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Creep and recovery tests in compression parallel to the grain were conducted on 2 in. by 2 in. Douglas-fir specimens 4 in. long at four different levels of moisture.

Specimens were loaded in stages up to a predetermined load. Instantaneous axial and lateral deformation as well as creep measurements were taken at each stage. Creep was observed over periods ranging from five minutes to twenty-five hours. Similarly, recovery was observed during unloading at successively lower stress levels.

There are indications that creep as well as negative creep and negative recovery were mainly due to moisture present in the cell walls.

Creep, in general, appeared to be more marked in the green specimens than in the intermediate and air-dry conditions. The only oven-dry specimen showed less creep than the air-dry specimens.

Results also show that the values of the coefficient of  $\frac{\Delta \mathcal{L}_{Lateral}}{\Delta \mathcal{L}_{Longitudinal}}$  (both radial and tangential), during the load rise were entirely different from those during the period of creep, indicating that the corresponding deformations were entirely different. The  $\mathcal{M}_{S}$  for the change in load were always higher than those for the periods of creep.

All specimens tested showed a recovery of more than 50% of the longitudinal creep. This indicates that creep in wood is made up of two parts, recoverable and permanent creep.

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#### Part I

#### INTRODUCTION

Wood, which was among the earliest materials to be used for construction, has long had an established place in the major field of structural engineering. Nowadays, through the wide application of results from fundamental and applied research, novel designs of large wooden structures are possible, making wood competitive with other materials such as steel and concrete.

Glued, laminated wooden construction, for example, has already been recognized to be of primary importance in this modern world of mechanization and automation. In the near future, prestressed or reinforced wooden members will likely join other engineered wood products for a more economical and satisfactory utilization of timber.

It has become desirable, therefore, that studies be conducted on some of the basic aspects concerning the strain behaviour of wood which take into account not only the effect of the applied stress, but in addition, the effect of time.

This present investigation was carried out primarily to provide fundamental information on the creep of wood loaded in compression parallel to the grain at four levels of moisture content. It was also concerned with the effect of sustained loading at successively higher stress intensities on the modulus of elasticity, ultimate compressive strength and maximum deformation.

In common with many materials, wood, when subjected to a constant force, exhibits an increase in deformation following an initial instantaneous strain. This subsequent increase in deformation, called creep or plastic flow, is time dependent and behaves in one of two ways. The creep may take place either at a continuously decreasing rate until it approaches an ultimate or limiting value; or, it may show gradually diminishing rate at first, then a constant rate and finally an increasing rate which eventually leads to failure.

Previous investigations elsewhere have shown that wood, under all types of loading, exhibits creep even when the stress is well below the standard proportional limit (1, 3, 4, 5, 6, 7, 8).

Tests at the U. S. Forest Products Laboratory as reported by Wood (8) have shown that Douglas-fir and white oak both exhibit noticeable creep when subjected to tension and compression parallel to the grain at stresses as low as 1500 psi. Wood also reported that there is an indication that creep in tension and compression is approximately proportional to stress.

Dietz (3) in his investigation of the creep behaviour of Douglas-fir found that for compressive stresses below the proportional limit, the creep is very low.

The Australian Forest Products Laboratory (1) reported

Numbers in parenthesis refer to literature cited.

observing increases in strain from 20 to 140 per cent in green mountain ash specimens subjected to compression for almost three years at stresses ranging from 10 to 35 per cent of the short time strength.

# Part II DESCRIPTION OF TEST MATERIAL

# Species Used

Douglas-fir (<u>Pseudotsuga menziesii</u>), a species of prime importance in world markets for structural grades of lumber, was the species chosen for this particular experiment.

As reported in F. P. L. Technical Note No. 3, "Strength and Related Properties of Woods Grown in Canada", Douglas-fir shows the following average values in compression parallel to the grain:

Moisture Content <sup>2</sup>	41%	12%
Nominal Specific Gravity	0.45	0.49
Stress at Proportional Limit	2810 psi.	4830 psi.
		7230 psi.
Modulus of Elasticity	1,670,000 psi.	1,950,000 psi.

Source and Selection of the Experimental Material

Material utilized in this work was obtained from a flatsawn plank selected from a stock of Douglas-fir on hand at the Civil Engineering Laboratory, University of British Columbia.

<sup>2</sup> Based on weight when oven-dry.

Based on volume at test and weight when oven-dry.

In order to simplify the interpretation of the results, it was considered desirable to use test pieces free of defects including knots, as straight grained as possible, and with sufficiently flat growth rings. When selected, the plank was in an air-seasoned condition registering a moisture content of about 12% with an electric moisture meter.

# Type and Preparation of Test Specimens

The specimens used in these experiments were nominally two-by-two inches in cross-section and four inches in length parallel to the grain (See Fig. 1).

The plank was surfaced on both flat grain faces to a nominal thickness of  $2\frac{1}{4}$  inches, the grain direction was determined, and sticks  $2\frac{1}{4}$ —inch wide were ripped parallel to the grain with growth rings as nearly as possible parallel and perpendicular to the end edges of the sticks. These were cross-cut into 5-inch lengths to produce end-matched blocks for the test specimens and their controls, and into 4-inch lengths for moisture content sampling. The individual pieces were reduced to final size only after they have already attained a moisture content very near the desired final condition.

In the final processing of the test specimens and their controls, particular care was taken not only to improve further the orientation of the annual rings so as to make them essentially at right angles to a pair of faces, but

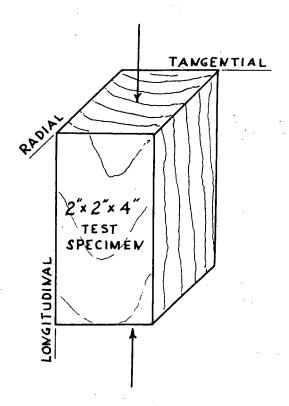


Fig. 1 - TEST SPECIMEN

also to make three sides of the blocks truly and mutually perpendicular to each other.

Each specimen was designated by a capital letter which indicated the moisture condition under which it was to be tested; air-dry condition was designated by the letter "A", intermediate moisture condition by "M", green condition by "G", and oven-dry condition by "O".

For the controls, the letter "C" was simply added. Thus, CA-1-6 indicates the control specimen in the air-dry condition.

## Part III

# MOISTURE CONDITIONS AT TEST

Creep and recovery tests in compression parallel to the grain were conducted at four different levels of moisture; namely:

- 1. Air-Dry Condition in which the moisture content of the specimens at test was in equilibrium with the atmosphere of the testing laboratory. The specimens were generally very close to 10% moisture content.
- 2. Intermediate Condition in which the moisture content was above 12% but below the fibre saturation point of the species which in this case is about 24% (2). Nominal moisture content for this condition was chosen to be 20%.
- 3. Green Condition in which the moisture content was above the fibre saturation point, the moisture content ranging

from 47% to 66%.

4. Oven-Dry Condition - in which the moisture content was very close to zero.

Conditioning of the Specimens

All test specimens were stored in temperature-andhumidity controlled rooms or chambers that would bring them to the desired final moisture content under which they were to be tested.

Those assigned to tests in the air-dry condition were stored exposed to the atmospheric conditions of the testing room until they reached constant weight. These specimens came to equilibrium at moisture contents ranging from 9.3% to 10.2%.

Those to be tested in the intermediate conditions were conditioned and stored in a chamber over a saturated solution of sodium sulphate. The chamber was maintained at a temperature of 75°F. At this temperature the relative vapor pressure above the salt solution was such as to result in moisture conditions that would give a nominal equilibrium moisture content in the blocks of 20%.

Those to be tested in the green conditions were stored in a controlled-humidity-temperature room. To facilitate the conditioning of these green specimens, they were first submerged in water under a vacuum for about three days.

Those to be tested in the oven-dry condition were conditioned in a thermostatically controlled electric oven

heated at 212° F and dried until no change in weight was observed for a period of twenty-four hours.

During the conditioning period, the specimens were weighed periodically, first at long intervals and finally, when they had very nearly attained their equilibrium moisture contents, more frequently. The weighing was continued until most of the specimens in each condition had maintained an almost constant weight indicating that they had attained the required equilibrium moisture content. Periodically, a moisture sample about an inch thich was taken from the centre of the moisture sample blocks and the moisture content, and its distribution throughout the cross-section, determined by the oven-dry method. (See Moisture Content and Specific Gravity Determination on page 24).

While every attempt was made to bring the air-dry, intermediate and green specimens to a uniform moisture content of 10%, 20%, and 60%, respectively, before testing, it was inevitable that some specimens exhibited a moisture content slightly higher or lower than the desired level. However, for the first two conditions, it was deemed imperative that no two specimens should have a difference in moisture of more than 10% of the nominal value, i. e., 1% and 2%, respectively, otherwise the test was discarded. For the green specimens, such limitation was not necessary since most of the strength properties of wood are influenced by moisture content only in the range below the

fibre saturation point.

The specimens were kept under these conditions until removed one at a time for testing.

#### Part IV

# EQUIPMENT USED

## Testing Machine

All the testing was performed on a Balwin Tate-Emery Testing Machine with a maximum capacity of four-hundred-thousand pounds and having three ranges of load. Only two ranges, the 80,000-lb. range, graduated at 100-lb. intervals, and the 16,000-lb. range, graduated at 20-lb. intervals, were used in these experiments.

A load maintainer, which is an accessory of the testing machine, was utilized in holding the load constant over a period of time in order that creep or recovery could be developed. With the use of the load maintainer, no fluctuations in the load have been observed even when held overnight.

A stop-watch was used to measure the time intervals while observing creep or recovery.

#### Stress-Strain Recorder

An autographic recorder, the Microformer Stress-Strain Recorder, was used to provide a record of load versus specimen deformation measured over a gauge length of two inches.

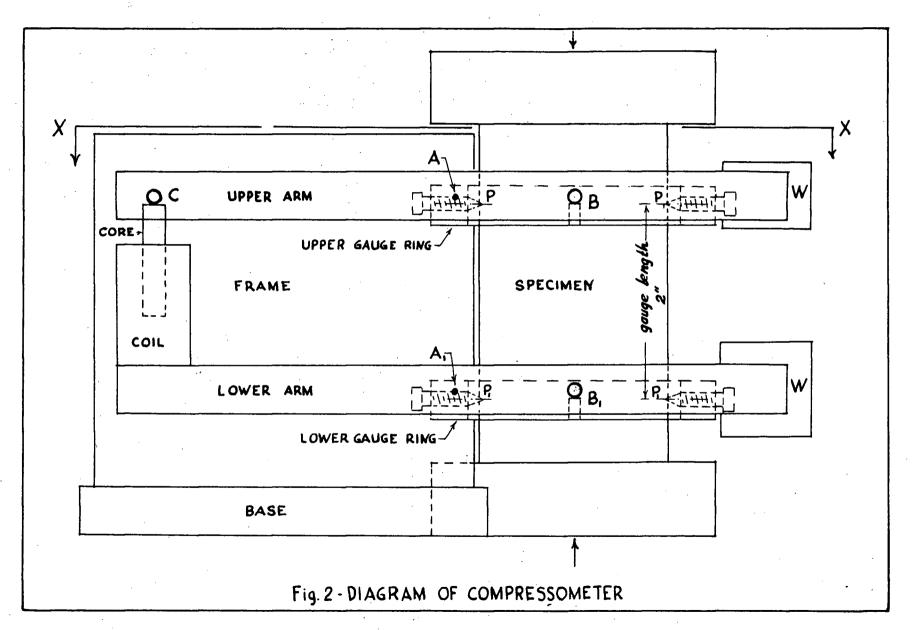
This apparatus consists essentially of two parts, the com-

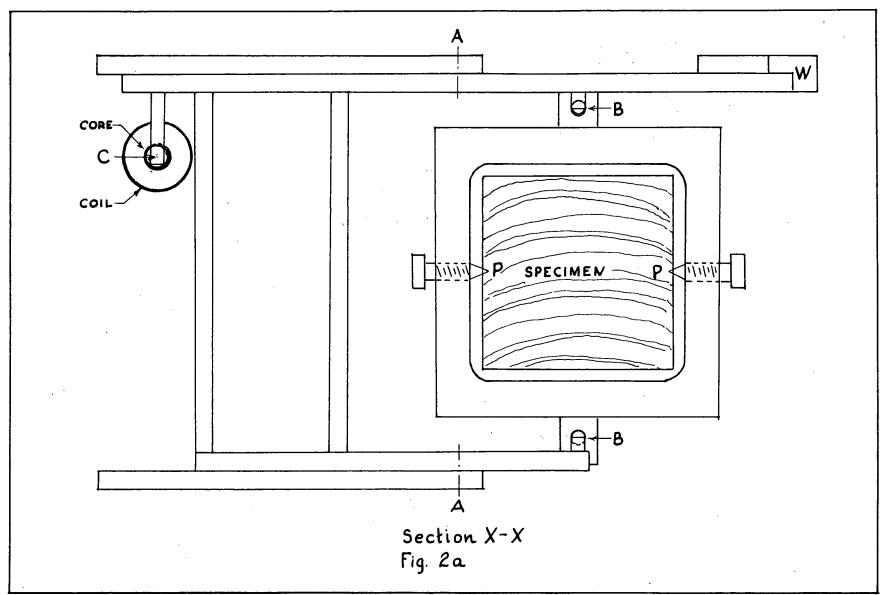
pressometer which is attached to the specimen and the microformer type recording equipment which produced autographic load-strain records of compression tests made parallel to the grain.

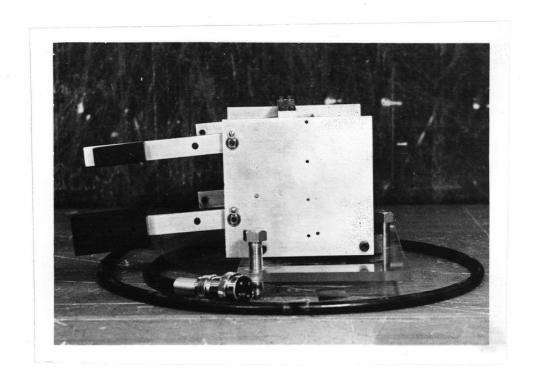
The recording equipment consists of a drum around which is wound a graph paper and a pen attached at the end of a push-rod which is in turn geared mechanically to the load indicating pointer of the testing machine. There are two coordinate axes in the graph; the load coordinate which is parallel to the axis of rotation of the drum and the strain coordinate which runs around the circumference of the drum. The load is marked by the pen which moves parallel to the axis of the drum while the drum rotates in proportion to specimen deformation.

The compressometer (Figs. 2 and 2a) includes a pair of gauge rings which are attached to the test specimen at a gauge length of two inches and a measuring assembly which measures the average values of displacement occurring between the gauge rings as the specimen is loaded. Each of the gauge rings is clamped to the test block by a pair of screws at points P and  $P_1$ .

The measuring assembly, which is also shown in Photograph 1, includes a heavy base supporting a vertical framework, two pairs of measuring arms and a microformer measuring unit. Each pair of measuring arms, upper arms and lower arms, pivots on a common axis A or  $A_1$ . Stable contact







Photograph 1
Measuring assembly of compressometer

between the measuring arms and the gauge rings at points Boor Book is maintained by the balance weights W.

The microformer measuring unit consisting of a coil unit and a movable core is fixed to the lower arm. The core, with a spring underneath, is forced to keep in contact with the upper arm at point Co.

When the test specimen shortens under the action of the load, the distance between the gauge points P and P<sub>1</sub> decreases and the upper and lower arms rotate about their respective axes A and A<sub>1</sub> causing their rear ends to move farther from each other. This results in the movement of the core relative to the coil. This motion of the core in relation to the coil unit actuates the rotation of the drum in the recording equipment proportional to the deformation of the gauge length of the specimen.

Two of the three available strain magnification settings of the Stress-Strain Recorder were used in recording the longitudinal deformation occurring within the gauge length of two inches. These are the "Intermediate" and the "High" which correspond to strain scales of 1 1000 micro-inches per inch and 1500 micro-inches per inch, respectively.

# Lateral Deformation Apparatus

To measure lateral deformation, creep or strain recovery, either in the radial or tangential directions, a special apparatus was devised. This apparatus was designed and patterned after the one developed and used by Dr. A. Hrennikoff of the Department of Civil Engineering of the University of British Columbia for measuring the lateral strain and creep of a concrete cylindrical specimen.

As shown in Fig. 3, it consists of two channel-shaped steel collars A joined by two bent iso-elastic4 strips which are fastened to the collars by means of small screws. The square ring formed by the collars and bent strips is clamped on the test specimens by two large thumb screws B passing through the mid-points of the collars. elastic springs form the sensitive elements which bend as the specimen undergoes distortion in the lateral dimension when it is subjected to a longitudinal compressive force. The bending strains in the iso-elastic springs are sensed by two Budd SR-4. Type CS-141-B electrical resistance strain gauges glued on the inner and the outer faces of both springs. All inside gauges C and separately all ouside gauges D are connected in series and attached to the active and compensating terminals of a Baldwin SR-4 Type L Strain Indicator whose readings are nearly proportional to the lateral strains in the specimen. With this arrangement, the inside gauges were also made active, but stressed differently from the outside ones. The indicator would then show the difference of the two strains which are of the opposite

Iso-elastic is an alloy of iron, nickel and chromium possessing particularly fine elastic properties.

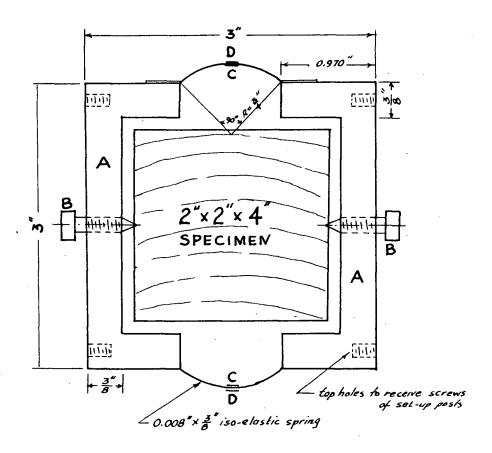


Fig. 3 - LATERAL DEFORMATION APPARATUS
ATTACHED TO THE RADIAL SIDES TO
MEASURE TANGENTIAL DEFORMATION

signs, thus increasing the sensitivity of the apparatus.

In order to be able to make simultaneous observations on the lateral strain, creep or recovery on both radial and tangential faces of the specimen, two apparatuses of this kind were made and used. It was also necessary to use a Baldwin switching and balancing unit to be able to read several gauges while using only one strain indicator.

The design of the lateral deformation apparatus consists mainly of determining the cross-sectional dimensions and geometry of the iso-elastic springs. Based on a useful range of strain of 3,000 micro-inches per inch, the spring was designed so that its stress will not exceed its elastic limit of 50 kips per square inch. In addition, a minimum force of about a third of a pound, was provided for as the initial force necessary to hold the apparatus in the specimen by friction alone.

When calibrated, the two apparatuses showed a sensitivity of 3.18 and 3.36 which means that a strain reading of 1 micro-inch per inch in the indicator corresponds to a strain of 3.18 or 3.36 micro-inches per inch on the test specimen.

Calibration of the apparatus was done by means of a 2-inch diameter cold rolled steel cylinder the lateral strains of which are known from strain gauges mounted directly on the cylinder.

#### Part V

## EXPERIMENTAL PROCEDURE

Measuring and Weighing of Test Specimens

Immediately before testing, each specimen was weighed to an accuracy of 0.01 gram and its cross-sectional dimension and length measured to the nearest 0.01 inch. After weighing and measuring, it was carefully wrapped and sealed completely with Saran Wrap, a polyvinyl plastic film, in order to prevent any increase or decrease in the moisture content of the specimen while the testing was in progress.

Immediately after each creep-recovery test, the specimen was weighed again to determine whether or not there was any change in moisture during the test. When a variation in weight of more than 1% was found, the test was cancelled. The change in weight before and after testing of the test specimens reported in this work was found to have a maximum value of only 0.3% which is an indication of the effectiveness of wrapping the specimens with Saran Wrap.

# Preparing the Specimen for Testing

The proper attachment to the test specimen of the four rings (two lateral deformation rings and the two compressometer rings) was accomplished with the aid of four set-up posts, which, together with the rings, formed a rigid frame (See Photograph 2). One lateral deformation ring is clamped on the tangential sides of the test block

to measure strain in the radial direction and the other on the radial faces to measure tangential distortion. Centre to centre distance of these two rings is five-eights of an inch and they are placed equidistant from the upper and lower compressometer rings. The posts are removed when the rings have been securely attached in their respective positions.

Photograph 2 shows the four rings and the set-up posts forming a rigid frame before attachment to the specimen.

Photograph 3 shows the rings attached to the specimen.

Compression Tests of Control Specimens

Initially, one end-matched control specimen was taken at random from each condition and tested in compression parallel to the grain according to standard procedures, except that the special compressometer was used to measure a deformation over a 2-inch gauge length. The specimen was subjected to progressive loading until failure, the rate of loading being maintained at a constant speed of approximately 0.012 inch per minute as per ASTM specifications. This gives the rate of loading from the following formula:

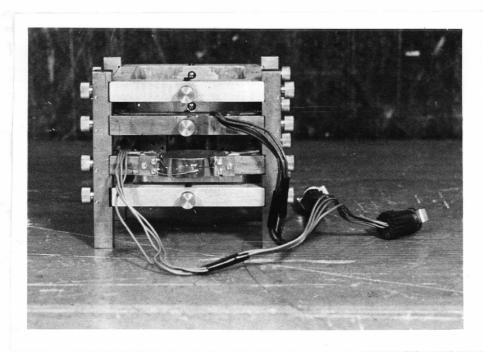
$$n = Z \times L$$

where m = speed of the movable head of the machine in inches per minute.

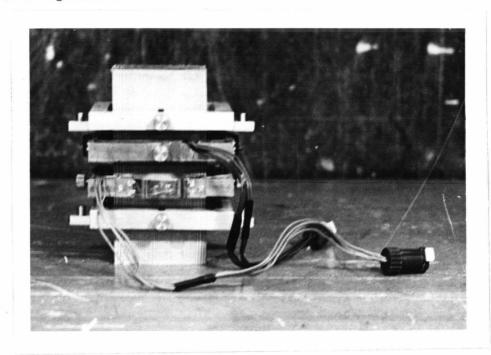
Z = rate of fibre strain per inch of fibre length

 $\mathcal{L}$  = length of compression specimen

The value commonly used for Z is 0.003, therefore, n = 0.012 inch per minute.



Photograph 2
Lateral deformation and compressometer rings rigidly connected with four set-up posts before attachment to a specimen.



Photograph 3
Lateral deformation and compressometer rings shown in their proper positions on the specimen.

While the recorder made possible the continuous and automatic recording of the load-axial strain graph, at least two persons were needed for the test, one reading the load and the other reading and recording the lateral strains from the strain indicator. The lateral strains were then plotted against load. From these sets of load-deformation curves, the two Poisson's ratios,  $\mathcal{M}_{LR}$  and  $\mathcal{M}_{LT}$ , were calculated. The Poisson's ratio  $\mathcal{M}_{LR}$  or  $\mathcal{M}_{LT}$  represents the numerical value of the ratio of the strain along the radial or tangential direction to that along the longitudinal direction due to a compressive stress parallel to the grain.

The results of these tests established the ultimate crushing strength of each of the three conditions of testing and permitted a selection of the various fixed stress-levels to which the other specimens were to be loaded in a step-wise manner.

# Step-by-Step Creep and Recovery Tests

In this type of test the specimen was subjected to compression parallel to the grain in successive steps at the desired various stress levels. A very rapid rate of loading was used so that no creep would come into play during the application of the load. The time of loading from one stress-level to the next higher one was about five seconds.

At every designated level of stress, the load was sustained for a desired period of time to record the

creep strain along the three perpendicular axes, after which the load was raised to the next step and at successively higher stress levels according to the loading schedule given in Tables 1, 2 and 3. Several load levels, chosen arbitrarily, were used, ranging from 13 to 91 per cent of the control's maximum load for the air-dry specimens, from 24 to 96 per cent for the intermediate conditions and from 23 to 81 per cent for the green specimens.

As in the testing of the control specimens, the load axial-strain relation was recorded automatically, and lateral strains, both in the radial and tangential directions, were read from the strain indicator and recorded throughout the duration of the test. In addition, time intervals were noted on the load-axial strain graph while simultaneous readings were made for any change of deformation in the lateral directions.

Creep was observed and recorded under a sustained loading for a minimum period of five minutes in some stress levels and a maximum of twenty-five hours at other levels of load. Measurements of creep were made every minute for the first five or ten minutes, every two or three minutes for the next ten or twenty minutes, and then as the creep rate became smaller, at convenient random time increments.

After stressing to a selected maximum stress-level had been completed, the specimen was unloaded in a similar step-by-step manner (See Tables 1, 2 and 3 for unloading schedule).

<u> </u>	<u> </u>														
Load Level		L	0	A	D	I., i	V G			U	N L	0 1	D	II	V. G.
kips	4	8	12	16	18	20	- 22	24	28	20	16	12	8	4	0
Per cent of max. load of controls	13	26	39	52	59	65	72	78	91	65	52	39	26	13	0
Specimen	1 .	Time duration of sustained load at each load level in minutes													
A-1-5 A-1-7 A-1-4 A-1-3(a) (b)	5	555555	5 15 30 15	5 15 5 25 25 25	1510	20 20 35 15	1500	225 15	5	5	555 5	555.55	5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	15 15 30	330 180 3095 135 195 135

SCHEDULE OF STEP-BY-STEP LOADING-UNLOADING TESTS (INTERMEDIATE CONDITION)

Load Level		L O	A	D	I N	G	UNI	COAD	ING
k1ps	4	8	10	12	15	16	8	4	0 :
Per cent of max. load of controls	24	48	60	72	89	96	48	24	0
Specimen		Time di	uration	of s	ustained in min		at each	n load	level
M-1-2 M-1-9 M-1-3 M-1-10 M-1-4(a) (b) M-1-7(a) (b)	5555555	60 50 5555555	30 5	60 50 5 60 30 5	75 1040 30	10	5550 10 555	5 10 15 20 10 10	215 145 155 1250 75 150 70 115

Table 3 - SCHEDULE OF STEP-BY-STEP LOADING-UNLOADING TESTS (GREEN CONDITION)

Load Level	L	O A	D I	N (	}	UN	LOAI	OING
kips	4	.8	10	12	14	-8	4	0
Per cent of max. load of controls	23	46	58	69	81	46	23	0
Specimen		ime dura	ation of		ined lo minute	ad at e	ach load	level
G-1-9 G-1-5 G-1-3 G-1-6 G-1-7(a)	555555	10 5 20 15 25 25	962 20 30 25 25	10 10 10 25	15	5 5 5	10 20 5 15 10	75 1025 90 135 95 150

<sup>(</sup>a) 1st. cycle of repetitive loading.(b) 2nd. cycle of repetitive loading.(c) 3rd. cycle of repetitive loading.

Longitudinal and lateral strain recovery - the creep on the unloading part of the cycle - were observed and recorded following the same technique used in creep observation.

operation without stops to failure at a machine speed of 0.012 inch per minute. Each test was continued until the load fell well below the maximum. This final test was done only after there was no further discernible recovery taking place in the specimen in its unloaded state after a considerable length of time which was in all cases equal or more than the testing time.

# Repetitive Loading

In this test, the specimen was loaded as in the previous step-by-step creep and recovery tests. Then after being unloaded, the specimen was loaded again in a similar step-wise fashion. For each succeeding loading cycle the selected maximum stress-level was made higher than that of the previous one. Specimens tested in this typesof repetitive loading were not reloaded until strain recovery, both axial and lateral, was virtually complete. A recovery period of attleast two hours at the unloaded state was allowed between loading cycles. As before, each specimen was finally loaded to failure.

#### Part VI

MOISTURE CONTENT AND SPECIFIC GRAVITY DETERMINATION

The moisture content and specific gravity of each test

specimen were determined from a small sample taken adjacent to the point of failure.

## Moisture Content

The moisture content was determined by the oven-dry method in the following manner:

- (1) After each final test, and after weighing the specimen, a moisture sample about an inch in thickness was taken and cut into sections as shown in Fig. 4. Moisture determinations were made for the core and each of the shells separately in order to determine the moisture distribution at the cross-section of the block.
- (2) Immediately after sawing, all loose splinters were removed and each section was weighed to the nearest 0.01 gram by means of a Mettler type balance graduated to the nearest 0.10 gram but which could be read by interpolation to the nearest 0.01 gram.
- (3) The material was then put into a thermostatically controlled electric oven heated at 212° F and dried until there was no variation in weight for a period of twenty-four hours.
- (4) Upon attaining this condition of constant weight,

  i. e., when all the moisture had been evaporated, the material
  was again carefully weighed.
- (5) The loss in weight expressed in per cent of the oven-dry weight indicates the moisture content of the specimen from which the sample was cut.

The variation in moisture content between the core and shells was limited to 0.1 %.

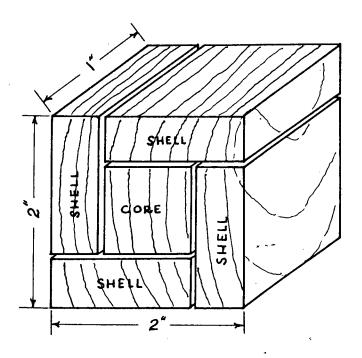


Fig. 4 - METHOD OF CUTTING MOISTURE SAMPLE FOR DETERMINATION OF SHELL AND CORE MOISTURE DISTRIBUTION

# Specific Gravity

The specific gravity for each specimen was calculated to the nearest 0.001 on the oven-dry weight, volume at test basis.

#### Part VII

# RESULTS AND DISCUSSIONS

# Creep and Recovery

Creep and recovery data, both axial and lateral, obtained from tests in compression parallel to the grain at the four moisture content conditions investigated are presented in Tables 4 and 5.

In general, creep, in the longitudinal and lateral directions, was found to be more marked in the green specimens than in either the intermediate or air-dry condition. The only oven-dry specimen showed lesser creep than the air-dry ones.

Apparently, creep in wood is due mainly to the moisture present in the cell walls. Upon application of the load, water in the walls is compressed so as to carry part of the load. While the load is held constant, the moisture is forced into the cell cavities and the pressure is relieved. The load, previously carried by the compressed water is transferred to the fibrous material, thus causing further deformation. The reason that the oven-dry specimen still showed some creep is probably due to the fact that all the moisture was not expelled.

During the rise in load, the specimen undergoes an increase in the lateral directions. When the load is sustained for a period of time, lateral creep occurs usually in the sense of bulging. In some cases, however, the lateral creep occurred in the opposite direction. This unusual phenomenon, in which the lateral dimension tends to contract instead of expanding further under a sustained longitudinal compressive stress, and which we will now call negative creep, has been observed to be more marked and predominant in the green specimens than in the specimens of intermediate moisture content. For the latter, only one, usually the upper level, in three stress levels at which creep was observed showed negative creep while in the green condition almost all of the stress levels showed negative creep.

Similar phenomena were also observed in the unloading part of the test. When the stress is reduced or released, a decrease in the lateral directions occurs, and a further contraction usually develops during the period of sustained load. At the low levels of stress in the intermediate and green conditions, however, an expansion in the lateral dimensions was noted during recovery following unloading. Again we will term this as negative recovery. As with negative creep, there was a preponderance of negative recovery observed in the green specimens.

However, not all the stress levels under which negative

creep was recorded exhibited a similar negative recovery.

For example, in Specimen M-1-2, there was a consistent negative creep both in the radial and tangential directions but no negative recovery either radially or tangentially was recorded. This was also found to be true with Specimen G-1-3 (Fig. 15), but in the tangential direction only. On the other hand, the creep and recovery in the two lateral directions of Specimen G-1-6 (Fig. 16) were found to be both negative. Such is the case, too, with Specimens M-1-4 and G-1-3 in their radial directions.

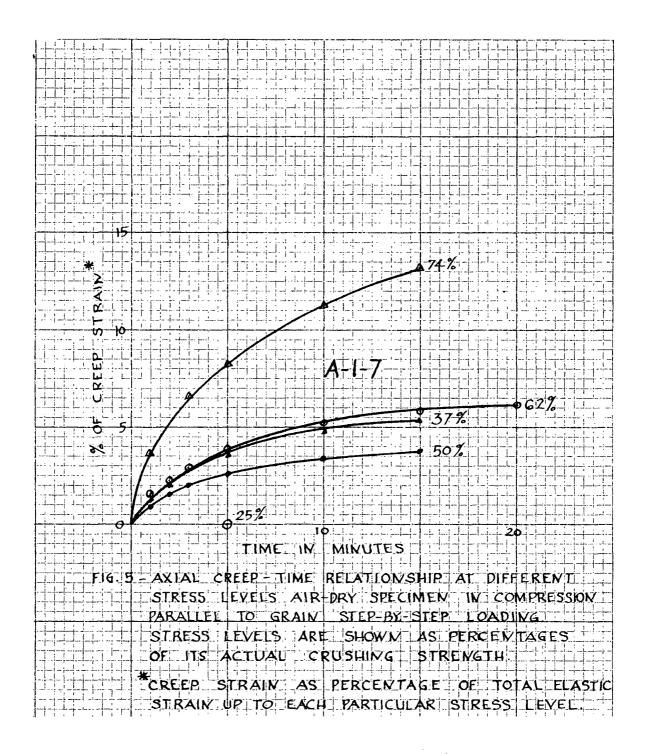
Since neither negative creep nor recovery was observed in the air-dry specimens tested and because such behaviour was found to have occurred more in the green than in the intermediate condition, it becomes obvious that the moisture content of the specimens had something to do with the negative creeping and recovery. It seems that during the rapid application of the load, pressure in the water within the cell structure builds up causing lateral expansion in addition to the normal distortion of the wood substance. When the load is sustained, some moisture finds its way into adjoining cavities. The water pressure in the cell walls is thus reduced, resulting in the lateral contraction: of the fibres. Hence, negative creep is developed. Negative recovery can be explained in the same way. After reduction or removal of the load, suction is developed and the moisture is attracted back to the cell walls, thus causing

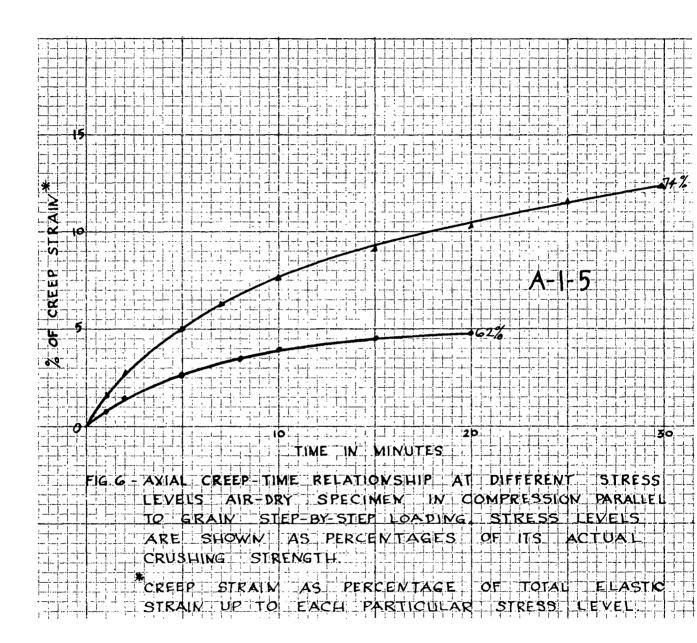
lateral expansion during the period of sustained load.

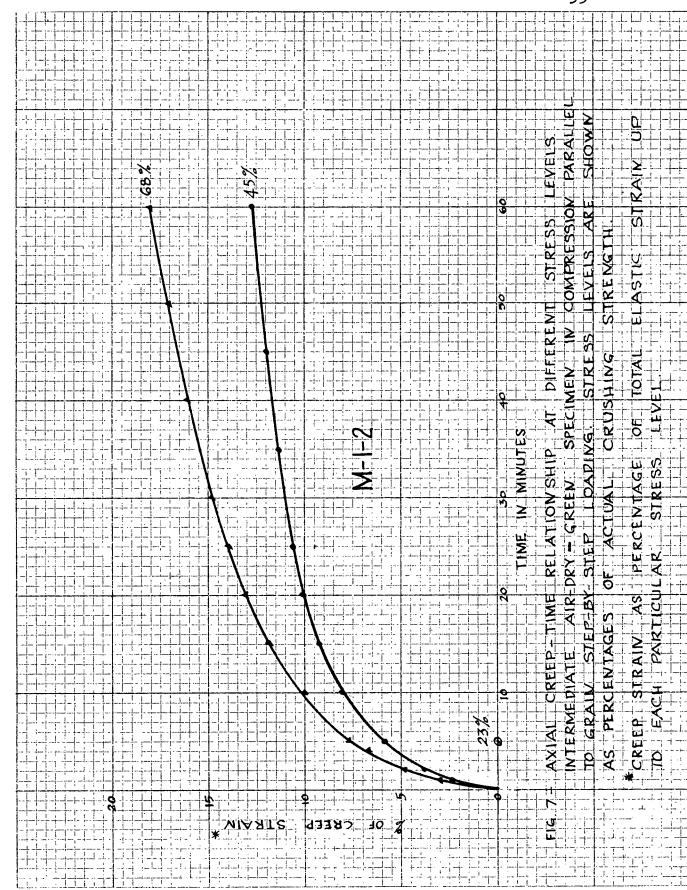
## Creep-Time Relationships

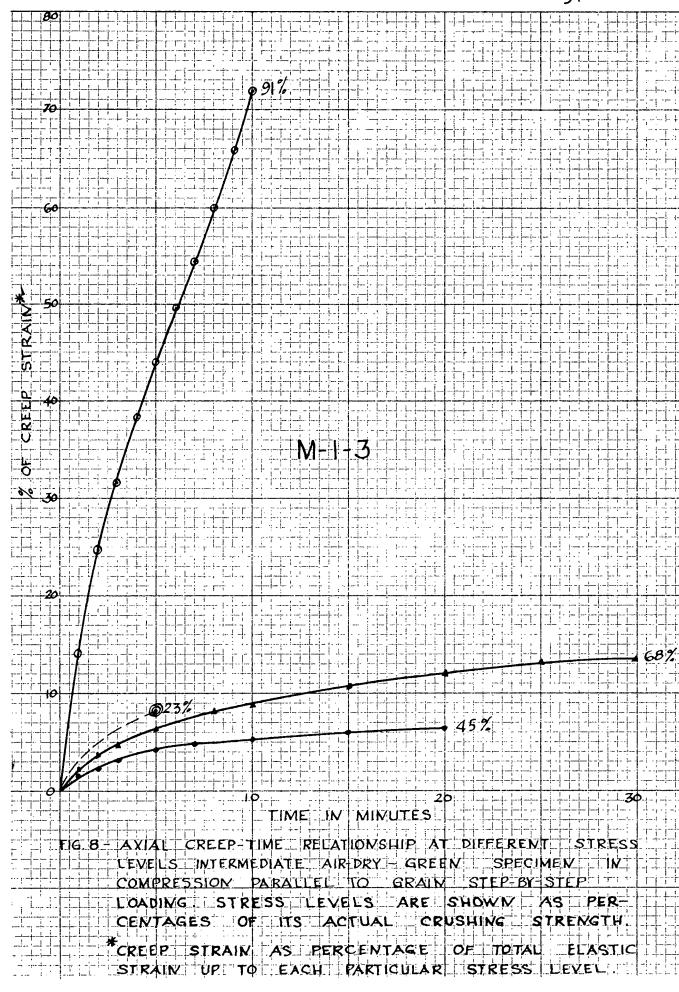
For each individual test specimen, axial creep values. expressed as percentages of the total elastic strain, from zero load to the stress level of the creep, were plotted against time at every creep level and smooth curve lines were fitted visually. The stress levels are expressed as percentages of the specimen's actual maximum crushing stress. By way of explanation, the specimen's actual maximum crushing stress, as used here, is the ultimate strength obtained from the static compression parallel to the grain test performed after the specimen had been subjected to the step-by-step creep and recovery tests as described earlier. Due to the natural variability of wood, and due to the effect of the first type of testing, the actual stress levels, expressed as the percentages of the ultimate compressive strength or each specimen, differed somewhat from the assumed stress levels which were based on the compressive strength of the control specimens.

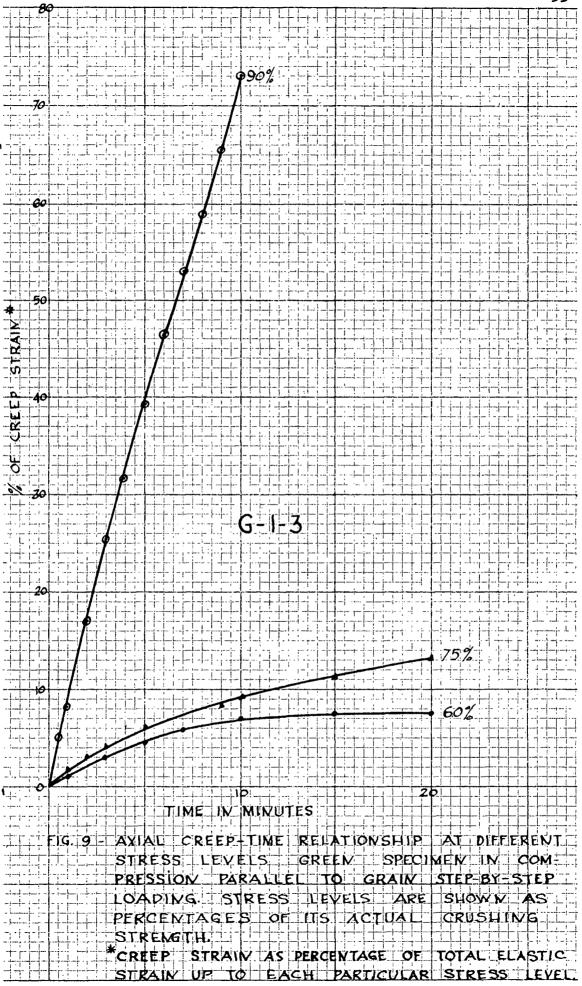
Typical creep curves showing the relationship of creeppercentage-of-elastic-strain and time, under various stresses,
are shown in Figs. 5 and 6 for the air-dry specimens, in Figs.
7 and 8 for the intermediate condition, and in Figs. 9 and 10
for the green specimens. Each set of curves is the result of
a test with an individual specimen.

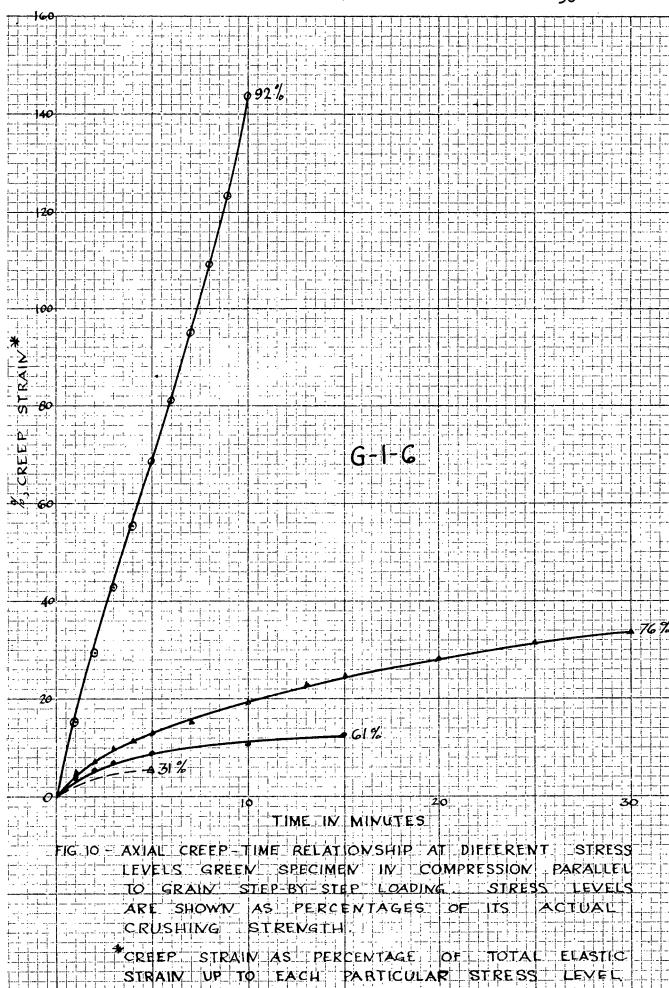












At about 90% stress level it appears that there is a rapid increase in creep, first at a decreasing rate, then at a constant rate and finally again at an increasing rate. Although none of the specimens was allowed to fail under a constant load, other investigators (8) have pointed out that the increase in the rate of creep is a sign of imminent failure.

continuous deformation at a diminishing rate, until an almost constant value is ultimately reached, is characteristic of the creep curves for stress levels up to about 70%. These curves are relatively flatter than those of the 70 - 77% stress levels. Within the stress levels and time range used in this investigation, these curves indicate that creep proceeds rapidly for the first few minutes after which the rate gradually diminishes with increasing time.

## Coefficient of Lateral Deformation

A specimen subjected to a compressive force undergoes deformation not only in the direction of the applied load but also in the lateral direction. Within the elastic limit of the material, the ratio of these deformations, lateral to longitudinal, is commonly known as the Poisson's ratio. In wood, this elastic property is obtained from standard tests made at a uniform rate of loading.

Similar ratios are given in Tables 6 and 7. They are herein referred to as the coefficients of lateral deformation,  $\mathcal{M} = \frac{\Delta \epsilon_{Lateral}}{\Delta \epsilon_{Longitudinal}}.$  These are of two kinds,

namely, the ratio of strain increments for the change in load and the ratio of strains during the period of creep or recovery.

In order to distinguish them from the usual Poisson's ratios,  $\mathcal{M}_{LR}$  and  $\mathcal{M}_{LT}$ , we designate the coefficients of radial and tangential deformation by  $\mathcal{M}_{R}$  and  $\mathcal{M}_{T}$ , respectively.

Typical  $\mathcal{M}$  diagrams are presented in Figs. 11 and 12 for the air-dry specimens, in Figs. 13 and 14 for the intermediate condition and in Figs. 15 and 16 for the green condition. Fig. 17 is that of the oven-dry specimen.

The values of the coefficient of lateral deformation, both radial and tangential, during the period of creep are entirely different from those during the load rise, indicating that the corresponding deformations are entirely different. In all the specimens tested, the Ms for the period of creep showed consistently lower values than those for the change in load.

At almost all stress levels, the coefficient of tangential deformation exhibited a higher value than its radial counterpart. This is probably due to the medullary rays running radially in the wood which restricts its deformation in the radial direction.

During the loading part of the cycle, some of the sum  $\mathcal{M}_R + \mathcal{M}_T$  have been observed to be greater than one, indicating that the material in a way opens up. On the other hand,

the negative sum  $\mathcal{U}_R + \mathcal{U}_T$  indicates a reduction in volume of the material.

Whenever feasible, the  $\mathcal{M}'$  during the period of creep were calculated separately for different parts of time intervals. These values appear to become smaller with increase in time for the intermediate condition while the reverse is true for the air-dry condition.

## Step-wise Loading and Modulus of Elasticity

Illustrated in Graphs 1, 2, 3, 4, and 5 are typical graphs traced automatically by the stress-strain recorder during the step-wise creep-recovery tests in compression parallel to the grain. At each level of load, short vertical lines were drawn on the horizontal portions of the graphs. These represent time increments in minutes during which creep or recovery, axial or lateral, was recorded. Whenever there was measurable creep or recovery, marking was done every minute for the first five or ten minutes, every two or three minutes for the next ten or twenty minutes, and so on, the time intervals gradually increasing thereafter.

Due to the rapid rate at which the load was raised, the load axial deformation relationship was, for all practical purposes, linear on each load increment. Such a linear behaviour was observed to exist in some specimens even up to stress regions beyond the standard proportional limit of the material, where it would normally plot as a

curved line under the ordinary rate of testing. Graphs 2 and 4, with load levels up to 95 and 92 per cent of the specimen's actual strength, respectively, will serve to illustrate the foregoing statement.

Obviously, the curved portion of a stress-strain graph as ordinarily obtained by the conventional testing method, is mainly due to creep that is taking place within the duration of the loading time.

In general, however, the slopes of the straight lines between two consecutive loads for each graph are not of equal magnitude. As one goes from one interval to the next higher interval, the slopes have a tendency to become smaller, though not consistently so. Consequently, because the modulus of elasticity is directly proportional to the slope, a similarly decreasing Young's modulus is evident from the figures in Table 6.

In the air-dry specimens, the decrease in the modulus of elasticity was of the order of 2 to 6 per cent with the exception of Specimen A-1-3 where a 12% reduction was noted from the first- to the second-step interval. Generally, the percentage decrease was greater in the intermediate condition and slightly higher for the green specimens, the former having a maximum reduction of 21% and the latter a maximum of 26%.

The said linearity in the graphs is also evident during unloading, although to a lesser degree, especially during the last unloading step where a pronounced curvilinear graph has been observed in all the intermediate and green specimens.

This curvilinear characteristic could be attributed to the moisture in the wood. Upon release of the load, suction is developed causing tension in the water at first and finally the water that had been forced out of the cell walls during the period of creep, is reverted. This accounts to the curvature of the unloading curve.

It will also be observed from the same graphs that in the formation of the hysteresis loop, the strain for a given load increment is generally greater upon releasing the stress than upon applying it. Or, putting it in another way, the modulus of elasticity is smaller on release of a load than on its application. This seems to indicate that upon decrease or removal of the compressive stress, there is, in addition to the elastic strain, an immediate recovery of a portion of the creep. This was found to be true in all the specimens tested in the intermediate and green conditions while the air-dry specimens did not show such a consistent trend.

Effect of Creep-Recovery Tests on the Subsequent Stress-Deformation Relation

Graphs 6, 7, 8 and 9 are examples of typical graphs showing the load-axial and load-lateral strain relationships obtained from a compression parallel to the grain test done at a uniform testing rate of about 0.012 inch per minute. As in the other graphs, the load-axial curve was automatically recorded while the two other graphs (radial and tangential) were plotted from experimental data obtained through the strain indicator.

With the exception of Graph 6, which is that of a control specimen, these graphs are all from specimens that have been previously subjected to the creep and recovery tests. It will be noticed from the last two graphs, which are those of the intermediate and green conditions, respectively, that the load-axial curves show a deviation from straightness almost from the origin, although the curvature is not very pronounced. This characteristic is present in all the curves of the intermediate and green specimens.

Graph 7, on the other hand, is typical for the air-dry specimens, with the exception of Specimen A-1-5, which also exhibited a non-proportional stress-strain relationship similar to that of either Graph 8 or 9. All the curves, however, showed a definite peak that indicated maximum load from which the maximum stress and strain values as given in Table 8 were obtained.

Also included in Table 8 are the moisture content and specific gravity of each specimen, and the modulus of elasticity and Poisson's ratios of the controls. From this table a comparison of the properties of the individual specimens with those of their corresponding controls can be made. All the air-dry specimens tested have shown higher ultimate compressive strength than their controls, the increase being of the order of 4 to 8 per cent. Likewise, for the intermediate condition, all but two showed greater strength than their controls, the percentage of increase varying from 5 to 16. For the green specimens, the opposite result was obtained

that is, the control showed higher crushing stress. A reduction of about 20 per cent was noted in all but one of the green specimens.

All the specimens tested at the three moisture conditions (air-dry, intermediate and green) exhibited strains at maximum load consistently greater than those of their respective controls.

### Permanent Set

Axial and lateral (radial and tangential) permanent set and strain recovery expressed as percentage of the axial creep are tabulated in Table 9.

Each of the residual strain values was taken at the time when no more measurable recovery was taking place. From the table it will readily be observed that, in all cases, more than half of the axial creep that had taken place had been ultimately recovered. Hence, creep of wood could then be classified into two kinds, namely, recoverable creep which other workers have considered as an elastic after-effect or delayed elasticity, and permanent creep or plastic deformation. Recoverable creep in the lateral direction was not determined because of the accidental disturbance of the lateral deformation apparatus at the end of the test.

#### Conclusions

The more important conclusions in this investigation are as follows:

- 1. Creep, both longitudinal and lateral, in the green specimens was, in general, more marked than in either the intermediate or air-dry condition. The oven-dry specimen showed creep response less than that of the air-dry specimens. Creep, therefore, could be attributed mainly to the presence of moisture in the cell walls.
- 2. Negative creep and negative recovery in the lateral directions observed in the intermediate and green conditions were due to moisture present in the cell walls.
- 3. Values of the coefficient of lateral deformation,  $\mathcal{M} = \frac{\Delta \mathcal{E}_{Lateral}}{\Delta \mathcal{E}_{Longitudinal}} \text{ (both radial and tangential), during the load rise are entirely different from those during the period of creep, indicating that the corresponding deformations are entirely different. The <math>\mathcal{M}$  in the tangential direction is usually greater than the  $\mathcal{M}$  in the radial direction. This is probably due to the medullary rays running radially in the wood which somehow restrict its deformation in that direction.
- 4. Decreasing moduli of elasticity have been observed during loading at successively higher stress levels.
- 5. Stress-strain curves from final tests of the intermediate and green specimens were found to be curvilinear from the beginning of loading, although the curvature was not very pronounced.
- 6. More than half of the longitudinal creep that had developed was ultimately recovered in all the specimens tested.

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Table 8- VALUES FROM FINAL STATIC TESTS IN COMPRESSION PARALLEL TO GRAIN FOLLOWING CREEP\_RECOVERY TESTS

	Moisture Content	Specific Gravity	Max. Stress	Max. Strain	Modulus of Elasti-	ľ	son's tio
	(%)		(psi.)	micro- in An )	'	MLR	$\mathcal{M}_{LT}$
CA-1-6* A-1-5	oven-dry 9.6 9.6	0.472 0.497 0.504	11,400 7300 7860	12200 6000 7300	1890	0.372	0.402
A-1-7 A-1-4 A-1-3 CM-1-1* M-1-2	10.2 9.5 9.6 19.9 20.8	0.541 0.492 0.492 0.491 0.491	7760 7750 7600 4190 4400	11,050 6700 6200 4400 4800	1610	0.210	0.596
M-1-9 M-1-3 M-1-10 M-1-4 CG-1-10 <sup>4</sup>	20.3 21.8 20.0 20.0 63.0	0.518 0.490 0.541 0.500 0.547	3720 4400 4850 4180 4100	4720 5500 7900 4650 3850	2000	0.342	0.651
G-1-9 G-1-5 G-1-3 G-1-6 G-1-7	46.9 60.5 65.5 62.1 57.4	0.545 0.498 0.504 0.499 0.547	3290 3280 3320 3250 3900	4300 6240 4250 6600 4850			

Control specimens were not subjected to creep-recovery tests.

Table 9- PERMANENT SET AND STRAIN RECOVERY OF DOUGLAS\_FIR TESTED FOR CREEP

	Per	maneni	t Set	Strain Recovery
Specimen		Tangential	Longitudinal	Percentage of Axial Creep
0-1-1 A-1-5 A-1-7 A-1-4 A-1-3(a) (b) (c) M-1-2 M-1-9 M-1-3 M-1-4(a) (b) M-1-7(a) G-1-9 G-1-5 G-1-5 G-1-6 G-1-7(a) (b)	0 48 48 4 4 - 27 - 27 138 - 32 - 34 - 30 - 22 - 67 - 4	17 77 91 * 10 232 -16 -127 29 43 -6 -19 * -35 -20 -29 * 60	270 300 370 270 0 40 700 80 750 490 1440 100 160 800 175 460 30 180	51.8 70.4 556.5 100.3 55.5 77.7 64.6 81.9 87.0 87.0 89.8 89.8

<sup>\*</sup> No values recorded due to accidental disturbance of the lateral deformation apparatus.

(a) Values of 1st. cycle.

(b) Values of 2nd. cycle.

(c) Values of 3rd. cycle.

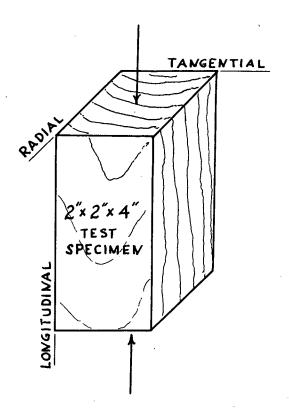
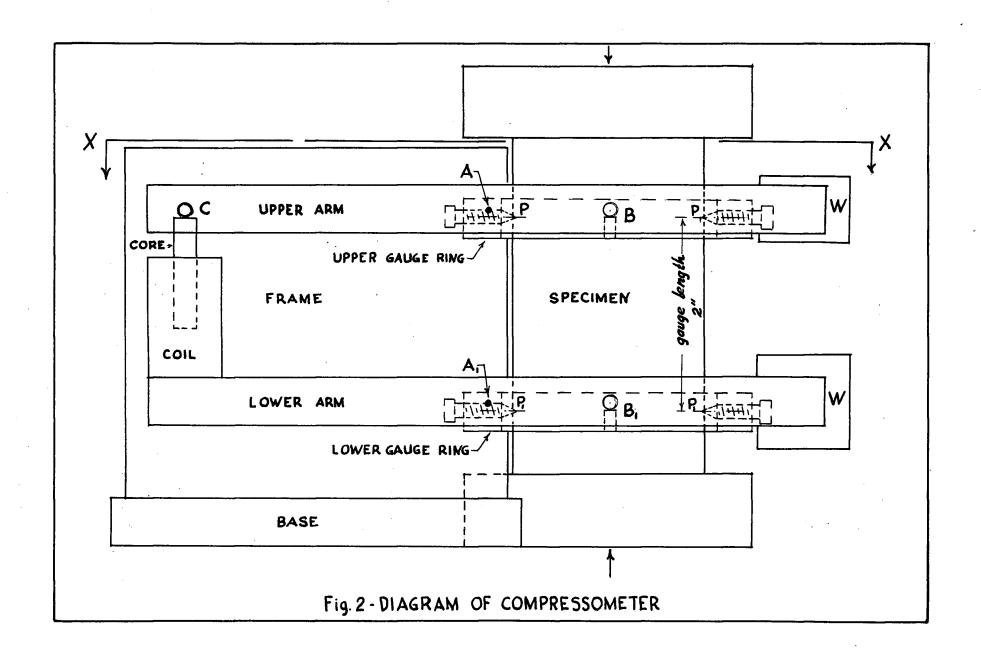
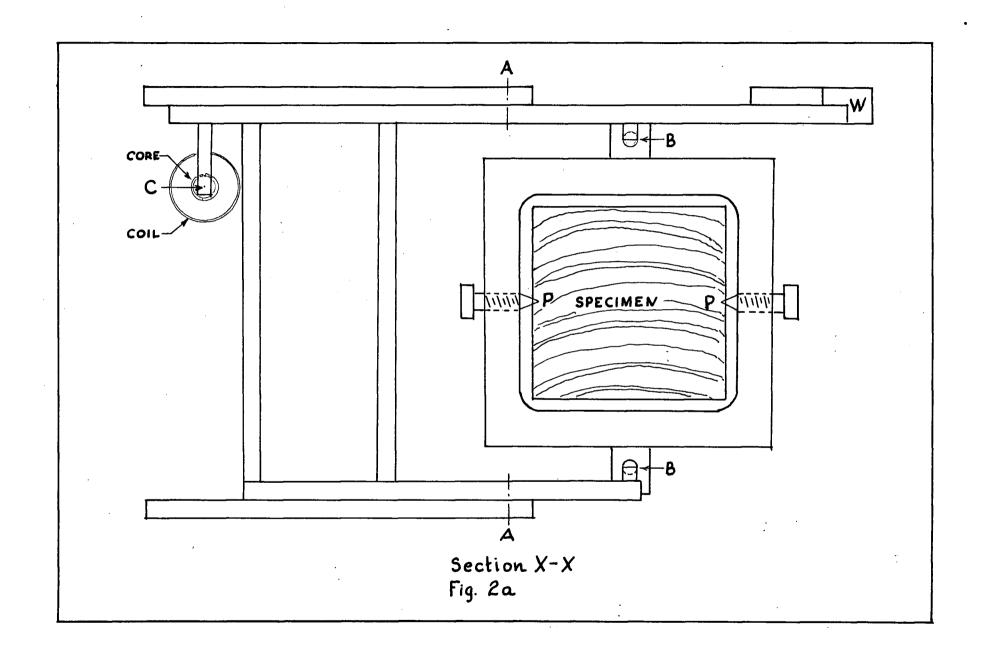


Fig. 1 - TEST SPECIMEN





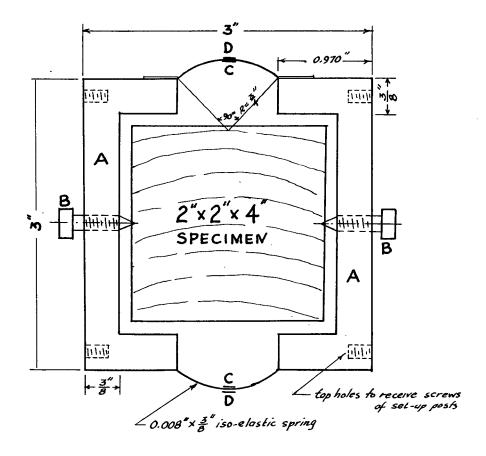


Fig. 3 - LATERAL DEFORMATION APPARATUS
ATTACHED TO THE RADIAL SIDES TO
MEASURE TANGENTIAL DEFORMATION

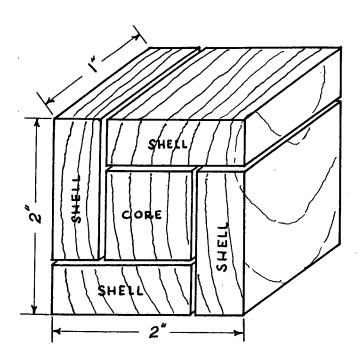
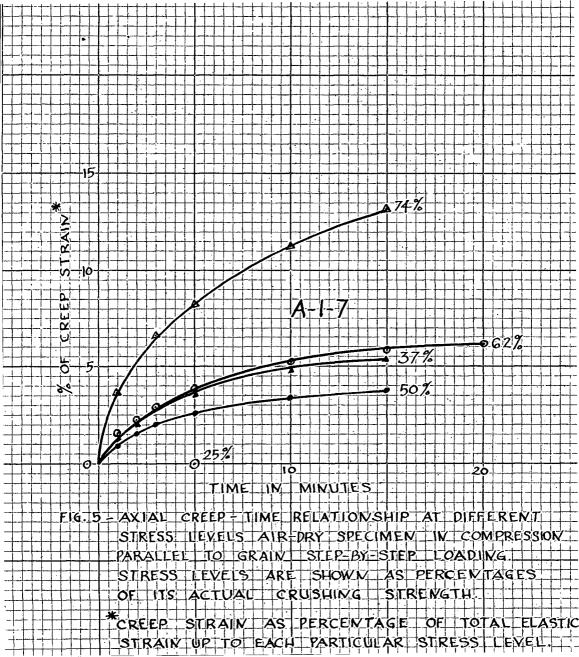
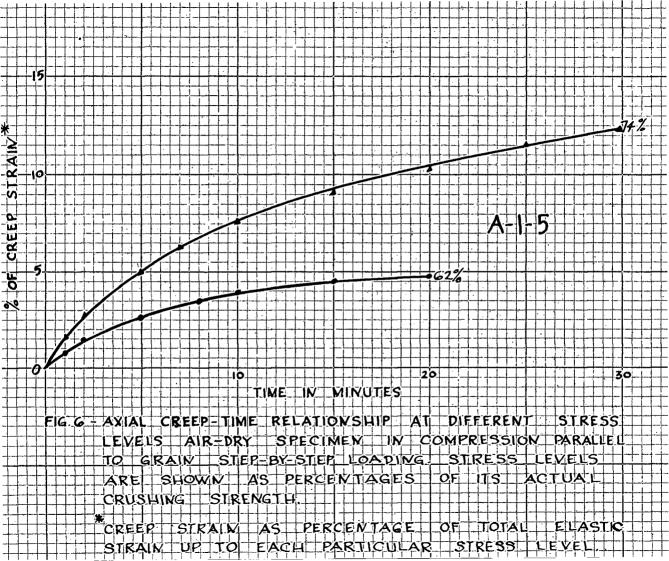
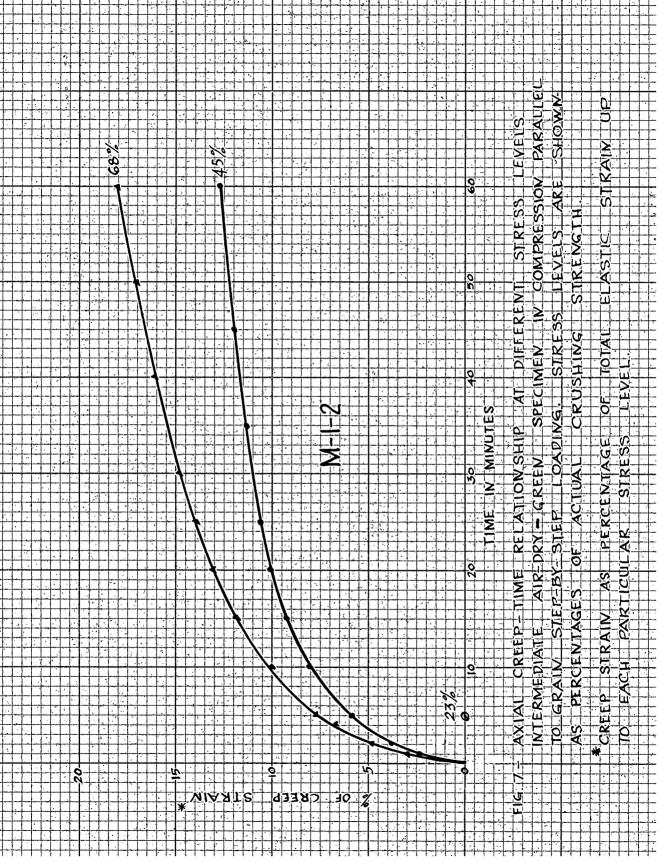
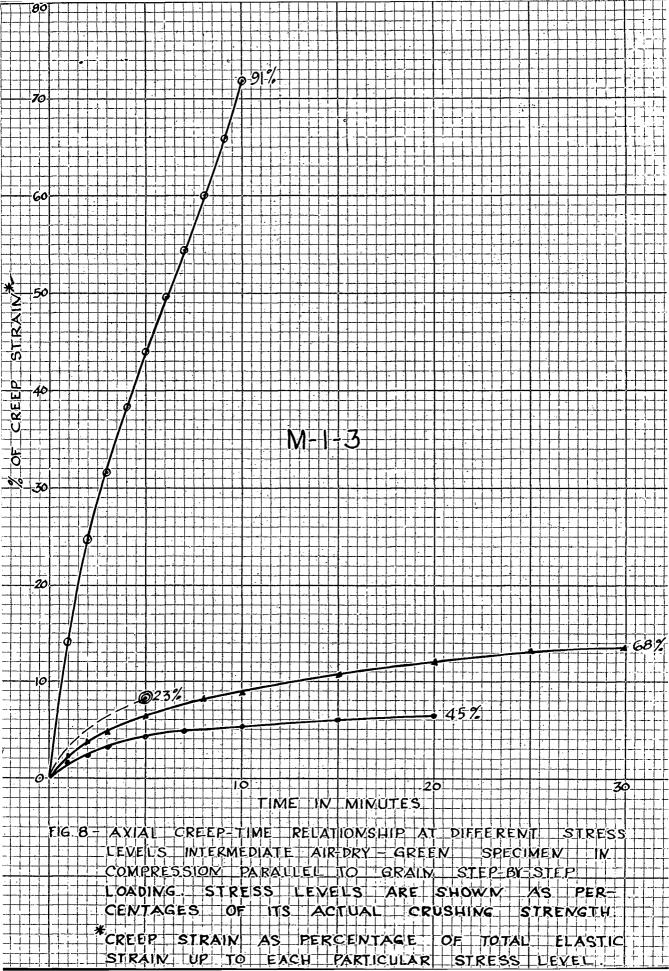


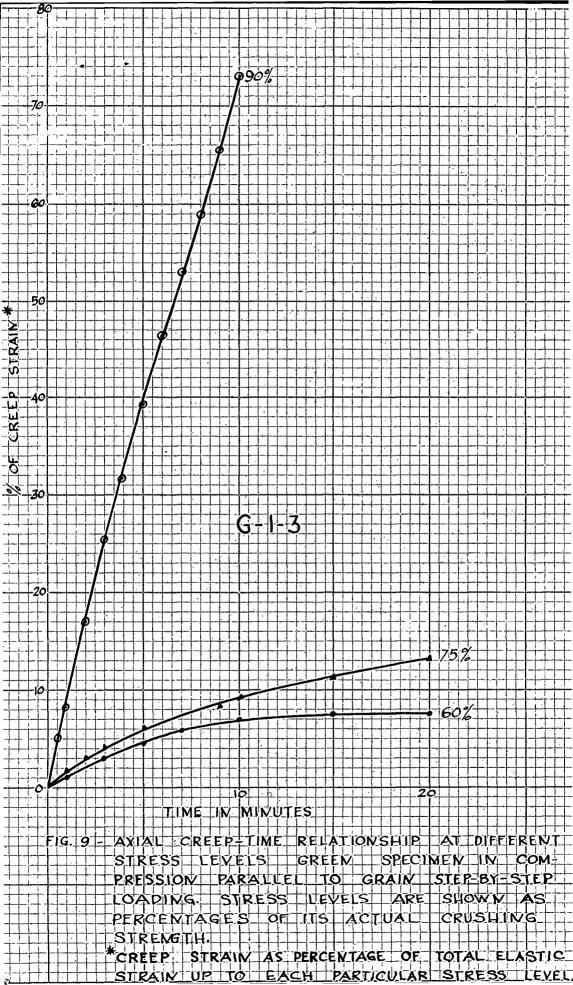
Fig. 4 - METHOD OF CUTTING MOISTURE SAMPLE FOR DETERMINATION OF SHELL AND CORE MOISTURE DISTRIBUTION











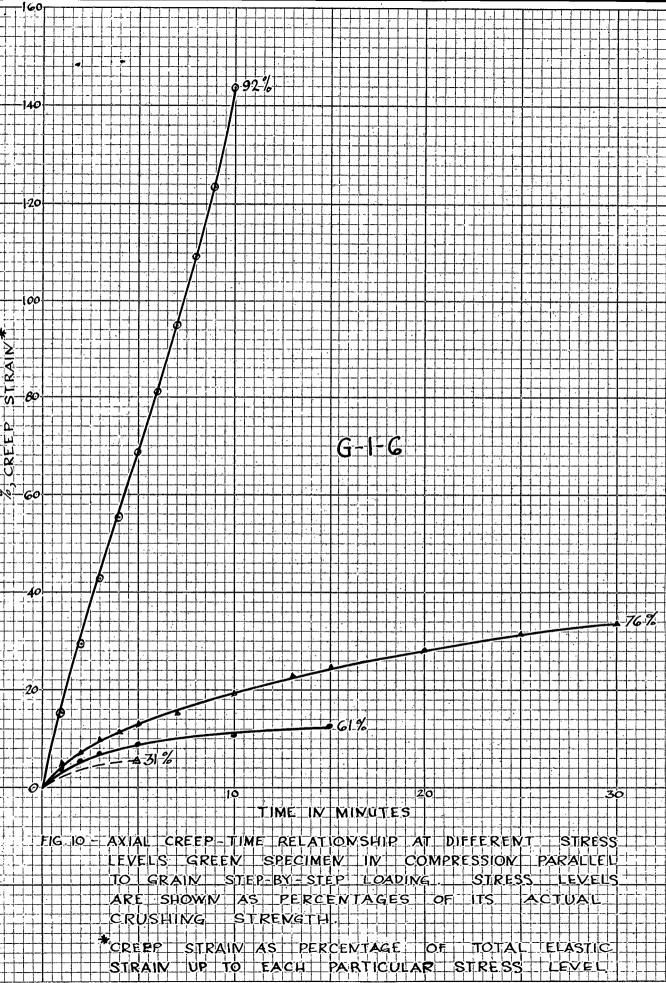


Table 1 - SCHEDULE OF STEP\_BY\_STEP LOADING\_UNLOADING TESTS (AIR\_DRY CONDITION)

oad Level		L	0	A	D	I. I	V G			U:	N L	0 1	A D	I.	N G;
kips	4	8	12	16	18	20	22	24	28	20	16	12	8	4	0
		26	39	52	59	65	72	78	91	65	52	39	26	13	0
	3		<u>,                                    </u>		<u> </u>	<u> </u>		<u> </u>				<u> </u>		<u> </u>	<u>                                     </u>
Specimen	I	Tim	e du:	rati	on of	süst	ained	load	d at	ea	ch :	loa	d l	eve:	1
			•			11	a min	utes							
A-1-5		5	5	5		20		225			-5	5	5	20	330
•	1	5	15	15	j	20	Ī	15	·		5	5	5	15	180
	5	5	5	5	1510	ł	1500				5	5	5	15	3095
A-1-3(a)		5	30		ļ ·								25	_	135
(b)	1	5	15	25	1	35						5	15	35	195
(c)	<u> </u>	5	15	15		15		15	5	5	5	5	5	10	135
	kips er cent of max. load f controls Specimen  A-1-5 A-1-7 A-1-4 A-1-3(a) (b)	kips 4 er cent of max. load 13 f controls Specimen  A-1-5 A-1-7 A-1-4 A-1-3(a) (b)	kips 4 8 er cent of max. load 13 26 f controls  Specimen Tim  A-1-5 A-1-7 A-1-4 A-1-3(a) (b)  5	kips     4     8     12       er cent of max. load f controls     13     26     39       Specimen     Time dual       A=1-5     5     5       A=1-7     5     5       A=1-4     5     5       A=1-3(a)     5     30       (b)     5     15	kips     4     8     12     16       er cent of max. load f controls     13     26     39     52       Specimen     Time durati       A=1-5     5     5     5       A-1-7     5     5     5       A-1-3(a)     5     30     25       (b)     5     15     25	kips     4     8     12     16     18       er cent of max. load f controls     13     26     39     52     59       f controls     Time duration of       A=1-5     5     5     5     5     15       A=1-7     5     5     5     5     1510       A=1-3(a)     5     30     25     1510       A=1-3(a)     5     15     25     1510	kips       4       8       12       16       18       20         er cent of max. load f controls       13       26       39       52       59       65         Specimen       Time duration of sustant and	kips       4       8       12       16       18       20       22         er cent of max. load f controls       13       26       39       52       59       65       72         f controls       Time duration of sustained in minumant         A=1-5       5       5       5       20         A=1-7       5       15       15       20         A=1-4       5       5       5       5       1510       1500         A=1-3(a)       5       30       25       35       35         (b)       5       15       25       35	kips       4       8       12       16       18       20       22       24         er cent of max. load for controls       13       26       39       52       59       65       72       78         Specimen       Time duration of sustained load in minutes         A=1-5       5       5       5       20       225         A=1-7       5       15       15       20       15         A=1-4       5       5       5       1510       1500         A=1-3(a)       5       30       25       35       1500         (b)       5       15       25       35       35	kips       4       8       12       16       18       20       22       24       28         er cent of max. load f controls       13       26       39       52       59       65       72       78       91         Specimen       Time duration of sustained load at in minutes         A=1-5       5       5       5       20       225       15         A=1-7       5       5       5       1510       1500       15         A=1-4       5       5       5       5       1510       1500       1500         A=1-3(a)       5       15       25       35       35       1500       1500	kips       4       8       12       16       18       20       22       24       28       20         er cent of max. load from 10 max.	kips       4       8       12       16       18       20       22       24       28       20       16         er cent of max. load from 10 max.	kips       4       8       12       16       18       20       22       24       28       20       16       12         er cent of max. load max. load f controls       13       26       39       52       59       65       72       78       91       65       52       39         f controls       Time duration of sustained load at each load in minutes         A-1-5       5       5       5       20       225       6       7       7       7       8       91       65       5       5       5       5       5       5       5       5       5       5       5       5       5	kips       4       8       12       16       18       20       22       24       28       20       16       12       8         er cent of max. load from 10 max. l	kips       4       8       12       16       18       20       22       24       28       20       16       12       8       4         er cent of max. load f controls       13       26       39       52       59       65       72       78       91       65       52       39       26       13         f controls       Time duration of sustained load at each load level in minutes         A=1-5       15       25       30       25       35       5       5       5       5       5       5       5       5       5       5       5

Table 2 - SCHEDULE OF STEP-BY-STEP LOADING UNLOADING TESTS (INTERMEDIATE CONDITION)

زديب									
Load Level		L 0	A	Ď	I N	G.	UN	LOAD	ING
kips	4	8	10	12	15	16	8	4	0 :
Per cent of max. load of controls	24	48	60	72	89	96	48	24	0
Specimen		Time du	uration	n of s	ustained in mir		at eac	h load	level
M-1-2 M-1-9 M-1-3 M-1-10 M-1-4(a) (b) M-1-7(a) (b)	55555555	65055555 8	<b>3</b> 0 5	60 5 30 5 60 30 5	75 1040 30	10	5550 555	5 10 15 20 10 10	215 145 155 1250 75 150 70

Table 3 - SCHEDULE OF STEP\_BY STEP LOADING UNLOADING TESTS (GREEN CONDITION)

							* · · · · · · · · · · · · · · · · · · ·	
Load Level	L	0 A	D I	N (	}	NU	LOAI	) I N G
kips	4	8	10	12	14	8	4	0
Per cent of max. load of controls	23	46	58	69	81	46	23	0
Specimen	T	ime dura	ation of		ined lo minute		ach load	level
G-1-9 G-1-5 G-1-3 G-1-6 G-1-7(a)	555555	10 5 20 15 25 25	962 20 30 25 25	10 10 10 25	15	5 55 5	10 20 5 15 10	75 1025 90 135 95 150

- (a) 1st. cycle of repetitive loading.(b) 2nd. cycle of repetitive loading.(c) 3rd. cycle of repetitive loading.

				, St	ер	•		2nd	, St	e p		<u>,</u>	3rd	. Ste	g (c	a 2		4th	. Ste	<b>∍</b> p	Providence in the control of the con		5th	. Sto	e p	unicita keluking papanana kendilike kala		6th	. Ste	n	
Spec	imen	Level	Duration of Sustained Stress (minutes)	Longi- tudinal		Tangen- tial	Level	Duration of Sustained Stress (minutes)	Longi- tudinal		tial	Level	Sustained Stress	Longi- tudinal	R E E Radial	Tangen- tial	Level	Duration of Sustained Stress	Longi- tudinal	Radial	Tangen- tial	Level	Duration of Sustained Stress (minutes)	Longi- tudinal	Radial	tial	Level	Duration of Sustained Stress	Longi- F tudinal	REEP Radial	tial
0-1-		2020	5	10			3030	5	10		0	4040	10	30	0,	0	5050	35	90	00	6	6060	65	180	1 2	6	<u></u>			- J	6
A-1-	1	(18%) <b>*</b> 1940	5	20	0		(27%) 2910	5	20	3	  	3880	5	45	6	7	(44%) 4850	20	120		20	(53%)	225	810	7 27	3.55	7070 (62%)	40	240	2	
A-1-		(25%) 1920	5	0	0	0	( 37% <b>)</b> 2880	15	90		7	(49%) 3840	15	85	6	20	(62%) 4800			19		5820 (74%)		:	137	155					
A-1-		(25%) 970	5	0	Õ		(37%) 1940	5	15	0	0	(50%)	15	30	7	20	(62%)	20 ·	175	32	60	5770 (74%)	15	445	95	161	C 7 1.0	3500	700	00	
		(12%) 1940		30			(25%)	30	60	0		(50%) 2910 (38%) 3880 (51%)	25			) . i	3890 (50%)	,	25			4360 (56%)	1510	250	51	77	5340 (69%)	1500	300	89	91
	(b)	(26%)	5	20		0	2910 (38%)	15	50	6	, , O .	(51%)		80	0		1,000		7.40												
	(c)		5	30	0	3		15	45	6	10	a series response	25	70			4860 (64%)	35	140	25	34	10 Table 10				· · · · · · · · · · · · · · · · · · ·					
M-1-		1000	2	50	0		2000	60		-17#	* **	7000	15	40	9	10		15	50	9	13	5830 (77%)	15	270	63	67	6800 (88%)	5	1140	267	336
		(23%)	· · · · · · · · · · · · · ·     · · · · · · · · · · · · · ·     ·	<b>D</b>			(45%) 1900	- OO	150		<b>-</b> 54	3000 (68%)	60	330	-24	<b>-4</b> 5						Author Victory					·				1
M-1-		950 (25%)	<b>2</b>	20	-3	-3	(51%)	20	50	10	13	2850 (76%) 3000 (68%)	2	150	37	38	3560 (95%)	<b>7</b> 5	2320	67	-38	arms A Mi									
M-1-		1000	5	50		3	(45%)	20	80	)		(68%)	30	270	-6	-7	4000 (91%)	10	2035	83	353					a and and a					
M-1	- 110	950 (19%)	<b>5</b>	40	3	0.	1900	5	65		10	2860 (58%) 2500 (60%)	5	80	10	1	3570 (73%)	1015	3870	309	232			3 · · ·							
M-1-	-4(a)	1000 (24%)	5	25	0	3	2000 (48%)	5	<b>6</b> 5	-13	24	(60%)	30	170	-57	74						10 m									
	(a)		5	40	, , , , , , , , , , , , , , , , , , ,	13	3000	5 	50	-3	20	100	5	35	-10	7	300 (72%)	60	760	-64	276										
M I	-7(a)	950 (20%)	5	20	1	0	1900 (40%)	5	25	3	-3	2860 (61%)	30	185	40	18															
	(0)		5	20	0	0		5	50	3	, 6. 		5	55	65	10	3570 (76%)	30	730	71	124										
G-1-		950 (28%)	5	0	0	***	1900 (57%) 2000 (61%)	10	40	-6		2860	10	1190	<b>-</b> 150									-							
G-1-	11(	1000	5	20	-3	-3	(61%)	5	75	-13	-10	2500 (76%)	962	3330	-124	-251															
G-1-	-3	1000 (30%) 1000 (31%)	5	0	0	0	(60%)	20	90	-19	-24	2860 (86%) 2500 (76%) 2500 (75%) 2500 (76%) 2380 (60%)	20	200	-29	-34	3000 (90%) 3000 (92%)	10	1350	-70	-218										
G-1-	-6	1000 (31%)	<b>5</b>	30	0	-10	2000	15	150	-30	-35	(76%)	30	530	-64	-38	3000 (92%)	10	2820	-94	-220						;				
G 1	-7(a)	950 (24%)	5	20	<b>~</b> 6	•	2000 (61%) 1900 (48%)	25	140	_41	0	2380 (60%)	25	120	-38	3															
	(b)		5. 5	30	-6			25	135	-25	0		25	75	-13	-13	2860 (72%)	25	275	-45	-47	3330	,15	1100	-41	-44					

<sup>&</sup>quot;Stress level expressed as percentage of specimen's actual crushing strength. #Negative sign indicates negative creep which is opposite to bulging,

<sup>(</sup>a) Values for 1st. cycle of loading.(b) Values for 2nd. cycle of loading.(c) Values for 3rd. cycle of loading.

				st. S t		entra de la composição	2nd					3rd	St	e p	
		Duration	RE		and the second s	,	Duration		COVE		Stress	Duration		COVE	, R Y
Specimen		of Sustained Stress		Radial	Tangen-:	Level	of Sustained Stress	Longi- tudinal	Radial	Tangen- tial	Level	of Sustained Stress	Longi- tudinal	Radial	Tängen tial
	(psi.)	(minutes)	(micro-	inches pe	er inch)	(psi.)		(micro-	inches po	er inch)	(psi.)	(minutes)	(micro-	inches pe	er inch
0-1-1	5050 (44%)*	5	20	0	0	4040 (35%)	5	0	Ö	0	3030 (27%)	5	0	0	0
A-1-5	3880 (49%)	5	40	0	0	2910 (37%)	5	50	0	7	1940 (25%)	5	50	0	7
A-1-7	3840 (50%)	5	50	13	10	2880 (3&%)		40	13	10	1920 (25%)	5	50	13	17
A-1-4	3890 (50%)	5	30	0	10	2910 (38%)	5	15	0,	3	1940	5	25	0	7
A-1-3(a)	1940 (26%)	25	0	0	0	970 (13%)	30	0	0	0	(0%)	135	100	3	13
(b)	2910	5	20	3	10	1940 (26%)	15	10	6	6	970 (13%)	35	25	10	10
(c)	4860 (64%)	5	25	6	10	3880 (51%)	5	30	-3	7	2910 (38%)	5	50	3	17
M-1-2	2000	5	20	, 0	6	1000	5	30	3	3	0 (0%)	215	170	27	32
M-1-9	1900	5	170	20	44	950 (25%)	10	400	17	80	(0%)	145	700	0 ;	-16 <sup>#</sup>
M-1-3	2000	5	195	0	6	1000	15	400	10	10	(0%)	155	1050	27	32
M-1-10	1900	10	185	<b>~</b> 3	57.	950 (19%)	20	530	34	105	(0%)	1250	1520	60	64
M-1-4(a)	1000	10	50	0	3	0 (0%)	75	20	0	3				as your as country	
(b)	2000	5	25	0	7	1000	10	140	-6	44	(0%)	150	390	-12	84
M-1-7(a)	1900	5	15	0	0	950 (20%)	10	25	13	3	(0%)	70	25	17	6
(ˈbː)	1900	5	120	3	6	950 (20%)	10	180	10	25	(0%)	115	165	37	54
G-1-9	1900	5	55	16	-	950 (28%)	10	260	13	•	(0%)	<b>7</b> 5	295	<b>-</b> 29	
G~1-5	1000	20	350	-47	10	(0%)	1025	900	-232	-22	* - /- /- /- /- /- /- /- /- /- /- /- /- /			-	
G-1-3	2000	5	55	0	12	1000 (30%)	5	200	-9	10	0 (0%)	85	600	<b>-</b> 29	3
G-1-6	2000	5	95	-7	0	1000	15	450	-40	-10	0	115	540	-67	-29
3-1-7(a)	950 (24%)	10~	50	-12	10	0 (0%)	95	10	-3	3	(0%)				
(b)	1900	5	100	-22	17	950 (24%)	10	350	-32	17	0 (0%)	150	430	-38	34

<sup>\*</sup>Stress level expressed as percentage of specimen's actual crushing strength.

										5 7 8 loc's see	The second second		*		1,
,													. ,		
TEP	MA	NNER	· A.T:		CESS	I V/E L			STRE	SS	LEV	ELS		!	
		4th		ер			5th						.Ste		
K Y	Stress	Duration	A STATE OF THE PARTY OF THE PAR	COVE			Duration		COVE		Stress				RY
Tängen-	Level	of		Radial		Level	of		Radial	Tangen-	Level	of		Radial	Tangen-
tial		Sustained	tudinal		tial		Sustained	tudinal		tial		Sustained	tudinal		tial
n inah	,	Stress				: /	Stress	/ 4 a a -			(mad )	Stress	1-1	<b> </b>	
r inch)	(psi.)	(minutes)	(micro-	inches p	er inch)	(psi,)	(minutes)	(micro-	inches p	er inch)	(bar.)	(minutes)	(micro-	inches p	er inch)
0	2020	5	10	0	0	1010	10	20	0	0	0	230	10	0	, and O
	(18%)		10			( 9%)	1				(0%)				
7	970	20	100	10	30	(9%)	330	205	41	44					
	(12%)					(0%)									
17	960	15	75	19	30	0.1	150	175	25	17					
	970 (12%) 960 (12%)					(0%)									
7	970	15	60	0	10	0	3095	180	10	60					
13	(12%)					(0%)									· · · · · · · · · · · · · · · · · · ·
1)	-		,												· · ·
10	0	195	45	6	7				, (1)						
	(0%)	400	'2												
17	(0%) 1940	5	80	0	23	970 (13%)	10	165	0 ' '	54	0	135	245	6	81
	(26%)	· · · · · ·				(13%)					(0%)				
32		ı.				•				ł ,	L	II		I	
- c#															
-16 <sup>#</sup>															
32															
26															
64															
								3							
84															

<sup>#</sup>Negative sign indicates negative recovery(bulging)
(a) Values for 1st. cycle of unloading.
(b) Values for 2nd. cycle of unloading.
(c) Values for 3rd. cycle of unloading.

Table 6 - MODULUS OF ELASTICITY AND COEFFICIENTS, OF LATERAL DEFORMATION DURING LOADING IN STEP-BY-STEP MANNER AT

SUCCESSIVELY HIGHER STRESS LEVELS

<b>X</b>	Modulus		S t e	p of La	terel	Modulus	والتناز بكفة فينفضون ويوري	S t e	p of Lat	eral	Modulus		S t e icient		erel	Modulus		S t e	p f Late:	ra l	Modulus	CONTRACTOR CONTRACTOR	S t e	p f Later	ദി	Modulus	6th.	S t o	e p of Lat	teral
Specimen	of Elasticity	D	eforms	tion .	cing	of Elasticity	De	forma	tion		of Elasticity	De	eformat	ion	ing	of Elasticity	Def	'ormati			of Elasticity	De	formating	tion	ring	of Elasticity	De	formating	tion	ring
	mias of of og		ding	Cre		ELES OF CLOY		d <b>i</b> ng	Cre		mras orer uy		ading		ep	E183 01010y	Load		Cree		LIAS UICTO	Loa	ding	Gre		Brasticity		ding	Cre	
	(1000 psi.)	$\mathcal{U}_{R}$	$\mathcal{M}_{T}$	$\mathcal{U}_{R}$	$M_{T}$	(1000 psi)	MR	$\mathcal{U}_{\tau}$	MR	$u_{\tau}$	(1000 psi)	$\mathcal{M}_{R}$	Mr	$M_R$	MT	(1000 psi)	U <sub>R</sub>	$\mathcal{U}_{\tau}$	MR	$\mathcal{M}_{T}$	(1000 psi)	$\mathcal{M}_{\mathcal{R}}$	Mr	$M_R$	$M_{T}$	(1000 psi)	$M_{R}$	MT	MA	MT
0-1-1	1800	0.232	0.406	0.000	0.000	1800	0.211	0.486	0.000	0.000	1740	0.186	0.490	0.000	0.000	1710	0.176	0.448	0.000	0.067	1710	0.229	0.458	0.017	0.033	1710	0.285	0.442	0.012	0.025
A-1-5	2110	.175	. 305	0	0	1980	.196	• 339	.150	.150	1940	.236	.362	.133	.156	1870	.227	.394	.158	.167	1850	.278	.410	.169	.191					
A-1-7	1690	. 349	.448	0	0	1780	. 359	.481	0	.078	1720	.352	.486	.071	.236	1720	.363	.493	.183	.343	1720	.432	.544	.214	.362			· ·		
A-1-4	2060	.528	. 336	0 1	0	2060	. 344	. 343	0	0	2020	.346	.369	.100	.100	2020	.346	. 392	.120	.280	1940	.392	.408	.204	.309	1870	.392	.427	.297	.304
A-1-3(a)	2200	•	•995	to the second of	0	1940	.183	.456	0	0	1870	.248	.485	0	0															
(b)	5050		.819	1	0	1940	.178	.450	.120	.140	1940	.236	.450	.086	.100	1940	.248	.456	.179	.243										
(0)	1900	.208	.600	0	.100	1870	.208	.423	.133	.222	1870	.238	.433	.225	.250	1870	.264	.423	.180	.260	1830	.258	.449	.234	.252	1830	.294	.564	.234	.295
M-1-2	1750	.530	.625	0	0	1670	.479	•525	113	360	1540	.415	.470	073	136															
M-1-9	1790	.444	.912	150	150	1500	. 330	.661	.200	.260	1460	.331	.626	.246	.254	1400	.310	•592	.029	016										
M-1-3	1600	.391	.444	0	.060	1480	.256	.388	.038	.082	1490	.253	.387	022	022	1470	.215	.328	.041	.179										
M-1-10	1900	.539	.808	.075	.150	1830	.531	.635	.108	.154	1800	.526	.623	.125	.162	1800	.505	.525	.080	.060										
M-14(a)	1790	.573	.475	0	.120	1610	.306	.460	200	. 369	1560	.298	.460	336	.435														,	
(b)	1660	.450	.465	0	. 325	1600	.292	.439	060	.400	1600	.250	.431	286	200	1600	.293	.446	084	.364		,						* **		
M-1-7(a)	1980	.775	.665	0	0,	1670	.475	.491	.120	120	1670	.466	.482	.216	.097		and and finding													
(b)	1980	.742	.621	0	0	1640	.466	.480	.060	.120	1610	.434	.463	.109	.182	1520	.451	.475	.097	.170		7 T	. 6						,	
G-1-9	1420	.461	<b>1990</b>	0		1400	.342		150	thear .	1240	.252	Botto	126											* *					
G-1-5	1560	.275	.710	150	150	1330	.234	.482	173	133	1390	.158	.450	037	076															
G-1-3	1850	. 338	.618	0	0	1670	.282	.598	211	267	1470	.309	.494	145	170	1430	.237	.444	052	161						·	i I			
G-1-6	1820	.360	.828	0	333	1540	.228	.440	200	233	1350	.227	•354	121	072	1280	.208	• 344	033	078					·	,				
G-1-7(a)	2210	.736	1. 1	300	***	1640	.405	•540	293	0	1590	. 347	•533	316	.025								,			, .				
(b)	1900	.621	1 1 1 1 1 1 1	200		1730	.382	.526	185	0	1700	-354	.504	173	173	1700	.307	.493	164	171	1700	.260	.470	037	040		,			

<sup>(</sup>a) Values for 1st. cycle of loading.(b) Values for 2nd. cycle of loading.(c) Values for 3rd.cycle of loading.

Table 7 - MODULUS OF ELASTICITY AND COEFFICIENTS

# OF LATERAL DEFORMATION DURING UNLOADING IN STEP-BY-STEP MANNER AT

SUCCESSIVELY LOWER STRESS LEVELS

			S t e					i t e p					S te				4th.	S t e				5th.	Ste	р			6th.	S t e	0 €	
	Modulus of	Coeffi De	cient format		ceral	Modulus of	Coeff:	lcient eformat		eral	Modulus of		icient eformat		teral	Modulus of		.cient format	of Lat :ion	ceral	Modulus of		icient eforma	of La tion	teral	Modulus of		ficient Deforma		teral
Specimen	Elasticity	Duri Unloa			ring overy	Elasticity	Dur: Unlo		Dur Recov		Elasticity		ring oading:	Dui Reco	ring very	Elasticity	Dur Unlos	ing ding	Dur Recov		Elasticity	Dur	ring ading	Du	ring overy	Elasticity	D	uring oading	Du	ring
	(1000 ps <b>i.</b> )	$M_R$	MT	$\mathcal{M}_{R}$	MT	(1000 ps <b>i</b> )	MR	Mr	MR	Mr	(1000 psi.)	MR	MT	MR	MT	(1000 psi.)	MR	MT	MR	MT	(1000 psi.)	MR	MT	MR	Mr	(1000 psi.)	MR	MT	MA	My
0-1-1	1800	0.264	0.464	0.000	0.000	1710	0.188	0.443	0.000	0.000	1740	0.169	0.531	0.000	0.000	1770	0.177	0.563	0.000	0.000;	1740	0.243	0.504	0.000	0.000	1660	0.430	0.430	0.000	0.000
A-1-5	1580	.163	.302	0	0	1900	.194	. 369	0	.140	1900	.194	.349	0	.140	1900	.186	.363	.100	. 300	1900	.318	.342	.164	.176					
A-1-7	1780	.291	. 384	.260	.200	1850	.405	•575	. 325	.250	1750	.376	.500	.260	. 340	1850	.410	.536	.253	.400	1750	.420	.500	.143	.097					
A-1-4	1910	. 328	.382	0	• 333	1900	.320	.341	.0	.200	1870	.343	.356	0	.280	1870	.330	.386	.0	.167	2060	.380	.392	.056	•333					
A-1-3(a)	1880	.292	.745	0	0	1940	.241	.450	0	0	2200	.241	.462	.030	.130								3							
(b)	1800	.301	.652	.150	.500	1830	.621	.650	.600	.600	1870	.560	.432	.400	.400	2040	.234	.445	.134	.155										
(c)	1800	.280	.602	.240	. 10	1800	.264	.429	.100	.233	1830	.241	.423	.060	.294	1760	.240	.416	0	.288	1800	.251	.508	0	.327	1800	.282	.542	.024	.331
M-1-2	1540	.406	.542	0	.300	1540	.384	.482	.100	.100	1600	.362	.461	.159	.188															
M-1-9	1240	.314	.536	.118	.259	1190	.297	.505	.042	.200	ficaces*	.142	.244	0	.023															
M-1-3	1350	.382	.424	0	.038	1370	.242	. 368	.025	.025	1250	.234	.360	.026	.030															
M-1-10	1590	.480	•532	016	.308	1490	.494	.646	.064	.198	* 1	• 354	.382	.046	.049							"								
M-1-4(a)	1600	.424	.461	0	.260	1600	.300	.451	0	.150										. 4 w										
(b)	1610	.430	.452	0 1	.280	1410	.341	. 392	043	.314	1450	.240		031	.216		,				,					A A A A A A A A A A A A A A A A A A A				
M-1-7(a)	1640	496	.521	0	0	1590	.424	.460	.520	.120		.388			.240					* * *	, ,	eng in			;					
(b)	1490	.490	.502	.025	.050	1440	.476	.492	.056	.139	1540		.475												:			;	5	
G-1-9	1150	.386		.291	**************************************	1270	. 334	(Copus	.050	Noner	-	.240		098	•••			1.5				*-								
G-1-5	1130	.211	.330	134	.028	•••	.041	.218	258	.024														ļ						· · · · · · · · · · · · · · · · · · ·
G-1-3	1390	.278	.401	0	.218	1370	.250	.416	045	.050	<b></b>	.101		129				*					,				,			,
G-1-6	1220	.182	,	074		1110	.135		089			.101	.110	124	054							,	<i>:</i>							
G-1-7(a)		.521		240			.400		300																					
(b)	1360	.489	.560	220	.170	1560	.440	.520	092	.049	<b>G</b>	. 346	.482	088	.079															

<sup>(</sup>a) Values for 1st. cycle of unloading.(b) Values for 2nd. cycle of unloading.(c) Values for 3rd. cycle of unloading.

Table 8- VALUES FROM FINAL STATIC TESTS IN COMPRESSION PARALLEL TO GRAIN FOLLOWING CREEP\_RECOVERY TESTS

Specimen	Moisture Content	Specific Gravity	Max. Stress	Max. Strain	Modulus of Elasti-	2	son's tio
ppecimen.	(%)		(psi.)	(micro- in An.)		MLR	$\mathcal{M}_{LT}$
0-1-1 CA-1-6 A-1-5 A-1-7	oven-dry 9.6 9.6 10.2	0.472 0.497 0.504 0.541	11,400 7300 7860 7760	12200 6000 7300 11050	1890	0.372	0.402
A-1-4 A-1-3 CM-1-1* M-1-2 M-1-9	9.3 9.6 19.9 20.8 20.3	0.492 0.492 0.491 0.491 0.518	7750 7600 4190 4400 3720	6700 6200 4400 4800 4720	1610	0.210	0.596
M-1-3 M-1-10 M-1-4 CG-1-10 <sup>©</sup> G-1-9 G-1-5 G-1-3	21.8 20.0 20.0 65.5 460.5 65.5	0.490 0.541 0.500 0.547 0.545 0.498 0.504	4400 4850 4180 4100 3290 3280 3320	5500 7900 4650 3850 4300 6240 4250	2000	0.342	0.651
G-1-6 G-1-7	62 <b>.1</b> 57.4	0.499 0.547	3250 3900	6600 4850			

Control specimens were not subjected to creep-recovery tests.

Table 9- PERMANENT SET AND STRAIN RECOVERY OF DOUGLAS-FIR-TESTED FOR CREEP

	Per	manent	Set	Strain Recovery
Specimen		Tangential	Longitudinal	Percentage of Axial Creep
0-1-1 A-1-5 A-1-4 A-1-3(a) (b) (c) M-1-2 M-1-9 M-1-3 M-1-10 M-1-4(a) (b) M-1-7(a) (b) G-1-5 G-1-5 G-1-5 G-1-6 G-1-7(a) (b)	0 48 48 48 48 48 48 49 27 138 232 49 49 49 49 49 49 49 49 49 49 49 49 49	17 77 91 * 10 232 * 127 29 40 -19 * -19 -19 -19 -10 -10 -10 -10 -10 -10 -10 -10 -10 -10	270 370 370 270 0 40 750 490 1440 100 160 50 80 220 800 175 460 180	51.8 70.1 50.3 56.0 87.5 77.9 64.6 81.7 90.1 89.8 89.8 89.8 89.8

<sup>\*</sup> No values recorded due to accidental disturbance of the lateral deformation apparatus. (a) Values of 1st. cycle. (b) Values of 2nd. cycle. (c) Values of 3rd. cycle.

## Note:

Pages 46-67 oversized and are contained in accompanying see fly leaf.

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