 FORMAT ('FRIEDMAN METHOD',10X,2D15.3)
    WRITE(3,320) RCR,ECR
    320 FORMAT('COATS-REDFERN METHOD',5X,2D15.3)
    WRITE(3,330) RCN;ECN
    330 FORMAT ('CHEN-NUTTALL METHOD',6X,2D15.3)
C
    WRITE (3,340)
    340 FORMAT(//'ANALYSIS WITHOUT THE ABNORMAL END DATA POINTS')
    WRITE (3,200)
    DO 350 I=1,M
    WRITE(3,210) T(I),V(I),VFIT1(I),VDD1 (I),VCR1 (I),VCN1 (I),VD(I)
350 CONTINUE
    WRITE (3,230)
    DO 360 I=M1,1,-1
    WRITE(3,240) X1(I),Y1(I),YFIT1(I),YCR1(I),YCRF1 (I),YCN1(I),
    1 YCNF1 (I)
360 CONTINUE
    WRITE (3,260)
    DO 370 I=ML1,1,-1
    WRITE(3,270) XD1(I), YDD1(I),YDDF1 (I)
370 CONTINUE
    WRITE (3,290)
    WRITE(3,300) RA1, E1
    WRITE (3,310) RAD1, ED1
    WRITE (3,320) RCR1, ECR1
    WRITE(3,330) RCN1, ECN1
    ENDFILE(UNIT = 3)
    CLOSE(UNIT = 3)
    STOP.
    END
    SUBROUTINE GAUSS (A,N,NDR,NDC,X,RNORM,IREEOR)
    IMPLICIT REAL*8(A-H,O-Z)
    DIMENSION A(NDR,NDC),X(N),B(50,51)
    NM=N-1
    NP}=N+
    DO 20 I=1,N
    DO 10 J=1,NP
    B(I,J)=A(I,J)
    CONTINUE
    CONTINUE
    DO 70 K=1,NM
        KP=K+1
        BIG=ABS (B (K,K))
        IPIVOT=K
        DO 30 I=KP,N
        AB=ABS (B (I,K))
        IE(AB.GT.BIG) THEN
            BIG=AB
                IPIVOT=I
            ENDIF
    CONTINUE
    IF(IPIVOT.NE.K) THEN
        DO 40 J=K,NP
            TEMP=B(IPIVOT,J)
            B(IPIVOT,J)=B(K,J)
            B(K,J)=TEMP
```

    CONTINUE
    IF(B(N,N).EQ.O.DO) THEN
        IERROR=2
        RETURN
    ENDIF
    X(N)=B(N,NP)/B (N,N)
    DO 90 I=NM, 1,-1
        SUM=0.DO
        DO }80\textrm{J}=\textrm{I}+1,\textrm{N
            SUM=SUM+B(I,J)*X(J)
        CONTINUE
        X(I)=(B(I,NP)-SUM)/B(I,I)
    90 CONTINUE
    RSQ=0.DO
    DO 110 I=1,N
        SUM=0.D0
        DO }100\textrm{J}=1,\textrm{N
        SUM=SUM+A (I,J) *X(J)
    100 CONTINUE
        RSQ=RSQ+(A (I,NP)-SUM) **2
    110 CONTINUE
    RNORM=DSQRT (RSQ)
    IERROR=1
    RETURN
    END
    SUBROUTINE FLSQP(X,Y,M,N,A,VAR)
    IMPLICIT REAL*8(A-H,O-Z)
    DIMENSION X(M),U(51),Y(M),V(51),A(N),B(11), COEFF(10,11),SUMU (18)
    NP=N+1
    NM2=2* (N-1)
    XMIN=X(1)
    XMAX=X(1)
    YMIN=Y(1)
    YMAX=Y(1)
    DO 10 K=2,M
        XMIN=DMIN1 (XMIN,X(K))
        XMAX=DMAX1 (XMAX,X(K))
        YMIN=DMIN1 (YMIN, Y (K))
        YMAX=DMAX1 (YMAX,Y(K))
    CONTINUE
    XP=XMIN+XMAX
    XM=XMAX-XMIN
    YP=YMIN+YMAX
    YM=YMAX-YMIN
    DO 20 K=1,M
        U(K)=(2.D0*X(K)-XP)/XM
        V(K)=(2.DO*Y(K)-YP)/YM
    CONTINUE
    ```

CONTINUE
VAR \(=\) SSUM/ \((M-N)\) RETURN
END
DO \(30 \mathrm{~L}=1\), NM2 SUMU (L) \(=0\). DO
CONTINUE
DO \(40 \mathrm{I}=1, \mathrm{~N}\) \(\operatorname{COEFF}(I, N P)=0 . D 0\)
CONTINUE
DO \(70 \mathrm{~K}=1, \mathrm{M}\)
TERMU=U (K)
DO \(50 \mathrm{~L}=1\), NM2
\(\operatorname{SUMU}(L)=\operatorname{SUMU}(L)+\) TERMU TERMU=TERMU*U(K)
CONTINUE
TERMV=V (K)
DO \(60 \mathrm{I}=1, \mathrm{~N}\)
\(\operatorname{COEFF}(I, N P)=\operatorname{COEFF}(I, N P)+T E R M V\) TERMV=TERMV*U(K)
CONTINUE
CONTINUE
DO \(90 \mathrm{I}=1, \mathrm{~N}\)
DO \(80 \mathrm{~J}=1, \mathrm{~N}\)
IF(I.EQ.1.AND.J.EQ.1) THEN
\(\operatorname{COEFF}(I, J)=M\)
ELSE
\(\operatorname{COEFF}(I, J)=\operatorname{SUMU}(I+J-2)\)
ENDIF
CONTINUE
CONTINUE
CALL GAUSS (COEFF,N,10,11, B, RNORM, IERROR)
DO \(110 \mathrm{I}=1 \mathrm{~N}\)
\(\mathrm{IM}=\mathrm{I}-1\)
\(S U M=B(I)\)
IF(I.NE.N) THEN
DO \(100 \mathrm{~J}=\mathrm{I}+1, \mathrm{~N}\)
SUM \(=\) SUM + NOMIAL \((I M, J-1) *(-X P / X M) * *(J-I) * B(J)\)
CONTINUE
ENDIF
\(A(I)=Y M^{*}(2 . D 0 / X M) * * I M^{\star} S U M / 2 . D 0\)
CONTINUE
\(\mathrm{A}(1)=\mathrm{A}(1)+\mathrm{YP} / 2 . \mathrm{D} 0\)
SSUM=0.DO
DO \(130 \mathrm{~K}=1, \mathrm{M}\)
SUM=A (1)
TEMP=1. D0
DO \(120 \mathrm{~J}=2, \mathrm{~N}\)
TEMP=TEMP \({ }^{\star} X(K)\)
SUM \(=\) SUM + A \((J)\) *TEMP
CONTINUE
CONTINUE
SSUM=SSUM \(+(\mathrm{Y}(\mathrm{K})-S U M) * * 2\)

FUNCTION NOMIAL (I, J)
NOMIAL=1
IF (J.LE.I.OR.I.EQ.O) RETURN
DO 10 ICOUNT=1,I
NOMIAL=NOMIAL* (J-ICOUNT+1) /ICOUNT
CONTINUE
RETURN
END

Fitting Results for Run\#61 with Single Overall First Order Reaction Models
\begin{tabular}{|c|c|c|c|c|c|c|c|}
\hline T & V & IT & VDD & VC & VCN & \multicolumn{2}{|l|}{VD} \\
\hline 273.160. & 0.390000 & 0.001835 & 0.002088 & 0.002148 & 0.002148 & \multicolumn{2}{|l|}{0.000000} \\
\hline 323.010. & 0.390000 & 0.023641 & 0.025793 & 0.023518 & 0.025587 & \multicolumn{2}{|l|}{0.000000} \\
\hline 397.660. & 0.390000 & 0.345736 & 0.361193 & 0.290610 & 0.344936 & \multicolumn{2}{|l|}{60.011796} \\
\hline 416.31 0. & 0.610000 & 0.585664 & 0.606667 & 0.476451 & 0.575114 & \multicolumn{2}{|l|}{40.035241} \\
\hline 449 & 800 & 438 & 1. & 1.042352 & 1.291552 & \multicolumn{2}{|l|}{0.055533} \\
\hline 474 & 60 & 2.335505 & 2.365809 & 1.747062 & 2.201061 & \multicolumn{2}{|l|}{10.066265} \\
\hline 499.264. & 4.810000 & 3.831136 & . 849982 & 2.785390 & 3.559491 & \multicolumn{2}{|l|}{10.083534} \\
\hline 524.166. & 6.890000 & 5.980990 & 5.967688 & 4.245088 & 5.488858 & \multicolumn{2}{|l|}{80.087500} \\
\hline 548.168. & 8.990000 & 8.805194 & 8.732291 & 6.133121 & 8.000267 & \multicolumn{2}{|l|}{\(7 \quad 0.110782\)} \\
\hline 571.8111. & 11.610000 & 12.411952 & 12.245013 & 8.524594 & 11.187995 & \multicolumn{2}{|l|}{50.118889} \\
\hline 598.8114. & 14.820000 & 17.595572 & 17.272655 & 11.962289 & 15.756300 & \multicolumn{2}{|l|}{0.130263} \\
\hline 621.6117. & 17.790000 & 22.839938 & 22.346199 & 15.476020 & 20.38 & \multicolumn{2}{|l|}{\[
0.147681
\]} \\
\hline 648.5621. & 21.770000 & 29.925783 & 29.196143 & 20.332535 & 26.678746 & \multicolumn{2}{|l|}{-0.175139} \\
\hline 675.5126. & 26.490000 & 37.688713 & 36.712478 & 25.870783 & 33.669396 & \multicolumn{2}{|l|}{6 0.206579} \\
\hline 698.3131. & 31.200000 & 44.476911 & 43.311675 & 30.979525 & \multirow[t]{2}{*}{39.903756
43.346999} & \multicolumn{2}{|l|}{0.235341} \\
\hline 710.7634. & 34.130000 & 48.156937 & 46.905714 & 33.886906 & & \multicolumn{2}{|l|}{0.326897} \\
\hline 725.2638. & 38.870000 & 52.337141 & 51.007448 & 37.339291 & \multirow[t]{2}{*}{\[
\begin{aligned}
& 47.326756 \\
& 51.215466
\end{aligned}
\]} & \multicolumn{2}{|l|}{0.373793} \\
\hline 39.7644. & 44.290000 & 56.333893 & 54.953606 & 40.825516 & & \multicolumn{2}{|l|}{-0.468269} \\
\hline 750.1649 & 49.160000 & 59.050742 & 57.653200 & 43.324175 & \[
\begin{aligned}
& 51.215466 \\
& 53.915945
\end{aligned}
\] & \multicolumn{2}{|l|}{\multirow[t]{2}{*}{\[
\begin{aligned}
& 0.522892 \\
& 0.591304
\end{aligned}
\]}} \\
\hline 758.4653. & 53.500000 & 61.113169 & 59.713761 & 45.305721 & 56.002775 & \multicolumn{2}{|l|}{\multirow[t]{2}{*}{5.591304

0.607229}} \\
\hline 768.8159. & 59.620000 & 63.538995 & 62.152120 & 47.748745 & 58.505153
60.422057 & & \\
\hline 777.1164 & 64.660000 & 65.358755 & 63.993571 & 49.676613 & 60.422057 & \multicolumn{2}{|l|}{0.563855} \\
\hline 785.4169 & 69.340000 & 67.061108 & 65.727472 & 51.569099 & \multirow[t]{2}{*}{62.251570
64.819982} & \multicolumn{2}{|l|}{0.389558} \\
\hline 797.8674 & 74.190000 & 69.387621 & \multirow[t]{2}{*}{\[
\begin{aligned}
& 68.118229 \\
& 70.242956
\end{aligned}
\]} & 54.327247 & & \multicolumn{2}{|l|}{0.237903} \\
\hline 810.2677 & 77.140000 & 71.430956 & & 56.960587 & \multirow[t]{2}{*}{67.156536
69.272460} & \multicolumn{2}{|l|}{-0.129317} \\
\hline 822.7178 & 78.750000 & 73.211498 & 72.118937 & 59.473007 & & \multicolumn{2}{|l|}{0.046753} \\
\hline 849.6680 & 80.010000 & 76.196155 & 75.337198 & 64.396759 & 69.272460
73.063267 & \multicolumn{2}{|l|}{0.013253} \\
\hline 874.5680 & 80.340000 & 78.041713 & 77.401913 & 68.261742 & \[
\begin{aligned}
& 73.063267 \\
& 75.662338
\end{aligned}
\] & \multicolumn{2}{|l|}{\multirow[t]{2}{*}{\[
\begin{aligned}
& 0.004019 \\
& 0.000000
\end{aligned}
\]}} \\
\hline 949.2180 & 80.640000 & 80.292291 & 80.117552 & 75.953570 & 79.566076 & & \\
\hline 973.1680 & 80.640000 & 80.485864 & 80.384632 & 1350 & 80.041927 & \multicolumn{2}{|l|}{\[
0.000000
\]} \\
\hline x & & Y & FIT & YCR YCR & YCRF & YCN & YCNF \\
\hline 00103 & 03-9.17 & 9704 & 708 -9 & 50486-10 & . 057252 & . 931758 & 0.694442 \\
\hline 0.00105 & \(05-9.153\) & 17 -9 & 84969 -9 & 800649-10. & . 153178 & 0.973291 & 0.594723 \\
\hline 0.00114 & \(14-9.464\) & 4879. -9.94 & \(43113-10.0\) & 043230-10. & . 485878 & 0.704377 & 0.248866 \\
\hline 0.00118 & \(18-9.566\) & 6304-10.07 & 76571 -10.1 & 122604 -10. & . 609854 & 0.616063 & 0.119986 \\
\hline 0.00122 & \(22-9.778\) & 232-10.23 & 30119-10.3 & 311201 -10. & . 752494 & 0.417698 & -0.028295 \\
\hline 0.00123 & \(23-9.936\) & 6654-10.30 & 04503-10.4 & 459026-10. & . 821594 & 0.265328 & -0.100127 \\
\hline 0.00125 & \(25-10.132\) & \(2381-10.38\) & 80896-10.6 & 644309-10. & . 892559 & 0.075499 & -0.173899 \\
\hline 0.00127 & \(27-10.361\) & 1963-10.46 & 60023-10.8 & \(863514-10\). & . 966065 & -0.148292 & -0.250311 \\
\hline 0.00129 & \(29-10.541\) & \(1413-10.51\) & 14183 -11 & 036105-11 & . 016377 & -0.323952 & -0.302613 \\
\hline 0.00130 & \(30-10.712\) & 2212-10.56 & 69512-11.2 & 200091-11. & . 067776 & -0.491018 & -0.356045 \\
\hline 0.00132 & \(32-10.904\) & 4251-10.64 & 40204 -11.3 & \(383700-11\). & . 133445 & -0.678478 & -0.424311 \\
\hline 0.00133 & \(33-11.035\) & 5355-10.69 & 98303-11.5 & 508095-11. & . 187417 & -0.805974 & -0.480417 \\
\hline 0.00135 & \(35-11.181\) & 1704-10.77 & 72943 -11.6 & 646099-11.25 & . 256753 & -0.947876 & -0.552496 \\
\hline 0.00138 & \(38-11.345\) & 5282-10.88 & \(80580-11.7\) & \(798158-11\). & . 356744 & -1.105396 & -0.656441 \\
\hline 0.00141 & \(41-11.494\) & 4671-10.992 & \(92609-11.93\) & 936160-11. & . 460814 & -1.248887 & -0.764627 \\
\hline 0.00143 & \(43-11.586\) & \(6662-11.092\) & 92512-12.0 & 018476-11. & . 553619 & -1.335942 & -0.861103 \\
\hline 0.00148 & \(48-11.743\) & 3532-11.28 & \(85013-12.1\) & 157867 -11. & . 732445 & -1.484068 & -1.047000 \\
\hline 0.00154 & \(54-11.917\) & \(7910-11.53\) & 30008-12.3 & 311971 -11. & . 960034 & -1.648598 & -1.283590 \\
\hline 0.00161 & \(61-12.085\) & \(5959-11.79\) & 96246 -12 & 60149-12. & . 207358 & -1.807311 & -1.540695 \\
\hline 0.00167 & \(67-12.232\) & 2599-12.04 & 40200-12.5 & 590281-12. & . 433980 & -1.946443 & -1.776279 \\
\hline 0.00175 & \(75-12.426\) & 6729-12.35 & 54252-12.7 & 765208-12.72 & . 725721 & -2.132134 & -2.079558 \\
\hline 0.00182 & 82-12.632 & 2724-12.65 & 54757-12.9 & 954679-13. & . 004877 & -2.331130 & -2.369754 \\
\hline 0.00191 & 91-12.839 & 9908-12.98 & 97432-13.1 & \(145370-13\). & . 313918 & -2.531581 & -2.691016 \\
\hline 0.00200 & \(00-13.132\) & 2499-13.36 & 66387-13.4 & 421133-13. & . 665951 & -2.817570 & -3.056972 \\
\hline 0.00211 & \(11-13.477\) & \(7522-13.78\) & 85127-13.7 & 749606-14. & . 054942 & -3.156376 & -3.461346 \\
\hline 0.00222 & \(22-13.968\) & 8892-14.24 & 49277-14.22 & 224727-14 & . 486117 & -3.641916 & -3.909573 \\
\hline 0.00240 & \(40-14.914\) & 4993-14.95 & 55857-15 & 49524-15 & 42499 & -4.580806 & -4 \\
\hline . 00251 & & & & & & 7 & \\
\hline
\end{tabular}
\begin{tabular}{crr}
X & \multicolumn{1}{c}{ YDD } & \multicolumn{1}{c}{ YDDF } \\
0.00123417 & 0.60807557 & 0.04009520 \\
0.00125335 & 0.60895559 & -0.03490237 \\
0.00127322 & 0.54271001 & -0.11258481 \\
0.00128682 & 0.56647688 & -0.16575592 \\
0.00130071 & 0.36674818 & -0.22007510 \\
0.00131846 & 0.08485328 & -0.28947602 \\
0.00133305 & -0.18634571 & -0.34651452 \\
0.00135179 & -0.44043312 & -0.41979110 \\
0.00137882 & -0.80468691 & -0.52546313 \\
0.00140694 & -1.04618577 & -0.63544672 \\
0.00143203 & -1.43585946 & -0.73352542 \\
0.00148036 & -1.65717687 & -0.92251219 \\
0.00154188 & -1.90582279 & -1.16303342 \\
0.00160873 & -2.14174758 & -1.42441033 \\
0.00166998 & -2.31340313 & -1.66390972 \\
0.00174883 & -2.45237380 & -1.97222821 \\
0.00182428 & -2.56023804 & -2.26724630 \\
0.00190781 & -2.82504563 & -2.59384664 \\
0.00200296 & -2.89923461 & -2.96588292 \\
0.00210810 & -3.15234745 & -3.37697689 \\
0.00222464 & -3.34667896 & -3.83265228 \\
0.00240206 & -3.81617456 & -4.52633118 \\
0.00251471 & -4.91334683 & -4.96681167
\end{tabular}

INTEGRAL METHOD FRIEDMAN METHOD COATS-REDFERN METHOD CHEN-NUTTALL METHOD

YDDF
0.04009520
-0.03490237
\(-0.16575592\)
-0.22007510
28947602
-0.41979110
-0.52546313
3544672
-0.92251219
\(-1.16303342\)
\(-1.42441033\)
-1. 66390972
7222821
-2. 59384664
-2.96588292
.37697689
-4.52633118
A.
\(0.151 \mathrm{D}+03 \quad 0.331 \mathrm{D}+05\)
\(0.130 \mathrm{D}+03 \quad 0.325 \mathrm{D}+05\)
\(0.591 D+02 \quad 0.308 D+05\)
\(0.104 \mathrm{D}+03\)

\section*{E} \(0.320 \mathrm{D}+05\)
```

Anthony-Howard Model FORTRAN Program with L-M Nonlinear Regression
C
INTEGER NDATA,MA,MFIT,NCA
PARAMETER(NDATA=30,MA=3,MFIT=3,NCA=3)
DOUBLE PRECISION X(NDATA),Y(NDATA),A(MA),SIG(NDATA),AF,BF,YF,
1 COVAR (NCA, NCA), ALPHA (NCA, NCA), CHISQ,ALAMDA, YFIT (NDATA),T,
2 EO,S,Z
COMMON /ZDATA/T,EO,S,Z
EXTERNAL MRQMIN,MRQCOF,GAUSSJ,COVSRT,FUNCS,FUNCO, FUNC1,
1 FUNC2, FUNC3, QROMB,TRAPZD, POLINT
INTEGER LISTA(MA)
DATA LISTA/1,2,3/
DATA SIG/12*1.D-1,6*1.D-5,12*1.D-1/
DATA A/115000.DO,15000.D0,500000000.0D0/
C DATA FOR RUN 61:CAN61.DAT X=T (C) + 273.16 K
DATA X/273.16D0,323.01D0,397.66D0,416.31D0,449.51DO,474.36DO,
499.26DO,524.16DO,548.16D0,571.81DO,598.81DO,621.61DO,
648.56D0,675.51D0,689.31D0,710.76D0,725.26D0,739.76D0,
750.16DO,758.46DO,768.81DO,777.11DO,785.41DO,797.86DO,
810.26D0,822.71DO,849.66DO,874.56DO,949.21DO,973.16D0/
DATA Y/ 0.39D0, 0.39D0, 0.39D0, 0.61D0, 1.78D0, 3.16D0, 4.81D0,
1 6.89D0, 8.99D0,11.61D0,14.82D0,17.79DO,21.77DO,26.49DO,
2 31.20DO,34.13D0,38.87D0,44.29DO,49.16DO,53.50D0,59.62DO,
364.66D0,69.34D0,74.19D0,77.14D0,78.75DO,80.01D0,80.34DO,
4 80.64D0,80.64D0/
OPEN(UNIT=3,FILE='SLM61.DAT',ACCESS='SEQUENTIAL',
1 STATUS='OLD')
DO 10 I=1,NDATA
X(I)=X(I)-273.16D0
10 CONTINUE
ALAMDA=-0.001D0
CALL MRQMIN (X,Y,SIG,NDATA,A,MA, LISTA,MFIT,COVAR,
1 ALPHA,NCA, CHISQ, FUNCS, ALAMDA)
WRITE (3,100) A
100 FORMAT (5X,3F20.6)
VSTAR=Y(NDATA)
AF=A(1)-4.DO*A(2)
BF=A(1)+4.D0*A(2)
EO=A(1)
S=A(2)
z=A(3)
DO 200 I=1,NDATA
T=X(I) +273.16D0
CALL QROMB (FUNCO,AF,BF,YF)
YFIT(I)=VSTAR*(1.DO-YF/(2.506628DO*A(2)))
200 CONTINUE
DO 400 I=1,NDATA
WRITE(3,300) X(I),Y(I),YFIT(I)
300 FORMAT (5X,3F20.5)
400 CONTINUE
ENDFILE(UNIT=3)
CLOSE (UNIT=3)
STOP
END
SUBROUTINE MRQMIN(X,Y,SIG,NDATA,A,MA, LISTA,MFIT,
*
COVAR, ALPHA, NCA, CHISQ, FUNCS, ALAMDA)
INTEGER MMAX,MA,MFIT,NCA, KK, K,J,IHIT
PARAMETER (MMAX=20)
INTEGER LISTA(MA)
DOUBLE PRECISION X(NDATA),Y(NDATA),SIG(NDATA),A(MA),ALAMDA,CHISQ,

* COVAR(NCA,NCA), ALPHA (NCA, NCA), ATRY (MMAX),BETA (MMAX),DA (MMAX),

```
```

                    OCHISQ
    EXTERNAL FUNCS
    IF(ALAMDA.LT.O.DO) THEN
        KK=MFIT+1
        DO }12\textrm{J}=1,\textrm{MA
            IHIT=0
            DO 11 K=1,MFIT
                IF(LISTA(K).EQ.J) IHIT=IHIT+1
            CONTINUE
            IF (IHIT.EQ.0) THEN
                LISTA (KK)=J
                KK=KK+1
            ELSE IF (IHIT.GT.1) THEN
                PAUSE 'Improper permutation in LISTA'
                ENDIF
        CONTINUE
        IF (KK.NE. (MA+1)) PAUSE 'Improper permutation in LISTA'
        ALAMDA=0.001D0
        CALL MRQCOF(X,Y,SIG,NDATA,A,MA,LISTA,MFIT,ALPHA,BETA,NCA,CHISQ,F
    * UNCS)
        OCHISQ=CHISQ
        DO 13 J=1,MA
            ATRY(J)=A(J)
        CONTINUE
    ENDIF
    100 DO 15 J=1,MFIT
        DO 14 K=1,MFIT
            COVAR (J,K) =ALPHA (J,K)
        CONTINUE
        COVAR (J,J) =ALPHA (J,J) * (1.DO+ALAMDA)
            DA (J) = BETA (J)
        CONTINUE
        CALL GAUSSJ(COVAR,MFIT,NCA,DA,1,1)
        IF (ALAMDA.EQ.O.DO)THEN
            CALL COVSRT (COVAR,NCA,MA, LISTA,MFIT)
            RETURN
    ENDIF
    DO 16 J=1,MFIT
    C WRITE(3,*) 'DA(',J,')=',DA(J)
IF((DABS (DA(1)) +DABS (DA(2))+DABS(DA(3))).LT.1.D1) THEN
ATRY(LISTA(J))=A(LISTA(J)) +DA (J)
ELSE
ATRY(LISTA (J)) =A (LISTA (J)) +DA (J) *1.0D-60
ENDIF
16 CONTINUE
IF((DABS (DA(1))+DABS (DA(2))+DABS (DA (3))).LT.1.D-250) RETURN
CALI MRQCOF(X,Y,SIG,NDATA,ATRY,MA,LISTA,MFIT, COVAR,DA,NCA, CHISQ,
* FUNCS)
IF((DABS (CHISQ-OCHISQ)/CHISQ).LT.1D-3) RETURN
IF(CHISQ.LT.OCHISQ) THEN
ALAMDA=0.1D0*ALAMDA
OCHISQ=CHISQ
DO 18 J=1,MFIT
DO 17 K=1,MFIT
ALPHA (J,K) =COVAR (J,K)
CONTINUE
BETA (J)=DA (J)
A(LISTA(J))=ATRY(LISTA(J))
CONTINUE
ELSE
ALAMDA=10.D0*ALAMDA
CHISQ=OCHISQ
DO 180 J=1,MFIT

```
```

            DO 170 K=1,MFIT
                    ALPHA (J,K) =COVAR (J,K)
                    CONTINUE
            BETA(J)=DA(J)
            A(LISTA(J))=ATRY(LISTA(J))
            CONTINUE
        ENDIF
    C WRITE(*,*) 'CHISQ=',CHISQ,' OCHISQ=',OCHISQ
WRITE(*,*) A
GOTO 100
RETURN
END
SUBROUTINE MRQCOF (X,Y,SIG,NDATA,A,MA,LISTA,MFIT,ALPHA, BETA,NALP,
* CHISQ,FUNCS)
INTEGER NDATA,MA,MFIT,NALP, I, J, K,MMAX
PARAMETER (MMAX=3)
INTEGER LISTA(MFIT)
DOUBLE PRECISION X(NDATA),Y(NDATA),SIG(NDATA),ALPHA(NALP,NALP),
* BETA(MA),DYDA (3),A(3),XI,DY,SIG2I,WT,CHISQ
DO 12 J=1,MFIT
DO 11 K=1,J
ALPHA (J,K) =0.DO
CONTINUE
BETA (J) =0.DO
12 CONTINUE
CHISQ=0.DO
DO 15 I=1,NDATA
XI=X(I)
CALL FUNCS (XI,A,YMOD,DYDA, 3)
SIG2I=1.DO/(SIG(I)*SIG(I))
DY=Y(I) -YMOD
DO 14 J=1,MFIT
WT=DYDA(LISTA(J))*SIG2I
DO 13 K=1,J
ALPHA (J,K) =ALPHA (J,K) +WT*DYDA (LISTA (K))
CONTINUE
BETA (J) = BETA (J) +DY*WT
CONTINUE
CHISQ=CHISQ+DY*DY*SIG2I
CONTINUE
DO 17 J=2,MFIT
DO 16 K=1,J-1
ALPHA (K,J) =ALPHA (J,K)
CONTINUE
17 CONTINUE
RETURN
END
C
SUBROUTINE GAUSSJ (A,N,NP,B,M,MP)
INTEGER NMAX,N,NP,M,MP,I,J,K,IROW,ICOL,I,LL
PARAMETER (NMAX=50)
INTEGER IPIV(NMAX),INDXR (NMAX),INDXC (NMAX)
DOUBLE PRECISION A(NP,NP),B(NP,MP),DUM,BIG,PIVINV
DO 11.J=1,N
IPIV(J)=0
11 CONTINUE
DO 22 I=1,N
BIG=0.D0
DO 13 J=1,N
IF (IPIV(J).NE.1) THEN
DO 12 K=1,N
IF (IPIV (K).EQ.0) THEN

```
```

                    IF (ABS (A(J,K)).GE.BIG) THEN
                        BIG=ABS (A (J,K))
                        IROW=J
                        ICOL=K
                    ENDIF
                ELSE IF (IPIV(K).GT.1) THEN
                    PAUSE 'Singular matrix'
                    ENDIF
            CONTINUE
        ENDIF
        CONTINUE
        IPIV (ICOL) =IPIV (ICOL) +1
        IF (IROW.NE.ICOL) THEN
    DO 14 L=1,N
        DUM=A(IROW,L)
        A (IROW,L) =A (ICOL,L)
        A(ICOL,L)=DUM
    CONTINUE
    DO }15\textrm{L}=1,
        DUM=B (IROW,L)
        B (IROW,L) = B (ICOL,L)
        B (ICOL,L) =DUM
    CONTINUE
        ENDIF
        INDXR(I) =IROW
        INDXC(I)=ICOL
        IF (A(ICOL,ICOL).EQ.O.DO) PAUSE 'Singular matrix.'
        PIVINV=1./A(ICOL,ICOL)
        A (ICOL,ICOL ) =1.
        DO 16 L=1,N
    A(ICOL,L)=A(ICOL,L)*PIVINV
    CONTINUE
    DO }17\textrm{L}=1,
    B}(ICOL,L)=B(ICOL,L)*PIVINV
    CONTINUE
    DO 21 LL=1,N
    IF (LL.NE.ICOL) THEN
        DUM=A (LL,ICOL)
        A (LL,ICOL ) = . DO
        DO }18\textrm{L}=1,\textrm{N
                    A (LL,L) =A (LL,L) - A (ICOL,L) *DUM
        CONTINUE
        DO }19\textrm{L}=1,
                    B(LL,L)=B (LL,L) - B (ICOL,L) *DUM
            CONTINUE
        ENDIF
        CONTINUE
    CONTINUE
    DO }24\textrm{L}=\textrm{N},1,-
        IF (INDXR (L).NE.INDXC (L)) THEN
            DO 23 K=1,N
                DUM=A (K,INDXR (L))
                A(K,INDXR(L))=A(K,INDXC (L))
                A(K,INDXC (L))=DUM
            CONTINUE
            ENDIF
    24 CONTINUE
    RETURN
    END
    C
SUBROUTINE COVSRT (COVAR,NCVM,MA,LISTA,MFIT)
INTEGER NCVM,MA,LISTA,MFIT, I,J
INTEGER LISTA(MFIT)

```
```

    DOUBLE PRECISION COVAR(NCVM,NCVM),SWAP
    DO 12 J=1,MA-1
        DO }11\textrm{I}=\textrm{J}+1,M
        COVAR (I,J) =0.D0
    CONTINUE
    CONTINUE
    DO 14 I=1,MFIT-1
        DO 13 J=I+1;MFIT
            IF(LISTA(J).GT.LISTA(I)) THEN
                COVAR(LISTA (J),LISTA(I))=COVAR(I,J)
            ELSE
                COVAR(LISTA (I),LISTA (J)) =COVAR(I,J)
                ENDIF
    CONTINUE
    CONTINUE
    SWAP=COVAR (1, 1)
    DO 15 J=1,MA
        COVAR (1,J)=COVAR (J,J)
    COVAR (J;J) =0. DO
    CONTINUE
    COVAR(LISTA(1),LISTA(1))=SWAP
    DO 16 J=2,MFIT
        COVAR(LISTA (J),LISTA (J))=COVAR(1,J)
    CONTINUE
    DO 18 J=2,MA
        DO }17\textrm{I}=1,\textrm{J}-
        COVAR(I,J)=COVAR(J,I)
    CONTINUE
    CONTINUE
    RETURN
    END
    C
SUBROUTINE FUNCS (XX,A,Y,DYDA,NA)
INTEGER NA,NAA
DOUBLE PRECISION A(3),DYDA(3),XX,Y,T,EO,S,Z,VSTAR,YY,AA,BB,
* Y1,Y2,Y3
EXTERNAL FUNC0,FUNC1,FUNC2,FUNC3,QROMB,POLINT,TRAPZD
COMMON /ZDATA/T,EO,S,Z
NAA=NA
VSTAR=80.64D0
R=8.314D0
B=50.0D0
AA=A(1)-4.DO*A(2)
BB}=\textrm{A}(1)+4.D0*A(2
EO=A(1)
S=A (2)
Z=A(3)
T=XX+273.13D0
CALL QROMB (FUNCO,AA,BB,Y)
YY=Y
Y=VSTAR*(1.D0-YY/(2.506628D0*S))
CALL QROMB (FUNC1,AA, BB,Y1)
DYDA (1) =Y1+DEXP (-Z*R*T**2/(B* (EO+4.DO*S))*DEXP (- (EO+4.DO*S)/(R*T)
* )*(1.D0-2.DO*R*T/(EO+4.DO*S)))*DEXP (-8.D0)-DEXP(-Z*R*T**2
* /(B*(EO-4.DO*S))*DEXP(-(EO-4.DO*S)/(R*T))* (1.D0-2.DO*R*T/
* (EO-4.DO*S)))*DEXP(-8.DO)
DYDA (1) =-DYDA (1)*VSTAR/ (2.506628D0*S)
CALL QROMB (FUNC2,AA, BB,Y2)
DYDA (2)=Y2+4.DO*DEXP (-Z*R*T**2/(B* (EO+4.DO*S))*DEXP(- (EO+4.DO*S)/
* (R*T))*(1.D0-2.DO*R*T/(EO+4.DO*S)))*DEXP(-8.DO) +4.DO*DEXP
* (-Z*R*T**2/(B* (EO-4.DO*S))*DEXP(-(EO-4.DO*S)/(R*T))* (I.DO
* -2.DO*R*T/(EO-4.DO*S)))*DEXP(-8.DO)
DYDA (2) =-VSTAR*(YY/S-DYDA (2))/(2.506628DO*S)

```
```

    CALL QROMB (FUNC3,AA,BB,Y3)
    DYDA(3)=-Y3*VSTAR/(2.506628D0*S)
    RETURN
    END
    SUBROUTINE QROMB (FUNC,A,B,SS)
    INTEGER JMAX, JMAXP,J, K, KM, L
    DOUBLE PRECISION EPS,A,B,SS,DSS
    PARAMETER(EPS=5.D-3,JMAX=500,JMAXP=JMAX+1,K=5, KM=4)
    DOUBLE PRECISION S(JMAXP),H(JMAXP)
    EXTERNAL FUNC
    H(1)=1.
    DO 11 J=1, JMAX
        CALL TRAPZD(FUNC,A,B,S(J),J)
        IF (J.GE.K) THEN
        L=J-KM
    CALL POLINT(H(L),S(L),K,O.DO,SS,DSS)
    IF (DABS(DSS).LT.EPS*DABS(SS)) RETURN
        ENDIF
        S (J+1)=S (J)
    H(J+1)=0.25D0*H(J)
    11 CONTINUE
    PAUSE 'Too many steps.'
    END
    C
SUBROUTINE TRAPZD(EUNC,A,B,S,N)
INTEGER N,IT,J
DOUBLE PRECISION A,B,S,DEL,TNM,SUM,X
IF (N.EQ.1) THEN
S=0.5DO* (B-A)* (FUNC (A) +FUNC (B))
IT=1
ELSE
TNM=IT
DEL=(B-A)/TNM
X=A+0.5D0*DEL
SUM=0.DO
DO }11\textrm{J}=1,I
SUM=SUM+FUNC (X)
X=X+DEL
CONTINUE
S=0.5D0*(S+(B-A)*SUM/TNM)
IT=2*IT
ENDIF
RETURN
END
C
SUBROUTINE POLINT (XA, YA,N,X,Y,DY)
INTEGER I,M,N,NS,NMAX
PARAMETER (NMAX=10)
DOUBLE PRECISION XA(N),YA(N),C(NMAX),D(NMAX),X,Y,DY,DIFT,HO,HP,W,
*
DEN
NS=1
DIF=DABS (X-XA(1))
DO 11 I=1,N
DIFT=DABS (X-XA(I))
IF (DIFT.LT.DIF) THEN
NS=I
DIF=DIFT
ENDIF
C(I)=YA(I)
D(I)=YA(I)
CONTINUE
Y=YA(NS)

```
```

    NS=NS-1
    DO 13 M=1,N-1
        DO }12\textrm{I}=1,\textrm{N}-\textrm{M
            HO=XA(I)-X
            HP=XA(I+M)-X
            W=C(I+1)-D(I)
            DEN=HO-HP
            IF(DEN.EQ.O.DO)PAUSE
            DEN=W/DEN
            D(I) =HP*DEN
            C(I)=HO*DEN
    12 CONTINUE
        IF (2*NS.LT.N-M) THEN
            DY=C (NS+1)
        ELSE
            DY=D(NS)
            NS=NS-1
        ENDIF
            Y=Y+DY
    CONTINUE
RETURN
END
C
DOUBLE PRECISION FUNCTION FUNCO(E)
DOUBLE PRECISION T,EO,S,Z,E
COMMON/ZDATA/T,EO,S,Z
R=8.314DO
B=50.0D0
FUNCO=DEXP(-Z*R*T**2/(B*E)*DEXP(-E/(R*T))* (1.DO-2.DO*R*T/E))
1
RETURN
END
C
C
DOUBLE PRECISION FUNCTION FUNC2(E)
DOUBLE PRECISION T,EO,S,Z,E
COMMON /ZDATA/T,EO,S,Z
R=8.314D0
B=50.0D0
FUNC2=DEXP (-Z*R*T**2/(B*E)* DEXP (-E/(R*T))*(1.D0-2.DO*R*T/E))*
1
RETURN
END
C
DOUBLE PRECISION FUNCTION FUNC3(E)
DOUBLE PRECISION T,EO,S,Z,E
COMMON /ZDATA/T,EO,S,Z
R=8.314D0
B=50.0D0
FUNC3=DEXP (-Z*R*T**2/(B*E)*DEXP (-E/ (R*T))*(1.D0-2.D0*R*T/E))* (-R

* *T**2/(B*E))*DEXP(-E/(R*T))* (1.D0-2.D0*R*T/E)*DEXP(-0.5DO* ((E-
* EO)/S)**2)
RETURN
END

```

Fitting Result with Anthony-Howard Model for RUN\#61
\begin{tabular}{|c|c|c|}
\hline E J/mol & \(5 \mathrm{~J} / \mathrm{mol}\) & \(\mathrm{k} \mathrm{s}^{-1}\) \\
\hline 114999.999951 & 14999.998064 & 500000004.536339 \\
\hline T \({ }^{\circ} \mathrm{C}\) & \(\mathrm{V}_{\text {exp }}\) & \(\mathrm{V}_{\text {mod }}\) \\
\hline . 00000 & . 39000 & -. 00868 \\
\hline 49.85000 & . 39000 & . 00783 \\
\hline 124.50000 & . 39000 & . 04946 \\
\hline 143.15000 & . 61000 & . 10393 \\
\hline 176.35000 & 1.78000 & . 34987 \\
\hline 201.20000 & 3.16000 & . 82673 \\
\hline 226.10000 & 4.81000 & 1.69501 \\
\hline 251.00000 & 6.89000 & 3.04457 \\
\hline 275.00000 & 8.99000 & 5.27093 \\
\hline 298.65000 & 11.61000 & 8.66743 \\
\hline 325.65000 & 14.82000 & 13.68217 \\
\hline 348.45000 & 17.79000 & 18.60192 \\
\hline 375.40000 & 21.77000 & 26.42453 \\
\hline 402.35000 & 26.49000 & 36.10305 \\
\hline 416.15000 & 31.20000 & 41.01006 \\
\hline 437.60000 & 34.13000 & 47.87946 \\
\hline 452.10000 & 38.87000 & 52.29991 \\
\hline 466.60000 & 44.29000 & 56.74668 \\
\hline 477.00000 & 49.16000 & 59.87655 \\
\hline 485.30000 & 53.50000 & 62.26663 \\
\hline 495.65000 & 59.62000 & 65.02800 \\
\hline 503.95000 & 64.66000 & 67.00712 \\
\hline 512.25000 & 69.34000 & 68.74093 \\
\hline 524.70000 & 74.19000 & 70.88771 \\
\hline 537.10000 & 77.14000 & 72.60551 \\
\hline 549.55000 & 78.75000 & 74.08491 \\
\hline 576.50000 & 80.01000 & 76.79565 \\
\hline 601.40000 & 80.34000 & 78.57816 \\
\hline 676.05000 & 80.64000 & 80.37077 \\
\hline 700.00000 & 80.64000 & 80.52666 \\
\hline
\end{tabular}

\section*{APPENDIX E FORTRAN Program for Two-Stage First Order Reaction Model}
\begin{tabular}{|c|c|}
\hline C & Original experime \\
\hline C & \(V(M) \quad\) Experimental volatile content \\
\hline C & \(\mathrm{VD}(\mathrm{M}) \quad\) Experimental \(\mathrm{dV} / \mathrm{dT}, \mathrm{VD}(\mathrm{I})=(\mathrm{V}(\mathrm{I}+1)-\mathrm{V}(\mathrm{I})) /(\mathrm{T}(\mathrm{I}+1)-\mathrm{T}(\mathrm{I})\) ) \\
\hline C & \(T(M)\) Experimental temperature \\
\hline C & \(\mathrm{X}(\mathrm{M}) \quad 1 / \mathrm{T}(\mathrm{M})\) in K \\
\hline C & M No. of experimental data points \\
\hline C & LN No. of experimental data points omitted at the beginning for \\
\hline C & fitting \\
\hline C & \\
\hline C & VF Fitting results in the entire temperature range \\
\hline C & \multirow[t]{2}{*}{First reaction fitting parameters (Section below temperature 450 oc)} \\
\hline C & \\
\hline C & \(N \quad N o\). of first reaction data points \\
\hline C & V1(N) Experimental volatile content of \\
\hline C & VD1 (N) Experimental dV/dT \\
\hline C & T1(N) Experimental temperature \\
\hline C & Xl (N) \(1 / \mathrm{T} 1\) (N) \\
\hline C & A1(2) Fitting array used in FLSQP subroutine for first reaction \\
\hline C & Y1(N) Fitting parameter derived from V1(N) using relative method \\
\hline C & Y1F(N) Fitted value for Y1(N) using relative method \\
\hline C & V1F(N) Fitted volatile using relative method \\
\hline C & \\
\hline C & First reaction fitting parameters (Section below temperature 450 OC) \\
\hline C & MN No. of second reaction data points \\
\hline C & V2(MN) Experimental volatile content \% \\
\hline C & VD2 (MN) Experimental dV/dT \\
\hline C & T2 (MN) Experimental temperature \\
\hline C & X 2 (MN) 1/T2 (MN) \\
\hline C & A2 (2) Fitting array used in FLSQP subroutine for second reaction \\
\hline C & Y2 (MN) Fitting parameter derived from V2 (MN) using relative method \\
\hline C & Y2F(MN) Fitted value for Y2 (MN) using relative method \\
\hline C & V2F(MN) Fitted volatile using relative method \\
\hline C & \\
\hline C & Subscript for each method \\
\hline C & \\
\hline C & NT Integral method \\
\hline C & CR Coats-Redfern method \\
\hline C & CN Chen-Nuttall method \\
\hline C & FM Friedman method \\
\hline C & \\
\hline & IMPLICIT REAL* 8 ( \(\mathrm{A}-\mathrm{H}, \mathrm{O}-\mathrm{Z}\) ) \\
\hline & PARAMETER ( \(\mathrm{M}=37, \mathrm{~N}=14, \mathrm{MN}=14, \mathrm{LN}=4\) ) \\
\hline & EXTERNAL NOMIAL \\
\hline & DIMENSION V(M), T(M), X (M), A1 (2), A2 (2), VD (M) \\
\hline & DIMENSION V1 (N), T1 (N), X1 (N), V2 (MN), T2 (MN), X2 (MN), VD1 (N), VD2 (MN), \\
\hline & \(1 \mathrm{Y} 1 \mathrm{NT}(\mathrm{N}), \mathrm{Y} 1 \mathrm{FNT}(\mathrm{N}), \mathrm{V} 1 \mathrm{FNT}(\mathrm{N}), \mathrm{Y} 2 \mathrm{NT}(\mathrm{MN}), \mathrm{Y} 2 \mathrm{FNT}(\mathrm{MN}), \mathrm{V} 2 \mathrm{ENT}(\mathrm{MN}), \mathrm{VFNT}(\mathrm{M})\), \\
\hline & 2. \(\mathrm{Y} 1 \mathrm{CR}(\mathrm{N}), \mathrm{Y} 1 \mathrm{FCR}(\mathrm{N}), \operatorname{V1FCR}(\mathrm{N}), \mathrm{Y} 2 \mathrm{CR}(\mathrm{MN}), \mathrm{Y} 2 \mathrm{FCR}(\mathrm{MN}), \operatorname{V2FCR}(\mathrm{MN}), \mathrm{VFCR}(\mathrm{M})\), \\
\hline & \(3 \mathrm{Y} 1 \mathrm{CN}(\mathrm{N}), \mathrm{Y} 1 \mathrm{FCN}(\mathrm{N}), \mathrm{V1FCN}(\mathrm{~N}), \mathrm{Y} 2 \mathrm{CN}(\mathrm{MN}), \mathrm{Y} 2 \mathrm{FCN}(\mathrm{MN}), \mathrm{V} 2 \mathrm{FCN}(\mathrm{MN}), \mathrm{VFCN}(\mathrm{M})\), \\
\hline & \(4 \mathrm{Y} 1 \mathrm{FM}(\mathrm{N}), \mathrm{Y} 1 \mathrm{FFM}(\mathrm{N}), \mathrm{V} 1 \mathrm{FFM}(\mathrm{N}), \mathrm{Y} 2 \mathrm{FM}(\mathrm{MN}), \mathrm{Y} 2 \mathrm{FFM}(\mathrm{MN}), \mathrm{V} 2 \mathrm{FFM}(\mathrm{MN}), \mathrm{VFFM}(\mathrm{M})\) \\
\hline & DIMENSION VDNT (M), VDCR (M), VDCN (M), VDFM (M) \\
\hline C & \multirow[t]{2}{*}{\begin{tabular}{l}
canmet pitch \(25 \mathrm{oC} / \mathrm{min}\) and 800 oC \\
DATA V/99.96D0,99.94D0,99.50D0,98.98D0,97.83D0,96.14D0,94.43D0,
\end{tabular}} \\
\hline & \\
\hline & \\
\hline & \[
\begin{aligned}
& 1 \text { 92.41D0,90.17D0,87.92D0,85.52D0,82.67D0,81.30D0,79.35D0,77.45DO, } \\
& 2 \quad 74.91 \mathrm{DO}, 72.41 \mathrm{DO}, 69.25 \mathrm{D} 0,65.96 \mathrm{D}, 62.65 \mathrm{D} 0,59.69 \mathrm{DO}, 56.33 \mathrm{D} 0,51.74 \mathrm{DO},
\end{aligned}
\] \\
\hline & 46.48D0, 41.57D0, 36.49D0, 32.58D0,28.48D0,25.14DO, 22.84D0,21.38D0, \\
\hline & 4 20.30D0,19.97D0,19.71D0,19.48D0,19.28D0,19.16D0/ \\
\hline & DATA T/ 50.22D0,100.32D0,135.40D0,153.77D0,175.47D0,200.52DO, \\
\hline & \(1225.60 \mathrm{DO}, 250.65 \mathrm{DO}, 275.70 \mathrm{DO}, 300.75 \mathrm{DO}, 325.80 \mathrm{DO}, 350.85 \mathrm{DO}, 362.55 \mathrm{DO}\), \\
\hline & \(2375.90 \mathrm{DO}, 387.60 \mathrm{DO}, 400.95 \mathrm{DO}, 414.32 \mathrm{D} 0,426.00 \mathrm{DO}, 437.70 \mathrm{DO}, 447.72 \mathrm{DO}\), \\
\hline & 456.07D0,464.42D0,474.45D0,484.45D0,492.80DO, 500.00D0,507.85DO, \\
\hline & \(4516.20 \mathrm{DO}, 526.22 \mathrm{DO}, 537.90 \mathrm{DO}, 551.27 \mathrm{D} 0,576.32 \mathrm{DO}, 601.37 \mathrm{DO}, 649.80 \mathrm{DO}\), \\
\hline
\end{tabular}
```

    5 699.90D0,750.00D0,800.12D0/
    DATA R,VO,C/8.314D0,80.84D0,25.D0/
    OPEN(UNIT = 3, FILE = 'FIT2RN1.DAT',
    1 ACCESS = 'SEQUENTIAL', STATUS = 'OLD')
    E1=250.D3
    DO 10 I=1,M
    V(I)=100.D0-V(I)
    T(I)=T(I)+273.16D0
    X(I)=1.D0/T(I)
    10 CONTINUE
DO 20 I=1,N
V1 (I) =V (LN+I)
T1(I)=T(LN+I)
X1(I) =X (LN+I)
20 CONTINUE
DO 30 I=1,MN
V2(I) = V(LN+N+I)
T2(I) =T(LN+N+I)
X2(I) =X (LN+N+I)
30 CONTINUE
C
C
C
40 E1OLD=E1
DO 50 I=1,N
Y1NT(I)=DLOG(-C*DLOG(1.D0-V1(I)/VO)/(R*T1 (I)*T1 (I)))
1-DLOG(1.DO-2.D0*R*T1(I)/E1)
50 CONTINUE
CALL FLSQP(X1,Y1NT,N,2,A1,VAR)
E1=-A1 (2)*R
IF(DABS (E1-E1OLD).LT.1.OD-8) THEN
RK1=EXP(A1 (1))*E1
VAR1=VAR
GOTO 60
ENDIF
GOTO 40
6 0 ~ C O N T I N U E ~
DO }70\textrm{I}=1,
Y1FNT (I) =A1 (1) +A1 (2)*X1 (I)
V1FNT(I) =VO*(1.D0-EXP(-RK1*R*T1(I)*T1(I)*EXP(-E1/(R*T1(I)))*(1.D0-
1 2.D0*R*T1(I)/E1)/(C*E1)))
7 0 ~ C O N T I N U E
C
C
C
E2=E1
80 E2OLD=E2
DO 90 I=1,MN
Y2NT (I) =DLOG (-C*DLOG (1.D0-V2 (I)/VO)/(R*T2(I) *T2 (I)))
1 -DLOG(1.D0-2.DO*R*T2(I)/E2)
90 CONTINUE
CALL FLSQP(X2,Y2NT,MN,2,A2,VAR)
E2=-A2 (2)*R
IF(DABS (E2-E2OLD).GE.1.0D-8) GOTO }8
RK2=EXP(A2 (1))*E2
DO }100\textrm{I}=1,\textrm{MN
Y2FNT (I) =A2(1) +A2(2)*X2(I)

```
```

        V2FNT(I)=VO*(1.D0-EXP(-RK2*R*T2(I)*T2(I)*EXP(-E2/(R*T2(I)))
        1 *(1.D0-2.D0*R*T2(I)/E2)/(C*E2)))
    100 CONTINUE
        WRITE(*,*) 'INTEGRAL METHOD SUCCESSFUL'
    C
C Coats-Redfern method
C
C
C
First reaction
DO 110 I=1,N
Y1CR(I)=DLOG(-C*DLOG(1.D0-V1(I)/VO)/(R*T1 (I)*T1 (I)))
110 CONTINUE
CALL FLSQP(X1,Y1CR,N,2,A1,VAR)
E1CR=-A1(2)*R
RK1CR=EXP (A1 (1)) *E1CR
DO 120 I=1,N
Y1FCR(I)=A1 (1) +A1 (2) *X1 (I)
V1FCR(I)=VO*(1.D0-EXP(-RK1CR*R*T1(I)*T1(I)*EXP(-E1CR/(R*T1 (I)))*
1 (1.D0-2.D0*R*T1(I)/E1CR)/(C*E1CR)))
120 CONTINUE
C
C
C
Second reaction
DO 130 I=1,MN
Y2CR(I)=DLOG (-C*DLOG(1.D0-V2(I)/VO)/(R*T2(I)*T2(I)))
130 CONTINUE
CALL FLSQP(X2,Y2CR,MN,2,A2,VAR)
E2CR=-A2(2)*R
RK2CR=EXP(A2 (1)) *E2CR
DO 140 I=1,MN
Y2FCR(I) =A2 (1) +A2 (2)*X2(I)
V2FCR(I)=VO*(1.D0-EXP (-RK2CR*R*T2(I)*T2(I)*EXP(-E2CR/(R*T2(I)))
1 *(1.DO-2.D0*R*T2 (I)/E2CR)/(C*E2CR)))
140 CONTINUE
WRITE(*,*) 'COATS-REDFERN METHOD SUCESSFUL'
C
C
C
Chen-Nuttall method
First reaction
E1CN=E1CR
150 EOLD=E1CN
DO 160 I=1,N
Y1CN (I) =DLOG (-C* (E1CN+2.D0*R*T1 (I))*DLOG(1.D0-V1 (I)/VO)/
1 (T1(I)*T1(I)*R))
160 CONTINUE
CALL FLSQP (X1, Y1CN,N,2,A1,VAR)
E1CN=-R*A1 (2)
IF(DABS (E1CN-EOLD).GE.1.OD-8) GOTO 150
RK1CN=EXP(A1 (1))
D0 170 I=1,N
Y1FCN(I) =A1(1) +A1 (2)*X1(I)
V1FCN(I)=VO*(1.D0-EXP(-RK1CN*R*T1(I)*T1(I)*EXP(-E1CN/(R*T1(I)))*
1 (1.D0-2.DO*R*T1(I)/E1CN)/(C*E1CN)))
170 CONTINUE
C
C
C
E2CN=E1CN
180 EOLD=E2CN
DO 190 I=1,MN
Y2CN (I)=DLOG (-C* (E2CN+2.D0*R*T2 (I))*DLOG(1.D0-V2 (I)/VO)/(T2 (I)*

```
```

        1
        T2(I)*R))
    190 CONTINUE
        CALL FLSQP (X2,Y2CN,MN,2,A2,VAR)
        E2CN=-R*A2 (2)
        IF(DABS (E2CN-EOLD).GE.1.0D-8) GOTO 180
        RK2CN=EXP(A2(1))
        DO 200 I=1,MN
        Y2FCN(I)=A2(1)+A2(2)*X2(I)
        V2FCN (I) =VO* (1.D0-EXP(-RK2CN*R*T2(I)*T2(I)*EXP(-E2CN/(R*T2(I)))*
    1 (1.D0-2.DO*R*T2(I)/E2CN)/(C*E2CN)))
    200 CONTINUE
    WRITE(*,*) 'CHEN-NUTTAL METHOD SUCESSFUL'
    C
C
C Friedman Method
C
DO 210 I=1,M-1
VD (I) = (V(I+1)-V(I))/.(T(I+1)-T(I))
210 CONTINUE
VD (M) = VD (M-1)
DO 220 I=1,N
VD1 (I) =VD (LN+I)
220 CONTINUE
DO 230 I=1,MN
VD2 (I) = VD (LN+N+I)
230 CONTINUE
C
C
C
First reaction
DO 240 I=1,N
Y1FM(I)=DLOG(C/VO*VD1 (I))-DLOG(1.DO-V1 (I)/VO)
240 CONTINUE
CALL FLSQP(X1,Y1FM,N,2,A1,VAR)
E1FM=-A1 (2)*R
RK1FM=EXP(A1 (1))
DO 250 I=1,N
Y1FFM(I)=A1 (1) +A1 (2) *X1 (I)
V1FFM(I) =VO* (1.D0-EXP(-RK1FM*R*T1(I)*T1(I)*EXP(-E1FM/(R*T1(I)))*
1 (1.0D0-2.D0*R*T1 (I)/E1FM)/(C*E1FM)))
250 CONTINUE
C
C Second reaction
C
DO 260 I=1,MN
Y2FM(I)=DLOG(C/VO*VD2 (I))-DLOG(1.D0-V2 (I)/VO)
260 CONTINUE
CALL FLSQP(X2,Y2FM,MN,2,A2,VAR)
E2FM=-A2(2)*R
RK2FM=EXP(A2 (1))
DO 270 I=1,MN
Y2FFM(I) =A2(1)+A2(2)*X2(I)
V2FFM(I)=VO*(1.D0-EXP(-RK2FM*R*T2(I)*T2(I)*EXP(-E2FM/(R*T2(I)))*
1 (1.D0-2.D0*R*T2(I)/E2FM)/(C*E2FM)))
270 CONTINUE
WRITE(*,*) 'FRIEDMAN METHOD SUCESSFUL'
Calculate the volatile content in the entire temperature range
DO $280 \mathrm{I}=1, \mathrm{M}$
IF ( (T (I) - 273.16 ). LT. 450.D0) THEN
$\operatorname{VFNT}(I)=V O^{*}\left(1 . \operatorname{DO}-\operatorname{EXP}\left(-R K 1 * R^{\star} T(I) * T(I) * \operatorname{EXP}(-E 1 /(R * T(I))) *(1 . D 0-\right.\right.$

```

1 2.D0*R*T(I)/E1)/(C*E1)))
\(\operatorname{VFCR}(I)=\operatorname{VO}\left(1 . \operatorname{DO}-\operatorname{EXP}\left(-R K 1 C R^{\star} R^{*} T(I) * T(I) * \operatorname{EXP}(-E 1 C R /(R * T(I))) *\right.\right.\)
1 (1.D0-2.D0*R*T(I)/E1CR)/(C*E1CR)))
\(\operatorname{VFCN}(\mathrm{I})=\mathrm{VO} *(1 . \operatorname{DO}-\operatorname{EXP}(-\mathrm{RKICN} * \mathrm{R} * \mathrm{~T}(\mathrm{I}) * T(I) * \operatorname{EXP}(-E 1 C N /(R * T(I))) *\)
1
(1.DO-2.D0*R*T(I)/E1CN)/(C*E1CN)))
\(\operatorname{VFFM}(I)=V O *(1 . \operatorname{DO}-\operatorname{EXP}(-R K 1 F M * R * T(I) * T(I) * E X P(-E 1 F M /(R * T(I))) *\)
1 (1.0D0-2.D0*R*T(I)/E1FM)/(C*E1FM)))
ELSE
\(\operatorname{VFNT}(I)=V O *(1 . D 0-E X P(-R K 2 * R * T(I) * T(I) * E X P(-E 2 /(R * T(I)))\)
1 *(1.D0-2.DO*R*T(I)/E2)/(C*E2)))
\(\operatorname{VFCR}(\mathrm{I})=\mathrm{VO} *\left(1 . \operatorname{DO}-\operatorname{EXP}\left(-\mathrm{RK} 2 \mathrm{CR} * \mathrm{R}^{\star} \mathrm{T}(\mathrm{I}) * \mathrm{~T}(\mathrm{I}) \star \operatorname{EXP}\left(-E 2 C R /\left(\mathrm{R}^{\star} \mathrm{T}(\mathrm{I})\right)\right)\right.\right.\)
1 *(1.D0-2.D0*R*T(I)/E2CR)/(C*E2CR)))
\(\operatorname{VFCN}(I)=\mathrm{VO}^{\star}\left(1 . \operatorname{DO}-\operatorname{EXP}\left(-\mathrm{RK} 2 \mathrm{CN}^{\star} \mathrm{R}^{\star} \mathrm{T}(\mathrm{I}) \star \mathrm{T}(\mathrm{I}) \star \operatorname{EXP}\left(-\mathrm{E} 2 \mathrm{CN} /\left(\mathrm{R}^{\star} \mathrm{T}(\mathrm{I})\right)\right) \star\right.\right.\)
1 (1.D0-2.D0*R*T(I)/E2CN)/(C*E2CN)))
\(\operatorname{VFFM}(I)=V O *(1 . \operatorname{DO}-E X P(-R K 2 F M * R * T(I) * T(I) * E X P(-E 2 F M /(R * T(I))) *\)
1
(1.DO-2.D0*R*T(I)/E2FM)/(C*E2FM)))

ENDIF
280 CONTINUE
c
Calculate the rate at entire temperature range
DO \(290 \mathrm{I}=1, \mathrm{M}\)
IF ( \((\mathrm{T}(\mathrm{I})-273.16 \mathrm{D} 0) . \mathrm{LT} .450 . \mathrm{D} 0)\) THEN
\(\operatorname{VDNT}(I)=R K 1 * E X P(-E 1 / R / T(I)) *(V O-V(I)) / C\)
\(\operatorname{VDCR}(I)=R K 1 C R * E X P(-E 1 C R / R / T(I)) *(\operatorname{VO}-V(I)) / C\)
\(\operatorname{VDCN}(I)=R K 1 C N * E X P(-E 1 C N / R / T(I)) *(V O-V(I)) / C\)
\(\operatorname{VDFM}(I)=R K 1 F M * E X P(-E 1 E M / R / T(I)) *(V O-V(I)) / C\)
ELSE
\(\operatorname{VDNT}(I)=\operatorname{RK} 2 * \operatorname{EXP}(-E 2 / R / T(I)) *(\operatorname{VO}-V(I)) / C\)
\(\operatorname{VDCR}(I)=R K 2 C R * E X P(-E 2 C R / R / T(I)) *(V O-V(I)) / C\)
\(\operatorname{VDCN}(I)=R K 2 C N * E X P(-E 2 C N / R / T(I)) *(V O-V(I)) / C\)
\(\operatorname{VDEM}(I)=R K 2 F M^{*} \operatorname{EXP}(-E 2 F M / R / T(I)) *(V O-V(I)) / C\)
ENDIF
290 CONTINUE
Calculate the absolute average deviation ERROR
SEENT=0.DO
SEECR=0.D0
SEECN=0.DO
SEEFM=0.DO
DO \(450 \mathrm{I}=1, \mathrm{M}\)
\(\operatorname{SEENT}=\operatorname{SEENT}+(\mathrm{V}(\mathrm{I})-\mathrm{VFNT}(\mathrm{I})) * * 2\)
\(\operatorname{SEECR}=\operatorname{SEECR}+(\mathrm{V}(\mathrm{I})-\operatorname{VFCR}(\mathrm{I})) * * 2\)
\(\operatorname{SEECN}=\operatorname{SEECN}+(\mathrm{V}(\mathrm{I})-\operatorname{VFCN}(\mathrm{I})) * * 2\)
SEEFM=SEEFM+(V(I)-VFFM(I))**2
450 CONTINUE
SEENT=SQRT ( (SEENT) /(DBLE (M) -2.DO))
SEECR=SQRT ( (SEECR) / (DBLE (M)-2.D0))
\(\operatorname{SEECN}=\operatorname{SQRT}((S E E C N) /(D B L E(M)-2 . D O))\)
SEEFM=SQRT ( (SEEFM) / (DBLE (M) -2.DO) )
WRITE(*,*) SEENT, SEECR, SEECN,SEEFM
C
C
```

Printing results of $\mathrm{V} 1, \mathrm{~V} 2, \mathrm{~V} 1 \mathrm{~F}$ and V 2 F

```

DO \(500 \mathrm{I}=1, \mathrm{M}\)
\(T(I)=T(I)-273.16 D 0\)
500 CONTINUE
WRITE \((3,550)\)
550 FORMAT (//'Fitting results in the selected temperature range')
\(\operatorname{WRITE}(3,600)\)
```

    600 FORMAT (4X,'T',8X,'V', 8X,'VFNT',5X, 'VFCR', 6X,'VFCN', 6X,'VFFM',
        1 8X,'VD')
        DO 620 I=1,N
        WRITE(3,610) T(LN+I),V(LN+I),V1FNT(I),V1FCR(I),V1FCN(I),V1FFM(I),
    1 VD(LN+I)
    610 FORMAT (F7.2,6F10.6)
    620 CONTINUE

```
        DO \(640 \mathrm{I}=1, \mathrm{MN}\)
        \(\operatorname{WRITE}(3,630) T(L N+N+I), V(L N+N+I), \operatorname{V2FNT}(I), \operatorname{V2FCR}(I), \operatorname{V2FCN}(I)\),
        \(1 \quad \operatorname{V2FFM}(I), \mathrm{VD}(\mathrm{LN}+\mathrm{N}+\mathrm{I})\)
    630 FORMAT (F7.2, 6F10.6)
    640 CONTINUE
        WRITE \((3,650)\)
    650 FORMAT (//'Activation energies and pre-exponential factor for both
    1 reactions')
        WRITE \((3,660)\)
    660 FORMAT (30X, 'E1', 10X, 'K1', 10X, 'E2', 10X, 'K2')
    WRITE \((3,670)\) E1, RK1, E2, RK2
    670 FORMAT ('Integral method', 10X, 4F12.3)
    WRITE \((3,680)\) E1CR, RK1CR, E2CR, RK2CR
    680 FORMAT ('Coats-Redfern method', 5X, 4F12.3)
    WRITE \((3,690)\) E1CN, RK1CN, E2CN, RK2CN
    690 FORMAT ('Chen-Nuttall method', 6X,4F12.3)
    WRITE \((3,700)\) E1FM, RK1FM, E2FM, RK2FM
    700 FORMAT ('Friedman method',10X,4F12.3)
C
C
C
    Printing the results of \(Y 1, Y 2, Y 1 F\) and \(Y 2 F\)
    WRITE \((3,701)\)
    701 FORMAT (//'Experimental results Y1, Y2 and fitted results Y1F,Y2F')
    WRITE \((3,702)\)
    702 FORMAT (4X,'T', 5X, 'YNT', 5X, 'YFNT', 4X, 'YCR', 5X, 'YFCR', 4X, 'YCN', 5X,
    1 'YFCN',4X,'YFM',5X,'YFFM')
        DO \(704 \mathrm{I}=\mathrm{MN}, 1,-1\)
        WRITE(3,703) X2(I), Y2NT(I), Y2FNT(I),Y2CR(I),Y2FCR(I),Y2CN(I),
        1 Y2FCN(I),Y2FM(I),Y2FFM(I)
    703 FORMAT (F8.6.8F8.3)
    704 CONTINUE
        DO \(705 \mathrm{I}=\mathrm{N}, 1,-1\)
        WRITE(3,703) X1 (I), Y1NT (I), Y1FNT (I), Y1CR(I), Y1FCR(I), Y1CN(I),
        \(1 \quad \mathrm{Y} 1 \mathrm{FCN}(\mathrm{I}), \mathrm{Y} 1 \mathrm{FM}(\mathrm{I}), \mathrm{Y} 1 \mathrm{FFM}(\mathrm{I})\)
    705 CONTINUE
C
C
C
    710 FORMAT (//'Fitting results in the entire temperature range')
    WRITE \((3,720)\)
    720 FORMAT (4X, 'T', 7X, 'V', 7X, 'VFNT', 6X, 'VFCR', 6X, 'VFCN', 6X, 'VFFM')
        DO \(740 \mathrm{I}=1, \mathrm{M}\)
        \(\operatorname{WRITE}(3,730) T(I), V(I), \operatorname{VFNT}(I), \operatorname{VFCR}(I), \operatorname{VFCN}(I), \operatorname{VFFM}(I)\)
    730 FORMAT (F7.2,5F10.6)
    740 CONTINUE
C
C Printing the standard deviation
C
    WRITE \((3,742)\)
    742 FORMAT (//'standard deviation for each method above')
    WRITE \((3,746)\) SEENT, SEECR, SEECN, SEEFM
    746 FORMAT (17X, 4F10.6)
C
```

C Printing the rates in the entire temperature range
C
WRITE(3,750)
750 FORMAT(//'Fitting rate dV/dT in the entire temperature reange')
WRITE (3,760)
760 FORMAT(4X,'T',6X,'VD',7X,'VDNT',6X,'VDCR',6X, 'VDCN', 6X, 'VDFM')
DO }770\textrm{I}=1,\textrm{M
WRITE(3,730) T(I),VD(I),VDNT(I),VDCR(I),VDCN(I),VNFM(I)
7 7 0 CONTINUE
ENDFILE(UNIT = 3)
CLOSE(UNIT = 3)
STOP
END
SUBROUTINE GAUSS (A,N,NDR,NDC,X, RNORM,IREEOR)
IMPLICIT REAL*8(A-H,O-2)
DIMENSION A(NDR,NDC),X(N),B(50,51)
NM=N-1
NP=N+1
IREEOR=3
DO 20 I=1,N
DO 10 J=1,NP
B(I,J)=A(I,J)
CONTINUE
CONTINUE
DO . }70\textrm{K}=1,\textrm{NM
KP=K+1
BIG=ABS (B (K,K))
IPIVOT=K
DO 30 I=KP,N
AB=ABS (B.(I,K))
IF(AB.GT.BIG) THEN
BIG=AB
IPIVOT=I
ENDIF
CONTINUE
IE(IPIVOT.NE.K) THEN
DO 40 J=K,NP
TEMP=B(IPIVOT,J)
B(IPIVOT,J)=B(K,J)
B (K,J) =TEMP
CONTINUE
ENDIF
IF(B(K,K).EQ.O.DO) THEN
IERROR=2
RETURN
ENDIF
DO 60 I=KP,N
QUOT=B(I,K)/B(K,K)
B}(I,K)=0.D
DO 50 J=KP,NP
B(I,J) = B (I,J) -QUOT * B (K,J)
CONTINUE
CONTINUE
CONTINUE
IF(B(N,N).EQ.O.DO) THEN
IERROR=2
RETURN

```
```

    ENDIF
    X(N)=B(N,NP)/B(N,N)
    DO 90 I=NM,1,-1
        SUM=0. DO
        DO 80 J=I+1,N
            SUM=SUM+B(I,J)*X(J)
    80 CONTINUE
        X(I)=(B(I,NP)-SUM)/B (I,I)
    O CONTINUE
    RSQ=0.DO
    DO 110 I=1,N
        SUM=0.DO
        DO }100\textrm{J}=1,
        SUM=SUM+A(I,J) *X(J)
    100 CONTINUE
RSQ=RSQ+(A (I,NP)-SUM)**2
110. CONTINUE
RNORM=DSQRT (RSQ)
IERROR=1
RETURN
END
SUBROUTINE FLSQP(X,Y,M,N,A,VAR)
IMPLICIT REAL*8(A-H,O-Z)
DIMENSION X (M),U(51),Y(M),V(51),A(N),B(11), COEFF(10,11),SUMU (18)
NP=N+1
NM2=2* (N-1)
XMIN=X(1)
XMAX=X (1)
YMIN=Y(1)
YMAX=Y(1)
DO 10 K=2,M
XMIN=DMIN1 (XMIN,X(K))
XMAX=DMAX1 (XMAX,X(K))
YMIN=DMIN1 (YMIN,Y(K))
YMAX=DMAXI (YMAX,Y(K))
CONTINUE
XP=XMIN+XMAX
XM=XMAX-XMIN
YP=YMIN+YMAX
YM=YMAX-YMIN
DO 20 K=1,M
U(K)=(2.DO*X(K)-XP)/XM
V(K)=(2.DO*Y(K)-YP)/YM
CONTINUE
DO 30 L=1,NM2
SUMU (L)=0.DO
CONTINUE
DO 40 I=1,N
COEFF(I,NP)=0.DO
40 CONTINUE
DO 70 K=1,M
TERMU=U (K)
DO 50 L=1,NM2
SUMU (L) =SUMU (L) +TERMU
TERMU=TERMU*U (K)
CONTINUE
TERMV=V (K)
DO 60 I=1,N
COEFF}(I,NP)=\operatorname{COEFF}(I,NP)+TERMV
TERMV=TERMV*U(K)

```
```

6 0
7 0
CONTINUE
DO }90\mathrm{ I=1,N
DO }80\textrm{J}=1,
IF(I.EQ.1.AND.J.EQ.1) THEN
COEFF}(I,J)=
ELSE
COEFF (I,J)=SUMU (I+J-2)
ENDIF
CONTINUE
CONTINUE
CALL GAUSS (COEFF,N,10,11,B,RNORM,IERROR)
DO 110 I=1,N
IM=I-1
SUM=B (I)
IF(I.NE.N) THEN
DO 100 J=I +1,N
SUM=SUM+NOMIAL (IM,J-1)* (-XP/XM)** (J-I)*B (J)
CONTINUE
ENDIF
A(I) =YM* (2.DO/XM)**IM*SUM/2.D0
110 CONTINUE
A(1)=A(1)+YP/2.D0
SSUM=0.DO
DO 130 K=1,M
SUM=A(1)
TEMP=1.D0
DO 120 J=2,N
TEMP=TEMP*X(K)
SUM=SUM+A (J) *TEMP
CONTINUE
SSUM=SSUM+(Y(K)-SUM)**2
CONTINUE
VAR=SSUM/ (M-N)
RETURN
END
C
FUNCTION NOMIAL (I,J)
NOMIAL=1
IF(J.LE.I.OR.I.EQ.0) RETURN
DO 10 ICOUNT=1,I
NOMIAL=NOMIAL* (J-ICOUNT+1)/ICOUNT
CONTINUE
RETURN
END

```

2-Stage First Order Reaction Model Fitting Results

\section*{CANMET pitch \(25^{\circ} \mathrm{C} / \mathrm{min}, 800^{\circ} \mathrm{C}\) final temperature RUN\# CAN48}
\begin{tabular}{ccccccccr} 
Experimental results Y1, Y2 and fitted results Y1F, Y2F \\
T & YNT & YFNT & YCR & YFCR & YCN & YFCN & YFM & YFFM \\
.001177 & -10.718 & -10.575 & -10.939 & -10.794 & .412 & .556 & -1.242 & -.179 \\
.001213 & -10.836 & -10.882 & -11.049 & -11.095 & .297 & .251 & -.723 & -.349 \\
.001233 & -10.958 & -11.053 & -11.168 & -11.263 & .176 & .081 & -.299 & -.443 \\
.001251 & -11.104 & -11.208 & -11.310 & -11.414 & .031 & -.073 & -.195 & -.529 \\
.001267 & -11.268 & -11.344 & -11.471 & -11.548 & -.132 & -.208 & -.112 & -.604 \\
.001280 & -11.434 & -11.460 & -11.635 & -11.662 & -.297 & -.324 & -.089 & -.668 \\
.001293 & -11.570 & -11.572 & -11.768 & -11.771 & -.432 & -.435 & -.331 & -.730 \\
.001306 & -11.736 & -11.676 & -11.932 & -11.873 & -.598 & -.539 & -.239 & -.788 \\
.001320 & -11.884 & -11.800 & -12.078 & -11.994 & -.745 & -.661 & -.620 & -.856 \\
.001338 & -12.037 & -11.951 & -12.228 & -12.143 & -.898 & -.812 & -.907 & -.940 \\
.001356 & -12.170 & -12.107 & -12.358 & -12.296 & -1.029 & -.967 & -1.178 & -1.026 \\
.001371 & -12.267 & -12.240 & -12.453 & -12.426 & -1.126 & -1.099 & -1.393 & -1.099 \\
.001387 & -12.354 & -12.377 & -12.538 & -12.560 & -1.213 & -1.235 & -1.591 & -1.175 \\
.001407 & -12.455 & -12.545 & -12.636 & -12.725 & -1.313 & -1.402 & -1.735 & -1.267 \\
.001430 & -11.979 & -12.050 & -12.736 & -12.779 & -2.380 & -2.429 & -1.964 & -2.384 \\
.001455 & -12.100 & -12.114 & -12.839 & -12.832 & -2.489 & -2.487 & -2.063 & -2.437 \\
.001483 & -12.199 & -12.190 & -12.916 & -12.896 & -2.573 & -2.556 & -2.479 & -2.501 \\
.001513 & -12.307 & -12.269 & -13.004 & -12.962 & -2.668 & -2.628 & -2.506 & -2.567 \\
.001541 & -12.392 & -12.341 & -13.071 & -13.022 & -2.742 & -2.693 & -2.696 & -2.628 \\
.001573 & -12.485 & -12.426 & -13.144 & -13.093 & -2.822 & -2.770 & -2.834 & -2.699 \\
.00603 & -12.551 & -12.504 & -13.193 & -13.158 & -2.878 & -2.840 & -3.077 & -2.764 \\
.001670 & -12.705 & -12.680 & -13.312 & -13.306 & -3.011 & -3.000 & -3.150 & -2.912 \\
.001742 & -12.853 & -12.872 & -13.425 & -13.466 & -3.138 & -3.173 & -3.357 & -3.073 \\
.001222 & -13.018 & -13.081 & -13.558 & -13.642 & -3.285 & -3.363 & -3.454 & -3.249 \\
.001909 & -13.231 & -13.311 & -13.738 & -13.834 & -3.480 & -3.571 & -3.489 & -3.441 \\
.002005 & -13.487 & -13.56 .3 & -13.963 & -14.045 & -3.719 & -3.799 & -3.620 & -3.653 \\
.002111 & -13.792 & -13.843 & -144.238 & -14.279 & -4.009 & -4.052 & -3.810 & -3.887 \\
.002229 & -14.299 & -14.153 & -14.716 & -14.538 & -4.502 & -4.333 & -3.843 & -4.147
\end{tabular}

Fitting results in the entire temperature range
\begin{tabular}{rrrrrr} 
T & \multicolumn{1}{c}{ V } & VFNT & VFCR & VFCN & \multicolumn{1}{l}{ VFFM } \\
50.22 & .040000 & .155567 & .143757 & .173164 & .254039 \\
100.32 & .060000 & .585928 & .446609 & .583486 & .789804 \\
135.40 & .500000 & 1.231859 & .842096 & 1.151879 & 1.488240 \\
153.77 & 1.020000 & 1.733712 & 1.126916 & 1.574572 & 1.990044 \\
175.47 & 2.170000 & 2.504867 & 1.541807 & 2.204314 & 2.719006 \\
200.52 & 3.860000 & 3.671561 & 2.135001 & 3.126396 & 3.756867 \\
225.60 & 5.570000 & 5.172992 & 2.858170 & 4.276503 & 5.014856 \\
250.65 & 7.590000 & 7.032889 & 3.711836 & 5.662237 & 6.489252 \\
275.70 & 9.830000 & 9.265615 & 4.693808 & 7.285796 & 8.170753 \\
300.75 & 12.080000 & 11.867032 & 5.795725 & 9.138123 & 10.038886 \\
325.80 & 14.480000 & 14.815040 & 7.003953 & 11.200199 & 12.064187 \\
350.85 & 17.330000 & 18.070076 & 8.299930 & 13.443768 & 14.209525 \\
362.55 & 18.700000 & 19.680585 & 8.928980 & 14.543901 & 15.240696 \\
375.90 & 20.650000 & 21.577142 & 9.660695 & 15.832638 & 16.431792 \\
387.60 & 22.550000 & 23.282974 & 10.311053 & 16.986420 & 17.482892 \\
400.95 & 25.090000 & 25.269189 & 11.059527 & 18.324429 & 18.683748 \\
414.32 & 27.590000 & 27.289795 & 11.811514 & 19.680511 & 19.880750 \\
426.00 & 30.750000 & 29.071447 & 12.466628 & 20.872552 & 20.915811 \\
437.70 & 34.040000 & 30.863205 & 13.117667 & 22.068282 & 21.937321 \\
447.72 & 37.350000 & 32.397342 & 13.668495 & 23.089827 & 22.796086 \\
456.07 & 40.310000 & 41.062072 & 35.863390 & 40.231548 & 52.110546 \\
464.42 & 43.670000 & 45.481618 & 39.961255 & 44.595503 & 54.815420 \\
474.45 & 48.260000 & 50.810762 & 45.020628 & 49.878827 & 57.945213 \\
484.45 & 53.520000 & 55.999556 & 50.100606 & 55.050838 & 60.904851 \\
492.80 & 58.430000 & 60.121639 & 54.273443 & 59.184363 & 63.232246
\end{tabular}
500.0063 .51000063 .44988657 .75513562 .54198565 .122477 \(507.8567 .42000066 .77902161 .365837 \quad 65.92312367 .051542\) 516.2071 .52000069 .92255864 .92892669 .14238668 .945288 \(526.2274 .86000073 .101714 \quad 68.74411372 .43378370 .996004\) 537.9077 .16000075 .97179972 .46875975 .45049473 .078863 551.2778 .62000078 .23058975 .72999477 .87482275 .066625 \(\begin{array}{lllll}576.32 & 79.700000 & 80.249327 & 79.254115 & 80.12422677 .755774\end{array}\) 601.3780 .03000080 .75951680 .50686380 .73331279 .351849 649.8080 .29000080 .83975480 .83627480 .83957080 .575569 699.9080 .52000080 .84000080 .83999880 .84000080 .812883 750.0080 .72000080 .84000080 .84000080 .84000080 .838394 800.1280 .84000080 .84000080 .84000080 .84000080 .839947
\begin{tabular}{rccccc} 
Fitting & rate dV/dT & in the entire temperature range \\
T & VD & VDNT & VDCR & VDCN & VDFM \\
50.22 & .000399 & .005196 & .004289 & .005420 & .007594 \\
100.32 & .012543 & .015487 & .010694 & .014563 & .018972 \\
135.40 & .028307 & .028220 & .017650 & .025052 & .031347 \\
153.77 & .052995 & .036999 & .022115 & .031991 & .039298 \\
175.47 & .067465 & .049144 & .027976 & .041305 & .049740 \\
200.52 & .068182 & .065595 & .035494 & .053529 & .063140 \\
225.60 & .080639 & .084826 & .043850 & .067409 & .078044 \\
250.65 & .089421 & .106264 & .052710 & .082443 & .093857 \\
275.70 & .089820 & .129585 & .061912 & .098369 & .110288 \\
300.75 & .095808 & .154713 & .071431 & .115130 & .127292 \\
325.80 & .113772 & .180904 & .080943 & .132188 & .144295 \\
350.85 & .117094 & .206557 & .089793 & .148424 & .160123 \\
362.55 & .146067 & .218426 & .093755 & .155798 & .167211 \\
375.90 & .162393 & .230390 & .097522 & .163007 & .173957 \\
387.60 & .190262 & .239737 & .100294 & .168466 & .178925 \\
400.95 & .186986 & .248123 & .102472 & .173057 & .182836 \\
414.32 & .270548 & .255705 & .104300 & .177062 & .186123 \\
426.00 & .281197 & .256427 & .103506 & .176486 & .184728 \\
437.70 & .330339 & .254908 & .101856 & .174410 & .181803 \\
447.72 & .354491 & .249397 & .098818 & .169807 & .176398 \\
456.07 & .402395 & .558800 & .454728 & .539926 & .539964 \\
464.42 & .457627 & .585505 & .475201 & .565276 & .533028 \\
474.45 & .526000 & .599901 & .485379 & .578632 & .509290 \\
484.45 & .588024 & .585343 & .472178 & .564076 & .464358 \\
492.80 & .705556 & .543249 & .437149 & .523123 & .407799 \\
500.00 & .498089 & .466302 & .374454 & .448745 & .334071 \\
507.85 & .491018 & .403712 & .323475 & .388251 & .275146 \\
516.20 & .333333 & .314930 & .251757 & .302658 & .203760 \\
526.22 & .196918 & .231569 & .184618 & .222364 & .140963 \\
537.90 & .109200 & .166329 & .132199 & .159569 & .094482 \\
551.27 & .043114 & .119123 & .094357 & .114163 & .062666 \\
576.32 & .013174 & .083148 & .065462 & .079540 & .038128 \\
601.37 & .005369 & .078905 & .061765 & .075349 & .031788 \\
649.80 & .004591 & .089660 & .069471 & .085356 & .028688 \\
699.90 & .003992 & .084197 & .064620 & .079924 & .021745 \\
750.00 & .002394 & .048626 & .037002 & .046039 & .010352 \\
800.12 & .002394 & .000000 & .000000 & .000000 & .000000
\end{tabular}

CANMET pitch \(50^{\circ} \mathrm{C} / \mathrm{min}, 800^{\circ} \mathrm{C}\) final temperature RUN\#33
Experimental results \(\mathrm{Y} 1, \mathrm{Y} 2\) and fitted results \(\mathrm{Y} 1 \mathrm{~F}, \mathrm{Y} 2 \mathrm{~F}\)
\begin{tabular}{crrrrrrrr} 
T & YNT & YFNT & YCR & YFCR & YCN & YFCN & YFM & YFFM \\
.001169 & -10.078 & -10.036 & -10.328 & -10.283 & .939 & .983 & -.645 & .522 \\
.001207 & -10.197 & -10.328 & -10.437 & -10.568 & .824 & .693 & -.131 & .277 \\
.001223 & -10.306 & -10.455 & -10.543 & -10.692 & .716 & .567 & .429 & .171 \\
.001240 & -10.450 & -10.587 & -10.683 & -10.820 & .574 & .437 & .488 & .061 \\
.001265 & -10.725 & -10.776 & -10.953 & -11.005 & .300 & .248 & .437 & -.098
\end{tabular}
\begin{tabular}{lllllllll} 
\\
.001290 & -11.035 & -10.973 & -11.258 & -11.197 & -.008 & .053 & .280 & -.263 \\
.001309 & -11.231 & -11.119 & -11.450 & -11.340 & -.203 & -.092 & .014 & -.385 \\
.001328 & -11.406 & -11.269 & -11.622 & -11.487 & -.377 & -.241 & -.262 & -.511 \\
.001344 & -11.526 & -11.393 & -11.739 & -11.607 & -.496 & -.363 & -.523 & -.614 \\
.001360 & -11.642 & -11.519 & -11.852 & -11.730 & -.611 & -.488 & -.688 & -.720 \\
.001369 & -11.695 & -11.583 & -11.904 & -11.793 & -.664 & -.552 & -.824 & -.773 \\
.001386 & -11.800 & -11.714 & -12.006 & -11.920 & -.768 & -.682 & -.946 & -.883 \\
.001407 & -11.916 & -11.882 & -12.118 & -12.085 & -.883 & -.849 & -1.186 & -1.024 \\
.001430 & -12.030 & -12.056 & -12.229 & -12.254 & -.995 & -1.021 & -1.329 & -1.169 \\
.001453 & -12.137 & -12.235 & -12.332 & -12.429 & -1.101 & -1.199 & -1.511 & -1.320 \\
.001486 & -12.249 & -12.496 & -12.440 & -12.684 & -1.212 & -1.458 & -1.868 & -1.539 \\
.001542 & -11.846 & -11.792 & -12.571 & -12.495 & -2.271 & -2.843 & -2.210 & -2.429 \\
.001603 & -11.993 & -11.945 & -12.678 & -12.623 & -2.393 & -3.038 & -2.502 & -2.517 \\
.001669 & -12.123 & -12.109 & -12.770 & -12.761 & -2.499 & -3.251 & -2.748 & -2.613 \\
.001746 & -12.277 & -12.305 & -12.885 & -12.924 & -2.629 & -3.509 & -2.842 & -2.726 \\
.001824 & -12.453 & -12.501 & -13.026 & -13.089 & -2.784 & -3.772 & -2.870 & -2.840 \\
.001910 & -12.645 & -12.716 & -13.184 & -13.269 & -2.956 & -4.039 & -3.016 & -2.964 \\
.002003 & -12.885 & -12.951 & -13.391 & -13.466 & -3.178 & -4.397 & -3.094 & -3.100 \\
.002117 & -13.206 & -13.237 & -13.677 & -13.704 & -3.481 & -4.806 & -3.260 & -3.265 \\
.002232 & -13.656 & -13.527 & -14.096 & -13.948 & -3.916 & -5.734 & -3.344 & -3.434
\end{tabular}

Fitting results in the entire temperature range
\begin{tabular}{|c|c|c|c|c|c|}
\hline & & & VFCR & VFC & VFI \\
\hline 50.10 & 320000 & 159 & 13889 & . 171486 & 591272 \\
\hline 99. & 310000 & 556368 & . 404340 & . 539215 & 3 \\
\hline 125.80 & 400000 & 71510 & . 650541 & . 898298 & 1.709785 \\
\hline 152.50 & 1.060000 & 1.585018 & . 98697 & 1.405 & 2.241225 \\
\hline 174.80 & 2.010000 & 2.282442 & . 34559 & 1.96170 & 60 \\
\hline 199.25 & 3.37000 & . 27105 & . 82600 & 2.7248 & 253745 \\
\hline 226.00 & 4.960000 & . 6504 & 459542 & 3.7563 & 23 \\
\hline 250.4 & 6.640000 & 6.202251 & 3.136557 & 4.883752 & 5 \\
\hline 274.95 & 8.420000 & 8.042902 & 3.904061 & 6.188210 & 4.678267 \\
\hline 299.45 & 10.430000 & 10.164 & 4.752172 & 7.65821 & 11 \\
\hline 26.15 & 12.620000 & 12.77613 & 5.753620 & 9.4295 & 2 \\
\hline & 14.76000 & 15.41611 & 6.726081 & 11.185 & 0 \\
\hline 375.15 & 17.410000 & 18.248549 & 7.729867 & 13.03 & 9 \\
\hline 399.60 & 20.810000 & 21.219655 & 8.740720 & 14.948457 & 9 \\
\hline 415.20 & 23.700000 & 23.166671 & 9.379280 & 16.184024 & 9 \\
\hline 426.35 & 26.510000 & 24.573239 & 9.828113 & 68 & 3 \\
\hline 437.45 & 29.700000 & 25.979937 & 10.265765 & 17.945 & 5 \\
\hline 448.60 & 33.180000 & 27.394181 & 10.693524 & 18.820292 & 09 \\
\hline 457.50 & 36.470000 & 39.517420 & 33.78151 & 38.494790 & 35.907049 \\
\hline 461.95 & 38.2000 & 41.620283 & 35.681101 & 40.558965 & 37.622476 \\
\hline 5 & 42.010000 & 45.875263 & 39.583898 & 44.747626 & 41.113278 \\
\hline 479.75 & 46.100000 & 50.133976 & 43.581692 & 48.958655 & 44.651810 \\
\hline 490.90 & 52.050000 & 55.353100 & 48.633913 & 54.150658 & 49.087966 \\
\hline 502.05 & 58.550000 & 60.295076 & 53.615079 & 59.107162 & 53.445751 \\
\hline 517 & 67.700000 & 66.450372 & 60.191753 & 65.355186 & 59.225680 \\
\hline 533.20 & 74.020000 & 71.494238 & 66.051103 & 70.566098 & 64.475386 \\
\hline 544.35 & 76.480000 & 74.320833 & 69.637644 & 73.542262 & 67.790289 \\
\hline 555.45 & 77.950000 & 76.495891 & 72.644783 & 75.875617 & 70.678095 \\
\hline 582.15 & 79.280000 & 79.517704 & 77.569040 & 79.235903 & 75.862118 \\
\hline 624.45 & 79.950000 & 80.709446 & 80.389877 & 80.675051 & 79.724209 \\
\hline 689.05 & 80.150000 & 80.789924 & 80.787785 & 80.789835 & 80.758907 \\
\hline 80 & 80.790000 & 80.79000 & 80. & 80.790000 & 80.789999 \\
\hline
\end{tabular}

VFFM
.591272
1.242183
1.709785
2.241225
2.718060
3.253745
3.823297

293595
4.934511
5.013092
4.845490
4.388759
. 578859
2.197203
1.445455
. 570909
5. 907049
41.113278
49.087966
53.445751
59.225680
67.790289
70.678095
75.862118
80.758907

Fitting rate \(\mathrm{dV} / \mathrm{d} T\) in the entire temperature range
\begin{tabular}{|c|c|c|c|c|c|}
\hline T & VD & VDNT & VDCR & VDCN & VDFM \\
\hline 50.10 & -. 000204 & . 005149 & . 004026 & . 005205 & . 01481 \\
\hline 99.10 & . 003371 & . 014337 & . 009492 & . 013161 & . 02681 \\
\hline 125.80 & . 024719 & . 022507 & . 013844 & . 019799 & . 03479 \\
\hline 152.50 & . 042601 & . 033148 & . 019120 & . 028094 & . 04 \\
\hline 174.80 & . 055624 & . 043952 & . 024167 & . 03623 & . 05 \\
\hline
\end{tabular}
\begin{tabular}{llllll} 
& & .057 \\
199.25 & .059439 & .057756 & .030292 & .046329 & .059121 \\
226.00 & .068712 & .075246 & .037675 & .058758 & .068311 \\
250.45 & .072653 & .093094 & .044861 & .071103 & .076549 \\
274.95 & .082041 & .112617 & .052407 & .084293 & .084603 \\
299.45 & .082022 & .133237 & .060054 & .097901 & .092158 \\
326.15 & .087347 & .156988 & .068543 & .113247 & .100004 \\
350.65 & .108163 & .179303 & .076215 & .127353 & .106566 \\
375.15 & .139059 & .200430 & .083110 & .140330 & .111726 \\
399.60 & .185256 & .218394 & .088506 & .150892 & .114728 \\
415.20 & .252018 & .226248 & .090434 & .155076 & .114691 \\
426.35 & .287387 & .228010 & .090280 & .155429 & .112787 \\
437.45 & .312108 & .227007 & .089065 & .153929 & .109668 \\
448.60 & .369663 & .223431 & .086886 & .150727 & .105486 \\
457.50 & .388764 & .532799 & .421759 & .510357 & .408986 \\
461.95 & .428090 & .545977 & .431532 & .522721 & .414753 \\
470.85 & .459551 & .563999 & .444443 & .539453 & .419756 \\
479.75 & .533632 & .570667 & .448386 & .545314 & .416309 \\
490.90 & .582960 & .549467 & .430191 & .524450 & .391181 \\
502.05 & .588424 & .492026 & .383888 & .469098 & .342084 \\
517.60 & .405128 & .352553 & .273787 & .335617 & .237411 \\
533.20 & .220628 & .220429 & .170413 & .209535 & .143936 \\
544.35 & .132432 & .160001 & .123312 & .151940 & .102276 \\
555.45 & .049813 & .119715 & .091986 & .113573 & .074962 \\
582.15 & .015839 & .085249 & .065050 & .080693 & .050908 \\
624.45 & .003096 & .072702 & .054917 & .068592 & .040506 \\
689.05 & .005768 & .098935 & .073711 & .092926 & .050168 \\
800.00 & .005768 & .000000 & .000000 & .000000 & .000000
\end{tabular}

\section*{CANMET pitch \(100^{\circ} \mathrm{C} / \mathrm{min}, 800^{\circ} \mathrm{C}\) final temperature RUN\# CAN41}
\begin{tabular}{|c|c|c|c|c|c|c|c|c|}
\hline \multicolumn{2}{|l|}{\multirow[t]{2}{*}{erimental}} & \multicolumn{2}{|r|}{Y2 and} & \multicolumn{3}{|r|}{Y1F, Y2F} & & \\
\hline & & YFNT & YCR & CR & YCN & YFCN & YFM & YFEM \\
\hline . 001215 & -9.9 & -10.10 & -10.192 & -10.319 & 1.162 & 1.036 & 1.132 & 1.357 \\
\hline . 001227 & -10.127 & -10.209 & -10.335 & -10.417 & 1.019 & . 937 & 1.252 & 1.207 \\
\hline . 001239 & -10.289 & -10.311 & -10.495 & -10.517 & . 857 & . 834 & 1.184 & 1.052 \\
\hline . 001251 & -10.449 & -10.416 & -10.653 & -10.620 & . 697 & . 730 & 1.023 & . 893 \\
\hline . 001263 & -10.600 & -10.523 & -10.801 & -10.725 & . 548 & . 624 & . 815 & 732 \\
\hline . 001276 & -10.732 & -10.632 & -10.931 & -10.832 & . 416 & . 51 & 567 & 568 \\
\hline . 001295 & -10.904 & -10.800 & -11.100 & -10.996 & . 245 & . 349 & . 260 & 315 \\
\hline . 001315 & -11.045 & -10.973 & -11.238 & -11.166 & . 105 & . 177 & -. 064 & . 054 \\
\hline . 001342 & -11.204 & -11.210 & -11.392 & -11.398 & -. 053 & -. 059 & . 377 & . 304 \\
\hline . 001378 & -11.375 & -11.523 & -11.558 & -11.705 & -. 222 & 370 & 5 & 76 \\
\hline . 001424 & -11.148 & -11.131 & -11.716 & -11.686 & -1.195 & -1.903 & -1.067 & -1.216 \\
\hline . 001482 & -11.340 & -11.317 & -11.879 & -11.853 & -1.370 & -2.145 & -1.370 & -1.367 \\
\hline . 001544 & -11.521 & -11.519 & -12.032 & -12.033 & -1.535 & -2.371 & -1.625 & -1.531 \\
\hline . 001602 & -11.693 & -11.706 & -12.181 & -12.201 & -1.695 & -2.661 & -1.754 & -1.684 \\
\hline . 001664 & -11.888 & -11.908 & -12.352 & -12.381 & -1.877 & -2.934 & -1.879 & -1.848 \\
\hline . 001743 & -12.146 & -12.164 & -12.583 & -12.609 & -2.121 & -3.288 & -2.041 & -2.056 \\
\hline . 001817 & -12.383 & -12.404 & -12.799 & -12.824 & -2.348 & -3.626 & -2.280 & -2.251 \\
\hline . 001912 & -12.691 & -12.711 & -13.081 & -13.099 & -2.643 & -4.065 & -2.521 & -2.501 \\
\hline 01 & -13.050 & -12.999 & -13.419 & -13.357 & -2.992 & -6.243 & -2.655 & -2.736 \\
\hline
\end{tabular}

Fitting results in the entire temperature range
\begin{tabular}{rrrll}
T & \multicolumn{1}{l}{V} & \multicolumn{1}{l}{ VFNT } & VFCR & \multicolumn{1}{l}{ VFCN } \\
51.20 & .090000 & .037815 & .037196 & .041788 \\
148.60 & .250000 & .590487 & .449749 & .571244 \\
175.80 & .570000 & 1.038072 & .750431 & .977237 \\
199.20 & 1.250000 & 1.603755 & 1.113857 & 1.478432 \\
226.50 & 2.410000 & 2.532160 & 1.686924 & 2.283696 \\
249.80 & 3.670000 & 3.599821 & 2.323655 & 3.192784 \\
277.10 & 5.330000 & 5.226674 & 3.265426 & 4.555680 \\
300.50 & 7.100000 & 6.980181 & 4.255858 & 6.004666
\end{tabular}

\footnotetext{
VFFM
.089455
.885730
1.416064
2.031820
2.965834 3.966103 5.392814 6.840918
}
\begin{tabular}{|c|c|c|c|c|c|}
\hline & & & & & \\
\hline & & & & & \\
\hline & & & & & \\
\hline & 18.310000 & & & & \\
\hline 29.00 & 22 & & & & \\
\hline & & & & & \\
\hline & 32 & & & 31.302714 & \\
\hline & & & & & \\
\hline & & & & 43.836819 & \\
\hline & 47.510000 & 50.417374 & & & \\
\hline & 1. & & & & \\
\hline & & & & 56.655377 & \\
\hline & 61 & 61 & & 60.037606 & \\
\hline & & & & 63.197157 & \\
\hline & 69 & & & & \\
\hline & & & & & \\
\hline & 77.000000 & & & & \\
\hline & & & & 239 & \\
\hline & 78 & & & & \\
\hline . 60 & 78 & & & 9 & \\
\hline & & & & & \\
\hline & & & & & \\
\hline
\end{tabular}
\begin{tabular}{rccccc} 
Fitting & rate \(d V / d T\) & in the entire temperature range \\
T & VD & VDNT & VDCR & VDCN & VDFM \\
51.20 & .001643 & .001454 & .001318 & .001541 & .002970 \\
148.60 & .011765 & .014541 & .010325 & .013569 & .019344 \\
175.80 & .029060 & .023057 & .015584 & .020970 & .028129 \\
199.20 & .042491 & .032670 & .021262 & .029134 & .037294 \\
226.50 & .054077 & .046802 & .029274 & .040884 & .049829 \\
249.80 & .060806 & .061441 & .037273 & .052823 & .061992 \\
277.10 & .075641 & .081694 & .047974 & .069056 & .077848 \\
300.50 & .093750 & .101366 & .058032 & .084558 & .092374 \\
327.70 & .106410 & .126246 & .070346 & .103839 & .109705 \\
351.10 & .116239 & .148975 & .081254 & .121176 & .124680 \\
374.50 & .126838 & .172396 & .092180 & .138785 & .139336 \\
401.70 & .154945 & .199582 & .104463 & .158895 & .155372 \\
429.00 & .195299 & .223821 & .114859 & .176369 & .168297 \\
452.40 & .260513 & .372701 & .304524 & .360559 & .240177 \\
471.90 & .323226 & .460019 & .373598 & .444225 & .347618 \\
487.40 & .394872 & .521182 & .421329 & .502595 & .444381 \\
499.10 & .486325 & .551505 & .444351 & .531302 & .513459 \\
510.80 & .560256 & .553150 & .444231 & .532368 & .560853 \\
518.60 & .619231 & .532058 & .426392 & .511745 & .570239 \\
526.40 & .628205 & .487780 & .390098 & .468866 & .552005 \\
534.20 & .578205 & .424184 & .338550 & .407489 & .506336 \\
542.00 & .461039 & .350260 & .278993 & .336273 & .440551 \\
549.70 & .298718 & .282713 & .224756 & .271267 & .374077 \\
565.30 & .136923 & .177523 & .140595 & .170142 & .259549 \\
584.80 & .046886 & .103931 & .081937 & .099474 & .171268 \\
612.10 & .013882 & .062954 & .049333 & .060145 & .121581 \\
651.00 & .002964 & .044749 & .034788 & .042650 & .106607 \\
701.60 & .003550 & .050080 & .038566 & .047596 & .152880 \\
752.30 & .003212 & .035344 & .026987 & .033506 & .134973 \\
799.00 & .003212 & .000000 & .000000 & .000000 & .000000
\end{tabular}

\section*{CANMET pitch \(150^{\circ} \mathrm{C} / \mathrm{min}, 800^{\circ} \mathrm{C}\) final temperature RUN\# CAN58}
\begin{tabular}{|c|c|c|c|c|c|c|c|c|}
\hline \multicolumn{2}{|l|}{Experimenta} & \multicolumn{2}{|l|}{Y1, Y2} & \multicolumn{3}{|r|}{Y1F, Y2F} & \multirow[b]{2}{*}{YFM} & \multirow[b]{2}{*}{YFEM} \\
\hline T & YNT & YFNT & YCR & YFCR & YCN & YFCN & & \\
\hline . 001129 & -9.519 & -9.531 & -9.685 & -9.696 & 1.934 & 1.922 & 1.899 & 52 \\
\hline 001152 & -9.801 & -9.796 & -9.963 & -9.958 & 1.652 & 1.657 & 1.496 & 1.600 \\
\hline
\end{tabular}
\begin{tabular}{rrrrrrrrr}
.001175 & -10.008 & -10.073 & -10.167 & -10.231 & 1.446 & 1.382 & 1.088 & 1.337 \\
.001192 & -10.245 & -10.261 & -10.401 & -10.417 & 1.210 & 1.194 & 1.356 & 1.158 \\
.001209 & -10.483 & -10.459 & -10.637 & -10.613 & .973 & .997 & 1.113 & .969 \\
.001226 & -10.704 & -10.658 & -10.855 & -10.810 & .752 & .798 & .853 & .779 \\
.001244 & -10.895 & -10.866 & -11.044 & -11.015 & .562 & .591 & .510 & .582 \\
.001262 & -11.155 & -11.079 & -11.302 & -11.226 & .303 & .379 & .536 & .379 \\
.001291 & -11.380 & -11.411 & -11.523 & -11.554 & .078 & .047 & -.201 & .062 \\
.001321 & -11.702 & -11.759 & -11.841 & -11.898 & -.243 & -.300 & -.202 & -.269 \\
.001363 & -11.895 & -12.082 & -12.198 & -12.383 & -1.225 & -2.063 & -.728 & -.731 \\
.001419 & -12.285 & -12.398 & -12.574 & -12.688 & -1.610 & -3.575 & -1.284 & -1.089 \\
.001480 & -13.148 & -12.742 & -13.424 & -13.021 & -2.468 & -5.479 & -1.244 & -1.479 \\
.001753 & -14.167 & -14.274 & -14.395 & -14.500 & -3.470 & -10.911 & -3.259 & -3.216
\end{tabular}
\begin{tabular}{rrrrrrr} 
Fitting & results in the entire temperature range & \\
\(T\) & \multicolumn{1}{c}{ V } & VFNT & VFCR & VFCN & \multicolumn{1}{l}{ VFFM } \\
51.80 & .060000 & .000151 & .000154 & .000159 & .000068 \\
203.75 & .000000 & .074681 & .063126 & .073340 & .070851 \\
297.20 & .780000 & .701829 & .55518 .4 & .672535 & .866054 \\
402.35 & 2.850000 & 4.237655 & 3.194776 & 3.986740 & 6.378226 \\
431.60 & 7.050000 & 6.330544 & 4.732698 & 5.934912 & 9.894506 \\
460.70 & 10.840000 & 9.115970 & 8.027129 & 8.983551 & 9.795881 \\
484.10 & 15.870000 & 15.080683 & 13.282723 & 14.852779 & 15.812360 \\
501.65 & 21.770000 & 21.204772 & 18.726693 & 20.883108 & 21.863492 \\
519.20 & 27.110000 & 28.796499 & 25.568963 & 28.370146 & 29.251362 \\
530.90 & 33.840000 & 34.576969 & 30.867080 & 34.082948 & 34.821159 \\
542.60 & 39.520000 & 40.770361 & 36.649493 & 40.218947 & 40.756949 \\
554.15 & 46.400000 & 47.076315 & 42.676873 & 46.486832 & 46.786606 \\
566.00 & 53.900000 & 53.464116 & 48.967499 & 52.863180 & 52.903960 \\
577.55 & 60.980000 & 59.309409 & 54.942318 & 58.729828 & 58.537902 \\
595.10 & 66.750000 & 66.854625 & 63.119819 & 66.368983 & 65.934092 \\
612.65 & 72.410000 & 72.240913 & 69.523827 & 71.899768 & 71.411071 \\
624.35 & 75.110000 & 74.582095 & 72.607495 & 74.342314 & 73.913015 \\
653.60 & 75.950000 & 77.155327 & 76.597077 & 77.095814 & 76.930663 \\
723.65 & 76.820000 & 77.589951 & 77.589461 & 77.589928 & 77.589731 \\
799.55 & 77.590000 & 77.590000 & 77.590000 & 77.590000 & 77.590000
\end{tabular}
\begin{tabular}{rccccc} 
Fitting & rate dV/dT & in the entire temperature range \\
\(T\) & VD & VDNT & VDCR & VDCN & VDFM \\
51.80 & -.000395 & .000009 & .000009 & .000009 & .000005 \\
203.75 & .008347 & .002221 & .001826 & .002159 & .002333 \\
297.20 & .019686 & .015110 & .011640 & .014342 & .020546 \\
402.35 & .143590 & .067985 & .049723 & .063308 & .113490 \\
431.60 & .130241 & .090575 & .065475 & .083982 & .158347 \\
460.70 & .214957 & .206107 & .178311 & .202428 & .213433 \\
484.10 & .336182 & .310924 & .267476 & .304952 & .314499 \\
501.65 & .304274 & .398138 & .341129 & .390105 & .396048 \\
519.20 & .575214 & .501982 & .428456 & .491390 & .491444 \\
530.90 & .485470 & .538596 & .458575 & .526912 & .521913 \\
542.60 & .595671 & .576667 & .489815 & .563825 & .553269 \\
554.15 & .632911 & .576449 & .488506 & .563295 & .547803 \\
566.00 & .612987 & .533933 & .451440 & .521455 & .502590 \\
577.55 & .328775 & .451831 & .381193 & .441036 & .421484 \\
595.10 & .322507 & .388681 & .326871 & .379098 & .357799 \\
612.65 & .230769 & .242157 & .203026 & .236009 & .220097 \\
624.35 & .028718 & .137569 & .115110 & .134011 & .124013 \\
653.60 & .012420 & .136916 & .114024 & .133220 & .121026 \\
723.65 & .010145 & .155215 & .127955 & .150649 & .131517 \\
799.55 & .010145 & .000000 & .000000 & .000000 & .000000
\end{tabular}

\section*{Syncrude pitch \(25^{\circ} \mathrm{C} / \mathrm{min}, 800^{\circ} \mathrm{C}\) final temperature RUN\# Syn43}

\footnotetext{
Experimental results Y1,Y2 and fitted results Y1F, Y2F
}
\begin{tabular}{ccccc}
T & YNT & YFNT & YCR & YFCR \\
.001206 & -10.639 & -10.787 & -10.867 & -11.014 \\
.001220 & -10.774 & -10.907 & -10.999 & -11.132 \\
.001233 & -10.937 & -11.009 & -11.159 & -11.232 \\
.001243 & -11.111 & -11.093 & -11.331 & -11.313 \\
.001251 & -11.227 & -11.156 & -11.446 & -11.375 \\
.001262 & -11.350 & -11.242 & -11.566 & -11.459 \\
.001275 & -11.487 & -11.351 & -11.701 & -11.566 \\
.001294 & -11.647 & -11.509 & -11.857 & -11.720 \\
.001314 & -11.773 & -11.671 & -11.980 & -11.878 \\
.001335 & -11.873 & -11.837 & -12.077 & -12.041 \\
.001359 & -11.964 & -12.035 & -12.164 & -12.234 \\
.001381 & -12.027 & -12.213 & -12.223 & -12.409 \\
.001430 & -11.847 & -11.692 & -12.320 & -12.146 \\
.001484 & -11.970 & -11.889 & -12.422 & -12.330 \\
.001541 & -12.111 & -12.102 & -12.542 & -12.530 \\
.001603 & -12.292 & -12.331 & -12.703 & -12.745 \\
.001670 & -12.504 & -12.580 & -12.894 & -12.978 \\
.001743 & -12.765 & -12.850 & -13.136 & -13.232 \\
.001822 & -13.063 & -13.145 & -13.414 & -13.509 \\
.001910 & -13.399 & -13.469 & -13.731 & -13.812 \\
.002005 & -13.761 & -13.824 & -14.074 & -14.146 \\
.002112 & -14.233 & -14.218 & -14.528 & -14.516 \\
.002230 & -14.714 & -14.655 & -14.991 & -14.926 \\
.002361 & -15.242 & -15.144 & -15.502 & -15.385
\end{tabular}
\begin{tabular}{rrrr} 
YCN & YFCN & \multicolumn{1}{l}{ YFM } & \multicolumn{1}{l}{ YFFM } \\
.435 & .287 & -.368 & .449 \\
.301 & .168 & .155 & .269 \\
.139 & .067 & .321 & .116 \\
-.034 & -.016 & .437 & -.009 \\
-.150 & -.079 & .282 & -.104 \\
-.272 & -.164 & -.057 & -.233 \\
-.409 & -.273 & -.294 & -.397 \\
-.567 & -.429 & -.626 & -.633 \\
-.692 & -.590 & -.951 & -.876 \\
-.791 & -.755 & -1.242 & -1.126 \\
-.881 & -.951 & -1.529 & -1.422 \\
-.943 & -1.129 & -1.783 & -1.690 \\
-1.684 & -2.380 & -2.092 & -1.879 \\
-1.796 & -2.644 & -2.211 & -2.023 \\
-1.927 & -2.930 & -2.278 & -2.177 \\
-2.097 & -3.245 & -2.295 & -2.344 \\
-2.299 & -3.590 & -2.412 & -2.525 \\
-2.551 & -3.973 & -2.525 & -2.722 \\
-2.840 & -4.398 & -2.746 & -2.937 \\
-3.168 & -4.874 & -3.015 & -3.172 \\
-3.522 & -5.410 & -3.361 & -3.431 \\
-3.987 & -6.017 & -3.619 & -3.718 \\
-4.461 & -6.711 & -4.149 & -4.036 \\
-4.983 & -7.514 & -4.653 & -4.392
\end{tabular}
\begin{tabular}{rrrlllll} 
Fitting & results & in & the entire temperature range & \\
T & V & VFNT & VFCR & VFCN & VFFM \\
50.05 & .100000 & .045729 & .041916 & .047220 & .183638 \\
75.12 & .150000 & .119218 & .103723 & .120245 & .379017 \\
100.17 & .190000 & .274938 & .228622 & .271701 & .712813 \\
125.22 & .600000 & .573755 & .458602 & .556947 & 1.242274 \\
150.30 & 1.000000 & 1.102366 & .851063 & 1.053327 & 2.032195 \\
175.35 & 1.860000 & 1.971729 & 1.476784 & 1.858037 & 3.146510 \\
200.40 & 3.270000 & 3.317881 & 2.420624 & 3.088735 & 4.648328 \\
225.47 & 5.630000 & 5.293673 & 3.776935 & 4.876388 & 6.592209 \\
250.52 & 8.600000 & 8.048237 & 5.638572 & 7.348210 & 9.012339 \\
275.57 & 12.650000 & 11.715766 & 8.094285 & 10.620223 & 11.926517 \\
300.62 & 17.690000 & 16.384554 & 11.214130 & 14.773021 & 15.326583 \\
325.70 & 23.580000 & 22.079089 & 15.044083 & 19.839594 & 19.182983 \\
350.75 & 29.640000 & 28.709136 & 19.576469 & 25.762834 & 23.426576 \\
375.80 & 35.840000 & 36.099426 & 24.769893 & 32.422129 & 27.979782 \\
400.87 & 41.510000 & 43.975909 & 30.533683 & 39.617613 & 32.748024 \\
425.92 & 46.950000 & 51.972562 & 36.716279 & 47.067418 & 37.615765 \\
450.97 & 52.400000 & 46.345424 & 40.187785 & 45.322347 & 22.184007 \\
462.67 & 55.440000 & 53.099613 & 46.439729 & 51.985524 & 28.545636 \\
476.05 & 59.570000 & 60.768890 & 53.788245 & 59.598577 & 37.089753 \\
487.72 & 63.810000 & 67.125449 & 60.146397 & 65.959294 & 45.439528 \\
499.42 & 68.730000 & 72.929307 & 66.244810 & 71.822067 & 54.279365 \\
511.12 & 74.310000 & 77.960176 & 71.854825 & 76.963617 & 63.075784 \\
519.47 & 78.470000 & 81.002113 & 75.458219 & 80.110255 & 68.995159 \\
526.15 & 81.640000 & 83.089095 & 78.061344 & 82.291746 & 73.348936 \\
531.17 & 84.140000 & 84.454969 & 79.842218 & 83.732460 & 76.333647 \\
537.85 & 86.990000 & 86.011279 & 81.970639 & 85.390236 & 79.863493 \\
546.20 & 88.850000 & 87.563757 & 84.241244 & 87.067081 & 83.504374 \\
556.22 & 89.870000 & 88.918606 & 86.412761 & 88.558719 .86 .728488 \\
574.60 & 90.460000 & 90.308221 & 89.019474 & 90.140012 & 89.881430 \\
599.65 & 90.560000 & 90.918100 & 90.551567 & 90.878470 & 90.944794 \\
624.72 & 90.580000 & 91.020123 & 90.955365 & 91.014822 & 91.028241 \\
651.45 & 90.750000 & 91.029652 & 91.024145 & 91.029359 & 91.029996 \\
674.82 & 90.800000 & 91.029991 & 91.029642 & 91.029980 & 91.030000 \\
699.90 & 90.950000 & 91.030000 & 91.029991 & 91.030000 & 91.030000 \\
724.95 & 90.900000 & 91.030000 & 91.030000 & 91.030000 & 91.030000
\end{tabular}
750.0090 .89000091 .03000091 .03000091 .03000091 .030000 775.0591 .04000091 .03000091 .03000091 .03000091 .030000 800.1291 .03000091 .03000091 .03000091 .03000091 .030000
\begin{tabular}{rcccccl} 
Fitting & rate dV/dT & in the entire temperature range \\
T & VD & VDNT & VDCR & VDCN & VDFM \\
50.05 & .001994 & .001964 & .001713 & .001984 & .006238 \\
75.12 & .001597 & .004483 & .003716 & .004425 & .011371 \\
100.17 & .016367 & .009153 & .007260 & .008858 & .019116 \\
125.22 & .015949 & .017014 & .012987 & .016182 & .029980 \\
150.30 & .034331 & .029395 & .021687 & .027532 & .044581 \\
175.35 & .056287 & .047477 & .033988 & .043869 & .063032 \\
200.40 & .094136 & .072355 & .050421 & .066049 & .085283 \\
225.47 & .118563 & .104373 & .070989 & .094241 & .110524 \\
250.52 & .161677 & .143776 & .095669 & .128543 & .138203 \\
275.57 & .201198 & .188875 & .123200 & .167355 & .166266 \\
300.62 & .234848 & .237368 & .152042 & .208607 & .192835 \\
325.70 & .241916 & .286149 & .180256 & .249594 & .215958 \\
350.75 & .247505 & .333937 & .207161 & .289273 & .235538 \\
375.80 & .226167 & .377615 & .230969 & .325032 & .250227 \\
400.87 & .217166 & .419042 & .252973 & .358567 & .262075 \\
425.92 & .217565 & .454277 & .270933 & .386595 & .269273 \\
450.97 & .259829 & .518990 & .417385 & .499692 & .285043 \\
462.67 & .308670 & .571709 & .457946 & .549754 & .343355 \\
476.05 & .363325 & .615731 & .491032 & .591258 & .408191 \\
487.72 & .420513 & .629331 & .500008 & .603604 & .453461 \\
499.42 & .476923 & .606242 & .479920 & .580793 & .473687 \\
511.12 & .498204 & .531899 & .419588 & .509003 & .449582 \\
519.47 & .474551 & .445717 & .350744 & .426201 & .397909 \\
526.15 & .498008 & .363080 & .285167 & .346972 & .338350 \\
531.17 & .426647 & .283894 & .222655 & .271176 & .273100 \\
537.85 & .222754 & .180931 & .141638 & .172724 & .181460 \\
546.20 & .101796 & .108144 & .084465 & .103164 & .114152 \\
556.22 & .032100 & .064882 & .050539 & .061841 & .072722 \\
574.60 & .003992 & .039440 & .030575 & .037535 & .049169 \\
599.65 & .000798 & .042837 & .033005 & .040689 & .061295 \\
624.72 & .006360 & .053212 & .040760 & .050450 & .086728 \\
651.45 & .002139 & .043029 & .032767 & .040720 & .079952 \\
674.82 & .005981 & .043910 & .033276 & .041490 & .090944 \\
699.90 & -.001996 & .019056 & .014370 & .017978 & .044086 \\
724.95 & -.000399 & .038199 & .028670 & .035983 & .098155 \\
750.00 & .005988 & .050227 & .037529 & .047247 & .142614 \\
775.05 & -.00399 & -.004339 & -.003228 & -.004076 & -.013548 \\
800.12 & -.000399 & .000000 & .000000 & .000000 & .000000
\end{tabular}

Syncrude pitch \(50^{\circ} \mathrm{C} / \mathrm{min}, 800^{\circ} \mathrm{C}\) final temperature RUN\# Syn29
\begin{tabular}{|c|c|c|c|c|c|c|c|c|}
\hline  & YNT & YFNT & YCR & YFCR & YCN & YFCN & YFM & YF \\
\hline 01230 & -10.282 & -10.38 & -10.47 & -10.578 & . 929 & . 826 & . 683 & 962 \\
\hline 01244 & -10.463 & -10.514 & -10.655 & -10.705 & . 748 & . 697 & . 855 & 775 \\
\hline . 001262 & -10.700 & -10.681 & -10.888 & -10.869 & . 512 & . 531 & 651 & 533 \\
\hline . 001281 & -10.930 & -10.852 & -11.116 & -11.038 & . 282 & . 361 & . 440 & 285 \\
\hline . 001292 & -11.047 & -10.957 & -11.231 & -11.141 & . 167 & . 256 & . 210 & 133 \\
\hline 001308 & -11.179 & -11.101 & -11.360 & -11.283 & . 035 & 113 & . 076 & 6 \\
\hline 001328 & -11.309 & -11.285 & -11.488 & -11.463 & . 09 & 070 & . 410 & 42 \\
\hline 001361 & -11.455 & -11.592 & -11.629 & -11.765 & -. 239 & -. 375 & . 870 & 787 \\
\hline 001391 & -11.336 & -11.124 & -11.718 & -11.493 & -. 921 & -1.148 & -1.216 & -1.112 \\
\hline 01442 & -11.480 & -11.354 & -11.846 & -11.713 & -1.057 & -1.384 & -1.451 & -1.275 \\
\hline 001492 & -11.625 & -11.578 & -11.977 & -11.927 & -1.196 & -1.637 & -1.547 & -1.434 \\
\hline 001545 & -11.795 & -11.818 & -12.132 & -12.157 & -1.359 & -1 & -1.630 & \\
\hline & & & & & & -2.266 & & \\
\hline
\end{tabular}
\begin{tabular}{lllllllll}
.001670 & -12.257 & -12.382 & -12.565 & -12.698 & -1.809 & -2.624 & -1.879 & -2.006 \\
.001743 & -12.570 & -12.716 & -12.863 & -13.018 & -2.116 & -2.943 & -2.063 & -2.243 \\
.001824 & -12.944 & -13.079 & -13.222 & -13.366 & -2.484 & -3.409 & -2.315 & -2.502 \\
.001896 & -13.300 & -13.404 & -13.566 & -13.677 & -2.836 & -3.884 & -2.606 & -2.732 \\
.002001 & -13.861 & -13.878 & -14.111 & -14.131 & -3.391 & -4.673 & -2.980 & -3.069 \\
.002107 & -14.375 & -14.361 & -14.611 & -14.594 & -3.900 & -6.428 & -3.596 & -3.413 \\
.002285 & -15.392 & -15.162 & -15.608 & -15.362 & -4.911 & -8.201 & -4.158 & -3.983
\end{tabular}
\begin{tabular}{rrrrrrr} 
Fitting & results in & the entire & temperature range & \\
T & V & VFNT & VFCR & VFCN & VFFM \\
51.65 & .140000 & .009799 & .009251 & .010091 & .069018 \\
100.00 & .190000 & .076477 & .066885 & .076391 & .315460 \\
164.50 & .480000 & .603673 & .489475 & .585025 & 1.453963 \\
201.35 & 1.520000 & 1.542227 & 1.209491 & 1.474579 & 2.903798 \\
226.70 & 2.760000 & 2.716734 & 2.089952 & 2.577077 & 4.403386 \\
254.35 & 5.230000 & 4.728916 & 3.574526 & 4.452674 & 6.612174 \\
275.10 & 7.850000 & 6.892639 & 5.154681 & 6.459578 & 8.712572 \\
300.40 & 11.990000 & 10.464453 & 7.749320 & 9.761545 & 11.826934 \\
325.75 & 17.060000 & 15.226500 & 11.209405 & 14.156507 & 15.572819 \\
351.10 & 22.760000 & 21.261362 & 15.629579 & 19.730895 & 19.927271 \\
374.15 & 28.370000 & 27.825962 & 20.516929 & 25.816720 & 24.368376 \\
397.20 & 34.000000 & 35.271960 & 26.202357 & 32.765722 & 29.197021 \\
420.20 & 39.550000 & 43.319420 & 32.57 .4092 & 40.353804 & 34.309169 \\
445.55 & 45.630000 & 52.465639 & 40.203254 & 49.112623 & 40.152213 \\
461.65 & 49.930000 & 45.544597 & 40.142579 & 44.732081 & 25.361219 \\
480.10 & 56.230000 & 57.049162 & 51.010700 & 56.134667 & 37.631151 \\
491.60 & 61.490000 & 64.046787 & 57.913998 & 63.119553 & 46.535234 \\
500.85 & 66.500000 & 69.320859 & 63.320820 & 68.418263 & 54.065860 \\
507.75 & 70.620000 & 72.949788 & 67.174942 & 72.086372 & 59.693159 \\
519.25 & 77.790000 & 78.265718 & 73.090112 & 77.503543 & 68.630844 \\
530.80 & 83.510000 & 82.552398 & 78.202721 & 81.925942 & 76.428641 \\
540.00 & 86.620000 & 85.182659 & 81.583735 & 84.676348 & 81.413289 \\
553.80 & 88.850000 & 87.920141 & 85.456989 & 87.589287 & 86.580788 \\
574.55 & 89.660000 & 89.941164 & 88.853273 & 89.808968 & 89.957255 \\
599.90 & 89.980000 & 90.609394 & 90.362395 & 90.584542 & 90.669984 \\
650.55 & 90.220000 & 90.699888 & 90.698331 & 90.699812 & 90.700000 \\
698.95 & 90.400000 & 90.700000 & 90.700000 & 90.700000 & 90.700000 \\
749.60 & 90.570000 & 90.700000 & 90.700000 & 90.700000 & 90.700000 \\
800.30 & 90.700000 & 90.700000 & 90.700000 & 90.700000 & 90.700000
\end{tabular}

Fitting rate \(d V / d T\) in the entire temperature range
\begin{tabular}{rlllll}
\(T\) & VD & VDNT & VDCR & VDCN & VDFM \\
51.65 & .001034 & .000490 & .000446 & .000497 & .002632 \\
100.00 & .004496 & .002968 & .002509 & .002925 & .009479 \\
164.50 & .028223 & .017630 & .013835 & .016868 & .033618 \\
201.35 & .048915 & .038858 & .029491 & .036688 & .058773 \\
226.70 & .089331 & .062112 & .046199 & .058174 & .081707 \\
254.35 & .126265 & .096965 & .070713 & .090106 & .111234 \\
275.10 & .163636 & .129992 & .093525 & .120147 & .135785 \\
300.40 & .200000 & .177651 & .125892 & .163208 & .167061 \\
325.75 & .224852 & .232004 & .162139 & .211965 & .198132 \\
351.10 & .243384 & .290795 & .200647 & .264330 & .227298 \\
374.15 & .244252 & .345237 & .235665 & .312477 & .250483 \\
397.20 & .241304 & .399266 & .269833 & .359941 & .270272 \\
420.20 & .239842 & .450452 & .301601 & .404580 & .285841 \\
445.55 & .267081 & .499500 & .331252 & .446924 & .296595 \\
461.65 & .341463 & .577103 & .476958 & .560143 & .371255 \\
480.10 & .457391 & .663253 & .545310 & .642807 & .489652 \\
491.60 & .541622 & .675488 & .553640 & .654083 & .541543 \\
500.85 & .597101 & .646257 & .528390 & .625343 & .552650 \\
507.75 & .623478 & .595681 & .486170 & .576110 & .533992 \\
519.25 & .495238 & .454474 & .369847 & .439180 & .439911 \\
530.80 & .338043 & .299108 & .242723 & .288809 & .312035
\end{tabular}
\begin{tabular}{llllll}
540.00 & .161594 & .193218 & .156450 & .186448 & .213631 \\
553.80 & .039036 & .105838 & .085423 & .102036 & .127370 \\
574.55 & .012623 & .078144 & .062780 & .075238 & .106271 \\
599.90 & .004738 & .074164 & .059265 & .071297 & .116186 \\
650.55 & .003719 & .088169 & .069767 & .084523 & .179029 \\
698.95 & .003356 & .090528 & .071034 & .086578 & .229655 \\
749.60 & .002564 & .062715 & .048819 & .059842 & .196353 \\
800.30 & .002564 & .000000 & .000000 & .000000 & .000000
\end{tabular}

\section*{Syncrude pitch \(100^{\circ} \mathrm{C} / \mathrm{min}, 800^{\circ} \mathrm{C}\) final temperature RUN\# Syn 18}
\begin{tabular}{|c|c|c|c|c|c|c|c|c|}
\hline \multicolumn{9}{|l|}{sperimental results Y1, Y2 and fitted results Y1F, Y2F} \\
\hline T & YNT & YFNT & YCR & YFCR & YCN & YFCN & YFM & YF \\
\hline . 001127 & -9.308 & -9.498 & -9.564 & -9.751 & 1.722 & 1.534 & 232 & 1.444 \\
\hline . 001158 & -9.516 & -9.743 & -9.764 & -9.990 & 1.517 & 1.291 & . 996 & 1.152 \\
\hline . 001176 & -9.785 & -9.889 & -10.028 & -10.132 & 1.250 & 1.146 & 1.482 & 78 \\
\hline 001191 & -10.007 & -10.001 & -10.247 & -10.241 & 1.029 & 1.035 & 1.452 & 844 \\
\hline . 001205 & -10.215 & -10.116 & -10.452 & -10.354 & . 822 & . 921 & 1.225 & 707 \\
\hline . 001220 & -10.398 & -10.235 & -10.631 & -10.469 & . 641 & . 803 & . 947 & 566 \\
\hline . 001236 & -10.552 & -10.356 & -10.782 & -10.587 & . 487 & . 683 & 653 & 422 \\
\hline . 001251 & -10.677 & -10.480 & -10.904 & -10.708 & . 364 & . 560 & 336 & 273 \\
\hline . 001267 & -10.772 & -10.606 & -10.996 & -10.831 & . 270 & . 435 & 016 & 123 \\
\hline . 001284 & -10.850 & -10.737 & -11.071 & -10.958 & . 192 & . 305 & . 242 & 033 \\
\hline . 001301 & -10.918 & -10.871 & -11.135 & -11.089 & . 126 & 172 & 435 & 3 \\
\hline . 001324 & -10.997 & -11.056 & -11.209 & -11.269 & . 049 & . 011 & 639 & 3 \\
\hline . 001343 & -11.055 & -11.199 & -11.264 & -11.408 & -. 009 & . 152 & . 724 & 583 \\
\hline . 001368 & -11.131 & -11.396 & -11.337 & -11.599 & . 084 & . 347 & 30 & . 818 \\
\hline . 001393 & -11.100 & -10.908 & -11.415 & -11.214 & . 489 & . 811 & . 882 & 69 \\
\hline . 001427 & -11.212 & -11.087 & -11.518 & -11.388 & -. 598 & -. 973 & 3 & 18 \\
\hline . 001455 & -11.318 & -11.236 & -11.617 & -11.532 & . 700 & -1.143 & . 977 & 924 \\
\hline . 001492 & -11.453 & -11.432 & -11.744 & -11.722 & . 832 & -1.319 & -1.114 & -1.064 \\
\hline . 001523 & -11.579 & -11.596 & -11.863 & -11.881 & 956 & -1.503 & -1.153 & -1.182 \\
\hline . 001555 & -11.720 & -11.768 & -11.997 & -12.047 & -1.094 & -1.696 & -1.228 & -1. 304 \\
\hline . 001589 & -11.868 & -11.946 & -12.139 & -12.220 & -1.239 & -1.950 & -1.357 & 1.432 \\
\hline . 001624 & -12.032 & -12.133 & -12.295 & -12.401 & -1.400 & -2.218 & -1.463 & -1.565 \\
\hline . 001661 & -12.210 & -12.327 & -12.467 & -12.589 & -1.576 & -2.562 & -1.593 & -1.704 \\
\hline . 001709 & -12.463 & -12.584 & -12.712 & -12.838 & -1.826 & -2.998 & -1.740 & -1.888 \\
\hline . 001760 & -12.760 & -12.855 & -13.000 & -13.101 & -2.120 & -3.473 & -1.922 & -2.082 \\
\hline . 001826 & -13.145 & -13.204 & -13.376 & -13.438 & -2.502 & -4.077 & -2.241 & -2.331 \\
\hline . 001909 & -13.593 & -13.644 & -13.813 & -13.865 & -2.946 & -5.319 & -2.719 & -2.646 \\
\hline . 001999 & -14.121 & -14.124 & -14.330 & -14.330 & -3.471 & -6.997 & -3.127 & -2.989 \\
\hline 114 & -15.006 & -14.735 & -15.203 & -14.922 & -4.353 & -9.207 & -3.556 & -3.426 \\
\hline
\end{tabular}
\begin{tabular}{|c|c|c|c|c|c|}
\hline T & V & VFNT & VFCR & VFCN & VFFM \\
\hline 50.40 & . 010000 & . 001540 & . 001496 & . 001588 & . 016485 \\
\hline 101.40 & . 000000 & . 018878 & . 017085 & . 018978 & . 105664 \\
\hline 152.30 & . 080000 & . 129853 & . 111371 & . 128043 & . 442528 \\
\hline 199.90 & . 420000 & . 550585 & . 453851 & . 535174 & 1.293837 \\
\hline 227.10 & 1.120000 & 1.116204 & . 902845 & 1.077492 & 2.185278 \\
\hline 250.80 & 2.050000 & 1.948625 & 1.553575 & 1.871033 & 3.301432 \\
\hline 274.60 & 3.440000 & 3.247629 & 2.557005 & 3.103598 & 4.816467 \\
\hline 295.00 & 5.330000 & 4.856552 & 3.789285 & 4.624784 & 6.483361 \\
\hline 312.00 & 7.450000 & 6.635186 & 5.144724 & 6.302401 & 8.163141 \\
\hline 329.00 & 9.930000 & 8.884788 & 6.854845 & 8.420862 & 10.127888 \\
\hline 342.60 & 12.160000 & 11.066733 & 8.513106 & 10.473818 & 11.914502 \\
\hline 356.20 & 14.630000 & 13.618970 & 10.455958 & 12.874559 & 13.896846 \\
\hline 369.80 & 17.290000 & 16.562265 & 12.704596 & 15.643977 & 16.076091 \\
\hline 383.40 & 20.210000 & 19.907563 & 15.275332 & 18.794581 & 18.449990 \\
\hline 397.00 & 23.230000 & 23.653224 & 18.177902 & 22.328055 & 21.012616 \\
\hline 414.00 & 26.990000 & 28.871191 & 22.274075 & 27.264504 & 24.466045 \\
\hline 427.50 & 30.220000 & 33.395664 & 25.885633 & 31.561762 & 27.389481 \\
\hline
\end{tabular}
\begin{tabular}{lllllll}
444.50 & 34.100000 & 39.480515 & 30.848632 & 37.371937 & 31.268862 \\
458.10 & 37.280000 & 30.294054 & 25.465104 & 29.445278 & 17.320647 \\
471.70 & 40.440000 & 36.300444 & 30.645522 & 35.289245 & 21.944588 \\
481.90 & 42.920000 & 41.121766 & 34.873027 & 39.993312 & 25.922760 \\
495.50 & 46.340000 & 47.837747 & 40.887724 & 46.571202 & 31.885258 \\
505.70 & 49.260000 & 52.973813 & 45.607671 & 51.626478 & 36.804164 \\
515.90 & 52.570000 & 58.079517 & 50.426936 & 56.678531 & 42.037160 \\
526.00 & 56.470000 & 62.995823 & 55.214767 & 61.574160 & 47.440367 \\
536.20 & 61.340000 & 67.710001 & 59.976033 & 66.304153 & 53.008302 \\
546.40 & 67.070000 & 72.070023 & 64.570176 & 70.718213 & 58.555721 \\
556.60 & 73.250000 & 75.990181 & 68.907305 & 74.728841 & 63.938346 \\
566.80 & 79.270000 & 79.408748 & 72.906769 & 78.269233 & 69.011118 \\
577.00 & 84.200000 & 82.293013 & 76.502948 & 81.298659 & 73.641053 \\
590.60 & 88.020000 & 85.308779 & 80.594647 & 84.526661 & 78.944401 \\
614.40 & 89.670000 & 88.564528 & 85.747974 & 88.135044 & 85.506564 \\
644.90 & 90.020000 & 90.176356 & 89.132030 & 90.041295 & 89.407948 \\
699.30 & 90.270000 & 90.574109 & 90.516899 & 90.569747 & 90.561938 \\
801.20 & 90.580000 & 90.580000 & 90.579999 & 90.580000 & 90.580000
\end{tabular}
\begin{tabular}{ccccccl} 
Fitting & rate dV/dT & in the entire temperature range \\
T. & VD & VDNT & VDCR & VDCN & VDFM \\
50.40 & -.000196 & .000089 & .000084 & .000091 & .000721 \\
101.40 & .001572 & .000832 & .000734 & .000829 & .003566 \\
152.30 & .007143 & .004537 & .003793 & .004433 & .011985 \\
199.90 & .025735 & .015877 & .012761 & .015295 & .029320 \\
227.10 & .039241 & .029010 & .022876 & .027756 & .045017 \\
250.80 & .058403 & .046410 & .036052 & .044166 & .062805 \\
274.60 & .092647 & .070968 & .054374 & .067202 & .084707 \\
295.00 & .124706 & .098349 & .074537 & .092766 & .106299 \\
312.00 & .145882 & .125836 & .094563 & .118330 & .125877 \\
329.00 & .163971 & .157746 & .117598 & .147911 & .146683 \\
342.60 & .181618 & .186382 & .138102 & .174379 & .163949 \\
356.20 & .195588 & .217506 & .160227 & .203072 & .181427 \\
369.80 & .214706 & .250916 & .183810 & .233795 & .198912 \\
383.40 & .222059 & .285889 & .208314 & .265870 & .215849 \\
397.00 & .221176 & .322448 & .233749 & .299315 & .232321 \\
414.00 & .239259 & .370407 & .266876 & .343077 & .252370 \\
427.50 & .228235 & .408066 & .292643 & .377325 & .266473 \\
444.50 & .233824 & .456957 & .325871 & .421681 & .283511 \\
458.10 & .232353 & .392638 & .311918 & .376596 & .235294 \\
471.70 & .243137 & .449670 & .355393 & .430556 & .279832 \\
481.90 & .251471 & .493090 & .388258 & .471541 & .315379 \\
495.50 & .286275 & .550524 & .431395 & .525615 & .364805 \\
505.70 & .324510 & .588064 & .459198 & .560798 & .399845 \\
515.90 & .386139 & .616536 & .479789 & .587278 & .429851 \\
526.00 & .477451 & .627702 & .486870 & .597254 & .448357 \\
536.20 & .561765 & .609276 & .471047 & .579092 & .445692 \\
546.40 & .605882 & .552983 & .426173 & .525030 & .414022 \\
556.60 & .590196 & .458760 & .352468 & .435121 & .351351 \\
566.80 & .483333 & .335994 & .257369 & .318359 & .263082 \\
577.00 & .280882 & .212114 & .162001 & .200783 & .169709 \\
590.60 & .069328 & .098484 & .074930 & .093104 & .081032 \\
614.40 & .011475 & .044710 & .033800 & .042177 & .038554 \\
644.90 & .004596 & .036952 & .027721 & .034769 & .033719 \\
699.30 & .003042 & .033062 & .024494 & .030979 & .033079 \\
801.20 & .003042 & .000000 & .000000 & .000000 & .000000
\end{tabular}

Syncrude pitch \(150^{\circ} \mathrm{C} / \mathrm{min}, 800^{\circ} \mathrm{C}\) final temperature RUN\# Syn08
\begin{tabular}{|c|c|c|c|c|c|c|c|c|}
\hline \multicolumn{2}{|l|}{\multirow[t]{2}{*}{Experimental}} & \multicolumn{2}{|l|}{Y1, Y2} & \multicolumn{3}{|r|}{Y1F, Y2F} & \multirow[b]{2}{*}{YFM} & \multirow[b]{2}{*}{YF} \\
\hline & & YFNT & YCR & YFCR & YCN & YFCN & & \\
\hline 112 & -8.944 & -9.092 & -9.181 & -9.327 & 2.158 & 2.011 & . 707 & 1.861 \\
\hline
\end{tabular}
\begin{tabular}{lrrrrrrrr}
.001153 & -9.102 & -9.290 & -9.333 & -9.521 & 2.002 & 1.814 & 1.507 & 1.643 \\
.001171 & -9.317 & -9.445 & -9.544 & -9.672 & 1.788 & 1.660 & 1.795 & 1.473 \\
.001190 & -9.581 & -9.605 & -9.804 & -9.829 & 1.526 & 1.501 & 1.742 & 1.297 \\
.001203 & -9.764 & -9.715 & -9.984 & -9.936 & 1.344 & 1.392 & 1.677 & 1.176 \\
.001217 & -9.942 & -9.828 & -10.160 & -10.046 & 1.166 & 1.280 & 1.494 & 1.052 \\
.001230 & -10.107 & -9.941 & -10.322 & -10.157 & 1.003 & 1.168 & 1.280 & .928 \\
.001244 & -10.243 & -10.058 & -10.456 & -10.272 & .867 & 1.051 & .965 & .798 \\
.001258 & -10.351 & -10.177 & -10.560 & -10.387 & .760 & .934 & .658 & .668 \\
.001273 & -10.437 & -10.300 & -10.644 & -10.508 & .675 & .812 & .364 & .533 \\
.001288 & -10.513 & -10.426 & -10.717 & -10.631 & .600 & .687 & .171 & .394 \\
.001311 & -10.610 & -10.619 & -10.810 & -10.820 & .504 & .494 & -.063 & .182 \\
.001335 & -10.695 & -10.818 & -10.892 & -11.014 & .420 & .297 & -.261 & -.037 \\
.001368 & -10.808 & -11.098 & -10.999 & -11.288 & .309 & .019 & -.415 & -.345 \\
.001394 & -10.787 & -10.634 & -11.086 & -10.926 & -.124 & -.269 & -.479 & -.297 \\
.001421 & -10.892 & -10.784 & -11.184 & -11.072 & -.227 & -.486 & -.517 & -.409 \\
.001449 & -11.001 & -10.938 & -11.287 & -11.222 & -.334 & -.714 & -.605 & -.525 \\
.001489 & -11.162 & -11.158 & -11.439 & -11.434 & -.491 & -.954 & -.715 & -.690 \\
.001530 & -11.349 & -11.387 & -11.617 & -11.657 & -.674 & -1.210 & -.800 & -.862 \\
.001574 & -11.553 & -11.630 & -11.813 & -11.893 & -.876 & -1.549 & -.956 & -1.044 \\
.001620 & -11.784 & -11.888 & -12.036 & -12.145 & -1.104 & -1.918 & -1.121 & -1.238 \\
.001682 & -12.125 & -12.231 & -12.366 & -12.477 & -1.440 & -2.314 & -1.309 & -1.496 \\
.001749 & -12.481 & -12.603 & -12.711 & -12.839 & -1.793 & -2.748 & -1.676 & -1.775 \\
.001821 & -12.947 & -13.004 & -13.167 & -13.228 & -2.255 & -3.218 & -1.906 & -2.076 \\
.001900 & -13.438 & -13.442 & -13.648 & -13.654 & -2.743 & -3.957 & -2.406 & -2.405 \\
.001986 & -13.882 & -13.916 & -14.083 & -14.115 & -3.185 & -6.081 & -3.021 & -2.761 \\
.002120 & -14.878 & -14.663 & -15.064 & -14.840 & -4.176 & -9.310 & -3.387 & -3.321
\end{tabular}
\begin{tabular}{|c|c|c|c|c|c|}
\hline T & V & VFNT & VFCR & VFCN & VFEM \\
\hline 50.00 & . 020000 & . 000891 & 000867 & . 000917 & 007966 \\
\hline 125.75 & . 070000 & . 034308 & . 030388 & . 034164 & 134342 \\
\hline 198.50 & . 320000 & . 396712 & . 330372 & . 386644 & 896348 \\
\hline 230.45 & . 970000 & . 938068 & . 764888 & . 907488 & 1.746136 \\
\hline 253.10 & 1.630000 & 1.623374 & 1.306550 & 1.563176 & 2.669480 \\
\hline 275.90 & 2.850000 & 2.694018 & 2.143508 & 2.583332 & 3.949116 \\
\hline 298.55 & 4.820000 & 4.276303 & 3.369713 & 4.085792 & 5.642770 \\
\hline 321.35 & 7.260000 & 6.553257 & 5.123781 & 6.242050 & 7.845678 \\
\hline 344.00 & 10.660000 & 9.663453 & 7.513555 & 9.182153 & 10.591947 \\
\hline 362.30 & 13.840000 & 12.900822 & 10.003840 & 12.240464 & 13.249213 \\
\hline 380.45 & 17.410000 & 16.819236 & 13.031557 & 15.943453 & 16.286221 \\
\hline 398.60 & 21.390000 & 21.474008 & 16.658307 & 20.348467 & 19.723079 \\
\hline 416.90 & 25.520000 & 26.901856 & 20.942446 & 25.498709 & 23.578196 \\
\hline 430.40 & 28.720000 & 31.342672 & 24.502038 & 29.726985 & 26.655275 \\
\hline 444.05 & 32.080000 & 36.156700 & 28.427027 & 34.329008 & 29.947086 \\
\hline 457.70 & 35.380000 & 28.033573 & 23.801644 & 27.333528 & 18.612215 \\
\hline 476.00 & 39.830000 & 36.320795 & 31.011915 & 35.421705 & 25.296279 \\
\hline 489.50 & 43.350000 & 43.062424 & 37.006635 & 42.024491 & 31.104254 \\
\hline 503.15 & 47.390000 & 50.205200 & 43.520235 & 49.050957 & 37.654348 \\
\hline 512.30 & 50.520000 & 55.048013 & 48.054191 & 53.837831 & 42.353987 \\
\hline 521.45 & 54.040000 & 59.830187 & 52.646333 & 58.587362 & 47.227206 \\
\hline 530.45 & 58.280000 & 64.382933 & 57.147460 & 63.134447 & 52.111450 \\
\hline 539.60 & 63.460000 & 68.766400 & 61.629215 & 67.541374 & 57.077576 \\
\hline 548.60 & 69.320000 & 72.755015 & 65.867186 & 71.582191 & 61.864936 \\
\hline 557.75 & 75.110000 & 76.412080 & 69.927104 & 75.320122 & 66.530691 \\
\hline 566.90 & 80.170000 & 79.617170 & 73.668578 & 78.629972 & 70.894557 \\
\hline 80.55 & 85.600000 & 83.505731 & 78.543562 & 82.705429 & 76.659233 \\
\hline 594.20 & 88.350000 & 86.344521 & 82.480538 & 85.744533 & 81.356480 \\
\hline 612.35 & 89.590000 & 88.711449 & 86.251528 & 88.354415 & 85.839815 \\
\hline 639.65 & 89.970000 & 90.211576 & 89.281375 & 90.095072 & 89.313126 \\
\hline 689.60 & 90.300000 & 90.613468 & 90.562245 & 90.609490 & 90.585096 \\
\hline 798.80 & 90.620000 & 90.620000 & 90.620000 & 90.620000 & 90.620000 \\
\hline
\end{tabular}

Fitting rate \(d V / d T\) in the entire temperature range
\begin{tabular}{rlllll}
\(T\) & VD & VDNT & VDCR & VDCN & VDFM \\
50.00 & .000660 & .000054 & .000051 & .000055 & .000376 \\
125.75 & .003436 & .001397 & .001208 & .001379 & .004351 \\
198.50 & .020344 & .011908 & .009689 & .011514 & .021734 \\
230.45 & .029139 & .024939 & .019867 & .023937 & .037795 \\
253.10 & .053509 & .039780 & .031266 & .038002 & .053566 \\
275.90 & .086976 & .060793 & .047191 & .057825 & .073400 \\
298.55 & .107018 & .088697 & .068073 & .084034 & .096922 \\
321.35 & .150110 & .125044 & .094960 & .118035 & .124536 \\
344.00 & .173770 & .168954 & .127063 & .158945 & .154501 \\
362.30 & .196694 & .210191 & .156917 & .197232 & .180199 \\
380.45 & .219284 & .255424 & .189377 & .239103 & .206139 \\
398.60 & .225683 & .303822 & .223798 & .283764 & .231577 \\
416.90 & .237037 & .355679 & .260370 & .331479 & .256700 \\
430.40 & .246154 & .394629 & .287620 & .367218 & .274068 \\
444.05 & .241758 & .433662 & .314724 & .402941 & .290120 \\
457.70 & .243169 & .389133 & .314606 & .375285 & .260778 \\
476.00 & .260741 & .473696 & .380539 & .455912 & .326435 \\
489.50 & .295971 & .537611 & .429941 & .516685 & .377864 \\
503.15 & .342077 & .596677 & .475083 & .572649 & .427533 \\
512.30 & .384699 & .627795 & .498431 & .601965 & .455504 \\
521.45 & .471111 & .647704 & .512801 & .620502 & .475739 \\
530.45 & .566120 & .644567 & .508947 & .616969 & .479042 \\
539.60 & .651111 & .608896 & .479497 & .582329 & .457857 \\
548.60 & .632787 & .534725 & .420008 & .510976 & .406636 \\
557.75 & .553005 & .435741 & .341385 & .416050 & .335093 \\
566.90 & .397802 & .327743 & .256132 & .312683 & .254815 \\
580.55 & .201465 & .184725 & .143840 & .176034 & .145923 \\
594.20 & .068320 & .097515 & .075665 & .092822 & .078226 \\
612.35 & .013919 & .053958 & .041679 & .051288 & .044148 \\
639.65 & .006607 & .045216 & .034702 & .042890 & .038054 \\
689.60 & .002930 & .035875 & .027236 & .033912 & .031660 \\
798.80 & .002930 & .000000 & .000000 & .000000 & .000000
\end{tabular}

\section*{APPENDIX F Summary of Kinetic Parameters of the 2-Stage Model}
\begin{tabular}{ll}
\(\mathrm{E}_{1}, \mathrm{E}_{2}\) & Reaction activation energy \(\mathrm{J} / \mathrm{mol}\) \\
\(\mathrm{k}_{\mathrm{o}}, \mathrm{k}_{\mathrm{o} 2}\) & pre-exponential factors \\
s.e.e & Standard deviation for each method, \(\%\)
\end{tabular}

CANMET pitch \(25^{\circ} \mathrm{C} / \mathrm{min}\)., \(800^{\circ} \mathrm{C}\)
\begin{tabular}{lcccccc} 
Activation energies & and pre-exponential factors & for both reactions \\
& \(\mathrm{E}_{1}\) & \(\mathrm{k}_{\mathrm{o1}}\) & \(\mathrm{E}_{2}\) & \(\mathrm{k}_{\circ 2}\) & s.e.e \\
2-Integral & 21895.871 & 5.534 & 71347.395 & 44484.445 & 1.425598 \\
2-Coats-Redfern & 18317.044 & 1.207 & 69931.248 & 28659.014 & 8.666605 \\
2-Chen-Nuttall & 19815.807 & 2.663 & 70918.682 & 40047.550 & 4.464558 \\
2-Friedman & 18356.762 & 2.169 & 39422.571 & 222.056 & 5.683572
\end{tabular}

CANMET Pitch \(50^{\circ} \mathrm{C} / \mathrm{min}\)., \(800^{\circ} \mathrm{C}\)
\begin{tabular}{|c|c|c|c|c|c|}
\hline & \(\mathrm{E}_{1}\) & \(\mathrm{k}_{01}\) & \(\mathrm{E}_{2}\) & \(\mathrm{k}^{\text {2 }}\) & s.e.e \\
\hline 2-Integral & 20907.935 & 7.649 & 64473.015 & 24448.637 & 2.128091 \\
\hline 2-Coats-Redfern & 17507.079 & 1.688 & 62943.175 & 15044.712 & 8.270525 \\
\hline 2-Chen-Nuttall & 18939.224 & 3.717 & 63977.704 & 21585.114 & 4.639823 \\
\hline & 12109.801 & 0.833 & 54006.641 & 3350.699 & 11.413135 \\
\hline
\end{tabular}

CANMET pitch \(100^{\circ} \mathrm{C} / \mathrm{min}\)., \(800^{\circ} \mathrm{C}\)
Activation energies and pre-exponential factor for both reactions
\begin{tabular}{lrrrrr} 
& \multicolumn{1}{c}{\(\mathrm{E}_{1}\)} & \(\mathrm{k}_{\mathrm{ol}}\) & \(\mathrm{E}_{2}\) & \(\mathrm{k}_{\mathrm{o} 2}\) & s.e.e. \\
2-Integral & 26914.040 & 39.637 & 72117.918 & 11143.981 & 1.446535 \\
2-Coats-Redfern & 24063.137 & 12.482 & 70720.507 & 72034.563 & 4.703861 \\
2-Chen-Nuttall & 25429.522 & 24.221 & 71699.067 & 100310.634 & 1.899520 \\
2-Friedman & 21904.369 & 12.635 & 108819.449 & 31432193.146 & 5.279363
\end{tabular}

CANMET pitch \(150^{\circ} \mathrm{C} / \mathrm{min}\)., \(800^{\circ} \mathrm{C}\)
Activation energies and pre-exponential factor for both reactions
\begin{tabular}{llrlll} 
2-Integral & 46648.168 & 552.339 & 96651.309 & 3510920.010 & 0.975426 \\
2-Coats-Redfern & 45065.482 & 304.766 & 95534.758 & 2529475.085 & 2.908742 \\
2-Chen-Nuttall & 46065.844 & 463.684 & 96377.125 & 3296730.811 & 1.027556 \\
2-Friedman & 52896.511 & 2804.999 & 92011.326 & 1699475.805 & 1.499851
\end{tabular}

Syncrude Pitch \(25^{\circ} \mathrm{C} / \mathrm{min}\)., \(800^{\circ} \mathrm{C}\)
\begin{tabular}{|c|c|c|c|c|c|}
\hline & \(\mathrm{E}_{1}\) & \(\mathrm{k}_{01}\) & \(\mathrm{E}_{2}\) & \(\mathrm{k}_{02}\) & s.e.e. \\
\hline 2-Integral & 30825.690 & 51.805 & 67665.168 & 25546.591 & 1.941470 \\
\hline 2-Coats-Redfern & 28925.970 & 22.282 & 66149.761 & 15973.274 & 5.238812 \\
\hline 2-Chen-Nuttall & 29969.525 & 38.048 & 67185.904 & 22714.511 & 2.066038 \\
\hline & 22436.425 & 7.251 & 101506.600 & 3875272.258 & 9.746673 \\
\hline
\end{tabular}

Syncrude pitch \(50^{\circ} \mathrm{C} / \mathrm{min}\)., \(800^{\circ} \mathrm{C}\)
Activation energies and pre-exponential factor for both reactions
\begin{tabular}{|c|c|c|c|c|c|}
\hline & \(\mathrm{E}_{1}\) & \(\mathrm{k}_{\mathrm{o} 1}\) & \(\mathrm{E}_{2}\) & \(\mathrm{k}_{0}\) & s.e.e \\
\hline 2-Integral & 37574.160 & 298.225 & 76570.633 & 196413.926 & 2.079083 \\
\hline 2-Coats-Redfern & 36007.909 & 152.170 & 75271.326 & 131230.169 & 4.555376 \\
\hline 2-Chen-Nuttall. & 36950.208 & 240.377 & 76200.343 & 179430.028 & 1.827761 \\
\hline 2-Friedman & 26717.957 & 28.783 & 110908.304 & 34882245.211 & 8.003341 \\
\hline
\end{tabular}

Syncrude pitch \(100^{\circ} \mathrm{C} / \mathrm{min}\)., \(800^{\circ} \mathrm{C}\)
\begin{tabular}{lcrcrrr} 
Activation energies and pre-exponential factor for both reactions \\
& \(\mathrm{E}_{1}\) & \(\mathrm{~K}_{\mathrm{ol}}\) & \(\mathrm{E}_{2}\) & \(\mathrm{k}_{02}\) & s.e.e \\
2-Integral & 44166.789 & 1326.666 & 65511.067 & 35233.527 & 2.970666 \\
2-Coats-Redfern & 42786.603 & 750.710 & 63799.493 & 21122.231 & 4.667832 \\
2-Chen-Nuttall & 43670.266 & 1126.495 & 64938.056 & 30754.384 & 2.683058 \\
2-Friedman & 31580.121 & 99.840 & 78074.842 & 166743.612 & 8.111536
\end{tabular}

Syncrude pitch \(150^{\circ} \mathrm{C} / \mathrm{min}\)., \(800^{\circ} \mathrm{C}\)
\begin{tabular}{lcrcrl} 
Activation energies and pre-exponential factor for both reactions \\
& \(\mathrm{E}_{1}\) & \(\mathrm{~K}_{\mathrm{ol}}\) & \(\mathrm{E}_{2}\) & \(\mathrm{k}_{\mathrm{oz}}\) & S.e.e \\
2-Integral & 46141.423 & 2549.212 & 69808.233 & 103113.970 & 2.859655 \\
2-Coats-Redfern & 44831.086 & 1485.078 & 68221.838 & 64209.953 & 4.405498 \\
2-Chen-Nuttall & 45685.309 & 2194.201 & 69303.236 & 91513.836 & 2.586208 \\
2-Friedman & 34645.511 & 248.057 & 76752.670 & 216684.308 & 6.605246
\end{tabular}

APPENDIX G Kinetic Reaction Rate Constant lnk - 1/T for CANMET and Syncrude Pitches

CANMET PITCH 2-stage model kinetic reaction rate lnk-1/T
\begin{tabular}{lllll} 
Integral method & & & \\
\(1 / \mathrm{T} \mathrm{K}^{-1}\) & \(25^{\circ} \mathrm{C} / \mathrm{min}\) & \(50^{\circ} \mathrm{C} / \mathrm{min}\) & \(100^{\circ} \mathrm{C} / \mathrm{min}\) & \(150{ }^{\circ} \mathrm{C} / \mathrm{min}\) \\
\(800-450{ }^{\circ} \mathrm{C}\) & & & & \\
0.93183 & 2.70633 & 2.87824 & 3.53565 & 4.23878 \\
0.97736 & 2.31555 & 2.52511 & 3.14066 & 3.70941 \\
1.02758 & 1.88462 & 2.13570 & 2.70507 & 3.12564 \\
1.08324 & 1.40700 & 1.70410 & 2.22230 & 2.47863 \\
1.14527 & 0.87469 & 1.22308 & 1.68424 & 1.75753 \\
1.21483 & 0.27771 & 0.68362 & 1.08081 & 0.94883 \\
1.29339 & -0.39649 & 0.07438 & 0.39933 & 0.03553 \\
1.38282 & -1.16391 & -0.61910 & -0.37638 & -1.00407 \\
\(450-50{ }^{\circ} \mathrm{C}\) & & & & \\
1.38282 & -1.93090 & -1.44292 & -0.79669 & -1.44456 \\
1.48553 & -2.20140 & -1.70122 & -1.12919 & -2.02085 \\
1.60472 & -2.51531 & -2.00096 & -1.51504 & -2.68962 \\
1.74471 & -2.88399 & -2.35301 & -1.96822 & -3.47507 \\
1.91146 & -3.32314 & -2.77234 & -2.50801 & -4.41066 \\
2.11345 & -3.85510 & -3.28030 & -3.16189 & -5.54398 \\
2.36317 & -4.51277 & -3.90830 & -3.97029 & -6.94512 \\
2.67982 & -5.34669 & -4.70459 & -4.99532 & -8.72174 \\
3.09444 & -6.43866 & -5.74729 & -6.33755 & -11.04810
\end{tabular}
\begin{tabular}{lllll} 
Chen-Nuttall method & & & \\
\(1 / \mathrm{T} \mathrm{K}\) & \(2{ }^{\circ} \mathrm{C} / \mathrm{min}\) & \(50^{\circ} \mathrm{C} / \mathrm{min}\) & \(100{ }^{\circ} \mathrm{C} / \mathrm{min}\) & \(150{ }^{\circ} \mathrm{C} / \mathrm{min}\) \\
\(800-450{ }^{\circ} \mathrm{C}\) & & & & \\
0.93183 & 2.64931 & 2.80918 & 3.48004 & 4.20656 \\
0.97736 & 2.26088 & 2.45877 & 3.08734 & 3.67869 \\
1.02758 & 1.83253 & 2.07235 & 2.65428 & 3.09658 \\
1.08324 & 1.35779 & 1.64407 & 2.17431 & 2.45141 \\
1.14527 & 0.82867 & 1.16674 & 1.63938 & 1.73235 \\
1.21483 & 0.23528 & 0.63142 & 1.03945 & 0.92594 \\
1.29339 & -0.43486 & 0.02687 & 0.36194 & 0.01523 \\
1.38282 & -1.19767 & -0.66128 & -0.40927 & -1.02141 \\
\(450-50{ }^{\circ} \mathrm{C}\) & & & & \\
1.38282 & -2.31640 & -1.83714 & -1.04233 & -1.52266 \\
1.48553 & -2.56120 & -2.07111 & -1.35648 & -2.09176 \\
1.60472 & -2.84529 & -2.34263 & -1.72105 & -2.75218 \\
1.74471 & -3.17894 & -2.66153 & -2.14923 & -3.52783 \\
1.91146 & -3.57637 & -3.04138 & -2.65925 & -4.45174 \\
2.11345 & -4.05780 & -3.50150 & -3.27706 & -5.57091 \\
2.36317 & -4.65299 & -4.07037 & -4.04087 & -6.95456 \\
2.67982 & -5.40769 & -4.79168 & -5.00937 & -8.70900 \\
3.09444 & -6.39592 & -5.73620 & -6.27756 & -11.00630
\end{tabular}

CANMET SINGLE OVERALL MODEL lnk-1/T
\begin{tabular}{lllll}
\(700-50^{\circ} \mathrm{C}\) & & & & \\
\(1 / \mathrm{T} \mathrm{K}\) & Integral & \(\mathrm{C}-\mathrm{R}\) & \(\mathrm{C}-\mathrm{N}\) & Friedman \\
1.02758 & 0.92624 & 0.27246 & 0.68931 & 0.85065 \\
1.08324 & 0.70466 & 0.06628 & 0.47509 & 0.63309 \\
1.14527 & 0.45771 & -0.16351 & 0.23635 & 0.39061 \\
1.21483 & 0.18075 & -0.42122 & -0.03141 & 0.11868 \\
1.29339 & -0.13202 & -0.71227 & -0.33379 & -0.18843 \\
1.38282 & -0.48805 & -1.04356 & -0.67799 & -0.53800 \\
1.48553 & -0.89697 & -1.42406 & -1.07331 & -0.93951 \\
1.60472 & -1.37151 & -1.86562 & -1.53208 & -1.40544 \\
1.74471 & -1.92884 & -2.38423 & -2.07089 & -1.95267 \\
1.91146 & -2.59270 & -3.00196 & -2.71269 & -2.60450 \\
2.11345 & -3.39686 & -3.75024 & -3.49013 & -3.39409 \\
2.36317 & -4.39107 & -4.67536 & -4.45129 & -4.37027 \\
2.67982 & -5.65170 & -5.84840 & -5.67003 & -5.60805 \\
3.09444 & -7.30243 & -7.38442 & -7.26590 & -7.22885
\end{tabular}
\begin{tabular}{|c|c|c|c|c|}
\hline \multicolumn{5}{|l|}{Coats-Redfern method} \\
\hline \(1 / \mathrm{T} \mathrm{K}^{-1}\) & \(25^{\circ} \mathrm{C} / \mathrm{min}\) & \(50^{\circ} \mathrm{C} / \mathrm{min}\) & \(100^{\circ} \mathrm{C} / \mathrm{min}\) & \(150{ }^{\circ} \mathrm{C} / \mathrm{min}\) \\
\hline \multicolumn{5}{|l|}{\(800-450{ }^{\circ} \mathrm{C}\)} \\
\hline 0.93183 & 2.42538 & 2.56415 & 3.25860 & 4.03605 \\
\hline 0.97736 & 2.04236 & 2.21941 & 2.87125 & 3.51280 \\
\hline 1.02758 & 1.61998 & 1.83923 & 2.44410 & 2.93577 \\
\hline 1.08324 & 1.15184 & 1.41788 & 1.97069 & 2.29624 \\
\hline 1.14527 & 0.63009 & 0.94827 & 1.44305 & 1.58347 \\
\hline 1.21483 & 0.04496 & 0.42161 & 0.85132 & 0.78411 \\
\hline 1.29339 & -0.61585 & -0.17317 & 0.18305 & -0.11864 \\
\hline 1.38282 & -1.36804 & -0.85019 & -0.57763 & -1.14622 \\
\hline \multicolumn{5}{|l|}{450-50 \({ }^{\circ} \mathrm{C}\)} \\
\hline 1.38282 & -2.85843 & -2.38831 & -1.47800 & -1.77594 \\
\hline 1.48553 & -3.08472 & -2.60459 & -1.77527 & -2.33268 \\
\hline 1.60472 & -3.34732 & -2.85558 & -2.12025 & -2.97876 \\
\hline 1.74471 & -3.65574 & -3.15036 & -2.52542 & -3.73756 \\
\hline 1.91146 & -4.02311 & -3.50149 & -3.00804 & -4.64140 \\
\hline 2.11345 & -4.46812 & -3.92682 & -3.59265 & -5.73627 \\
\hline 2.36317 & -5.01830 & -4.45267 & -4.31542 & -7.08987 \\
\hline 2.67982 & -5.71592 & -5.11944 & -5.23188 & -8.80622 \\
\hline 3.09444 & -6.62940 & -5.99253 & -6.43193 & -11.05370 \\
\hline \multicolumn{5}{|l|}{Friedman method} \\
\hline \(1 / \mathrm{T} \mathrm{K}^{-1}\) & \(25^{\circ} \mathrm{C} / \mathrm{min}\) & \(50^{\circ} \mathrm{C} / \mathrm{min}\) & \(100^{\circ} \mathrm{C} / \mathrm{min}\) & \(150{ }^{\circ} \mathrm{C} / \mathrm{min}\) \\
\hline \multicolumn{5}{|l|}{\(800-450{ }^{\circ} \mathrm{C}\)} \\
\hline 0.93183 & 0.98447 & 2.06390 & 5.06693 & 4.03326 \\
\hline 0.97736 & 0.76855 & 1.76810 & 4.47092 & 3.52931 \\
\hline 1.02758 & 0.53044 & 1.44190 & 3.81365 & 2.97356 \\
\hline 1.08324 & 0.26654 & 1.08037 & 3.08520 & 2.35762 \\
\hline 1.14527 & -0.02759 & 0.67743 & 2.27331 & 1.67114 \\
\hline 1.21483 & -0.35744 & 0.22555 & 1.36279 & 0.90126 \\
\hline 1.29339 & -0.72997 & -0.28479 & 0.33451 & 0.03180 \\
\hline 1.38282 & -1.15400 & -0.86569 & -0.83597 & -0.95788 \\
\hline \multicolumn{5}{|l|}{\(450-50^{\circ} \mathrm{C}\)} \\
\hline 1.38282 & -2.27891 & -2.19688 & -1.10676 & -0.85881 \\
\hline 1.48553 & -2.50569 & -2.34648 & -1.37736 & -1.51230 \\
\hline 1.60472 & -2.76886 & -2.52009 & -1.69139 & -2.27065 \\
\hline 1.74471 & -3.07795 & -2.72399 & -2.06022 & -3.16131 \\
\hline 1.91146 & -3.44611 & -2.96687 & -2.49953 & -4.22221 \\
\hline 2.11345 & -3.89209 & -3.26108 & -3.03170 & -5.50733 \\
\hline 2.36317 & -4.44346 & -3.62481 & -3.68963 & -7.09615 \\
\hline 2.67982 & -5.14259 & -4.08602 & -4.52387 & -9.11074 \\
\hline 3.09444 & -6.05806 & -4.68995 & -5.61626 & -11.74870 \\
\hline
\end{tabular}

SYNCRUDE PITCH 2-STAGE MODEL KINETIC reaction rate \(\operatorname{lnk}\)-1/T
\begin{tabular}{|c|c|c|c|c|}
\hline \multicolumn{5}{|l|}{Integral method} \\
\hline \(1 / \mathrm{T} \mathrm{K}^{-1}\) & \(25^{\circ} \mathrm{C} /\) min & \(50^{\circ} \mathrm{C} / \mathrm{min}\) & \(100^{\circ} \mathrm{C} / \mathrm{min}\) & \(150^{\circ} \mathrm{C} / \mathrm{min}\) \\
\hline \multicolumn{5}{|l|}{\(800.450{ }^{\circ} \mathrm{C}\)} \\
\hline 0.93183 & 2.56439 & 3.60600 & 3.12732 & 3.71953 \\
\hline 0.97736 & 2.19378 & 3.18661 & 2.76851 & 3.33718 \\
\hline 1.02758 & 1.78509 & 2.72413 & 2.37282 & 2.91555 \\
\hline 1.08324 & 1.33213 & 2.21155 & 1.93428 & 2.44824 \\
\hline 1.14527 & 0.82729 & 1.64027 & 1.44551 & 1.92741 \\
\hline 1.21483 & 0.26112 & 0.99958 & 0.89736 & 1.34330 \\
\hline 1.29339 & -0.37828 & 0.27603 & 0.27832 & 0.68366 \\
\hline 1.38282 & -1.10610 & -0.54757 & -0.42633 & -0.06721 \\
\hline \multicolumn{5}{|l|}{\(450-50{ }^{\circ} \mathrm{C}\)} \\
\hline 1.38282 & -1.17957 & -0.55165 & -0.15558 & 0.16910 \\
\hline 1.48553 & -1.56039 & -1.01584 & -0.70122 & -0.40093 \\
\hline 1.60472 & -2.00232 & -1.55452 & -1.33442 & -1.06243 \\
\hline 1.74471 & -2.52136 & -2.18718 & -2.07809 & -1.83935 \\
\hline 1.91146 & -3.13961 & -2.94078 & -2.96391 & -2.76478 \\
\hline 2.11345 & -3.88852 & -3.85364 & -4.03694 & -3.88578 \\
\hline 2.36317 & -4.81441 & -4.98224 & -5.36355 & -5.27170 \\
\hline 2.67982 & -5.98842 & -6.41327 & -7.04567 & -7.02903 \\
\hline 3.09444 & -7.52573 & -8.28713 & -9.24830 & -9.33014 \\
\hline
\end{tabular}
\begin{tabular}{lllll}
\multicolumn{5}{c}{ Coats-Redfern method } \\
\(1 / \mathrm{T} \mathrm{K}\) & & & & \\
800 \\
\(800-450^{\circ} \mathrm{C} / \mathrm{Cin}\) & & \(50^{\circ} \mathrm{C} / \mathrm{min}\) & \(100^{\circ} \mathrm{C} / \mathrm{min}\) & \(150{ }^{\circ} \mathrm{C} / \mathrm{min}\) \\
0.93183 & 2.26465 & 3.34835 & 2.80748 & 3.42366 \\
0.97736 & 1.90234 & 2.93608 & 2.45804 & 3.05000 \\
1.02758 & 1.50280 & 2.48145 & 2.07270 & 2.63794 \\
1.08324 & 1.05998 & 1.97756 & 1.64561 & 2.18125 \\
1.14527 & 0.56645 & 1.41598 & 1.16961 & 1.67226 \\
1.21483 & 0.01296 & 0.78616 & 0.63578 & 1.10143 \\
1.29339 & -0.61212 & 0.07489 & 0.03291 & 0.45677 \\
1.38282 & -1.32364 & -0.73474 & -0.65332 & -0.27703 \\
\(450-50{ }^{\circ} \mathrm{C}\) & & & & \\
1.38282 & -1.70731 & -0.96399 & -0.49543 & -0.15327 \\
1.48553 & -2.06466 & -1.40883 & -1.02402 & -0.70712 \\
1.60472 & -2.47936 & -1.92506 & -1.63742 & -1.34984 \\
1.74471 & -2.96641 & -2.53135 & -2.35786 & -2.10469 \\
1.91146 & -3.54655 & -3.25353 & -3.21599 & -3.00384 \\
2.11345 & -4.24931 & -4.12835 & -4.25550 & -4.09301 \\
2.36317 & -5.11814 & -5.20989 & -5.54065 & -5.43957 \\
2.67982 & -6.21980 & -6.58128 & -7.17020 & -7.14699 \\
3.09444 & -7.66237 & -8.37702 & -9.30401 & -9.38275
\end{tabular}
\begin{tabular}{|c|c|c|c|c|}
\hline \multicolumn{5}{|l|}{Chen-Nuttall method} \\
\hline 1/T K \({ }^{-1}\) & \(25^{\circ} \mathrm{C} / \mathrm{min}\) & \(50^{\circ} \mathrm{C} / \mathrm{min}\) & \(100^{\circ} \mathrm{C} / \mathrm{min}\) & \(150^{\circ} \mathrm{C} / \mathrm{min}\) \\
\hline \multicolumn{5}{|l|}{\(800-450{ }^{\circ} \mathrm{C}\)} \\
\hline 0.93183 & 2.50061 & 3.55706 & 3.05557 & 3.65679 \\
\hline 0.97736 & 2.13262 & 3.13970 & 2.69990 & 3.27720 \\
\hline 1.02758 & 1.72683 & 2.67945 & 2.30768 & 2.85862 \\
\hline 1.08324 & 1.27707 & 2.16935 & 1.87297 & 2.39469 \\
\hline 1.14527 & 0.77580 & 1.60083 & 1.38848 & 1.87763 \\
\hline 1.21483 & 0.21364 & 0.96325 & 0.84513 & 1.29775 \\
\hline 1.29339 & -0.42123 & 0.24320 & 0.23150 & 0.64287 \\
\hline 1.38282 & -1.14389 & -0.57642 & -0.46699 & -0.10256 \\
\hline \multicolumn{5}{|l|}{\(450-50^{\circ} \mathrm{C}\)} \\
\hline 1.38282 & -1.34581 & -0.66351 & -0.23656 & 0.09500 \\
\hline 1.48553 & -1.71605 & -1.11999 & -0.77606 & -0.46940 \\
\hline 1.60472 & -2.14571 & -1.64973 & -1.40214 & -1.12436 \\
\hline 1.74471 & -2.65033 & -2.27188 & -2.13745 & -1.89360 \\
\hline 1.91146 & -3.25141 & -3.01297 & -3.01331 & -2.80988 \\
\hline 2.11345 & -3.97952 & -3.91067 & -4.07428 & -3.91980 \\
\hline 2.36317 & -4.87969 & -5.02052 & -5.38598 & -5.29202 \\
\hline 2.67982 & -6.02110 & -6.42779 & -7.04918 & -7.03197 \\
\hline 3.09444 & -7.51571 & -8.27053 & -9.22706 & -9.31034 \\
\hline
\end{tabular}
\begin{tabular}{|c|c|c|c|c|}
\hline \multicolumn{5}{|l|}{Friedman method} \\
\hline I/T K \({ }^{-1}\) & \(25^{\circ} \mathrm{C} / \mathrm{min}\) & \(50^{\circ} \mathrm{C} / \mathrm{min}\) & \(100^{\circ} \mathrm{C} / \mathrm{min}\) & \(150{ }^{\circ} \mathrm{C} / \mathrm{min}\) \\
\hline \multicolumn{5}{|l|}{\(800-450{ }^{\circ} \mathrm{C}\)} \\
\hline 0.93183 & 3.79334 & 4.93696 & 3.27364 & 3.68381 \\
\hline 0.97736 & 3.23737 & 4.32950 & 2.84601 & 3.26343 \\
\hline 1.02758 & 2.62428 & 3.65962 & 2.37445 & 2.79985 \\
\hline 1.08324 & 1.94477 & 2.91718 & 1.85180 & 2.28605 \\
\hline 1.14527 & 1.18745 & 2.08971 & 1.26929 & 1.71341 \\
\hline 1.21483 & 0.33812 & 1.16171 & 0.61602 & 1.07120 \\
\hline 1.29339 & -0.62106 & 0.11369 & -0.12174 & 0.34593 \\
\hline 1.38282 & -1.71288 & -1.07925 & -0.96153 & -0.47963 \\
\hline \multicolumn{5}{|l|}{\(450-50{ }^{\circ} \mathrm{C}\)} \\
\hline 1.38282 & -1.75058 & -1.08406 & -0.64897 & -0.24873 \\
\hline 1.48553 & -2.02776 & -1.41413 & -1.03911 & -0.67674 \\
\hline 1.60472 & -2.34942 & -1.79717 & -1.49186 & -1.17343 \\
\hline 1.74471 & -2.72720 & -2.24704 & -2.02360 & -1.75679 \\
\hline 1.91146 & -3.17719 & -2.78291 & -2.65698 & -2.45165 \\
\hline 2.11345 & -3.72228 & -3.43202 & -3.42422 & -3.29336 \\
\hline 2.36317 & -4.39619 & -4.23453 & -4.37277 & -4.33399 \\
\hline 2.67982 & -5.25070 & -5.25210 & -5.57551 & -5.65348 \\
\hline 3.09444 & -6.36962 & -6.58455 & -7.15044 & -7.38128 \\
\hline
\end{tabular}

\section*{Compensation Effect of Kinetic Parameters Derived other 2-Stage Methods}

The logarithms of reaction rate constants, calculated with the kinetic parameters derived from the 2-stage Coats-Redfern method, were plotted in Figures G. 1 and G. 4 for CANMET and Syncrude pitch respectively. The logarithms of reaction rate constants, calculated with the kinetic parameters derived from the 2-stage Chen-Nuttall method, were plotted in Figures G. 2 and G. 5 for CANMET and Syncrude pitch respectively. The logarithms of reaction rate constants, calculated with the kinetic parameters derived from the 2 -stage Friedman method, were plotted in Figures G. 3 and G. 6 for CANMET and Syncrude pitch respectively. Examination of these graphs reveals that the second criterion of compensation effect is not met.


Figure G. 1 CANMET pitch pyrolysis kinetic reaction rate as a function of temperature at different heating rates and final temperature \(800^{\circ} \mathrm{C}\) with 2-stage Coats-Redfern method


Figure G. 2 CANMET pitch pyrolysis kinetic reaction rate as a function of temperature at different heating rates and final temperature \(800^{\circ} \mathrm{C}\) with 2-stage Chen-Nuttall method


Figure G. 3 CANMET pitch pyrolysis kinetic reaction rate as a function of temperature at different heating rates and final temperature \(800^{\circ} \mathrm{C}\) with 2-stage Friedman method


Figure G. 4 Syncrude pitch pyrolysis kinetic reaction rate as a function of temperature at different heating rates and final temperature \(800^{\circ} \mathrm{C}\) with 2-stage Coats-Redfern method


Figure G. 5 Syncrude pitch pyrolysis kinetic reaction rate as a function of temperature at different heating rates and final temperature \(800^{\circ} \mathrm{C}\) with 2-stage Chen-Nuttall method


Figure G. 6 Syncrude pitch pyrolysis kinetic reaction rate as a function of temperature at different heating rates and final temperature \(800^{\circ} \mathrm{C}\) with 2-stage Friedman method

APPENDIX H Volatile Yield Predicted via the Single Set Kinetic Parameters for Different Heating Rates
\(800^{\circ} \mathrm{C}\) CANMET pitch
\begin{tabular}{|c|c|c|c|c|c|c|c|c|c|c|c|}
\hline \multirow[t]{2}{*}{\[
\begin{aligned}
& 25^{\circ} \mathrm{C} / \mathrm{min} \\
& \mathrm{t}_{\text {min }}
\end{aligned}
\]} & \multirow[b]{2}{*}{Vexp} & \multirow[b]{2}{*}{Vmod} & \multicolumn{2}{|l|}{\(50^{\circ} \mathrm{C} / \mathrm{min}\)} & \multicolumn{3}{|c|}{\(100{ }^{\circ} \mathrm{C} / \mathrm{min}\)} & \multicolumn{4}{|c|}{\(150{ }^{\circ} \mathrm{C} / \mathrm{min}\)} \\
\hline & & & t min & Vexp & Vmod & \(t\) min & Vexp & Vmod & \(t \mathrm{~min}\) & Vexp & Vmod \\
\hline 0.01 & 0.04 & 0.00 & 0.00 & 0.32 & 0.00 & 0.01 & 0.09 & 0.00 & 0.01 & 0.06 & 0.00 \\
\hline 2.01 & 0.06 & 0.04 & 0.98 & 0.31 & 0.02 & 0.99 & 0.25 & 0.05 & 1.03 & 0.00 & 0.15 \\
\hline 3.42 & 0.50 & 0.13 & 1.52 & 0.40 & 0.05 & 1.26 & 0.57 & 0.11 & 1.65 & 0.78 & 1.09 \\
\hline 4.15 & 1.02 & 0.24 & 2.05 & 1.06 & 0.12 & 1.49 & 1.25 & 0.21 & 2.35 & 2.85 & 5.25 \\
\hline 5.02 & 2.17 & 0.45 & 2.50 & 2.01 & 0.22 & 1.77 & 2.41 & 0.40 & 2.54 & 7.05 & 7.45 \\
\hline 6.02 & 3.86 & 0.88 & 2.99 & 3.37 & 0.43 & 2.00 & 3.67 & 0.67 & 2.74 & 10.84 & 10.68 \\
\hline 7.02 & 5.57 & 1.60 & 3.52 & 4.96 & 0.81 & 2.27 & 5.33 & 1.16 & 2.89 & 15.87 & 16.75 \\
\hline 8.03 & 7.59 & 2.75 & 4.01 & 6.64 & 1.38 & 2.51 & 7.10 & 1.77 & 3.01 & 21.77 & 22.70 \\
\hline 9.03 & 9.83 & 4.49 & 4.50 & 8.42 & 2.25 & 2.78 & 9.65 & 2.78 & 3.13 & 27.11 & 29.85 \\
\hline 10.03 & 12.08 & 7.00 & 4.99 & 10.43 & 3.50 & 3.01 & 12.14 & 3.97 & 3.21 & 33.84 & 35.18 \\
\hline 11.03 & 14.48 & 10.45 & 5.52 & 12.62 & 5.43 & 3.25 & 14.86 & 5.52 & 3.28 & 39.52 & 40.82 \\
\hline 12.03 & 17.33 & 14.97 & 6.01 & 14.76 & 7.84 & 3.52 & 18.31 & 7.85 & 3.36 & 46.40 & 46.53 \\
\hline 12.50 & 18.70 & 17.47 & 6.50 & 17.41 & 10.96 & 3.79 & 22.54 & 10.83 & 3.44 & 53.90 & 52.33 \\
\hline 13.04 & 20.65 & 20.63 & 6.99 & 20.81 & 14.84 & 4.02 & 27.11 & 13.38 & 3.52 & 60.98 & 57.70 \\
\hline 13.50 & 22.55 & 23.64 & 7.30 & 23.70 & 17.73 & 4.22 & 32.19 & 19.52 & 3.63 & 66.75 & 64.88 \\
\hline 14.04 & 25.09 & 27.36 & 7.53 & 26.51 & 19.99 & 4.37 & 37.20 & 25.62 & 3.75 & 72.41 & 70.40 \\
\hline 14.57 & 27.59 & 31.32 & 7.75 & 29.70 & 22.40 & 4.49 & 41.82 & 30.90 & 3.83 & 75.11 & 73.05 \\
\hline 15.04 & 30.75 & 34.95 & 7.97 & 33.18 & 24.96 & 4.61 & 47.51 & 36.67 & 4.02 & 75.95 & 76.57 \\
\hline 15.51 & 34.04 & 38.69 & 8.15 & 36.47 & 27.39 & 4.69 & 51.88 & 40.71 & 4.49 & 76.82 & 77.59 \\
\hline 15.91 & 37.35 & 41.95 & 8.24 & 38.20 & 29.60 & 4.76 & 56.71 & 44.85 & 5.00 & 77.59 & 77.59 \\
\hline 16.24 & 40.31 & 44.60 & 8.42 & 42.01 & 34.30 & 4.84 & 61.61 & 49.01 & & & \\
\hline 16.58 & 43.67 & 49.98 & 8.60 & 46.10 & 39.31 & 4.92 & 66.12 & 53.12 & & & \\
\hline 16.98 & 48.26 & 56.32 & 8.82 & 52.05 & 45.84 & 5.00 & 69.67 & 57.04 & & & \\
\hline 17.38 & 53.52 & 62.24 & 9.04 & 58.55 & 52.43 & 5.15 & 74.33 & 64.27 & & & \\
\hline 17.71 & 58.43 & 66.67 & 9.35 & 67.70 & 61.14 & 5.35 & 77.00 & 71.35 & & & \\
\hline 18.00 & 63.51 & 70.01 & 9.66 & 74.02 & 68.59 & 5.62 & 78.28 & 76.99 & & & \\
\hline 18.31 & 67.42 & 73.09 & 9.89 & 76.48 & . 72.78 & 6.01 & 78.82 & 79.15 & & & \\
\hline 18.65 & 71.52 & 75.69 & 10.11 & 77.95 & 75.92 & 6.52 & 78.97 & 79.30 & & & \\
\hline 19.05 & 74.86 & 77.93 & 10.64 & - 79.28 & 79.80 & 7.02 & 79.15 & 79.30 & & & \\
\hline 19.52 & 77.16 & 79.52 & 11.49 & 79.95 & 80.77 & 7.49 & 79.30 & 79.30 & & & \\
\hline 20.05 & 78.62 & 80.40 & 12.78 & 80.15 & 80.79 & & & & & & \\
\hline 21.05 & 79.70 & 80.81 & 15.00 & 80.79 & 80.79 & & & & & & \\
\hline 22.05 & 80.03 & 80.84 & & & & & & & & & \\
\hline 23.99 & 80.29 & 80.84 & & & & & & & & & \\
\hline 26.00 & 80.52 & 80.84 & & & & & & & & & \\
\hline 28.00 & 80.72 & 80.84 & & & & & & & & & \\
\hline 30.00 & 80.84 & 80.84 & & & & & & & & & \\
\hline
\end{tabular}

Syncrude pitch \(800^{\circ} \mathrm{C}\)
\begin{tabular}{|c|c|c|c|c|c|c|c|c|c|c|c|}
\hline \[
25^{\circ} \mathrm{C} / \mathrm{min}
\] & & & \(50^{\circ} \mathrm{C} / \mathrm{min}\) & & & \(100^{\circ}\) & & & \(150{ }^{\circ}\) & & \\
\hline \(t \mathrm{~min}\) & Vexp & Vmod & t min & Vexp & Vmod & \(t\) min & Vexp & Vmod & \(t\) min & Vexp & Vmod \\
\hline 0.00 & 0.10 & 0.00 & 0.03 & 0.14 & 0.00 & 0.00 & 0.01 & 0.00 & 0.00 & 0.02 & 0.00 \\
\hline 1.00 & 0.15 & 0.02 & 1.00 & 0.19 & 0.03 & 0.51 & 0.00 & 0.02 & 0.51 & 0.07 & 0.03 \\
\hline 2.01 & 0.19 & 0.06 & 2.29 & 0.48 & 0.37 & 1.02 & 0.08 & 0.12 & 0.99 & 0.32 & 0.35 \\
\hline 3.01 & 0.60 & 0.18 & 3.03 & 1.52 & 1.12 & 1.50 & 0.42 & 0.54 & 1.20 & 0.97 & 0.81 \\
\hline 4.01 & 1.00 & 0.46 & 3.53 & 2.76 & 2.20 & 1.77 & 1.12 & 1.12 & 1.35 & 1.63 & 1.40 \\
\hline 5.01 & 1.86 & 1.04 & 4.09 & 5.23 & 4.26 & 2.01 & 2.05 & 1.98 & 1.51 & 2.85 & 2.32 \\
\hline 6.02 & 3.27 & 2.18 & 4.50 & 7.85 & 6.67 & 2.25 & 3.44 & 3.36 & 1.66 & 4.82 & 3.67 \\
\hline 7.02 & 5.63 & 4.23 & 5.01 & 11.99 & 10.94 & 2.45 & 5.33 & 5.08 & 1.81 & 7.26 & 5.61 \\
\hline 8.02 & 8.60 & 7.67 & 5.52 & 17.06 & 17.02 & 2.62 & 7.45 & 7.00 & 1.96 & 10.66 & 8.27 \\
\hline 9.02 & 12.65 & 13.01 & 6.02 & 22.76 & 25.09 & 2.79 & 9.93 & 9.45 & 2.08 & 13.84 & 11.04 \\
\hline 10.02 & 17.69 & 20.71 & 6.48 & 28.37 & 34.08 & 2.93 & 12.16 & 11.84 & 2.20 & 17.41 & 14.41 \\
\hline 11.03 & 23.58 & 30.93 & 6.94 & 34.00 & 44.25 & 3.06 & 14.63 & 14.64 & 2.32 & 21.39 & 18.45 \\
\hline 12.03 & 29.64 & 43.21 & 7.40 & 39.55 & 54.88 & 3.20 & 17.29 & 17.88 & 2.45 & 25.52 & 23.21 \\
\hline 13.03 & 35.84 & 56.42 & 7.91 & 45.63 & 66.08 & 3.33 & 20.21 & 21.57 & 2.54 & 28.72 & 27.14 \\
\hline 14.03 & 41.51 & 68.85 & 8.23 & 49.93 & 45.92 & 3.47 & 23.23 & 25.69 & 2.63 & 32.08 & 31.45 \\
\hline 15.04 & 46.95 & 78.83 & 8.60 & 56.23 & 56.23 & 3.64 & 26.99 & 31.42 & 2.72 & 35.38 & 17.86 \\
\hline 16.04 & 52.40 & 62.72 & 8.83 & 61.49 & 62.52 & 3.78 & 30.22 & 36.37 & 2.84 & 39.83 & 23.56 \\
\hline 16.51 & 55.44 & 69.40 & 9.02 & 66.50 & 67.33 & 3.95 & 34.10 & 42.97 & 2.93 & 43.35 & 28.44 \\
\hline 17.04 & 59.57 & 76.10 & 9.16 & 70.62 & 70.70 & 4.08 & 37.28 & 25.56 & 3.02 & 47.39 & 33.91 \\
\hline 17.51 & 63.81 & 80.89 & 9.39 & 77.79 & 75.77 & 4.22 & 40.44 & 31.05 & 3.08 & 50.52 & 37.85 \\
\hline 17.98 & 68.73 & 84.61 & 9.62 & 83.51 & 80.08 & 4.32 & 42.92 & 35:55 & 3.14 & 54.04 & 41.96 \\
\hline 18.44 & 74.31 & 87.27 & 9.80 & 86.62 & 82.90 & 4.46 & 46.34 & 41.98 & 3.20 & 58.28 & 46.13 \\
\hline 18.78 & 78.47 & 88.59 & 10.08 & 88.85 & 86.12 & 4.56 & 49.26 & 47.01 & 3.26 & 63.46 & 50.44 \\
\hline 19.05 & 81.64 & 89.37 & 10.49 & 89.66 & 88.98 & 4.66 & 52.57 & 52.14 & 3.32 & 69.32 & 54.69 \\
\hline
\end{tabular}
\begin{tabular}{|c|c|c|c|c|c|c|c|c|c|c|c|}
\hline 19.25 & 84.14 & 89.81 & 11.00 & 89.98 & 90.34 & 4.76 & 56.47 & 57.21 & 3.39 & 75.11 & 58.97 \\
\hline 19.51 & 86.99 & 90.25 & 12.01 & 90.22 & 90.70 & 4.86 & 61.34 & 62.21 & 3.45 & 80.17 & 63.13 \\
\hline 19.85 & 88.85 & 90.60 & 12.98 & 90.40 & 90.70 & 4.96 & 67.07 & 66.98 & 3.54 & 85.60 & 68.98 \\
\hline 20.25 & 89.87 & 90.84 & 13.99 & 90.57 & 90.70 & 5.07 & 73.25 & 71.41 & 3.63 & 88.35 & 74.24 \\
\hline 20.98 & 90.46 & 91.00 & 15.01 & 90.70 & 90.70 & 5.17 & 79.27 & 75.41 & 3.75 & 89.59 & 80.05 \\
\hline 21.99 & 90.56 & 91.03 & & & & 5.27 & 84.20 & 78.91 & 3.93 & 89.97 & 86.01 \\
\hline 22.99 & 90.58 & 91.03 & & & & 5.41 & 88.02 & 82.76 & 4.26 & 90.30 & 90.11 \\
\hline 24.06 & 90.75 & 91.03 & & & & 5.64 & 89.67 & 87.26 & 4.99 & 90.62 & 90.62 \\
\hline 24.99 & 90.80 & 91.03 & & & & 5.95 & 90.02 & 89.80 & & & \\
\hline 26.00 & 90.95 & 91.03 & & & & 6.49 & 90.27 & 90.56 & & & \\
\hline 27.00 & 90.90 & 91.03 & & & & 7.51 & 90.58 & 90.58 & & & \\
\hline 28.00 & 90.89 & 91.03 & & & & & & & & & \\
\hline 29.00 & 91.04 & 91.03 & & & & & & & & & \\
\hline 30.00 & 91.03 & 91.03 & & & & & & & & & \\
\hline
\end{tabular}

Volatile Yield Predicted via the Single Set Kinetic Parameters for Different Final Temperature
\begin{tabular}{|c|c|c|c|c|c|c|c|c|}
\hline \multicolumn{3}{|l|}{\(100^{\circ} \mathrm{C} / \mathrm{min}\) CANMET pitch} & \multirow[b]{2}{*}{\(850{ }^{\circ} \mathrm{C}\)} & \multirow[b]{2}{*}{\(V_{\text {ex }} \%\)} & \multirow[b]{2}{*}{\(\mathrm{V}_{\text {mod }}\) \%} & \multirow[b]{2}{*}{\(950{ }^{\circ} \mathrm{C}\)} & \multirow[b]{2}{*}{\(\mathrm{V}_{\text {tep }} \%\)} & \multirow[b]{2}{*}{\(\mathrm{V}_{\text {mod }} \%\)} \\
\hline \(750{ }^{\circ} \mathrm{C}\) & \(\mathrm{V}_{\text {ep }}\) \% & \(\mathrm{V}_{\text {mod }}\) \% & & & & & & \\
\hline 50.6 & 0.16 & 0.00 & 49.3 & 0.00 & 0.00 & 51.1 & 0.29 & 0.00 \\
\hline 151.8 & 0.34 & 0.06 & 150.6 & 0.00 & 0.05 & 152.2 & 0.53 & 0.06 \\
\hline 202.4 & 1.04 & 0.23 & 197.4 & 0.50 & 0.20 & 175.6 & 1.45 & 0.11 \\
\hline 225.8 & 1.90 & 0.40 & 232.5 & 1.35 & 0.46 & 202.8 & 2.55 & 0.24 \\
\hline 249.2 & 2.85 & 0.67 & 252.0 & 2.26 & 0.70 & 222.2 & 3.79 & 0.38 \\
\hline 276.4 & 4.37 & 1.15 & 291.0 & 4.26 & 1.49 & 249.5 & 5.43 & 0.68 \\
\hline 299.8 & 6.02 & 1.75 & 322.1 & 6.55 & 2.53 & 276.7 & 7.28 & 1.18 \\
\hline 327.0 & 8.47 & 2.76 & 349.4 & 9.00 & 3.86 & 300.1 & 9.17 & 1.80 \\
\hline 350.4 & 10.95 & 3.95 & 376.7 & 11.98 & 5.67 & 327.3 & 11.47 & 2.83 \\
\hline 377.6 & 14.05 & 5.78 & 400.1 & 14.61 & 7.67 & 350.6 & 13.82 & 4.04 \\
\hline 401.0 & 17.09 & 7.81 & 423.5 & 17.82 & 10.14 & 377.9 & 16.81 & 5.92 \\
\hline 424.3 & 20.48 & 10.30 & 446.9 & 21.57 & 13.10 & 401.2 & 19.67 & 7.99 \\
\hline 451.6 & 25.64 & 13.22 & 466.3 & 25.85 & 17.52 & 424.6 & 23.09 & 10.55 \\
\hline 474.9 & 31.58 & 20.70 & 481.9 & 29.96 & 23.25 & 447.9 & 27.46 & 13.61 \\
\hline 490.5 & 36.67 & 27.07 & 497.5 & 34.59 & 30.04 & 467.4 & 32.51 & 18.39 \\
\hline 502.2 & 41.70 & 32.51 & 509.2 & 39.07 & 35.73 & 482.9 & 37.44 & 24.32 \\
\hline 510.0 & 45.81 & . 36.40 & 520.9 & 44.43 & 41.77 & 494.6 & 41.67 & 29.51 \\
\hline 521.7 & 53.18 & 42.51 & 532.6 & 50.68 & 47.98 & 506.3 & 46.50 & 35.24 \\
\hline 525.6 & 55.86 & 44.59 & 540.4 & 55.22 & 52.09 & 517.9 & 52.13 & 41.33 \\
\hline 533.3 & 61.28 & 48.72 & 548.2 & 59.87 & 56.08 & 525.7 & 56.46 & 45.56 \\
\hline 541.1 & 66.20 & 52.85 & 556.0 & 64.33 & 59.87 & 533.5 & 60.85 & 49.82 \\
\hline 552.8 & 71.47 & 58.78 & 563.8 & 68.12 & 63.40 & 541.3 & 65.12 & 54.04 \\
\hline 568.4 & 75.54 & 65.81 & 571.6 & 71.21 & 66.58 & 545.2 & 67.09 & 56.10 \\
\hline 583.9 & 77.50 & 71.34 & 591.0 & 75.40 & 72.79 & 552.9 & 70.59 & 60.04 \\
\hline 607.3 & 78.67 & 76.62 & 610.6 & 77.19 & 76.52 & 564.6 & 74.52 & 65.53 \\
\hline 626.8 & 78.88 & 78.64 & 649.5 & 78.02 & 78.84 & 580.2 & 77.73 & 71.61 \\
\hline 650.1 & 79.06 & 79.44 & 700.2 & 78.38 & 79.01 & 595.7 & 79.05 & 75.98 \\
\hline 700.7 & 79.40 & 79.60 & 750.8 & 78.58 & 79.01 & 623.0 & 80.06 & 79.98 \\
\hline 750.0 & 79.60 & 79.60 & 801.5 & 78.74 & 79.01 & 650.2 & 80.33 & 81.07 \\
\hline & & & 850.8 & 79.01 & 79.01 & 751.4 & 80.77 & 81.23 \\
\hline & & & & & & 852.5 & 81.07 & 81.23 \\
\hline & & & & & & 949.8 & 81.23 & 81.23 \\
\hline
\end{tabular}
\(50^{\circ} \mathrm{C} / \mathrm{min}\) Syncrude pitch
\begin{tabular}{|c|c|c|c|c|c|c|c|c|}
\hline \(750{ }^{\circ} \mathrm{C}\) & \(\mathrm{V}_{\text {ep }} \%\) & \(\mathrm{V}_{\text {mad }} \%\) & \(850{ }^{\circ} \mathrm{C}\) & \(\mathrm{V}_{\text {ap }} \%\) & \(\mathrm{V}_{\text {mad }} \%\) & \(950{ }^{\circ} \mathrm{C}\) & \(\mathrm{V}_{\text {¢ }} \%\) & \(\mathrm{V}_{\text {mod }} \%\) \\
\hline 50.4 & 0.03 & 0.00 & 51.6 & 0.06 & 0.00 & 51.1 & 0.21 & \\
\hline 99.9 & 0.10 & 0.03 & 102.2 & 0.02 & 0.03 & 99.7 & 0.29 & 0.03 \\
\hline 158.1 & 0.56 & 0.30 & 164.4 & 0.48 & 0.37 & 148.3 & 0.56 & 0.21 \\
\hline 190.4 & 1.46 & 0.82 & 201.3 & 1.36 & 1.12 & 182.4 & 1.52 & 0.65 \\
\hline 224.9 & 3.35 & 2.11 & 228.9 & 2.80 & 2.32 & 206.7 & 2.71 & 1.31 \\
\hline 250.8 & 5.88 & 3.94 & 252.0 & 4.80 & 4.03 & 233.5 & 5.31 & 2.61 \\
\hline 276.6 & 9.79 & 6.90 & 277.3 & 7.88 & 6.97 & 257.8 & 8.42 & 4.61 \\
\hline 300.3 & 14.02 & 10.95 & 300.4 & 11.70 & 10.92 & 277.2 & 11.62 & 6.99 \\
\hline 326.2 & 19.16 & 17.19 & 325.7 & 16.76 & 16.99 & 301.5 & 16.41 & 11.20 \\
\hline 349.9 & 24.83 & 24.73 & 348.7 & 21.89 & 24.21 & 325.8 & 22.06 & 17.09 \\
\hline 375.8 & 30.94 & 34.85 & 374.1 & 28.07 & 34.00 & 350.2 & 28.00 & 24.83 \\
\hline 399.5 & 36.59 & 45.41 & 399.4 & 34.44 & 45.21 & 376.9 & 34.79 & 35.36 \\
\hline 425.3 & 42.40 & 57.38 & 424.8 & 40.76 & 56.91 & 401.2 & 40.74 & 46.24 \\
\hline 449.0 & 47.91 & 67.69 & 447.8 & 46.75 & 66.92 & 425.5 & 46.39 & 57.50 \\
\hline 462.0 & 51.12 & 46.22 & 466.2 & 52.50 & 48.40 & 447.4 & 52.04 & 67.07 \\
\hline 474.9 & 54.96 & 53.45 & 482.3 & 59.46 & 57.39 & 464.5 & 57.61 & 47.64 \\
\hline 483.5 & 58.00 & 58.28 & 493.8 & 65.60 & 63.63 & 476.6 & 62.93 & 54.46 \\
\hline 492.0 & 61.50 & 62.92 & 503.1 & 71.20 & 68.36 & 486.3 & 67.86 & 59.86 \\
\hline
\end{tabular}
\begin{tabular}{|c|c|c|c|c|c|c|c|c|}
\hline 498.6 & 64.42 & 66.35 & 514.6 & 78.24 & 73.71 & 496.1 & 73.42 & 65.09 \\
\hline 507.2 & 69.23 & 70.64 & 526.1 & 83.85 & 78.33 & 505.8 & 78.51 & 69.98 \\
\hline 515.8 & 74.30 & 74.54 & 537.6 & 87.32 & 82.14 & 513.1 & 82.10 & 73.38 \\
\hline 526.6 & 80.25 & 78.82 & 549.1 & 88.80 & 85.07 & 520.4 & 85.05 & 76.48 \\
\hline 535.2 & 84.12 & 81.72 & 567.6 & 89.49 & 88.16 & 530.1 & 87.68 & 80.10 \\
\hline 546.0 & 87.41 & 84.68 & 599.8 & 89.78 & 90.24 & 542.3 & 88.98 & 83.79 \\
\hline 561.1 & 89.45 & 87.61 & 650.5 & 90.03 & 90.61 & 556.8 & 89.49 & 86.97 \\
\hline 574.0 & 89.93 & 89.19 & 701.2 & 90.23 & 90.61 & 600.6 & 90.17 & 90.66 \\
\hline 599.8 & 90.38 & 90.59 & 749.5 & 90.39 & 90.61 & 651.7 & 90.27 & 91.01 \\
\hline 651.6 & 90.53 & 90.96 & 800.2 & 90.50 & 90.61 & 700.3 & 90.65 & 91.01 \\
\hline 750.7 & 90.96 & 90.96 & 850.9 & 90.61 & 90.61 & 751.4 & 90.88 & 91.01 \\
\hline & & & & & & 800.0 & 90.75 & 91.01 \\
\hline & & & & & & 851.1 & 90.97 & 91.01 \\
\hline & & & & & & 897.3 & 90.89 & 91.01 \\
\hline & & & & & & 952.9 & 91.01 & 91.01 \\
\hline
\end{tabular}

\section*{APPENDIX I The Effect of the Number of Significant Digits and Sample Weight Analysis}

The effect of number of significant digits and change of reaction rate constant \(k \pm 1 \%\) and \(\pm 2 \%\) was checked. The following temperature T is in OC and the volatile content V is in \(\%\) of the original sample weight. Each symbol is defined in the FORTRAN program. Run Can 48 is fitted in the following results.

The results show that a change of the number of significant digits from 5 to 2 caused \(0.0052 \%, 0.005239 \%, 0.5548 \%, 1.742 \%, 26.66 \%\) of s.e.e. but made no noticeable effect on the fitting of the volatile content vs temperature curve, as shown in Figure I.1.

The results also show that a change of k from the best fitting values caused \(3.629 \%\), \(1.053 \%, 0.5186 \%, 2.496 \%\) for \(-2 \%,-1 \%,+1,+2 \%\) change of \(k\), but the made no noticeable effect on the fitting of the volatiles vs temperature curve, as shown in Figure I.1.

Fitting results in the entire temperature range with the change of significant digits
\begin{tabular}{rrrcrrr}
T & V & V & V 5 & V 4 & V & V 2 \\
50.22 & .040000 & .155612 & .155925 & .155187 & .148259 \\
100.32 & .060000 & .586071 & .587093 & .584630 & .561560 \\
135.40 & .500000 & 1.232127 & 1.234082 & 1.229293 & 1.184507 \\
153.77 & 1.020000 & 1.734067 & 1.736689 & 1.730211 & 1.669683 \\
175.47 & 2.170000 & 2.505345 & 2.508926 & 2.499990 & 2.416585 \\
200.52 & 3.860000 & 3.672206 & 3.677123 & 3.664705 & 3.548955 \\
225.60 & 5.570000 & 5.173826 & 5.180311 & 5.163737 & 5.009436 \\
250.65 & 7.590000 & 7.033928 & 7.042167 & 7.020847 & 6.822622 \\
275.70 & 9.830000 & 9.266864 & 9.276985 & 9.250461 & 9.004178 \\
300.75 & 12.080000 & 11.868485 & 11.880535 & 11.848537 & 11.551843 \\
325.80 & 14.480000 & 14.816680 & 14.830612 & 14.793103 & 14.445798 \\
350.85 & 17.330000 & 18.071870 & 18.087541 & 18.044737 & 17.648988 \\
362.55 & 18.700000 & 19.682439 & 19.698846 & 19.653715 & 19.236753 \\
375.90 & 20.650000 & 21.579050 & 21.596222 & 21.548593 & 21.108932 \\
387.60 & 22.550000 & 23.284918 & 23.302686 & 23.253032 & 22.795035 \\
400.95 & 25.090000 & 25.271160 & 25.289514 & 25.237764 & 24.760882 \\
414.32 & 27.590000 & 27.291778 & 27.310609 & 27.257016 & 26.763636 \\
426.00 & 30.750000 & 29.073425 & 29.092580 & 29.037604 & 28.531953 \\
437.70 & 34.040000 & 30.865168 & 30.884554 & 30.828418 & 30.312592 \\
447.72 & 37.350000 & 32.399281 & 32.418790 & 32.361847 & 31.839053 \\
456.07 & 40.310000 & 41.063754 & 41.020945 & 40.747694 & 42.488916 \\
464.42 & 43.670000 & 45.483339 & 45.439130 & 45.158472 & 46.936603 \\
474.45 & 48.260000 & 50.812486 & 50.767717 & 50.485304 & 52.261405 \\
484.45 & 53.520000 & 56.001230 & 55.957289 & 55.681786 & 57.400808 \\
492.80 & 58.430000 & 60.123230 & 60.081089 & 59.818183 & 61.447106 \\
500.00 & 63.510000 & 63.451378 & 63.411572 & 63.164263 & 64.686734 \\
507.85 & 67.420000 & 66.780378 & 66.743855 & 66.517930 & 67.898483 \\
516.20 & 71.520000 & 69.923750 & 69.891393 & 69.692140 & 70.899546 \\
526.22 & 74.860000 & 73.102690 & 73.075913 & 72.911869 & 73.895247 \\
537.90 & 77.160000 & 75.972522 & 75.952435 & 75.830062 & 76.553672 \\
551.27 & 78.620000 & 78.231054 & 78.217958 & 78.138651 & 78.599589 \\
576.32 & 79.700000 & 80.249473 & 80.245260 & 80.219989 & 80.361467 \\
601.37 & 80.030000 & 80.759543 & 80.758742 & 80.753972 & 80.779498 \\
649.80 & 80.290000 & 80.839754 & 80.839749 & 80.839723 & 80.839849 \\
699.90 & 80.520000 & 80.840000 & 80.840000 & 80.840000 & 80.840000 \\
750.00 & 80.720000 & 80.840000 & 80.840000 & 80.840000 & 80.840000 \\
800.12 & 80.840000 & 80.840000 & 80.840000 & 80.840000 & 80.840000
\end{tabular}
standard deviation for each method above
the s.e.e. relative change in \(\%\) with \(5,4,3,2\) digits .005239 . 5548011.74241626 .663709
\begin{tabular}{|c|c|c|c|c|c|c|}
\hline \multicolumn{7}{|l|}{Fitting results in the entire temperature range with the change of \(k\)} \\
\hline & & VM2 & & & & \\
\hline 0.22 & 0000 & 152449 & 154003 & 155557 & . 157111 & . 158665 \\
\hline 100.3 & . 060000 & 574217 & 580055 & 585893 & 591731 & 597568 \\
\hline 135.40 & 500000 & 1.207336 & 1.219562 & 1.231787 & 1.244009 & 1.256230 \\
\hline 153.77 & 1.020000 & 1.699306 & 1.716460 & 1.733611 & 1.750758 & 1.767901 \\
\hline 175.47 & 2.170000 & . 455396 & 2.480063 & 2.504722 & 2.529373 & 2.554017 \\
\hline 200.52 & 3.860000 & 3.599583 & 3.63547 & 3.671350 & 720 & 3.743050 \\
\hline 225 & 70000 & 5.072560 & 5.12264 & 5. & 5.222717 & 5.272703 \\
\hline 250 & 590 & 6.898025 & 6529 & 7.032493 & 7.099636 & 7.166718 \\
\hline 275.70 & 83000 & 9.090637 & 9.177923 & 9.265102 & 352175 & 3 \\
\hline 300.75 & 12.080000 & 11.647051 & 11.756806 & 11.866387 & 11.975795 & 12.085028 \\
\hline 325.80 & 14.480000 & 14.546404 & 14.680464 & 14.814254 & 14.947 & 15.081021 \\
\hline 350.85 & 17.330000 & 17.750 & 17.910142 & 18.069141 & 18.227 & 36 \\
\hline 362 & 18 & 19 & 19.50872 & 19.679582 & 19 & 20.019872 \\
\hline 375.90 & 20.650 & 21.206920 & 21.39177 & 21.576059 & 21.7597 & 21.942914 \\
\hline 387.60 & 22.550000 & 22.889470 & 23.085980 & 23.281823 & 23.477003 & 20 \\
\hline 400.95 & 25.090000 & 24.849839 & 25.059293 & 25.267963 & 25.47585 & 5 \\
\hline 414.32 & 27.590000 & 26 & 27.067503 & 27.288 & 27.5085 & 27.727767 \\
\hline 0 & 30 & 28 & 28.838 & 29.07008 & 29.3002 & \\
\hline 437.70 & 34 & 30.3787 & 30.620875 & 30.861791 & 31.101552 & 31.340163 \\
\hline 7.72 & 37.350000 & 31.897207 & 32.147183 & 32.395883 & 32.643312 & 32.889478 \\
\hline 456.07 & 40.310000 & 40.493876 & 40.778982 & 41.062074 & 41.343165 & 9 \\
\hline 464.42 & 43.670000 & 44.891975 & 45.188016 & 45.481620 & 45.772805 & 3 \\
\hline 474.45 & 48.260000 & 50.210075 & 50.511906 & 50.810763 & 51.106675 & 51.399671 \\
\hline 484.45 & 53.520000 & 55.406352 & 55.704705 & 55.999558 & 56.290952 & 56.578928 \\
\hline 0 & 58.430 & 59.549749 & 59.837641 & 60.121640 & 60.401799 & 60.678169 \\
\hline 0.00 & 63.510000 & 62.907169 & 63.180613 & 63.449888 & 63.715057 & 63.976182 \\
\hline 507.85 & 67.420000 & 66.278446 & 66.530923 & 66.779023 & 67.022820 & 67.262391 \\
\hline 516.20 & 71.520000 & 69.476530 & 69.701777 & 69.922559 & 70.138965 & 70.351080 \\
\hline 526.22 & 74.860000 & 72.729934 & 72.918005 & 73.101715 & 73.281164 & 73.456452 \\
\hline 537.90 & 77.160000 & 75.690399 & 75.833076 & 75.971800 & 76.10668 & 76.237823 \\
\hline 551.27 & 78.620000 & 78.045114 & 78.139443 & 78.230589 & 78.31865 & 78.403756 \\
\hline 576.32 & 79.700000 & 80.188263 & 80.219546 & 80.249327 & 80.27767 & 80.304670 \\
\hline 601.37 & 80.030000 & 80.747584 & 80.753756 & 80.759516 & 80.764891 & 80.769907 \\
\hline 649.80 & 80.290000 & 80.839682 & 80.839720 & 80.839754 & 80.839783 & 80.839809 \\
\hline 699.90 & 80.520000 & 80.840000 & 80.840000 & 80.840000 & 80.840000 & 80.840000 \\
\hline 50.00 & 80.720000 & 80.840000 & 80.840000 & 80.840000 & 80.84000 & 80.840000 \\
\hline 800.12 & 80.840000 & 80 & 80. & 80.840000 & 80.840000 & \\
\hline
\end{tabular}
standard deviation for each method above
\(1.477343 \quad 1.440612 \quad 1.4258191 .432991 \quad 1.461189\)
the s.e.e. relative change in \(\%\) with \(k:-/+1 \%\) and \(-/+2 \%\)
\(3.6296871 .053194 \quad .015486\)
.518604
2.496592


Figure I. 1 Effect of the number of significant digits (a) and the change of \(k\) in the range of \(\pm 1 \%\) and \(\pm 2 \%\) from the best fit values (b)

The Statistical Analysis of Sample Size Effect
The statistical analysis of sample size effect is examined with results in Tables 4.1.1 and 4.1.2 and the analysis results are listed in the following Table I.1. The results show that the volatile yield is roughly constant and the deviation is small in the sample size range for both heating rates. It is therefore believed that results reflect the intrinsic kinetics and are not significantly affected by mass transfer in these sample size ranges.

Table I. 1 Statistical Analysis of Sample Size Effect
\begin{tabular}{|c|c|c|c|c|}
\hline \multirow{2}{*}{} & \multicolumn{2}{|c|}{\begin{tabular}{c}
\(7.774 \sim 12.034 \mathrm{mg}\) at \\
heating rate \(100^{\circ} \mathrm{C} / \mathrm{min}\)
\end{tabular}} & \multicolumn{2}{c|}{\begin{tabular}{c}
\(8.011 \sim 13.157 \mathrm{mg} \mathrm{at}\) \\
heating rate \(50{ }^{\circ} \mathrm{C} / \mathrm{min}\)
\end{tabular}} \\
\cline { 2 - 5 } & \(\mathrm{V}_{\mathrm{t}=0} \%\) & \(\mathrm{~V}_{\mathrm{t}=10} \%\) & \(\mathrm{~V}_{\mathrm{t}=0} \%\) & \(\mathrm{~V}_{\mathrm{t}=10} \%\) \\
\hline Average V yield & 80.12 & 80.43 & 80.57 & 81.01 \\
\hline Standard Deviation & 0.8 & 0.7 & 0.3 & 0.5 \\
\hline No. of Data Points & 5 & 5 & 8 & 8 \\
\hline
\end{tabular}```

