DAVID FRANKLIN ALDRIDGE
B.A., Amherst College, 1975
M.A., The University of California, Berkeley, 1980

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Department of Geophysics and Astronomy
The University of British Columbia
Vancouver, Canada
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#### Abstract

Techniques for forward modeling and inversion of head wave traveltimes within the framework of one and two dimensional earth models are well developed. The first portion of this thesis extends these methods to encompass three dimensional layered models. Each critically refracting horizon of the model is approximated by a plane interface with arbitrary strike and dip. An advantage of this simple representation is that rapid computation of head wave traveltimes for arbitrary source-receiver geometries can be achieved with a minimum of ray tracing. Inversion methods are then developed for estimating the parameters defining single-layer and multilayer earth models. For the single-layer model, an algebraic solution to the inverse problem exists if refraction traveltimes are observed along two independent line profiles. For multilayer models and/or nonprofile recording geometries, the inversion is formulated as a constrained parameter optimization problem and solved via linear programming. Inclusion of constraints, in the form of inequality relations satisfied by the model parameters, often governs the ability of the algorithm to converge to a realistic solution. The procedure is tested with traveltimes recorded on broadside profiles in a deep crustal seismic experiment.

The second part of this thesis provides specific improvements to various two dimensional refraction traveltime inversion techniques. The generalized reciprocal method (GRM) is reformulated on the basis of an earth model characterized by vertical, rather than normal, layer thicknesses. This allows point values of interface depth to be inferred from the observed traveltimes. A novel interpretation method (critical offset refraction profiling) is described that yields point values of interface depth, interface dip, and refractor velocity. A smooth depth profile of the refracting horizon is then constructed using techniques of linear inverse theory. Finally, an automated version of the classical wavefront method for interpreting refraction traveltimes is developed. Recorded arrival times are downward continued through a near surface heterogeneous velocity structure with a finite-difference propagation algorithm. The locus of a refracting horizon is then obtained by applying a simple imaging condition involving the reciprocal time (the source-to-source traveltime). The method is


tested, apparently successfully, on a shallow refraction dataset recorded at an archeological site.

The final portion of this thesis develops an iterative tomographic inversion procedure for reconstructing a two dimensional $P$-wave velocity field from measured first arrival times. Two key features of this technique are (i) use of a finite-difference algorithm for rapid and accurate forward modeling of traveltimes, and (ii) incorporation of constraint information into the inversion to restrict the nonuniqueness inherent in large scale, nonlinear tomographic inverse problems. Analysis of a simulated vertical seismic profile (VSP) plus crosswell experiment indicates that the inversion algorithm can accurately reconstruct a smoothly varying interwell velocity field. Inclusion of constraints, in the form of horizontal and vertical firstdifference regularization, allows the solution of a traveltime tomography problem that is otherwise severely underdetermined.

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## CHAPTER 1

INTRODUCTION

The traveltimes of head waves propagating in layered earth models have been studied extensively since the inception of applied seismology in the 1920's. The earliest English language publications that examine this topic appear to be the classic works by Barton (1929) and Heiland (1929). These authors derive head wave traveltime formulae appropriate for a simple model consisting of a layer overlying a halfspace. The two media have uniform P-wave velocities and are separated by a plane interface. If the critically refracting interface is horizontal, then head wave traveltime is a linear function of source-receiver offset distance $X$ :

$$
\begin{equation*}
T(X)=m X+b \tag{1.1}
\end{equation*}
$$

The slope $m$ and intercept $b$ are independent of the recording geometry and are easily determined from the known earth model parameters. Numerous investigators, beginning with Barton (1929), have extended this expression to multilayered one dimensional models.

If the subsurface interface is dipping, then the same general mathematical formula applies. However, the slope and intercept are no longer invariant. Rather, the intercept depends on the horizontal coordinates of the source $\left(x_{S}, y_{S}\right)$ and the slope depends on the azimuth angle $\Psi$ of the receiver relative to the source. Barton (1929) and Heiland (1929) examine the case where an inline array of receivers is oriented normal to the strike direction of the refracting horizon. Hence, the azimuth angle $\Psi$ is restricted to two values (say, $\Psi=0$ and $\Psi=\pi$ ) that collectively refer to the updip and downdip directions. Surface-to-surface head wave traveltime is

$$
\begin{equation*}
T\left(X, \Psi, x_{S}\right)=m(\Psi) X+b\left(x_{S}\right) \tag{1.2}
\end{equation*}
$$

In this two dimensional situation, the plane of the head wave raypath is vertical. Barton (1929) and Heiland (1929) derive closed form mathematical expressions for the slope and
intercept in terms of the parameters defining the single-layer earth model. Thus, equation (1.2) is easily evaluated for this case.

Head wave traveltime analysis is extended to multilayered two dimensional models with plane, dipping interfaces by Ewing et al. (1939), Dooley (1952), Adachi (1954), Mota (1954), Ocola (1972), Johnson (1976), and Diebold and Stoffa (1981). The linear traveltime expression (1.2) also applies in this situation. Once again, the recording profile must be oriented perpendicular to the common strike direction of the subsurface refracting horizons. If this condition is not maintained, then the raypath is not confined to a single vertical plane, and a full three dimensional treatment of the problem is necessary. For the multilayered case, the slope and intercept in (1.2) cannot be simply expressed in terms of the specified earth model parameters. Rather, each depends implicitly on the model parameters through a set of raypath orientation angles. It is possible to determine these angles via repeated application of Snell's law of refraction, starting at the critically refracting horizon and working upward to the surface. Hence, equation (1.2) can be evaluated numerically.

Methods for inverting observed head wave arrival times to obtain the earth model parameters are also discussed by the above authors. The techniques assume that traveltimes are recorded by inline forward and reverse spreads, or by a split spread. If the P-wave velocity of the uppermost layer is known, then the measured slope of a particular traveltime branch can be used to infer the angle of incidence of the head wave at the surface. Repeated application of Snell's law of refraction from the surface downward determines the raypath orientation angle within the layer immediately above the critically refracting interface. The dip and velocity of the critical refractor are then obtained from raypath orientation angles calculated for forward and reverse propagating head waves. The method assumes that all velocities and dip angles overlying the target horizon have been previously determined. Finally, interface depths below the source locations are calculated from the measured intercept times.

An alternative two dimensional inversion method exploits the variation of intercept time $b\left(x_{S}\right)$ with source position. Cunningham (1974) demonstrates that head wave arrival times observed on two inline spreads with the same source-receiver azimuth $\Psi$ can be analyzed to
yield refractor dip, depth, and velocity. Although the slopes of the two traveltime curves are the same, the intercept times associated with the different source locations provide sufficient independent information for a solution. The method has practical utility in marine seismic exploration, where refracted arrivals are recorded on single-ended (i.e., unreversed) spreads. Cunningham (1974) examines simple models consisting of one and two layers overlying a halfspace. However, it is possible to demonstrate the validity of the method for general multilayered models. The fact that both forward and reverse arrivals are not required for a successful two dimensional inversion is not widely appreciated, even by specialists in seismic refraction exploration (e.g., Lankston, 1990; Palmer, 1991).

The purpose of Chapters 2, 3, and 4 of this thesis is to extend forward modeling and inversion of head wave arrival times to three dimensional layered models. This class of earth models is characterized by uniform velocity layers bounded by plane interfaces possessing arbitrary dip and strike. The linear traveltime expression remains valid in this three dimensional situation:

$$
\begin{equation*}
T\left(X, \Psi, x_{S}, y_{S}\right)=m(\Psi) X+b\left(x_{S}, y_{S}, \Psi\right) \tag{1.3}
\end{equation*}
$$

The source-receiver azimuth angle is not restricted to two values, but may range throughout the interval $0 \leq \Psi<2 \pi$. Note that the slope in equation (1.3) depends only on the recording azimuth $\Psi$. In general, the intercept depends on both the source location and the azimuth angle to the receiver. However, for the simple model consisting of a single layer overlying a halfspace, the dependence on $\Psi$ vanishes.

Chapter 2 presents a rigorous derivation of equation (1.3) for the single-layer earth model. Closed form mathematical expressions are obtained for the slope and intercept in terms of the specified model parameters. Also, equations for critical and crossover distances are derived. All of these expressions are generalizations of more familiar formulae that apply to one and two dimensional models, and are quite useful for forward modeling purposes. Unlike the analogous situation for seismic reflection, there is very little published information on the traveltimes of critically refracted waves in this basic three dimensional model. Dix
(1935), using geometric arguments, demonstrates that head wave isochrons (i.e., loci of fixed arrival time at the surface) are conic sections. A similar result is obtained by Dunkin and Levin (1971) although their analysis is limited to a particular areal recording configuration. Chapter 2 provides a derivation of the same result, using a completely different methodology from that in Dix (1935). No restrictions are placed on the recording geometry.

The simple 'layer over a halfspace' earth model is defined by five parameters: two P-wave velocities, two interface orientation angles, and a depth to the interface below a reference point. If the overburden velocity $v_{1}$ is known, then only four model parameters need to be determined by inverting the observed head wave arrival times. Heiland (1940, p. 525-527) describes an inversion method that utilizes traveltimes measured by four line profiles emanating from a single source location. Russell et al. (1982) require three profiles. However, Chapter 2 demonstrates that traveltimes recorded by only two profiles are sufficient to determine the attitude, velocity, and depth of the refractor. Two independent line profiles provide four measured data (two slopes and two intercepts) that allow an algebraic solution for the four unknown model parameters. Inversion methods that are not based on conventional line profile recording geometries are also possible. For example, Chapter 2 indicates that all five model parameters can be obtained by analysis of the geometric properties of head wave isochrons. This result suggests interesting possibilities for determining overburden velocity $v_{1}$ from the refraction data alone.

In Chapter 3, the derivation of the linear traveltime formula (1.3) is extended to multilayered earth models. A substantial simplification in the mathematical proof is obtained by using a novel three dimensional form of Snell's law of refraction. A previous proof given by Diebold (1987) is very cumbersome, and may actually be incorrect because it yields an expression that does not generalize to arbitrary source-receiver geometries properly. As in the two dimensional multilayered case, the slope and intercept in (1.3) depend on the orientation angles of the raypath within each layer. However, two angles are now necessary to describe the orientation of a raypath segment in three dimensional space. Chapter 3 describes a rapid computational procedure for obtaining these angles by applying Snell's law repeatedly from
the critically refracting horizon upward to the surface. Hence, the traveltime equation (1.3) is readily evaluated.

Finally, inversion of head wave traveltimes for three dimensional planar structure is addressed in Chapter 4. Single layer and multiple layer algorithms are developed for data acquisition geometries where sources and receivers are located on the surface. The inversion is posed as a constrained parameter optimization problem. An initial estimate of the earth model parameters is iteratively refined until an acceptable match is obtained between observed and predicted arrival times. Similiar parametric inversion schemes have been recently applied to three dimensional reflection traveltimes by Chiu et al. (1986), Lin (1989), and Phadke and Kanasewich (1990). Chiu and Stewart (1987) invert combined VSP and surface-to-surface reflection traveltimes. Kanasewich and Chiu (1985) invert combined reflection and refraction arrivals. They use the iterative ray-bending approach of Chander (1977a) to calculate the traveltime derivatives needed for the inversion. The computational procedure developed in Chapter 3 is not iterative. Hence, it provides a rapid alternative to ray-bending or ray-shooting methods for obtaining the necessary head wave traveltime derivatives.

A novel feature of the inversion algorithms discussed in Chapter 4 is the introduction of constraint information via inequality relations that are imposed on the model parameters. Often, a priori geological or geophysical data are available to guide and constrain an inversion. For example, many shallow seismic refraction projects are undertaken in conjuction with a drilling program. Interface depth and layer velocity data may be obtained from borehole logs. Constraints are particularly useful for the inversion of head wave traveltimes, because the problem can be very ill-conditioned and admit numerous solutions. Application of the constrained inversion algorithm to a crustal seismic refraction dataset from northern Alberta indicates that it can be a useful tool for analysis of broadside recorded traveltimes. Currently, there is a lack of techniques for the effective analysis of such data.

Following the analysis of three dimensional earth models in Chapters 2, 3, and 4, the central portion of this thesis (Chapters 5, 6, and 7) develops specific improvements to various two dimensional refraction traveltime inversion methods. By restricting consideration to two
dimensional models, examination of geological realities like undulating interfaces and/or variable velocity media becomes mathematically tractable.

Chapter 5 demonstrates that point estimates of interface depth and dip can be inferred from the observed refraction traveltimes. Two interpretation procedures are used for the discussion. The Generalized Reciprocal Method (GRM) is a technique for delineating an undulating subsurface interface from refracted arrivals recorded on inline forward and reverse spreads. It was developed by Palmer $(1980,1981)$ as an extension of the conventional reciprocal method (Hawkins, 1961) of refraction traveltime interpretation. Although the GRM has been successfully applied to the problem of mapping undulating horizons, the mathematical formulation of the method is based on a two dimensional earth model with plane interfaces. Hence, the improved head wave traveltime formula derived in Chapter 3 can be applied to GRM analysis. A useful traveltime expression is obtained by specializing the general equation to a two dimensional situation. As demonstrated in Chapter 5, all of the common GRM analysis tools can be derived from this novel 2D traveltime formula in a straightforward manner. Moreover, these new expressions allow point depth estimates of the refracting interface to be calculated from the measured traveltimes. A refractor depth profile can then be obtained by interpolation. In contrast, the current GRM depth estimation procedure involves the laborious task of constructing an envelope to a set of circular arcs (Hatherly, 1980).

A traveltime inversion technique that explicitly incorporates nonplane refracting horizons into the model should yield more accurate results. Thus, Chapter 5 also proposes a head wave traveltime interpretation method tentatively named critical offset refraction profiling. This inversion technique accommodates undulations in both the surface and the refracting interface, as well as horizontal variations in the velocity of the refracting layer. Analysis reveals that a mathematically exact solution for the earth model parameters exists: point values of interface depth, interface dip, and refractor velocity can all be obtained from the observed traveltimes. Hence, as with the modified GRM developed in the same chapter, a continuous depth profile of the interface can be constructed by interpolation. The dip
estimates obtained from critical offset refraction profiling provide additional constraints on the interpolant.

A detailed description of a method for calculating a smooth refractor depth profile from a set of point estimates of depth and dip is presented in Chapter 6. Determination of the interface depth profile is treated as an interpolation problem. Hence, the technique differs significantly from traditional methods of depth profile computation via envelope construction (Thornburgh, 1930; Dix, 1941; Tarrant, 1956; Hales, 1958; Hawkins, 1961; Hatherly, 1980; Palmer, 1980, 1990, 1991). Linear inverse theory provides the mathematical framework for solving this problem. The smoothest (i.e., minimum curvature) interpolant is the natural model to adopt in this situation because refraction traveltime inversion methods assume, either explicitly or implicitly, that local interface curvature is negligible. Thus, the final model for the refracting horizon is consistent with prior assumptions used for inferring its depth and/or dip. Posing the problem as an interpolation issue also has several specific advantages regarding treatment of the data: (i) additional depth and dip values arising from a variety of geological, geophysical, or engineering techniques are readily incorporated into the model construction, (ii) variable weighting of all data is easily achieved, and (iii) an adjustable misfit to error contaminated data is possible.

Recent advances in seismic refraction interpretation involve constructing a subsurface image directly from the observed first break waveforms (Clayton and McMechan, 1981; Hill, 1987). Chapter 7 describes an alternative imaging procedure that works only with the picked arrival times. The technique is an automated implementation of the classical Wavefront Method for interpreting refraction traveltimes (Thornburgh, 1930; Hagedoorn, 1959; Rockwell, 1967). Modern finite-difference propagation algorithms are used to downward continue recorded refraction arrival times through a near-surface heterogeneous velocity structure. Two such subsurface traveltime fields need to be reconstructed from the arrivals recorded on a forward and reverse geophone spread. The locus of a shallow refracting horizon is then defined by a simple imaging condition involving the reciprocal time (the traveltime between source positions at either end of the spread). Refractor velocity is estimated in a subsequent step by calculating the directional derivative of the reconstructed subsurface wavefronts
along the imaged interface. The principal limitation of the technique arises from imprecise knowledge of the overburden velocity distribution. This velocity information must be obtained from uphole times, direct arrivals, shallow refractions, and borehole data. Analysis of synthetic data examples indicates that the technique can accurately image both synclinal and anticlinal structures. The method is also tested, apparently successfully, on a shallow refraction dataset acquired at an archeological site in western Crete.

Recently, tomographic techniques have been applied to the shallow seismic refraction problem (Hampson and Russell, 1984; De Amorim et al., 1987; Olsen, 1989; Scales et al., 1990). Traveltime tomography is a general and versatile method of velocity analysis. The final portion of this thesis (Chapter 8) presents an iterative tomographic inversion procedure for reconstructing a two dimensional velocity field from measured first arrival times. Two key features of this technique are (i) use of a finite-difference algorithm for rapid and accurate forward modeling of traveltimes, and (ii) incorporation of constraint information into the inversion in order to restrict the nonuniqueness inherent in large scale, nonlinear tomographic inverse problems.

Finite-difference traveltime computation (Vidale, 1988) provides a useful alternative to conventional raytracing in tomography. All first arrival wave types (direct and refracted arrivals, head waves, diffractions) are handled with relative ease. Curved raypaths, needed for subsequent updating of the velocity model, are generated by following the steepest descent direction through a computed traveltime field from each receiver back to the source. Since arrival times are computed throughout a gridded two dimensional velocity field, very general recording geometries are easily accommodated. The main limitation of the method is that it is restricted to first arrivals. Hence, the algorithm developed in Chapter 8 cannot be applied to reflection tomography.

Constraint information may arise from known values of the subsurface velocity (e.g., from outcrops or well logs) or the imposition of reasonable geological characteristics, like smoothness, on the constructed velocity model. In either case, the data equations are augmented with additional linear constraint equations and a least squares solution of the entire
system is obtained. In Chapter 8, the constrained inversion algorithm is applied to a simulated double-well VSP plus crosswell experiment. The results indicate that the algorithm can accurately reconstruct a smoothly varying interwell velocity field. Inclusion of constraint information, in the form of horizontal and vertical first-difference regularization, allows the solution of a traveltime tomography problem that is otherwise severely underdetermined ( 243 data and 10000 unknowns). For this example, no obvious improvement is obtained by the addition of borehole velocity constraints.

## CHAPTER 2

## HEAD WAVE TRAVELTIMES IN A THREE DIMENSIONAL SINGLE-LAYER EARTH

### 2.1 Introduction

Many refraction seismologists believe that a minimum of two reversed profiles, preferentially oriented at right angles to each other, are necessary to determine the attitude, velocity, and depth of a plane subsurface refractor (e.g., Heiland, 1940, p. 525-527). Russell et al. (1982) demonstrate that traveltimes recorded along three unreversed spreads can be analyzed to yield this same information. They state that similiar measurements made by only two such profiles cannot define the three dimensional attitude of the dipping horizon. This statement, however, is incorrect. Russell et al. (1982) do not fully utilize the information contained in the intercept times of the traveltime curves. One purpose of this chapter is to demonstrate that, in many cases, two refraction profiles are sufficient to define the three dimensional attitude, true velocity, and depth of a plane refractor. Generally speaking, the main condition required is that the two lines provide independent traveltime information about the subsurface.

### 2.2 Earth model and recording geometry

The earth model consists of a single layer with P -wave speed $v_{1}$ overlying a halfspace with P-wave speed $v_{2}$. The plane interface separating the two media possesses, in general, a three dimensional dipping attitude. Let the $x y$ plane of a right handed, rectangular coordinate system be coincident with the free surface of the layer; depth $z$ is measured positive in the downward direction. If $\mathbf{r}=x \mathbf{i}+y \mathbf{j}+z \mathbf{k}$ is the position vector of an arbitrary point on the subsurface interface, then the equation defining this dipping plane is

$$
\begin{equation*}
\mathbf{r} \cdot \mathbf{n}=d \tag{2.1}
\end{equation*}
$$

$d$ is the perpendicular distance from the coordinate origin $O$ to the plane and $\mathbf{n}$ is a unit vector pointing from $O$ along the normal to the plane. Figure 2.1 illustrates that $\mathbf{n}$ is conveniently described in terms of polar and azimuthal angles $\phi$ and $\theta$ :

$$
\begin{equation*}
\mathbf{n}=(\sin \phi \cos \theta) \mathbf{i}+(\sin \phi \sin \theta) \mathbf{j}+(\cos \phi) \mathbf{k} . \tag{2.2}
\end{equation*}
$$

$\phi(0 \leq \phi<\pi / 2)$ is the dip angle of the interface and $\theta(0 \leq \theta<2 \pi)$ defines the updip direction. If the $+x$ and $+y$ axes are taken to point toward geographic north and east, respectively, then the strike angle of the interface is $\theta+\pi / 2$ (modulo $2 \pi$ ).


Fig. 2.1. Earth model and coordinate reference frame.

Figure 2.2a is a plan view of the surface recording geometry. The position vectors of the source $S$ and receiver $R$ are given by

$$
\begin{equation*}
\mathbf{r}_{S}=x_{S} \mathbf{i}+y_{S} \mathbf{j}, \quad \mathbf{r}_{R}=x_{R} \mathbf{i}+y_{R} \mathbf{j} \tag{2.3a,b}
\end{equation*}
$$

The azimuth angle of the receiver relative to the source is $\alpha(0 \leq \alpha<2 \pi)$. Azimuth is measured positive in the clockwise sense from geographic north. If the horizontal offset between source and receiver is denoted by $X(X \geq 0)$, then the receiver coordinates can be expressed in terms of the source coordinates as follows:

$$
\begin{equation*}
x_{R}=x_{S}+X \cos \alpha, \quad y_{R}=y_{S}+X \sin \alpha \tag{2.4a,b}
\end{equation*}
$$

Profile recording geometry is defined by the condition that the azimuth angle $\alpha$ remains fixed for a set of receivers that record energy from a single source.

b)


Fig. 2.2. (a) Plan view of surface recording geometry. $S$ and $R$ denote a point source and a point receiver, respectively. (b) Spatial relationship of the two sources $S_{1}$ and $S_{2}$.

### 2.3 Traveltime derivation

Consider the critically refracted raypath from surface source $S$ to surface receiver $R$ (Figure 2.3). This raypath is confined to a single plane referred to herein as the raypath plane. In general, the raypath plane is not a vertical plane (parallel to the $z$ axis). Only in the particular case where the profile line is oriented directly updip or downdip is the raypath plane vertical. The traveltime of a head wave propagating along the critical raypath from source to receiver can be worked out from simple geometric considerations. It is

$$
\begin{equation*}
T=\frac{L}{v_{2}}+\left(d_{S}+d_{R}\right) \frac{\cos i_{c}}{v_{1}}, \quad\left(L \geq L_{c r i t}\right) \tag{2.5}
\end{equation*}
$$

$L$ is the source-receiver range measured parallel to the refracting interface and $d_{S}$ and $d_{R}$ are perpendicular distances from $S$ and $R$ to this interface. The critical refraction angle $i_{c}$ is given by the usual expression: $\sin i_{c}=v_{1} / v_{2}$. Of course, the head wave exists only for ranges exceeding the critical range $L_{\text {crit }}$ given by

$$
\begin{equation*}
L_{c r i t}=\left(d_{S}+d_{R}\right) \tan i_{c} \tag{2.6}
\end{equation*}
$$



Fig. 2.3. Head wave raypath. The plane of this diagram is perpendicular to the refracting interface and thus is not necessarily vertical.

The head wave traveltime formula is more useful to the geophysicist if it is expressed in terms of the horizontal offset $X$. A simple analytical derivation of the desired relation is given in this section. An advantage of this method is that it does not require visualization of the three dimensional geometry of the problem. Starting with the general equation for the dipping plane, it is easy to demonstrate that the perpendicular distances from $S$ and $R$ to this plane are given by $d_{S}=d-\mathbf{r}_{S} \cdot \mathbf{n}$ and $d_{R}=d-\mathbf{r}_{R} \cdot \mathbf{n}$, respectively. Hence

$$
d_{R}=d_{S}+\left(\mathbf{r}_{S}-\mathbf{r}_{R}\right) \cdot \mathbf{n}
$$

Substituting from equations (2.2), (2.3), and (2.4) and reducing yield

$$
d_{R}=d_{S}-X \sin \phi \cos (\alpha-\theta)
$$

Now define an angle $\delta(-\pi / 2<\delta<\pi / 2)$ as follows: $\sin \delta \equiv \sin \phi \cos (\alpha-\theta)$. Then, the above expression becomes

$$
\begin{equation*}
d_{R}=d_{S}-X \sin \delta \tag{2.7}
\end{equation*}
$$

It is stressed that the angle $\delta$ is not the apparent dip of the refracting interface along the profile line azimuth $\alpha$. Apparent dip $\gamma$ is related to true dip $\phi$ via the following equation:

$$
\begin{equation*}
\tan \gamma=\tan \phi \cos (\alpha-\theta) \tag{2.8}
\end{equation*}
$$

Obviously, $\delta \neq \gamma$. The difference between $\delta$ and $\gamma$ arises from the fact that the former is measured in the raypath plane, whereas apparent dip is measured in a vertical plane. For small values of true $\operatorname{dip}, \delta \approx \gamma$.

The range $L$ is equal to the distance $\overline{P Q}$ in Figure 2.3. Hence

$$
\begin{aligned}
L^{2}=\overline{P Q}^{2} & =\left|\left(\mathbf{r}_{S}+d_{S} \mathbf{n}\right)-\left(\mathbf{r}_{R}+d_{R} \mathbf{n}\right)\right|^{2} \\
& =\left|\left(\mathbf{r}_{S}-\mathbf{r}_{R}\right)+\left(d_{S}-d_{R}\right) \mathbf{n}\right|^{2} \\
& =\left|\mathbf{r}_{S}-\mathbf{r}_{R}\right|^{2}+2\left(d_{S}-d_{R}\right)\left(\mathbf{r}_{S}-\mathbf{r}_{R}\right) \cdot \mathbf{n}+\left(d_{S}-d_{R}\right)^{2} \\
& =\left|\mathbf{r}_{S}-\mathbf{r}_{R}\right|^{2}-\left(d_{S}-d_{R}\right)^{2} \\
& =X^{2}-X^{2} \sin ^{2} \delta
\end{aligned}
$$

Thus

$$
\begin{equation*}
L=X \cos \delta \tag{2.9}
\end{equation*}
$$

Equations (2.7) and (2.9) are the desired expressions. It is evident that these formulae could have been derived by purely geometrical reasoning based on the raypath diagram of Figure 2.3. However, the relationship of the angle $\delta$ to the profile line azimuth and the orientation angles of the refracting plane would not have been easy to ascertain. Substituting expressions (2.7) and (2.9) into equation (2.5) and reducing yield an expression for the head wave traveltime:

$$
\begin{equation*}
T=\frac{\sin \left(i_{c}-\delta\right)}{v_{1}} X+\frac{2 d_{S} \cos i_{c}}{v_{1}}, \quad\left(X \geq X_{c r i t}\right) \tag{2.10}
\end{equation*}
$$

An expression for the critical offset distance $X_{c r i t}$ is obtained by similiar manipulations. Evaluating equations (2.7) and (2.9) at the critical distance and then substituting into (2.6) gives

$$
\begin{equation*}
X_{c r i t}=2 d_{S} \frac{\sin i_{c}}{\cos \left(i_{c}-\delta\right)} \tag{2.11}
\end{equation*}
$$

The crossover distance is defined as the particular source-receiver offset at which direct wave and head wave arrival times are identical. Setting $T$ in (2.10) equal to $X / v_{1}$ and solving for $X$ yields

$$
\begin{equation*}
X_{c r o s s}=2 d_{S} \frac{\cos i_{c}}{1-\sin \left(i_{c}-\delta\right)} \tag{2.12}
\end{equation*}
$$

Finally, the traveltime formula (2.10) indicates that refracted arrivals propagate along the receiver spread with an apparent velocity $V_{a}$ given by

$$
\begin{equation*}
V_{a}=\frac{v_{1}}{\sin \left(i_{c}-\delta\right)}=\frac{\sin i_{c}}{\sin \left(i_{c}-\delta\right)} v_{2} \tag{2.13}
\end{equation*}
$$

The variation of apparent velocity with profile azimuth is depicted in Figure 2.4 for the specific case where $v_{1} / v_{2}=3 / 5$.

Equations (2.10) through (2.13) indicate that the three dimensional refraction formulae are straightforward generalizations of those appropriate for the 2 D problem: the angle $\delta$ replaces the true dip $\phi$ in the relevant expressions. It is emphasized again that $\delta$ is not the apparent dip of the subsurface interface along the profile line. However, for a gently dipping refractor, the practical difference between the two is small.


Fig. 2.4. Normalized apparent refractor velocity $V_{a} / v_{2}$. Each curve refers to a specific value of the interface dip angle $\phi$. The P-wave velocity ratio is $v_{1} / v_{2}=3 / 5$.

### 2.4 Traveltime inversion

Traveltimes recorded by a set of refraction profiles can be inverted to recover the 3D attitude, true velocity, and depth of the refractor. Let the measured slope and intercept time of the $i^{\text {th }}$ such traveltime curve be $m_{i}$ and $\tau_{i}$, respectively. From equation (2.10), these are related to the assumed earth model parameters via

$$
\begin{equation*}
m_{i}=\frac{\sin \left(i_{c}-\delta_{i}\right)}{v_{1}}, \quad \tau_{i}=\frac{2 d_{S_{i}} \cos i_{c}}{v_{1}} \tag{2.14a,b}
\end{equation*}
$$

with:

$$
\begin{gather*}
\sin \delta_{i}=\sin \phi \cos \left(\alpha_{i}-\theta\right)  \tag{2.14c}\\
d_{S_{i}}=d-\sin \phi\left(x_{S_{i}} \cos \theta+y_{S_{i}} \sin \theta\right) \tag{2.14d}
\end{gather*}
$$

( $x_{S_{i}}, y_{S_{i}}$ ) are the position coordinates of source $i$ and $\alpha_{i}$ is the azimuth angle of profile line $i$. The perpendicular depth to the interface below source $i$ is designated $d_{S_{i}}$.

How many refraction lines are required in order to successfully invert for the earth model parameters? Assuming that the overburden velocity $v_{1}$ is known (perhaps from borehole data or traveltime analysis of the direct arrivals), then the earth model is defined by the four parameters $\left(v_{2}, \phi, \theta, d\right)$. Intuition suggests that two refraction profiles will yield the four data (two slopes and two intercepts) required to solve the problem unambiguously. Indeed, if the line index $i$ is set equal to 1 and then 2 , the system (2.14) above becomes a set of 8 equations in 8 unknowns. Since the equations are nonlinear in the unknowns, a definitive statement on the existence and uniqueness of a solution cannot be made. However, it can be demonstrated by algebraic techniques that, in many cases, two refraction profiles are sufficient to solve the problem.

The method proposed by Russell et al. (1982) requires three lines to obtain the refractor velocity, attitude, and depth. They utilize the slopes of the three traveltime curves ( $m_{1}, m_{2}, m_{3}$ ) and one intercept time ( $\tau_{1}$ ) as measured data in the inversion. For the case where the index $i$ runs from 1 to 3 , system (2.14) represents 12 equations in 10 unknowns;
hence, it would seem that redundant information exists. Although redundant data are always valuable in any practical inversion scheme, the theoretical minimum set of conditions under which the inversion is possible are of interest here.

In the following analysis, only two refraction profiles are employed. First, define the angles $\mu_{1}$ and $\mu_{2}$ :

$$
\begin{equation*}
\mu_{1} \equiv \sin ^{-1}\left(m_{1} v_{1}\right), \quad \quad \mu_{2} \equiv \sin ^{-1}\left(m_{2} v_{1}\right) \tag{2.15a,b}
\end{equation*}
$$

Next, the position of source 2 is expressed in terms of the coordinates of source 1 :

$$
\begin{equation*}
x_{S_{2}}=x_{S_{1}}+H \cos \beta, \quad y_{S_{2}}=y_{S_{1}}+H \sin \beta \tag{2.16a,b}
\end{equation*}
$$

$\beta(0 \leq \beta<2 \pi)$ is the azimuth of source 2 relative to source 1 (measured positive in the clockwise sense from geographic north) and $H$ is the horizontal distance between the two source locations (see Figure 2.2b). Using these expressions, the angles $\delta_{1}, \delta_{2}$ and the depths $d, d_{S_{1}}, d_{S_{2}}$ can be quickly eliminated from the system (2.14). A reduced system is obtained consisting of three nonlinear equations in the three unknown angles $i_{c}, \phi$, and $\theta$ :

$$
\begin{align*}
\sin \left(i_{c}-\mu_{1}\right) & =\sin \phi \cos \left(\alpha_{1}-\theta\right)  \tag{2.17a}\\
\sin \left(i_{c}-\mu_{2}\right) & =\sin \phi \cos \left(\alpha_{2}-\theta\right)  \tag{2.17b}\\
v_{1}\left(\tau_{1}-\tau_{2}\right) & =2 H \cos i_{c} \sin \phi \cos (\beta-\theta) \tag{2.17c}
\end{align*}
$$

### 2.4.1 Special cases

Three particular cases of the above system are examined before a more general solution technique is described in the next section.

1) Coincident sources. If the two sources occupy the same position $(H=0)$, then equation $(2.17 \mathrm{c})$ above reduces to $0=0$ (coincident source locations imply identical intercept times).

The two remaining equations contain three unknowns and cannot have a unique solution. Hence, two refraction profiles emanating from the same source location supply insufficient independent information for a successful inversion. The split recording spread is a particular example of coincident-source refraction profiles (Johnson, 1976).
2) Parallel profiles. If the azimuths of both refraction lines are the same ( $\alpha_{1}=\alpha_{2}$ ), then equations (2.17a) and (2.17b) above are not independent (identical azimuths imply identical slopes of the traveltime curves and thus $\mu_{1}=\mu_{2}$ ). Hence, the parallel profile recording configuration is not adequate to solve the problem either.
3) Anti-parallel profiles. Suppose that line 2 is recorded in a direction opposite to that of line $1\left(\alpha_{2}=\alpha_{1} \pm \pi\right)$. Then equations (2.17a) and (2.17b) can be solved immediately for the critical refraction angle:

$$
\begin{equation*}
i_{c}=\frac{\mu_{1}+\mu_{2}}{2} \tag{2.18}
\end{equation*}
$$

The refractor velocity is obtained via $v_{2}=v_{1} / \sin i_{c}$. This procedure for calculating $v_{2}$ is identical to that used with the classical reversed spread in the two dimensional situation. It is clear that it is generally valid for anti-parallel profiles (colinear or otherwise) recorded over a three dimensional dipping interface.

With the critical angle $i_{c}$ determined, the system (2.17) can be solved for the remaining unknowns. The azimuth of line 1 is designated by $\alpha$. Straightforward, but tedious, algebraic manipulation then yields

$$
\begin{gather*}
\tan \theta=-\left[\frac{\left(\tau_{1}-\tau_{2}\right) \cos \alpha+H\left(m_{1}-m_{2}\right) \cos \beta}{\left(\tau_{1}-\tau_{2}\right) \sin \alpha+H\left(m_{1}-m_{2}\right) \sin \beta}\right]  \tag{2.19}\\
\sin \phi=F \sqrt{\left(\tau_{1}-\tau_{2}\right)^{2}+2\left(\tau_{1}-\tau_{2}\right) H\left(m_{1}-m_{2}\right) \cos (\alpha-\beta)+H^{2}\left(m_{1}-m_{2}\right)^{2}} \tag{2.20}
\end{gather*}
$$

where the multiplicative factor $F$ is given by

$$
\begin{equation*}
F=\left|\frac{v_{1}}{2 H \cos i_{c} \sin (\alpha-\beta)}\right| \tag{2.21}
\end{equation*}
$$

A minor ambiguity associated with selecting the proper branch of the inverse tangent function in equation (2.19) is easily resolved by ensuring that the angle $\theta$ satisfies the original system (2.17). After $\theta$ and $\phi$ are determined, one of the equation pairs (2.14b,d) can be solved to yield the distance $d$ from the reference point $O$ to the.interface.

The solution for the two orientation angles given by equations (2.19) and (2.20) contains an important indeterminacy when the azimuth angle $\beta$ of source 2 equals $\alpha$ or $\alpha \pm \pi$. In this case, the recording geometry consists of two colinear, anti-parallel profiles. The identity of reciprocal times recorded at the two inline shot positions (i.e., $m_{1} H+\tau_{1}=m_{2} H+\tau_{2}$ ) implies that each equation reduces to $0 / 0$. Hence, two colinear anti-parallel profiles are inadequate for defining three dimensional refractor attitude. The reversed spread is a typical example of this configuration. As long as there exists some perpendicular offset distance between the two lines recorded in opposite directions, the inverse problem is well posed, at least in a theoretical sense. The latter data acquisition geometry is a fairly common arrangement for many land and marine seismic surveys.

### 2.4.2 Arbitrary line azimuths

A combination of algebraic and algorithmic techniques yields a solution of the system (2.17) for the case of arbitrary profile line azimuths $\alpha_{1}$ and $\alpha_{2}$. The special situations treated in the previous section are excluded. Eliminating $\theta$ and $\phi$ results in a quadratic equation in $\tan i_{c}:$

$$
\begin{equation*}
A \tan ^{2} i_{c}+B \tan i_{c}+C=0 \tag{2.22}
\end{equation*}
$$

The coefficients $A, B$, and $C$ depend upon measured quantities and recording geometry as follows:

$$
\begin{gather*}
A=\frac{v_{1}\left(\tau_{1}-\tau_{2}\right)}{2 H} \sin \left(\alpha_{2}-\alpha_{1}\right)  \tag{2.23a}\\
B=-\cos \mu_{1} \sin \left(\alpha_{2}-\beta\right)+\cos \mu_{2} \sin \left(\alpha_{1}-\beta\right)  \tag{2.23b}\\
C=\sin \mu_{1} \sin \left(\alpha_{2}-\beta\right)-\sin \mu_{2} \sin \left(\alpha_{1}-\beta\right)+A \tag{2.23c}
\end{gather*}
$$

In the case of anti-parallel profiles the constant $A$ vanishes and $\tan i_{c}$ is obtained by solving a linear equation. It is easy to verify that the expression for the critical angle $i_{c}$ given by the previous equation (2.18) is reproduced. In general, solution of equation (2.22) is via the quadratic formula:

$$
\begin{equation*}
\tan i_{c}=\frac{-B+\operatorname{sgn}(B) \sqrt{B^{2}-4 A C}}{2 A} \tag{2.24}
\end{equation*}
$$

where the sign of the radical has been chosen to yield an indeterminacy for $A=0$. Application of L'Hopital's rule then indicates that (2.18) is recovered as $A \rightarrow 0$. After $i_{c}$ is determined, simply back substitute sequentially into the expressions for the remaining unknowns. This is done numerically rather than symbolically since the formulae rapidly become unwieldly. Hence

$$
\tan \theta=-\left[\begin{array}{l}
\frac{H \cos i_{c} \sin \left(i_{c}-\mu_{1}\right) \cos \beta-\left[v_{1}\left(\tau_{1}-\tau_{2}\right) / 2\right] \cos \alpha_{1}}{H \cos i_{c} \sin \left(i_{c}-\mu_{1}\right) \sin \beta-\left[v_{1}\left(\tau_{1}-\tau_{2}\right) / 2\right]} \sin \alpha_{1} \tag{2.25}
\end{array}\right] .
$$

An analogous (and completely equivalent) formula involving $\mu_{2}$ and $\alpha_{2}$ could be used. Again, care must be exercised to ensure that the angle $\theta$ obtained by inverting (2.25) satisfies the original system of equations. Any one of the expressions ( $2.17 \mathrm{a}, \mathrm{b}, \mathrm{c}$ ) can then be solved for the true dip angle $\phi$. Finally, depth $d$ is determined as previously described.

### 2.5 Traveltime isochrons

Heretofore, the analysis is concerned with an inversion method for critically refracted traveltime data observed on two line profiles. An alternate approach to interpreting three dimensional refraction data entails recording arrival times on an areal grid of receivers. The measured data are posted and contoured on a map display of the receiver array. Characteristic patterns observed in the contoured data are used to qualitatively infer subsurface structural style. If the traveltime data are sufficiently precise, a quantitative estimate of various structural or stratigraphic parameters (dip and strike angles, locations of faults or
edges, magnitude and sense of fault throw, refractor velocity, etc.) is possible (Dunkin and Levin, 1971; Palmer, 1986, p. 246-249).

For a fixed traveltime $T$, the head wave wavefront occupies a three dimensional spatial locus consisting of a frustum of a cone. The intersection of this locus with the recording plane $z=0$ is defined to be a traveltime isochron. A closed form expression for a head wave traveltime isochron in polar coordinates is easily obtained by solving equation (2.10) for offset $X$ as a function of the azimuthal angle $\alpha$ :

$$
\begin{equation*}
X(\alpha)=\frac{v_{1}(T-\tau)}{\sin \left[i_{c}-\delta(\alpha)\right]} \tag{2.26}
\end{equation*}
$$

where $\tau$ is the intercept time $\left(\tau=2 d_{S} \cos i_{c} / v_{1}\right)$ and $\delta(\alpha) \equiv \sin ^{-1}[\sin \phi \cos (\alpha-\theta)]$. Straightforward algebraic manipulation reduces this expression to a quadratic form in the distance $X$ :

$$
\begin{equation*}
X^{2}\left[1-\left(\frac{\sin \phi}{\sin i_{c}}\right)^{2} \cos ^{2}(\alpha-\theta)\right]-2 X\left[\frac{v_{1}(T-\tau) \cos i_{c} \sin \phi}{\sin ^{2} i_{c}} \cos (\alpha-\theta)\right]-\left[\frac{v_{1}(T-\tau)}{\sin i_{c}}\right]^{2}=0 \tag{2.27}
\end{equation*}
$$

The mathematical structure of this expression is identical to that of equation (A2) in Appendix A. Hence, a locus of fixed head wave arrival time $T$ is a conic section (either an ellipse or a hyperbola) with principal axes parallel to the dip and strike directions of the refracting interface. Furthermore, the parameters defining the geometry of the curve can be determined by equating coefficients in (2.27) and (A5). Solving for the eccentricity $e$, semimajor axis $a$, and the origin-to-center distance $l_{c}$ yields

$$
\begin{equation*}
e=\frac{\sin \phi}{\sin i_{c}}, \quad a=\left|\frac{v_{1}(T-\tau) \sin i_{c} \cos \phi}{\sin ^{2} i_{c}-\sin ^{2} \phi}\right|, \quad \quad l_{c}=\frac{v_{1}(T-\tau) \cos i_{c} \sin \phi}{\sin ^{2} i_{c}-\sin ^{2} \phi} \tag{2.28a,b,c}
\end{equation*}
$$

These three expressions are consistent with equivalent formulae derived in an entirely different manner by Dix (1935). Two distinct cases must now be examined.

Case 1: $i_{c}>\phi$. If the critical angle exceeds the interface dip angle, then the eccentricity of the conic section is less than one $(e<1)$. Hence, traveltime isochrons are ellipses that encompass the source position. However, the shotpoint is not located at a center or a focus of any ellipse. Rather, the center is displaced in the updip direction by the distance $l_{c}$, and the nearest focus is displaced downdip by the amount $a e-l_{c}$. For a horizontal interface, the above expressions reduce to $e=0, a=v_{2}(T-\tau)$ and $l_{c}=0$. In this simple situation, head wave traveltime isochrons are concentric circles (centered at the source position) with radii that increase linearly with $T-\tau$.
Case 2: $\quad i_{c}<\phi$. In this situation, the eccentricity exceeds one (e>1) and isochrons are hyperbolae. This case may arise where there is a strong velocity contrast between overburden and refractor, together with a large dip on the intervening interface. Head wave traveltimes actually decrease with offset distance in a wedge of azimuths centered about the updip direction (i.e., $T-\tau<0$ for $\alpha$ bounded by $\theta \pm \cos ^{-1}(1 / e)$ ).

Elliptic and hyperbolic isochrons for head wave traveltime are illustrated in Figure 2.5. These contours are associated with actual head wave arrivals only if the source-receiver offset exceeds the critical distance given by equation (2.11). The curve defined by $X_{c r i t}(\alpha)$ delimits the precritical offset zone, within which no head waves are observed. Expression (2.11) easily reduces to
$X_{c r i t}^{2}\left[1-\left(\frac{\sin \phi}{\cos i_{c}}\right)^{2} \cos ^{2}(\alpha-\theta)\right]-2 X_{c r i t}\left[-2 d_{S} \tan ^{2} i_{c} \sin \phi \cos (\alpha-\theta)\right]-\left[2 d_{S} \tan i_{c}\right]^{2}=0$.
Hence, the limit of the precritical offset zone is also a conic section. By correspondence with equation (A4), the parameters defining this conic are

$$
e=\frac{\sin \phi}{\cos i_{c}}, \quad a=\left|\frac{2 d_{S} \sin i_{c} \cos i_{c} \cos \phi}{\cos ^{2} i_{c}-\sin ^{2} \phi}\right|, \quad \quad l_{c}=\frac{-2 d_{S} \sin ^{2} i_{c} \sin \phi}{\cos ^{2} i_{c}-\sin ^{2} \phi}
$$



Fig. 2.5. Head wave traveltime isochrons. In each panel, the isochrons (contour interval $=$ 25 ms ) are represented by solid curves, the critical offset distance by a dotted curve, and the crossover distance by a dashed curve. The source (asterisk) is located at the origin, and the straight line indicates the outcrop locus of the refractor. (a) Earth model: $v_{1}=1500$ $\mathrm{m} / \mathrm{s}, v_{2}=3000 \mathrm{~m} / \mathrm{s}, \phi=15^{\circ}, \theta=45^{\circ}, h=100 \mathrm{~m}$. Isochron eccentricity $=0.518$. (b) Earth model: $v_{1}=1500 \mathrm{~m} / \mathrm{s}, v_{2}=5796 \mathrm{~m} / \mathrm{s}, \phi=30^{\circ}, \theta=45^{\circ}, h=100 \mathrm{~m}$. Isochron eccentricity $=1.932$.


Fig. 2.5. (c) Earth model: $v_{1}=1500 \mathrm{~m} / \mathrm{s}, v_{2}=1655 \mathrm{~m} / \mathrm{s}, \phi=30^{\circ}, \theta=45^{\circ}, h=100 \mathrm{~m}$. Isochron eccentricity $=0.552$.

If $\phi<\pi / 2-i_{c}$, the eccentricity is less than one and the curve defined by (2.11) is an ellipse surrounding the source; its center is shifted by an amount $\left|l_{c}\right|$ in the downdip direction. However, if the interface dip angle exceeds the complement of the critical angle ( $\phi>\pi / 2-i_{c}$ ), then the critical offset curve is a hyperbola. This occurs, for example, in a high dip, low velocity contrast environment where rays critically refracted downdip never return to the surface.

The limit of the precritical offset zone is displayed in Figure 2.5 as a dotted curve. Fictitious arrival times within this curve are obtained by extrapolating or phantoming real traveltimes. Figure 2.5a illustrates elliptic traveltime isochrons ( $i_{c}>\phi$ ) and an elliptic critical distance curve ( $i_{c}+\phi<\pi / 2$ ). In Figure 2.5b, the isochrons are hyperbolic ( $i_{c}<\phi$ ) and the critical offset is still elliptic. Finally, in Figure 2.5c, the traveltime contours are
elliptic ( $i_{c}>\phi$ ) and the critical distance curve is hyperbolic ( $i_{c}+\phi>\pi / 2$ ). The remaining case (hyperbolic isochrons and hyperbolic critical offset) is not illustrated. The crossover distance $X_{\text {cross }}(\alpha)$, computed from equation (2.12), is depicted by a dashed curve in each panel. Since expression (2.12) cannot be put into the form of equation. (A4), this curve is not a conic section.

### 2.6 Inversion of traveltime isochrons

If the geometric parameters of a traveltime isochron can be determined from head wave arrival times observed on an areal recording array, then it is possible to invert for the earth model parameters. The principle axes of an elliptic or hyperbolic isochron are parallel to the dip and strike directions of the subsurface refracting horizon. Hence, the azimuth angle $\theta$ can be found. Furthermore, solving equations (2.28) for $\sin i_{c}, \sin \phi$, and $v_{1}|T-\tau|$ in terms of $e, a$, and $l_{c}$ yield

$$
\begin{gather*}
\sin i_{c}=\frac{1}{e} \sqrt{\frac{a^{2} e^{2}-l_{c}^{2}}{a^{2}-l_{c}^{2}}}, \quad \sin \phi=\sqrt{\frac{a^{2} e^{2}-l_{c}^{2}}{a^{2}-l_{c}^{2}}}  \tag{2.30a,b}\\
v_{1}|T-\tau|=\frac{1}{e} \sqrt{\left(1-e^{2}\right)\left(a^{2} e^{2}-l_{c}^{2}\right)} \tag{2.30c}
\end{gather*}
$$

For an ellipse, these expressions are all well defined because $e<1, a>l_{c}$, and $a e>l_{c}$. However, the relations remain valid for a hyperbola where $e>1, a<\left|l_{c}\right|$, and $a e<\left|l_{c}\right|$; the argument of each square root is still positive.

Equations (2.30a,b) indicate that the critical and dip angles can be determined from the geometric parameters of an observed traveltime isochron. If the overburden velocity $v_{1}$ is known from ancillary data, then (2.30c) can be solved for the intercept time $\tau$. The normal depth to the refractor below the shotpoint is then calculated via $d_{S}=v_{1} \tau / 2 \cos i_{c}$. However, an alternative approach considers the intercept time to be a measured quantity. The basic traveltime equation (2.10) indicates that intercept time is invariant with respect to recording profile azimuth $\alpha$. Thus, a value for $\tau$ can be obtained by linearly extrapolating
to zero offset the arrival times observed along any radial profile emanating from the source. Equation (2.30c) is then a single equation in the unknown layer velocity $v_{1}$. Solving for $v_{1}$ gives

$$
\begin{equation*}
v_{1}=\frac{\sqrt{\left(1-e^{2}\right)\left(a^{2} e^{2}-l_{c}^{2}\right)}}{e|T-\tau|} \tag{2.31}
\end{equation*}
$$

Hence, the overburden velocity can be calculated from the refraction data alone! With $v_{1}$ determined, the refractor velocity is obtained via $v_{2}=v_{1} / \sin i_{c}$. Interestingly, the above solution method requires the critically refracting interface to have nonzero dip. Otherwise, the eccentricity $e$ and origin-to-center distance $l_{c}$ vanish, and formula (2.30c) becomes indeterminate. Overburden velocity $v_{1}$ must then be obtained from independent measurements.

Practical issues related to fitting a conic section to a finite set of spatially sampled arrival times are not addressed here in any detail. Rather, the intention is to demonstrate the existence of a mathematical solution to the specified inverse problem. In principle, the situation is similiar to fitting a straight line segment to a set of error contaminated traveltimes observed along a line profile. However, there are obvious practical differences. Since a conic curve is nonlinear in its defining parameters, an iterative fitting procedure may be necessary. Finally, it may be possible to fit several conics with different isochron times $T$ to the measured data. Statistically robust estimates of the earth model parameters can then be obtained by using some form of averaging.

### 2.7 Conclusion

Although the earth model examined in this chapter is quite simple, there are undoubtedly many situations where it constitutes an adequate representation of geophysical reality. Also, the model is very useful as a pedagogical tool. Analysis of head wave propagation within the model yields closed form mathematical expressions for traveltime, critical and crossover distances, intercept time, and apparent refractor velocity. These expressions are generalizations of more familiar formulae that apply in one and two dimensional situations, and are valuable for forward modeling purposes.

Algebraic solutions of the traveltime inverse problem for the same earth model are also derived. There are numerous practical recording geometries where the analysis of critically refracted arrival times recorded on only two unreversed spreads can yield the three dimensional attitude, true velocity, and depth of a plane subsurface horizon. The well known method for determining refractor velocity in a 2D situation (measurement of apparent velocities on colinear forward and reverse profiles) applies directly to the more complicated 3D model. However, it is not necessary that the profiles be colinear; anti-parallel profiles are sufficient. This knowledge can be quite useful for the interpretation of apparent velocity data obtained from a 3D dipping interface. Extension of these methods to include multilayered earth models will be valuable for defining the minimum requirements for a successful inversion of three dimensional refraction traveltime data.

Finally, study of head wave arrivals recorded in an unconventional data acquisition geometry (i.e., an areal receiver array) reveals an interesting possibility for estimating overburden velocity from the refraction data alone. At present, this is primarily a theoretical result, but it does provide some justification for the numerical inversion results obtained in Chapter 4.

## CHAPTER 3

## HEAD WAVE TRAVELTIMES IN A THREE DIMENSIONAL MULTILAYERED EARTH

### 3.1 Introduction

There is a notable paucity of papers on the subject of head wave propagation in three dimensional, multilayered earth models. Chander (1977b) examines a model consisting of uniform velocity layers separated by plane interfaces with arbitrary strike and dip, and describes a method for calculating head wave traveltimes between specified source and receiver positions on a horizontal surface. If an array of receivers is colinear with the source, the head wave arrival time curve is a straight line. Hence, if two points on this line are established, then arrival times at all offsets can be determined simply by drawing the connecting straight line. Chander (1977b) locates the two initial points via raytracing techniques.

Chander's work is purely numerical and does not provide much insight into the dependence of head wave traveltime on the parameters that define the earth model. Moreover, it is restricted to conventional data acquisition geometries. Buried sources and/or receivers as well as nonprofile recording geometries require a more general treatment. Diebold's (1987) recent work constitutes the seminal contribution on this topic. He also considers a three dimensional multilayered earth, but derives traveltime formulae for reflected and critically refracted waves. These formulae are logical extensions of familiar traveltime expressions that are appropriate for one and two dimensional layered models. Thus, they offer the possibility for extending several known traveltime inversion techniques to accommodate 3 D planar structure. Unfortunately, Diebold's derivations are very ambiguous. Furthermore, his generalization to arbitrary source-receiver geometries yields an incorrect traveltime formula. Finally, he does not present a numerical technique for actually computing the traveltimes. These deficiencies are addressed in this chapter. Nevertheless, Diebold (1987) should be credited with an original contribution to traveltime analysis for this particular class of earth models.

This chapter provides a rigorous derivation of the three dimensional head wave traveltime formula. A related expression for the traveltime of a reflected wave propagating in the same earth model is obtained as a byproduct of the analysis. The mathematical proofs of the formulae are simplified by using a novel form of Snell's law of refraction and reflection. Various generalizations of the basic traveltime equation extend its applicability to arbitrary 3D recording geometries and/or mode-converted waves. Finally, a rapid numerical method for computing the arrival times of critical refractions is presented, and is illustrated with simulated examples from shallow refraction exploration and vertical seismic profiling (VSP).

### 3.2 Earth model

Consider an earth model consisting of a set of homogeneous and isotropic layers bounded by plane interfaces. In general, each interface may possess a three dimensional dipping attitude. The $i^{\text {th }}$ interface of the model is illustrated in Figure 3.1. $O$ is the origin of a right-handed, rectangular Cartesian coordinate system with orthonormal basis triad $\mathbf{i j k}$. The $x y$ plane is defined to be the horizontal plane and the depth coordinate $z$ increases in the downward direction. The locus of plane interface $i$ satisfies the equation

$$
\begin{equation*}
\mathbf{r} \cdot \mathbf{n}_{\boldsymbol{i}}=d_{i} \tag{3.1}
\end{equation*}
$$

where $\mathbf{n}_{\mathbf{i}}$ is a unit vector normal to the interface and $d_{i}$ is the perpendicular distance from $O$ to the interface. Figure 3.1 indicates that $\mathbf{n}_{\boldsymbol{i}}$ is conveniently described by two interface orientation angles:

$$
\mathbf{n}_{i}=\left(\sin \phi_{i} \cos \theta_{i}\right) \mathbf{i}+\left(\sin \phi_{i} \sin \theta_{i}\right) \mathbf{j}+\left(\cos \phi_{i}\right) \mathbf{k}
$$

$\phi_{i}\left(0 \leq \phi_{i}<\pi / 2\right)$ is the dip angle and $\theta_{i}\left(0 \leq \theta_{i}<2 \pi\right)$ is the azimuth angle of the interface. If the $+x$ and $+y$ axes are taken to point toward geographic north and east, respectively, then the interface strike angle is $\theta_{i}+\pi / 2$ (modulo $2 \pi$ ). Although these angular coordinates


Fig. 3.1. Orientation of the $i^{\text {th }}$ interface of a multilayered earth model.
are descriptive, a certain compactness in notation is achieved by specifying $\mathbf{n}_{i}$ in terms of its Cartesian components:

$$
\mathbf{n}_{i}=n_{i, x} \mathbf{i}+n_{i, y} \mathbf{j}+n_{i, z} \mathbf{k}
$$

with $n_{i, x}^{2}+n_{i, y}^{2}+n_{i, z}^{2}=1$. This convention is followed in the sequel.

Solving equation (3.1) for $z$ as a function of $x$ and $y$ yields the vertical depth of interface $i$ :

$$
\begin{equation*}
z_{i}(x, y)=z_{i}(0,0)-x\left[\frac{n_{i, x}}{n_{i, z}}\right]-y\left[\frac{n_{i, y}}{n_{i, z}}\right] \tag{3.2}
\end{equation*}
$$

where $z_{i}(0,0) \equiv d_{i} / n_{i, z}$ is the vertical depth of the $i^{t h}$ interface below the coordinate origin.
The surface (not necessarily horizontal) is interface 1 , and subsequent interfaces are numbered sequentially in the downward direction. Interface $i$ overlies layer $i$. The vertical thickness of layer $i$ is defined to be $h_{i}(x, y) \equiv z_{i+1}(x, y)-z_{i}(x, y)$. Thus

$$
\begin{equation*}
h_{i}(x, y)=h_{i}(0,0)+x\left[\frac{n_{i, x}}{n_{i, z}}-\frac{n_{i+1, x}}{n_{i+1, z}}\right]+y\left[\frac{n_{i, y}}{n_{i, z}}-\frac{n_{i+1, y}}{n_{i+1, z}}\right] \tag{3.3}
\end{equation*}
$$

where $h_{i}(0,0) \equiv z_{i+1}(0,0)-z_{i}(0,0)$ is the vertical thickness of the $i^{\text {th }}$ layer beneath the coordinate origin $O$.

Finally, the seismic wave propagation speed assigned to layer $i$ is given by $v_{i}$. This may be either the compressional wave speed $\alpha_{i}$ or the shear wave speed $\beta_{i}$. This flexibility allows the resulting traveltime equations to apply either to $P, S$, or mode-converted waves.

### 3.3 Raypath geometry

Initially, the analysis is restricted to the case where both the source and receiver are located on the surface. Generalization to an arbitrary data acquisition geometry is straightforward and is given in a later section. The horizontal coordinates of the point source $S$ and point receiver $R$ are $\left(x_{S}, y_{S}\right)$ and $\left(x_{R}, y_{R}\right)$, respectively. Their vertical coordinates are easily obtained from equation (3.2): $z_{S}=z_{1}\left(x_{S}, y_{S}\right)$ and $z_{R}=z_{1}\left(x_{R}, y_{R}\right)$.

In order to facilitate computation of the traveltime, the total head wave raypath is divided into three major portions: the downgoing, critically refracted, and upgoing paths. In Figure 3.2, these correspond to raypath segments $S P, P Q$, and $Q R$, respectively. The propagation time along each portion is calculated, and then all three are summed to obtain the surface-to-surface head wave traveltime.


Fig. 3.2. Schematic representation of the raypath of a wave critically refracted on interface $k$ of a multilayered earth model. $S$ and $R$ denote a surface source and receiver, respectively. In the 3D situation, the raypath is not confined to a plane.

Within each layer, the raypath is a straight line segment. On the downward portion of the raypath, the propagation direction within layer $i$ of the wave critically refracted at subsurface interface $k$ is described by the unit vector $p_{i k}$ :

$$
\mathbf{p}_{i k}=p_{i k, x} \mathbf{i}+p_{i k, y} \mathbf{j}+p_{i k, z} \mathbf{k}
$$

Similarly, the upward propagation direction within layer $i$ of the wave critically refracted
from interface $k$ is specified by another unit vector $q_{i k}$ :

$$
\mathbf{q}_{i k}=q_{i k, x^{1}}+q_{i k, y} \mathbf{j}+q_{i k, z} \mathbf{k}
$$

The complete head wave raypath is described by the set of unit vectors $\mathbf{p}_{i k}$ and $\mathbf{q}_{i k}(i=$ $1,2, \ldots, k-1$ ) together with a critically refracted propagation direction $\mathbf{p}_{k k}=\mathbf{q}_{k k}$.

At interface $i$ in the overburden, the wave is refracted in accordance with Snell's law. The situation for the downgoing wave as it encounters the $i^{t h}$ interface from above is depicted in Figure 3.3a. The plane of this diagram is the plane of incidence defined by the incident propagation direction $\mathbf{p}_{i-1, k}$ and the interface normal $\mathbf{n}_{\boldsymbol{i}}$. Snell's law of refraction consists of the following two conditions:
(i) the unit propagation vector of the transmitted ray $\left(\mathbf{p}_{i k}\right)$ is contained in the plane of incidence,
(ii) $\sin \mu / v_{i-1}=\sin \nu / v_{i}$, where $\mu$ and $\nu$ are positive acute angles measured from the interface normal to the incident and transmitted propagation directions $\mathbf{p}_{i-1, k}$ and $\mathbf{p}_{i k}$, respectively.

Both conditions are contained in the single vector equation

$$
\begin{equation*}
\frac{\mathbf{n}_{i} \times \mathbf{p}_{i-1, k}}{v_{i-1}}=\frac{\mathbf{n}_{i} \times \mathbf{p}_{i k}}{v_{i}} \tag{3.4}
\end{equation*}
$$

The vector formed by these cross products points out of the plane of the diagram in Figure 3.3a. In component form, equation (3.4) is


Fig. 3.3. Noncritical refraction of the raypath at interface $i$ in the overburden. (a) Downgoing raypath. (b) Upgoing raypath. The unit vector $\mathbf{n}_{i}$ is normal to the interface.

$$
\begin{align*}
& \frac{1}{v_{i-1}}\left[n_{i, y} p_{i-1, k, z}-n_{i, z} p_{i-1, k, y}\right]=\frac{1}{v_{i}}\left[n_{i, y} p_{i k, z}-n_{i, z} p_{i k, y}\right]  \tag{3.5a}\\
& \frac{1}{v_{i-1}}\left[n_{i, z} p_{i-1, k, x}-n_{i, x} p_{i-1, k, z}\right]=\frac{1}{v_{i}}\left[n_{i, z} p_{i k, x}-n_{i, x} p_{i k, z}\right],  \tag{3.5b}\\
& \frac{1}{v_{i-1}}\left[n_{i, x} p_{i-1, k, y}-n_{i, y} p_{i-1, k, x}\right]=\frac{1}{v_{i}}\left[n_{i, x} p_{i k, y}-n_{i, y} p_{i k, x}\right] \tag{3.5c}
\end{align*}
$$

Similarly, when the upgoing wave encounters interface $i$ from below, Snell's law in the form

$$
\begin{equation*}
\frac{\mathbf{n}_{i} \times \mathbf{q}_{i-1, k}}{v_{i-1}}=\frac{\mathbf{n}_{i} \times \mathbf{q}_{i k}}{v_{i}} \tag{3.6}
\end{equation*}
$$

holds (see Figure 3.3b). In this case, the angles $\mu$ and $\nu$ exceed $\pi / 2$ radians. The component form of expression (3.6) is analogous to equations (3.5).

The three dimensional statement of Snell's law of refraction given by the above expressions is quite different from the form typically used in raytracing applications (Sorrells et al., 1971; Shah, 1973; Chander, 1977b). However, it can be demonstrated that these expressions are equivalent to the raytracing formulae (equations (3.27) and (3.28) below). The value of the current formulation is that it leads to a substantial simplification in the mathematical proof of the traveltime equations.

### 3.4 Traveltime derivation

### 3.4.1 Downgoing traveltime

An expression for the traveltime increment of the downgoing wave as it traverses layer $i$ is derived first. Let the position vectors $\mathbf{r}_{\boldsymbol{i}}$ and $\mathbf{r}_{\boldsymbol{i}+1}$ denote the intersection points of the downgoing ray with interfaces $i$ and $i+1$, respectively. Then $\mathbf{r}_{i+1}=\mathbf{r}_{i}+l_{i} \mathbf{p}_{i k}$, where
$l_{i}$ is the length of the (straight line) raypath segment within layer $i$. Solving for $l_{i}$ gives $l_{i}=\mathbf{p}_{i k} \cdot\left(\mathbf{r}_{i+1}-\mathbf{r}_{i}\right)$. In component form, this expression is

$$
l_{i}=p_{i k, x}\left(x_{i+1}-x_{i}\right)+p_{i k, y}\left(y_{i+1}-y_{i}\right)+p_{i k, z}\left(z_{i+1}-z_{i}\right)
$$

The traveltime increment is obtained by dividing the path length segment $l_{i}$ by the layer velocity $v_{i}$ :

$$
t_{i}=\frac{l_{i}}{v_{i}}=\frac{p_{i k, x}}{v_{i}}\left(x_{i+1}-x_{i}\right)+\frac{p_{i k, y}}{v_{i}}\left(y_{i+1}-y_{i}\right)+\frac{p_{i k, z}}{v_{i}}\left(z_{i+1}-z_{i}\right)
$$

The vertical ( $z$ ) coordinates of the intersection points can be expressed in terms of the horizontal ( $x, y$ ) coordinates by using the equation for a dipping plane interface. Equation (3.2) yields

$$
\begin{gathered}
z_{i} \equiv z_{i}\left(x_{i}, y_{i}\right)=z_{i}(0,0)-x_{i}\left[\frac{n_{i, x}}{n_{i, z}}\right]-y_{i}\left[\frac{n_{i, y}}{n_{i, z}}\right] \\
z_{i+1} \equiv z_{i+1}\left(x_{i+1}, y_{i+1}\right)=z_{i+1}(0,0)-x_{i+1}\left[\frac{n_{i+1, x}}{n_{i+1, z}}\right]-y_{i+1}\left[\frac{n_{i+1, y}}{n_{i+1, z}}\right]
\end{gathered}
$$

Hence

$$
z_{i+1}-z_{i}=h_{i}(0,0)+x_{i}\left[\frac{n_{i, x}}{n_{i, z}}\right]+y_{i}\left[\frac{n_{i, y}}{n_{i, z}}\right]-x_{i+1}\left[\frac{n_{i+1, x}}{n_{i+1, z}}\right]-y_{i+1}\left[\frac{n_{i+1, y}}{n_{i+1, z}}\right]
$$

where $h_{i}(0,0)=z_{i+1}(0,0)-z_{i}(0,0)$ has been used. Substituting this result into the equation for $t_{i}$ and grouping terms yields the required expression for the traveltime increment:

$$
\begin{gathered}
t_{i}=\frac{h_{i}(0,0) p_{i k, z}}{v_{i}}+\frac{x_{i+1}}{n_{i+1, z}}\left[\frac{n_{i+1, z} p_{i k, x}-n_{i+1, x} p_{i k, z}}{v_{i}}\right]+\frac{y_{i+1}}{n_{i+1, z}}\left[\frac{n_{i+1, z} p_{i k, y}-n_{i+1, y} p_{i k, z}}{v_{i}}\right] \\
-\frac{x_{i}}{n_{i, z}}\left[\frac{n_{i, z} p_{i k, x}-n_{i, x} p_{i k, z}}{v_{i}}\right]-\frac{y_{i}}{n_{i, z}}\left[\frac{n_{i, z} p_{i k, y}-n_{i, y} p_{i k, z}}{v_{i}}\right] .
\end{gathered}
$$

The total downgoing traveltime is obtained by summing all of the individual traveltime increments $t_{i}, i=1,2, \ldots, k-1$ :

$$
\begin{gathered}
T_{d o w n}=\sum_{i=1}^{k-1} \frac{h_{i}(0,0) p_{i k, z}}{v_{i}} \\
+\sum_{i=1}^{k-1}\left\{\frac{x_{i+1}}{n_{i+1, z}}\left[\frac{n_{i+1, z} p_{i k, x}-n_{i+1, x} p_{i k, z}}{v_{i}}\right]+\frac{y_{i+1}}{n_{i+1, z}}\left[\frac{n_{i+1, z} p_{i k, y}-n_{i+1, y} p_{i k, z}}{v_{i}}\right]\right\} \\
\quad-\sum_{i=1}^{k-1}\left\{\frac{x_{i}}{n_{i, z}}\left[\frac{n_{i, z} p_{i k, x}}{v_{i}}-\frac{n_{i, x} p_{i k, z}}{v_{i}}\right]+\frac{y_{i}}{n_{i, z}}\left[\frac{n_{i, z} p_{i k, y}}{v_{i}}-\frac{n_{i, y} p_{i k, z}}{v_{i}}\right]\right\}
\end{gathered}
$$

The sum involving $x_{i+1}$ and $y_{i+1}$ is now reindexed and combined with the other sum. The result is

$$
\begin{gathered}
T_{d o w n}=\sum_{i=1}^{k-1} \frac{h_{i}(0,0) p_{i k, z}}{v_{i}} \\
+\left[\frac{x_{k}\left(n_{k, z} p_{k-1, k, x}-n_{k, x} p_{k-1, k, z}\right)+y_{k}\left(n_{k, z} p_{k-1, k, y}-n_{k, y} p_{k-1, k, z}\right)}{v_{k-1} n_{k, z}}\right] \\
\quad-\left[\frac{x_{1}\left(n_{1, z} p_{1 k, x}-n_{1, x} p_{1 k, z}\right)+y_{1}\left(n_{1, z} p_{1 k, y}-n_{1, y} p_{1 k, z}\right)}{v_{1} n_{1, z}}\right] \\
+\sum_{i=2}^{k-1} \frac{x_{i}}{n_{i, z}}\left\{\left[\frac{n_{i, z} p_{i-1, k, x}-n_{i, x} p_{i-1, k, z}}{v_{i-1}}\right]-\left[\frac{n_{i, z} p_{i k, x}-n_{i, x} p_{i k, z}}{v_{i}}\right]\right\} \\
+\sum_{i=2}^{k-1} \frac{y_{i}}{n_{i, z}}\left\{\left[\frac{n_{i, z} p_{i-1, k, y}-n_{i, y} p_{i-1, k, z}}{v_{i-1}}\right]-\left[\frac{n_{i, z} p_{i k, y}-n_{i, y} p_{i k, z}}{v_{i}}\right]\right\}
\end{gathered}
$$

At interfaces $2,3, \ldots, k-1$ in the overburden, Snell's law of refraction must apply. Equations (3.5a,b) then imply that all terms in the summations involving $x_{i}$ and $y_{i}$ vanish! The downgoing traveltime from $\left(x_{1}, y_{1}\right)=\left(x_{S}, y_{S}\right)$ to $\left(x_{k}, y_{k}\right)=\left(x_{P}, y_{P}\right)$ reduces to

$$
\begin{gather*}
T_{d o w n}=\sum_{i=1}^{k-1} \frac{h_{i}(0,0) p_{i k, z}}{v_{i}} \\
+\left[\frac{x_{P}\left(n_{k, z} p_{k-1, k, x}-n_{k, x} p_{k-1, k, z}\right)+y_{P}\left(n_{k, z} p_{k-1, k, y}-n_{k, y} p_{k-1, k, z}\right)}{v_{k-1} n_{k, z}}\right] \\
-\left[\frac{x_{S}\left(n_{1, z} p_{1 k, x}-n_{1, x} p_{1 k, z}\right)+y_{S}\left(n_{1, z} p_{1 k, y}-n_{1, y} p_{1 k, z}\right)}{v_{1} n_{1, z}}\right] \tag{3.7}
\end{gather*}
$$

### 3.4.2 Upgoing traveltime

The traveltime along the upward propagating portion of the total raypath is derived by similiar techniques. Snell's law in the form (3.6) is used at each interface in the overburden. The result is

$$
\begin{gather*}
T_{u p}=-\sum_{i=1}^{k-1} \frac{h_{i}(0,0) q_{i k, z}}{v_{i}} \\
-\left[\frac{x_{Q}\left(n_{k, z} q_{k-1, k, x}-n_{k, x} q_{k-1, k, z}\right)+y_{Q}\left(n_{k, z} q_{k-1, k, y}-n_{k, y} q_{k-1, k, z}\right.}{v_{k-1} n_{k, z}}\right] \\
+\left[\frac{x_{R}\left(n_{1, z} q_{1 k, x}-n_{1, x} q_{1 k, z}\right)+y_{R}\left(n_{1, z} q_{1 k, y}-n_{1, y} q_{1 k, z}\right)}{v_{1} n_{1, z}}\right] \tag{3.8}
\end{gather*}
$$

### 3.4.3 Critically refracted traveltime

The final traveltime increment needed for the derivation corresponds to the critically refracted segment of the total raypath. Let the position vectors $\mathbf{r}_{P}$ and $\mathbf{r}_{Q}$ refer to the intersection points of the downgoing and upgoing portions of the raypath with interface $k$, respectively. The critically refracted raypath segment is a straight line connecting these two points (and thus lying entirely within the plane of interface $k$ ). Then $\mathbf{r}_{Q}=\mathbf{r}_{P}+l_{k} \mathbf{p}_{k k}$ and hence $l_{k}=\mathbf{p}_{k k} \cdot\left(\mathbf{r}_{Q}-\mathbf{r}_{P}\right)$. The propagation time along this path length is

$$
t_{k}=\frac{l_{k}}{v_{k}}=\frac{p_{k k, x}}{v_{k}}\left(x_{Q}-x_{P}\right)+\frac{p_{k k, y}}{v_{k}}\left(y_{Q}-y_{P}\right)+\frac{p_{k k, z}}{v_{k}}\left(z_{Q}-z_{P}\right)
$$

Since points $P$ and $Q$ reside on the same plane interface, equation (3.2) yields

$$
z_{Q}-z_{P}=z_{k}\left(x_{Q}, y_{Q}\right)-z_{k}\left(x_{P}, y_{P}\right)=-\left(x_{Q}-x_{P}\right)\left[\frac{n_{k, x}}{n_{k, z}}\right]-\left(y_{Q}-y_{P}\right)\left[\frac{n_{k, y}}{n_{k, z}}\right]
$$

The expression for the critically refracted traveltime increment then reduces to

$$
\begin{equation*}
T_{c r i t} \equiv t_{k}=\left[\frac{\left(x_{Q}-x_{P}\right)\left(n_{k, z} p_{k k, x}-n_{k, x} p_{k k, z}\right)+\left(y_{Q}-y_{P}\right)\left(n_{k, z} p_{k k, y}-n_{k, y} p_{k k, z}\right)}{v_{k} n_{k, z}}\right] \tag{3.9}
\end{equation*}
$$

### 3.4.4 Total traveltime

The total surface-to-surface traveltime of the wave critically refracted on interface $k$ is obtained by adding the traveltime contributions along the downgoing, critically refracted, and upgoing raypath portions: $T_{t o t a l}=T_{\text {down }}+T_{c r i t}+T_{u p}$. Summing expressions (3.7), (3.8), and (3.9) yields

$$
\begin{gathered}
T_{k}\left(x_{S}, y_{S}, x_{R}, y_{R}\right)=\sum_{i=1}^{k-1} \frac{h_{i}(0,0)\left(p_{i k, z}-q_{i k, z}\right)}{v_{i}} \\
+\left[\frac{x_{R}\left(n_{1, z} q_{1 k, x}-n_{1, x} q_{1 k, z}\right)+y_{R}\left(n_{1, z} q_{1 k, y}-n_{1, y} q_{1 k, z}\right)}{v_{1} n_{1, z}}\right] \\
-\left[\frac{x_{S}\left(n_{1, z} p_{1 k, x}-n_{1, x} p_{1 k, z}\right)+y_{S}\left(n_{1, z} p_{1 k, y}-n_{1, y} p_{1 k, z}\right)}{v_{1} n_{1, z}}\right] \\
+F\left(x_{P}, y_{P}, x_{Q}, y_{Q}\right)
\end{gathered}
$$

The quantity $F$ depends on the horizontal coordinates of the two points of critical refraction and is given by

$$
\begin{gathered}
F\left(x_{P}, y_{P}, x_{Q}, y_{Q}\right)= \\
\frac{x_{P}}{n_{k, z}}\left[\frac{1}{v_{k-1}}\left(n_{k, z} p_{k-1, k, x}-n_{k, x} p_{k-1, k, z}\right)-\frac{1}{v_{k}}\left(n_{k, z} p_{k k, x}-n_{k, x} p_{k k, z}\right)\right] \\
+\frac{y_{P}}{n_{k, z}}\left[\frac{1}{v_{k-1}}\left(n_{k, z} p_{k-1, k, y}-n_{k, y} p_{k-1, k, z}\right)-\frac{1}{v_{k}}\left(n_{k, z} p_{k k, y}-n_{k, y} p_{k k, z}\right)\right] \\
-\frac{x_{Q}}{n_{k, z}}\left[\frac{1}{v_{k-1}}\left(n_{k, z} q_{k-1, k, x}-n_{k, x} q_{k-1, k, z}\right)-\frac{1}{v_{k}}\left(n_{k, z} p_{k k, x}-n_{k, x} p_{k k, z}\right)\right] . \\
-\frac{y_{Q}}{n_{k, z}}\left[\frac{1}{v_{k-1}}\left(n_{k, z} q_{k-1, k, y}-n_{k, y} q_{k-1, k, z}\right)-\frac{1}{v_{k}}\left(n_{k, z} p_{k k, y}-n_{k, y} p_{k k, z}\right)\right] .
\end{gathered}
$$

Since the wave is critically refracted at interface $k$, the propagation direction vectors $\mathbf{p}_{k k}$ and $\mathbf{q}_{k k}$ are identical. Then, requiring Snell's law to be satisfied at points $P$ and $Q$ results
in $F\left(x_{P}, y_{P}, x_{Q}, y_{Q}\right)=0$. The final formula for the surface-to-surface traveltime of the head wave becomes

$$
\begin{gather*}
T_{k}\left(x_{S}, y_{S}, x_{R}, y_{R}\right)=\sum_{i=1}^{k-1} \frac{h_{i}(0,0)\left(p_{i k, z}-q_{i k, z}\right)}{v_{i}} \\
+\left[\frac{x_{R}\left(n_{1, z} q_{1 k, x}-n_{1, x} q_{1 k, z}\right)+y_{R}\left(n_{1, z} q_{1 k, y}-n_{1, y} q_{1 k, z}\right)}{v_{1} n_{1, z}}\right] \\
-\left[\frac{x_{S}\left(n_{1, z} p_{1 k, x}-n_{1, x} p_{1 k, z}\right)+y_{S}\left(n_{1, z} p_{1 k, y}-n_{1, y} p_{1 k, z}\right)}{v_{1} n_{1, z}}\right] . \tag{3.10}
\end{gather*}
$$

### 3.4.5 Variants of the basic formula

Obviously, if the source or receiver is located at the coordinate origin, then the traveltime formula (3.10) simplifies considerably. Another simplification arises with a horizontal surface ( $n_{1, x}=n_{1, y}=0, n_{1, z}=1$ ). Expression (3.10) reduces to

$$
\begin{align*}
& T_{k}\left(x_{S}, y_{S}, x_{R}, y_{R}\right)=\sum_{i=1}^{k-1} \frac{h_{i}(0,0)\left(p_{i k, z}-q_{i k, z}\right)}{v_{i}} \\
& +\left[\frac{\left(x_{R} q_{1 k, x}+y_{R} q_{1 k, y}\right)-\left(x_{S} p_{1 k, x}+y_{S} p_{1 k, y}\right)}{v_{1}}\right] \tag{3.11}
\end{align*}
$$

This is equivalent to Diebold's (1987) surface-to-surface traveltime formula.
Conventionally, seismic refraction traveltime is presented as a function of the sourcereceiver offset distance. The current expression is easily converted to this form by specifying the receiver position in terms of an offset distance $X(X \geq 0)$ and an azimuth angle $\Psi$ $(0 \leq \Psi<2 \pi)$ relative to the source. The receiver coordinates are

$$
x_{R}=x_{S}+X \cos \Psi, \quad y_{R}=y_{S}+X \sin \Psi
$$

Substituting these expressions into equation (3.10) yields

$$
\begin{gather*}
T_{k}\left(x_{S}, y_{S}, X, \Psi\right)=\sum_{i=1}^{k-1} \frac{h_{i}(0,0)\left(p_{i k, z}-q_{i k, z}\right)}{v_{i}} \\
-x_{S}\left[\frac{n_{1, z}\left(p_{1 k, x}-q_{1 k, x}\right)-n_{1, x}\left(p_{1 k, z}-q_{1 k, z}\right)}{v_{1} n_{1, z}}\right] \\
-y_{S}\left[\frac{n_{1, z}\left(p_{1 k, y}-q_{1 k, y}\right)-n_{1, y}\left(p_{1 k, z}-q_{1 k, z}\right)}{v_{1} n_{1, z}}\right] \\
+X\left[\frac{\cos \Psi\left(n_{1, z} q_{1 k, x}-n_{1, x} q_{1 k, z}\right)+\sin \Psi\left(n_{1, z} q_{1 k, y}-n_{1, y} q_{1 k, z}\right)}{v_{1} n_{1, z}}\right] . \tag{3.12}
\end{gather*}
$$

Note that $X$ is the horizontal distance between source and receiver; the actual distance is larger since it is measured within the plane of interface 1 . It is straightforward to demonstrate that the true source-receiver distance is $L=X \sqrt{1+\tan ^{2} \phi_{1} \cos ^{2}\left(\Psi-\theta_{1}\right)}$, where $\phi_{1}$ and $\theta_{1}$ are the dip and azimuth angles of the surface.

Equation (3.12) is an extension of the common 'slope and intercept' traveltime formula for head waves to three dimensional, multilayered earth models. For the particular case of a model with only two layers and a horizontal surface, it can be shown that this equation reduces to the closed form expression (2.10) derived in the previous chapter. This serves as an important check on the correctness of the general formula. However, the proof entails some cumbersome algebra, and hence is relegated to Appendix B.

Finally, consider the specialization of the general traveltime formula to a two dimensional earth model. In the model parameterization utilized here, all of the $y$ components of the interface normal vectors vanish (equivalently, all interface azimuth angles are restricted to $\theta_{i}=0$ or $\theta_{i}=\pi$ ). Additionally, the recording profile must be oriented perpendicular to the strike directions of the subsurface horizons. Hence, $\Psi=0$ or $\pi$ also. If this second condition
is not satisfied, the wave propagation vectors $\mathbf{p}_{i k}$ and $\mathbf{q}_{i k}$ are not confined to the $x z$ plane and a full three dimensional treatment is necessary.

Since many previous investigators have assumed a horizontal surface, equation (3.11) is used as the point of departure for the analysis. Setting $y_{S}=y_{R}=0$ gives

$$
\begin{equation*}
T_{k}\left(x_{S}, x_{R}\right)=\sum_{i=1}^{k-1} \frac{h_{i}(0,0)\left(p_{i k, z}-q_{i k, z}\right)}{v_{i}}+\left[\frac{x_{R} q_{1 k, x}-x_{S} p_{1 k, x}}{v_{1}}\right] \tag{3.13}
\end{equation*}
$$

where $p_{i k, x}^{2}+p_{i k, z}^{2}=q_{i k, x}^{2}+q_{i k, z}^{2}=1$. This expression is compatible with analogous equations developed by Diebold and Stoffa (1981) and Diebold (1987). If the source is located at the coordinate origin, then (3.13) is also consistent with earlier two dimensional head wave traveltime formulae published by Dooley (1952), Adachi (1954), Ocola (1972), and Johnson (1976). All of these investigators prescribe vertical layer thicknesses, either beneath the origin or the shotpoint. In contrast, Ewing et al. (1939) and Mota (1954) measure thickness normal to the basal interface bounding a layer. Hence, their traveltime equations, although designed to treat an equivalent situation, differ in mathematical detail.

### 3.5 Generalizations of the traveltime formula

Heretofore, both the source and the receiver have been restricted to the surface. More versatile formulae are needed to model data acquisition geometries with buried sources and/or receivers. These situations arise in surface-to-borehole, borehole-to-surface, and borehole-to-borehole seismic experiments, as well as with placement of sources and/or receivers in underground mines.

### 3.5.1 Source and receiver on separate interfaces

Let the source $S$ be located on the $j^{\text {th }}$ interface with $1 \leq j<k$. The downgoing traveltime is obtained by summing the layer traveltime increments $t_{i}, i=j, j+1, \ldots, k-1$.

Equation (3.7) generalizes to

$$
\begin{gather*}
T_{d o w n}=\sum_{i=j}^{k-1} \frac{h_{i}(0,0) p_{i k, z}}{v_{i}} \\
+\left[\frac{x_{P}\left(n_{k, z} p_{k-1, k, x}-n_{k, x} p_{k-1, k, z}\right)+y_{P}\left(n_{k, z} p_{k-1, k, y}-n_{k, y} p_{k-1, k, z}\right)}{v_{k-1} n_{k, z}}\right] \\
-\left[\frac{x_{S}\left(n_{j, z} p_{j k, x}-n_{j, x} p_{j k, z}\right)+y_{S}\left(n_{j, z} p_{j k, y}-n_{j, y} p_{j k, z}\right)}{v_{j} n_{j, z}}\right] . \tag{3.14}
\end{gather*}
$$

Similarly, if the receiver $R$ is located on the $l^{t h}$ interface ( $1 \leq l<k$ ), then the upgoing traveltime becomes

$$
\begin{gather*}
T_{u p}=-\sum_{i=l}^{k-1} \frac{h_{i}(0,0) q_{i k, z}}{v_{i}} \\
-\left[\frac{x_{Q}\left(n_{k, z} q_{k-1, k, x}-n_{k, x} q_{k-1, k, z}\right)+y_{Q}\left(n_{k, z} q_{k-1, k, y}-n_{k, y} q_{k-1, k, z}\right)}{v_{k-1} n_{k, z}}\right] \\
+\left[\frac{x_{R}\left(n_{l, z} q_{l k, x}-n_{l, x} q_{l k, z}\right)+y_{R}\left(n_{l, z} q_{l k, y}-n_{l, y} q_{l k, z}\right)}{v_{l} n_{l, z}}\right] \tag{3.15}
\end{gather*}
$$

The critically refracted traveltime increment is still given by equation (3.9). Summing this expression together with (3.14) and (3.15) yields the total traveltime:

$$
\begin{align*}
& T_{k}\left(x_{S}, y_{S}, z_{S}, x_{R}, y_{R}, z_{R}\right)=\sum_{i=j}^{k-1} \frac{h_{i}(0,0) p_{i k, z}}{v_{i}}-\sum_{i=l}^{k-1} \frac{h_{i}(0,0) q_{i k, z}}{v_{i}} \\
& \quad+\left[\frac{x_{R}\left(n_{l, z} q_{l k, x}-n_{l, x} q_{l k, z}\right)+y_{R}\left(n_{l, z} q_{l k, y}-n_{l, y} q_{l k, z}\right)}{v_{l} n_{l, z}}\right] \\
& \quad-\left[\frac{x_{S}\left(n_{j, z} p_{j k, x}-n_{j, x} p_{j k, z}\right)+y_{S}\left(n_{j, z} p_{j k, y}-n_{j, y} p_{j k, z}\right)}{v_{j} n_{j, z}}\right] \tag{3.16}
\end{align*}
$$

where $z_{S}=z_{j}\left(x_{S}, y_{S}\right)$ and $z_{R}=z_{l}\left(x_{R}, y_{R}\right)$. This is the proper expression for head wave traveltime when source and receiver are located on different interfaces of the model. It differs significantly from the analogous formula published by Diebold (1987). His expression is actually a special case of the general equation (3.16); in particular, it is only valid if both the source and receiver interfaces are horizontal ( $n_{j, x}=n_{j, y}=n_{l, x}=n_{l, y}=0, n_{j, z}=n_{l, z}=1$ ). The difference between these two formulae is clearly revealed by a detailed examination of a simple two dimensional earth model in Appendix C. The analysis demonstrates that equation (3.16) reduces to the known traveltime solution for this situation, whereas Diebold's expression yields an erroneous result.

### 3.5.2 Arbitrary source and receiver locations

A further generalization is obtained by allowing the source and receiver to be located within designated layers. Assume that the source is located in layer $j$ at a vertical depth $d_{S}$ below the the immediately overlying interface (the $j^{t h}$ ). Similarly, let the receiver be located within layer $l$ at a depth $d_{R}$ beneath interface $l$. These incremental source and receiver depths must satisfy $0 \leq d_{S} \leq h_{j}\left(x_{S}, y_{S}\right)$ and $0 \leq d_{R} \leq h_{l}\left(x_{R}, y_{R}\right)$, respectively. The previously developed techniques can be used to derive the head wave traveltime for this situation. Traveltime increments induced by the source layer $j$ on the downward path and the receiver layer $l$ on the upward path must be treated separately, because the wave does not propagate across the full thickness of each layer. The result of the analysis is

$$
\begin{gather*}
T_{k}\left(x_{S}, y_{S}, z_{S}, x_{R}, y_{R}, z_{R}\right)=\sum_{i=j}^{k-1} \frac{h_{i}(0,0) p_{i k, z}}{v_{i}}-\frac{d_{S} p_{j k, z}}{v_{j}}-\sum_{i=l}^{k-1} \frac{h_{i}(0,0) q_{i k, z}}{v_{i}}+\frac{d_{R} q_{l k, z}}{v_{l}} \\
+\left[\frac{x_{R}\left(n_{l, z} q_{l k, x}-n_{l, x} q_{l k, z}\right)+y_{R}\left(n_{l, z} q_{l k, y}-n_{l, y} q_{l, z}\right)}{v_{l} n_{l, z}}\right] \\
 \tag{3.17}\\
-\left[\frac{x_{S}\left(n_{j, z} p_{j k, x}-n_{j, x} p_{j k, z}\right)+y_{S}\left(n_{j, z} p_{j k, y}-n_{j, y} p_{j k, z}\right)}{v_{j} n_{j, z}}\right]
\end{gather*}
$$

where $z_{S}=z_{j}\left(x_{S}, y_{S}\right)+d_{S}$ and $z_{R}=z_{l}\left(x_{R}, y_{R}\right)+d_{R}$. Note that the prior expression (3.16) is recovered in the limit as $d_{S} \rightarrow 0$ and $d_{R} \rightarrow 0$, as expected. As an additional check, examine the case where the source and receiver approach the basal interfaces of their respective layers (interfaces $j+1$ and $l+1$ ). The layer thicknesses $h_{j}(0,0)$ and $h_{l}(0,0)$ at the origin can be rewritten in terms of the thicknesses at the source and receiver positions ( $h_{j}\left(x_{S}, y_{S}\right)$ and $h_{l}\left(x_{R}, y_{R}\right)$ ) via expression (3.3). Then, by appealing to Snell's law at each basal interface, equation (3.17) is recast as

$$
\begin{gather*}
T_{k}\left(x_{S}, y_{S}, z_{S}, x_{R}, y_{R}, z_{R}\right)=\sum_{i=j+1}^{k-1} \frac{h_{i}(0,0) p_{i k, z}}{v_{i}}+\frac{\left[h_{j}\left(x_{S}, y_{S}\right)-d_{S}\right] p_{j k, z}}{v_{j}} \\
-\sum_{i=l+1}^{k-1} \frac{h_{i}(0,0) q_{i k, z}}{v_{i}}-\frac{\left[h_{l}\left(x_{R}, y_{R}\right)-d_{R}\right] q_{l k, z}}{v_{l}} \\
+\left[\frac{x_{R}\left(n_{l+1, z} q_{l+1, k, x}-n_{l+1, x} q_{l+1, k, z}\right)+y_{R}\left(n_{l+1, z} q_{l+1, k, y}-n_{l+1, y} q_{l+1, k, z}\right)}{v_{l+1} n_{l+1, z}}\right] \\
-\left[\frac{x_{S}\left(n_{j+1, z} p_{j+1, k, x}-n_{j+1, x} p_{j+1, k, z}\right)+y_{S}\left(n_{j+1, z} p_{j+1, k, y}-n_{j+1, y} p_{j+1, k, z}\right)}{v_{j+1} n_{j+1, z}}\right] . \tag{3.18}
\end{gather*}
$$

Comparing equations (3.18) and (3.16) indicates that the proper form of the traveltime expression is obtained in the limit as $d_{S} \rightarrow h_{j}\left(x_{S}, y_{S}\right)$ and $d_{R} \rightarrow h_{l}\left(x_{R}, y_{R}\right)$.

An additional benefit accrues from separating the downward and upward sums in the traveltime formulae: asymmetric wave propagation paths can be treated. An asymmetric raypath is defined as one where the mode of downgoing wave propagation in the $i^{\text {th }}$ layer differs from the mode of upgoing wave propagation across the same layer. Strictly, different symbols should be used to designate the wave speeds within layer $i$ in the downward and upward sums (e.g., $v_{i}^{d}$ and $v_{i}^{u}$ for the velocities of the downgoing and upgoing waves, respectively). However, this complication is avoided for the time being in order to maintain notational simplicity. The velocity $v_{i}$ appearing in each sum is simply interpreted as the propagation speed of the the desired mode ( P or S ) across layer i. Equations (3.17) and (3.18) then constitute general formulae for point-to-point traveltimes of head waves propagating in a three dimensional layered earth model.

### 3.5.3 Arbitrary reference points for layer thickness

Individual layer thicknesses enter the traveltime expressions evaluated at the coordinate origin $O$. An alternate form of the traveltime equation is characterized by layer thicknesses specified below the source and the receiver. This variant is particularly suitable for the time term, delay time, and reciprocal time inversion methods. Hence, the previous derivation is now modified to incorporate layer thicknesses prescribed at arbitrary reference locations; these points can then be specialized to the source and receiver positions. The resulting traveltime expression forms the point of departure for a three dimensional extension of the aforementioned inversion techniques.

The depth of the $i^{t h}$ interface, referred to an arbitrary location $A$ with horizontal coordinates $\left(x_{A}, y_{A}\right)$, is given by

$$
\begin{equation*}
z_{i}(x, y)=z_{i}\left(x_{A}, y_{A}\right)-\left(x-x_{A}\right)\left[\frac{n_{i, x}}{n_{i, z}}\right]-\left(y-y_{A}\right)\left[\frac{n_{i, y}}{n_{i, z}}\right] \tag{3.19}
\end{equation*}
$$

Consider the downgoing raypath first. The previous expression for the traveltime increment $t_{i}$ induced by wave propagation across the $i^{\text {th }}$ layer is trivially modified to

$$
t_{i}=\frac{p_{i k, x}}{v_{i}}\left[\left(x_{i+1}-x_{A}\right)-\left(x_{i}-x_{A}\right)\right]+\frac{p_{i k, y}}{v_{i}}\left[\left(y_{i+1}-y_{A}\right)-\left(y_{i}-y_{A}\right)\right]+\frac{p_{i k, z}}{v_{i}}\left(z_{i+1}-z_{i}\right) .
$$

The vertical ( $z$ ) coordinates of the ray intersection points are now expressed in terms of the corresponding horizontal coordinates by using equation (3.19):

$$
\begin{gathered}
z_{i+1}-z_{i}=h_{i}\left(x_{A}, y_{A}\right)+\left(x_{i}-x_{A}\right)\left[\frac{n_{i, x}}{n_{i, z}}\right]+\left(y_{i}-y_{A}\right)\left[\frac{n_{i, y}}{n_{i, z}}\right] \\
-\left(x_{i+1}-x_{A}\right)\left[\frac{n_{i+1, x}}{n_{i+1, z}}\right]-\left(y_{i+1}-y_{A}\right)\left[\frac{n_{i+1, y}}{n_{i+1, z}}\right]
\end{gathered}
$$

where $h_{i}\left(x_{A}, y_{A}\right) \equiv z_{i+1}\left(x_{A}, y_{A}\right)-z_{i}\left(x_{A}, y_{A}\right)$ is the vertical thickness of layer $i$ below the reference point $A$. Substituting this expression into the equation for $t_{i}$ yields

$$
\begin{gathered}
t_{i}=\frac{h_{i}\left(x_{A}, y_{A}\right) p_{i k, z}}{v_{i}} \\
+\frac{\left(x_{i+1}-x_{A}\right)}{n_{i+1, z}}\left[\frac{n_{i+1, z} p_{i k, x}-n_{i+1, x} p_{i k, z}}{v_{i}}\right]+\frac{\left(y_{i+1}-y_{A}\right)}{n_{i+1, z}}\left[\frac{n_{i+1, z} p_{i k, y}-n_{i+1, y} p_{i k, z}}{v_{i}}\right] \\
-\frac{\left(x_{i}-x_{A}\right)}{n_{i, z}}\left[\frac{n_{i, z} p_{i k, x}-n_{i, x} p_{i k, z}}{v_{i}}\right]-\frac{\left(y_{i}-y_{A}\right)}{n_{i, z}}\left[\frac{n_{i, z} p_{i k, y}-n_{i, y} p_{i k, z}}{v_{i}}\right]
\end{gathered}
$$

The traveltime increments $t_{i}$ are now summed over all layers in the overburden. Applying Snell's law at each plane interface results in

$$
T_{d o w n}=\sum_{i=1}^{k-1} \frac{h_{i}\left(x_{A}, y_{A}\right) p_{i k, z}}{v_{i}}
$$

$$
\begin{align*}
& +\left[\frac{\left(x_{P}-x_{A}\right)\left(n_{k, z} p_{k-1, k, x}-n_{k, x} p_{k-1, k, z}\right)+\left(y_{P}-y_{A}\right)\left(n_{k, z} p_{k-1, k, y}-n_{k, y} p_{k-1, k, z}\right)}{v_{k-1} n_{k, z}}\right] \\
& \quad-\left[\frac{\left(x_{S}-x_{A}\right)\left(n_{1, z} p_{1 k, x}-n_{1, x} p_{1 k, z}\right)-\left(y_{S}-y_{A}\right)\left(n_{1, z} p_{1 k, x}-n_{1, x} p_{1 k, z}\right)}{v_{1} n_{1, z}}\right] \tag{3.20}
\end{align*}
$$

A similiar analysis yields the upgoing traveltime. Layer thicknesses are now referred to a different arbitrary position $B$ with horizontal coordinates $\left(x_{B}, y_{B}\right)$ :

$$
\begin{gather*}
T_{u p}=-\sum_{i=1}^{k-1} \frac{h_{i}\left(x_{B}, y_{B}\right) q_{i k, z}}{v_{i}} \\
-\left[\frac{\left(x_{Q}-x_{B}\right)\left(n_{k, z} q_{k-1, k, x}-n_{k, x} q_{k-1, k, z}\right)-\left(y_{Q}-y_{B}\right)\left(n_{k, z} q_{k-1, k, y}-n_{k, y} q_{k-1, k, z}\right)}{v_{k-1} n_{k, z}}\right] \\
+\left[\frac{\left(x_{R}-x_{B}\right)\left(n_{1, z} q_{1 k, x}-n_{1, x} q_{1 k, z}\right)+\left(y_{R}-y_{B}\right)\left(n_{1, z} q_{1 k, y}-n_{1, y} q_{1 k, z}\right)}{v_{1} n_{1, z}}\right] \tag{3.21}
\end{gather*}
$$

Finally, the critically refracted traveltime increment is given by a simple alteration to the prior equation (3.9):

$$
\begin{gather*}
T_{c r i t}=\frac{\left[\left(x_{Q}-x_{B}\right)-\left(x_{P}-x_{A}\right)\right]\left(n_{k, z} p_{k k, x}-n_{k, x} p_{k k, z}\right)}{v_{k} n_{k, z}} \\
+\frac{\left[\left(y_{Q}-y_{B}\right)-\left(y_{P}-y_{A}\right)\right]\left(n_{k, z} p_{k k, y}-n_{k, y} p_{k k, z}\right)}{v_{k} n_{k, z}} \\
-\left[\frac{\left(x_{A}-x_{B}\right)\left(n_{k, z} p_{k k, x}-n_{k, x} p_{k k, z}\right)+\left(y_{A}-y_{B}\right)\left(n_{k, z} p_{k k, y}-n_{k, y} p_{k k, z}\right)}{v_{k} n_{k, z}}\right] \tag{3.22}
\end{gather*}
$$

The total head wave traveltime is now obtained in the usual manner by summing the contributions along all three raypath portions. Adding equations (3.20), (3.21), and (3.22) gives

$$
\begin{gathered}
T_{k}\left(x_{S}, y_{S}, x_{R}, y_{R}\right)=\sum_{i=1}^{k-1} \frac{h_{i}\left(x_{A}, y_{A}\right) p_{i k, z}-h_{i}\left(x_{B}, y_{B}\right) q_{i k, z}}{v_{i}} \\
+\left[\frac{\left(x_{R}-x_{B}\right)\left(n_{1, z} q_{1 k, x}-n_{1, x} q_{1 k, z}\right)+\left(y_{R}-y_{B}\right)\left(n_{1, z} q_{1 k, y}-n_{1, y} q_{1 k, z}\right)}{v_{1} n_{1, z}}\right] \\
-\left[\frac{\left(x_{S}-x_{A}\right)\left(n_{1, z} p_{1 k, x}-n_{1, x} p_{1 k, z}\right)+\left(y_{S}-y_{A}\right)\left(n_{1, z} p_{1 k, y}-n_{1, y} p_{1 k, z}\right)}{v_{1} n_{1, z}}\right] \\
+\left[\frac{\left(x_{B}-x_{A}\right)\left(n_{k, z} p_{k k, x}-n_{k, x} p_{k k, z}\right)+\left(y_{B}-y_{A}\right)\left(n_{k, z} p_{k k, y}-n_{k, y} p_{k k, z}\right)}{v_{k} n_{k, z}}\right] \\
+F\left(x_{P}-x_{A}, y_{P}-y_{A}, x_{Q}-x_{B}, y_{Q}-y_{B}\right)
\end{gathered}
$$

where the function $F$ has been previously defined. Once again, applying Snell's law at the critically refracting horizon demonstrates that $F$ vanishes identically. Furthermore, it is common (but not mandatory) practice to select the reference points $A$ and $B$ to be coincident with the source $S$ and receiver $R$, respectively. Thus $\left(x_{A}, y_{A}\right)=\left(x_{S}, y_{S}\right)$ and $\left(x_{B}, y_{B}\right)=\left(x_{R}, y_{R}\right)$. With these substitutions, the above expression simplifies dramatically to

$$
\begin{gather*}
T_{k}\left(x_{S}, y_{S}, x_{R}, y_{R}\right)=\sum_{i=1}^{k-1} \frac{h_{i}\left(x_{S}, y_{S}\right) p_{i k, z}-h_{i}\left(x_{R}, y_{R}\right) q_{i k, z}}{v_{i}} \\
+\left[\frac{\left(x_{R}-x_{S}\right)\left(n_{k, z} p_{k k, x}-n_{k, x} p_{k k, z}\right)+\left(y_{R}-y_{S}\right)\left(n_{k, z} p_{k k, y}-n_{k, y} p_{k k, z}\right)}{v_{k} n_{k, z}}\right] . \tag{3.23}
\end{gather*}
$$

The two dimensional form of equation (3.23) has some similarity to an analogous expression published by Ocola (1972), in that individual layer thicknesses are measured vertically
below the source and receiver. However, it is quite different from Palmer's (1980) 2D traveltime equation, which currently forms the theoretical basis of the generalized reciprocal method (GRM) of seismic refraction interpretation. Following Ewing et al. (1939), Palmer measures thickness perpendicular to the base of each layer, and thus arrives at a different expression. Nevertheless, the GRM can be formulated on the basis of the two dimensional version of equation (3.23); a distinct advantage is then gained in constructing the refractor depth profile (see Chapters 5 and 6).

### 3.5.4 Reflection traveltime

The previous analysis has concentrated on waves that are critically refracted at the $k^{t h}$ subsurface horizon. However, it is straightforward to demonstrate that the traveltime formulae also apply to waves that are reflected from interface $k$ of the model. Snell's law of reflection for this situation is illustrated in Figure 3.4 and is compactly expressed as

$$
\begin{equation*}
\frac{\mathbf{n}_{k} \times \mathbf{p}_{k-1, k}}{v_{k-1}^{d}}=\frac{\mathbf{n}_{k} \times \mathbf{q}_{k-1, k}}{v_{k-1}^{u}} \tag{3.24}
\end{equation*}
$$

Note that equation (3.24) allows for a possible mode conversion upon reflection, i.e. $v_{k-1}^{d}$ is not necessarily equal to $v_{k-1}^{u}$.

For a wave reflected at interface $k$, points $P$ and $Q$ are coincident; there is no intervening critically refracted raypath segment. The total traveltime is obtained by simply adding the downgoing and components. Hence, setting $\left(x_{Q}, y_{Q}\right)=\left(x_{P}, y_{P}\right)$ and summing equations (3.14) and (3.15) yields


Fig. 3.4. Reflection of a raypath at interface $k$ of the earth model. If a mode conversion occurs, then downgoing velocity $v_{k-1}^{d}$ differs from upgoing velocity $v_{k-1}^{u}$.

$$
\begin{gather*}
T_{d o w n}+T_{u p}=\sum_{i=j}^{k-1} \frac{h_{i}(0,0) p_{i k, z}}{v_{i}^{d}}-\sum_{i=l}^{k-1} \frac{h_{i}(0,0) q_{i k, z}}{v_{i}^{u}} \\
+\left[\frac{x_{R}\left(n_{l, z} q_{l k, x}-n_{l, x} q_{l k, z}\right)+y_{R}\left(n_{l, z} q_{l k, y}-n_{l, y} q_{l k, z}\right)}{v_{l}^{u} n_{l, z}}\right] \\
-\left[\frac{x_{S}\left(n_{j, z} p_{j k, x}-n_{j, x} p_{j k, z}\right)+y_{S}\left(n_{j, z} p_{j k, y}-n_{j, y} p_{j k, z}\right)}{v_{j}^{d} n_{j, z}}\right] \\
+G\left(x_{P}, y_{P}\right) \tag{3.25}
\end{gather*}
$$

where the quantity $G\left(x_{P}, y_{P}\right)$ depends on the horizontal coordinates of the reflection point and is given by

$$
\begin{gathered}
G\left(x_{P}, y_{P}\right)= \\
\frac{x_{P}}{n_{k, z}}\left[\frac{1}{v_{k-1}^{d}}\left(n_{k, z} p_{k-1, k, x}-n_{k, x} p_{k-1, k, z}\right)-\frac{1}{v_{k-1}^{u}}\left(n_{k, z} q_{k-1, k, x}-n_{k, x} q_{k-1, k, z}\right)\right] \\
+\frac{y_{P}}{n_{k, z}}\left[\frac{1}{v_{k-1}^{d}}\left(n_{k, z} p_{k-1, k, y}-n_{k, y} p_{k-1, k, z}\right)-\frac{1}{v_{k-1}^{u}}\left(n_{k, z} q_{k-1, k, y}-n_{k, y} q_{k-1, k, z}\right)\right] .
\end{gathered}
$$

Distinct downgoing and upgoing layer velocities are explicitly incorporated into the above expressions in order to emphasize the possibility of asymmetric mode-converted raypaths. The component form of Snell's law of reflection implies that $G\left(x_{P}, y_{P}\right)$ vanishes. Equation (3.25) is then identical in form to the previous traveltime expression (3.16) that was derived for head waves!

### 3.6 Rapid traveltime computation

In order to compute traveltimes via the above formulae, the unit propagation vectors $\mathbf{p}_{i k}$ and $\mathbf{q}_{i k}$ overlying the refracting/reflecting interface must be determined. For the reflection problem, this set of vectors depends on both the offset distance $X$ and the azimuth angle $\Psi$ of the receiver relative to the source. However, the critical refraction problem is qualitatively different; the propagation vectors depend only on the azimuth $\Psi$. This particular feature can be exploited to yield a rapid computational procedure for head wave traveltimes.

Since the propagation vectors depend on the recording azimuth, they should be written as $\mathbf{p}_{i k}(\Psi)$ and $\mathbf{q}_{i k}(\Psi)$, although the explicit dependence on $\Psi$ is often suppressed for notational convenience. The functional form of this dependence is not known. However, with a minimal amount of raytracing, it is possible to numerically generate the function $\Psi\left(\mathbf{p}_{i k}, \mathbf{q}_{i k}\right)$ over the full range of possible recording azimuths ( $2 \pi$ radians). Inversion of this function then yields the propagation vectors for a prescribed value of the source-receiver azimuth angle. This technique is discussed in both general and mathematical terms in the following two sections.

### 3.6.1 General description

The computational algorithm is based on the close relationship between a critically reflected and critically refracted raypath. For a given source-receiver azimuth angle, both raypath types possess the same unit propagation vectors $\mathbf{p}_{i k}(\Psi)$ and $\mathbf{q}_{i k}(\Psi)$ for $i<k$. Consider the following six-step calculation procedure, with reference to the critically reflected/refracted raypath segments depicted in Figure 3.5:
(i) Select a point $P$ on the critically refracting interface. Since the location of $P$ is arbitrary, it is convenient to position it beneath the coordinate origin $O$.
(ii) Choose a critically refracted propagation direction $\mathbf{p}_{k k}$ through point $P$. The orientation of this vector within the plane of interface $k$ is defined by an angle $\chi$ measured from an arbitrary reference line.
(iii) Forward raytrace from $P$ along the set of upward unit propagation vectors $\mathrm{q}_{i k}(i=$ $k-1, k-2, \ldots, 2,1)$ to establish the position of a receiving point $R$ on the surface. The departing vector $\mathbf{q}_{k-1, k}$ is oriented at the critical angle $i_{k}=\sin ^{-1}\left(v_{k-1} / v_{k}\right)$ relative to the interface normal and is contained in the plane defined by $\mathbf{p}_{k k}$ and $\mathrm{n}_{k}$. The appropriate three dimensional form of Snell's law is applied at all interfaces intervening between the refractor and the surface.
(iv) Reverse raytrace from $P$ along the set of downward unit propagation vectors $\mathbf{p}_{i k}(i=$ $k-1, k-2, \ldots, 2,1)$ to establish the position of a source point $S$ on the surface. The incidence angle of the arriving vector $\mathbf{p}_{k-1, k}$ (in the plane defined by $\mathbf{p}_{k k}$ and $\mathbf{n}_{k}$ ) also equals the critical angle. A three dimensional 'backward propagating' form of Snell's law is applied at all refracting interfaces.
(v) Calculate the azimuth angle $\Psi$ of $R$ with respect to $S$.
(vi) Increment angle $\chi$ by a small amount and repeat steps (ii) through (v). Stop after $\chi$ has been incremented by a total of $\pi$ radians.


Fig. 3.5. Schematic representation of a raypath critically reflected/refracted at point $P$ on interface $k . S$ and $R$ denote a surface source and receiver, respectively.

This procedure numerically defines a function $\Psi=f(\chi)$ over an interval $\left[\chi_{0}, \chi_{0}+\pi\right]$. It is not necessary to perform raytracing to determine the azimuth angle for $\chi \in\left[\chi_{0}+\pi, \chi_{0}+2 \pi\right]$. Rather, values are easily generated from the symmetry relation $\Psi=f(\chi-\pi)+\pi$. This symmetry condition arises from raypath reciprocity: reversing the direction of the critically refracted propagation vector $\mathbf{p}_{k k}$ merely interchanges the positions of the source and receiver on the surface.

If the propagation vector $\mathbf{p}_{k k}$ makes one complete rotation on the critically refracting interface (i.e., $\chi$ increments by $2 \pi$ radians), points $R$ and $S$ make one complete closed circuit on the surface (i.e., $\Psi$ also increments by $2 \pi$ ). This functional dependence is designated $\Psi=g(\chi)$. The inverse function $\chi=g^{-1}(\Psi)$ can then be used to determine the appropriate value of $\chi$ for a specified source-receiver azimuth angle. Finally, unit propagation vectors $\mathbf{p}_{i k}$ and $\mathbf{q}_{i k}$ corresponding to this value of $\chi$ are regenerated via points (iii) and (iv) above.

Note that the function $\Psi=g(\chi)$ needs to be calculated only once. All recording azimuths contained in the data acquisition geometry are treated by this same function. The numerical method for inverting $g(\chi)$ is briefly stated in the next section.

### 3.6.2 Raytracing technique

The pertinent mathematical details of the aforementioned computation procedure are now described. A distinction must be made between the downgoing and upgoing portions of the critically reflected raypath. Thus, superscripts ' $d$ ' and ' $u$ ' refer to these raypath segments, respectively.

The Cartesian components of the critically refracted propagation vector $p_{k k}$ are chosen to be

$$
\begin{align*}
& p_{k k, x}=\cos \chi \cos \phi_{k} \cos \theta_{k}-\sin \chi \sin \theta_{k}  \tag{3.26a}\\
& p_{k k, y}=\cos \chi \cos \phi_{k} \sin \theta_{k}+\sin \chi \cos \theta_{k}  \tag{3.26b}\\
& p_{k k, z}=-\cos \chi \sin \phi_{k} \tag{3.26c}
\end{align*}
$$

It is straightforward to verify that $\left\|\mathbf{p}_{k k}\right\|=1$ and $\mathbf{p}_{k k} \cdot \mathbf{n}_{k}=\mathbf{0}$. Hence, $\mathbf{p}_{k k}$ is a unit vector perpendicular to the interface normal $n_{k}$. Its orientation within the plane of interface $k$ is determined by the angle $\chi$. If $\chi=0$, then $\mathbf{p}_{k k}$ points in the maximum updip direction along the critically refracting horizon. Increasing the angle $\chi$ rotates $\mathbf{p}_{k k}$ in a clockwise sense when viewed from the coordinate origin $O$.

Consider the downgoing portion of the raypath first (segment $S P$ ). Shah (1973) uses the following three dimensional form of Snell's law for the ray refracted at interface $i$ :

$$
\mathbf{p}_{i k}=\frac{v_{i}}{v_{i-1}} \mathbf{p}_{i-1, k}+\left(\cos \nu^{d}-\frac{v_{i}}{v_{i-1}} \cos \mu^{d}\right) \mathbf{n}_{i}
$$

Hence, the transmitted propagation vector is a linear combination of the incident propagation vector and the interface normal vector. Solving for $\mathbf{p}_{i-1, k}$ yields

$$
\begin{equation*}
\mathbf{p}_{i-1, k}=\frac{v_{i-1}}{v_{i}} \mathbf{p}_{i k}+\left(\cos \mu^{d}-\frac{v_{i-1}}{v_{i}} \cos \nu^{d}\right) \mathbf{n}_{i} . \tag{3.27}
\end{equation*}
$$

This form gives the incident vector in terms of the transmitted vector and the interface normal. The cosines of the incidence angles can be obtained by the following four-point prescription: (i) $\cos \nu^{d}=\mathbf{p}_{i k} \cdot \mathbf{n}_{i}$, (ii) $\sin \nu^{d}=\sqrt{1-\cos ^{2} \nu^{d}}$, (iii) $\sin \mu^{d}=\left(v_{i-1} / v_{i}\right) \sin \nu^{d}$, (iv) $\cos \mu^{d}=\sqrt{1-\sin ^{2} \mu^{d}}$.

Similarly, the upgoing ray (segment $P R$ ) at interface $i$ satisfies Snell's law in the form

$$
\begin{equation*}
\mathbf{q}_{i-1, k}=\frac{v_{i-1}}{v_{i}} \mathbf{q}_{i k}+\left(\cos \nu^{u}-\frac{v_{i-1}}{v_{i}} \cos \mu^{u}\right) \mathbf{n}_{i} \tag{3.28}
\end{equation*}
$$

where the cosines of the incidence angles are evaluated via: (i) $\cos \mu^{u}=\mathbf{q}_{i k} \cdot \mathbf{n}_{i}$, (ii) $\sin \mu^{u}=$ $\sqrt{1-\cos ^{2} \mu^{u}}$, (iii) $\sin \nu^{u}=\left(v_{i-1} / v_{i}\right) \sin \mu^{u}$, (iv) $\cos \nu^{u}=-\sqrt{1-\sin ^{2} \nu^{u}}$. Note that a negative sign is used to calculate $\cos \nu^{u}$ since the angle $\nu^{u}$ exceeds $\pi / 2$ radians. Starting with $\mathbf{p}_{k k}\left(=\mathbf{q}_{k k}\right)$ given by expression (3.26), equations (3.27) and (3.28) are evaluated recursively for $i=k, k-1, k-2, \ldots, 3,2$ to generate the unit propagation direction vectors in all layers overlying the critically refracting interface.

Intersection points of the critically reflected raypath with the plane interfaces of the model are easily determined after the propagation directions are known. For the downward path segment within the $i^{\text {th }}$ layer, these are related via

$$
\begin{equation*}
\mathbf{r}_{i}^{d}=\mathbf{r}_{i+1}^{d}-l_{i}^{d} \mathbf{p}_{i k} \tag{3.29}
\end{equation*}
$$

Hence, $\mathbf{r}_{i}^{d}$ can be calculated if $\mathbf{r}_{i+1}^{d}$ is already known and the path length $l_{i}^{d}$ can be found. Taking the dot product with $\mathbf{n}_{i}$ and using $\mathbf{r}_{i}^{d} \cdot \mathbf{n}_{i}=d_{i}$ (see equation (3.1)) yields

$$
l_{i}^{d}=\frac{\left(\mathbf{r}_{i+1}^{d} \cdot \mathbf{n}_{i}\right)-d_{i}}{\left(\mathbf{p}_{i k} \cdot \mathbf{n}_{i}\right)}
$$

Similarly, on the upward propagating raypath segment within the same layer:

$$
\begin{equation*}
\mathbf{r}_{i}^{u}=\mathbf{r}_{i+1}^{u}+l_{i}^{u} \mathbf{q}_{i k} \tag{3.30}
\end{equation*}
$$

with

$$
l_{i}^{u}=\frac{d_{i}-\left(\mathbf{r}_{i+1}^{u} \cdot \mathbf{n}_{i}\right)}{\left(\mathbf{q}_{i k} \cdot \mathbf{n}_{i}\right)}
$$

Thus, starting with $\mathbf{r}_{k}^{d}=\mathbf{r}_{k}^{u}=0 \mathbf{i}+0 \mathbf{j}+z_{k}(0,0) \mathbf{k}$, equations (3.29) and (3.30) can be evaluated recursively with $i=k-1, k-2, \ldots, 2,1$ to yield the ray intersection points $\mathbf{r}_{1}^{d}$ and $\mathbf{r}_{1}^{u}$ on the surface. Finally, the azimuth angle $\Psi$ of $\mathbf{r}_{1}^{u}$ (the receiver point) with respect to $r_{1}^{d}$ (the source point) is calculated by

$$
\begin{equation*}
\Psi=\tan ^{-1}\left[\frac{y_{1}^{u}-y_{1}^{d}}{x_{1}^{u}-x_{1}^{d}}\right] \tag{3.31}
\end{equation*}
$$

$\Psi$ is obviously a function of the reference angle $\chi: \Psi=g(\chi)$. In principle, the inverse function $\chi=g^{-1}(\Psi)$ can be determined. In practice, numerical calculation of the complete inverse function $g^{-1}$ is unecessary. Rather, inverse interpolation between neighboring
computed values of $\Psi$ is used to locate the particular value $\hat{\chi}$ corresponding to a specified source-receiver azimuth angle $\hat{\Psi}$. Linear interpolation is adequate, provided that the sampling in $\chi$ is sufficiently fine. The value $\hat{\chi}$ is then used to regenerate the complete set of propagation vectors $\mathbf{p}_{i k}$ and $\mathbf{q}_{i k}$ needed to evaluate the head wave traveltime formula.

A comparison with an example presented by Chander (1977b) validates the accuracy of the calculation. He considers a four layer model defined by the parameters

$$
\begin{array}{llll}
\phi_{1}=0^{\circ}, & \theta_{1}=0^{\circ}, & z_{1}(0,0)=0 \mathrm{~m}, & v_{1}=4000 \mathrm{~m} / \mathrm{s} \\
\phi_{2}=0.81^{\circ}, & \theta_{2}=225^{\circ}, & z_{2}(0,0)=50 \mathrm{~m}, & v_{2}=4500 \mathrm{~m} / \mathrm{s} \\
\phi_{3}=4.04^{\circ}, & \theta_{3}=225^{\circ}, & z_{3}(0,0)=150 \mathrm{~m}, & v_{3}=5000 \mathrm{~m} / \mathrm{s} \\
\phi_{4}=11.98^{\circ}, & \theta_{4}=225^{\circ}, & z_{4}(0,0)=350 \mathrm{~m}, & v_{4}=5600 \mathrm{~m} / \mathrm{s}
\end{array}
$$

Using the above procedure, unit propagation vectors corresponding to the raypath critically refracted on interface 4 are determined for a recording azimuth $\Psi=33.69^{\circ}$ (the sampling interval used for $\chi$ is $0.25^{\circ}$ ). In particular, the downgoing propagation vector at the source is

$$
\mathbf{p}_{14}\left(33.69^{\circ}\right)=(0.523126) \mathbf{i}+(0.322360) \mathbf{j}+(0.788938) \mathbf{k}
$$

Chander (1977b) iteratively solves two coupled nonlinear equations for the raypath takeoff angles at the source, given the above azimuth angle to the receiver. The orientation of the takeoff vector $p_{14}$ can be described by two angles in a manner analogous to the interface normals (see Figure 3.1):

$$
\mathbf{p}_{14}=\left(\sin \alpha_{14} \cos \gamma_{14}\right) \mathbf{i}+\left(\sin \alpha_{14} \sin \gamma_{14}\right) \mathbf{j}+\left(\cos \alpha_{14}\right) \mathbf{k}
$$

$\alpha_{14}\left(0^{\circ} \leq \alpha_{14}<180^{\circ}\right)$ is a polar angle measured from the $+z$ axis and $\gamma_{14}\left(0^{\circ} \leq \gamma_{14}<360^{\circ}\right)$ is an azimuthal angle measured from the $+x$ axis. Then, the above results yield $\alpha_{14}=37.91^{\circ}$ and $\gamma_{14}=31.64^{\circ}$, in agreement with Chander's (1977b) solution for the same two angles.

### 3.7 Forward modeling examples

Two examples presented in this section illustrate the utility, as well as some of the limitations, of the head wave traveltime formulae for forward modeling applications. The recording geometry in the first example is the common reversed profile; all sources and receivers are located on the surface. The second example considers a typical offset VSP geometry (surface source and downhole receivers).

### 3.7.1 Profile geometry

Equation (3.12) expresses the head wave traveltime in terms of the horizontal offset distance between a surface source and a surface receiver. It is written in condensed form as

$$
\begin{equation*}
T_{k}\left(x_{S}, y_{S}, X, \Psi\right)=m_{k}(\Psi) X+b_{k}\left(x_{S}, y_{S}, \Psi\right) \tag{3.32}
\end{equation*}
$$

where the definitions of the slope $m_{k}(\Psi)$ and intercept $b_{k}\left(x_{S}, y_{S}, \Psi\right)$ are obvious. This expression is a three dimensional extension of the slope/intercept formulae that are commonly used to describe head wave traveltimes. Equation (3.32) is evaluated for a shallow three-layer model defined by the parameters

$$
\begin{array}{llll}
\phi_{1}=0^{\circ}, & \theta_{1}=0^{\circ}, & z_{1}(0,0)=0 \mathrm{~m}, & v_{1}=1000 \mathrm{~m} / \mathrm{s} \\
\phi_{2}=8^{\circ}, & \theta_{2}=0^{\circ}, & z_{2}(0,0)=5 \mathrm{~m}, & v_{2}=1800 \mathrm{~m} / \mathrm{s} \\
\phi_{3}=5^{\circ}, & \theta_{3}=60^{\circ}, & z_{3}(0,0)=12 \mathrm{~m}, & v_{3}=3200 \mathrm{~m} / \mathrm{s}
\end{array}
$$

Figure 3.6 displays the head wave arrival times observed by a set of four reversed refraction profiles. Direct wave traveltimes are also included in each plot. Shots are located at each end of the recording spreads and the maximum source-receiver offset distance is 50 m . The profile pairs are all centered on the coordinate origin and are oriented in the N-S, NE-SW, E-W, and SE-NW directions (panels a through d, respectively).


Fig. 3.6. Direct and head wave arrival times recorded by a set of four reversed profiles over a shallow three-layer earth model. (a) N-S profile. (b) NE-SW profile. (c) E-W profile. (d) SE-NW profile. All profiles are centered at the coordinate origin.

The straight line arrival time curves plotted in each panel convey an impression of a two dimensional earth model. The full three dimensionality of the subsurface is only appreciated by comparing traveltime curves recorded along several separate azimuths. Note that the reciprocal times (the shot-to-shot traveltimes) for forward and reverse arrivals in all panels agree, as expected. Refraction traveltime curves are extended to zero offset distance, even though head waves do not exist in the precritical offset zone. Since the traveltime computation method does not locate the critical offset distance, equation (3.32) is evaluated over the full offset range covered by the receiver array. If traveltime analysis is concerned solely with first arrivals, then these calculated nonphysical traveltimes do not pose any problems, because they are always associated with later arrivals. However, precritical offset arrivals do have interpretive significance (Ackermann et al., 1986) and thus their inclusion in the current algorithm may be useful for some studies.

Note that the intercept time in equation (3.32) depends on the recording azimuth angle $\Psi$ (through the propagation vectors $\mathbf{p}_{i k}$ and $\mathbf{q}_{i k}$ ) in addition to the source coordinates $\left(x_{S}, y_{S}\right)$. This unusual feature appears to be peculiar to three dimensional, multilayered earth models. Intercept time is obviously independent of profile azimuth for all one dimensional models. For two dimensional earth models (recorded normal to strike), the identity of intercept times observed on split spread profiles is an interpretive rule (Johnson, 1976; Merrick et al., 1978; Ackermann et al., 1986; Brückl, 1987). Finally, in the case of the simplest three dimensional model consisting of a single layer overlying a halfspace, the intercept time is also independent of recording profile azimuth (see equation (2.10)). Dependence on the azimuth angle $\Psi$ only arises when multiple layers in three dimensions are analyzed. However, raypath reciprocity requires that the propagation direction vectors satisfy

$$
\begin{equation*}
\mathbf{p}_{i k}(\Psi+\pi)=-\mathbf{q}_{i k}(\Psi), \quad \mathbf{q}_{i k}(\Psi+\pi)=-\mathbf{p}_{i k}(\Psi) \tag{3.33}
\end{equation*}
$$

Substituting these results into the expression for the intercept time yields

$$
\begin{equation*}
b_{k}\left(x_{S}, y_{S}, \Psi+\pi\right)=b_{k}\left(x_{S}, y_{S}, \Psi\right) \tag{3.34}
\end{equation*}
$$

Hence, the two dimensional interpretive rule also holds for the particular three dimensional model examined here (uniform velocity layers bounded by plane interfaces). However, it is not true that that intercept times recorded on all line profiles emanating from the same shotpoint are identical. For realistic earth models (maximum dip $\sim 10^{\circ}$ ) this variation in intercept time with profile azimuth appears to be minute. In the current example, $b_{3}(0,0, \Psi)$ varies by only $0.03 \%$ as the azimuth $\Psi$ increases from $0^{\circ}$ to $360^{\circ}$.

### 3.7.2 VSP geometry

The final example examines head wave traveltimes recorded in a simulated VSP experiment. The computation procedure described in section 3.6 is readily generalized to a situation where the source and receiver are located on different interfaces of the model. An azimuth angle function $\Psi=g(\chi)$ can be calculated for a hypothetical source located on interface $j$ and a hypothetical receiver located on interface $l$. In an offset VSP survey, the source is located on the surface $(j=1)$ and a borehole geophone is lowered continuously down the well. Thus, an azimuth angle function is computed for all interfaces $l$, including the critically refracting horizon $(l=k)$. These functions are then inverted at the known source-receiver azimuth angle, and the resulting propagation vectors are used in equation (3.26) to obtain head wave arrival times at each interface intersection with the well. The traveltime to a geophone located within a layer is then calculated by linearly interpolating times computed at the bounding interfaces.

This example considers a four layer earth model defined by the parameters:

$$
\begin{array}{llll}
\phi_{1}=0^{\circ}, & \theta_{1}=0^{\circ}, & z_{1}(0,0)=0 \mathrm{~m}, & v_{1}=1800 \mathrm{~m} / \mathrm{s}, \\
\phi_{2}=3^{\circ}, & \theta_{2}=90^{\circ}, & z_{2}(0,0)=50 \mathrm{~m}, & v_{2}=2500 \mathrm{~m} / \mathrm{s}, \\
\phi_{3}=4^{\circ}, & \theta_{3}=270^{\circ}, & z_{3}(0,0)=120 \mathrm{~m}, & v_{3}=3200 \mathrm{~m} / \mathrm{s}, \\
\phi_{4}=0^{\circ}, & \theta_{4}=90^{\circ}, & z_{4}(0,0)=155 \mathrm{~m}, & v_{4}=3900 \mathrm{~m} / \mathrm{s} .
\end{array}
$$



Fig. 3.7. Head wave arrival times recorded in a VSP configuration. Surface source is offset from the well by 500 m to the north, east, and west in panels (a), (b), and (c), respectively.

The model is strictly two dimensional; the strike directions of all interfaces are north-south. However, since sources are deployed at various azimuths around the well, the three dimensional formulae are needed to compute accurate arrival times. Figure 3.7 displays head wave traveltime curves as a function of geophone depth within a borehole positioned at the coordinate origin. Surface sources are offset from the well by 500 m to the north, east, and west (panels $a, b$, and $c$, respectively). For this 2D model, a source offset 500 m to the south generates traveltimes that are identical to those plotted in panel a. The arrival time of a given head wave decreases as the geophone is lowered in the well, because the receiver approaches the critically refracting interface more closely. As the geophone passes through an interface, the slope of the traveltime curve changes. Expressions for these slopes can be obtained from equations (3.17) or (3.18) above. Each curve terminates at the depth of the critical refractor in the well since the associated head wave is not observable below this level. Note that the critically refracted waves depicted in Figure 3.7 are not necessarily the initial arrivals. Other waves that are neglected in the modeling procedure (e.g., direct waves) may actually arrive first over certain ranges of receiver depth.

### 3.8 Conclusion

The traveltime formulae derived in this chapter are useful for a variety of forward modeling applications involving head waves propagating in three dimensional, multilayered earth models. Obviously, construction of traveltime curves for a trial earth model can assist in the interpretation of field recorded data. The equations also provide means for evaluating the importance of three dimensional effects on head wave traveltimes in various seismological contexts. For example, Merrick et al. (1978) investigate the hidden layer phenomenon using a 2D earth model. Hunter and Pullan (1990), using a 1D model, compare the sensitivities of vertical and horizontal receiver arrays for discriminating layer velocities. Many investigators are concerned about the possibility that head waves might constitute first arrivals in crosswell transmission tomography experiments. All of these studies can benefit from a full three dimensional analysis.

Several straightforward extensions of the results described in this chapter can enhance the utility of the formulae for forward modeling purposes. Inclusion of multiple reflections within the overburden of the critically refracting interface does not pose any special problems (note that multiple raypath segments along the critical horizon are not allowed; see Červený and Ravindra, 1971, p. 210). Also, well known tools of asymptotic ray theory can be applied to calculate the particle displacement amplitude and waveform of a head wave in this three dimensional situation. Care must be exercised in treating shear wave propagation, because SV and SH modes are not globally decoupled in 3D. Richards et al. (1991) examine some of these phenomena and propose a particular computing procedure. Interestingly, they claim (but do not prove!) that Diebold's (1987) traveltime formula is, in general, correct. Thus, their results should be treated with caution.

Finally, since head wave traveltime can be expressed by a simple mathematical formula, inverse methods designed to recover the earth model parameters from measured data are facilitated. Chapter 4 describes an inversion procedure that exploits the rapid forward modeling capability developed above. Other head wave traveltime inversion procedures are probably possible. In particular, the 'slope and intercept' equation (3.32) may allow the extension of the two dimensional methods of Dooley (1952), Adachi (1954), and Johnson (1976) to three dimensional models. Although considerable effort has been devoted to discovering this generalization, success has not yet been achieved.

## CHAPTER 4

## INVERSION OF HEAD WAVE TRAVELTIMES FOR THREE DIMENSIONAL PLANAR STRUCTURE

### 4.1 Introduction

Numerous investigators have studied the inversion of reflection traveltimes for three dimensional subsurface structure (Hubral, 1976; Gjøystal and Ursin, 1981; Chiu et al., 1986; Chiu and Stewart, 1987; Lin, 1989; Phadke and Kanasewich, 1990). However, the analogous situation for refraction traveltime data has not been thoroughly examined. Kanasewich and Chiu (1985) present a method for jointly inverting reflection and refraction traveltimes to recover 3D structure. Forward calculation of traveltimes is achieved with the iterative ray bending method of Chander (1977a).

This chapter describes an algorithm for inverting head wave arrival times for three dimensional planar structure. The earth model is characterized by a stack of uniform velocity layers bounded by plane, dipping interfaces. The inversion method is iterative; an initial estimate of the model parameters is refined until an acceptable match is obtained between observed and predicted traveltimes. Rapid forward modeling of both traveltimes and traveltime derivatives is achieved with the numerical technique developed in Chapter 3. A novel feature of the inversion procedure is the inclusion of constraint information in the form of inequality relations satisfied by the model parameters. Often, a priorigeological or geophysical information is available to guide and constrain a traveltime inversion. This is particularly useful for the inversion of head wave arrival times because the problem can be very ill-posed and admit numerous solutions.

After a discussion of the mathematical basis of the inversion technique, the algorithm is tested on both synthetic and field acquired traveltime data. Inversion of refraction data acquired in the Peace River Arch region of northern Alberta indicates that the algorithm can be a valuable tool for the analysis of traveltimes recorded in a broadside configuration.

### 4.2 Inversion mathematics

### 4.2.1 General theory

Let the observed arrival times for the experiment be organized into an $M$-dimensional column vector $\mathrm{t}_{\text {obs }}$. The earth model is characterized by a finite set of scalar parameters $m_{j}, j=1,2, \ldots, N$. These are organized into an $N$-dimensional column vector m . Predicted head wave traveltimes generated by this model are designated by the $M$-dimensional vector $\mathbf{t}_{\text {prd }}(\mathbf{m})$. Then, a first order Taylor series expansion of the observed data about the particular model $\mathbf{m}^{n}$ yields the expression

$$
\begin{equation*}
\mathbf{A}\left(\mathbf{m}^{n}\right) \Delta \mathbf{m}^{n+1}=\Delta \mathbf{t}\left(\mathbf{m}^{n}\right) \tag{4.1}
\end{equation*}
$$

where $\Delta \mathrm{t}\left(\mathrm{m}^{n}\right) \equiv \mathrm{t}_{\text {obs }}-\mathrm{t}_{\text {prd }}\left(\mathrm{m}^{n}\right)$ is the data discrepancy vector, $\Delta \mathrm{m}^{n+1} \equiv \mathrm{~m}^{n+1}-\mathrm{m}^{n}$ is the parameter update vector, and the elements of the $M \times N$ sensitivity matrix $\mathbf{A}\left(\mathbf{m}^{n}\right)$ are given by

$$
\begin{equation*}
\left[\mathbf{A}\left(\mathbf{m}^{n}\right)\right]_{i j}=\left.\frac{\partial t_{i}^{p r d}}{\partial m_{j}}\right|_{\mathbf{m}=\mathbf{m}^{n}} \tag{4.2}
\end{equation*}
$$

In these and subsequent expressions, the superscript $n$ denotes the iteration index.
In many crustal seismic reflection and refraction experiments, the system (4.1) is overdetermined, underconstrained, and inconsistent. A popular solution technique for $\Delta \mathbf{m}^{n+1}$ is the damped least squares method (Braile, 1973; Kanasewich and Chiu, 1985; Chiu et al., 1986; Chiu and Stewart, 1987; Phadke and Kanasewich, 1990). The updated parameter vector is then obtained via $\mathbf{m}^{n+1}=\mathbf{m}^{n}+\Delta \mathbf{m}^{n+1}$. Iterations continue until an acceptable fit to the observed traveltime data is achieved. This strategy is termed 'creeping' (Scales et al., 1990) because the final solution is obtained by the addition of (possibly many) small perturbations to an initial guess.

An alternative approach is used in this study to obtain the improved model parameter vector. Substituting the definition of the parameter update vector into (4.1) and rearranging
terms gives

$$
\begin{equation*}
\mathbf{A}\left(\mathbf{m}^{n}\right) \mathbf{m}^{n+1}=\Delta \mathbf{t}\left(\mathbf{m}^{n}\right)+\mathbf{A}\left(\mathbf{m}^{n}\right) \mathbf{m}^{n} . \tag{4.3}
\end{equation*}
$$

The right hand side of this expression consists of known quantities. Hence, equation (4.3) can be solved directly for the new model parameter vector $\mathbf{m}^{n+1}$. Scales et al. (1990) refer to this technique as 'jumping' because, in the absence of additional constraints, the size of the model change between iterations $\left\|\mathbf{m}^{n+1}-\mathbf{m}^{n}\right\|$ is not restricted to be small. Since the inversion is formulated in terms of the model itself, rather than a model perturbation, the jumping strategy facilitates the incorporation of constraint information into the algorithm. Constraints are mathematically expressed in the form of inequality relations satisfied by the model parameter vector:

$$
\begin{equation*}
\mathbf{m}_{-} \leq \mathbf{m}^{n+1} \leq \mathbf{m}_{+} \tag{4.4a}
\end{equation*}
$$

where the vectors $\mathbf{m}_{-}$and $\mathbf{m}_{+}$are lower and upper bounds on the model, respectively. These bounds arise from a priori geological or geophysical knowledge (or assumptions) about the earth model. For example, the model parameters in this study are strictly nonegative. Thus, if the lower bound is set equal to 0 , negative values are excluded on each iteration of the inversion procedure. This is required for meaningful forward modeling of traveltimes. The inequality bounds in (4.4a) can also be used to severely restrict (or even eliminate) the variation of a certain parameter on succesive iterations of the inversion. In this situation, upper and lower bounds are narrowly established about an accurately known (or preferred) value for the particular parameter.

The inversion algorithm is also stabilized by limiting the size of the model change between iterations. Hence, if $\delta \mathrm{m}$ is a vector of upper bounds on the parameter increments, the updated parameter vector must satisfy the additional inequality constraints

$$
\begin{equation*}
\mathbf{m}^{n}-\delta \mathbf{m} \leq \mathbf{m}^{n+1} \leq \mathbf{m}^{n}+\delta \mathbf{m} \tag{4.4b}
\end{equation*}
$$

These constraints fulfill the same regularizing role as the damping parameter in a damped least squares solution of the original equation (4.1). If a reasonably good initial estimate
$\mathrm{m}^{0}$ for the model is available, and if $\|\delta \mathrm{m}\|$ is sufficiently small, then the constraints (4.4b) assure that the algorithm iterates toward a solution in the neighborhood of $\mathrm{m}^{0}$, rather than jumping to a remote region of model space.

The observed traveltimes on the right hand side of the linearized data equation (4.3) are contaminated with random picking errors. Hence, an exact solution may not exist. A robust solution to this inconsistent system can be obtained by minimizing the $l_{1}$ norm of the misfit. Linear programming provides a convenient solution method because the model parameters are intrinsically non-negative and are constrained by inequalities (4.4a,b). In order to pose the problem in the context of linear programming, an $M$-dimensional residual vector $r$ is introduced into equation (4.3) as follows:

$$
\begin{equation*}
\mathbf{A}\left(\mathbf{m}^{n}\right) \mathbf{m}^{n+1}+\mathbf{r}=\Delta \mathbf{t}\left(\mathbf{m}^{n}\right)+\mathbf{A}\left(\mathbf{m}^{n}\right) \mathbf{m}^{n} \tag{4.4c}
\end{equation*}
$$

The elements of $\mathbf{r}$ constitute additional unknown variables that must be solved for. The problem now consists of determining the model parameter vector $\mathrm{m}^{n+1}$ and the residual vector $r$ that simultaneously satisfy the inequality constraints (4.4a,b), the equality constraints ( 4.4 c ), and that minimize the $l_{1}$ norm of the residual

$$
\begin{equation*}
\|\mathbf{r}\|=\sum_{i=1}^{M}\left|r_{i}\right| \tag{4.4d}
\end{equation*}
$$

A standard linear programming routine is used here to solve the constrained optimization problem specified by equations (4.4a,b,c,d). After an improved model parameter vector $\mathbf{m}^{n+1}$ is obtained, the $l_{1}$ norm of the misfit between observed and predicted traveltimes $\left\|\mathrm{t}_{\text {obs }}-\mathrm{t}_{\text {prd }}\left(\mathrm{m}^{n+1}\right)\right\|$ is computed. Iterations cease when the misfit reaches some acceptable level, or exhibits negligable change on successive iterations. The elements of the residual vector $r$ play no further role after equations (4.4) are solved, and are discarded. However, the $l_{1}$ norm of the residual (4.4d) is monitored on each iteration in order to assess how closely the linearized data equations have been fit.

Finally, the flexibility of the inversion algorithm is enhanced by including variable weighting of both the traveltime data and the model parameters. Equation (4.3) is modified to

$$
\left[\mathbf{W}_{d} \mathbf{A}\left(\mathbf{m}^{n}\right) \mathbf{W}_{p}^{-1}\right]\left[\mathbf{W}_{p} \mathbf{m}^{n+1}\right]=\mathbf{W}_{d} \Delta \mathbf{t}\left(\mathbf{m}^{n}\right)+\left[\mathbf{W}_{d} \mathbf{A}\left(\mathbf{m}^{n}\right) \mathbf{W}_{p}^{-1}\right]\left[\mathbf{W}_{p} \mathbf{m}^{n}\right]
$$

where $\mathbf{W}_{d}$ and $\mathbf{W}_{p}$ are data and parameter weighting matrices, respectively. Currently, these are restricted to be diagonal matrices. Thus, premultiplication of the sensitivity matrix by $\mathbf{W}_{d}$ corresponds to row weighting and postmultiplication by $\mathbf{W}_{\boldsymbol{p}}^{-1}$ corresponds to column weighting.

The data weighting matrix can be used to emphasize those particular traveltimes judged to be more significant for the inversion. The parameter weighting matrix serves to nondimensionalize and normalize the elements of the model parameter vector $\mathbf{m}^{n+1}$. This is a practical concern in inversion algorithms where the model is characterized by parameters with different physical dimensions and/or widely varying numerical magnitudes. The conditioning of the sensitivity matrix $\mathbf{A}\left(\mathbf{m}^{n}\right)$ is improved by column scaling by $\mathbf{W}_{p}^{-1}$. Thus, numerical roundoff error associated with the linear programming solution is reduced. Suitable units of measure are chosen for the various model parameters; the reciprocals of these scalars form the diagonal elements of $W_{p}$. The inequality constraints (4.4a,b) on the model must also be nondimensionalized in the same manner. If the weighted parameter vector calculated by the inversion algorithm is designated $\hat{\mathbf{m}}^{n+1}$, then the physical parameters required for forward modeling of traveltimes are obtained via $\mathbf{m}^{n+1}=\mathbf{W}_{\boldsymbol{p}}^{-1} \hat{\mathbf{m}}^{n+1}$.

### 4.2.2 Calculation of sensitivities

A simple 3D earth model consisting of a single layer overlying a halfspace is characterized by the five-element parameter vector $\mathrm{m}=\left[v_{1}, v_{2}, \phi, \theta, h\right]^{T}$, where $v_{1}$ and $v_{2}$ are P-wave velocities of the two media, $\phi$ and $\theta$ are interface orientation angles, and $h$ is the vertical depth to the refractor below the coordinate origin (Figure 2.1). A closed form expression for head wave traveltime in terms of these parameters is given in Chapter 2 (equation (2.10)). Hence, formulae for the elements of the sensitivity matrix can be derived by straightforward
partial differentiation. However, a substantial simplification arises by redefining the set of model parameters as $\mathbf{m}=\left[s_{1}, i_{c}, \phi, \theta, h\right]^{T}$, where $s_{1}=1 / v_{1}$ is the layer slowness and $i_{c}=\sin ^{-1}\left(v_{1} / v_{2}\right)$ is the critical refraction angle. The traveltime equation is then linear in the slowness $s_{1}$. In terms of these new parameters, head wave traveltime is

$$
\begin{equation*}
T=s_{1} \sin \left(i_{c}-\delta\right) X+2 s_{1} \cos i_{c}\left[h \cos \phi-\sin \phi\left(x_{S} \cos \theta+y_{S} \sin \theta\right)\right] \tag{4.5}
\end{equation*}
$$

where $\delta=\sin ^{-1}[\sin \phi \cos (\Psi-\theta)] . x_{S}$ and $y_{S}$ are the horizontal coordinates of the source, and $X$ and $\Psi$ are the offset and azimuth to the receiver, respectively (Figure 2.2a). Differentiating expression (4.5) yields the traveltime sensitivities

$$
\begin{gathered}
\frac{\partial T}{\partial s_{1}}=\sin \left(i_{c}-\delta\right) X+2 \cos i_{c}\left[h \cos \phi-\sin \phi\left(x_{S} \cos \theta+y_{S} \sin \theta\right)\right] \\
\frac{\partial T}{\partial i_{c}}=s_{1} \cos \left(i_{c}-\delta\right) X-2 s_{1} \sin i_{c}\left[h \cos \phi-\sin \phi\left(x_{S} \cos \theta+y_{S} \sin \theta\right)\right] \\
\frac{\partial T}{\partial \phi}=\frac{-s_{1} \cos \left(i_{c}-\delta\right) \cos \phi \cos (\Psi-\theta)}{\cos \delta} X-2 s_{1} \cos i_{c}\left[h \sin \phi+\cos \phi\left(x_{S} \cos \theta+y_{S} \sin \theta\right)\right] \\
\frac{\partial T}{\partial \theta}=\frac{-s_{1} \cos \left(i_{c}-\delta\right) \sin \phi \sin (\Psi-\theta)}{\cos \delta} X+2 s_{1} \cos i_{c} \sin \phi\left(x_{S} \sin \theta-y_{S} \cos \theta\right) \\
\frac{\partial T}{\partial h}=2 s_{1} \cos i_{c} \cos \phi
\end{gathered}
$$

The $i^{\text {th }}$ row of the $M \times 5$ sensitivity matrix $\mathbf{A}\left(\mathrm{m}^{n}\right)$ is obtained by evaluating these relations with the current model vector $\mathrm{m}^{n}$ and with the geometric parameters $\left\{x_{S i}, y_{S i}, X_{i}, \Psi_{i}\right\}$ appropriate for the $i^{\text {th }}$ recorded traveltime. The number of traveltimes $M$ usually exceeds 5 , and thus the matrix is overdetermined. However, in some situations, the matrix is also rank deficient. If the refracting horizon is horizontal $(\phi=0)$ then the derivative $\partial T / \partial \theta$ vanishes, and the fourth column of the sensitivity matrix is identically zero. This will occur, for
example, if a one dimensional earth model is used for the initial parameter vector estimate $\mathrm{m}^{0}$.

Calculation of head wave traveltime sensitivities for a multilayered earth model is more complicated because a closed form mathematical expression for the arrival time does not exist. However, Chapter 3 describes a rapid computing procedure for obtaining critically refracted traveltimes for arbitrary recording geometries. A combination of analytical and numerical techniques can then be used to calculate the sensitivities. In this study, sources and receivers are restricted to the surface. Moreover, the orientation angles of the surface ( $\phi_{1}$ and $\theta_{1}$ ) are assumed to be known. Thus, if there are $K$ interfaces in the earth model, then there are a total of $N=4 K-3$ model parameters ( $K$ layer slownesses, $2 K-2$ interface orientation angles, and $K-1$ layer thicknesses). These parameters are organized into the $N$-dimensional column vector

$$
\mathbf{m}=\left[s_{1}, \ldots, s_{K}, \phi_{2}, \ldots, \phi_{K}, \theta_{2}, \ldots, \theta_{K}, h_{1}, \ldots, h_{K-1}\right]^{T}
$$

where $s_{l}$ is a layer slowness, $\phi_{l}$ is an interface dip angle, $\theta_{l}$ is an interface azimuth angle, and $h_{l}$ is a vertical layer thickness measured at the origin. The numbering convention for layers and interfaces is described at the beginning of Chapter 3.

A 'slope and intercept' expression for the traveltime of a head wave critically refracted at interface $k(2 \leq k \leq K)$ is given by equation (3.12). In the current notation, it is

$$
\begin{equation*}
T=m(\Psi) X+b\left(x_{S}, y_{S}, \Psi\right) \tag{4.6a}
\end{equation*}
$$

where the slope $m(\Psi)$ and intercept time $b\left(x_{S}, y_{S}, \Psi\right)$ are

$$
\begin{equation*}
m(\Psi)=s_{1}\left[\cos \Psi q_{1 k, x}+\sin \Psi q_{1 k, y}-\tan \phi_{1} \cos \left(\Psi-\theta_{1}\right) q_{1 k, z}\right] \tag{4.6b}
\end{equation*}
$$

$$
\begin{gather*}
b\left(x_{S}, y_{S}, \Psi\right)=\sum_{i=1}^{k-1} s_{i} h_{i}\left(p_{i k, z}-q_{i k, z}\right)-s_{1}\left[x_{S}\left(p_{1 k, x}-q_{1 k, x}\right)+y_{S}\left(p_{1 k, y}-q_{1 k, y}\right)\right. \\
\left.-\tan \phi_{1}\left(x_{S} \cos \theta_{1}+y_{S} \sin \theta_{1}\right)\left(p_{1 k, z}-q_{1 k, z}\right)\right] \tag{4.6c}
\end{gather*}
$$

Note that the slope and intercept simplify if the surface is horizontal ( $\phi_{1}=0$ ). Equation (4.6) is not a 'closed form' expression for head wave traveltime because the unit propagation vectors $\mathbf{p}_{i k}$ and $\mathbf{q}_{i k}$ characterizing the raypath depend implicitly on both the model $\mathbf{m}$ and the recording azimuth $\Psi$. Nevertheless, if $m$ and $\Psi$ are specified, then all of the propagation vectors can be accurately computed.

Many of the elements of the sensitivity matrix $\mathbf{A}(\mathbf{m})$ can be determined by analysis. The traveltime of a head wave formed on interface $k$ does not depend on the parameters $s_{l}$, $\phi_{l}$, and $\theta_{l}$ for $l>k$, or on $h_{l}$ for $l \geq k$. Hence, these traveltime derivatives are identically zero. Moreover, the sensitivity to layer thickness $h_{l}$ for $l<k$ is derived directly from the above expression for intercept time: $\partial T / \partial h_{l}=s_{l}\left(p_{l k, z}-q_{l k, z}\right)$. The remaining sensitivities must be evaluated by a finite-difference technique. Let $\mathbf{d m}_{j}$ be a model perturbation vector with zeros in all element positions except the $j^{\text {th }}$. Then, the partial derivative of head wave traveltime with respect to parameter $m_{j}$ is approximated by the forward finite-difference

$$
\begin{equation*}
\frac{\partial T}{\partial m_{j}} \simeq \frac{T\left(\mathbf{m}+\mathbf{d m}_{j}\right)-T(\mathbf{m})}{\left\|\mathbf{d m}_{j}\right\|} \tag{4.7}
\end{equation*}
$$

Unit propagation vectors are generated for both the perturbed model $\mathbf{m}+\mathbf{d m} \mathbf{m}_{j}$ and the unperturbed model $m$ via the procedure described in Chapter 3. These are then used in formulae ( $4.6 \mathrm{a}, \mathrm{b}, \mathrm{c}$ ) above to calculate traveltimes for each model. Finally, substituting these traveltimes into (4.7) yields the required derivative. The size of the model perturbation $\left\|\mathbf{d m}_{\boldsymbol{j}}\right\|$ is typically about $1 \%$ of the value of the associated model parameter. Although a centered finite-difference scheme would yield greater accuracy, the one-sided approximation
used here requires less computational effort. Only two traveltime calculations are necessary for each model parameter, instead of three.

### 4.3 Synthetic examples

The two examples discussed in this section are representative of a large number of computational experiments conducted with the inversion algorithm. Both the single-layer and multiple-layer variants of the algorithm are examined. The first example demonstrates that the single-layer version is capable of returning the correct solution under a variety of operating conditions. However, as indicated in the second example, the multilayer version appears to require fairly restrictive constraints in order to iterate toward the correct model.

### 4.3.1 Single layer

The earth model used for the first example consists of a single layer overlying a halfspace and is defined by the parameter values:

$$
v_{1}=1500 \mathrm{~m} / \mathrm{s}, \quad v_{2}=2500 \mathrm{~m} / \mathrm{s}, \quad \phi=5^{\circ}, \quad \theta=45^{\circ}, \quad h=100 \mathrm{~m} .
$$

Critically refracted arrival times for two areal recording geometries are generated from equation (2.10). Inversions are performed with both accurate and error contaminated traveltimes.

Figure 4.1a is a plan view of a triangular data acquisition geometry. Thirty receivers are deployed around the perimeter of an equilateral triangle with sides 500 m long. Each side contains 11 receivers separated by 50 m . All receivers record energy from a source that is activated sequentially at the three vertices of the triangle. However, some source-receiver offsets are less than the critical offset distance (equation (2.11)) and thus the receivers do not detect a head wave arrival. These fictitious times are excluded from the inversions. A total of $M=73$ uniformly weighted traveltimes are used to recover the $N=5$ earth model parameters.


Fig. 4.1. Plan views of two areal recording arrays. Sources are indicated by asterisks and receivers by small crosses. (a) Triangular geometry. (b) Swath geometry. The strike and dip symbol in the upper right corner of each panel refers to a subsurface interface located 100 m below the coordinate origin (large cross).

Numerical results from a few typical inversion runs are described here. The iterative inversion procedure is initiated with a one dimensional model given by

$$
v_{1}=1000 \mathrm{~m} / \mathrm{s}, \quad v_{2}=2000 \mathrm{~m} / \mathrm{s}, \quad \phi=0^{\circ}, \quad \theta=0^{\circ}, \quad h=85 \mathrm{~m}
$$

After eight iterations, an excellent estimate of the true earth model is returned:

$$
v_{1}=1500.21 \mathrm{~m} / \mathrm{s}, \quad v_{2}=2499.98 \mathrm{~m} / \mathrm{s}, \quad \phi=5^{\circ}, \quad \theta=45^{\circ}, \quad h=100.02 \mathrm{~m}
$$

The initial traveltime misfit of 79.6 ms is reduced to 0.1 ms ; iterations are terminated when the relative change in the misfit is less than $1 \%$. A wide variety of starting models yields essentially the same final solution, although the number of iterations required for convergence varies. Also, the inversion is stable when the traveltimes are contaminated with small random errors. Random numbers drawn from a uniform probability distribution on $\pm 4 \mathrm{~ms}$ are added to the accurate times, and the algorithm is initiated with the same starting model. After five iterations, the following solution is obtained:

$$
v_{1}=1496.93 \mathrm{~m} / \mathrm{s}, \quad v_{2}=2549.46 \mathrm{~m} / \mathrm{s}, \quad \phi=5.22^{\circ}, \quad \theta=51.93^{\circ}, \quad h=99.65 \mathrm{~m}
$$

Iterations cease when the misfit decreases below 2.3 ms , or one standard deviation of the noise. Note that the overburden velocity $v_{1}$ has been correctly estimated from the refraction data alone! This interesting (and unusual) result is consistent with the theoretical analysis presented in Chapter 2.

These results suggest that the triangular recording array is a useful tool for determining three dimensional planar structure. This particular geometry combines an adequate distribution of offset and azimuth with three reciprocal time pairs. These are favorable attributes for a successful inversion of refraction arrival times via the time term method (Scheidegger and Willmore, 1957; Willmore and Bancroft, 1960; Berry and West, 1966; Smith et al., 1966; Reiter, 1970). The triangle was first investigated by Gardner (1939) who demonstrated that
it yields an exact solution for the delay times at the three vertices. More recently, the triangular array has been used for deep crustal seismic exploration in Saskatchewan (Kanasewich and Chiu, 1985) and British Columbia (Zelt et al., 1990).

Other recording geometries are considerably less robust in detecting and resolving the three dimensional dipping structure. Figure 4.1b depicts two parallel line arrays, separated by 200 m , oriented along the strike direction of the subsurface refractor (NW-SE). Each spread contains 11 geophones (receiver interval $=50 \mathrm{~m}$ ) that record head wave arrivals from a source located at the center of the opposite array. This broadside recording pattern simulates aspects of the 'swath geometry' commonly used for 3D seismic reflection surveys. Using the same starting model, inversion of the 22 broadside traveltimes yields

$$
v_{1}=1285.85 \mathrm{~m} / \mathrm{s}, \quad v_{2}=2500.37 \mathrm{~m} / \mathrm{s}, \quad \phi=4.96^{\circ}, \quad \theta=45.02^{\circ}, \quad h=80.00 \mathrm{~m}
$$

The initial traveltime misfit of 66.4 ms is reduced to $\sim 0 \mathrm{~ms}$ in eight iterations. Although this model generates an exact fit to the data, the overburden velocity $v_{1}$ and vertical depth $h$ are incorrect. A similar effect is observed when the inversion is initiated from numerous different starting models. This situation illustrates the classical tradeoff between overburden velocity and refractor depth in seismic refraction interpretation. If additional a priori data are introduced into the inverse problem, then a correct solution is possible. For example, the interface depth may be known from a borehole drilled at the coordinate origin. Constraining the depth to satisfy $99 \mathrm{~m}<h<101 \mathrm{~m}$ yields the model

$$
v_{1}=1490.01 \mathrm{~m} / \mathrm{s}, \quad v_{2}=2500.31 \mathrm{~m} / \mathrm{s}, \quad \phi=5.12^{\circ}, \quad \theta=45^{\circ}, \quad h=99.00 \mathrm{~m}
$$

which is substantially correct. Alternately, constraining the overburden velocity with the inequalities $1450 \mathrm{~m} / \mathrm{s}<v_{1}<1550 \mathrm{~m} / \mathrm{s}$ yields the erroneous one dimensional model

$$
v_{1}=1450 \mathrm{~m} / \mathrm{s}, \quad v_{2}=2493.96 \mathrm{~m} / \mathrm{s}, \quad \phi=0^{\circ}, \quad \theta=0^{\circ}, \quad h=94.25 \mathrm{~m} .
$$

Evidently, the broadside geometry illustrated in Figure 4.1b allows many solutions to this nonlinear traveltime inverse problem. However, if the inline head wave arrivals recorded by each spread are also included in the inversion $(M=32)$ then the correct model is recovered in seven iterations:

$$
v_{1}=1499.85 \mathrm{~m} / \mathrm{s}, \quad v_{2}=2499.99 \mathrm{~m} / \mathrm{s}, \quad \phi=5^{\circ}, \quad \theta=45^{\circ}, \quad h=99.98 \mathrm{~m} .
$$

Head waves recorded along strike provide excellent control on the velocity of the refracting medium. Equation (2.13) and Figure 2.4 indicate that the measured apparent velocity equals the true velocity in this situation. As expected, the inversion is degraded when the exact traveltimes are contaminated with uniformly distributed random errors on $\pm 4 \mathrm{~ms}$. The model

$$
v_{1}=1558.76 \mathrm{~m} / \mathrm{s}, \quad v_{2}=2664 \mathrm{~m} / \mathrm{s}, \quad \phi=4.99^{\circ}, \quad \theta=42.72^{\circ}, \quad h=106.86 \mathrm{~m},
$$

is obtained in seven iterations with a traveltime misfit of 2.4 ms .

### 4.3.2 Multiple layers

The earth model used for the second example consists of two layers overlying a halfspace and is defined by the parameter values

$$
\begin{array}{llll}
v_{1}=1500 \mathrm{~m} / \mathrm{s}, & \phi_{1}=0^{\circ}, & \theta_{1}=0^{\circ}, & z_{1}=0 \mathrm{~m} \\
v_{2}=2000 \mathrm{~m} / \mathrm{s}, & \phi_{2}=3^{\circ}, & \theta_{2}=180^{\circ}, & z_{2}=40 \mathrm{~m} \\
v_{3}=2500 \mathrm{~m} / \mathrm{s}, & \phi_{3}=5^{\circ}, & \theta_{3}=45^{\circ}, & z_{3}=100 \mathrm{~m} .
\end{array}
$$

As indicated previously, the model parameters $\phi_{1}, \theta_{1}$, and $z_{1}$ are not allowed to vary in the inversion. Hence, there are only $N=9$ parameters to estimate. Head wave traveltimes for
each critically refracting horizon are generated by evaluating equation (4.6a). The recording geometry used is the triangular array displayed in Figure 4.1a. Also, precritical offset arrivals are included in the traveltime dataset (in an actual field experiment, these times could be estimated by extrapolation or phantoming). Hence, there are a total of $M=174$ arrival times input to the inversion procedure.

If restrictive constraints are imposed on the model parameters, then it is possible to recover the correct solution with the multilayer inversion algorithm. For example, constraining the velocity $v_{1}$ and the depth $z_{2}$ to be equal to the true values allows the algorithm to iterate to the known solution from a one dimensional starting model. Also, if narrow bounds are placed on the velocities and depths ( $\pm 50 \mathrm{~m} / \mathrm{s}$ and $\pm 5 \mathrm{~m}$ about the true values, respectively) then the correct solution is obtained from a nearby initial model. However, a relatively unconstrained inversion invariably yields an erroneous result for this simulated experiment. It is probable that the objective function (4.4d), which only measures the misfit in the linearized data equations, has many local minima that preclude convergence to the desired global minimum.

### 4.4 Field data example

### 4.4.1 Peace River Arch broadside data

Deep seismic refraction data were acquired in the Peace River Arch (PRA) region of northern Alberta in 1985. In addition to four inline profiles, two broadside profiles were recorded. Figure 4.2 illustrates the source-receiver geometry for these two profiles. Shot A4 in the west is recorded by the north-south trending line A, and shot B4 in the north is recorded by the east-west trending line B. Offset distances range from $249-313 \mathrm{~km}$ on line A, and from $262-344 \mathrm{~km}$ on line B. Each receiver array subtends an azimuthal angle of $\sim 64^{\circ}$ relative to its source. The first arrivals at the recording sites are interpreted to be waves that are critically or near-critically refracted at the Moho (Zelt, 1989). Hence, inversion of the first break traveltimes can provide an estimate of the regional depth and dip of the Moho beneath the Peace River Arch.


Fig. 4.2. Broadside recording geometry for the Peace River Arch seismic experiment. Sources are indicated by asterisks and receivers by small crosses.

In this example, the broadside arrival times are inverted within the framework of a simple 'layer over a halfspace' earth model. No attempt is made to infer a detailed structural picture of the crust and Moho in the Peace River Arch region. Rather, the intent is to recover a large scale 3D model that can be subsequently refined by other traveltime interpretation/inversion methods. Initial estimates of the model parameters are obtained from crustal sections along lines A and B given by Zelt (1989, p. 103 and 112). These sections were derived by interpreting the inline refraction data of the PRA experiment via a trial-and-error forward modeling approach.

There are 83 first break picks from line A and 52 first break picks from line B . Thus, $M=135$ uniformly weighted traveltimes are input to the iterative inversion procedure. Since the first arrivals recorded at these long offset distances are emergent, the estimated picking error is relatively large ( $\pm 50 \mathrm{~ms}$ ).

### 4.4.2 Static corrections

The effect of variable near surface structure on the head wave arrival times can be reduced be applying static corrections to the first break picks. Static corrections are commonly applied to seismic reflection data for this same purpose. However, there is an important distinction between the corrections for the two types of data. In the refraction case, the static is designed to remove the refraction delay time influence of the near surface structure, and then replace it with the delay time contribution of a constant velocity medium. In the reflection case, the correction pertains to the vertical traveltime through the actual and replacement media. Statics application is an important preprocessing step for the PRA broadside data because the subsequent inversion assumes a very simple earth model. In effect, an attempt is made to 'make the data fit the model' more closely. The following development assumes that near surface velocity information is available. For the PRA experiment, this is obtained from well log data along the two profile lines.

An expression for the static correction at each receiver site is derived by approximating the local near surface velocity structure by a one dimensional stack of layers. The $i^{t h}$ layer is bounded from above and below by plane, horizontal interfaces at depths $z_{i}$ and $z_{i+1}$,
respectively. The P -wave velocity of the layer depends linearly on depth $z$ and is given by $v_{i}(z)=v_{i}(0)+a_{i} z$. In this situation, the raypath associated with a particular ray parameter $p$ is a circular arc. The upward propagation time across the layer is

$$
\begin{equation*}
\Delta t_{i}(p)=\frac{1}{a_{i}}\left[\cosh ^{-1}\left(\frac{1}{p v_{i}}\right)-\cosh ^{-1}\left(\frac{1}{p v_{i}^{\prime}}\right)\right] \tag{4.8}
\end{equation*}
$$

where $v_{i} \equiv v_{i}\left(z_{i}\right)=v_{i}(0)+a_{i} z_{i}$ and $v_{i}^{\prime} \equiv v_{i}\left(z_{i+1}\right)=v_{i}(0)+a_{i} z_{i+1}$. Symbols $v_{i}$ and $v_{i}^{\prime}$ denote the velocities at the top and bottom of the $i^{\text {th }}$ layer, respectively. The horizontal distance accumulated by this wave traversing layer $i$ is

$$
\begin{equation*}
\Delta x_{i}(p)=\frac{1}{p a_{i}}\left[\sqrt{1-\left(p v_{i}\right)^{2}}-\sqrt{1-\left(p v_{i}^{\prime}\right)^{2}}\right] \tag{4.9}
\end{equation*}
$$

If the wave is critically refracted on a horizontal interface at greater depth, then the ray parameter $p$ equals $1 / v_{c}$, where $v_{c}$ is the critical refraction velocity. The angle $\eta(z)$ that the raypath makes with the vertical at any depth is then determined by a velocity ratio: $\eta(z)=\sin ^{-1}\left[p v_{i}(z)\right]=\sin ^{-1}\left[v_{i}(z) / v_{c}\right]$. Using this result, equations (4.8) and (4.9) reduce to

$$
\begin{equation*}
\Delta t_{i}=\frac{1}{a_{i}} \ln \left[\frac{\left(1+\cos \eta_{i}\right) \sin \eta_{i}^{\prime}}{\left(1+\cos \eta_{i}^{\prime}\right) \sin \eta_{i}}\right], \quad \Delta x_{i}=\frac{v_{c}}{a_{i}}\left[\cos \eta_{i}-\cos \eta_{i}^{\prime}\right] \tag{4.10a,b}
\end{equation*}
$$

where $\eta_{i}=\sin ^{-1}\left(v_{i} / v_{c}\right)$ and $\eta_{i}^{\prime}=\sin ^{-1}\left(v_{i}^{\prime} / v_{c}\right) . \quad \eta_{i}$ and $\eta_{i}^{\prime}$ are the incident angles of the circular raypath segment at the top and bottom of layer $i$, respectively.

The refraction delay time contribution of the $i^{t h}$ layer is defined as

$$
\Delta \tau_{i} \equiv \Delta t_{i}-\frac{\Delta x_{i}}{v_{c}}
$$

Substituting from equations (4.10) yields

$$
\begin{equation*}
\Delta \tau_{i}=\frac{1}{a_{i}} \ln \left[\frac{Q_{i}}{Q_{i}^{\prime}}\right] \tag{4.11}
\end{equation*}
$$

where quantities $Q_{i}$ and $Q_{i}^{\prime}$ are given by

$$
Q_{i}=\frac{1+\cos \eta_{i}}{\sin \eta_{i} \exp \left(\cos \eta_{i}\right)}, \quad \quad Q_{i}^{\prime}=\frac{1+\cos \eta_{i}^{\prime}}{\sin \eta_{i}^{\prime} \exp \left(\cos \eta_{i}^{\prime}\right)} .
$$

As a check on the correctness of this result, examine the case where the vertical velocity gradient of the layer vanishes $\left(a_{i} \rightarrow 0\right)$. In this situation, the raypath becomes a straight line segment, and the angles $\eta_{i}$ and $\eta_{i}^{\prime}$ both approach the same angle $\eta_{0}=\sin ^{-1}\left[v_{i}(0) / v_{c}\right]$. Application of L'Hopital's rule to equation (4.11) then yields

$$
\lim _{a_{i} \rightarrow 0} \Delta \tau_{i}=\frac{\left(z_{i+1}-z_{i}\right) \cos \eta_{0}}{v_{i}(0)}=\frac{h_{i} \cos \eta_{0}}{v_{i}(0)}
$$

This is the proper expression for the one-way delay time contribution of a layer with thickness $h_{i}$ and uniform velocity $v_{i}(0)$.

The one-way refraction delay time associated with a stack of $n$ horizontal layers is obtained by summing expression (4.11):

$$
\begin{equation*}
\Delta \tau=\sum_{i=1}^{n} \Delta \tau_{i}=\sum_{i=1}^{n} \frac{1}{a_{i}} \ln \left[\frac{Q_{i}}{Q_{i}^{\prime}}\right] \tag{4.12}
\end{equation*}
$$

This formula naturally accommodates any velocity discontinuities at the interfaces (i.e., $v_{i-1}^{\prime} \neq v_{i}$ ). However, if the velocity function is continuous between all layers (as in the PRA
example), then $Q_{i}^{\prime}=Q_{i+1}$ and expression (4.12) simplifies to

$$
\begin{equation*}
\Delta \tau=\sum_{i=1}^{n} \frac{1}{a_{i}} \ln \left[\frac{Q_{i}}{Q_{i+1}}\right] \tag{4.13}
\end{equation*}
$$

The static correction applied to the picked arrival times is designed to remove the oneway refraction delay time influence of the layered near surface structure, and add the delay time associated with a uniform replacement medium of equal thickness. If the replacement velocity is designated $v_{r}$, then the static shift is

$$
\begin{equation*}
t_{s t a t}=-\Delta \tau+\frac{\left(\sum_{i=1}^{n} h_{i}\right) \cos \eta_{r}}{v_{r}} \tag{4.14}
\end{equation*}
$$

where $\eta_{r}=\sin ^{-1}\left(v_{r} / v_{c}\right)$. Although this formula has been derived within the context of upward wave propagation through a set of horizontal layers, it also applies to the downward propagating portion of the total raypath. Hence, it may be used for the computation of either a source site static or a receiver site static.

Near-surface velocity information along lines $A$ and $B$ of the PRA experiment are obtained from models developed by Zelt (1989, p. 40) from well logs. There are eight wells along line $A$ and seven wells along line $B$. At each wellsite, the velocity model is approximated by a one dimensional stack of layers with linear velocities, and a static correction is calculated via formula (4.14). The critical and replacement velocities are assumed to be $v_{c}=8.25 \mathrm{~km} / \mathrm{s}$ and $v_{r}=6.6 \mathrm{~km} / \mathrm{s}$, respectively. These wellsite static corrections are then interpolated/extrapolated to the receiver locations on each line using the following procedure. First, a straight line is fitted to the coordinates of all recording stations and wellsites along a profile using the York algorithm (York, 1966, 1967, 1969; Williamson, 1968). This is the proper line fitting technique to use in this situation because both the northing and easting coordinates are subject to positioning errors. The York algorithm also automatically returns the coordinates of the perpendicular projections of the points to the fitted straight
line. Finally, the wellsite statics are linearly interpolated/extrapolated in these projected coordinates to the receiver sites. Figure 4.3a displays the computed receiver static corrections for both lines; squares denote the wellsite statics calculated from expression (4.14). The magnitudes of the statics are approximately the same on both lines ( $\sim 300 \mathrm{~ms}$ ), although there is a noticeable increase toward the southern end of line A.

The static corrections calculated with the above procedure apply only to receiver sites. However, since shot A4 is located at the end of profile line B, a source static is readily estimated for this shot from the nearest computed receiver correction ( -290 ms ). A source static for the isolated shot B4 must be assumed; the value adopted here is -300 ms .

### 4.4.3. Inversion results

Initial estimates of the model parameters are inferred from the interpreted crustal sections in Zelt (1989, p. 103 and 112). These starting parameter values are

$$
v_{1}=6.5 \mathrm{~km} / \mathrm{s}, \quad v_{2}=8.25 \mathrm{~km} / \mathrm{s}, \quad \phi=2^{\circ}, \quad \theta=270^{\circ}, \quad h=40 \mathrm{~km}
$$

The same sections are also used to provide lower and upper bounds on the earth model parameters:

$$
\begin{aligned}
& v_{1}^{-}=6.0 \mathrm{~km} / \mathrm{s}, \quad v_{2}^{-}=7.5 \mathrm{~km} / \mathrm{s}, \quad \phi^{-}=0^{\circ}, \quad \theta^{-}=0^{\circ}, \quad h^{-}=38 \mathrm{~km}, \\
& v_{1}^{+}=7.5 \mathrm{~km} / \mathrm{s}, \quad v_{2}^{+}=8.5 \mathrm{~km} / \mathrm{s}, \quad \phi^{+}=15^{\circ}, \quad \theta^{+}=360^{\circ}, \quad h^{+}=42 \mathrm{~km} .
\end{aligned}
$$

The bounds on the velocities must be transformed to equivalent bounds on the parameters $s_{1}$ (slowness) and $i_{c}$ (critical angle) that are used by the inversion algorithm. After five iterations, the following model is returned:

$$
v_{1}=6.21 \mathrm{~km} / \mathrm{s}, \quad v_{2}=8.50 \mathrm{~km} / \mathrm{s}, \quad \phi=2.75^{\circ}, \quad \theta=299.74^{\circ}, \quad h=42 \mathrm{~km}
$$



Fig. 4.3. Receiver static functions for lines $A$ and $B$ of the PRA experiment. Squares refer to well locations where near surface velocity logs are available. (a) Statics calculated with $v_{c}=8.25 \mathrm{~km} / \mathrm{s}$ and $v_{r}=6.6 \mathrm{~km} / \mathrm{s}$. (b) Statics calculated with $v_{c}=8.5 \mathrm{~km} / \mathrm{s}$ and $v_{r}=6.2$ $\mathrm{km} / \mathrm{s}$.

The initial traveltime misfit of 658 ms is reduced to 172 ms ; iterations cease when the relative change in the misfit is less than $1 \%$. The velocities $v_{1}$ and $v_{2}$ of this solution differ from the replacement and critical velocities previously used for calculating the static corrections. Hence, the statics are recomputed with $v_{r}=6.2 \mathrm{~km} / \mathrm{s}$ and $v_{c}=8.5 \mathrm{~km} / \mathrm{s}$ and applied again to the picked arrival times. Figure 4.3b illustrates this second set of receiver static corrections. New source statics are assumed to be -210 ms for shot A4 and -250 ms for shot B4. Then, initiating the inversion algorithm with the same starting model yields

$$
v_{1}=6.17 \mathrm{~km} / \mathrm{s}, \quad v_{2}=8.50 \mathrm{~km} / \mathrm{s}, \quad \phi=2.73^{\circ}, \quad \theta=300.86^{\circ}, \quad h=42 \mathrm{~km},
$$

in five iterations with a misfit of 171 ms . Although this model is not significantly different from the previous one, it is consistent with the assumptions used for calculating statics.

A comparison between the observed arrival times (after static corrections) and the traveltimes predicted by the model produced by the inversion is displayed in Figure 4.4. Evidently, the simple five parameter earth model provides an adequate explanation for the gross character of the broadside arrival time curves. Small scale variations in the predicted times (solid curves) are due strictly to recording geometry irregularities, rather than any subsurface structural complications. However, a large component of the total traveltime misfit must be attributed to structure or velocity variations that are not modeled in the inversion procedure. This misfit is too large to be accounted for by random picking errors alone. For example, the predicted times are systematically greater than the observed times throughout the central portion of line B. This suggests that the Moho north of line B is not adequately represented by a plane interface bounded by uniform velocities.

Vertical depths to the Moho calculated from the inversion results are posted on a plan view of the PRA recording geometry in Figure 4.5. A depth trend is readily apparent, although the depth at the southern end of line A is probably too large. A southeastward dip of the Moho is suggested by interpretations of the inline refraction data on lines $A$ and $B$, but not on the other lines of the PRA experiment (Zelt, 1989). Also, the velocities recovered by the 3D inversion are broadly consistent with those obtained by Zelt (1989). Of course,


Fig. 4.4. Comparison between predicted traveltimes (solid lines) and observed traveltimes (triangles) on broadside lines A and B of the PRA experiment.


Fig. 4.5. Vertical depths (km) to the Moho in the PRA region inferred fom the inversion results.
precise agreement cannot be expected. The mean crustal velocity is lower ( $6.17 \mathrm{~km} / \mathrm{s}$ vs. $\sim 6.6 \mathrm{~km} / \mathrm{s}$ ) and the sub-Moho velocity is higher ( $8.5 \mathrm{~km} / \mathrm{s}$ vs. $\sim 8.2 \mathrm{~km} / \mathrm{s}$ ). However, Zelt inferred a $P_{n}$ velocity of $8.4 \mathrm{~km} / \mathrm{s}$ along the northern half of line A and the eastern quarter of line $B$.

An independent check on the validity of the 3D inversion is provided by a method proposed by Zelt (1989). Using a simple two dimensional earth model, each arrival time recorded along a broadside profile can be inverted for an estimate of Moho depth beneath the associated receiver site. A depth profile for the Moho can then be constructed by plotting these depth estimates side-by-side. Zelt (1989) accomplished this inversion using ray tracing techniques. However, as the following derivation demonstrates, a straightforward mathematical solution to the problem is possible. In the two dimensional situation, head wave traveltime is

$$
\begin{equation*}
T(X)=\frac{\sin \left(i_{c}+\varphi\right)}{v_{1}} X+\frac{2 h\left(x_{S}\right) \cos \varphi \cos i_{c}}{v_{1}} \tag{4.15}
\end{equation*}
$$

where $i_{c}=\sin ^{-1}\left(v_{1} / v_{2}\right)$ is the critical angle and $h\left(x_{S}\right)$ is the vertical thickness of the layer at the source location. The interface dip angle $\varphi$ may be positive, zero, or negative. Substituting $h\left(x_{S}\right)=h(0)+x_{S} \tan \varphi$ into this expression yields

$$
T(X)=\frac{\sin \left(i_{c}+\varphi\right)}{v_{1}} X+\frac{2 h(0) \cos \varphi \cos i_{c}}{v_{1}}+\frac{2 x_{S} \sin \varphi \cos i_{c}}{v_{1}}
$$

$h(0)$ is the layer thickness at the coordinate origin. This relation is rewritten as a quadratic form in $\cos \varphi$ :

$$
\begin{equation*}
A \cos ^{2} \varphi+B \cos \varphi+C=0 \tag{4.16}
\end{equation*}
$$

The three coefficients depend on earth model parameters $\left\{v_{1}, v_{2}, h(0)\right\}$ and measured data $\left\{x_{S}, X, T\right\}$ :

$$
\begin{gathered}
A=1+\frac{4 h(0) \cos i_{c}}{X}\left[\sin i_{c}+\left[\frac{h(0)}{X}+\left(1+\frac{x_{S}}{X}\right) \frac{x_{S}}{h(0)}\right] \cos i_{c}\right] \\
B=-2\left[\frac{v_{1} T}{X}\right]\left[\sin i_{c}+\frac{2 h(0)}{X} \cos i_{c}\right] \\
C=\left[\frac{v_{1} T}{X}\right]^{2}-\left[1+\frac{2 x_{S}}{X}\right]^{2} \cos ^{2} i_{c}
\end{gathered}
$$

If values for $v_{1}, v_{2}$, and $h(0)$ are known (or assumed), then these coefficients can be evaluated numerically. Solution of equation (4.16) for $\cos \varphi$ is via the quadratic formula:

$$
\begin{equation*}
\cos \varphi=\frac{-B+\sqrt{B^{2}-4 A C}}{2 A} \tag{4.17}
\end{equation*}
$$

where the positive root is chosen by analyzing the form of the right hand side as $X \rightarrow+\infty$. There is still a two-fold ambiguity in determining the dip angle $\varphi$ from $\cos \varphi$. This ambiguity is easily resolved by ensuring that the traveltime predicted by formula (4.15) agrees with the known traveltime $T$. Finally, once the dip angle is determined, the vertical depth to the refractor at any inline position $x$ can be calculated via $h(x)=h(0)+x \tan \varphi$.

Figure 4.6 compares Moho depths calculated via the above method with those inferred from the parameters recovered by the 3D inversion. Vertical depths beneath the recording stations on each line are plotted. The coefficients in the quadratic (4.16) are evaluated with the parameters obtained from the 3 D inversion procedure ( $v_{1}=6.17 \mathrm{~km} / \mathrm{s}, v_{2}=8.50 \mathrm{~km} / \mathrm{s}$, $h(0)=37.6 \mathrm{~km}$ for shot A 4 , and $h(0)=34.8 \mathrm{~km}$ for shot B 4 ). Short wavelength variations in the computed curves are artifacts of the traveltime picking errors, and should be ignored. There is close agreement in the Moho depth trends calculated by these two completely
different techniques, especially along line A. The larger departure of the two depth curves along line $B$ suggests structural complexity north of this line in the Peace River Arch region.

Although the two dimensional inversion method is simple and appears to yield reasonable depth estimates, it requires the assumption of numerical values for the three unknown model parameters $\left\{v_{1}, v_{2}, h(0)\right\}$. In contrast, the 3D inversion technique yields simultaneous estimates of all of the relevant earth model parameters. Moreover, since it accomplishes a joint inversion of all of the error contaminated traveltime data, it is more robust than the 2D method.



Fig. 4.6. Comparison of vertical depths to the Moho beneath the receivers of lines A and B calculated by two different methods. Smooth curves in each panel are depths inferred from 3D inversion results. Jagged curves are obtained from a simple 2D inversion method.

### 4.5 Conclusion

The inversion of head wave arrival times for three dimensional planar structure is posed as a constrained, parameter optimization problem. The iterative inversion algorithm de-
scribed here has the ability to converge to a realistic solution provided that (i) the data acquisition geometry is adequate, and (ii) sufficient a priori constraints are available. However, precise definitions of 'adequate' and 'sufficient' in this context are not known. Nevertheless, the inversion procedure provides a useful tool for examining these phenomena. In particular, the single-layer variant of the algorithm can be used as an aid in designing recording geometries for detecting and resolving three dimensional dipping structure. Also, investigation of both synthetic and field recorded datasets indicates that it can be successfully applied to the inversion of broadside refraction data. Currently, there is a lack of techniques for effective -interpretation of such data. Although the multiple-layer version of the algorithm exhibits a greater tendency to converge to an erroneous result, several successful inversions have been achieved with the inclusion of sufficient constriaints. Generalization of these algorithms to three dimensional recording geometries would be a useful extension. As indicated in Chapter 3 , the forward modeling procedure can easily accommodate buried sources and/or receivers. Moreover, well log information regarding interface depths and layer velocities would provide the necessary constraint information for the multilayer algorithm.

## CHAPTER 5

## POINT DEPTH AND DIP ESTIMATES FROM REFRACTION TRAVELTIMES

### 5.1 Introduction

The Generalized Reciprocal Method (GRM) is a technique for delineating an undulating subsurface interface from refraction traveltime data recorded by inline forward and reverse profiles. After the refracted arrivals from a particular horizon are identified, simple linear combinations of the picked times are used to form two analysis functions: (i) the velocity analysis function, and (ii) the generalized time-depth function. Each is plotted with respect to receiver position along the profile line. The slope of the velocity analysis function is a measure of the P -wave velocity of the refracting layer. The generalized time-depth function is a measure of the depth to the critically refracting interface, expressed in units of time. Hence, it is analogous to a reflection event on a seismic reflection time section. Conversion of the time-depth function to actual depth values requires velocity information for both the overburden and the refracting layer.

The GRM was developed by Palmer $(1980,1981)$ as an extension of the conventional reciprocal time method for interpreting refraction arrival times (Hawkins, 1961). Although it has been successfully applied to the problem of mapping undulating refractors, the mathematical formulation of the method is based on a two dimensional earth model with plane interfaces. Palmer adopts a model parameterization originally proposed by Ewing et al. (1939): the thickness of a layer is measured normal to its basal interface. Moreover, Palmer (1980, p. 3) claims that an alternative parameterization characterized by vertical layer thicknesses is "not suitable" for the derivation of the GRM analysis functions. This statement is incorrect. All of the GRM analysis tools can be rigorously derived from the two dimensional version of equation (3.23) for head wave traveltime. This development is the subject of the first portion of this chapter. The use of vertical layer thicknesses results in substantially simplified derivations. Hence, this reformulation of the theoretical basis of the GRM is quite
useful for educational purposes. However, a practical benefit related to refractor depth profile construction is also realized. The equations developed here allow point depth estimates of the interface to be calculated from the generalized time-depth function. Computation of a refractor depth profile then reduces to an interpolation problem (treated in Chapter 6). In contrast, Palmer's time-depth function yields a circular locus of possible refractor positions. Depth profile calculation then involves the more complicated task of constructing an envelope to a set of circular arcs.

As indicated above, the GRM is founded upon a two dimensional earth model with plane interfaces. A traveltime inversion technique that explicitly incorporates nonplane refracting horizons into the model should yield more accurate results. Hence, the second portion of this chapter proposes a 2D head wave traveltime interpretation method tentatively named critical offset refraction profiling. This inversion technique accommodates undulations in both the refracting interface and the surface, as well as horizontal variations in the velocity of the refracting layer. Detection and definition of such structural and velocity variations are subjects of current research (Palmer, 1991). Critical offset refraction profiling yields a mathematically exact solution to the 2D seismic refraction problem; point values of interface depth, interface dip, and refractor velocity are obtained. A continuous refractor depth profile can then constructed by interpolation. The dip estimates provide additional constraints on the interpolant.

### 5.2 Specialization to a 2D model

Prior to deriving the GRM analysis functions, the head wave traveltime equation (3.23) must be specialized to a two dimensional earth model. In the three dimensional model, the orientation of interface $i$ is described by a dip angle $\phi_{i}\left(0 \leq \phi_{i}<\pi / 2\right)$ and an azimuth angle $\theta_{i}\left(0 \leq \theta_{i}<2 \pi\right)$. Note that the dip angle $\phi_{i}$ is non-negative. In terms of these two angles, the unit normal to the $i^{\text {th }}$ interface is

$$
\begin{equation*}
\mathbf{n}_{\boldsymbol{i}}=\left(\sin \phi_{i} \cos \theta_{i}\right) \mathbf{i}+\left(\sin \phi_{i} \sin \theta_{i}\right) \mathbf{j}+\left(\cos \phi_{i}\right) \mathbf{k} \tag{5.1a}
\end{equation*}
$$

and the vertical depth of the interface is

$$
\begin{equation*}
z_{i}(x, y)=z_{i}(0,0)-\tan \phi_{i}\left(x \cos \theta_{i}+y \sin \theta_{i}\right) \tag{5.1b}
\end{equation*}
$$

The vertical thickness of layer $i$ (bounded from above and below by interfaces $i$ and $i+1$, respectively) is
$h_{i}(x, y)=h_{i}(0,0)+x\left(\cos \theta_{i} \tan \phi_{i}-\cos \theta_{i+1} \tan \phi_{i+1}\right)+y\left(\sin \theta_{i} \tan \phi_{i}-\sin \theta_{i+1} \tan \phi_{i+1}\right)$.

If the earth model is two dimensional, then all interface azimuth angles $\theta_{i}$ are restricted to the two values 0 or $\pi$. In this case, the above equations reduce to

$$
\begin{gather*}
\mathbf{n}_{i}=\left(\sin \phi_{i} \cos \theta_{i}\right) \mathbf{i}+\left(\cos \phi_{i}\right) \mathbf{k}  \tag{5.2a}\\
z_{i}(x)=z_{i}(0)-x \cos \theta_{i} \tan \phi_{i}  \tag{5.2b}\\
h_{i}(x)=h_{i}(0)+x\left(\cos \theta_{i} \tan \phi_{i}-\cos \theta_{i+1} \tan \phi_{i+1}\right) . \tag{5.2c}
\end{gather*}
$$

Thus, the normal $n_{i}$ has no $y$-component, and the depth $z_{i}(x)$ and thickness $h_{i}(x)$ are independent of the $y$-coordinate. For subsequent analysis of the two dimensional refraction problem, it is convenient to reparameterize the earth model in terms of interface dip angles that may be positive or negative. Hence, let the symbol $\varphi_{i}$ refer to an interface dip angle in a two dimensional layered earth. $\varphi_{i}$ is an acute angle ( $0 \leq\left|\varphi_{i}\right|<\pi / 2$ ) measured with respect to the $+x$ axis; it is considered positive (negative) if the angle opens in the clockwise (counterclockwise) sense. This angle is related to the 3D interface orientation angles via $\varphi_{i}=-\phi_{i}\left(\cos \theta_{i}\right)$, where the azimuth $\theta_{i}$ is restricted to the two values 0 or $\pi$. Rewriting this relation as $\phi_{i}=-\varphi_{i}\left(\cos \theta_{i}\right)$ and then substituting into equations (5.2a,b,c) gives the familiar expressions

$$
\begin{gather*}
\mathbf{n}_{i}=\left(-\sin \varphi_{i}\right) \mathbf{i}+\left(\cos \varphi_{i}\right) \mathbf{k}  \tag{5.3a}\\
z_{i}(x)=z_{i}(0)+x \tan \varphi_{i}  \tag{5.3b}\\
h_{i}(x)=h_{i}(0)+x\left[\frac{\sin \left(\varphi_{i+1}-\varphi_{i}\right)}{\cos \varphi_{i+1} \cos \varphi_{i}}\right] . \tag{5.3c}
\end{gather*}
$$

The head wave traveltime formula relevant to GRM analysis is equation (3.23). This is specialized to a two dimensional model by setting $y_{S}=y_{R}=0, n_{k, x}=-\sin \varphi_{k}$, and $n_{k, z}=\cos \varphi_{k}$. Equation (3.23) becomes

$$
\begin{equation*}
T_{k}\left(x_{S}, x_{R}\right)=\sum_{i=1}^{k-1} \frac{h_{i}\left(x_{S}\right) p_{i k, z}-h_{i}\left(x_{R}\right) q_{i k, z}}{v_{i}}+\frac{\left(x_{R}-x_{S}\right)\left(p_{k k, x} \cos \varphi_{k}+p_{k k, z} \sin \varphi_{k}\right)}{v_{k} \cos \varphi_{k}} \tag{5.4}
\end{equation*}
$$

A further simplification is obtained by expressing the components of the unit propagation vectors $\mathbf{p}_{i k}$ and $\mathbf{q}_{i k}$ in terms of orientation angles. In the three dimensional situation, each of these vectors is described by two angles:

$$
\begin{align*}
& \mathbf{p}_{i k}=\left(\sin \alpha_{i k} \cos \gamma_{i k}\right) \mathbf{i}+\left(\sin \alpha_{i k} \sin \gamma_{i k}\right) \mathbf{j}+\left(\cos \alpha_{i k}\right) \mathbf{k}  \tag{5.5a}\\
& \mathbf{q}_{i k}=\left(\sin \beta_{i k} \cos \delta_{i k}\right) \mathbf{i}+\left(\sin \beta_{i k} \sin \delta_{i k}\right) \mathbf{j}+\left(\cos \beta_{i k}\right) \mathbf{k} \tag{5.5b}
\end{align*}
$$

$\alpha_{i k}$ and $\beta_{i k}$ are polar angles measured from the $+z$ axis $\left(0 \leq \alpha_{i k}, \beta_{i k} \leq \pi\right)$, and $\gamma_{i k}$ and $\delta_{i k}$ are azimuthal angles measured from the $+x$ axis ( $0 \leq \gamma_{i k}, \delta_{i k}<2 \pi$ ). If the recording profile is oriented perpendicular to the strike of the subsurface horizons, then all propagation vectors are confined to the $x z$ plane. The azimuth angles equal 0 or $\pi$, and expressions $(5.5 \mathrm{a}, \mathrm{b})$ reduce to

$$
\begin{equation*}
\mathbf{p}_{i k}=\left( \pm \sin \alpha_{i k}\right) \mathbf{i}+\left(\cos \alpha_{i k}\right) \mathbf{k}, \quad \quad \mathbf{q}_{i k}=\left( \pm \sin \beta_{i k}\right) \mathbf{i}+\left(\cos \beta_{i k}\right) \mathbf{k} \tag{5.6a,b}
\end{equation*}
$$

The propagation direction of the critically refracted raypath segment is obtained by straightforward geometric analysis:

$$
\begin{equation*}
\mathbf{p}_{k k}= \pm\left[\left(\cos \varphi_{k}\right) \mathbf{i}+\left(\sin \varphi_{k}\right) \mathbf{k}\right] \tag{5.6c}
\end{equation*}
$$

where the plus sign is used for $x_{R}>x_{S}$ and the negative sign for $x_{R}<x_{S}$. Finally, substituting equations ( $5.6 \mathrm{a}, \mathrm{b}, \mathrm{c}$ ) into (5.4) gives the remarkably simple result

$$
\begin{equation*}
T_{k}\left(x_{S}, x_{R}\right)=\sum_{i=1}^{k-1} \frac{h_{i}\left(x_{S}\right) \cos \alpha_{i k}-h_{i}\left(x_{R}\right) \cos \beta_{i k}}{v_{i}}+\frac{\left|x_{R}-x_{S}\right|}{v_{k} \cos \varphi_{k}} \tag{5.7}
\end{equation*}
$$

Equation (5.7) is a novel expression for surface-to-surface head wave traveltime that forms the basis for an alternative development of the GRM. It differs from the analogous equation given by Palmer ( 1980 , p. 5) in several important ways: (i) layers are characterized by vertical thicknesses below source and receiver, (ii) raypath orientation angles are measured with respect to the vertical, (iii) the coefficient multiplying the source-receiver offset distance depends only on critical refractor quantities (i.e., velocity $v_{k}$ and $\operatorname{dip} \varphi_{k}$ ), and (iv) the earth's surface may be nonhorizontal. These attributes facilitate a straightforward derivation of the various GRM analysis tools.

The raypath angles in equation (5.7) depend on the recording profile azimuth $\Psi$, which in turn is restricted to the two values 0 or $\pi$. However, raypath reciprocity requires that $\alpha_{i k}(\pi)=\pi-\beta_{i k}(0)$ and $\beta_{i k}(\pi)=\pi-\alpha_{i k}(0)$. These expressions imply that

$$
\begin{equation*}
\cos \alpha_{i k}(\pi)=-\cos \beta_{i k}(0), \quad \cos \beta_{i k}(\pi)=-\cos \alpha_{i k}(0) \tag{5.8a,b}
\end{equation*}
$$

Using these relations, it is easy to demonstrate that the traveltime formula (5.7) satisfies source-receiver reciprocity: $T_{k}\left(x_{S}, x_{R}\right)=T_{k}\left(x_{R}, x_{S}\right)$.

### 5.3 Derivation of GRM parameters

The two dimensional earth model used to derive the GRM analysis functions consists of uniform velocity layers bounded by plane, dipping interfaces. Figure 5.1 depicts the relevant critically refracted raypaths. Although the subsequent mathematical development assumes a multilayered earth, this single layer model is useful for illustrative purposes. Receivers are deployed between two sources $S_{1}$ and $S_{2}$ with horizontal coordinates $x_{S_{1}}$ and $x_{S_{2}}$, respectively; $R$ denotes a typical receiver with coordinate $x_{R}$. The velocity analysis function and the generalized time-depth function are constructed from the traveltimes of forward and reverse propagating head waves that arrive at the positions $x_{R}+L / 2$ and $x_{R}-L / 2$, respectively. $L$ is an adjustable separation distance. If $L=0$, then arrivals at the single receiver at $x_{R}$ are examined, and the GRM reduces to the conventional reciprocal method. GRM analysis seeks to determine the particular separation $L$ such that the upgoing segments of the two raypaths depart from the same point on the critically refracting horizon. This determination is made by evaluating the velocity analysis function and the time-depth function over a range of separation distances $L$, and then interpreting the plotted curves. The interpretive criteria that are used for selecting the appropriate distance $L$ are described by Palmer $(1980,1981,1990,1991)$ and are not investigated here.

### 5.3.1 Velocity analysis function

The velocity analysis function for layer $k$ is defined as

$$
\begin{equation*}
T_{v}\left(x_{R} ; L\right) \equiv \frac{1}{2}\left[T_{k}\left(x_{S_{1}}, x_{R}+L / 2\right)-T_{k}\left(x_{S_{2}}, x_{R}-L / 2\right)+T_{k}\left(x_{S_{1}}, x_{S_{2}}\right)\right] \tag{5.9}
\end{equation*}
$$

The last term on the right hand side is the reciprocal time (the shot-to-shot traveltime).


Fig. 5.1. Critically refracted raypaths used to derive the GRM analysis functions. $R$ denotes a receiver located between the two surface sources $S_{1}$ and $S_{2}$.

Substituting the head wave traveltime formula (5.7) into this definition yields

$$
\begin{align*}
& T_{v}\left(x_{R} ; L\right)=\sum_{i=1}^{k-1} \frac{h_{i}\left(x_{S_{1}}\right) \cos \alpha_{i k}(0)}{v_{i}}-\sum_{i=1}^{k-1} \frac{h_{i}\left(x_{S_{2}}\right)}{v_{i}}\left[\frac{\cos \alpha_{i k}(\pi)+\cos \beta_{i k}(0)}{2}\right] \\
& +\sum_{i=1}^{k-1} \frac{h_{i}\left(x_{R}-L / 2\right) \cos \beta_{i k}(\pi)-h_{i}\left(x_{R}+L / 2\right) \cos \beta_{i k}(0)}{2 v_{i}}+\frac{x_{R}-x_{S_{1}}}{v_{k} \cos \varphi_{k}} \tag{5.10}
\end{align*}
$$

where the dependence of the raypath angles on recording profile azimuth is explicit. Using equation (5.3c), the layer thicknesses at the horizontal coordinates $x_{R} \pm L / 2$ are written as

$$
h_{i}\left(x_{R} \pm L / 2\right)=h_{i}\left(x_{S_{1}}\right)+\left(x_{R}-x_{S_{1}} \pm L / 2\right)\left[\frac{\sin \left(\varphi_{i+1}-\varphi_{i}\right)}{\cos \varphi_{i+1} \cos \varphi_{i}}\right]
$$

Substituting this expression and relations (5.8a,b) into (5.10) and reducing yield the result

$$
\begin{align*}
& \quad T_{v}\left(x_{R} ; L\right)=\sum_{i=1}^{k-1} \frac{1}{v_{i}}\left\{h_{i}\left(x_{S_{1}}\right)+\frac{L}{2}\left[\frac{\sin \left(\varphi_{i+1}-\varphi_{i}\right)}{\cos \varphi_{i+1} \cos \varphi_{i}}\right]\right\} \frac{\cos \alpha_{i k}(0)-\cos \beta_{i k}(0)}{2} \\
& +\left(x_{R}-x_{S_{1}}\right)\left\{\frac{1}{v_{k} \cos \varphi_{k}}-\sum_{i=1}^{k-1} \frac{1}{v_{i}}\left[\frac{\sin \left(\varphi_{i+1}-\varphi_{i}\right)}{\cos \varphi_{i+1} \cos \varphi_{i}}\right] \frac{\cos \alpha_{i k}(0)+\cos \beta_{i k}(0)}{2}\right\} . \tag{5.11}
\end{align*}
$$

Hence, for a model consisting of uniform velocity layers bounded by plane interfaces, the velocity analysis function is a linear function of the distance $\left(x_{R}-x_{S_{1}}\right)$. The slope of this line is used to estimate the velocity of the critically refracting layer. The intercept has a simple interpretation that is given in section 5.3.5.

### 5.3.2 Apparent horizontal velocity

The apparent horizontal velocity of layer $k$ is defined as

$$
\begin{equation*}
\frac{1}{V_{a}} \equiv \frac{\partial T_{v}\left(x_{R} ; L\right)}{\partial x_{R}} \tag{5.12}
\end{equation*}
$$

Using expression (5.11) for the velocity analysis function immediately results in

$$
\begin{equation*}
\frac{1}{V_{a}}=\frac{1}{v_{k} \cos \varphi_{k}}-\sum_{i=1}^{k-1} \frac{1}{v_{i}}\left[\frac{\sin \left(\varphi_{i+1}-\varphi_{i}\right)}{\cos \varphi_{i+1} \cos \varphi_{i}}\right] \frac{\cos \alpha_{i k}(0)+\cos \beta_{i k}(0)}{2} \tag{5.13}
\end{equation*}
$$

Thus, the apparent horizontal velocity of a refracting layer is influenced by all velocities and dips in the overburden, as well as by the dip of the critically refracting horizon. However, for dip angles commonly encountered in refraction exploration, formula (5.12) yields a reasonably accurate estimate of $v_{k}$. A quantitative understanding of the accuracy of (5.12) can be obtained by analyzing a specific numerical example. Palmer's (1980, p. 10) shallow subsurface model is quite useful for this purpose. Since the interfaces have unusually large
dips, this model provides a strong test of the utility of (5.12) for velocity estimation. Furthermore, a comparison with the numerical values of $V_{a}$ computed via Palmer's (1980, p. 9) formula serves as a check on the correctness of the current derivation.

Palmer (1980) examines a four layer model defined by the parameters

$$
\begin{array}{lll}
z_{1}(0)=0.00 \mathrm{~m}, & \varphi_{1}=0^{\circ}, & v_{1}=1000 \mathrm{~m} / \mathrm{s} \\
z_{2}(0)=20.31 \mathrm{~m}, & \varphi_{2}=-10^{\circ}, & v_{2}=1800 \mathrm{~m} / \mathrm{s} \\
z_{3}(0)=59.55 \mathrm{~m}, & \varphi_{3}=5^{\circ}, & v_{3}=2500 \mathrm{~m} / \mathrm{s}, \\
z_{4}(0)=80.25 \mathrm{~m}, & \varphi_{4}=-15^{\circ}, & v_{4}=4000 \mathrm{~m} / \mathrm{s}
\end{array}
$$

(He does not give the vertical depths to the interfaces, but these may be derived from the specified geometric parameters). In order to evaluate equation (5.13), the raypath angles $\alpha_{i k}(0)$ and $\beta_{i k}(0)$ must be determined. Application of Snell's law of refraction at interface $i$ yields the two recursion relations
$\alpha_{i-1, k}=\sin ^{-1}\left[\frac{v_{i-1}}{v_{i}} \sin \left(\alpha_{i k}+\varphi_{i}\right)\right]-\varphi_{i}, \quad \quad \beta_{i-1, k}=\pi-\sin ^{-1}\left[\frac{v_{i-1}}{v_{i}} \sin \left(\beta_{i k}+\varphi_{i}\right)\right]-\varphi_{i}$.

Starting with $\alpha_{k k}(0)+\varphi_{k}=\beta_{k k}(0)+\varphi_{k}=\pi / 2$, these relations are evaluated successively for $i=k, k-1, \ldots, 3,2$ to obtain

$$
\begin{array}{ll}
\alpha_{12}=43.7490^{\circ}, & \beta_{12}=156.2510^{\circ}, \\
\alpha_{13}=26.6536^{\circ}, & \beta_{13}=160.9118^{\circ}, \\
\alpha_{23}=41.0545^{\circ}, & \beta_{23}=128.9455^{\circ}, \\
\alpha_{14}=22.5156^{\circ}, & \beta_{14}=174.7119^{\circ}, \\
\alpha_{24}=32.9588^{\circ}, & \beta_{24}=161.6659^{\circ}, \\
\alpha_{34}=53.6822^{\circ}, & \beta_{34}=156.3178^{\circ},
\end{array}
$$

Finally, using these numerical values for the raypath angles in formula (5.13) yields

$$
k=2: V_{a}=1827.77 \mathrm{~m} / \mathrm{s}, \quad k=3: \quad V_{a}=2578.59 \mathrm{~m} / \mathrm{s}, \quad k=4: \quad V_{a}=4209.65 \mathrm{~m} / \mathrm{s}
$$

These calculated apparent velocities agree closely with Palmer's (1980, p. 11) values, and thus confirm the validity of equation (5.13). Moreover, it is evident that $V_{a}$ is a good approximation to the true refractor velocity $v_{k}$; percentage errors are reasonably small $(+1.5 \%$, $+3.1 \%$, and $+5.2 \%$ for layers 2,3 , and 4, respectively). As Palmer (1980, p. 9) points out, an improved estimate of refractor velocity can be obtained if the $\operatorname{dip} \varphi_{k}$ of the critically refracting interface is known. This estimate is derived as follows. Assume that the dip angles of all interfaces overlying the critically refracting horizon vanish ( $\varphi_{i}=0$ for $i=1,2, \ldots, k-1$ ). Then, equation (5.13) becomes

$$
\frac{1}{V_{a}}=\frac{1}{v_{k} \cos \varphi_{k}}-\frac{\tan \varphi_{k}}{v_{k-1}}\left[\frac{\cos \alpha_{k-1, k}(0)+\cos \beta_{k-1, k}(0)}{2}\right]
$$

Geometric analysis reveals that the raypath angles within the layer immediately above interface $k$ are given by $\alpha_{k-1, k}(0)=\left|i_{k}-\varphi_{k}\right|$ and $\beta_{k-1, k}(0)=\pi-\left|i_{k}+\varphi_{k}\right|$, where $i_{k}=\sin ^{-1}\left(v_{k-1} / v_{k}\right)$ is the critical angle at the $k^{t h}$ interface. Substituting these relations into the above expression for $V_{a}$ and reducing yield

$$
\begin{equation*}
V_{a}=\frac{v_{k}}{\cos \varphi_{k}} \tag{5.15}
\end{equation*}
$$

Thus, under the stated assumptions, the apparent velocity exceeds the true refractor velocity. Equation (5.15) holds in an approximate sense if the dip angles of all interfaces overlying the critical horizon are small, rather than identically zero. A dip corrected apparent horizontal velocity can be defined as $V_{a}^{\prime} \equiv V_{a} \cos \varphi_{k}$. Using the above model parameters, the following values are obtained:

$$
k=2: \quad V_{a}^{\prime}=1800.00 \mathrm{~m} / \mathrm{s}, \quad k=3: \quad V_{a}^{\prime}=2568.77 \mathrm{~m} / \mathrm{s}, \quad k=4: \quad V_{a}^{\prime}=4066.21 \mathrm{~m} / \mathrm{s}
$$

Obviously, $V_{a}^{\prime}$ is an improved estimate of the true refractor velocity; percentage errors are $0 \%,+2.8 \%$, and $+1.7 \%$ for layers 2,3 , and 4 , respectively. Note that $V_{a}^{\prime}$ calculated for layer 2 is exactly correct, because the dip of the overlying interface (the surface) is zero. Interestingly, the error does not increase monotonically with layer depth.

As indicated previously, the condition for exact validity of equation (5.15) is the dip angles of all interfaces above the critical horizon vanish. Palmer (1980, p. 9) states that (5.15) is approximately correct if the differences in the dip angles between successive interfaces $\left(\varphi_{i+1}-\varphi_{i}\right)$ are negligible. However, this assertion is easily disproved. Expression (5.13) for the apparent horizontal velocity is evaluated for a multilayered model with velocities $v_{i}=1000+100 i \mathrm{~m} / \mathrm{s}$ and dips $\varphi_{i}=i^{\circ},(i=1,2, \ldots, 50)$. All differential dip angles equal $1^{\circ}$ and thus are small. A comparison between the apparent velocities and the true layer velocity is illustrated in Figure 5.2. In particular, both $V_{a}$ and $V_{a}^{\prime}$ depart appreciably from the correct velocity value as the layer index increases, i.e., as the dips of the bounding interfaces become large. This behaviour confirms that the approximation (5.15) requires small absolute dip angles, rather than small differential dip angles.


Fig. 5.2. Comparison of apparent velocity $V_{a}$ and dip-corrected apparent velocity $V_{a}^{\prime}$ with the true layer velocity $v_{k}$, for a multilayered earth model.

Another model which can be used to establish the veracity of (5.13) consists of a set of parallel, but dipping layers. Assume that all interfaces (including surface and critical refractor) have the same dip angle ( $\varphi_{i}=\varphi$, for $i=1,2, \ldots, k$ ). Obviously, the differential
dip angles are identically zero. Expression (5.13) immediately yields the expected result: $V_{a}=v_{k} \cos \varphi$. In this case, the apparent velocity is less than the true refractor velocity.

Finally, an alternate representation of the apparent horizontal velocity is derived from the slope/intercept formula for head wave traveltime:

$$
\begin{equation*}
T_{k}\left(x_{S}, X, \Psi\right)=m_{k}(\Psi) X+b_{k}\left(x_{S}\right) \tag{5.16}
\end{equation*}
$$

Explicit expressions for the slope $m_{k}(\Psi)$ and intercept $b_{k}\left(x_{S}\right)$ can be obtained by specializing equation (3.12) to a two dimensional earth model. Using this equation in the definition of the velocity analysis function yields

$$
\begin{equation*}
\frac{1}{V_{a}} \equiv \frac{\partial T_{v}}{\partial x_{R}}=\frac{m_{k}(0)+m_{k}(\pi)}{2} \tag{5.17}
\end{equation*}
$$

Thus, the apparent horizontal slowness $V_{a}^{-1}$ is merely the arithmetic mean of the slopes of the forward and reverse traveltime curves. This implies that $V_{a}$ is the harmonic mean of the velocities measured on the forward and reverse spreads. Defining $v_{f} \equiv 1 / m_{k}(0)$ and $v_{r} \equiv 1 / m_{k}(\pi)$, then

$$
\frac{1}{V_{a}}=\frac{1}{2}\left[\frac{1}{v_{f}}+\frac{1}{v_{r}}\right], \quad \text { or } \quad V_{a}=2 \frac{v_{f} v_{r}}{v_{f}+v_{r}}
$$

Lankston (1990) and Palmer (1990) state that the harmonic mean of $v_{f}$ and $v_{r}$ equals the true layer velocity. The present analysis reveals that claim to be erroneous. Rather, the harmonic mean $V_{a}$ is an approximation to the true velocity $v_{k}$.

### 5.3.3 Generalized time-depth

The generalized time-depth function for interface $k$ is defined as

$$
\begin{equation*}
T_{d}\left(x_{R} ; L\right) \equiv \frac{1}{2}\left[T_{k}\left(x_{S_{1}}, x_{R}+L / 2\right)+T_{k}\left(x_{S_{2}}, x_{R}-L / 2\right)-T_{k}\left(x_{S_{1}}, x_{S_{2}}\right)-\frac{L}{V_{a}}\right] \tag{5.18}
\end{equation*}
$$

Substituting from equation (5.7) for the head wave traveltimes gives

$$
\begin{gathered}
T_{d}\left(x_{R} ; L\right)=\sum_{i=1}^{k-1} \frac{h_{i}\left(x_{R}\right)}{v_{i}}\left[\frac{\cos \alpha_{i k}(0)-\cos \beta_{i k}(0)}{2}\right] \\
+\frac{L}{2}\left\{\frac{1}{v_{k} \cos \varphi_{k}}-\sum_{i=1}^{k-1} \frac{1}{v_{i}}\left[\frac{\sin \left(\varphi_{i+1}-\varphi_{i}\right)}{\cos \varphi_{i+1} \cos \varphi_{i}}\right] \frac{\cos \alpha_{i k}(0)+\cos \beta_{i k}(0)}{2}-\frac{1}{V_{a}}\right\} .
\end{gathered}
$$

However, from equation (5.13) for the apparent horizontal velocity, the term in braces is identically zero. The above expression reduces to

$$
\begin{equation*}
T_{d}\left(x_{R} ; L\right)=\sum_{i=1}^{k-1} \frac{h_{i}\left(x_{R}\right)}{v_{i}}\left[\frac{\cos \alpha_{i k}-\cos \beta_{i k}}{2}\right] \tag{5.19}
\end{equation*}
$$

where the explicit dependence of the ray angles on recording profile azimuth has been suppressed because, from relations (5.8a,b), the difference $\cos \alpha_{i k}(\Psi)-\cos \beta_{i k}(\Psi)$ is independent of azimuth. The time-depth of a critically refracting interface at coordinate $x_{R}$ is related to the vertical thicknesses of all overlying layers at the same position. Plotted time-depth sections give a qualitative indication of refractor structure, in a manner analogous to reflection time sections. Conversion of the time-depth to actual depth requires velocity information, and is briefly discussed in the next section.

The slope/intercept formula for head wave traveltime also yields an alternative expression for the time-depth function. Substituting equation (5.16) into the definition of the generalized time-depth and reducing gives

$$
T_{d}\left(x_{R} ; L\right)=\left[\frac{m_{k}(\pi)-m_{k}(0)}{2}\right]\left(x_{S_{2}}-x_{R}\right)+\frac{b_{k}\left(x_{S_{2}}\right)}{2}
$$

The reciprocal time (or shot-to-shot traveltime) is $T_{k}\left(x_{S_{1}}, x_{S_{2}}\right)=T_{k}\left(x_{S_{2}}, x_{S_{1}}\right)$. Thus, from the slope/intercept formula

$$
m_{k}(0)\left(x_{S_{2}}-x_{S_{1}}\right)+b_{k}\left(x_{S_{1}}\right)=m_{k}(\pi)\left(x_{S_{2}}-x_{S_{1}}\right)+b_{k}\left(x_{S_{2}}\right) .
$$

Hence

$$
m_{k}(\pi)-m_{k}(0)=\frac{b_{k}\left(x_{S_{1}}\right)-b_{k}\left(x_{S_{2}}\right)}{x_{S_{2}}-x_{S_{1}}}
$$

Finally, substituting this result into the above expression for the time-depth function yields

$$
\begin{equation*}
T_{d}\left(x_{R} ; L\right)=\frac{b_{k}\left(x_{S_{1}}\right)}{2}\left[\frac{x_{S_{2}}-x_{R}}{x_{S_{2}}-x_{S_{1}}}\right]+\frac{b_{k}\left(x_{S_{2}}\right)}{2}\left[\frac{x_{R}-x_{S_{1}}}{x_{S_{2}}-x_{S_{1}}}\right] \tag{5.20}
\end{equation*}
$$

The time-depth function at receiver coordinate $x_{R}$ is a linear interpolation of the halfintercept times at the two bounding source coordinates. Thus, for the chosen earth model, the GRM time-depth section is identical to the half-intercept time section used in classical refraction traveltime analysis (Wyrobek, 1956).

### 5.3.4 Depth conversion factor

Expression (5.19) for the generalized time-depth may be rewritten as

$$
\begin{equation*}
T_{d}\left(x_{R} ; L\right)=\sum_{i=1}^{k-1} \frac{h_{i}\left(x_{R}\right)}{v_{i k}}, \tag{5.21}
\end{equation*}
$$

where the depth conversion factor is defined as

$$
\begin{equation*}
v_{i k} \equiv \frac{2 v_{i}}{\cos \alpha_{i k}-\cos \beta_{i k}} \tag{5.22a}
\end{equation*}
$$

The depth conversion factor has physical dimension of velocity. Since the raypath angles $\alpha_{i k}$ and $\beta_{i k}$ are usually not determined in a GRM interpretation, $v_{i k}$ cannot be evaluated numerically. However, various approximations to the actual depth conversion factor are useful. If the cosines of the raypath angles are replaced with the expressions appropriate for horizontal layering $\left(\cos \alpha_{i k}=-\cos \beta_{i k}=\sqrt{1-\left(v_{i} / v_{k}\right)^{2}}\right.$ ), then equation (5.22a) is approximated by

$$
\begin{equation*}
\hat{v}_{i k}=\frac{v_{i} v_{k}}{\sqrt{v_{k}^{2}-v_{i}^{2}}} \tag{5.22b}
\end{equation*}
$$

Alternately, the true layer velocities in (5.22b) can be replaced by apparent horizontal velocities to obtain

$$
\begin{equation*}
\tilde{v}_{i k}=\frac{V_{a i} V_{a k}}{\sqrt{V_{a k}^{2}-V_{a i}^{2}}} \tag{5.22c}
\end{equation*}
$$

where $V_{a i}$ denotes the apparent horizontal velocity of layer $i$. Evaluating these expressions with the parameters defining the shallow four-layer model yields the following numerical values (in $\mathrm{m} / \mathrm{s}$ ):

| $i$ | $k$ | $v_{i k}$ | $\hat{v}_{i k}$ | $\tilde{v}_{i k}$ |
| :---: | :---: | :---: | :---: | :---: |
| 1 | 2 | 1221.23 | 1202.68 | 1194.66 |
| 1 | 3 | 1087.69 | 1091.09 | 1084.91 |
| 2 | 3 | 2603.66 | 2593.76 | 2591.17 |
|  |  |  |  |  |
| 1 | 4 | 1041.93 | 1032.80 | 1029.47 |
| 2 | 4 | 2013.08 | 2015.61 | 2029.00 |
| 3 | 4 | 3315.54 | 3202.56 | 3262.22 |

The approximations appear to be reasonable; the largest percentage error is $-3.4 \%$ (for $\hat{v}_{34}$ ).
The depth conversion factors are used to calculate layer thicknesses from the observed time-depths via expression (5.21). For $k=2$, this equation yields

$$
\begin{equation*}
h_{1}\left(x_{R}\right)=v_{12} T_{d}\left(x_{R} ; L\right) \tag{5.23a}
\end{equation*}
$$

while for $k>2$ :

$$
\begin{equation*}
h_{k-1}\left(x_{R}\right)=v_{k-1, k}\left[T_{d}\left(x_{R} ; L\right)-\sum_{i=1}^{k-2} \frac{h_{i}\left(x_{R}\right)}{v_{i k}}\right] \tag{5.23b}
\end{equation*}
$$

Vertical depths to the refracting interfaces are obtained simply by summing the layer thicknesses: $z_{k}\left(x_{R}\right)=z_{1}\left(x_{R}\right)+\sum_{i=1}^{k-1} h_{i}\left(x_{R}\right)$ for $k \geq 2$. Note that the depth $z_{k}\left(x_{R}\right)$ is an unambiguously defined function of the receiver coordinate $x_{R}$. In contrast, the 'depth' calculated via Palmer's (1980) time-depth function is a radius of a circular arc centered at the receiver position.

Continuing with the illustrative example, numerical values of the time-depth for the three subsurface interfaces can be obtained by evaluating expression (5.20). Sources are located at the horizontal coordinates $x_{S_{1}}=-100 \mathrm{~m}$ and $x_{S_{2}}=+60 \mathrm{~m}$, and the receiver is positioned at the coordinate origin. The results are

$$
k=2: T_{d}(0 ; L)=16.63 \mathrm{~ms}, \quad k=3: T_{d}(0 ; L)=33.74 \mathrm{~ms}, \quad k=4: T_{d}(0 ; L)=45.23 \mathrm{~ms}
$$

Finally, values of interface depths at $x_{R}=0$ are calculated using the exact and approximate depth conversion factors. The results are (in m):

| $i$ | $z_{i}(0)$ | $\hat{z}_{i}(0)$ | $\tilde{z}_{i}(0)$ |
| :---: | :---: | :---: | :---: |
| 2 | 20.31 | $20.00(-1.5 \%)$ | $19.87(-2.2 \%)$ |
| 3 | 59.55 | $59.98(+0.7 \%)$ | $59.85(+0.5 \%)$ |
| 4 | 80.25 | $79.29(-1.2 \%)$ | $80.15(-0.1 \%)$ |

The associated percentage errors indicate that, for this example, an adequate result is obtained by using the approximate depth conversion factors.

### 5.3.5 Time-depth near a shotpoint

The intercept on the time axis of the velocity analysis function (5.11) has a simple, but nevertheless useful, interpretation. From formula (5.3c), the vertical thickness of the $i^{\text {th }}$ layer at the coordinate $x_{S_{1}}+L / 2$ can be written as

$$
h_{i}\left(x_{S_{1}}+L / 2\right)=h_{i}\left(x_{S_{1}}\right)+\frac{L}{2}\left[\frac{\sin \left(\varphi_{i+1}-\varphi_{i}\right)}{\cos \varphi_{i+1} \cos \varphi_{i}}\right] .
$$

Substituting this result into expression (5.11) for the velocity analysis function yields

$$
T_{v}\left(x_{R} ; L\right)=\sum_{i=1}^{k-1} \frac{h_{i}\left(x_{S_{1}}+L / 2\right)}{v_{i}}\left[\frac{\cos \alpha_{i k}-\cos \beta_{i k}}{2}\right]+\frac{x_{R}-x_{S_{1}}}{V_{a}}
$$

However, from equation (5.19), the summation term is recognized as the generalized timedepth at the coordinate $x_{S_{1}}+L / 2$. Thus

$$
\begin{equation*}
T_{v}\left(x_{R} ; L\right)=T_{d}\left(x_{S_{1}}+L / 2 ; L\right)+\frac{x_{R}-x_{S_{1}}}{V_{a}} \tag{5.24}
\end{equation*}
$$

The time axis intercept of the linear velocity analysis function equals the interface timedepth at a point offset a distance $L / 2$ from source 1 . This interesting relationship provides a tool for estimating refractor depths near a source, where head waves are either nonexistent or difficult to pick accurately.

### 5.4 Critical offset refraction profiling

### 5.4.1 Earth model

The two dimensional earth model considered in this section is illustrated in Figure 5.3. A single layer with uniform velocity $v_{1}$ is underlain by a medium where velocity $v_{2}(x)$ depends solely on the horizontal coordinate. Surface and subsurface horizons are undulating and are defined by the depth functions $z_{1}(x)$ and $z_{2}(x)$, respectively. The dip angles of these two interfaces are given by

$$
\varphi_{1}(x)=\tan ^{-1}\left[\frac{d z_{1}(x)}{d x}\right], \quad \quad \varphi_{2}(x)=\tan ^{-1}\left[\frac{d z_{2}(x)}{d x}\right]
$$

These relations define the sign convention associated with interface dip.
Forward and reverse traveltime data are recorded by receivers distributed along the surface. The observed traveltime curves are designated $T_{f}(x)$ and $T_{r}(x)$, respectively. The slopes of the two traveltime curves are

$$
m_{f}(x) \equiv \frac{d T_{f}(x)}{d x}, \quad \quad m_{r}(x) \equiv-\frac{d T_{r}(x)}{d x}
$$

Note that a negative sign is included in the definition of $m_{\tau}(x)$. From the measured slopes of the traveltime curves, the angles of incidence of rays arriving at the surface can be calculated. Let the functions $\mu_{f}(x)$ and $\mu_{r}(x)$ refer to the incidence angles of the forward profile and reverse profile raypaths, respectively. Then, straightforward geometric analysis yields the expressions


Fig. 5.3. A 2D earth model with undulating surface and refractor topography. Forward and reverse raypaths exiting from the same point $C$ on the critical horizon intersect the surface at $B$ and $A$, respectively. $x_{M}$ is the midpoint coordinate of $A$ and $B$.

$$
\begin{aligned}
\mu_{f}(x) & =\sin ^{-1}\left[v_{1} \cos \varphi_{1}(x) m_{f}(x)\right]+\varphi_{1}(x) \\
\mu_{r}(x) & =\sin ^{-1}\left[v_{1} \cos \varphi_{1}(x) m_{r}(x)\right]-\varphi_{1}(x)
\end{aligned}
$$

$\mu_{f}(x)$ and $\mu_{r}(x)$ are acute angles measured with respect to the vertical. The previously established sign conventions for surface dip and traveltime slope determine the positive sense of these two incidence angles. In particular, $\mu_{f}(x)$ is positive if the angle opens in the clockwise sense. The converse holds for $\mu_{r}(x)$; it is positive if the angle opens in the counterclockwise sense (see Figure 5.4).

The measured traveltime data pertain to waves that are refracted at the subsurface horizon $z_{2}(x)$. In general, the raypaths of these waves may undergo noncritical refractions at the interface. However, the inversion procedure discussed here adopts the following fundamental assumption: the critically refracted portion of the total raypath is coincident with the undulating subsurface interface. Although this is an approximation, it is sufficiently accurate if the structure on the interface is not too severe. In the presence of synclinal structure, the approximation is reasonable because there is a tendency for the diffracted raypath to follow the interface (see, for example, Figure 7.1). Since raypaths penetrate beneath anticlinal structures, the approximation is less accurate in these cases. Nevertheless, refraction seismologists commonly adopt this assumption as a useful working hypothesis. In fact, it underlies the popular interpretive rule known as the 'law of parallelism' of refraction traveltime curves (Rockwell, 1967; Sjögren, 1980; Ackermann et al., 1986; Brückl, 1987).

Figure 5.3 depicts forward and reverse raypaths exiting from the same point $C$ on the subsurface interface. If $i_{2}(x) \equiv \sin ^{-1}\left[v_{1} / v_{2}(x)\right]$ is a variable critical angle defined along interface 2 , then these rays depart at an angle $i_{2}\left(x_{C}\right)$ with respect to the normal to $z_{2}(x)$. Points of intersection of the reverse and forward rays with the surface $z_{1}(x)$ are designated $A$ and $B$, respectively. These two points are separated by the critical offset distance, i.e., the minimum horizontal distance between a surface source and receiver at which a critical refraction can be observed. Obviously, the critical offset varies along the line of profile due to the changing depth and dip of the two interfaces, as well as the the varying velocity $v_{2}(x)$ of the lower medium. If $x_{M}$ denotes midpoint coordinate of $A$ and $B$, then a critical offset function $L_{c}\left(x_{M}\right)$ can be defined at every midpoint position on the surface. The inversion technique outlined below is based on the fact that an exact solution to the refraction problem is possible if this function is known. A procedure for estimating the critical offset function from the measured traveltime data is discussed in a subsequent section.

### 5.4.2 Inversion method

If the two points $A$ and $B$ on the surface are separated by the critical offset distance, then a closed form solution for the subsurface model parameters can be obtained. Figure
5.4 indicates that

$$
\tan \mu_{f}\left(x_{B}\right)=\frac{x_{B}-x_{C}}{z_{C}-z_{B}}, \quad \tan \mu_{r}\left(x_{A}\right)=\frac{x_{C}-x_{A}}{z_{C}-z_{A}}
$$

Solving these equations for the two unknown coordinates ( $x_{C}, z_{C}$ ) yields

$$
\begin{array}{r}
x_{C}=\frac{x_{B} \cos \mu_{f}\left(x_{B}\right) \sin \mu_{r}\left(x_{A}\right)+x_{A} \sin \mu_{f}\left(x_{B}\right) \cos \mu_{r}\left(x_{A}\right)}{\sin \left(\mu_{f}\left(x_{B}\right)+\mu_{r}\left(x_{A}\right)\right)} \\
+\frac{\left(z_{B}-z_{A}\right) \sin \mu_{f}\left(x_{B}\right) \sin \mu_{r}\left(x_{A}\right)}{\sin \left(\mu_{f}\left(x_{B}\right)+\mu_{r}\left(x_{A}\right)\right)}, \\
z_{C}=\frac{z_{B} \sin \mu_{f}\left(x_{B}\right) \cos \mu_{r}\left(x_{A}\right)+z_{A} \cos \mu_{f}\left(x_{B}\right) \sin \mu_{r}\left(x_{A}\right)}{\sin \left(\mu_{f}\left(x_{B}\right)+\mu_{r}\left(x_{A}\right)\right)} \\
 \tag{5.25b}\\
+\frac{\left(x_{B}-x_{A}\right) \cos \mu_{f}\left(x_{B}\right) \cos \mu_{r}\left(x_{A}\right)}{\sin \left(\mu_{f}\left(x_{B}\right)+\mu_{r}\left(x_{A}\right)\right)}
\end{array}
$$

These relations establish the position of the point of critical refraction $C$ in terms of known quantities. Furthermore, since the sum of the angles of triangle $A B C$ equals $\pi$ radians, the critical angle at $C$ is given by

$$
\begin{equation*}
i_{2}\left(x_{C}\right)=\frac{\mu_{f}\left(x_{B}\right)+\mu_{r}\left(x_{A}\right)}{2} \tag{5.26}
\end{equation*}
$$

Hence, the refractor velocity at point $C$ is determined by $v_{2}\left(x_{C}\right)=v_{1} / \sin i_{2}\left(x_{C}\right)$. Finally, the local dip of the refracting interface is calculated by analysis of either triangle $A C D$ or triangle $B C E$ in Figure 5.4. The result is

$$
\begin{equation*}
\varphi_{2}\left(x_{C}\right)=\frac{\mu_{f}\left(x_{B}\right)-\mu_{r}\left(x_{A}\right)}{2} \tag{5.27}
\end{equation*}
$$



Fig. 5.4. An expanded view of triangle $A B C$ in Fig. 5.3. Angles $\mu_{f}$ and $\mu_{r}$ describe the orientation of raypath segments $\overline{C B}$ and $\overline{C A}$ relative to vertical, respectively. The critical and dip angles at point $C$ on the refractor are $i_{2}$ and $\varphi_{2}$, respectively.

Expressions (5.25) through (5.27) constitute a mathematically exact solution of the specified seismic refraction problem. The formulae are not restricted to the particular geometric situation depicted in Figure 5.4, but remain valid if the refracting point $C$ does not reside between the surface points $A$ and $B$ (i.e., if either $\mu_{f}\left(x_{B}\right)<0$ or $\mu_{\tau}\left(x_{A}\right)<0$ ).

The above inversion formulae for critical angle and dip are similar to the classical refraction solution for an earth model with plane interfaces and uniform velocities. In that case, the forward and reverse traveltime curves are straight lines with slopes that are independent of the horizontal coordinate $x$. Hence, the forward and reverse angles of incidence are also constant: $\mu_{f}(x)=\mu_{f}$ and $\mu_{r}(x)=\mu_{r}$. Equations (5.26) and (5.27) reduce to the familiar expressions $i_{2}=\left(\mu_{f}+\mu_{r}\right) / 2$ and $\varphi_{2}=\left(\mu_{f}-\mu_{r}\right) / 2$. There are no analogues to
equations (5.25) in the classical solution, because vertical depths to the refracting interface are determined only at the source locations.

A set of point estimates of the position of the critically refracting horizon is obtained by repeated application of formulae ( $5.25 \mathrm{a}, \mathrm{b}$ ) at various midpoint locations along the line. A profile of the interface can then be constructed by interpolating these points. Additional constraints on the interpolant are provided by the interface dip estimates at each refracting point $C$. Assuming an error-free inversion, the constructed depth profile should pass through the coordinates $\left(x_{C}, z_{C}\right)$ and have slope $\tan \varphi_{2}\left(x_{C}\right)$ at this point. A procedure for constructing a smooth interpolation of a set of point estimates of depth and dip is presented in Chapter 6.

### 5.4.3. Critical offset determination

The inversion method outlined above requires the critical offset distance $L_{c}\left(x_{M}\right)$ to be known at each analysis midpoint along the profile line. Jones and Jovanovich (1985) describe a computational technique for estimating critical offsets at the source positions on a line. This section discusses an extension of their method to arbitrary locations along a reversed refraction profile. Moreover, an analytical, rather than computational solution to the problem is given. Consider two surface locations with a common midpoint coordinate $x_{M}$ and separated by the horizontal distance $L$. A candidate critical refraction raypath can be constructed from these two points by projecting the incoming raypath segments downward until they intersect (Figure 5.5). The projection directions are determined from the known incidence angles $\mu_{f}\left(x_{M}+L / 2\right)$ and $\mu_{r}\left(x_{M}-L / 2\right)$. The propagation time along this hypothetical raypath is called a 'predicted' traveltime and can be derived via geometric analysis. It is


Fig. 5.5. Determination of the critical offset distance $L_{c}\left(x_{M}\right)$ at the midpoint coordinate $x_{M}$. Raypath segments projected downward from $A$ and $B$ on the surface intersect at point $C$ on the refracting interface.

$$
\begin{gather*}
T_{p r d}\left(x_{M} ; L\right)=\frac{L}{v_{1}}\left[\frac{\cos \mu_{f}\left(x_{M}+L / 2\right)+\cos \mu_{r}\left(x_{M}-L / 2\right)}{\sin \left(\mu_{f}\left(x_{M}+L / 2\right)+\mu_{r}\left(x_{M}-L / 2\right)\right)}\right] \\
+\frac{z_{1}\left(x_{M}+L / 2\right)-z_{1}\left(x_{M}-L / 2\right)}{v_{1}}\left[\frac{\sin \mu_{f}\left(x_{M}+L / 2\right)-\sin \mu_{r}\left(x_{M}-L / 2\right)}{\sin \left(\mu_{f}\left(x_{M}+L / 2\right)+\mu_{r}\left(x_{M}-L / 2\right)\right)}\right] . \tag{5.28}
\end{gather*}
$$

Note that this expression simplifies considerably if the surface is horizontal (i.e., $z_{1}(x)=$ $z_{1}(0)=$ constant $)$. At the same two surface locations, an 'observed' traveltime is determined from the measured forward and reverse traveltime curves as follows:

$$
\begin{equation*}
T_{o b s}\left(x_{M} ; L\right)=T_{f}\left(x_{M}+L / 2\right)+T_{r}\left(x_{M}-L / 2\right)-T_{R} \tag{5.29}
\end{equation*}
$$

where $T_{R}$ is the reciprocal time: $T_{R}=T_{f}\left(x_{S_{2}}\right)=T_{r}\left(x_{S_{1}}\right)$. For a fixed midpoint coordinate $x_{M}$, expressions (5.28) and (5.29) are evaluated over a range of values of the separation distance $L$, extending from well below to well above the expected critical offset distance. Figure 5.4 indicates that when $T_{p r d}\left(x_{M} ; L\right)=T_{o b s}\left(x_{M} ; L\right)$, then $L=L_{c}\left(x_{M}\right)$. The common time calculated at this separation distance is the critical offset traveltime $T_{c}\left(x_{M}\right)$ associated with the midpoint position. Application of this analysis technique at various midpoint locations along the profile line yields several point estimates of the critical offset function $L_{c}(x)$.

### 5.5 Conclusion

The theoretical foundation of the generalized reciprocal method of refraction traveltime inversion has been reformulated in terms of layers characterized by vertical, rather than normal, thicknesses. This reformulation results in a simplification of the derivations of the two GRM analysis functions, and thus has educational value. However, it also allows point estimates of interface depth to be calculated from the measured traveltimes. A refractor depth profile can then be calculated by interpolating the depth points. Although the GRM is based on a simple 2D model with plane interfaces, interpreters apply the technique in more realistic environments with nonplane refracting horizons. The results that are obtained should then be viewed as approximate.

In contrast, the proposed method of critical offset refraction profiling explicitly incorporates undulating interfaces and variable refractor velocities into the model. A mathematically exact solution to the traveltime inverse problem indicates that point estimates of interface depth, interface dip, and refractor velocity can all be obtained from the observed arrival times. These point estimates are positioned properly in space; there is no need for a subsequent migration of the solution. In common with many other refraction interpretation methods, independent knowledge of the overburden velocity $v_{1}$ is required. Although the derivations in this chapter assume a uniform layer velocity, this is not a practical necessity. Rather, it is only necessary that $v_{1}$ be reasonably constant over the extent of the triangle $A B C$ in Figure 5.3. Lateral variations with a characteristic scale greater than the horizontal
dimension of the triangle are allowed. The primary approximation adopted by this technique relates to the refracted raypath: it remains coincident with the undulating interface. The implications of this approximation for the inversion should be investigated more thoroughly. Finally, extension of the method to multilayered earth models represents another avenue of developement. Using the present algorithm, this could be achieved via a layer stripping approach.

## CHAPTER 6

## CONSTRUCTION OF A SMOOTH REFRACTOR DEPTH PROFILE

### 6.1 Introduction

Linear inverse theory provides techniques for constructing acceptable models that are consistent with the available data. However, if acceptability is judged by the data misfit alone, then there are usually infinitely many equally valid solutions. A specific solution can be obtained only by extremizing some functional of the model, subject to a requirement that the final model is in satisfactory agreement with the observations. Since different functionals can produce models with significantly different character, an essential step in the solution of any inverse problem is to select the 'right' functional. Minimization of the $l_{2}$ norm of the model yields a 'smallest' model that often has high variability and hence may be difficult to interpret. Similarly, models can be constructed by minimizing the $l_{2}$ norm of the first or second derivative; these are often referred to as 'flattest' or 'smoothest' models, respectively. These latter types are particularly useful because they can be interpreted as minimum structure solutions for many geophysical inverse problems. It is anticipated that the true earth model that gives rise to the measured data has at least as much structure as observed on these constructed models.

Methods for calculating minimum structure solutions have been described by various authors. These methods differ in the manner in which the required additional information regarding the model is treated. Johnson and Gilbert (1972) construct a smoothest model by explicitly incorporating endpoint values of the model and its derivative into the objective functional to be minimized. Oldenburg (1984), in computing a flattest model, specifies an a priori model value at an endpoint. Parker, Shure, and Hildebrand (1987) minimize a seminorm of the model. This chapter discusses two generalizations to the conventional method for constructing a smoothest model. In this technique, the governing equations are integrated by parts twice; each such integration introduces extra parameters into the model construction problem. Numerical values for these additional parameters are obtained either by (i) prescribing a priori values at any abscissae within the interval of model definition,
or (ii) optimizing the appropriate objective function with respect to the parameters. These generalizations are readily extended to model construction problems that involve minimizing the $l_{2}$ norm of other derivatives of the model.

In this chapter, the smoothest model construction formalism is applied to the problem of determining a depth profile for a critically refracting horizon. It has only recently been demonstrated that refraction traveltime data can be processed to yield point estimates of interface depth (Jones and Jovanovich, 1986; Pavlenkin et al., 1986; Ali Ak, 1990). Moreover, as shown in the previous chapter, local dip estimates can also be derived from the traveltimes. A refractor depth profile can then be calculated by smoothly interpolating all of these data. This procedure provides a useful alternative to the popular time-to-depth conversion method associated with the GRM (Hatherly, 1980; Hatherly and Neville, 1986; Kilty et al., 1986; Lankston and Lankston, 1986; Lankston, 1989; Palmer, 1990, 1991). In the GRM technique, circular arcs with radii given by expression (5.23a) are centered at each analysis location along the seismic line. The position of the refracting interface is then estimated by the envelope of the set of arcs. This envelope is approximated by tangent lines to pairs of adjacent arcs; problems specific to this method are identified and discussed by Hatherly (1980). The technique has antecedents in the depth determination procedures of Thornburgh (1930), Dix (1941), Tarrant (1956), Hales (1958), and Hawkins (1961) where possible refractor loci, rather then point depth estimates, are calculated from the traveltime data.

The interpolation method described in this chapter circumvents those problems related to calculating an envelope of a set of circular arcs. It also possesses several specific advantages regarding treatment of the data. However, perhaps the most important difference pertains directly to the derived models: the method proposed here yields the smoothest refractor depth profile consistent with the given data. This attribute is not shared by the model constructed via the GRM technique.

A general theoretical developement of the smoothest model construction method is presented before discussing the particular application to interpolation. Hence, the results in this chapter are useful for the solution of other geophysical inverse problems that may require a
smooth model. Assume that an $N$-vector of observed data $\mathbf{e}_{o b s}$ is related to the true earth model $m_{t}(x)$ via a Fredholm integral equation of the first kind:

$$
\begin{equation*}
\mathbf{e}_{o b s}=\int_{a}^{b} \mathbf{g}(x) m_{t}(x) d x+\delta \mathbf{e} \tag{6.1}
\end{equation*}
$$

$\mathbf{g}(x)$ is a vector of known kernel functions and $\delta \mathbf{e}$ is a vector of additive random errors. The true earth model $m_{t}(x)$ is unknown and is to be estimated. Predicted data generated by a constructed earth model $m(x)$ are given by the linear functional

$$
\begin{equation*}
\mathbf{e}_{p r d}(m)=\int_{a}^{b} \mathbf{g}(x) m(x) d x \tag{6.2}
\end{equation*}
$$

In the following development, the $l_{2}$ norm is used to quantitatively measure both model structure and data misfit. For example, the square of the $l_{2}$ norm of the second derivative of the model is

$$
\left\|m^{\prime \prime}\right\|^{2}=\int_{a}^{b} m^{\prime \prime}(x)^{2} d x
$$

Similarly, the square of the $l_{2}$ norm of the misfit between observed and predicted data is given by

$$
\left\|\mathbf{e}_{o b s}-\mathbf{e}_{p r d}(m)\right\|^{2}=\left[\mathbf{e}_{o b s}-\mathbf{e}_{p r d}(m)\right]^{T}\left[\mathbf{e}_{o b s}-\mathbf{e}_{p r d}(m)\right]
$$

where the superscript $T$ indicates transposition.

### 6.2 Smoothest model construction

A detailed discussion of the method for calculating the flattest model is given by Aldridge et al. (1991). The analogous derivation for the smoothest model is conceptually similiar, although algebraically more complicated.

### 6.2.1 Modified data equation

Initially, equation (6.2) is integrated by parts to obtain a modified data equation satisfied by the first derivative of the constructed model:

$$
\begin{equation*}
\mathbf{e}_{p r d}(m)=m(b) \mathbf{h}(b)-\int_{a}^{b} \mathbf{h}(x) m^{\prime}(x) d x \tag{6.3}
\end{equation*}
$$

where $\mathbf{h}(x)$ is the indefinite integral of the original kernel function vector $\mathbf{g}(x)$ :

$$
\begin{equation*}
\mathbf{h}(x)=\int_{a}^{x} \mathbf{g}(u) d u \tag{6.4}
\end{equation*}
$$

The value of the model at the endpoint $b$ is required to evaluate the right-hand side of equation (6.3). However, the expression is easily rewritten in terms of a model value $m\left(c_{1}\right)$ specified at an arbitrary abscissa $c_{1}$ within the closed interval $[a, b]$. From the fundamental theorem of calculus

$$
\begin{equation*}
m(b)-m\left(c_{1}\right)=\int_{c_{1}}^{b} m^{\prime}(x) d x=\int_{a}^{b} H\left(x-c_{1}\right) m^{\prime}(x) d x \tag{6.5}
\end{equation*}
$$

where $H(x)$ is the Heaviside unit step function. Eliminating $m(b)$ from equations (6.3) and (6.5) yields

$$
\begin{equation*}
\mathbf{e}_{p r d}(m)=m\left(c_{1}\right) \mathbf{h}(b)-\int_{a}^{b}\left[\mathbf{h}(x)-\mathbf{h}(b) H\left(x-c_{1}\right)\right] m^{\prime}(x) d x \tag{6.6}
\end{equation*}
$$

In the model construction problem, $m\left(c_{1}\right)$ is considered to be a parameter independent of the abscissa $c_{1}$. This independence is emphasized by writing $d_{1}$ for $m\left(c_{1}\right)$. Equation (6.6) becomes

$$
\begin{equation*}
\mathbf{e}_{p r d}\left(m^{\prime}\right)=d_{1} \mathbf{h}(b)-\int_{a}^{b} \mathbf{p}\left(x ; c_{1}\right) m^{\prime}(x) d x \tag{6.7}
\end{equation*}
$$

where a new kernel function vector is defined by

$$
\begin{equation*}
\mathbf{p}\left(x ; c_{1}\right) \equiv \mathbf{h}(x)-\mathbf{h}(b) H\left(x-c_{1}\right) \tag{6.8}
\end{equation*}
$$

Note that the predicted data are now considered to be a functional of the first derivative $m^{\prime}$, rather than the model $m$ itself. Furthermore, due to the presence of the term $d_{1} h(b)$, these data are a nonlinear functional of $m^{\prime}$.

If the above procedure is repeated, then a modified data equation satisfied by the second derivative of the constructed model can be derived. Thus, expression (6.7) is integrated by parts and the fundamental theorem is used to transfer endpoint information to another arbitrary location $c_{2}$ with the interval $[a, b]$. The result is

$$
\begin{equation*}
\mathbf{e}_{p r d}\left(m^{\prime \prime}\right)=d_{1} \dot{\mathbf{h}}(b)-d_{2}\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right]+\int_{a}^{b} \mathbf{p}\left(x ; c_{1}, c_{2}\right) m^{\prime \prime}(x) d x \tag{6.9}
\end{equation*}
$$

where the kernel function vector is defined by

$$
\begin{equation*}
\mathbf{p}\left(x ; c_{1}, c_{2}\right) \equiv \mathbf{k}(x)-\mathbf{h}(b) R\left(x-c_{1}\right)-\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right] H\left(x-c_{2}\right) \tag{6.10}
\end{equation*}
$$

$c_{1}$ and $c_{2}$ are two abscissae in $[a, b]$ where the parameters $d_{1} \equiv m\left(c_{1}\right)$ and $d_{2} \equiv m^{\prime}\left(c_{2}\right)$ are defined. Also, $\mathbf{k}(x)$ is the indefinite integral of $h(x)$ :

$$
\begin{equation*}
\mathbf{k}(x)=\int_{a}^{x} \mathbf{h}(u) d u \tag{6.11}
\end{equation*}
$$

Finally, $R(x)$ in equation (6.10) is the ramp function with unit slope: $R(x)=x H(x)$.
Expression (6.9) is the desired result. It indicates that the predicted data are a nonlinear functional of the second derivative of the constructed model. In the particular case where both $c_{1}$ and $c_{2}$ are restricted to the upper endpoint $b$, formulae (6.9) and (6.10) reduce to familiar forms.

### 6.2.2 Objective function

In order to construct a smooth model that simultaneously minimizes the misfit between observed and predicted data, consider the objective functional

$$
\begin{equation*}
\Phi\left(m^{\prime \prime}\right)=\mu\left\|w m^{\prime \prime}\right\|^{2}+\left\|\mathbf{W}\left[\mathbf{e}_{o b s}-\mathbf{e}_{p r d}\left(m^{\prime \prime}\right)\right]\right\|^{2} \tag{6.12}
\end{equation*}
$$

The scalar $\mu(0 \leq \mu<+\infty)$ is a tradeoff parameter that controls the relative importance of the two terms. Equation (6.12) also includes a model derivative weighting function $w(x)$ and a data weighting matrix $W$. The function $w(x)$ can be used to emphasize certain portions of $m^{\prime \prime}(x)$ during subsequent minimization of $\Phi$. Similarly, the matrix $W$ allows a point-bypoint weighting of the input data according to prescribed criteria. Commonly, $\mathrm{W}^{2}$ is taken to be the inverse of the covariance matrix of the observational error $\delta$. However, the matrix in (6.12) is not restricted to this specific type of weighting.

Appendix $D$ demonstrates that the second derivative function that extremizes $\Phi$ must be a linear combination of the weighted kernels $p_{j}\left(x ; c_{1}, c_{2}\right) / w(x)^{2}, j=1,2, \ldots, N$ :

$$
\begin{equation*}
m^{\prime \prime}(x)=\alpha^{T} \frac{\mathbf{p}\left(x ; c_{1}, c_{2}\right)}{w(x)^{2}} \tag{6.13}
\end{equation*}
$$

where $\alpha$ is an $N$-vector of coefficients. Using this representation for $m^{\prime \prime}(x)$, it is straightforward to demonstrate that

$$
\begin{equation*}
\left\|w m^{\prime \prime}\right\|^{2}=\alpha^{T} \Gamma\left(c_{1}, c_{2}\right) \alpha \tag{6.14}
\end{equation*}
$$

and

$$
\begin{equation*}
\mathbf{e}_{p r d}=d_{1} \mathbf{h}(b)-d_{2}\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right]+\Gamma\left(c_{1}, c_{2}\right) \alpha \tag{6.15}
\end{equation*}
$$

where $\Gamma\left(c_{1}, c_{2}\right)$ is an inner product matrix formed from the kernel function vector $\mathbf{p}\left(x ; c_{1}, c_{2}\right)$ :

$$
\begin{equation*}
\Gamma\left(c_{1}, c_{2}\right) \equiv \int_{a}^{b} w(x)^{-2} \mathbf{p}\left(x ; c_{1}, c_{2}\right) \mathbf{p}\left(x ; c_{1}, c_{2}\right)^{T} d x \tag{6.16}
\end{equation*}
$$

The matrix $\Gamma\left(c_{1}, c_{2}\right)$ is symmetric and, if the original kernel functions $g_{j}(x)$ are linearly independent on $[a, b]$, positive definite. Substituting expressions (6.14) and (6.15) into (6.12) converts the objective functional $\Phi$ into a quadratic form in $\boldsymbol{\alpha}, d_{1}$, and $d_{2}$ :

$$
\begin{gather*}
\Phi\left(\boldsymbol{\alpha}, c_{1}, c_{2}, d_{1}, d_{2}\right)=\boldsymbol{\alpha}^{T} \Gamma\left(c_{1}, c_{2}\right)\left[\mathbf{W}^{T} \mathbf{W} \Gamma\left(c_{1}, c_{2}\right)+\mu \mathbf{I}\right] \boldsymbol{\alpha} \\
+2\left[d_{1} \mathbf{h}(b)-d_{2}\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right]-\mathbf{e}_{o b s}\right]^{T} \mathbf{W}^{T} \mathbf{W} \Gamma\left(c_{1}, c_{2}\right) \boldsymbol{\alpha} \\
+d_{1}^{2} \mathbf{h}(b)^{T} \mathbf{W}^{T} \mathbf{W} \mathbf{h}(b)-2 d_{1} d_{2} \mathbf{h}(b)^{T} \mathbf{W}^{T} \mathbf{W}\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right] \\
+d_{2}^{2}\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right]^{T} \mathbf{W}^{T} \mathbf{W}\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right] \\
-2 d_{1} \mathbf{h}(b)^{T} \mathbf{W}^{T} \mathbf{W} \mathbf{e}_{o b s}+2 d_{2}\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right]^{T} \mathbf{W}^{T} \mathbf{W} \mathbf{e}_{o b s}+\mathbf{e}_{o b s}^{T} \mathbf{W}^{T} \mathbf{W} \mathbf{e}_{o b s} . \tag{6.17}
\end{gather*}
$$

Dependence on the two abscissae $c_{1}$ and $c_{2}$ is via the inner product matrix $\Gamma\left(c_{1}, c_{2}\right)$ (as well as explicit dependence on $c_{1}$ ). Note that expression (6.17) is a complete quadratic form in the independent variables $\alpha, d_{1}$, and $d_{2}$ in the sense that all possible terms are present. In contrast, the analogous quadratic objective function given by Johnson and Gilbert (1972) lacks the cross product term in $d_{1}$ and $d_{2}$, the linear terms in both $d_{1}$ and $d_{2}$, and the constant term. Additionally, the remaining terms differ in detail from those given above.

### 6.2.3 Extremizing the objective function

Straightforward methods of multivariable calculus are now used to extremize the objective function $\Phi$ with respect to all of the variables relevant to the problem. Thus, calculating $\partial \Phi / \partial \boldsymbol{\alpha}$ and setting the result equal to 0 yields the linear system

$$
\begin{equation*}
\left[\Gamma\left(c_{1}, c_{2}\right)+\mu\left(\mathbf{W}^{T} \mathbf{W}\right)^{-1}\right] \boldsymbol{\alpha}=\mathbf{e}_{o b s}-d_{1} \mathbf{h}(b)+d_{2}\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right] \tag{6.18}
\end{equation*}
$$

If values for the four scalars $\left(c_{1}, d_{1}, c_{2}, d_{2}\right)$ are prescribed, then (6.18) can be solved for the coefficient vector $\alpha$. This offers an interesting alternative to the conventional method for constructing a smoothest model, where $c_{1}$ and $c_{2}$ are typically restricted to one of the endpoints $a$ or $b$. However, for the current analysis, optimum values of the four parameters are desired. Thus, extremizing $\Phi$ with respect to $d_{1}$ and $d_{2}$ yields the two additional expressions

$$
\begin{gathered}
{\left[\mathbf{W}^{T} \mathbf{W} \mathbf{h}(b)\right]^{T}\left[d_{1} \mathbf{h}(b)-d_{2}\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right]-\mathbf{e}_{o b s}+\Gamma\left(c_{1}, c_{2}\right) \boldsymbol{\alpha}\right]=0} \\
{\left[\mathbf{W}^{T} \mathbf{W}\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right]\right]^{T}\left[d_{1} \mathbf{h}(b)-d_{2}\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right]-\mathbf{e}_{o b s}+\Gamma\left(c_{1}, c_{2}\right) \boldsymbol{\alpha}\right]=0 .}
\end{gathered}
$$

Substituting from equation (6.18) immediately reduces these expressions to the simpler forms

$$
\begin{align*}
& \alpha^{T} \mathbf{h}(b)=0  \tag{6.19}\\
& \alpha^{T} \mathbf{k}(b)=0 \tag{6.20}
\end{align*}
$$

Geometrically, these conditions imply that the $N$-dimensional coefficient vector $\boldsymbol{\alpha}$ must be orthogonal to the 2 -dimensional subspace spanned by $\mathbf{h}(b)$ and $\mathbf{k}(b)$.

Finally, the objective function $\Phi$ must be extremized with respect to the two abscissae $c_{1}$ and $c_{2}$. However, Appendix E demonstrates the remarkable result that the derivatives $\partial \Phi / \partial c_{1}$ and $\partial \Phi / \partial c_{2}$ both vanish if the prior conditions (6.18), (6.19), and (6.20) hold. Thus, these three equations are sufficient to extremize $\Phi$ with respect to the five quantities ( $\alpha, c_{1}, d_{1}, c_{2}, d_{2}$ ). As long as these conditions are imposed, any convenient pair of abscissae $c_{1}$ and $c_{2}$ can be chosen to evaluate the inner product matrix $\Gamma\left(c_{1}, c_{2}\right)$. Expressions (6.18) through (6.20) are then a system of $N+2$ equations in the $N+2$ unknowns $\alpha_{j}, d_{1}$, and $d_{2}$, and can be solved simultaneously with standard techniques of linear algebra. However,
additional insight is obtained by first eliminating the coefficient vector $\boldsymbol{\alpha}$ to form a simple $2 \times 2$ system for $d_{1}$ and $d_{2}$ :

$$
\begin{equation*}
\mathbf{A d}=\mathbf{b} \tag{6.21}
\end{equation*}
$$

In this expression, $\mathbf{d}=\left(d_{1}, d_{2}\right)^{T}$, and the elements of the matrix $\mathbf{A}$ and the vector $\mathbf{b}$ are given by

$$
\begin{array}{ll}
a_{11}=\mathbf{h}(b)^{T} \Gamma^{-1}(\mu) \mathbf{h}(b), & a_{12}=\left(b-c_{1}\right) a_{11}-\mathbf{k}(b)^{T} \Gamma^{-1}(\mu) \mathbf{h}(b) \\
a_{21}=\mathbf{h}(b)^{T} \Gamma^{-1}(\mu) \mathbf{k}(b), & a_{22}=\left(b-c_{1}\right) a_{21}-\mathbf{k}(b)^{T} \Gamma^{-1}(\mu) \mathbf{k}(b) \\
b_{1}=\mathbf{h}(b)^{T} \Gamma^{-1}(\mu) \mathbf{e}_{o b s}, & b_{2}=\mathbf{k}(b)^{T} \Gamma^{-1}(\mu) \mathbf{e}_{o b s} \tag{6.22}
\end{array}
$$

Here $\boldsymbol{\Gamma}^{-1}(\mu)$ stands for $\left[\boldsymbol{\Gamma}\left(c_{1}, c_{2}\right)+\mu\left(\mathbf{W}^{T} \mathbf{W}\right)^{-1}\right]^{-1}$. If the special cases $\mathbf{h}(b)=\mathbf{0}, \mathbf{k}(b)=\mathbf{0}$, and $\mathbf{k}(b)=\beta \mathbf{h}(b)$ (where $\beta$ is a constant) are excluded, then it can be shown via the CauchySchwartz inequality that the determinant of this system is nonzero. The optimum values of $d_{1}$ and $d_{2}$ obtained by solving (6.21) are linear combinations of the observed data $\mathbf{e}_{\text {obs }}$. After these two parameters are determined, the coefficient vector $\alpha$ is found by solving equation (6.18).

### 6.2.4 Constructing the model

The model $m(x)$ is obtained by integrating the second derivative $m^{\prime \prime}(x)$ in equation (6.13) twice. Integration constants are chosen so that the predicted data $\mathbf{e}_{p r d}(m)$ reproduce the observed data $\mathbf{e}_{o b s}$ in the limit as the tradeoff parameter $\mu$ approaches zero. Thus

$$
\begin{equation*}
m(x)=d_{1}+d_{2}\left(x-c_{1}\right)+\boldsymbol{\alpha}^{T} \int_{c_{1}}^{x} \int_{c_{2}}^{v} \mathbf{p}\left(u ; c_{1}, c_{2}\right) d u d v \tag{6.23}
\end{equation*}
$$

Optimizing the values of the parameters $d_{1}$ and $d_{2}$ in the above manner has beneficial implications for the curvature of the model. In general, the second derivative of the model
is given by expression (6.13). Substituting in the explicit form for the kernel function vector $\mathrm{p}\left(x ; c_{1}, c_{2}\right)$ yields
$m^{\prime \prime}(x)=w(x)^{-2}\left\{\boldsymbol{\alpha}^{T} \mathbf{k}(x)-\boldsymbol{\alpha}^{T} \mathbf{h}(b)\left[R\left(x-c_{1}\right)-\left(b-c_{1}\right) H\left(x-c_{2}\right)\right]-\boldsymbol{\alpha}^{T} \mathbf{k}(b) H\left(x-c_{2}\right)\right\}$.

If conditions (6.19) and (6.20) hold, then the second derivative simplifies considerably to $m^{\prime \prime}(x)=\alpha^{T} \mathbf{k}(x) / w(x)^{2}$. These extra conditions remove a step discontinuity in the second derivative at $x=c_{2}$ due to the Heaviside function. Assuming that $\mathbf{k}(x)$ and $w(x)$ are continuous, then the constructed model is twice continuously differentiable, i.e. $m(x) \in C^{2}$ on $(a, b)$.

This model is also unique except in those cases where the determinant of the $2 \times 2$ matrix A vanishes. Three particular cases are readily identified by analysis of $\operatorname{det} \mathbf{A}$. From equations (6.22):

$$
\operatorname{det} \mathbf{A}=\left[\mathbf{h}(b)^{T} \boldsymbol{\Gamma}^{-1}(\mu) \mathbf{k}(b)\right]^{2}-\left[\mathbf{h}(b)^{T} \Gamma^{-1}(\mu) \mathbf{h}(b)\right]\left[\mathbf{k}(b)^{T} \boldsymbol{\Gamma}^{-1}(\mu) \mathbf{k}(b)\right]
$$

Obviously, this determinant vanishes if either $\mathbf{h}(b)$ or $\mathbf{k}(b)$ equals $\mathbf{0}$. The sole remaining case arises in the following manner. Since $\Gamma(\mu)$ is positive definite, $\Gamma^{-1 / 2}(\mu)$ exists. Define two vectors $\mathbf{u}$ and $\mathbf{v}$ as:

$$
\mathbf{u} \equiv \Gamma^{-1 / 2}(\mu) \mathbf{h}(b), \quad \mathbf{v} \equiv \Gamma^{-1 / 2}(\mu) \mathbf{k}(b)
$$

The above expression for the determinant becomes $\operatorname{det} \mathbf{A}=\left(\mathbf{u}^{T} \mathbf{v}\right)^{2}-\left(\mathbf{u}^{T} \mathbf{u}\right)\left(\mathbf{v}^{T} \mathbf{v}\right)$. If $\mathbf{u} \neq \beta \mathbf{v}$ (where $\beta$ is a constant), then the Cauchy-Schwartz inequality implies that $\|\mathbf{u}\|^{2}$ $\|\mathbf{v}\|^{2}>\left(\mathbf{u}^{T} \mathbf{v}\right)^{2}$. Hence, the determinant is nonvanishing in this situation, which is equivalent to the condition $\mathbf{h}(b) \neq \beta \mathbf{k}(b)$.

In these three cases, the constructed model is unique only to within an arbitrary additive constant $a_{0}$ or linear function $a_{0}+a_{1} x$. These situations are now described in detail.

Case 1: $\mathbf{h}(b)=\mathbf{0}$. This situation arises if the original kernel functions $g_{j}(x)$ all possess zero area (e.g., Oldenburg, 1981). The system $\mathbf{A d}=\mathbf{b}$ reduces to a single equation that can be solved for $d_{2}$ :

$$
d_{2}=-\frac{\mathbf{k}(b)^{T} \Gamma^{-1}(\mu) \mathbf{e}_{o b s}}{\mathbf{k}(b)^{T} \boldsymbol{\Gamma}^{-1}(\mu) \mathbf{k}(b)}
$$

Also, the linear system for the coefficient vector $\alpha$ simplifies to

$$
\left[\Gamma\left(c_{1}, c_{2}\right)+\mu\left(\mathbf{W}^{T} \mathbf{W}\right)^{-1}\right] \boldsymbol{\alpha}=\mathbf{e}_{o b s}+d_{2} \mathbf{k}(b)
$$

Since the right hand side is known, solution for $\alpha$ is possible. The model $m(x)$ is then calculated from (6.23) by picking any value for the parameter $d_{1}$. Thus, constructed models differ by an arbitrary additive constant.

Case 2: $\mathbf{k}(b)=\mathbf{0}$. The $2 \times 2$ system $\mathbf{A d}=\mathbf{b}$ reduces to a single relation between $d_{1}$ and $d_{2}$ :

$$
d_{1}+\left(b-c_{1}\right) d_{2}=\frac{\mathbf{h}(b)^{T} \Gamma^{-1}(\mu) \mathbf{e}_{o b s}}{\mathbf{h}(b)^{T} \Gamma^{-1}(\mu) \mathbf{h}(b)} \equiv \lambda
$$

Also, expression (6.18) becomes

$$
\left[\Gamma\left(c_{1}, c_{2}\right)+\mu\left(\mathbf{W}^{T} \mathbf{W}\right)^{-1}\right] \boldsymbol{\alpha}=\mathbf{e}_{o b s}-\lambda \mathbf{h}(b)
$$

It is again possible to solve for the coefficient vector $\boldsymbol{\alpha}$. The model is constructed via (6.23) after picking a value for either $d_{1}$ or $d_{2}$; the remaining parameter is obtained from the above relation between $d_{1}$ and $d_{2}$. Constructed models then differ by an arbitrary linear function of $x$.

Case 3: $\mathbf{k}(b)=\beta \mathbf{h}(b)$. This case is a generalization of the previous one. The two equations in (6.21) are no longer independent, but reduce to the single expression

$$
d_{1}+\left(b-c_{1}-\beta\right) d_{2}=\lambda
$$

Once again, equation (6.18) becomes

$$
\left[\Gamma\left(c_{1}, c_{2}\right)+\mu\left(\mathbf{W}^{T} \mathbf{W}\right)^{-1}\right] \boldsymbol{\alpha}=\mathbf{e}_{o b s}-\lambda \mathbf{h}(b)
$$

and can be solved for the coefficients. Model construction proceeds as indicated for Case 2 above. Hence, the model is unique only to within an arbitrary linear function.

This analysis highlights a particular advantage of the current formulation, where the separate system (6.21) is derived for the two parameters $d_{1}$ and $d_{2}$. The specific mathematical conditions for nonuniqueness in the calculated model can be determined simply by examining the equation $\operatorname{det} \mathbf{A}=0$. Moreover, as indicated above, model construction can still proceed in these situations. In contrast, the alternative development of Parker et al. (1987) is very ambiguous regarding both of these issues.

### 6.3 Interpolation via the smoothest model

The theory developed in the previous section is clearly illustrated by the problem of constructing a smooth interpolation of a set of discrete samples. This is a common problem in many branches of geophysics. The particular application considered here consists of calculating a continuous depth profile of a refracting interface from a set of point estimates of the refractor depth and dip. These point estimates can be derived by analyzing the first break traveltimes of a seismic experiment by various techniques. There are many such techniques available to the practicing interpreter, and these are not examined individually here. However, the methods typically employ the assumption that the critically refracting horizon is plane or nearly plane. Under these circumstances, it is logical to construct an interface depth profile that possesses minimum curvature. Thus, the final model for the interface adheres closely to the prior theoretical assumptions used for inferring its depth.

Many shallow seismic refraction projects are undertaken in conjunction with a drilling program (e.g., Hawkins, 1961; Hasselström, 1969; Hatherly and Neville, 1986; Schwarz, 1990). Accurate depths to an interface are determined at the drillholes and the refraction
method is then used to extend this information between or beyond the wells. In some cases, outcrops of the target horizon provide additional geologic control on both depth and dip (Kilty et al., 1986). When constructing a profile of the interface, the depth and dip information from all available sources should be integrated together. Extra data from drillholes, trenching, outcrops, etc. then contribute directly to the solution, rather than merely providing 'tiepoints' to the profile determined from refraction data. This goal can be readily achieved if the problem of calculating the refractor profile is posed as an interpolation issue. In this case, the data weighting matrix $\mathbf{W}$ assumes an important role. This matrix can be used to nondimensionalize different data types, and to emphasize those particular data deemed more important in the inversion. In contrast, there is no provision in the GRM depth conversion technique for including additional geological or geophysical data with variable weighting.

### 6.3.1 General theory

The complete set of $N$ measured data is divided into two distinct subsets: $n$ depth samples and $N-n \operatorname{dip}$ samples, where $0 \leq n \leq N$. These data are observed at abscissae $x_{j}$ that are ordered as follows:

$$
\begin{array}{ll}
\text { Depth : } & a \leq x_{1}<x_{2}<\cdots<x_{n-1}<x_{n} \leq b, \\
\text { Dip : } & a \leq x_{n+1}<x_{n+2}<\cdots<x_{N-1}<x_{N} \leq b
\end{array}
$$

The abscissae associated with the dip samples do not necessarily coincide with those of the depth samples. In particular, note that the extreme cases $n=N$ (no dip samples) and $n=0$ (no depth samples) are allowed.

Expressions for the kernel functions $p_{j}\left(x ; c_{1}, c_{2}\right)$ are now derived. The depth data are treated first. For $j=1,2, \ldots, n$, a predicted datum is considered to be a sample from a
function $e(x)$ :

$$
e_{j}^{p r d} \equiv e\left(x_{j}\right)=\int_{a}^{b} \delta\left(x-x_{j}\right) e(x) d x
$$

The $j^{\text {th }}$ component of the kernel function vector $\mathrm{g}(x)$ is a Dirac delta function centered at the sampling point $x_{j}: g_{j}(x)=\delta\left(x-x_{j}\right)$. Hence, the singly and doubly integrated kernels are

$$
h_{j}(x)=H\left(x-x_{j}\right), \quad \quad k_{j}(x)=R\left(x-x_{j}\right)
$$

An expression for the more general kernel function $p_{j}\left(x ; c_{1}, c_{2}\right)$ is then obtained from equation (6.10):

$$
\begin{equation*}
p_{j}\left(x ; c_{1}, c_{2}\right)=R\left(x-x_{j}\right)-R\left(x-c_{1}\right)+\left(x_{j}-c_{1}\right) H\left(x-c_{2}\right) . \tag{6.24}
\end{equation*}
$$

The dip data are examined next. A simplification arises if these data are redefined as slope, rather than dip samples. The transformation is easily effected by calculating the tangent of each dip angle. From the fundamental theorem of calculus, the difference in the slope of the function $e(x)$ between two locations $x_{j}$ and $c_{2}$ is given by

$$
e^{\prime}\left(x_{j}\right)-e^{\prime}\left(c_{2}\right)=\int_{c_{2}}^{x_{j}} e^{\prime \prime}(x) d x=\int_{a}^{b}\left[H\left(x-c_{2}\right)-H\left(x-x_{j}\right)\right] e^{\prime \prime}(x) d x
$$

Thus, for $j=n+1, n+2, \ldots, N$, a predicted slope datum is a functional of the second derivative of $e(x)$ :

$$
e_{j}^{p r d} \equiv e^{\prime}\left(x_{j}\right)=d_{2}+\int_{a}^{b} p_{j}\left(x ; c_{1}, c_{2}\right) e^{\prime \prime}(x) d x
$$

where $d_{2}=e^{\prime}\left(c_{2}\right)$ and the kernel function is defined by

$$
\begin{equation*}
p_{j}\left(x ; c_{1}, c_{2}\right)=H\left(x-c_{2}\right)-H\left(x-x_{j}\right) \tag{6.25}
\end{equation*}
$$

Comparison of the above two expressions with equations (6.9) and (6.10) indicates that $h_{j}(b)=0$ and $k_{j}(b)=-1$ for the index range $j=n+1, n+2, \ldots, N$. Hence, if the dataset consists entirely of slope samples ( $n=0$ ), then the interpolation problem is an example of the Case 1 phenomenon described in the previous section (i.e., the constructed interpolant $e(x)$ is unique only to within an additive constant).

If the weighting function $w(x)$ equals unity, then explicit formulae for the elements of the inner product matrix $\Gamma\left(c_{1}, c_{2}\right)$ can be derived by substituting the above kernel functions into (6.16) and integrating. Arbitrary nonuniform weighting requires that the integrals be evaluated by numerical techniques. The examples presented below adopt uniform weighting. For $c_{1} \leq c_{2}$, there are 15 distinct formulae for the matrix elements $\Gamma_{i j}$, corresponding to various locations of the coordinates $x_{i}$ and $x_{j}$ relative to $c_{1}$ and $c_{2}$. These equations are omitted here for brevity. If $c_{1}>c_{2}$, then a transformation of the independent variable via $x^{\prime}=a+b-x$ and a reindexing of the data allow the previously established formulae to be used.

As indicated previously, it is possible to solve for $\boldsymbol{\alpha}, d_{1}$, and $d_{2}$ simultaneously. This is the preferred method of solution for situations with a large amount of data. It eliminates the need to invert $\Gamma(\mu)$ in order to obtain the elements of the matrix $\mathbf{A}$ and the vector $\mathbf{b}$ as in (6.22). However, for small scale problems like the following examples, calculation of $\Gamma^{-1}(\mu)$ is not computationally burdensome. Singular value decomposition (SVD) is used here to compute the inverse. An advantageous feature of this technique is that it allows the singular vectors associated with very small eigenvalues to be winnowed from the solution simply by truncating the SVD sum. Small eigenvalues arise when the matrix $\Gamma(\mu)$ is nearly singular, and contribute to a large variance in the computed solution.

Let $\Gamma_{i j}^{-1}(\mu)$ denote the $i j^{t h}$ element of the matrix $\left[\Gamma\left(c_{1}, c_{2}\right)+\mu\left(\mathbf{W}^{T} \mathbf{W}\right)^{-1}\right]^{-1}$. Then, define three sums as follows:

$$
P_{i}=\sum_{j=1}^{n} \Gamma_{i j}^{-1}(\mu), \quad Q_{i}=\sum_{j=1}^{n} x_{j} \Gamma_{i j}^{-1}(\mu), \quad R_{i}=\sum_{j=n+1}^{N} \Gamma_{i j}^{-1}(\mu) .
$$

These quantities are all defined for the full index range $i=1,2, \ldots, N$. After some algebraic manipulation, the elements of the matrix $\mathbf{A}$ and the vector $\mathbf{b}$ in equation (6.21) reduce to the simple forms

$$
\begin{array}{ll}
a_{11}=\sum_{i=1}^{n} P_{i}, & a_{12}=\sum_{i=1}^{n}\left(x_{i}-c_{1}\right) P_{i}+\sum_{i=n+1}^{N} P_{i} \\
a_{21}=\sum_{i=1}^{n}\left(Q_{i}+R_{i}\right), & a_{22}=\sum_{i=1}^{n}\left(x_{i}-c_{1}\right)\left(Q_{i}+R_{i}\right)+\sum_{i=r}^{N} \\
b_{1}=\sum_{i=1}^{N} e_{i}^{o b s} P_{i}, & b_{2}=\sum_{i=1}^{N} e_{i}^{o b s}\left(Q_{i}+R_{i}\right) .
\end{array}
$$

Solution for $d$ and then the coefficient vector $\boldsymbol{\alpha}$ is now possible. The problem is completed by constructing the function $e(x)$ via equation (6.23). Expressions for the doubly integrated kernel functions in (6.23) are obtained by integrating equations (6.24) and (6.25) for $p_{j}\left(x ; c_{1}, c_{2}\right)$. There are five distinct cases that must be examined, each corresponding to different location of the coordinate $x_{j}$ relative to the two abscissae $c_{1}$ and $c_{2}$.

### 6.3.2 Numerical example

Figures 6.1 through 6.4 present a synthetic example from shallow refraction seismology. The top panel in each figure depicts the elevation profile of a near surface alluvium/bedrock interface (dashed line). A small channel is located adjacent to a larger and broader anticline. Elevations are referenced to an arbitrary horizontal datum plane. Nonuniformly spaced elevation and slope samples from the interface are indicated by asterisks. Initially, these are considered to be error free.

If accurate estimates of interface depth and dip are available (say, from a shallow borehole) then these values can be used to construct a smoothest elevation model. Figure 6.1 depicts this situation. The prescribed elevation and slope values are indicated by open


Fig. 6.1. A smoothest refractor model $e(x)$ constructed from six error-free depth samples (indicated by asterisks). Dashed lines (--) refer to the true earth model and solid lines (-) refer to the constructed model. The independent values of the refractor elevation and slope are taken from the true earth model and are indicated by small squares at abscissa $c_{1}=c_{2}=23.75 \mathrm{~m}$. The $l_{2}$ norm of the second derivative of the constructed model is $\left\|e^{\prime \prime}\right\|=0.228 \mathrm{~m}^{-1 / 2}$.


Fig. 6.2. Same as Fig. 6.1 except that optimum values of the refractor elevation and slope are calculated at $c_{1}=c_{2}=23.75 \mathrm{~m}$. The $l_{2}$ norm of the second derivative is reduced to $\left\|e^{\prime \prime}\right\|=0.129 \mathrm{~m}^{-1 / 2}$.
squares at horizontal coordinate $c_{1}=c_{2}=23.75 \mathrm{~m}$. The constructed model and its derivative (solid curves) pass through these points. However, the second derivative of the model possesses a jump discontinuity at this location. This discontinuity arises from the Heaviside step function with onset at $x=c_{2}$ in equation (6.24). If the depth and dip values are specfied erroneously (i.e., unequal to the true earth model values) then the cusp in $e^{\prime}(x)$ and the discontinuity in $e^{\prime \prime}(x)$ can be magnified (Aldridge et al., 1991).

Figure 6.2 displays the smoothest elevation model constructed by using the optimum values of $d_{1}$ and $d_{2}$. There are obvious differences from the previous model on the flank of the anticline and on the flat portion between anticline and syncline. The second derivative of the elevation profile is now a continuous, piecewise linear function. It is emphasized that the same model is obtained if the parameters $d_{1}$ and $d_{2}$ are optimized at any pair of horizontal positions $c_{1}$ and $c_{2}$ along the profile (including outside of the range encompassed by the six elevation samples). The $l_{2}$ norm of the second derivative is reduced by optimizing the parameters, as expected. Aldridge et al. (1991) discuss the relation of these interpolants to cubic splines. In particular, the constructed elevation profile in Figure 6.2 is a cubic spline on $[a, b]$, whereas $e(x)$ in Figure 6.1 consists of two cubic splines joined end-to-end at the coordinate $x=c_{2}$.

Inclusion of slope samples, as in Figure 6.3, results in a more accurate reconstruction of the refracting interface and its first derivative. However, continuity of $e^{\prime \prime}(x)$ is now sacrificed. As predicted by equation (6.25), a jump discontinuity is introduced into the second derivative at each abscissa $x_{j}$ associated with a slope datum. Hence, the constructed elevation profile is no longer a cubic spline on $[a, b]$.

In actual practice, the calculated interface depth and dip samples will contain some random error. This situation is simulated in Figure 6.4. Random numbers drawn from a uniform probability distribution on $\pm 0.50 \mathrm{~m}$ (standard deviation $=0.29 \mathrm{~m}$ ) are added to the accurate elevations. Also, each slope sample is perturbed by an amount corresponding to a uniformly distributed random dip angle on $\pm 5^{\circ}$ (standard deviation $=2.9^{\circ}$ ). A model that is an exact fit to these error contaminated samples would contain spurious structure induced by the noise. Hence, a model is constructed that reproduces the erroneous data


Fig. 6.3. A smoothest refractor model constructed from six depth samples and six dip samples. All data are error-free. Optimum values of the refractor elevation and slope are calculated at $c_{1}=c_{2}=23.75 \mathrm{~m} .\left\|e^{\prime \prime}\right\|=0.354 \mathrm{~m}^{-1 / 2}$.


Fig. 6.4. A smoothest refractor model constructed from error contaminated data. Depth and dip samples are located at the same abscissae as in Fig. 6.3. Optimum values of the refractor elevation and slope are calculated at $c_{1}=c_{2}=23.75 \mathrm{~m} .\left\|e^{\prime \prime}\right\|=0.259 \mathrm{~m}^{-1 / 2}$.
approximately, rather than exactly. The tradeoff parameter $\mu$ and the weighting matrix $\mathbf{W}$ can be used advantageously for this purpose. Since a constructed smoothest model depends on the value of the tradeoff parameter, predicted data generated by this model will also be a function of $\mu$. The notation $\mathbf{e}_{\text {prd }}(\mu)$ is used to explicitly denote this dependence. If the weighting matrix is diagonal with elements $W_{i}, i=1,2, \ldots, N$, then the degree of misfit between observed and predicted data is quantified by

$$
\begin{gathered}
\epsilon_{d e p t h}(\mu) \equiv \sqrt{\frac{1}{n} \sum_{i=1}^{n} W_{i}^{2}\left[e_{i}^{o b s}-e_{i}^{p r d}(\mu)\right]^{2}}, \\
\epsilon_{s l o p e}(\mu) \equiv \sqrt{\frac{1}{N-n} \sum_{i=n+1}^{N} W_{i}^{2}\left[e_{i}^{o b s}-e_{i}^{p r d}(\mu)\right]^{2}} .
\end{gathered}
$$

A value for the tradeoff parameter is sought such that these misfits are approximated by

$$
\epsilon_{\text {depth }}(\mu) \simeq \sqrt{\frac{1}{n} \sum_{i=1}^{n} W_{i}^{2} \sigma_{i}^{2}}, \quad \epsilon_{s l o p e}(\mu) \simeq \sqrt{\frac{1}{N-n} \sum_{i=n+1}^{N} W_{i}^{2} \sigma_{i}^{2}}
$$

Note that, for uniform weighting, the right hand side of each expression above is proportional to the rms value of the standard deviations of the data errors. Selection of a search procedure for $\mu$ is largely a matter of personal preference. For small scale problems like the present example, a few trial runs with guessed values of $\mu$ allow a sufficiently precise estimate to be made. Alternately, it is possible to implement an automated iterative scheme that converges upon the desired tradeoff parameter.

Figure 6.4 displays a refractor elevation model constructed with the misfit values $\epsilon_{\text {depth }}=$ 0.30 m and $\epsilon_{\text {slope }}=0.0063$. The slope data are weighted by 0.12 relative to the depth data. Hence, the misfit to the slope samples corresponds to one standard deviation of a uniform distribution of dip angles on $\pm 5^{\circ}$. Also, optimum values for the two parameters $d_{1}$ and $d_{2}$ are used. Interest in detecting the shallow channel may stem from various exploration
objectives: placer ore localization, groundwater accumulation, contaminant migration, etc. However, the figure suggests that the channel is just at the limit of detectability for the given horizontal sampling and error statistics.

### 6.4 Conclusion

The conventional method for constructing the smoothest model is generalized in two important ways:

1) The additional information required to calculate the smoothest model may be specified anywhere within the interval of definition $[a, b]$. This improvement can be particularly useful if the best known value of the model (or its slope) is not located at an endpoint.
2) Maximum smoothness can be obtained by calculating optimum values of the model and its slope directly from the observed data. This technique is advisable if independent values of these parameters are unknown or are difficult to estimate accurately.

It is evident that these techniques can be extended to model construction situations where the $l_{2}$ norm of a higher derivative of the model is minimized. The modified data equation satisfied by the $n^{\text {th }}$ derivative of the model $m^{(n)}(x)$ will entail $n$ parameters $d_{i}$ defined at the abscissae $c_{i}$ as follows: $d_{i} \equiv m^{(i-1)}\left(c_{i}\right), i=1,2, \ldots, n$. Extremizing the relevant objective function leads to an $n \times n$ system of equations $\mathbf{A d}=\mathbf{b}$ for the unknown $d_{i}$. A unique solution exists when $\operatorname{det} \mathbf{A} \neq 0$, and the model constructed with these parameters will possess a continuous $n^{t h}$ derivative $\left(m(x) \in C^{n}\right.$ ) provided that the original kernel function vector $\mathrm{g}(x) \in C^{-n}$ (this assumes that the weighting function $w(x)$ is also continuous). If the determinant vanishes, then the model that yields an absolute minimum of $\left\|m^{(n)}\right\|^{2}$, subject to constraints provided by the data, can still be constructed but is only unique to within an additive polynomial of at most degree $n-1$. Only the smallest model $(n=0)$ is unique in all situations.

Application of this model construction formalism to the problem of calculating an interface depth profile yields a useful tool for seismic refraction exploration. The smoothest model is the natural model to adopt in this situation because refraction traveltime inversion
methods assume, either explicitly or implicitly, that local interface curvature is negligable. The resulting depth profile and its derivatives are mathematically well defined throughout the horizontal range $(a, b)$ and thus can be used for other forward modeling purposes. The method is also flexible, as evidenced by the following specific advantages compared with depth determination via the GRM:

1) Additional depth and dip data arising from a variety of geological, geophysical, or engineering techniques are readily incorporated into the computation.
2) Variable weighting of all data is easily achieved.
3) An adjustable misfit to error contaminated data is possible.

Numerical examples presented here and in Aldridge et al. (1991) demonstrate the feasibility of the technique for refractor depth profile construction. Obviously, there is still much to learn about issues such as weighting of different data types, resolvability of small scale features, optimum horizontal sampling, etc. However, the algorithm presented here provides a flexible tool for investigating these and related phenomena.

## CHAPTER 7

## REFRACTOR IMAGING USING AN AUTOMATED WAVEFRONT RECONSTRUCTION METHOD

### 7.1 Introduction

The Wavefront Method is one of the earliest of the many techniques for interpreting refraction arrival times. In 1930, Thornburgh demonstrated that subsurface wavefronts could be reconstructed from surface arrival times by applying Huygens' principle in reverse. Subsequently, Hagedoorn (1959) elucidated an imaging condition for delineating a refracting horizon. First, two oppositely propagating wavefront systems are reconstructed from the arrival times recorded on a forward and reverse spread, respectively. Then, pairs of these subsurface wavefronts intersect on or slightly below the refracting interface when the sum of their times equals the known reciprocal time (the shot-to-shot traveltime). This imaging principle yields the correct spatial locus of a critically refracting horizon if the earth consists of constant velocity layers bounded by plane dipping interfaces. However, several investigators have demonstrated that the imaging condition is reasonably accurate even if the measured arrival times are due to diving rays, rather than true critically refracted rays (Hagedoorn, 1959; Rockwell, 1967; Schenck, 1967; Hill, 1987). Diving rays may arise from nonplane structure on the refracting interface, or a velocity gradient within the underlying medium.

Extensive application of the wavefront method has been limited by two factors: i) laborious graphical techniques are required to construct the subsurface wavefront loci, and ii) detailed knowledge of the near surface velocity structure is necessary. This study directly addresses the first of these two issues. Instead of defining the wavefronts by a tedious graphical application of Huygens' principle (e.g., Rockwell, 1967), a finite-difference computer algorithm is used to downward continue surface arrival times through a specified velocity field. The algorithm is rapid and accurate, and is capable of handling a heterogeneous velocity structure.

Recently, Hill (1987) downward continued refracted waveforms to obtain a two dimensional image of shallow structure. Although the technique presented here works only with arrival times, the goal is identical. The advantage of this approach resides in its computational simplicity. Since the propagation algorithm operates directly in the space-time domain, no transformations of the recorded wavefield, with attendent concerns about sampling adequacy (Clayton and McMechan, 1981; Hill, 1987) are necessary. Furthermore, true amplitude recording and processing of seismic traces are not required. However, prior picking of these traces to obtain the arrival times is necessary, and this may be a time consuming job in some situations.

### 7.2 Finite-difference traveltimes

### 7.2.1 Wavefront Construction

Vidale $(1988,1990)$ has recently developed an algorithm for calculating the first arrival times of a seismic wave propagating through a two or three dimensional velocity structure. The velocity field is sampled on a uniformly spaced 2D or 3D grid; plane wave finite-difference operators are used to extrapolate the traveltimes from point to point throughout this grid. Calculations are initiated at a source point within the predefined velocity field. The algorithm properly handles the various wave types that comprise first arrivals (body waves, head waves, and diffractions). Subsequent contouring of the computed traveltime field yields a visual impression of propagating wavefronts. The phrase wavefront construction is used here to refer to traveltime loci calculated in this manner. Figure 7.1 depicts the subsurface wavefront systems generated by a sequence of shots buried in an earth model with undulating surface and refractor topography. The direct wave through the overburden is the initial arrival near each shot location. Beyond the crossover distance, the wave refracted by the higher velocity bedrock arrives first. The traveltimes recorded along the nonplane surface of the model are accurately computed by assigning a P -wave velocity to the uppermost layer equal to the speed of sound in air ( $\sim 350 \mathrm{~m} / \mathrm{s}$ ). Wavefront contours are then suppressed in


Fig. 7.1. Subsurface first arrival wavefronts (contour interval $=2 \mathrm{~ms}$ ) for an earth model with undulating surface and refractor topography. Layer P-wave speeds are $v_{0}=350 \mathrm{~m} / \mathrm{s}$, $v_{1}=1500 \mathrm{~m} / \mathrm{s}$, and $v_{2}=2500 \mathrm{~m} / \mathrm{s}$. Shots are buried 1 m deep and grid cell size is 0.2 m . For visual clarity, contours within the uppermost layer (air) are suppressed.


Fig. 7.2. Surface arrival time curves for the five wavefront systems depicted in Fig. 7.1.
this region for visual clarity. The surface arrival time curves displayed in Figure 7.2 illustrate that nonplane topography has a complicating effect on an interpretation.

Vidale's wavefront construction algorithm has been altered in two important ways in order to improve its suitability for the shallow refraction problem. First, the traveltime calculations are initiated from a spatially extended source, rather than a point source. In the 2D case, source activation times are specified on and inside a rectangular region located within the velocity field; arrival times at grid points outside this rectangle are generated by the normal working of the algorithm. Since wavefronts are strongly curved in the immediate vicinity of a point source, use of the plane wave finite-difference operators will yield inaccurate traveltimes in this region. Moreover, these inaccuracies will be propagated to greater distances, where the plane wave extrapolators are locally valid. In order to avoid this problem, near source traveltimes are calculated via mathematically exact formulae appropriate
for either a constant or linear velocity field. Although more complicated velocity distributions can be considered, these particular velocity functions provide sufficient flexibility for many traveltime computation problems.

Second, the mathematical form of the traveltime extrapolation operator is modified in those cases where there is a large velocity increase across a grid cell. This situation is relatively common in the shallow refraction environment. The interface between unconsolidated overburden and consolidated bedrock, or between saturated and unsaturated alluvium, often represents a sharp velocity increase. In these cases, as the following analysis indicates, the conventional traveltime extrapolation formula may fail.

Figure 7.3a depicts a system of plane wavefronts propagating across a square grid cell with side length $h$. The arrival time at the corner numbered 4 must be calculated from the known arrival times $t_{1}, t_{2}$, and $t_{3}$ at the other three corners of the cell. Assuming a plane wave advancing with a constant slowness $s$, this time is given by $t_{4}=t_{1}+(\sqrt{2} h \cos \theta) s$, where the angle $\theta$ describes the ray direction relative to the cell diagonal. Simple geometric analysis yields

$$
\cos \theta=\sqrt{1-\left(\frac{t_{3}-t_{2}}{\sqrt{2} h s}\right)^{2}}
$$

Hence

$$
\begin{equation*}
t_{4}=t_{1}+\sqrt{2(h s)^{2}-\left(t_{3}-t_{2}\right)^{2}} \tag{7.1}
\end{equation*}
$$

This expression is identical to Vidale's (1988) equation (3), which was derived by approximating the partial derivatives in the 2D eikonal equation by finite differences and then solving algebraically for $t_{4}$. The present derivation clearly reveals the underlying geometric assumption of plane wave propagation.

If the argument of the square root in equation (7.1) becomes negative, then the plane wave extrapolation formula is obviously invalid. This may occur, for example, if there is a dramatic velocity increase across the cell (implying that the slowness $s$ assigned to the cell


Fig. 7.3. Finite-difference traveltime extrapolation operators. (a) Locally plane wavefronts; $\theta$ is the angle between the ray (heavy line) and the cell diagonal. (b) Circular wavefronts. In each case, the square grid cell has side length $h$ and assigned slowness $s$.
is quite small). In these cases, the arrival time $t_{4}$ is calculated via the alternative formula

$$
\begin{equation*}
t_{4}=\min \left\{t_{1}+\sqrt{2} h s, t_{2}+h s, t_{3}+h s\right\} \tag{7.2}
\end{equation*}
$$

The geometric basis of equation (7.2) is illustrated in Figure 7.3b. In effect, the plane wavefront approximation is abandoned and Huygens' principle is used directly to calculate the next traveltime. Although this computed time is not exactly correct, extensive numerical testing indicates that equation (7.2) is superior to the fix advocated by Vidale (1990) (i.e., if the argument of the square root becomes negative, take $t_{4}=t_{1}$ ).

### 7.2.2 Wavefront reconstruction

Figure 7.4 displays the forward and reverse wavefront systems generated by shooting over a shallow syncline. These are the wavefronts that give rise to the first arrival times observed on the surface. The finite-difference traveltime algorithm can now be used to recreate subsurface wavefronts from knowledge of the arrival times recorded at the surface. The source rectangle is placed at zero depth and is greatly elongated in the horizontal dimension. This line source is then activated sequentially (rather than simultaneously) with an initiation function $T_{S}(x)$ derived from the recorded refraction arrival times $T(x)$ :

$$
\begin{equation*}
T_{S}(X)=T_{R}-T(X) \tag{7.3}
\end{equation*}
$$

where $T_{R}$ is the reciprocal time. At source-receiver offsets $X$ less than the crossover distance, phantom arrival times $T(X)$ can be constructed from parallel traveltime curves recorded from distant shotpoints (Rockwell, 1967; Ackermann et al., 1986). The line source generates a set of wavefronts radiating downward into the specified velocity field (Figure 7.5). The downward continuation velocity function $v(x, z)$ is selected as a good approximation to the actual near surface velocity structure. Hence, within the overburden, the calculated wavefronts coincide with the emerging refracted wavefronts of Figure 7.4. Since the position of the refracting interface is initially unknown, the wavefronts are continued to greater depth using the known velocity field $v(x, z)$. Rockwell (1967) referred to these traveltime loci as a "directed wavefront system"; in this study, the phrase wavefront reconstruction is used to describe the process of creating an emergent wavefront system from recorded surface arrival times.

### 7.3 Refractor imaging

Let $t_{f}(x, z)$ and $t_{r}(x, z)$ refer to the subsurface traveltime fields reconstructed from the forward and reverse arrival times, respectively. Then, according to Hagedoorn's (1959) imaging principle, the refracting interface is implicitly defined by the relation


Fig. 7.4. Forward and reverse first arrival wavefronts for a shallow syncline model. Overburden velocity $v_{1}=1500 \mathrm{~m} / \mathrm{s}$, bedrock velocity $v_{2}=2500 \mathrm{~m} / \mathrm{s}$. Grid cell size is 5 m and contour interval for wavefronts is 30 ms .


Fig. 7.5. Emergent wavefronts reconstructed from the surface arrival times recorded over the shallow syncline. Downward continuation velocity $v(x, z)=1500 \mathrm{~m} / \mathrm{s}$. Grid cell size is 5 m and wavefront contour interval is 30 ms .

$$
\begin{equation*}
t_{f}(x, z)+t_{r}(x, z)=T_{R} \tag{7.4}
\end{equation*}
$$

Figure 7.6 graphically illustrates the superposition of the two reconstructed wavefront systems shown in the prior figure. This 2D array of superposed traveltimes is systematically searched to locate grid points where the imaging condition (7.4) is satisfied. If equation (7.4) does not hold on a grid point, linear interpolation between adajacent points is used to find the proper depth. The resulting depth locus $z(x)$ (dashed line in Figure 7.6) is an accurate spatial image of the original refracting horizon, except near the edges of the input velocity field where the subsurface wavefronts are not reconstructed correctly.

Figure 7.7 indicates that the technique is also capable of imaging anticlinal structure. The apex of the anticline is imaged slightly too deep because the refracted rays penetrate beneath this stucture, rather than propagating along the undulating interface (e.g., Hagedoorn, 1959, figures 2 and 3 ). Note that a similiar problem does not occur with the syncline, because there is a tendency for the diffracted ray to follow the interface in the presence of synclinal structure.

The calculated locus for the refracting horizon depends on the reciprocal time $T_{R}$ and velocity field $v(x, z)$ used for downward continuation of the surface arrival times. Variations in these quantitites from their correct values will induce variations in the depth and position of the refractor.

It is relatively easy to assess the dependence of the refractor image on the value of the reciprocal time. The forward and reverse subsurface traveltime fields are added together and the result is contoured for various candidate "imaging times". Figure 7.8a illustrates this situation for the buried syncline. If the imaging time used is less than or greater than the true reciprocal time, then the interface image is too shallow or too deep, respectively. This particular dependence upon the reciprocal time is the converse of that predicted by the classical wavefront method (Rockwell, 1967, p. 378). The difference arises from the method of reconstructing the subsurface wavefronts. Currently, finite-difference traveltime computations are initiated with the source function (7.3) and the algorithm is run forward


Fig. 7.6. (a) Superposition of the two reconstructed wavefront systems of Fig. 7.5. Dashed line is the locus satisfying the refractor imaging condition. (b) Comparsion of the true (solid) and imaged (dashed) refracting interfaces.


Fig. 7.7. (a) Superposition of the forward and reverse reconstructed wavefronts (contour interval $=30 \mathrm{~ms}$ ) over a shallow anticline. Dashed line is the refractor image. (b) Comparison of the true (solid) and imaged (dashed) refracting interfaces.


Fig. 7.8. Dependence of the syncline locus on imaging time (top) and downward continuation velocity (bottom). Different images correspond to increments of 10 ms in imaging time and $100 \mathrm{~m} / \mathrm{s}$ in velocity, respectively. Heavy lines are the images corresponding to the correct values of $T_{R}$ and $v$.
in time. Hence, subsurface wavefronts are labeled with times later than the surface source values. In contrast, the classical wavefront reconstruction methods label the subsurface wavefronts with times earlier than the surface measured times. In either case, the true position of the interface corresponds to an imaging time equal to $T_{R}$.

Quantifying the dependence of the refractor image on the downward continuation velocity is more complicated. Forward and reverse subsurface wavefront systems must be reconstructed for each velocity function used in the analysis. Figure 7.8 b displays a set of images of the shallow syncline calculated for various values of a constant downward continuation velocity. If the velocity is less than or greater than the actual overburden velocity, then the interface image is too shallow or too deep, respectively. Moreover, a grossly incorrect continuation velocity distorts the shape of the interface structure. Hence, in common with many other seismic refraction interpretation techniques, accurate time-to-depth conversion with the wavefront method requires good knowledge of the overburden velocity distribution. This information can be obtained from uphole times, direct and reflected arrivals, shallow refractions, and borehole data.

Finally, the accuracy of the solution depends on the reliability of the picked first arrival times. The ability of the method to resolve small scale features on the refracting horizon is also limited by the field geophone interval. These phenomena are analyzed by performing the inversion with noisy traveltime data sampled at an assumed geophone interval. Figure 7.9 displays the refractor image obtained by downward continuing error contaminated arrival times sampled every 25 m . Spatially correlated, normally distributed time errors (standard deviation $=5 \mathrm{msec}$; correlation distance $=100 \mathrm{~m}$ ) are added to the theoretically exact refraction picks. Appendix F describes the method used for computing correlated random noise. A cubic spline is then loosely fitted to the noisy arrival times and is used in equation (7.3) for the source initiation function. The experiment has detected the presence of the syncline, and its lateral position and depth are approximately correct. Long wavelength undulations on the refractor are artifacts of the spatially correlated noise, but are not unduly harmful to the structural interpretation.


Fig. 7.9. Reconstructed wavefonts and syncline image formed from noisy traveltime data sampled at a geophone interval of 25 m . Grid cell size is 5 m and wavefront contour interval is 30 ms . Long wavelength undulations on the refractor arise from spatially correlated traveltime errors.

### 7.4 Refractor velocity estimation

A particular advantage of the wavefront method is that the interface depth calculation is independent of the refractor velocity. Rather, the velocity of the substratum can be estimated after the position of the refracting horizon is determined. The distance between two points on the interface divided by the difference in the reconstructed wavefront times at these points is an estimate of the refractor velocity. This value is assigned to the midpoint of the two points for plotting purposes. In effect, the directional derivative of the subsurface traveltime field along the interface locus is computed by a centered finite-difference formula; the reciprocal of this value corresponds to the local velocity of the refractor. Either the forward or reverse wavefront systems may be used for the computation. Figure 7.10 illustrates that the refractor velocity estimated in this manner possesses systematic errors related to


Fig. 7.10. Refractor velocity estimates for the syncline (top) and anticline (bottom) models. The spatial differencing interval used for the calculation is 200 m . Small amplitude, short wavelength oscillations are artifacts of the grid interval.
the interface structure. However, for the two synthetic examples examined here, the inferred velocity values are everywhere within $3 \%$ (or $\pm 75 \mathrm{~m} / \mathrm{s}$ ) of the correct values.

### 7.5 Field data example

The interface imaging procedure is tested with a shallow refraction dataset acquired at the archeological site of Phalasarna in western Crete. Hadjidaki (1988) discusses the historical and archeological significance of this site and also gives a detailed description of the surface and near subsurface conditions. Forward and reverse refraction profiles were recorded along an inline spread of 18 geophones (geophone interval $=0.5 \mathrm{~m}$, near source offset $=0.5 \mathrm{~m}$ ) during the summer of 1989. The data acquisition system consisted of a portable signal stacking seismograph with a hammer energy source. First arrival time picks were made on stacked traces in order to reduce random errors induced by ambient noise.

The arrival times observed at reciprocal source positions at opposite ends of the spread differ slightly ( 16.40 msec vs. 16.70 msec on the forward and reverse profiles, respectively). Since a successful inversion requires consistency in the measured reciprocal times, reciprocal time corrections (Hatherly, 1982) are applied to the picked arrival times. A constant time shift is added to the raw time picks on each source gather in order to adjust the observed reciprocal times to the average value of 16.55 msec .

Figure 7.11 displays the 18 first break picks recorded on the forward and reverse profiles after application of these reciprocal time corrections. A preliminary interpretation of the plotted traveltime curves identifies the direct and refracted branches. Overburden velocity, determined from the slopes of the direct arrival segments, exhibits a weak lateral variation ( $\sim 8 \%$ ) over the 9 m spread length. This information is used to construct a near surface velocity function for subsequent downward continuation of the refracted arrival times. A cubic spline is fitted to the 16 refraction picks on each spread and is extrapolated to zero offset as a straight line. These curves are then used in equation (7.3) to calculate the source initiation functions required for the wavefront reconstruction algorithm.


Fig. 7.11. Shallow refraction traveltime data acquired at the dry harbor of Phalasarna, Crete. First break picks are indicated by triangles, and interpreted arrival time branches by smooth curves.

Figure 7.12a depicts the subsurface wavefronts generated by downward continuing the refracted arrival times through a near surface velocity field given by $v(x)=252-2.11 x$ ( $x$ in m and $v$ in $\mathrm{m} / \mathrm{s}$ ). A shallow undulating interface is then imaged using the corrected reciprocal time $T_{R}=16.55 \mathrm{msec}$. The refractor velocity estimate (Figure 7.12b) exhibits two distinct zones: i) abrupt variations about $800 \mathrm{~m} / \mathrm{s}$ on the left, and ii) a low velocity zone slower than $800 \mathrm{~m} / \mathrm{s}$ on the right.

The validity of various refractor velocity functions can be tested by using the wavefront construction algorithm to compare predicted traveltimes with the observed traveltimes. A uniform refractor velocity of $800 \mathrm{~m} / \mathrm{s}$ leads to unacceptably large differences (Figure 7.13b). When a low velocity zone is introduced into the refractor, good agreement is obtained (Figure 7.13c). The effect of the low velocity zone is evident in the plotted forward and reverse wavefronts of Figure 7.14 between 6 m and 8 m horizontal position. Note that


Fig. 7.12. (a) Forward and reverse subsurface wavefronts (contour interval $=1 \mathrm{~ms}$ ) reconstructed by downward continuing the refracted arrival times of Fig. 7.11. Grid cell size is 0.02 m . Heavy line is the refractor image. (b) Refractor velocity function (solid curve) calculated with a differencing interval of 0.8 m .


Fig. 7.13. Comparison of observed and predicted first break times for two refractor velocity functions. (a) Constant velocity (dashed) equals $800 \mathrm{~m} / \mathrm{s}$ and variable velocity (solid) includes a low velocity zone between 6 and 8 m . (b) Predicted traveltimes calculated with the constant refractor velocity. (c) Predicted traveltimes calculated with the variable refractor velocity.


Fig. 7.14. Subsurface wavefronts (contour interval $=0.5 \mathrm{~ms}$ ) constructed from the earth model with the laterally varying refractor velocity. Grid cell size is 0.02 m . Note the effect of the low velocity zone between 6 and 8 m .
the strong variations in refractor velocity displayed on the left in Figure 7.12b need not be incorporated into the model to obtain an adequate fit to the measured arrival times. Further adjustment of the refractor velocity to obtain a closer fit is probably unwarranted.

Although the recovered model of Figure 7.14 generates acceptable predicted traveltimes, it cannot be stated with certainty that this is the correct earth model. Other interpretations of the observed traveltime data, incorporating multiple layers or lateral changes in structure and/or velocity, are possible. Since the refraction dataset does not include arrival times recorded from far offset shotpoints, it is not possible to distinguish between these alternatives (Ackermann et al., 1986). However, the inferred model is consistent with known subsurface information from the vicinity of the refraction profile. Archeological trenching conducted about 17 m distant encountered dipping sandstone bedrock at approximately 1.8 m depth (Hadjidaki, 1988). Various earthen and gravel layers overlie the bedrock. A porous, aerated sandstone might have a P-wave velocity as low as $\sim 800 \mathrm{~m} / \mathrm{s}$. Hence, the preliminary interpretation is that the interface imaged in Figure 7.14 is the upper surface of the sandstone bedrock. The very low velocity between 6 m and 8 m may be a zone of more extensive weathering, fracturing, or aeration. Alternately, it is possible that one of the overlying shallow gravel layers has been imaged.

### 7.6 Conclusion

The essential requirements for reconstructing shallow refracted wavefronts are:

1) arrival times $T(X)$ from a given marker horizon recorded (or phantomed) on a forward and reverse spread,
2) a reciprocal time $T_{R}$,
3) a near surface velocity function $v(x, z)$.

A simple modification of Vidale's finite-difference traveltime algorithm then allows the rapid calculation of the subsurface wavefront systems that give rise to the recorded arrival times.

Although the synthetic examples described here use a uniform near surface velocity, downward continuation through a varying velocity field is also possible with no increase in computation time. The buried refracting horizon is delineated in a subsequent step by applying Hagedoorn's imaging principle. No prior assumption regarding the refractor velocity is required. Rather, the velocity of the substratum can be estimated by calculating the directional derivative of the reconstructed wavefront systems along the imaged interface.

Picking of first arrival times and assignment of these picks to specific refractors are necessary in this method. The final locus for the refracting interface is sensitive to errors in the picked times, as well as to an incorrect choice of the reciprocal time and velocity field. However, since the technique is not computationally intensive, it is possible to assess the magnitude of the position and depth uncertainty by performing the inversion repetitively. The forward modeling capabilities of the finite-difference traveltime algorithm can also be used to quickly generate predicted arrival times from the inferred subsurface model. Comparison of these times with the observed data is a powerful method of establishing the significance of various features of the recovered earth model.

Finally, two specific problem areas with the automated wavefront reconstruction method have been identified that merit further research: i) downward continuation of traveltimes recorded along a nonplane surface, and ii) correction of the refractor velocity function for the effects of structurally induced errors. Although a fully automated solution to these problems is not yet available, this should not prevent the immediate application of the method in shallow seismic refraction exploration.

## CHAPTER 8 <br> TWO DIMENSIONAL TOMOGRAPHIC INVERSION WITH FINITE-DIFFERENCE TRAVELTIMES

### 8.1 Introduction

Curved ray traveltime tomography was originally developed by Bois et al. (1972) for the purpose of estimating the seismic velocity distribution between two boreholes. Following their seminal work, several investigators advanced the technology of tomographic imaging with curved raypaths (Lytle and Dines, 1980; Bishop et al., 1985; McMechan et al., 1987; Bregman et al., 1989; White, 1989). Curved ray methods are necessary for accurately reconstructing the velocity field in highly refractive media. Although straight ray techniques are adequate in media with relatively small velocity variations, it is difficult to decide when the simplifying assumption of straight line raypaths becomes invalid. Hence, there is a compelling reason to use curved ray methods in all situations: they are based on a more accurate model of wave propagation through variable velocity media.

Traveltime tomography is a nonlinear inverse problem that can be solved by local linearization and iteration. Since the velocity field is updated on each cycle of the tomographic imaging procedure, rays have to be retraced between all source-receiver pairs. This raytracing constitutes a large part of the computational cost of curved ray tomographic inversion. Two-point raytracing between a specific source and receiver is an iterative process, and may encounter difficulties due to shadow zones and multipathing. In this study, the problems associated with conventional raytracing are circumvented by calculating traveltimes to all points of a two dimensional slowness field with a rapid finite-difference algorithm (Vidale, 1988). Raypaths are then generated by following the steepest descent direction through the computed traveltimes from each receiver back to the source. This method yields the raypaths of all wave types that comprise first arrivals (body waves, head waves, and diffractions). Moreover, since arrival times are calculated throughout the slowness field, arbitrary recording geometries are easily accommodated.

In addition to rapid and accurate forward modeling, tomographic inversion requires the solution of a system of linear algebraic equations to obtain the improved velocity field. This solution may exhibit erratic and unphysical behavior due to noise in the observed traveltime data and/or ill-conditioning in the equations. Hence, some form of regularization is often used to stabilize the inversion. For example, Lytle and Dines (1980) introduce Laplacian smoothing into the system when calculating a perturbation to a slowness model. Bishop et al. (1985), Bregman et al. (1989), and White (1989) limit the size of the model perturbation by using the damped least squares method. Macrides et al. (1988) impose inequality constraints on a perturbation calculated via an ART algorithm. The present approach is to apply linear equality constraints directly to the slowness model, rather than to a model perturbation, on each iteration of the inversion procedure. In addition to improving the mathematical conditioning of the system, the constraint equations allow the introduction of a priori geological or geophysical knowledge about the model into the inversion. In particular, the constraints may arise from a desire to impose a preferred character, like flatness or smoothness, on the slowness solution. Alternately, one may seek a model that is close, in some quantitative sense, to a prescribed base model. Inclusion of these constraint equations restricts the nonuniqueness that is common in realistic tomographic inverse problems.

The nonlinear tomographic inversion procedure described in this chapter consists of four basic steps:

1) calculation of first arrival traveltimes from each source location to all points of a gridded slowness field,
2) generation of raypaths between all source-receiver pairs,
3) solution of a large and sparse system of linear equations for a perturbation to the existing slowness model,
4) updating and (optionally) smoothing the slowness model.

This four-step process is initiated with an estimate of the true slowness function, and is repeated until an acceptable match is obtained between observed and calculated traveltimes.

The initial estimate is usually a uniform slowness field. Subsequent sections describe these steps in more detail, and demonstrate the inversion procedure using synthetic traveltime data from simulated VSP and crosswell experiments.

### 8.2 General theory and model representation

The traveltime of a seismic wave propagating through a slowness field $s$ is given by the path integral

$$
\begin{equation*}
t(s)=\int_{\Gamma(s)} s d l \tag{8.1}
\end{equation*}
$$

where $\Gamma(s)$ is the raypath connecting source and receiver, and $d l$ is an incremental path length. Since the raypath locus depends on the slowness, the traveltime $t(s)$ is a nonlinear functional of $s$. The problem can be linearized by considering the traveltime difference $\Delta t \equiv t(s+\Delta s)-t(s)$, where $\Delta s$ is a perturbation to the slowness field $s$. For a sufficiently small perturbation, the raypath $\Gamma(s+\Delta s)$ is approximated by the original raypath $\Gamma(s)$. Fermat's Principle then implies that the traveltime through the perturbed slowness field can be evaluated by integration along the unperturbed raypath. The traveltime difference becomes

$$
\begin{equation*}
\Delta t=\int_{\Gamma(s)} \Delta s d l \tag{8.2}
\end{equation*}
$$

Hence, a small difference in traveltime is linearly related to a small difference in slowness.
The two dimensional slowness function $s(x, z)$ is represented by a set of $K$ square cells, each with a uniform slowness value $m_{k}(k=1,2, \ldots, K)$. Thus, within a cell, a raypath is a straight line segment. For a collection of raypaths, equation (8.2) is expressed as the matrix/vector product

$$
\begin{equation*}
\Delta \mathbf{t}=\mathbf{A}(\mathbf{m}) \Delta \mathbf{m} \tag{8.3}
\end{equation*}
$$

where $\mathbf{A}(\mathbf{m})$ is a matrix of raypath length segments within the square cells of the slowness model m . In the tomographic inversion problem, a model perturbation $\Delta \mathrm{m}$ is sought such
that the improved model $m+\Delta m$ approximates the true slowness model that generates the measured traveltime data. Hence, the traveltime difference vector $\Delta t$ is given by

$$
\begin{equation*}
\Delta \mathbf{t}=\mathbf{t}_{o b s}-\mathbf{t}_{p r d}(\mathbf{m}) \tag{8.4}
\end{equation*}
$$

where $t_{o b s}$ is a vector of observed arrival times, and $t_{p r d}(\mathbf{m})$ is a vector of predicted traveltimes computed from the known slowness model $m$. In principle, equation (8.3) can be solved for the required model perturbation. In practice, solution difficulties arise because the raypath matrix is commonly nonsquare, large, sparse, and rank deficient. Moreover, since the observed traveltimes contain random errors, the system (8.3) may be inconsistent. In this case, an exact solution does not exist and a minimum misfit solution is usually sought. Finally, if the initial model $m$ is a poor approximation to the true slowness, several iterations of the model updating procedure may be necessary before the magnitude of the traveltime residual vector $\Delta t$ becomes acceptably small.

### 8.3 Forward modeling

The first arrival times of a seismic wave propagating through a two dimensional velocity structure are computed by Vidale's (1988) finite-difference scheme. This algorithm uses plane wavefront traveltime operators to extrapolate arrival times from point to point throughout a uniformly spaced grid. The method is rapid and accurate, and can be applied to a heterogeneous medium with moderate to strong velocity variations. Podvin and Lecomte (1991) describe improvements to the local traveltime extrapolators that allow models with very strong velocity contrasts to be examined. The traveltimes of all wave types that comprise first arrivals (body waves, head waves, and diffractions) are calculated. Reflections and other later arrivals are not included; this represents a limitation of the technique as currently formulated.

Vidale's method is based on a centered finite-difference solution of the eikonal equation on each square cell of a gridded slowness field. Thus, the associated discretization error is second order in the grid cell size. An input slowness function $s(x, z)$ is sampled on a
uniformly spaced two dimensional grid. If the grid interval is $h$, then the sampled slowness values are given by $s_{i j}=s\left(x_{i}, z_{j}\right)$, where $x_{i}=x_{\min }+(i-1) h$ and $z_{j}=z_{\min }+(j-1) h$ (with $i=1,2, \ldots, I$ and $j=1,2, \ldots, J)$. Since there are $I J$ grid points, there are $K=(I-1)(J-1)$ square cells. The slowness assigned to a particular cell is the arithmetic mean of the slowness values at the four bounding grid points:

$$
\begin{equation*}
m_{k}=\frac{1}{4}\left(s_{i j}+s_{i+1, j}+s_{i, j+1}+s_{i+1, j+1}\right) \tag{8.5}
\end{equation*}
$$

where $k=i+(I-1)(j-1)$. With this indexing scheme, the elements of the vector $m$ constitute a row-ordered sequence of the two dimensional array of cell slowness values. An individual cell is referenced either by the coordinate indices of its upper left corner (ij), or by its sequential index $(k)$.

Calculations are initiated at a designated source point ( $x_{s}, z_{s}$ ) (not necessarily coincident with a grid point) within the slowness model. Since wavefronts are strongly curved in the immediate vicinity of a point source, plane wavefront traveltime extrapolators are inappropriate in this region. Furthermore, near source inaccuracies are propagated to all greater distances. In order to mitigate these effects, the traveltimes in a near source rectangle are calculated via mathematically exact formulae derived from certain simple velocity distributions. Hence, consider the linear velocity function

$$
\begin{equation*}
v(x, z)=v_{s}+a_{x}\left(x-x_{s}\right)+a_{z}\left(z-z_{s}\right) \tag{8.6}
\end{equation*}
$$

where $v_{s}$ is the velocity at the source location, and $a_{x}$ and $a_{z}$ are the horizontal and vertical components of the velocity gradient vector. Numerical values for these constants are obtained by performing a least squares fit to the velocity samples $v_{i j}=1 / s_{i j}$ surrounding the source. The gradient components can be expressed in terms of the magnitude $a$ and direction angle $\phi$ (relative to vertical) of the gradient: $a_{x}=a \sin \phi$ and $a_{z}=a \cos \phi$, where $a=\sqrt{a_{x}^{2}+a_{z}^{2}}$ and $\tan \phi=a_{x} / a_{z}$. If the frame of reference is rotated through the angle $\phi$, then the velocity
field (8.6) is transformed into a one dimensional function. The spatial coordinates in the rotated system are designated by primes and are given by

$$
x^{\prime}=x \cos \phi-z \sin \phi, \quad z^{\prime}=x \sin \phi+z \cos \phi
$$

In the rotated frame, the velocity function becomes $v\left(z^{\prime}\right)=v_{s}+a\left(z^{\prime}-z_{s}^{\prime}\right)$. The arrival time at an arbitrary near source location $\left(x^{\prime}, z^{\prime}\right)$ can then be calculated by standard 1D techniques. The result is

$$
\begin{equation*}
t\left(x^{\prime}, z^{\prime}\right)=\frac{2}{a} \sinh ^{-1} \sqrt{\frac{a^{2}\left[\left(x^{\prime}-x_{s}^{\prime}\right)^{2}+\left(z^{\prime}-z_{s}^{\prime}\right)^{2}\right]}{4 v_{s}\left[v_{s}+a\left(z^{\prime}-z_{s}^{\prime}\right)\right]}} \tag{8.7}
\end{equation*}
$$

Wavefronts associated with this time field are eccentric circles with centers that are displaced along a straight line through the source point, in the direction of the velocity gradient vector. In the limit $a \rightarrow 0$, it can be shown via L'Hopital's rule that equation (8.7) reduces to the proper expression for a uniform velocity field (i.e., circular concentric wavefronts).

The finite-difference algorithm calculates a traveltime $t_{i j}$ at every grid point of the slowness field. If a receiver is not located on a grid node, then an interpolator is needed to estimate the arrival time at the actual receiver position. Simple bilinear interpolation provides adequate accuracy. Hence, if a receiver with coordinates $\left(x_{r}, z_{r}\right)$ is located within cell $i j$, then define the dimensionless quantities $p=\left(x_{r}-x_{i}\right) / h$ and $q=\left(z_{r}-z_{j}\right) / h$. The interpolated traveltime is given by

$$
\begin{equation*}
t\left(x_{r}, z_{r}\right)=(1-p)(1-q) t_{i j}+p(1-q) t_{i+1, j}+q(1-p) t_{i, j+1}+p q t_{i+1, j+1} \tag{8.8}
\end{equation*}
$$

If the four arrival times bounding cell $i j$ are due to local plane wave propagation, then equation (8.8) is an exact expression for the traveltime at the interior point ( $x_{r}, z_{r}$ ). In other cases, the interpolator (8.8) has accuracy $O\left(h^{2}\right)$ (Dahlquist and Björk, 1974, p. 319) and thus is consistent with the level of accuracy associated with the forward modeling scheme.

### 8.4 Ray generation

Raypaths are generated by following the steepest descent direction through a computed traveltime field from each receiver back to the corresponding source point. This strategy was originally suggested by Vidale (1988), and has recently been implemented by Podvin and Lecomte (1991).

The horizontal and vertical components of the traveltime gradient vector within cell $i j$ are approximated by the centered finite-difference formulae

$$
\begin{align*}
\frac{\partial t}{\partial x} & \simeq \frac{\left(t_{i+1, j}+t_{i+1, j+1}\right)-\left(t_{i j}+t_{i, j+1}\right)}{2 h}  \tag{8.9a}\\
\frac{\partial t}{\partial z} & \simeq \frac{\left(t_{i, j+1}+t_{i+1, j+1}\right)-\left(t_{i j}+t_{i+1, j}\right)}{2 h} \tag{8.9b}
\end{align*}
$$

Assignment of a constant traveltime gradient to a cell is compatible with the assumption of locally plane wavefronts used in the forward modeling algorithm. The steepest descent direction is opposite to the gradient direction of the traveltime field. Hence, within cell $i j$ (or $k$ ), the steepest descent direction is defined by the angle

$$
\begin{equation*}
\theta_{k}=\tan ^{-1}\left[\frac{\partial t / \partial z}{\partial t / \partial x}\right]+\pi \quad(\text { modulo } 2 \pi) \tag{8.10}
\end{equation*}
$$

$\theta_{k}$ is measured clockwise from the positive horizontal axis. For a fixed source, all raypaths that cross cell $i j$ have this same orientation angle. The lengths of the raypath segments within the cell range from zero to a maximum of $\sqrt{2} h$.

A representative situation for cell $i j$ is depicted in Figure 8.1. The raypath enters the cell at point A on its right boundary with coordinates ( $x_{a}, z_{a}$ ). Depending on the value of the steepest descent angle assigned to the cell, the ray may exit on any of the remaining three sides or one of the four corners. The logic that selects one of these seven possibilities


Fig. 8.1. Raypath (heavy line) traced through square cell $i j$ of a 2D slowness model. Ray enters cell at point A, follows the local steepest descent direction across the cell, and exits at point B.
is given in the first column of Table 8.1, where $\alpha_{1}$ and $\alpha_{2}$ are positive acute angles defined by

$$
\alpha_{1}=\tan ^{-1}\left(\frac{z_{j+1}-z_{a}}{h}\right), \quad \quad \alpha_{2}=\tan ^{-1}\left(\frac{z_{a}-z_{j}}{h}\right)
$$

These two angles are illustrated in Figure 8.1. After an exit option is selected, the coordinates of the exit point $B$ are easily determined (columns 2 and 3 in Table 8.1). These coordinates $\left(x_{b}, z_{b}\right)$ constitute the entry point coordinates for the next cell that the ray crosses (columns 4 and 5 in Table 8.1). In the particular case displayed, the raypath enters cell $i, j+1$ on its top boundary. Hence, a different logical scheme is required to extend the raypath across this next cell. A total of eight logical tables are necessary to handle all of the possibilities.

Angle Range

$$
\begin{aligned}
& 0<\theta_{k} \leq \pi / 2 \\
& \pi / 2<\theta_{k}<\pi-\alpha_{1} \\
& \theta_{k}=\pi-\alpha_{1} \\
& \pi-\alpha_{1}<\theta_{k}<\pi+\alpha_{2} \\
& \theta_{k}=\pi+\alpha_{2} \\
& \pi+\alpha_{2}<\theta_{k}<3 \pi / 2 \\
& 3 \pi / 2 \leq \theta_{k}<2 \pi
\end{aligned}
$$

Exit Coordinates

| $x_{i+1}$ | $z_{j+1}$ |
| :--- | :--- |
| $x_{a}+\left(z_{j+1}-z_{a}\right) \cot \theta_{k}$ | $z_{j+1}$ |
| $x_{i}$ | $z_{j+1}$ |
| $x_{i}$ | $z_{a}-h \tan \theta_{k}$ |
| $x_{i}$ | $z_{j}$ |
| $x_{a}+\left(z_{j}-z_{a}\right) \cot \theta_{k}$ | $z_{j}$ |
| $x_{i+1}$ | $z_{j}$ |

Next Cell Indices

| $i+1$ | $j+1$ |
| :--- | :--- |
| $i$ | $j+1$ |
| $i-1$ | $j+1$ |
| $i-1$ | $j$ |
| $i-1$ | $j-1$ |
| $i$ | $j-1$ |
| $i+1$ | $j-1$ |

Table 8.1. Ray tracing logic for an entry point on the right side of a square grid cell. Column 1 gives possible ranges for the steepest descent angle $\theta_{k}$ assigned to the cell. Angles $\alpha_{1}$ and $\alpha_{2}$ are defined in the text and illustrated in Fig. 8.1. Columns 2 and 3 give the horizontal and vertical coordinates of the ray exit point, respectively. ( $x_{a}, z_{a}$ ) are the ray entry point coordinates. Columns 4 and 5 give the indices of the next cell that the ray enters.

These correspond to a raypath entering a cell on the right, bottom, left, and top sides, and the upper right, lower right, lower left, and upper left corners.

The length of the raypath segment within cell $i j$ is easily calculated once the entry and exit coordinates are known. Although tomographic inversion requires only the value of the segment length, the actual coordinates $\left(x_{a}, z_{a}\right)$ and $\left(x_{b}, z_{b}\right)$ are also retained in order to create raypath plots for diagnostic purposes.

Ray tracing is initiated at each receiver position. If a receiver is located on a grid node or grid line, then an average of the steepest descent angles from the neighboring cells determines which cell the raypath enters first. Tracing then begins using the logic outlined above. However, if a receiver is positioned within a cell, a separate logical scheme generates the initial path segment to the enclosing cell boundary. The technique is similiar to that described above and is given in Appendix G. Iterative generation of path segments continues until the raypath arrives at the boundary of a defined near-source zone. The size and shape
of this region vary slightly depending on whether the point source resides on a grid node, a grid line, or within a cell (see Appendix G). The final portion of the raypath is then taken to be a straight line from the boundary point directly to the source position ( $x_{s}, z_{s}$ ), regardless of the local values of the steepest descent angle. This ray termination procedure is designed to overcome difficulties associated with nonuniformity of the traveltime gradient vector in close proximity to the source.

Figure 8.2 a is a contour plot of a velocity model bounded by two vertical boreholes. A shallow low velocity anomaly overlies a dipping, higher velocity zone. The first arrival wavefronts from a surface source located between the boreholes are illustrated in Figure 8.2b. The wavefronts are retarded by the low velocity zone and advance more rapidly through the high velocity zone. In addition, Figure 8.2 b displays the raypaths traced through this time field from 18 downhole receivers back to the source. The raypaths are orthogonal to the wavefronts, as expected.

### 8.5 Inversion mathematics

As indicated previously, system (8.3) is typically ill-conditioned and inconsistent. Hence, an undamped least squares solution ( $\left.\Delta \mathrm{m}=\left(\mathbf{A}^{T} \mathbf{A}\right)^{-1} \mathbf{A}^{T} \Delta t\right)$ may yield a model update vector with relatively large and unrealistic cell-to-cell variations in slowness. The conditioning of these equations can be improved by incorporating model constraint information into the inversion. Hence, the system (8.3) is augmented with sets of linear equality constraints on the updated model $m+\Delta m$. In addition to mathematically stabilizing the inversion, these equality constraints allow the convenient introduction of a priori geological or geophysical knowledge (or bias) into the problem.

In general, a set of linear constraint equations applied to the improved model $m+\Delta m$ is written as the matrix/vector multiplication

$$
\begin{equation*}
\mathbf{B}_{n}(\mathbf{m}+\Delta \mathbf{m})=\mathbf{b}_{n} \tag{8.11}
\end{equation*}
$$



Fig. 8.2. (a) Velocity model (contour interval $=100 \mathrm{~m} / \mathrm{s}$ ) with a horizontal low velocity zone and a dipping high velocity zone. Maximum velocity $=2800 \mathrm{~m} / \mathrm{s}$; minimum velocity $=$ $1633 \mathrm{~m} / \mathrm{s}$. (b) Wavefronts (contour interval $=25 \mathrm{~ms}$ ) and raypaths generated by a surface source. The raypaths are traced from 18 downhole receivers back to the surface source.
where the coefficient matrix $\mathbf{B}_{n}$ and the right hand side vector $\mathbf{b}_{n}$ are prescribed. The subscript $n(n=1,2, \ldots, N)$ is used to refer to a particular set of constraints. A perturbation $\Delta \mathrm{m}$ that simultaneously satisfies, in the least squares sense, the linearized data equations and the model constraint equations is sought. Hence, the relevant objective function is

$$
\begin{equation*}
\Phi(\Delta \mathbf{m})=\|\mathbf{A} \Delta \mathbf{m}-\Delta \mathrm{t}\|^{2}+\sum_{n=1}^{N} \mu_{n}\left\|\mathbf{B}_{n}(\mathbf{m}+\Delta \mathrm{m})-\mathbf{b}_{n}\right\|^{2} \tag{8.12}
\end{equation*}
$$

where the scalars $\mu_{n}\left(0<\mu_{n}<+\infty\right)$ are adjustable tradeoff parameters that control the relative importance of the various terms. Extremizing $\Phi$ with respect to $\Delta \mathrm{m}$ yields the linear algebraic equations

$$
\begin{equation*}
\left[\mathbf{A}^{T} \mathbf{A}+\sum_{n=1}^{N} \mu_{n} \mathbf{B}_{n}^{T} \mathbf{B}_{n}\right] \Delta \mathbf{m}=\mathbf{A}^{T} \Delta \mathbf{t}+\sum_{n=1}^{N} \mu_{n} \mathbf{B}_{n}^{T}\left(\mathbf{b}_{n}-\mathbf{B}_{n} \mathbf{m}\right) \tag{8.13}
\end{equation*}
$$

For nonzero $\mu_{n}$, the coefficient matrix in this expression is usually nonsingular. The required model perturbation $\Delta \mathrm{m}$ can be obtained by solving (8.13) using standard techniques of numerical linear algebra. However, the coefficient matrix is large ( $K \times K$, where $K$ is the number of slowness cells) and may be dense even if the original raypath and constraint matrices are sparse. Hence, it is advantageous to seek a solution method that avoids explicitly forming the square matrices $\mathbf{A}^{T} \mathbf{A}$ and $\mathbf{B}_{n}^{T} \mathbf{B}_{n}$.

It is straightforward to demonstrate that (8.13) are the normal equations associated with the least squares solution of the rectangular system

$$
\left[\begin{array}{c}
\mathbf{A}  \tag{8.14}\\
\sqrt{\mu_{1}} \mathbf{B}_{1} \\
\vdots \\
\sqrt{\mu_{N}} \mathbf{B}_{N}
\end{array}\right] \Delta \mathbf{m}=\left[\begin{array}{c}
\Delta \mathbf{t} \\
\sqrt{\mu_{1}}\left(\mathbf{b}_{1}-\mathbf{B}_{1} \mathbf{m}\right) \\
\vdots \\
\sqrt{\mu_{N}}\left(\mathbf{b}_{N}-\mathbf{B}_{N} \mathbf{m}\right)
\end{array}\right]
$$

Algorithm LSQR (Paige and Saunders, 1982) is used to solve equation (8.14) directly for $\Delta m . \operatorname{LSQR}$ is an iterative solution technique for large and sparse systems of linear equations that is closely related to the conjugate gradient method. It is designed to seek the minimum norm least squares solution of a set of equations. Numerical studies of LSQR applied to the tomographic inversion problem indicate that it is both rapid and accurate (Nolet, 1985; Scales, 1987). A simple FORTRAN version of the LSQR algorithm is given by Nolet (1987).

The dimensions of the coefficient matrix in (8.14) are $\left(N_{\text {data }}+N_{\text {cons }}\right) \times K$, where $N_{\text {data }}$ and $N_{\text {cons }}$ are the number of data and constraint equations, respectively. Since this matrix may be large and sparse, a significant reduction in storage space is achieved by storing only the nonzero elements in a one dimensional array. The full index scheme described by Scales (1987) is used here to store and address the matrix elements. With this storage method, the sparse matrix/vector multiplications required by the LSQR algorithm are particularly simple to implement.

Finally, the correction to the slowness value at grid point $i j$ is determined by averaging the slowness perturbations calculated for the four surrounding cells:

$$
\begin{equation*}
\Delta s_{i j}=\frac{1}{4}\left(\Delta m_{k-I}+\Delta m_{k-I+1}+\Delta m_{k-1}+\Delta m_{k}\right) \tag{8.15}
\end{equation*}
$$

where $k=i+(I-1)(j-1)$. Grid points located on the edges (corners) of the slowness model are updated by adding the average of the perturbations associated with the neighboring two (one) cells. After all grid points are updated, forward modeling of traveltimes for the next iteration of the inversion can proceed.

Several investigators apply a spatial filter to the slowness field between tomographic iterations in order to suppress short wavelength variations in the computed values (Dines and Lytle, 1979; Radcliffe et al., 1984; Gersztenkorn and Scales, 1988). These variations may arise from noise in the traveltime data and/or instabilities in the inversion. Smoothing then serves to condition the slowness field for the next forward modeling step. Hence, an optional 9 -point square smoother with prescribed weights $w_{l}$ is included here. The smoother
is applied to the gridded values. The smoothed slowness value assigned to grid point $i j$ is given by

$$
\begin{align*}
\hat{s}_{i j}= & w_{1} s_{i-1, j-1}+w_{2} s_{i, j-1}+w_{3} s_{i+1, j-1} \\
& w_{4} s_{i-1, j}+w_{5} s_{i j}+w_{6} s_{i+1, j} \\
& w_{7} s_{i-1, j+1}+w_{8} s_{i, j+1}+w_{9} s_{i+1, j+1} \tag{8.16}
\end{align*}
$$

where $\sum w_{l}=1$. Grid points residing on the edges of the model are smoothed by conceptually extending the grid by one point with the local values. In the examples to follow, the filter weights are $w_{5}=4 / 12$ and all other $w_{l}=1 / 12$.

### 8.6 Model constraint equations

There is a wide variety of linear equality constraints that can be employed in the tomographic inversion problem. The particular type of constraint used in this study is characterized by a two dimensional 'operator' or 'filter' with five specified constants $c_{1}, c_{2}, c_{3}, c_{4}$, and $c_{5}$. Figure 8.3 depicts the application of this operator to interior cell $i j$ of a slowness model. Using the row-ordered indexing scheme, the $k^{\text {th }}$ component of the filtered image $\mathbf{B m}$ is given by

$$
c_{1} m_{k+1}+c_{2} m_{k+I-1}+c_{3} m_{k-1}+c_{4} m_{k-I+1}+c_{5} m_{k}
$$

Edge cells are handled by conceptually extending the slowness model beyond the defined region with the local cell slowness values. For example, application of the 5 -point operator to the upper left corner cell ( $i=1, j=1$ corresponding to $k=1$ ) of the model m is via the formula

$$
c_{1} m_{2}+c_{2} m_{I}+\left(c_{3}+c_{4}+c_{5}\right) m_{1}
$$

One such constraint is applied to each cell of the slowness model. Development of the matrix representation $\mathbf{B}$ for this set of equations is merely a matter of proper row and column


Fig. 8.3. Schematic representation of the 5 -point constraint operator applied to cell $i j$ of a slowness model. The slowness values in five neighboring cells are multiplied by constants $c_{1}$ through $c_{5}$ and the products are summed.
indexing. In general, the components of the right hand side vector $b$ may differ, implying spatial variation in the value of the applied model constraints.

The 5-point operator is an extremely simple and flexible mechanism for introducing various types of constraint information into tomographic imaging. Some examples include:

Case 1: $c_{5}=1$ and all other $c_{l}=0$. An individual cell slowness is not constrained by its immediate neighbors. Rather, inversion produces a model that is closest, in the least squares sense, to a given target model $\mathbf{b}$.

Case 2: $c_{1}=1 / 2 h, c_{3}=-1 / 2 h$ and all other $c_{l}=0$. This operator is a centered finitedifference approximation to the horizontal derivative of the slowness field. A similar approximation to the vertical derivative is achieved by setting $c_{2}=1 / 2 h, c_{4}=-1 / 2 h$
and all other $c_{l}=0$. These operators introduce first-difference regularization, or flattening, into the inversion.

Case 3: $c_{1}=1 / h^{2}, c_{5}=-2 / h^{2}, c_{3}=1 / h^{2}$ and all other $c_{l}=0$. This operator, as well as its vertical counterpart, introduces second-difference regularization, or smoothing, into the inversion.

Case 4: $c_{1}=c_{2}=c_{3}=c_{4}=1 / h^{2}$ and $c_{5}=-4 / h^{2}$. This operator incorporates Laplacian smoothing into the inversion.

These constraints are inherently local in character; they consist of linear relations between adjacent cell slowness values. Obviously, an operator with larger spatial extent (say, a 9point square filter) would provide more options. However, the current 5-point operator offers a reasonable compromise among the competing issues of flexibility, accuracy, and ease of implementation.

Combinations of constraints applied by the simple 5-point operator can also be considered. For example, the next section illustrates situations where both horizontal and vertical first-difference regularization is applied (i.e., $N=2$ ). The first system of constraint equations corresponds to $c_{1}=1 / 2 h$ and $c_{3}=-1 / 2 h$, while the second set corresponds to $c_{2}=1 / 2 h$ and $c_{4}=-1 / 2 h$. In each case, the right hand side vector $b_{n}$ is set equal to 0 .

### 8.7 Synthetic examples

The examples presented in this section demonstrate the ability of the tomographic inversion procedure to image a smoothly varying velocity field. Figure 8.2a displays the 500 m $\times 500 \mathrm{~m}$ velocity model used to generate synthetic traveltime data. Since the grid interval is $h=5 \mathrm{~m}$, there are $I J=10201$ grid points used for the forward modeling and $K=10000$ square cells used in the inversion.

The data acquisition geometry used for the first example simulates a double-well VSP plus crosswell experiment. Nine surface sources, located between two vertical boreholes, are spaced 50 m apart. Traveltimes are recorded by 18 borehole receivers ( 9 per well; 50 m
separation) from each source position. In addition, the 9 downhole receivers in the right well record traveltimes from 9 sources symmetrically placed in the left well. The complete set of 243 raypaths linking all source-receiver pairs is illustrated in Figure 8.4. Note that these first arrival raypaths tend to avoid the low velocity zone, resulting in a region of reduced ray coverage. Furthermore, no raypaths penetrate below 450 m depth.

Contoured velocity tomograms obtained by inverting the combined VSP and crosswell traveltimes are displayed in Figure 8.5. As indicated above, both horizontal and vertical first-difference constraints are imposed in the iterative inversions. Additionally, in Figure 8.5 b , the slowness values of the cells adjacent to the two boreholes are constrained by the true slowness function. In an actual field experiment, this information may be available from borehole velocity logs. Each of these reconstructions produces an rms traveltime error of $\sim 0.5 \mathrm{~ms}$, which is about $0.25 \%$ of the rms value of the synthetic traveltimes ( 200.7 ms ).

The main features of the true velocity model are recovered by both of the inversions depicted in Figure 8.5. The location and amplitude of the dipping high velocity anomaly are approximately correct. Also, the shallow low velocity zone has been detected and correctly positioned, although the actual velocity value at its center is about $90 \mathrm{~m} / \mathrm{s}$ too high. This effect is associated with the reduced raypath density in this area of the model. Interestingly, inclusion of the borehole velocity constraints does not yield a dramatic improvement. The principal difference between the two reconstructions appears in the region below 400 m depth, where raypath coverage is negligible or nonexistent.

Six (Figure 8.5a) and nine (Figure 8.5b) iterations are required to reduce the initial rms traveltime misfit to $\sim 0.5 \mathrm{~ms}$, with most of the improvement actually occurring on the first iteration. As stated above, the conjugate gradient solver LSQR is also an iterative algorithm. Theoretically, LSQR requires at most $K$ iterations to converge to the solution of a system with $K$ unknowns (assuming exact arithmetic can be performed). For the tomographic inversions described in this section, an acceptable solution is obtained with about 10 iterations in LSQR, which is three orders of magnitude less than the theoretical value $K=10000$. This results in an appreciable saving in computation time.


Fig. 8.4. 243 raypaths traced through the velocity field of Fig. 8.2a. These rays link all source-receiver pairs of a combined double-well VSP and crosswell experiment.


Fig. 8.5. Reconstructed velocity models (contour interval $=100 \mathrm{~m} / \mathrm{s}$ ) obtained by inverting the combined VSP and crosswell traveltimes. (a) No borehole velocity constraints applied. Max velocity $=2819 \mathrm{~m} / \mathrm{s}$; min velocity $=1724 \mathrm{~m} / \mathrm{s}$. (b) Velocity constraints imposed at boreholes. Max velocity $=2801 \mathrm{~m} / \mathrm{s} ; \min$ velocity $=1724 \mathrm{~m} / \mathrm{s}$.

Figure 8.6 illustrates that the tomographic inversion is stable when the synthetic traveltimes are contaminated with small amounts of random noise. Random numbers drawn from a uniform probability distribution on $\pm 4 \mathrm{~ms}$ are added to the exact traveltimes. The iterative inversion is initiated with the same constant slowness model used for the previous example. The convergence criterion for terminating iterations is arbitrarily selected to be 2.5 ms of rms traveltime error, or approximately one standard deviation of the noise. No borehole constraints are imposed. An accurate velocity reconstruction results when the weights of the flattening constraints ( $\sqrt{\mu_{1}}$ and $\sqrt{\mu_{2}}$ in equation (8.14)) are set sufficiently high.


Fig. 8.6. Reconstructed velocity model (contour interval $=100 \mathrm{~m} / \mathrm{s}$ ) obtained by inverting noise contaminated VSP and crosswell traveltimes. Max velocity $=2810 \mathrm{~m} / \mathrm{s}$; min velocity $=1731 \mathrm{~m} / \mathrm{s}$.

The final example examines the ability of the constrained inversion algorithm to reconstruct the interwell velocity field when only crosshole traveltimes are available. Figure 8.7 displays the 81 crosswell raypaths. The zone of reduced ray coverage at shallow depths is much more extensive. Images obtained by inverting error free traveltimes are illustrated in Figure 8.8. Once again, both horizontal and vertical flattening constraints are applied. The image in Figure 8.8a displays a shallow low velocity zone and a deeper, dipping high velocity zone. However, the peak of the high velocity anomaly is shifted downdip by a significant distance. An inversion including borehole velocity constraints is displayed in Figure 8.8b. Surprisingly, the reconstruction is not improved; rather, a spurious region with high velocity gradients develops near the right well.

The tomographic inversions in this section are all initiated with uniform slowness models calculated via the method described in Appendix H. Similar results are obtained from a range of nearby slowness values. Alternately, a nonuniform starting model obtained by horizontal interpolation of the borehole slowness values can be be used. For the examples considered here, this latter technique does not yield any obvious improvement in the velocity reconstructions.

### 8.8 Conclusion

Finite-difference traveltime computation offers an attractive alternative to conventional raytracing for tomographic inversion purposes. The method is sufficiently rapid and accurate, and handles all of the various wave types that constitute first arrivals. Moreover, since traveltimes are computed throughout a slowness model, very general recording geometries are easily accommodated. The main limitation of the technique is that it is restricted to first arriving waves. Hence, the present formulation cannot be applied to the reflection tomography problem. However, current efforts to generalize finite-difference computation methods to reflection traveltimes (e.g., Podvin and Lecomte, 1991) are encouraging.

The introduction of constraint information into traveltime tomography is a responsible way to address the nonuniqueness inherent in this inverse problem. Constraining information may arise from known geological or geophysical properties of the subsurface velocity model


Fig. 8.7. 81 crosswell raypaths traced through the velocity model of Fig. 8.2a. 9 sources are in left well and 9 receivers are in right well.


Fig. 8.8. Reconstructed velocity models (contour interval $=100 \mathrm{~m} / \mathrm{s}$ ) obtained by inverting only the crosswell traveltimes. (a) No borehole velocity constraints applied. Max velocity $=2811 \mathrm{~m} / \mathrm{s} ; \min$ velocity $=1792 \mathrm{~m} / \mathrm{s}$. (b) Velocity constraints imposed at boreholes. Max velocity $=2862 \mathrm{~m} / \mathrm{s} ; \min$ velocity $=1728 \mathrm{~m} / \mathrm{s}$.
(say, from outcrops or borehole logs). Alternately, constraints may derive from a desire to impose certain reasonable attributes, like flatness or smoothness, on the constructed model. The method of incorporating constraints into the mathematical inversion procedure is adaptable to either viewpoint. Linear equality constraints are applied directly to the constructed model, rather than to a model perturbation, and are satisfied in the least squares sense. The examples illustrate the imposition of flattening constraints and known borehole information in the reconstruction of a smoothly varying interwell velocity field. Inclusion of this constraint information allows the solution of a problem that is strongly underdetermined (243 data and 10000 unknowns). Nevertheless, a superior result is not necessarily achieved by the addition of the borehole velocity constraints.

## CHAPTER 9

## SUMMARY

This thesis contributes to knowledge in three specific areas of seismic refraction traveltime analysis:
(1) Forward modeling and inversion of head wave arrival times is extended to three dimensional layered earth models in Chapters 2, 3, and 4. This class of models is characterized by uniform velocity layers bounded by plane interfaces with arbitrary strike and dip. Computational procedures are developed for both single-layer and multilayered models. In the single-layer case, closed form mathematical solutions to the forward and inverse problems exist. A rigorous derivation of a traveltime equation for critically refracted waves propagating within a multilayered model is also given. The resulting expression is not strictly 'closed form'; it requires a minimal amount of raytracing to evaluate numerically. This formula forms the basis of an iterative head wave traveltime inversion algorithm designed to recover the parameters defining a single-layer or multilayered earth model. Inclusion of constraint information in the procedure, in the form of inequality relations satisfied by the model parameters, often governs the ability of the algorithm to converge to a realistic solution. Tests with simulated and field data, acquired in various recording geometries, indicate that the single-layer version of the algorithm is reasonably robust. However, the multilayered inversion algorithm appears to require fairly restrictive constraints in order to operate effectively.
(2) Improvements to various two dimensional refraction traveltime inversion methods that allow for nonplane interfaces and/or variable velocity media are developed in Chapters 5, 6, and 7. Two existing techniques (the generalized reciprocal method and the wavefront method) are extensively analyzed and a new method (critical offset refraction profiling) is proposed. This new technique is an improvement over the GRM in that it explicitly incorporates undulating interfaces and horizontally varying refractor velocity into the model. Moreover, point values of interface depth, interface dip, and refractor velocity are obtained from the observed arrival times. Hence, a depth profile of the critically refracting horizon can
be constructed via interpolation techniques. A rational procedure for calculating a smooth depth profile, based on methods of linear inverse theory, is presented. This latter technique is not limited for use solely with the critical offset refraction profiling method, but may augment any traveltime analysis procedure that yields point depth or dip estimates of the subsurface interface (such as the modified GRM developed here). Finally, an automated implementation of the classical wavefront method for interpreting refraction arrival times indicates that an image of an interface can be obtained directly from the picked times, with no intermediate computational steps. A finite-difference propagation algorithm is used to downward continue the observed times through a heterogeneous near-surface velocity structure. A simple imaging condition involving the reciprocal time then defines the locus of the subsurface refracting horizon. Tests with synthetic data indicate that both anticlinal and synclinal structures can be imaged accurately. Shallow refraction data acquired at an archeological site is also used to assess the workability of the algorithm.
(3) Application of finite-difference traveltime computation methods to the two dimensional tomographic inverse problem is treated in Chapter 8. An iterative algorithm is developed for reconstructing a P -wave velocity field from measured first arrival times. Rapid and accurate forward modeling of all first arrival types (direct waves, head waves, and diffractions) for arbitrary source-receiver geometries is achieved with a finite-difference algorithm. Curved raypaths, needed for converting traveltime residuals into localized updates to the slowness model, are generated by following the steepest descent direction through the computed traveltime field from each receiver back to the source. Incorporation of constraint information into the procedure, in the form of horizontal and vertical first difference regularization, serves to stabilize the inversion and drive the solution toward a model with a preferred character (i.e., a 'flat' slowness model). Tests with simulated vertical seismic profile and crosswell arrival times indicate that the algorithm can successfully reconstruct a smoothly varying interwell velocity field, even when the problem is severely underdetermined.

These contributions can be applied to the solution of practical problems in seismic refraction exploration, for various objectives and at many different scales. The stage is
now set for generalizing the techniques to more complicated three dimensional models with nonplane interfaces and/or nonuniform velocities.

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## APPENDICES

## Appendix A: Conic sections in polar coordinates

Expressions for conic sections in plane polar coordinates can be found in most textbooks on analytic geometry. However, these formulae typically assume that the coordinate origin is located either at the center of symmetry or at a focus of the curve. In this appendix, an alternate representation of the ellipse or hyperbola in polar coordinates is derived. The principle axes of the conic are rotated by an arbitrary angle $\theta$ with respect to the coordinate axes, and the origin resides at any point along the major axis of symmetry.

In Figure A1, the primed coordinate system is obtained by a clockwise rotation of the unprimed system through the angle $\theta$. The coordinates of a given point relative to each system are related via

$$
\begin{equation*}
x^{\prime}=x \cos \theta+y \sin \theta, \quad y^{\prime}=-x \sin \theta+y \cos \theta \tag{A1}
\end{equation*}
$$

Consider an ellipse with center located on the $x^{\prime}$ axis at $\left(x^{\prime}, y^{\prime}\right)=\left(l_{c}, 0\right)$. The semimajor axis $a$ and the semiminor axis $b$ are parallel to the $x^{\prime}$ and $y^{\prime}$ coordinate axes, respectively. Figure A1 depicts the case where $a<\left|l_{c}\right|$. However, the situation $a>\left|l_{c}\right|$ is also allowed; in the case, the ellipse encompasses the origin. In the primed reference frame, the equation of the elliptical locus is

$$
\begin{equation*}
\left(\frac{x^{\prime}-l_{c}}{a}\right)^{2}+\left(\frac{y^{\prime}}{b}\right)^{2}=1 \tag{A2}
\end{equation*}
$$

Substituting from relations (A1) yields

$$
\left(\frac{x \cos \theta+y \sin \theta-l_{c}}{a}\right)^{2}+\left(\frac{y \cos \theta-x \sin \theta}{b}\right)^{2}=1
$$

However $b^{2}=a^{2}\left(1-e^{2}\right)$ where $e$ is the ellipse eccentricity $(0 \leq e<1)$. Thus

$$
\left(1-e^{2}\right)\left(x \cos \theta+y \sin \theta-l_{c}\right)^{2}+(y \cos \theta-x \sin \theta)^{2}=a^{2}\left(1-e^{2}\right)
$$

This expression is converted to plane polar coordinates $(X, \alpha)$ via the substitutions $x=$ $X \cos \alpha$ and $y=X \sin \alpha$. After some algebraic reduction, a quadratic form in the radial coordinate $X$ is obtained:

$$
\begin{equation*}
X^{2}\left[1-e^{2} \cos ^{2}(\alpha-\theta)\right]-2 X\left[\left(1-e^{2}\right) l_{c} \cos (\alpha-\theta)\right]+\left[\left(1-e^{2}\right)\left(l_{c}^{2}-a^{2}\right)\right]=0 \tag{A3}
\end{equation*}
$$

The quadratic formula can be used to solve equation (A3) for $X$ as a function of the azimuthal angle $\alpha$. However, this is algebraically tedious as well as unecessary for present purposes. Note that the conventional expressions for an ellipse in polar coordinates are obtained as special cases of equation (A3). Thus, setting $l_{c}=0$ (i.e., origin coincident with the ellipse center) immediately yields

$$
X(\alpha)=a \sqrt{\frac{1-e^{2}}{1-e^{2} \cos ^{2}(\alpha-\theta)}}
$$

Also, if $l_{c}=a e$, then the origin is located at a focus of the ellipse. In this case (A3) gives

$$
X(\alpha)=a \frac{1-e^{2}}{1-e \cos (\alpha-\theta)}
$$

This form is common in mechanics; it describes the trajectory of a particle moving under the influence of an inverse square central force.

A similiar analysis reveals that equation (A3) also applies to a hyperbola with the same center location and principle axis orientation. The only difference is that the numerical value of the eccentricity exceeds one $(e>1)$ for a hyperbola.


Fig. A1. A noncentered, rotated ellipse.

## Appendix B: Traveltime analysis for a simple 3D model

A compact formula for head wave traveltime in a simple three dimensional earth model is derived in Chapter 2 (equation (2.10)). The purpose of this appendix is to verify that the more general expression (3.12) reduces to this known result in the special case of a two media model with a horizontal surface. Evaluating equation (3.12) with $n_{1, x}=n_{1, y}=0, n_{1, z}=1$, and with the interface index $k=2$ gives

$$
\begin{gather*}
T_{2}\left(x_{S}, y_{S}, X, \Psi\right)=\left[\frac{\cos \Psi q_{12, x}+\sin \Psi q_{12, y}}{v_{1}}\right] X \\
+\frac{h_{1}(0,0)\left(p_{12, z}-q_{12, z}\right)}{v_{1}}-\frac{x_{S}\left(p_{12, x}-q_{12, x}\right)}{v_{1}}-\frac{y_{S}\left(p_{12, y}-q_{12, y}\right)}{v_{1}} \tag{B1}
\end{gather*}
$$

Expressions for the $x y z$ components of the unit propagation vectors are now determined.
As indicated in Chapter 2, the entire head wave raypath for this situation (depicted in Figure 2.2) is confined to a single plane. The two vectors $\mathbf{n}_{\mathbf{2}}$ and $\mathbf{p}_{22}$ form an orthonormal basis for all vectors in this plane. Hence, the propagation vectors $\mathbf{p}_{12}$ and $\mathbf{q}_{12}$ can be resolved along this basis. Geometric analysis of Figure 2.2 yields

$$
\begin{equation*}
\mathbf{p}_{12}=\left(\cos i_{c}\right) \mathbf{n}_{2}+\left(\sin i_{c}\right) \mathbf{p}_{22}, \quad \quad \mathbf{q}_{12}=\left(-\cos i_{c}\right) \mathbf{n}_{2}+\left(\sin i_{c}\right) \mathbf{p}_{22} \tag{B2}
\end{equation*}
$$

where $i_{c}$ is the critical angle. The unit normal $n_{2}$ to the refracting interface is given by

$$
\begin{equation*}
\mathbf{n}_{2}=\left(\sin \phi_{2} \cos \theta_{2}\right) \mathbf{i}+\left(\sin \phi_{2} \sin \theta_{2}\right) \mathbf{j}+\left(\cos \phi_{2}\right) \mathbf{k} \tag{B3}
\end{equation*}
$$

An expression for the critically refracted propagation direction vector $p_{22}$ is obtained by recognizing that

$$
\mathbf{p}_{22}=\frac{\mathbf{r}_{Q}-\mathbf{r}_{P}}{\left\|\mathbf{r}_{Q}-\mathbf{r}_{P}\right\|}
$$

where $\mathbf{r}_{P}$ and $\mathbf{r}_{Q}$ are the position vectors of points $P$ and $Q$ in Figure 2.2. Thus

$$
\begin{aligned}
\mathbf{r}_{Q}-\mathbf{r}_{P} & =\left(\mathbf{r}_{R}+d_{R} \mathbf{n}_{2}\right)-\left(\mathbf{r}_{S}+d_{S} \mathbf{n}_{2}\right) \\
& =\left(x_{R}-x_{S}\right) \mathbf{i}+\left(y_{R}-y_{S}\right) \mathbf{j}+\left(d_{R}-d_{S}\right) \mathbf{n}_{2} \\
& =(X \cos \Psi) \mathbf{i}+(X \sin \Psi) \mathbf{j}+(-X \sin \delta) \mathbf{n}_{2}
\end{aligned}
$$

where equation (2.7) has been used and the angle $\delta$ is defined by $\sin \delta=\sin \phi_{2} \cos \left(\Psi-\theta_{2}\right)$. Also, equation (2.9) implies $\left\|\mathbf{r}_{Q}-\mathbf{r}_{P}\right\|=X \cos \delta$. Hence

$$
\begin{equation*}
\mathbf{p}_{22}=\frac{(\cos \Psi) \mathbf{i}+(\sin \Psi) \mathbf{j}-(\sin \delta) \mathbf{n}_{2}}{\cos \delta} \tag{B4}
\end{equation*}
$$

Substituting these expressions for $\mathbf{n}_{2}$ and $\mathbf{p}_{22}$ into equations (B2) yields the rather formidable formulae

$$
\begin{align*}
& \mathbf{p}_{12}=\left[\frac{\sin \phi_{2} \cos \theta_{2} \cos \left(i_{c}+\delta\right)+\sin i_{c} \cos \Psi}{\cos \delta}\right] \mathbf{i} \\
& +\left[\frac{\sin \phi_{2} \sin \theta_{2} \cos \left(i_{c}+\delta\right)+\sin i_{c} \sin \Psi}{\cos \delta}\right] \mathbf{j}+\left[\frac{\cos \phi_{2} \cos \left(i_{c}+\delta\right)}{\cos \delta}\right] \mathbf{k} \tag{B5}
\end{align*}
$$

and

$$
\begin{align*}
& \mathbf{q}_{12}=\left[\frac{-\sin \phi_{2} \cos \theta_{2} \cos \left(i_{c}-\delta\right)+\sin i_{c} \cos \Psi}{\cos \delta}\right] \mathbf{i} \\
& +\left[\frac{-\sin \phi_{2} \sin \theta_{2} \cos \left(i_{c}-\delta\right)+\sin i_{c} \sin \Psi}{\cos \delta}\right] \mathbf{j}+\left[\frac{-\cos \phi_{2} \cos \left(i_{c}-\delta\right)}{\cos \delta}\right] \mathbf{k} \tag{B6}
\end{align*}
$$

It is straightforward to verify that $\left\|\mathbf{p}_{12}\right\|=\left\|\mathbf{q}_{12}\right\|=1, \mathbf{p}_{12} \cdot \mathbf{n}_{2}=\cos i_{c}$, and $\mathbf{q}_{12} \cdot \mathbf{n}_{2}=$ $-\cos i_{c}$, as is required. Moreover, equations (B5) and (B6) imply that

$$
\begin{array}{lr}
p_{12, x}-q_{12, x}=2 \cos i_{c} \sin \phi_{2} \cos \theta_{2}, & p_{12, y}-q_{12, y}=2 \cos i_{c} \sin \phi_{2} \sin \theta_{2}, \\
p_{12, z}-q_{12, z}=2 \cos i_{c} \cos \phi_{2}, & \cos \Psi q_{12, x}+\sin \Psi q_{12, y}=\sin \left(i_{c}-\delta\right) .
\end{array}
$$

These relations are substituted into the traveltime expression (B1) to obtain the final result

$$
\begin{equation*}
T_{2}\left(x_{S}, y_{S}, X, \Psi\right)=\frac{\sin \left(i_{c}-\delta\right)}{v_{1}} X+\frac{2 d_{S} \cos i_{c}}{v_{1}} \tag{B7}
\end{equation*}
$$

where $d_{S}$ is the perpendicular distance from the source $S$ to the refracting interface:

$$
d_{S}=h_{1}(0,0) \cos \phi_{2}-\sin \phi_{2}\left(x_{S} \cos \theta_{2}+y_{S} \sin \theta_{2}\right)
$$

Except for some minor notational changes, equation (B7) agrees exactly with the traveltime formula (2.10) previously developed for this simple model situation.

## Appendix C: Traveltime analysis for a simple 2D model

A useful check on the validity of the general head wave traveltime formulae in Chapter 3 is provided by analyzing a specific situation for which a closed form traveltime solution exists. This is especially important because several of these expressions disagree with analogous formulae published recently by Diebold (1987). The inconsistency between the two results is evident when source and receiver are located on separate interfaces with different dip angles. In this case, it is not possible to effect a coordinate frame rotation such that both interfaces become horizontal (i.e., parallel to the $x y$ plane).

Consider the simple two dimensional earth model depicted in Figure C1. Two subsurface interfaces have the same dip angle $\varphi$. The source $S$ is located at the coordinate origin on the surface and the receiver $R$ is located on subsurface interface 2. $L$ is the source-receiver range measured parallel to the dipping interfaces. The perpendicular distances from the origin to the subsurface interfaces are $d_{2}$ and $d_{3}$; vertical layer thicknesses at the same point are $h_{1}$ and $h_{2}$.

The traveltime of a head wave formed on interface 3 can be derived from first principles of refraction traveltime analysis. Rotating the perspective through the small angle $\varphi$ transforms the situation into a one dimensional problem. The total traveltime is easily obtained by summing the refraction delay time associated with each layer that the wave traverses, together with $L / v_{3}$. The result is

$$
\begin{equation*}
T_{3}=\frac{L}{v_{3}}+\frac{d_{2} \cos \theta_{13}}{v_{1}}+\frac{2\left(d_{3}-d_{2}\right) \cos \theta_{23}}{v_{2}} \tag{C1}
\end{equation*}
$$

where $\sin \theta_{i j}=v_{i} / v_{j}$. The following geometric relationships are evident from Figure C 1 :

$$
d_{2}=h_{1} \cos \varphi, \quad d_{3}-d_{2}=h_{2} \cos \varphi, \quad L=\frac{x_{R}}{\cos \varphi}+d_{2} \tan \varphi
$$



Fig. C1. A simple 2D earth model with three media. $S$ is a surface source and $R$ is a buried receiver.
where $x_{R}$ is the horizontal coordinate of the receiver. Substituting these expressions into equation (C1) yields

$$
\begin{equation*}
T_{3}\left(x_{R}\right)=\frac{x_{R}}{v_{3} \cos \varphi}+\frac{h_{1} \cos \left(\theta_{13}-\varphi\right)}{v_{1}}+\frac{2 h_{2} \cos \theta_{23} \cos \varphi}{v_{2}} \tag{C2}
\end{equation*}
$$

The traveltime predicted by the general formula (3.16) is now compared with this specific traveltime solution. For the situation being examined, (3.16) simplifies considerably. The following conditions hold:
(i) $x_{S}=y_{S}=y_{R}=0$.
(ii) $j=1, \quad l=2, \quad k=3$.
(iii) $h_{1}(0,0)=h_{1}, \quad h_{2}(0,0)=h_{2}$.
(iv) $n_{2, x}=-\sin \varphi, \quad n_{2, z}=\cos \varphi$.

Evaluating formula (3.16) with these parameters yields

$$
\begin{equation*}
T_{3}\left(x_{R}\right)=x_{R}\left[\frac{\cos \varphi q_{23, x}+\sin \varphi q_{23, z}}{v_{2} \cos \varphi}\right]+\frac{h_{1} p_{13, z}}{v_{1}}+\frac{h_{2}\left(p_{23, z}-q_{23, z}\right)}{v_{2}} \tag{C3}
\end{equation*}
$$

Since the earth model is strictly two dimensional, and the recording profile is oriented normal to the common strike direction of the interfaces, the unit propagation direction vectors are contained entirely within the $x z$ plane. Thus, the components of these vectors can be obtained by further geometric analysis of Figure C1:

$$
\begin{array}{ll}
p_{13, z}=\cos \left(\theta_{13}-\varphi\right), & p_{23, z}=\cos \left(\theta_{23}-\varphi\right) \\
q_{23, x}=\sin \left(\theta_{23}+\varphi\right), & q_{23, z}=-\cos \left(\theta_{23}+\varphi\right)
\end{array}
$$

Substituting these expressions into equation (C3) gives

$$
T_{3}\left(x_{R}\right)=\frac{x_{R} \sin \theta_{23}}{v_{2} \cos \varphi}+\frac{h_{1} \cos \left(\theta_{13}-\varphi\right)}{v_{1}}+\frac{2 h_{2} \cos \theta_{23} \cos \varphi}{v_{2}} .
$$

But since $\sin \theta_{23}=v_{2} / v_{3}$, this equation reduces immediately to

$$
\begin{equation*}
T_{3}\left(x_{R}\right)=\frac{x_{R}}{v_{3} \cos \varphi}+\frac{h_{1} \cos \left(\theta_{13}-\varphi\right)}{v_{1}}+\frac{2 h_{2} \cos \theta_{23} \cos \varphi}{v_{2}} \tag{C4}
\end{equation*}
$$

This is in complete agreement with the known solution (C2)! In contrast, equation (21) in Diebold (1987), when applied to the model depicted in Figure C1, becomes

$$
\begin{equation*}
T_{3}\left(x_{R}\right)=\frac{x_{R} \sin \left(\theta_{23}+\varphi\right)}{v_{2}}+\frac{h_{1} \cos \left(\theta_{13}-\varphi\right)}{v_{1}}+\frac{2 h_{2} \cos \theta_{23} \cos \varphi}{v_{2}} \tag{C5}
\end{equation*}
$$

Clearly, this expression differs from the expected solution (C2). It reduces to the correct traveltime only if the dip angle $\varphi$ equals zero, i.e., all interfaces are horizontal.

## Appendix D: Derivation of formula (6.13)

A standard technique from the calculus of variations is used to demonstrate the validity of equation (6.13). If the function $m^{\prime \prime}(x)$ is perturbed by an arbitrary, but small, amount $\delta m^{\prime \prime}(x)$, then a variation $\delta \Phi$ is induced in the functional $\Phi\left(m^{\prime \prime}\right)$. An expression for this variation is obtained by evaluating equation (6.12) at $m^{\prime \prime}+\delta m^{\prime \prime}$. Hence

$$
\begin{gather*}
\delta \Phi \equiv \Phi\left(m^{\prime \prime}+\delta m^{\prime \prime}\right)-\Phi\left(m^{\prime \prime}\right) \\
=2 \int_{a}^{b}\left\{\mu w(x)^{2} m^{\prime \prime}(x)-\left[\mathbf{e}_{o b s}-\mathbf{e}_{p r d}\left(m^{\prime \prime}\right)\right]^{T} \mathbf{W}^{T} \mathbf{W} \mathbf{p}\left(x ; c_{1}, c_{2}\right)\right\} \delta m^{\prime \prime}(x) d x+O\left(\delta m^{\prime \prime}\right)^{2} \tag{D1}
\end{gather*}
$$

$\Phi\left(m^{\prime \prime}\right)$ is extremized by the particular $m^{\prime \prime}(x)$ such that $\delta \Phi$ is second order in $\delta m^{\prime \prime}$. Thus, the integral in equation (D1) must vanish identically. Since $\delta m^{\prime \prime}(x)$ is an arbitrary small perturbation, this can be achieved only if

$$
\begin{equation*}
m^{\prime \prime}(x)=\frac{1}{\mu}\left[\mathbf{e}_{o b s}-\mathbf{e}_{p r d}\left(m^{\prime \prime}\right)\right]^{T} \mathbf{W}^{T} \mathbf{W} \frac{\mathbf{p}\left(x ; c_{1}, c_{2}\right)}{w(x)^{2}} \tag{D2}
\end{equation*}
$$

The $m^{\prime \prime}(x)$ that extremizes $\Phi\left(m^{\prime \prime}\right)$ is a linear combination of the functions $p_{j}\left(x ; c_{1}, c_{2}\right) / w(x)^{2}$. Moreover, the coefficient vector in this linear superposition is proportional to the difference between observed and predicted data. Parker et al. (1987) obtain an analogous result for a different inverse problem formulation.

## Appendix E: Vanishing of two derivatives

This appendix proves that the extremum of the objective function in equation (6.17) is invariant with respect to the two abscissae $c_{1}$ and $c_{2}$. Differentiating $\Phi$ with respect to these parameters yields the two equations

$$
\begin{gather*}
\frac{\partial \Phi}{\partial c_{1}}=\boldsymbol{\alpha}^{T}\left[2 \Gamma \mathbf{W}^{T} \mathbf{W}+\mu \mathbf{I}\right] \frac{\partial \boldsymbol{\Gamma}}{\partial c_{1}} \boldsymbol{\alpha} \\
+2\left[d_{1} \mathbf{h}(b)-d_{2}\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right]-\mathbf{e}_{o b s}\right]^{T} \mathbf{W}^{T} \mathbf{W} \frac{\partial \Gamma}{\partial c_{1}} \boldsymbol{\alpha} \\
-2 d_{2} \mathbf{h}(b)^{T} \mathbf{W}^{T} \mathbf{W}\left[d_{1} \mathbf{h}(b)-d_{2}\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right]-\mathbf{e}_{o b s}+\mathbf{\Gamma} \boldsymbol{\alpha}\right] \tag{E1}
\end{gather*}
$$

and

$$
\begin{gather*}
\frac{\partial \Phi}{\partial c_{2}}=\boldsymbol{\alpha}^{T}\left[2 \mathbf{\Gamma} \mathbf{W}^{T} \mathbf{W}+\mu \mathrm{I}\right] \frac{\partial \Gamma}{\partial c_{2}} \boldsymbol{\alpha} \\
+2\left[d_{1} \mathbf{h}(b)-d_{2}\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right]-\mathbf{e}_{o b s}\right]^{T} \mathbf{W}^{T} \mathbf{W} \frac{\partial \Gamma}{\partial c_{2}} \boldsymbol{\alpha} \tag{E2}
\end{gather*}
$$

where $\Gamma$ stands for $\Gamma\left(c_{1}, c_{2}\right)$. These expressions are simplified by using the previous equation (6.18) obtained by extremizing the objective function with respect to the coefficient vector $\alpha$ :

$$
\begin{equation*}
\boldsymbol{\Gamma} \boldsymbol{\alpha}+\mu\left(\mathbf{W}^{T} \mathbf{W}\right)^{-1} \boldsymbol{\alpha}=\mathbf{e}_{o b s}-d_{1} \mathbf{h}(b)+d_{2}\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right] \tag{E3}
\end{equation*}
$$

Substituting (E3) into (E1) and (E2) immediately yields the simpler forms

$$
\begin{equation*}
\frac{\partial \Phi}{\partial c_{1}}=-\mu \boldsymbol{\alpha}^{T} \frac{\partial \Gamma}{\partial c_{1}} \boldsymbol{\alpha}+2 \mu d_{2} \mathbf{h}(b)^{T} \boldsymbol{\alpha} \tag{E4}
\end{equation*}
$$

$$
\begin{equation*}
\frac{\partial \Phi}{\partial c_{2}}=-\mu \boldsymbol{\alpha}^{\boldsymbol{T}} \frac{\partial \mathbf{\Gamma}}{\partial c_{2}} \boldsymbol{\alpha} \tag{E5}
\end{equation*}
$$

Expressions for the partial derivatives of the inner product matrix $\Gamma$ are required. Differentiating equation (6.16) with respect to $c_{1}$ yields

$$
\begin{equation*}
\frac{\partial \Gamma}{\partial c_{1}}=\int_{a}^{b} w(x)^{-2}\left[\frac{\partial \mathbf{p}}{\partial c_{1}} \mathbf{p}^{T}+\mathbf{p}\left(\frac{\partial \mathbf{p}}{\partial c_{1}}\right)^{T}\right] d x \tag{E6}
\end{equation*}
$$

where $\mathbf{p}$ stands for $\mathbf{p}\left(x ; c_{1}, c_{2}\right)$. An analogous formula is obtained for $\partial \Gamma / \partial c_{2}$. The derivatives of the kernel function vector $p\left(x ; c_{1}, c_{2}\right)$ are obtained by differentiating expression (6.10). Hence

$$
\begin{align*}
\frac{\partial \mathbf{p}\left(x ; c_{1}, c_{2}\right)}{\partial c_{1}}= & \mathbf{h}(b)\left[H\left(x-c_{1}\right)-H\left(x-c_{2}\right)\right]+\mathbf{h}(b)\left(x-c_{1}\right) \delta\left(x-c_{1}\right)  \tag{E7}\\
& \frac{\partial \mathbf{p}\left(x ; c_{1}, c_{2}\right)}{\partial c_{2}}=\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right] \delta\left(x-c_{2}\right) \tag{E8}
\end{align*}
$$

Substituting (E7) into (E6) and integrating gives the expression

$$
\begin{equation*}
\frac{\partial \mathbf{\Gamma}}{\partial c_{1}}=\mathbf{h}(b) \mathbf{q}^{T}+\mathbf{q} \mathbf{h}(b)^{T} \tag{E9}
\end{equation*}
$$

where $\mathbf{q}$ is an auxilliary vector defined by $\mathbf{q} \equiv \int_{c_{1}}^{c_{2}} w(x)^{-2} \mathbf{p}\left(x ; c_{1}, c_{2}\right) d x$. Similarly, substituting (E8) into the analogue of (E6) for $\partial \Gamma / \partial c_{2}$ gives the result

$$
\begin{align*}
\frac{\partial \boldsymbol{\Gamma}}{\partial c_{2}}=w\left(c_{2}\right)^{-2} & {\left[\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right] \mathbf{p}\left(c_{2} ; c_{1}, c_{2}\right)^{\boldsymbol{T}}\right.} \\
& \left.+\mathbf{p}\left(c_{2} ; c_{1}, c_{2}\right)\left[\mathbf{k}(b)-\left(b-c_{1}\right) \mathbf{h}(b)\right]^{T}\right] \tag{E10}
\end{align*}
$$

Relations (E9) and (E10) are the required formulae. Substituting these expressions into (E4) and (E5) and reducing yield

$$
\begin{gather*}
\frac{\partial \Phi}{\partial c_{1}}=2 \mu\left(d_{2}-\boldsymbol{\alpha}^{T} \mathbf{q}\right)\left[\boldsymbol{\alpha}^{T} \mathbf{h}(b)\right]  \tag{E11}\\
\frac{\partial \Phi}{\partial c_{2}}=-2 \mu w\left(c_{2}\right)^{-2}\left[\boldsymbol{\alpha}^{T} \mathbf{p}\left(c_{2} ; c_{1}, c_{2}\right)\right]\left[\boldsymbol{\alpha}^{T} \mathbf{k}(b)-\left(b-c_{1}\right) \boldsymbol{\alpha}^{T} \mathbf{h}(b)\right] . \tag{E12}
\end{gather*}
$$

However, if the objective function $\Phi$ is already extremized with respect to the two parameters $d_{1}$ and $d_{2}$, then $\boldsymbol{\alpha}^{T} \mathbf{h}(b)=\boldsymbol{\alpha}^{T} \mathbf{k}(b)=0$. Thus, equations (E11) and (E12) reduce to the desired results

$$
\begin{equation*}
\frac{\partial \Phi}{\partial c_{1}}=0, \quad \frac{\partial \Phi}{\partial c_{2}}=0 \tag{E13}
\end{equation*}
$$

Note that both of the conditions $\boldsymbol{\alpha}^{T} \mathbf{h}(b)=0$ and $\boldsymbol{\alpha}^{T} \mathbf{k}(b)=0$ are necessary for $\partial \Phi / \partial c_{2}$ to vanish.

## Appendix F: Spatially correlated traveltime errors

Suppose that x is a vector of $n$ random variables with covariance matrix $\mathrm{C}_{x}$. Then, the vector of $n$ random variables $y$ given by the linear transformation $\mathbf{y} \doteq \mathbf{A x}$ possesses the covariance matrix

$$
\begin{equation*}
\mathbf{C}_{y}=\mathbf{A} \mathbf{C}_{x} \mathbf{A}^{T} \tag{F1}
\end{equation*}
$$

If $\mathbf{C}_{x}$ and $\mathbf{C}_{y}$ are prescribed, then equation (F1) can be solved for the ( $n \times n$ ) transformation matrix $\mathbf{A}$.

Since the covariance matrices are symmetric and positive definite, they may be factored via Cholesky decomposition as follows:

$$
\mathbf{C}_{x}=\mathbf{L}_{x} \mathbf{L}_{x}^{T}, \quad \mathbf{C}_{y}=\mathbf{L}_{y} \mathbf{L}_{y}^{T}
$$

where $\mathrm{L}_{x}$ and $\mathrm{L}_{y}$ are lower triangular matrices. Then, one solution of (F1) is

$$
\begin{equation*}
\mathbf{A}=\mathbf{L}_{y} \mathbf{L}_{x}^{-1} \tag{F2}
\end{equation*}
$$

which can be readily verified by substitution.
An important special case occurs when the $x_{i}$ are a set of independent random variables drawn from the one dimensional normal distribution with zero mean and unit standard deviation. In this situation, the covariance matrix $\mathbf{C}_{x}$ equals the identity matrix. The transformed random variables are given simply by $\mathbf{y}=\mathrm{L}_{y} \mathbf{x}$. Moreover, each $y_{i}$ is normally distributed because it is a linear combination of independent normal variates.

Within the context of the arrival time picking problem, each $y_{i}$ is considered to be a rândom time error with an assigned standard deviation $\sigma_{i}$. A double-tailed exponential function is used to evaluate the correlation coefficient between the arrival time errors at
geophone stations $i$ and $j$ (Berni and Roever, 1989). The elements of the covariance matrix become

$$
\begin{equation*}
\left[\mathbf{C}_{y}\right]_{i j}=\exp \left(-d_{i j} / D\right) \sigma_{i} \sigma_{j} \tag{F3}
\end{equation*}
$$

where $d_{i j}$ is the distance between stations $i$ and $j$, and $D$ is an adjustable parameter called the correlation distance. A large value for $D$ (relative to the $d_{i j}$ ) implies that the individual arrival time errors are highly correlated, and vice versa. Standard FORTRAN subroutines (Press et al., 1986, p. 192-199) are used to generate independent normal variates $x_{i}$ with zero mean and unit variance. Hence, the $y_{i}$ calculated via the above procedure become a set of correlated, normally distributed traveltime errors.

## Appendix G: Ray initiation and termination logic

If a receiver is located on a grid node, on a horizontal grid line, or on a vertical grid line (points 1, 2, and 3 in Figure G1a, respectively), then the average of the steepest descent angles from the immediately adjacent cells determines which cell the raypath enters first. However, if the receiver is located within a cell, as in Figure G1b, then the logic contained in Table G1 is used to determine the initial path segment and the cell that the raypath subsequently enters. Angles $\beta_{1}$ through $\beta_{8}$ are all positive acute angles and are illustrated in Figure G1b.

Now assume that points 1,2 , and 3 are sources, rather than receivers. Then, the set of immediately adjacent cells constitutes a near source zone. Tracing of an incoming ray continues until the boundary of this zone is encountered. For example, the four cells surrounding point 1 in Figure G1a form a square with side length $2 h$. For source points 2 and 3, the near source zones are rectangles with vertical and horizonal dimensions of $2 h \times h$ and $h \times 2 h$, respectively. Finally, if the source is located within a cell (Figure G1b), then iterative ray tracing proceeds to the boundary of the enclosing square cell with side length $h$. In all cases, the final raypath segment is obtained by drawing a straight line from the boundary point directly to the source position.


Fig. G1 (a) Points 1, 2, and 3 denote receivers (or sources) located on a grid node or grid lines. (b) A receiver located within a square grid cell.

| Angle Range |  | Exit Coordinates | Next Cell Indice |  |
| :--- | :--- | :--- | :--- | :--- |
| $0 \leq \theta_{k}<\beta_{1}$ | $x_{i+1}$ | $z_{r}+\left(x_{i+1}-x_{r}\right) \tan \theta_{k}$ | $i+1$ | $j$ |
| $\theta_{k}=\beta_{1}$ | $x_{i+1}$ | $z_{j+1}$ | $i+1$ | $j+1$ |
| $\pi / 2-\beta_{2}<\theta_{k}<\pi / 2+\beta_{3}$ | $x_{\tau}+\left(z_{j+1}-z_{r}\right) \cot \theta_{k}$ | $z_{j+1}$ | $i$ | $j+1$ |
| $\theta_{k}=\pi / 2+\beta_{3}$ | $x_{i}$ | $z_{j+1}$ | $i-1$ | $j+1$ |
| $\pi-\beta_{4}<\theta_{k}<\pi+\beta_{5}$ | $x_{i}$ | $z_{r}+\left(x_{i}-x_{r}\right) \tan \theta_{k}$ | $i-1$ | $j$ |
| $\theta_{k}=\pi+\beta_{5}$ | $x_{i}$ | $z_{j}$ | $i-1$ | $j-1$ |
| $3 \pi / 2-\beta_{6}<\theta_{k}<3 \pi / 2+\beta_{7}$ | $x_{r}+\left(z_{j}-z_{r}\right) \cot \theta_{k}$ | $z_{j}$ | $i$ | $j-1$ |
| $\theta_{k}=3 \pi / 2+\beta_{7}$ | $x_{i+1}$ | $z_{j}$ | $i+1$ | $j-1$ |
| $2 \pi-\beta_{8}<\theta_{k}<2 \pi$ | $x_{i+1}$ | $z_{\tau}+\left(x_{i+1}-x_{r}\right) \tan \theta_{k}$ | $i+1$ | $j$ |

Table G1. Ray initiation logic for a receiver located within a cell. Column 1 gives possible ranges for the steepest descent direction $\theta_{k}$ assigned to the cell. Angles $\beta_{1}$ through $\beta_{8}$ are illustrated in Fig. G1b. Columns 2 and 3 give the horizontal and vertical coordinates of the ray exit point on the cell boundary, respectively. $\left(x_{r}, z_{r}\right)$ are the receiver coordinates. Columns 4 and 5 give the indices of the next cell that the ray enters.

## Appendix H: Optimum starting slowness

In the absence of a priori information, a uniform slowness model is used to initiate the tomographic inversion procedure. The value of this slowness can be chosen to minimize the rms traveltime residual on the first iteration. This tends to reduce the total number of iterations required for convergence to a specified misfit level.

Let $t_{i}$ be the $i^{\text {th }}$ observed traveltime, and let the straight line distance between the associated source and receiver be $d_{i}$. Then, the difference between the observed traveltime and the time predicted by a uniform slowness $s$ is $t_{i}-s d_{i}$. The $l_{2}$ norm (squared) of all such differences is

$$
\phi(s)=\sum_{i=1}^{N}\left(t_{i}-s d_{i}\right)^{2}
$$

where $N$ is the total number of observed arrival times. Extremizing $\phi(s)$ with respect to $s$ yields

$$
s=s_{o p t}=\frac{\sum_{i=1}^{N} d_{i} t_{i}}{\sum_{i=1}^{N} d_{i}^{2}}
$$

Alternately, the problem can be posed in terms of an optimum velocity instead of an optimum slowness. The result is $v_{o p t}=1 / s_{o p t}$. For the combined VSP and crosswell datasets used in Chapter $8, v_{\text {opt }}$ equals $2191 \mathrm{~m} / \mathrm{s}$ (for both exact and error contaminated traveltimes). For the crosshole data only, $v_{\text {opt }}=2310 \mathrm{~m} / \mathrm{s}$.

